DATA-DRIVEN IDENTIFICATION

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Find $(\mathbf{u}, \boldsymbol{\sigma})$ such that

Admissibility for u

 $\boldsymbol{u}=\boldsymbol{u}_{\text{d}}$ on Γ_{u} + regularity

 $\operatorname{div}(\boldsymbol{\sigma})=0$

Compatibility

$$\boldsymbol{\varepsilon} = \frac{1}{2}(\nabla \mathbf{u} + \nabla^{\mathsf{T}}\mathbf{u})$$

Balance of external forces

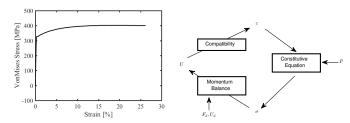
Balance of momentum

$${\pmb \sigma}({\sf n}) = {\sf F}_d \ {\sf on} \ {\sf \Gamma}_{\it F}$$

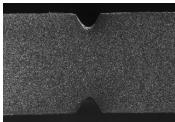
Regularization

Sample with a geometry such that the solution is unique. Usual specimen geometry for uni-, bi-, tri-axial testing,...

▶ "Engineering" approach

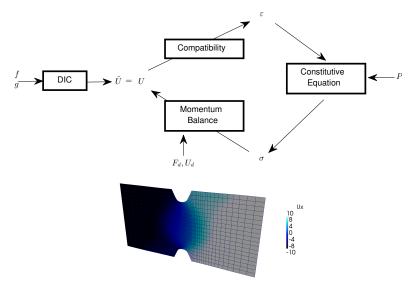


► Validation ?

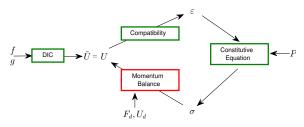


by G. Portemont at ONERA Lille

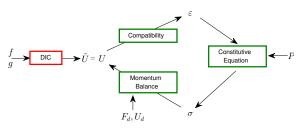
► Photomechanics



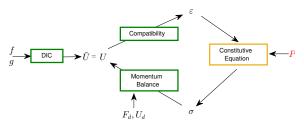
► Stress calculation



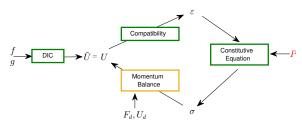
▶ Numerical simulation



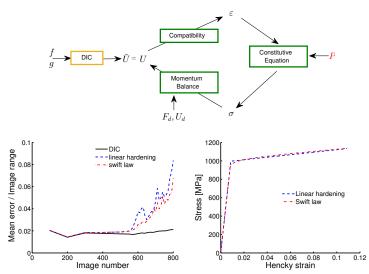
► Constitutive Equation Gap [Chrysochoos et al.]



▶ Equilibrium gap [Claire et al., 2004], VFM [Grédiac et al., 2006]

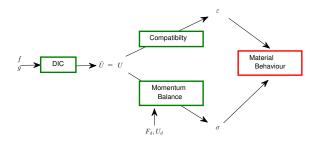


▶ FEMU [Lecompte et al., 2007, Leclerc et al., 2009]...



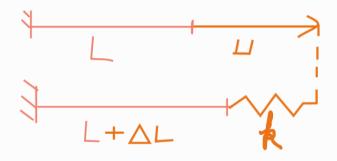
PARAMETRIC V.S. NON-PARAMETRIC

- ▶ Parametric techniques (using a constitutive equation)
 - + provide for the optimal set of parameters
- +/- tell that the constitutive equation is not correct
 - how to improve it
 - kinematically consistent direct FEA
- ▶ Non-parametric (without using a constitutive equation)



SIMPLE PROBLEM

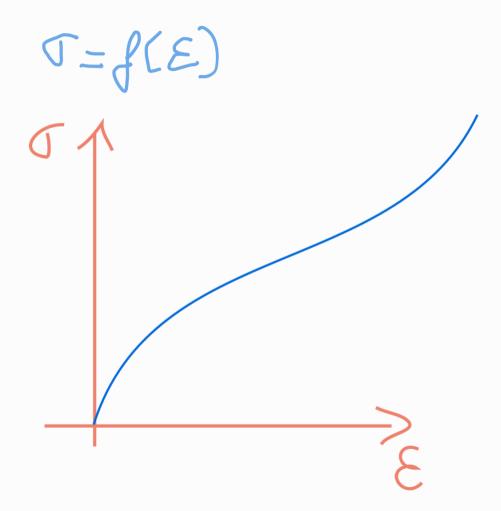
Problem definition



Compatibility

Balance of momentum

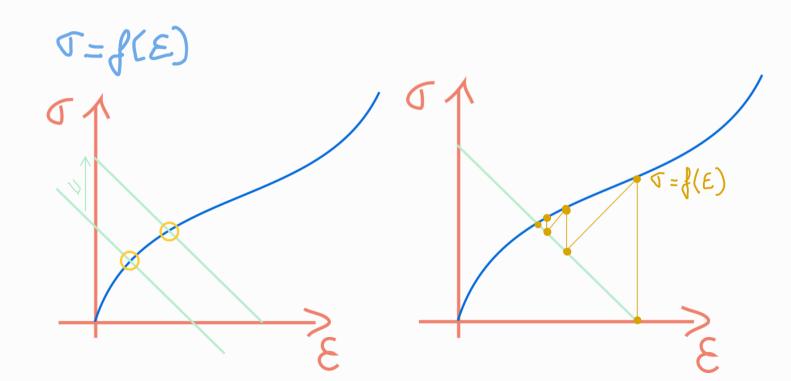
CONSTITUTIVE LAW



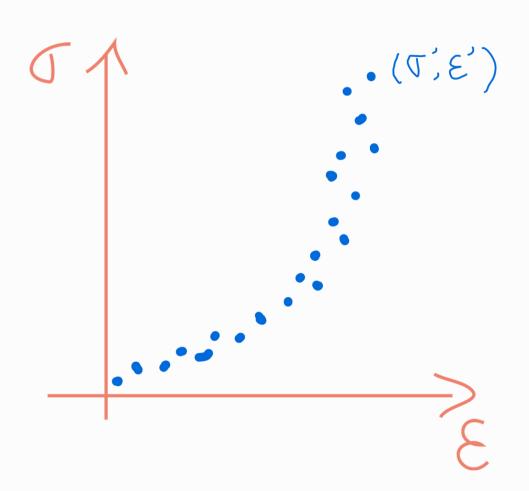
RESOLUTION

Expected solution

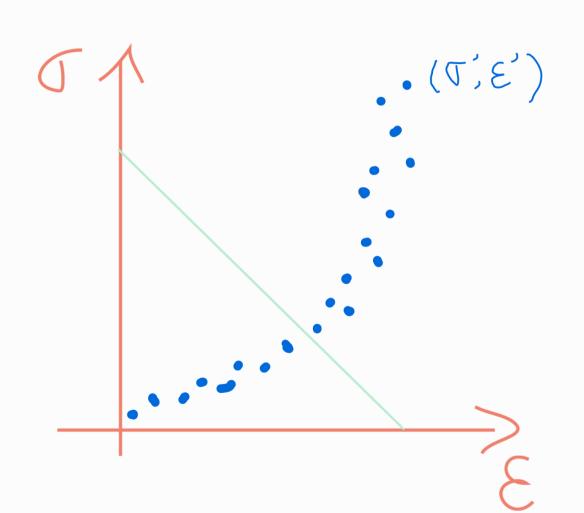
Non linear problem resolution



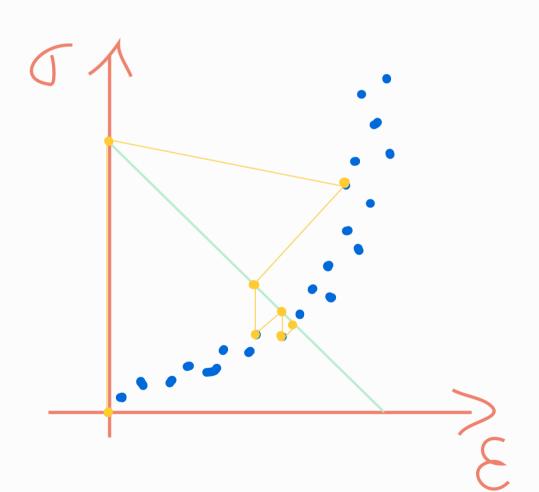
CONSTITUTIVE DATA



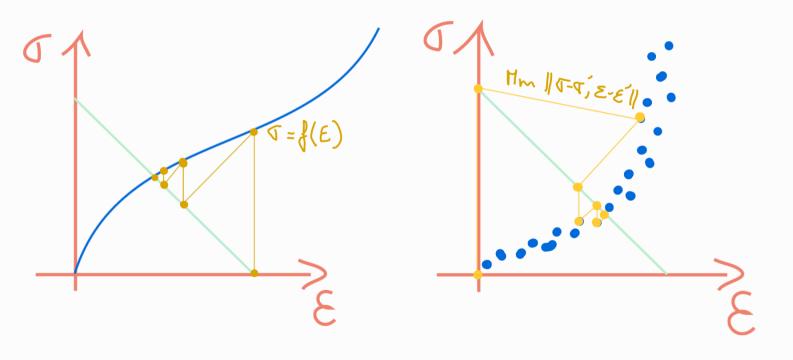
DATA-DRIVEN COMPUTATIONAL MECHANICS



DATA-DRIVEN SOLVER



INTRISINC DIFFERENCE



Find $(\mathbf{u}, \boldsymbol{\sigma})$ such that:

Admissibility for u

 $\mathbf{u} = \mathbf{u}_d$ on Γ_u + regularity

$div(\boldsymbol{\sigma}) = 0$

Compatibility

$$\boldsymbol{\varepsilon} = \frac{1}{2}(\nabla \mathbf{u} + \nabla^{\mathsf{T}}\mathbf{u})$$

Balance of external forces

Balance of momentum

$$\sigma(\mathsf{n}) = \mathsf{F}_d \; \mathsf{on} \; \mathsf{\Gamma}_{\mathit{F}}$$

Regularization

Sample with a geometry such that the solution is unique. Usual specimen geometry for uni-, bi-, tri-axial testing,...

CONTINUUM MECHANICS

Find $(\mathbf{u}, \boldsymbol{\sigma})$ such that:

Admissibility for u

 $\mathbf{u} = \mathbf{u}_d$ on Γ_u + regularity

Compatibility

$$\boldsymbol{\varepsilon} = \frac{1}{2}(\nabla \mathbf{u} + \nabla^{\mathsf{T}}\mathbf{u})$$

Balance of momentum

$$div(\boldsymbol{\sigma}) = 0$$

Balance of external forces

$$\sigma(\mathsf{n}) = \mathsf{F}_d \; \mathsf{on} \; \mathsf{\Gamma}_{\mathit{F}}$$

Regularization

Constitutive relation

$$\sigma = f(\varepsilon, ...)$$

NON-PARAMETRIC IDENTIFICATION

From a measured **u**, find (σ) such that:

Admissibility for u

 $\mathbf{u} = \mathbf{u}_d$ on Γ_u + regularity

Balance of momentum

 $\operatorname{div}(\boldsymbol{\sigma})=0$

Compatibility

 $\boldsymbol{\varepsilon} = \frac{1}{2} (\nabla \mathbf{u} + \nabla^{\mathsf{T}} \mathbf{u})$

Balance of external forces

 ${\pmb \sigma}({\sf n}) = {\sf F}_{d} \; {\sf on} \; {\sf \Gamma}_{\it F}$

Regularization

Minimizing the deviation (energy norm in the phase space) from the mean response (ϵ^*, σ^*) .



DDI implementation

October 9, 2025

1 How DDI Works?

Authors: Outlaw Group, Centrale Nantes, France

[Leygue et al., Data-based derivation of material response, CMAME 2018]

1.1 Basics and an illustrative example

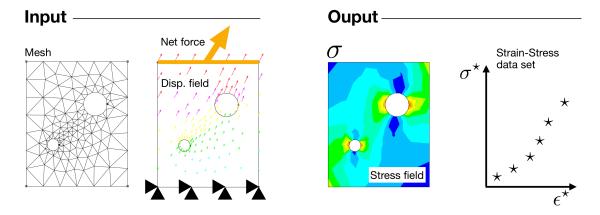
1.1.1 DDI at a glance

Data Driven Identification (DDI) aims at computing stresses on a structure where displacement field (u) and applied forces (f) are known but not the mechanical behaviour law. DDI gives us, first, the **balanced stress field** (σ) in the structure (stress tensor in each element) and second, a **Strain-Stress data set** $(\varepsilon^{\star}, \sigma^{\star})$. The Strain-Stress data set is needed to solve the ill-posed problem of equilibrium in the absense of a material law, its size, (N_{ss}) , is smaller than the number of elements (N_e) of the structure. This data set can also be seen as an attractor allowing the minimisation of the variance of the stress field.

Input * 2D mesh of a structure * Node displacement field u of size N_e - Typically from DIC * Applied net forces - Typically from a load cell

Output* Stress field (σ) - preserving balance of momentum * Clusturized Strain-Stress data base $(\varepsilon^\star,\sigma^\star)$ of size N_{ss}

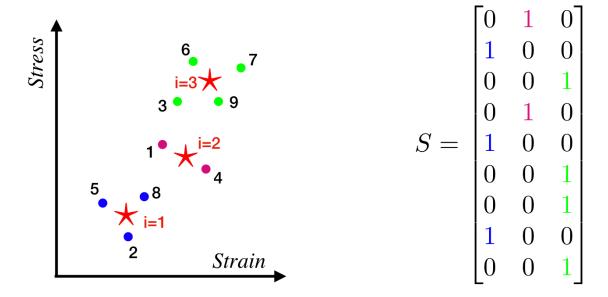
Parameters 1. r: clustering ratio [no unit] $\left(r = \frac{N_e}{N_{ss}}\right)$ 2. \mathcal{C}_0 : Algorithmic stiffness [Pa]



1.2 Definitions, Problem & Equations to be solved

1.2.1 1. Definitions:

- The Clustering is the pairing between each couple $(\varepsilon, \sigma)_e$ of an element e with a couple i of the data-set $(\varepsilon^*, \sigma^*)_i$. $(\varepsilon^*, \sigma^*)_i$ is a weighted average of a subset of (ε, σ) . Obviously, $N_e > N_{ss}$.
- Clusterization is defined by the pairing operator S such as $\sigma_e^* = P_{ie}\sigma_i^*$. S is an unknown of the problem.
- Illustration of S:



• \mathcal{C}_0 **norm** is define on a couple (\mathbf{a}, \mathbf{b}) of tensors of order 2 (typically a strain and a stress) by the definition :

 $\|\mathbf{a},\mathbf{b}\|_{\mathcal{C}_0} = \left[\mathbf{a}:\mathcal{C}_0:\mathbf{a}+\mathbf{b}:\mathcal{C}_0^{-1}:\mathbf{b}\right]^{\frac{1}{2}}, \quad \mathcal{C}_0 \text{ a 4th order symmetric positive definite tensor.}$

1.2.2 2. Problem:

 ε and f being known, find $(\sigma, \varepsilon^{\star}, \sigma^{\star}, S)$ minimizing the \mathcal{C}_0 norm between (ε, σ) and $(\varepsilon^{\star}, \sigma^{\star})$ and which preserve the equilibrium of the structure.

$$\|\varepsilon - S\varepsilon^\star, \sigma - S\sigma^\star\|_{\mathcal{C}_0} = \left[(\varepsilon - S\varepsilon^\star) : \mathcal{C}_0 : (\varepsilon - S\varepsilon^\star) + (\sigma - S\sigma^\star) : \mathcal{C}_0^{-1} : (\sigma - S\sigma^\star) \right]^{\frac{1}{2}},$$

and $\operatorname{div} \sigma = f$, $\forall M \in \text{Structure}$

 \mathcal{C}_0 has the dimension of a stiffness tensor.

That can be rewritten in a variational form:

Find, $(\sigma, \varepsilon^*, \sigma^*, \eta)$ that make the following functional stationary, for a given S

$$\mathcal{E}(\sigma, \boldsymbol{\varepsilon^{\star}}, \boldsymbol{\sigma^{\star}}, \boldsymbol{S}, \boldsymbol{\eta}) = \frac{1}{2} \int_{V} \left[\| \boldsymbol{\varepsilon} - \boldsymbol{S} \boldsymbol{\varepsilon^{\star}}, \sigma - \boldsymbol{S} \boldsymbol{\sigma^{\star}} \|_{\mathcal{C}_{0}}^{2} - \left(\operatorname{div} \sigma - f \right) \boldsymbol{\eta} \right] dV$$

With η a Lagrange Multiplier (dimension of a displacement vector in [m]).

1.2.3 3. Derivation of variational equation:

The derivation of the above functional = 0 gives us the following equations needed to find the unknowns $(\sigma, \varepsilon^*, \sigma^*, \eta)$:

$$\begin{array}{llll} \delta \varepsilon^{\star} & \quad \Rightarrow & \int_{V} \mathcal{C}_{0} : \left(\varepsilon - S \varepsilon^{\star} \right) dV & = 0 & \forall V & (1) \\ \delta \sigma^{\star} & \quad \Rightarrow & \int_{V} \mathcal{C}_{0}^{-1} : \left(\sigma - S \sigma^{\star} \right) dV & = 0 & \forall V & (2) \\ \delta \eta & \quad \Rightarrow & \int_{V} \left(\operatorname{div} \sigma - f \right) dV & = 0 & \forall V & (3) \\ \delta \sigma & \quad \Rightarrow & \left(\sigma - S \sigma^{\star} \right) = \mathcal{C}_{0} : \operatorname{grad}^{s}(\eta) & \forall M & (4) \end{array}$$

- Equations 1 and 2 state that ε^* (resp. σ^*) is a weighted average of a cluster of ε (resp. σ)
- Equation 3 is the standard balance of momentum equation on the whole structure.
- Equation 4 states the distance between σ and σ^* is proportional to the gradient of \$ grad^s()\$ (a kind of strain)

1.3 Discrete format

In 1D using bar elements (instead 2D elements): > - (ε, σ) become scalar (instead of tensors) > - \mathcal{C}_0 is just a scalar C_0 > - Only one snapshot is considered > - the bar elements are assigned a unit heigh > - the unit for length is pixel

The variational form can be rewritten as:

$$\begin{split} \mathcal{E}(S_e, \boldsymbol{E^{\star}}, \boldsymbol{S^{\star}}, \boldsymbol{S}, \boldsymbol{L}) = & \frac{1}{2} C_0 \left[E_e - \boldsymbol{S} \ \boldsymbol{E^{\star}} \right]^T \ W \ \left[E_e - \boldsymbol{S} \ \boldsymbol{E^{\star}} \right] \\ + \frac{1}{2} C_0^{-1} \left[S_e - \boldsymbol{S} \ \boldsymbol{S^{\star}} \right]^T \ W \ \left[S_e - \boldsymbol{S} \ \boldsymbol{S^{\star}} \right] \\ - \left[\boldsymbol{B}^T \ W \ \boldsymbol{S_e} - \boldsymbol{F_{ext}} \right]^T \ \boldsymbol{L} \end{split}$$

where B is such that $E_e = B U$, U being a vector collecting the nodal displacements (data), W is a diagonal matrix collecting the surface of each bar element (E_e, S_e) are vectors collecting the strain and stress in each element, (S^*, S^*) are vectors containing the values of $(\varepsilon^*, \sigma^*)$ and L a vector collecting the value of nodal Lagrange multipliers. The stationarity conditions recast as:

$$\delta E^{\star} \qquad \Rightarrow (S^T \ S)E^{\star} = S^T \ E_e \tag{I}$$

$$\delta S^{\star} \qquad \qquad \Rightarrow \quad (S^T \; S) S^{\star} = S^T \; S_e \qquad \qquad (II)$$

$$\delta L \qquad \qquad \Rightarrow \quad B^T \ W \ S_e = F_{ext} \qquad \qquad (III)$$

$$\delta S_e \qquad \qquad \Rightarrow \quad C_0^{-1} W \left(S_e - S \ S^\star \right) + W \ B \ L = 0 \tag{IV}$$

In practice the full F_{ext} vector is not known. Only one component of the resulting force applied on one side of the sample can be measured. A two matrices D and D_c are introduced.

$$D = B^T W$$
 $D_c = \Lambda B^T W$

where Λ is such that: >- the lines of D corresponding to vanishing nodal forces are kept, >- the lines of D corresponding to non-vanishing nodal forces are removed, >- lines corresponding to measuring resulting force (sum of one component of nodal forces along one edge e.g.) are added. The external force vector is modified accordingly and named \bar{F}_{ext} .

The first two systems of equations (I, II):

$$(S^T S)E^* = S^T E_e \tag{2}$$

$$(S^T \ S)S^{\star} = S^T \ S_e \tag{3}$$

(4)

(1)

are easily solved independently as $(S^T S)$ is diagonal. In the following they will be solved together when updating S the selection matrix obtained from the labelling of a k-means algorithm.

The two next lines (III, IV) are included in the following linear system

$$\begin{bmatrix} C_0^{-1}W & D_c^T \\ D_c & 0 \end{bmatrix} \begin{bmatrix} S_e \\ L \end{bmatrix} = \begin{bmatrix} C_0^{-1}W & S & S^* \\ \bar{F}_{ext} \end{bmatrix}$$

The resolution is performed by computing first the Lagrange multipliers L:

$$(Dc~W^{-1}~Dc^T)L = Dc~S~S^{\star} - \bar{F}_{ext}$$

and then the stress:

$$S_e = S \ S^{\star} - C_0 \ W^{-1} \ D_c^T \ L$$

The following algorithm is used latter on >- #### While > - Update S using k-means > - Solve $(S^T\ S)E^\star = S^T\ E_e$ > - Solve $(S^T\ S)S^\star = S^T\ S_e$ > - Solve $(Dc\ W^{-1}\ Dc^T)L = Dc\ S\ S^\star - \bar{F}_{ext}$ > - Update \$\$S_e=\$S_c-c_0_w-(-1)^D_c^TL\$\$

Note that the three steps could be performed in a single one but, usually k-means function use a L_2 norm for evaluating the distance between clusters and samples. Here the norm for this distance is $\|\mathbf{a}, \mathbf{b}\|_{C_0}$. K-means is thus called with $(E_e \sqrt{C_0}, S_e / \sqrt{C_0})$ instead of (E_e, S_e) as input. The coordinates of the clusters (E^\star, S^\star) are thus not a direct output of kmeams. Their are udpated by solving systems (I, II) suing the new S.

1.4 Numerical implementation

```
[1]: # Import Libraries
  import numpy as np
  import matplotlib.pyplot as plt
  import scipy
  from scipy import ndimage
  from scipy.sparse import csr_matrix as smatrix
  import scipy.sparse.linalg as splinalg
  import h5py
  import scipy.io as sio
  from sklearn.cluster import KMeans
  import fem
```

1.4.1 Application parameters

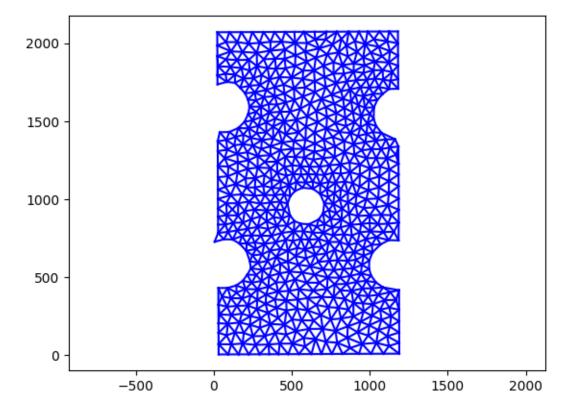
```
[2]: # Units m->pixel=m/pix2m
# Pa=kg/m/s^2->kg/pixel/s^2=Pa*pix2m
# N = kg.m/s^2->kg.pixel/s^2=N/pix2m
# N/m = kg/s^2->kg/s^2=N/m
stdu=0.1 # noise level displacement in pixel
stdf=.0 # N noise level on force
pix2m=25.e-6; # pixel to m conversion
thickness=3e-3 # specimen thicness in m
```

1.4.2 DDI parameters

```
[3]: Ns=50 # number of material states
     inp='dic-coarse.res'# input file
     (X,conn)=fem.readDICmesh(inp)# loading the mesh
     model=fem.FEModel() # instantiating a model
     model.X=X # Nodes
     model.conn=conn # Connectivity
     Nnodes=X.shape[0]
     Nelems=conn.shape[0]
     model.Assemble() # Assembly
     W=model.W # Weighting matrix
     B=model.B # B matrix for computing strain from displacement
     npz=np.load('fem-from-dic.npz')
     U=npz['U'] # Input displacement from FE simulation
     Fres=(npz['Fres']+stdf)/thickness # Input force from FE simulation
     Sref=npz['Sref']*pix2m # Stress field from the FE simulation used as imput data
     E_e=B.dot(U) # Input strain
     Eref=E_e # considered as the reference
     dE_e=B.dot(stdu*np.random.randn(U.size))# noise
     E_e=Eref+dE_e # strain to be considered as imput for DDI
```

```
# Setting the algorithmic stiffness
L=max(X[:,1])-min(X[:,1])
dL=np.max(U[Nnodes::])-np.min(U[Nnodes::])
section=max(X[:,0])-min(X[:,0])
Co=(Fres/section)/(dL/L)*pix2m

# Display the sample
plt.plot(X[conn,0].T,X[conn,1].T,'b-');
plt.axis('equal');
```



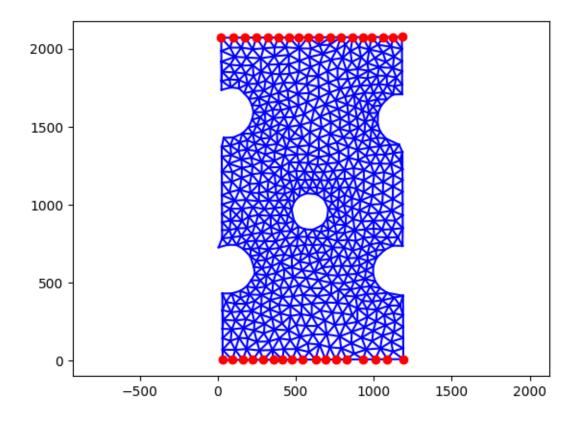
1.4.3 Boundary conditions

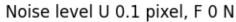
The internal force vector is 0 everywhere except: >- on the bottom where the distribution of its x component is unknown >- on the top where the distribution of its x component is unknown >- on the top where the distribution of its x component is unknown >- on the top where the sum of the distribution of its y component equals the measured load

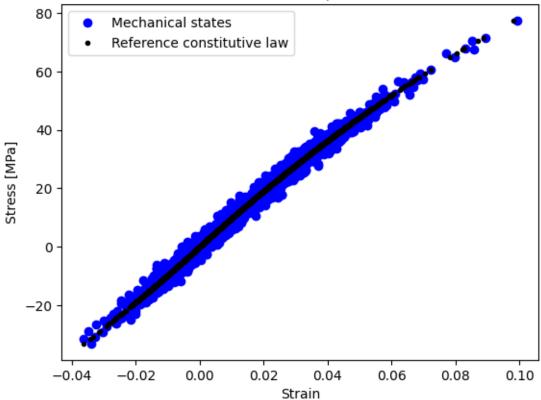
```
[4]: top=X[:,1]>max(X[:,1])*0.99
bot=X[:,1]<max(X[:,1])*0.01

nodes_index=np.arange(Nnodes)
top_nodes=nodes_index[top]</pre>
```

```
ntop=top_nodes.size
free=np.logical_not(np.logical_or(top, bot))
free_nodes=nodes_index[free]
nfree=free_nodes.size
Free_x=smatrix((np.ones(nfree),(np.
 ⇒arange(nfree),free_nodes)),shape=(2*nfree+1,2*Nnodes))
Free_y=smatrix((np.ones(nfree),(np.
 →arange(nfree)+nfree,free_nodes+Nnodes)),shape=(2*nfree+1,2*Nnodes))
Const_y=smatrix((np.ones(ntop),(2*nfree*np.
 ⇔ones(ntop),top_nodes+Nnodes)),shape=(2*nfree+1,2*Nnodes))
plt.plot(X[conn,0].T,X[conn,1].T,'b-');
plt.plot(X[top,0],X[top,1].T,'ro');
plt.plot(X[bot,0],X[bot,1].T,'ro');
plt.axis('equal');
ff=plt.figure()
plt.plot(E_e,Sref/pix2m*1.e-6,'bo',label='Mechanical states')
plt.plot(Eref,Sref/pix2m*1.e-6,'k.',label='Reference constitutive law');
plt.xlabel('Strain')
plt.ylabel('Stress [MPa]');
plt.title('Noise level U %g pixel, F %g N' % (stdu,stdf))
plt.legend();
```







```
[5]: # Operator assembly
D=B.T*W
Dc=(Free_x+Free_y+Const_y)*(B.T*W)
Fext=np.zeros(2*nfree+1)
Fext[-1]=Fres

iW=scipy.sparse.spdiags(1/W.diagonal(),0,Nelems,Nelems)
C=Dc*(iW*Dc.T)
LU=splinalg.splu(C)
```

/tmp/ipykernel_81593/876515232.py:9: SparseEfficiencyWarning: splu converted its
input to CSC format
 LU=splinalg.splu(C)

1.4.4 Resolution

```
[6]:
# Initialisation
E_e=B.dot(U)
S_e=np.zeros(Nelems)
```

```
E_e=Eref+dE_e
#ff=plt.figure()
#plt.yscale('log')
#plt.ylabel('DDI norm')
#plt.xlabel('Number of iteration')
ic=0
for resampling in range(5):
    ## selection matrix from k-means
    samples=np.c_[np.squeeze(E_e*np.sqrt(Co)),np.squeeze(S_e/np.sqrt(Co))]
   kmeans = KMeans(Ns).fit(samples)
    #.reshape(-1,1))
   ie=kmeans.labels_
   val=np.ones(Nelems)
   ii=np.arange(Nelems)
   S=smatrix((val,(ii,ie)),shape=(Nelems,Ns))
   STS=S.T*S
   STS=STS.diagonal()
    ## Material states
   Estar=S.T.dot(E_e)/STS
   Sstar=S.T.dot(S_e)/STS
   ff=plt.figure()
   plt.subplot(121)
   plt.plot(E e,S e/pix2m*1.e-6,'bo',label='Mechanical states')
   plt.plot(Estar,Sstar/pix2m*1.e-6,'r+',label='Material states');
   plt.xlabel('Strain')
   plt.ylabel('Stress [MPa]');
   plt.title('Iteration %d after clustering' % (resampling))
   plt.legend();
    # Projection
   Estar_e=S.dot(Estar)
   Sstar_e=S.dot(Sstar)
   b=Dc.dot(Sstar_e)-Fext
   Lag=LU.solve(b)
   S_e=Sstar_e-iW*(Dc.T.dot(Lag))
   Estar_e=S.dot(Estar)
   Sstar e=S.dot(Sstar)
   ddi_norm=0.5*(Co*np.dot(E_e-Estar_e,W.dot(E_e-Estar_e))+1/Co*np.
 →dot(S_e-Sstar_e,W.dot(S_e-Sstar_e)))
   print('***DDI loop Iteration %02d: DDI norm %6.3e ***' %
 ⇒(resampling,ddi_norm))
   plt.subplot(122)
   plt.plot(E_e,S_e/pix2m*1.e-6,'bo',label='Mechanical states')
   plt.plot(Estar,Sstar/pix2m*1.e-6,'r+',label='Material states');
   plt.xlabel('Strain')
```

```
plt.ylabel('Stress [MPa]');
plt.title('Iteration %d after projection' % (resampling))
plt.legend();
```

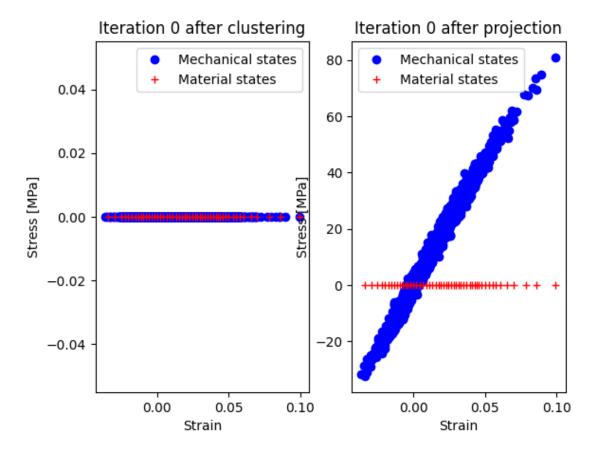
```
***DDI loop Iteration 00: DDI norm 2.619e+12 ***

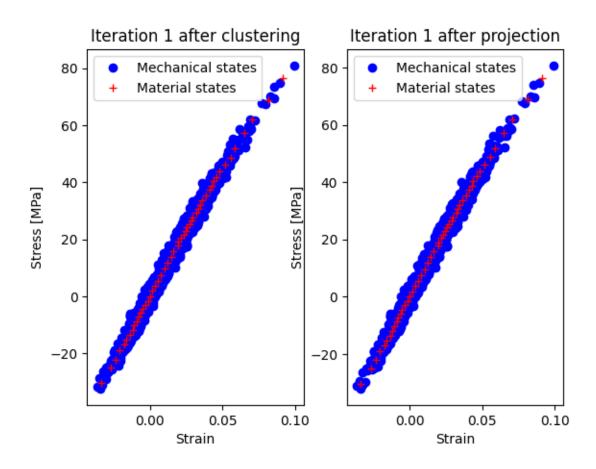
***DDI loop Iteration 01: DDI norm 9.214e+08 ***

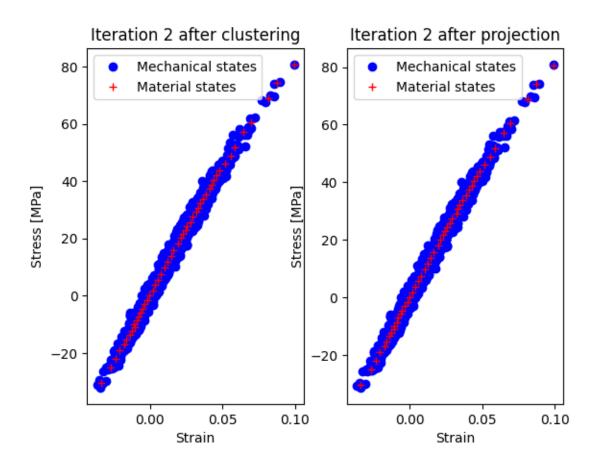
***DDI loop Iteration 02: DDI norm 7.138e+08 ***

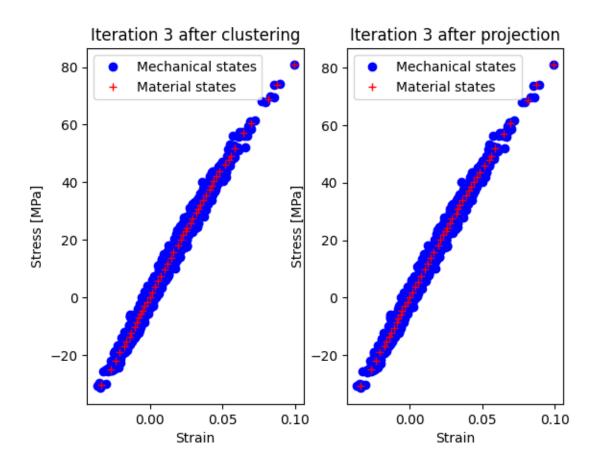
***DDI loop Iteration 03: DDI norm 6.544e+08 ***

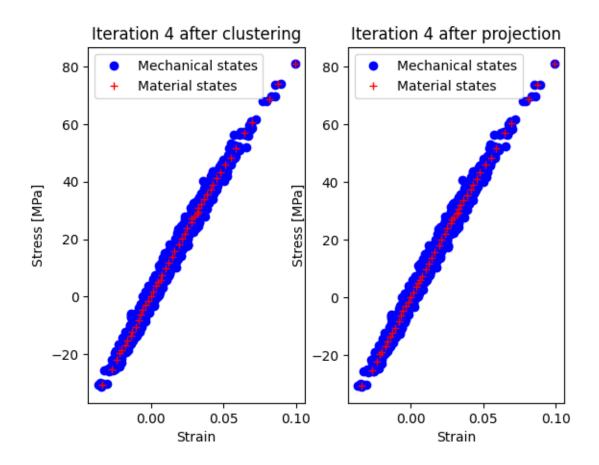
***DDI loop Iteration 04: DDI norm 6.437e+08 ***
```







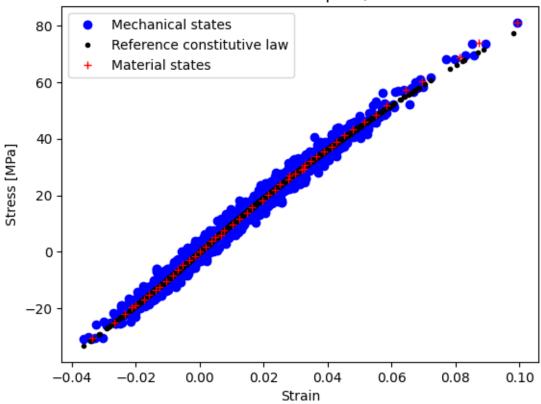




2 Comparison with reference

```
[7]: ff=plt.figure()
    plt.plot(E_e,S_e/pix2m*1.e-6,'bo',label='Mechanical states')
    plt.plot(Eref,Sref/pix2m*1.e-6,'k.',label='Reference constitutive law');
    plt.plot(Estar,Sstar/pix2m*1.e-6,'r+',label='Material states');
    plt.xlabel('Strain')
    plt.ylabel('Stress [MPa]');
    plt.title('Noise level U %g pixel, F %g N' % (stdu,stdf))
    plt.legend();
```

Noise level U 0.1 pixel, F 0 N

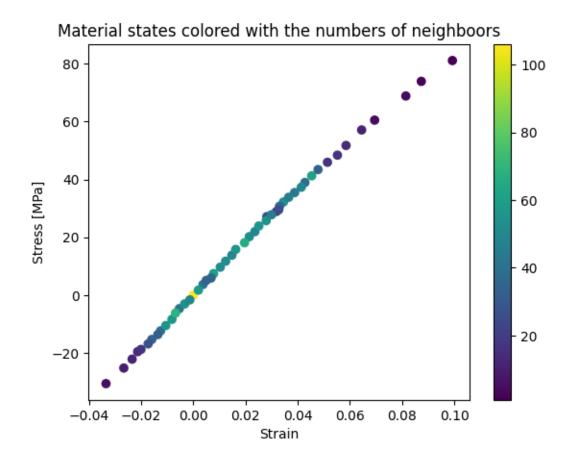


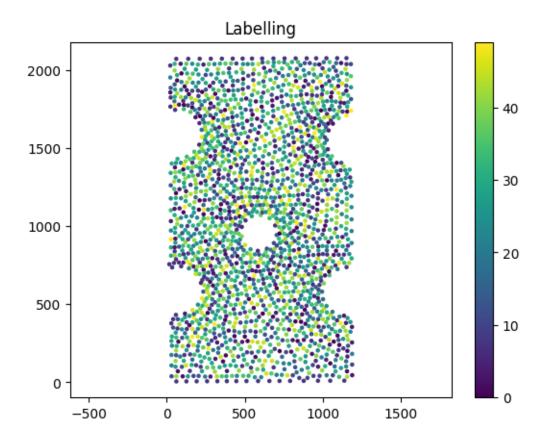
2.1 Illustrations

```
[8]: STS=S.T*S
    iSTS=STS.diagonal()
    ff=plt.figure()
    plt.scatter(Estar,Sstar/pix2m*1.e-6,c=iSTS);
    plt.xlabel('Strain')
    plt.ylabel('Stress [MPa]');
    plt.title('Material states colored with the numbers of neighboors')
    plt.colorbar();

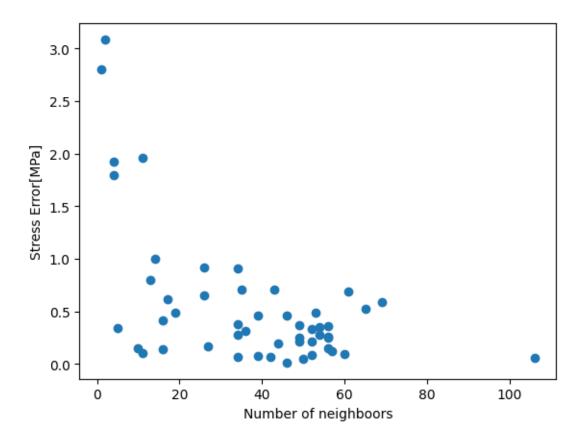
    Xg=0.5*(X[conn[:,0]]+X[conn[:,1]])
    ff=plt.figure();
    plt.scatter(Xg[:,0],Xg[:,1],c=ie,s=5);
    plt.colorbar();
    plt.axis('equal');
    plt.title('Labelling')
```

[8]: Text(0.5, 1.0, 'Labelling')





```
[9]: Sref=2e8*(1-np.exp(-np.abs(Estar)/0.2))*np.sign(Estar)*pix2m
plt.scatter(iSTS,np.abs(Sref-Sstar)/pix2m*1.e-6);
plt.xlabel('Number of neighboors')
plt.ylabel('Stress Error[MPa]');
```



[]:

Class's references i



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