

Homework/Programming Assignment #1

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Homework Due: 02/24/2022- 5:00PM

Signature (required)

I/We have followed the rules in completing this Assignment.

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I/We have followed the rules in completing this

Assignment

Question	Points	Total
HA1		
HA 2		
HA 3a		
HA 3b		
HA 4a		
HA 4b		
HA 4c		
HA 5a		
HA 5b		
HA 6		
HA 7		
HA 8		
HA 9		
PA 1		
PA 2		
PA 3		

Instruction:

- 1. Remember that this is a graded homework assignment. It is the equivalent of a mini take-home exam.
- 2. You are to work alone or in teams of two and are not to discuss the problems with anyone other than the TAs or the instructor.
- 3. It is open book, notes, and web. But you should cite any references you consult.
- 4. Unless I say otherwise in class, it is due before the start of class on the due date mentioned in the P/H Assignment.
- 6. **Sign and append** this score sheet as the first sheet of your assignment.
- 7. Remember to submit your assignment in Canvas.

ME 397: Algorithm for Sensor-Based Robotics. Spring 2022 THA 1, Programming Assignment

Bryant Zhou / Jonathan Salfity

Below is a description of our functions, test cases, and output we used to answer in THA1 PA. Links to functions:

```
rotation 2 axis angle
rotation 2 quaternion
rotation 2 ZYZ
rotation 2 rpy
axis angle 2 rotation
quaternion 2 rotation
Screw Axis
```

The input params, output values, and references for each function are within the function comments.

For most test cases, we compared the output values of our functions with the builtin Matlab functions. We picked specific values for the inputs and outputs that reflect corner cases and singularities.

To account for numerical precision, we used a global epsilon value:

```
function eps = getGlobaleps
  eps = 1e-3;
return
```

For all of our functions that input a Rotation matrix, we wrote an Rinso3 function to check the input rotation matrix is in SO(3). We used an epsilon value for tolerance to account for computational error. In most cases, if the Rotation matrix is not in SO(3), the functions calling Rinso will return NaN for their function output values.

rotation_2_axis_angle

This function takes in a rotation matrix and outputs the corresponding axis and angle of rotation. There are a few checks to perform: The first singularity case is checked if R = I, because there is no rotation in this case. The second case checks trace(R) == -1, as it would be a rotation by pi. Finally, if those checks pass, the axis and angle are returned.

```
% PA 1a
function [theta, omega] = rotation 2 axis angle(R)
   % param: R (3x3 rotation matrix)
   % return: omega, theta (axis-angle representation)
   % reference: ASBR notes, W2L2, slide 9. MR Section 3.2
   % check R is in SO(3)
   if ~RinSO3(R)
       theta = nan;
       omega = nan;
       return
   end
   if R == eye(3)
       theta = 0;
       omega = NaN;
   elseif sum(diag(R)) == -1
       theta = pi;
       if R(3,3) \sim -1
           omega = (1/sqrt(2*(1+R(3,3)))) * [R(1,3); R(2,3); 1+R(3,3)];
       elseif R(2,2) \sim -1
           omega = (1/sqrt(2*(1+R(2,2)))) * [R(1,2); 1+R(2,2); R(3,2)];
       else %R(1,1) \sim= 0
           omega = (1/sqrt(2*(1+R(1,1)))) * [1+R(1,1); R(2,1); 1+R(3,1)];
       end
   else
       theta = acos(0.5 * (sum(diag(R)) - 1));
       omega hat = (R - R') / (2 * sin(theta)); % screw symmetric matrix
       omega 1 = \text{omega hat}(3,2);
       omega 2 = \text{omega hat}(1,3);
       omega 3 = \text{omega hat}(2,1);
       omega = [omega 1; omega 2; omega 3];
   end
   return
end
Test:
% 1a rotation_2_axis_angle
% To find an axis angle from a rotation matrix:
```

```
% The first singularity case is checked if R = I, because there is no
% rotation in this case.
% The second case checks trace(R) == -1, as it would be a rotation by pi.
% Finally, if those checks pass, the axis and angle are returned.
   %case 1
   R = eye(3);
   [theta, omega] = rotation 2 axis angle(R);
   assert(isnan(omega))
   assert(theta == 0)
   % case 2
   R = [1 \ 0 \ 0; \ 0 \ -1 \ 0; \ 0 \ 0 \ -1];
   [theta, omega] = rotation 2 axis angle(R);
   assert( abs(theta - pi) < eps)</pre>
   % case 3 (MR Example 3.12)
   R = [0.8658 -0.2502 0.4333;
        0.2502 0.9665 0.0581;
        -0.4333 0.0581 0.8994];
   [theta, omega] = rotation_2_axis_angle(R);
   assert(norm(omega - [0;0.866;0.5]) < eps);</pre>
   assert(abs(theta - 0.524) < eps);
```

rotation_2_quaternion

This function takes in a rotation matrix and uses the formula provided by ASBR notes W3L1.

```
function q = rotation_2_quaternion(R)
   % param: R (rotation matrix)
   % return: q (4x1 quaternion matrix)
   % reference: ASBR W3L1 pg11, MR Appendix B3
   % check R is in SO(3)
   if ~RinSO3(R)
       q = nan;
       return
   end
   q0 = 0.5 * sqrt(R(1,1) + R(2,2) + R(3,3) + 1);
   q1 = 0.5 * sign(R(3,2) - R(2,3)) * sqrt(R(1,1) - R(2,2) - R(3,3) + 1);
   q2 = 0.5 * sign(R(1,3) - R(3,1)) * sqrt(R(2,2) - R(3,3) - R(1,1) + 1);
   q3 = 0.5 * sign(R(2,1) - R(1,2)) * sqrt(R(3,3) - R(1,1) - R(2,2) + 1);
   q = [q0; q1; q2; q3];
   return
end
<u>Test:</u>
% 1b rotation 2 quaternion
% reference: rotm2quat() provided by matlab
   R = eye(3);
   q = rotation 2 quaternion(R);
   qmatlab = rotm2quat(R);
   assert(norm(q - qmatlab') < eps);</pre>
  R = [0.8658 -0.2502 0.4333;
        0.2502
                 0.9665 0.0581;
        -0.4333 0.0581 0.89941;
   q = rotation_2_quaternion(R);
   qmatlab = rotm2quat(R);
   assert(norm(q - qmatlab') < eps);</pre>
   R = [0 \ 0 \ 1; \ 0 \ 1 \ 0; \ -1 \ 0 \ 0];
   q = rotation 2 quaternion(R);
   qmatlab = rotm2quat(R);
   assert(norm(q - qmatlab') < eps);</pre>
```

rotation_2_ZYZ

This function takes in a rotation matrix and outputs the ZYZ angles. The additional input is the input angle, which designates the input angle quadrant to perform calculations.

```
% PA 1c
function [phi, theta, psi] = rotation 2 ZYZ(R, theta in)
   % param: R (3x3 rotation matrix)
   % param: theta in (designates the input angle quadrant
   % return: phi
   % return: theta
   % return: psi
   % reference: ASBR W3L1 Pg4
   % check R is in SO(3)
   if ~RinSO3(R)
      phi = nan;
       theta = nan;
      psi = nan;
       return
   % check for range of theta in
   if theta in > 0
      phi = atan2(R(2,3), R(1,3));
       theta = atan2(sqrt(R(1,3)^2 + R(2,3)^2), R(3,3));
      psi = atan2(R(3,2), -R(3,1));
   else
     phi = atan2(-R(2,3), -R(1,3));
      theta = atan2(-sqrt(R(1,3)^2 + R(2,3)^2), R(3,3));
     psi = atan2(-R(3,2), R(3,1));
   end
end
Test:
% 1c rotation 2 ZYZ
  R = eye(3);
   [phi, theta, psi] = rotation_2_ZYZ(R, -pi/2);
   angles matlab = rotm2eul(R, 'ZYZ');
   assert(norm([angles matlab(1) - phi; ...
                angles_matlab(2) - theta; ...
                angles matlab(3) - psi]) < eps);</pre>
  R = [0.8658 -0.2502 0.4333;
        0.2502
                 0.9665 0.0581;
        -0.4333 0.0581 0.8994];
   [phi, theta, psi] = rotation 2 ZYZ(R, -pi/2);
   angles matlab = rotm2eul(R, 'ZYZ');
```

rotation_2_rpy

This function takes in a rotation matrix and outputs the roll-pitch-yaw angles. The additional input is the input angle, which designates the input angle quadrant to perform calculations.

```
% PA 1c
function [phi, theta, psi] = rotation_2_rpy(R, theta_in)
   % param: R (3x3 rotation matrix)
   % param: theta in (designates the input angle quadrant
   % return: phi
   % return: theta
   % return: psi
   % reference: ASBR W3L1 Pg7
   % check R is in SO(3)
   if ~RinSO3(R)
      phi = nan;
       theta = nan;
      psi = nan;
       return
   end
   if theta in > -pi/2 && theta in < pi/2</pre>
      phi = atan2(R(2,1), R(1,1));
       theta = atan2(-R(3,1), sqrt(R(3,2)^2 + R(3,3)^2));
       psi = atan2(R(3,2), R(3,3));
   elseif theta in > pi/2 && theta in < 3*pi/2
      phi = atan2(-R(2,1), -R(1,1));
       theta = atan2(-R(3,1), -sqrt(R(3,2)^2 + R(3,3)^2));
      psi = atan2(-R(3,2), -R(3,3));
   end
end
Test:
% 1d rotation 2 rpy
% reference: rotm2eul() provided by matlab
  R = eye(3);
   [phi, theta, psi] = rotation 2 rpy(R, 0);
   angles matlab = rotm2eul(R);
   assert(norm([angles matlab(1) - phi; ...
                angles_matlab(2) - theta; ...
                angles matlab(3) - psi]) < eps);</pre>
  R = [0.8658 -0.2502 0.4333;
        0.2502
                 0.9665 0.0581;
        -0.4333 0.0581 0.8994];
   [phi, theta, psi] = rotation 2 rpy(R, 0);
   angles matlab = rotm2eul(R);
   assert(norm([angles matlab(1) - phi; ...
                angles matlab(2) - theta; ...
```

axis_angle_2_rotation

This function takes in an axis angle and rotation angle to compute Rodrigues' formula for a rotation matrix.

```
% PA 2a
function R = axis angle 2 rotation(omega, theta)
   % param: omega (3x1 orientation vector)
   % param: theta (angle [radian])
   % return: R (rotation matrix)
   % reference: MR Proposition 3.11, Example
                   0
   omega hat = [
                             -omega(3) omega(2);
                omega(3)
                               0
                                        -omega(1);
                -omega(2)
                                        0
                                                 ];
                             omega(1)
  R = eye(3) + sin(theta) * omega hat + (1 - cos(theta))*omega hat^2;
   return
end
Test:
% 2a axis angle 2 rotation (MR Example 3.12)
  omega = [0 \ 0.866 \ 0.5];
   theta = 0.524;
  R = axis_angle_2_rotation(omega, theta);
  R book = [0.8658 -0.2502 0.4333;
                    0.9665 0.0581;
             0.2502
             -0.4333 0.0581 0.8994];
   assert(norm(R - R_book) < eps);</pre>
```

quaternion_2_rotation

This function takes in an axis angle and rotation angle and computes the quaternion output.

```
function R = quaternion_2_rotation(q)
   % param: q (4x1 quaternion)
   % return: R (3x3 rotation matrix)
   % reference: ASBR W3L1 pg11, MR Appendix B.3
  q0 = q(1);
  q1 = q(2);
   q2 = q(3);
  q3 = q(4);
  R = zeros(3,3);
   % 1st column
  R(1, 1) = q0*q0 + q1*q1 - q2*q2 - q3*q3;
  R(2, 1) = 2 * (q0*q3 + q1*q2);
  R(3, 1) = 2 * (q1*q3 - q0*q2);
   % 2nd column
  R(1, 2) = 2 * (q1*q2 - q0*q3);
  R(2, 2) = q0*q0 - q1*q1 + q2*q2 - q3*q3;
  R(3, 2) = 2 * (q0*q1 + q2*q3);
  % 3rd column
  R(1, 3) = 2 * (q0*q2 + q1*q3);
  R(2, 3) = 2 * (q2*q3 - q0*q1);
  R(3, 3) = q0*q0 - q1*q1 - q2*q2 + q3*q3;
   return
end
Test:
% 2b quaternion 2 rotation
   q = [0.7071 \ 0.7071 \ 0 \ 0];
   R = quaternion 2 rotation(q);
   Rmatlab = quat2rotm(q);
   assert(norm(R - Rmatlab) < eps);</pre>
   q = [1 \ 0 \ 0 \ 0];
  R = quaternion 2 rotation(q);
  Rmatlab = quat2rotm(q);
   assert(norm(R - Rmatlab) < eps);</pre>
   q = [0.9659 \ 0 \ 0.2243 \ 0.1295];
   R = quaternion 2 rotation(q);
   Rmatlab = quat2rotm(q);
   assert(norm(R - Rmatlab) < eps);</pre>
```

Screw Axis

In this part of the programming exercise, we used the functions we wrote to plot transformations given an initial configuration T, screw axis specified by $\{q, \hat{s}, h\}$, and a range of desired angles to rotate, i.e [0, pi/4, pi/2, 3*pi/4, pi].

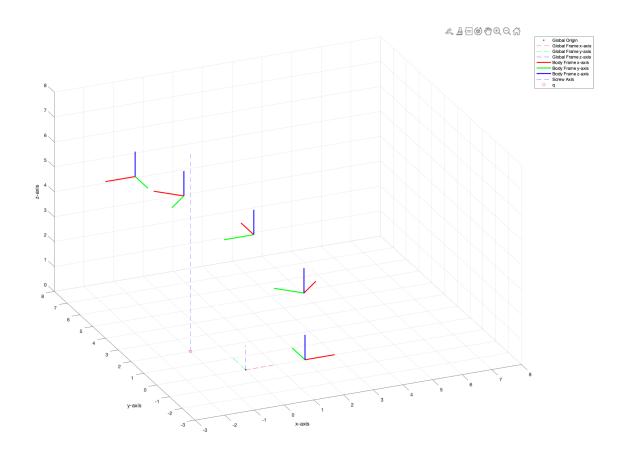
This function plots the body frame axis for all thetas by first computing a rotation matrix from our <code>axis_angle_2_rotation</code> function and a p vector using Proposition 3.25 in MR. For each theta in the list of thetas, we now have a Rotation matrix and p vector, which composes a 4x4 transformation matrix, then used to plot each body frame axis.

To compute the screw axis, we used the algorithm in section 3.3.3.2 Matrix Logarithm of Rigid-Body Motions. We then plot the screw axis of the final Transformation.

The function returns the screw axis, S, being a normalized twist, composed of the angular velocity of the transformation, \omega, and the linear velocity of the transformation, v. and the distance \theta following S that takes the rigid body, the final Transformation, to the origin.

In this example, the input parameters are:

The output looks like below:



screw axis:

Λ

0

1.0000

2.0000

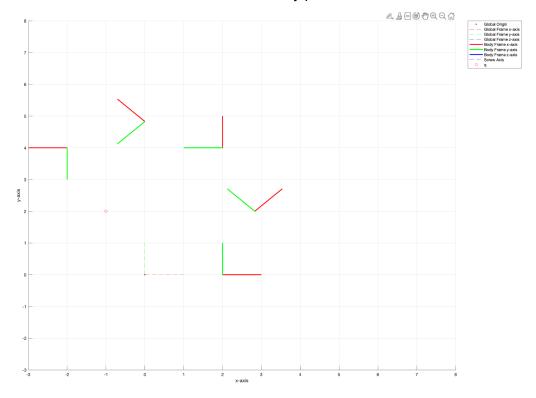
1.0000

2.0000

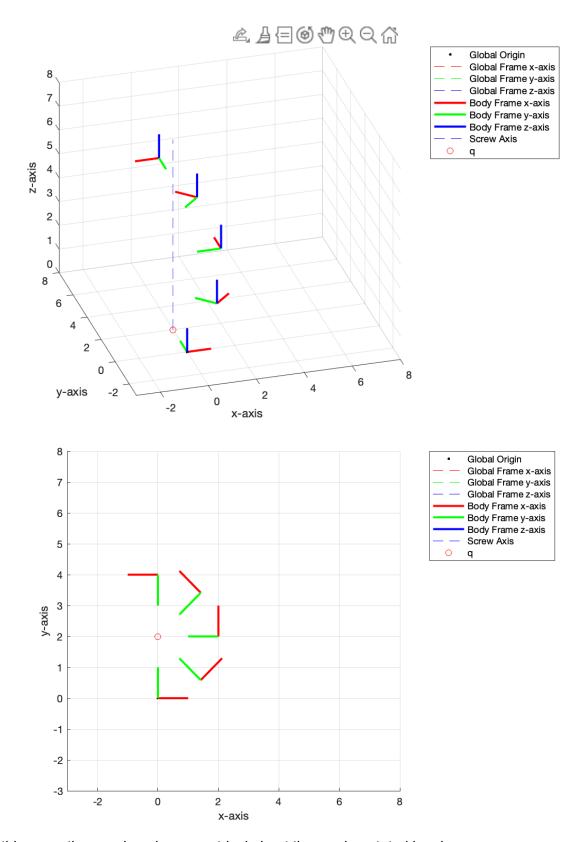
theta:

3.1416

Because there was only rotation around z, we found it interesting to view from the top down view, i.e. the x,y plane.



One method to check that our matlab code was correct was to set the input parameters to rotate around the global origin. To achieve this we set the initial T matrix as identity.



In this case, the x-y plane is symmetrical about the z axis, rotated by pi.

```
function [T_final, S_final, q_final] = screw(T_input, q, shat, h, thetas)
   % param: T input (4x4 transformation matrix)
   % param: s (screw axis [q, shat, h] in fixed frame s)
   % param: thetas (range of distances traveled along screw axis theta)
   % return: T final
   % reference: ASBR W5L1
  S input = [shat; -cross(shat,q)+h*shat];
  omega = S input(1:3);
  v = S input(4:6);
   omega hat = [
                            -omega(3) omega(2);
                             0
                                      -omega(1);
               omega(3)
               -omega(2)
                           omega(1)
                                        0
                                               ];
  figure(1)
  % origin frame
  plot handle(1) = plot3(0, 0, 0, '.k');
  hold on
  plot handle(2) = line([0 1], [0 0], [0 0], 'Color', 'r', 'LineStyle',
'--');
  plot_handle(3) =line([0 0], [0 1], [0 0], 'Color', 'g', 'LineStyle', '--');
  plot_handle(4) = line([0 0], [0 0], [0 1], 'Color', 'b', 'LineStyle',
'--');
  T final = nan; % return value
  %%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
   for theta = thetas
      R = axis angle 2 rotation(shat, theta);
(eye(3)*theta +(1-cos(theta))*omega hat+(theta -sin(theta))*omega hat^2)*v;
      T t = [R p; 0 0 0 1]*T input;
      R t = T t(1:3,1:3);
      P t = [T t(1,4), T t(2,4), T t(3,4)];
      P_x = P_t(1);
      Py = Pt(2);
      Pz = Pt(3);
      % plot rotated frame
      plot handle(5) = line(([0 R t(1,1)]+P x), ...
                            ([0 R t(2,1)]+P_y), ...
                            ([0 R t(3,1)]+P z), \dots
                            'Color', 'r', 'LineWidth', 2);
      plot handle(6) = line(([0 R t(1,2)]+P x), ...
                            ([0 R t(2,2)]+P y), \dots
                            ([0 R t(3,2)]+P z), ...
                            'Color', 'g', 'LineWidth', 2);
      plot handle(7) = line(([0 R t(1,3)]+P x), ...
                            ([0 R t(2,3)]+P y), ...
                            ([0 R_t(3,3)]+P_z), \dots
                            'Color', 'b', 'LineWidth', 2);
      hold on
```

```
if theta_ == thetas(end)
          T final = T t;
      end
  end
  %%%%% plot screw axis for T final %%%%%
  R final = T final(1:3,1:3);
  p final = T final(1:3,4);
  S final = zeros(6,1); % Return value
  % Reference MR 3.3.3.2 Matrix Logarithm of Rigid-Body Motions
  if R_final == eye(3)
      omega = [0;0;0];
      v = p_final/norm(p_final);
      theta = norm(p final);
      theta dot = norm(v);
      h = omega'*v/theta_dot;
      q = cross(omega, v)/theta dot;
      % screw axis
      S_final = [omega;v];
  else
      [theta, omega] = rotation 2 axis angle(R final);
      omega hat = [0 - omega(3) omega(2);
                  omega(3) 0 - omega(1);
                  -omega(2) omega(1) 0];
      % Rodrigues Formula to obtain axis and angle
      G inv = 1/theta*eye(3) - 1/2*omega hat +
(1/theta-1/2*cot(theta/2))*omega_hat^2;
      v = G inv * p final;
      theta dot = norm(omega);
      h = omega'*v/theta_dot;
      q = cross(omega, v)/theta dot;
      % screw axis
      S final = [omega;v] / norm(omega);
  disp('screw axis: ')
  disp(S final)
  disp('theta: ')
  disp(theta)
  x1 = q(1);
  y1 = q(2);
  z1 = q(3);
  if omega == [0;0;0]
      omega = v;
  end
  x2 = q(1) + omega(1);
  y2 = q(2) + omega(2);
  z2 = q(3) + omega(3);
  nx = x2 - x1;
```

```
ny = y2 - y1;
  nz = z2 - z1;
  len = 100;
  xx = [x1 - len*nx, x2 + len*nx];
  yy = [y1 - len*ny, y2 + len*ny];
   zz = [z1 - len*nz, z2 + len*nz];
  plot_handle(8) = plot3(xx, yy, zz,'--','Color','b');
  hold on
  plot_handle(9) = plot3(q(1),q(2),q(3),'o','Color','r');
  xlim([-3 8])
  ylim([-3 8])
  zlim([0 8])
  xlabel('x-axis')
  ylabel('y-axis')
  zlabel('z-axis')
  grid on
  legend(plot_handle([1, 2, 3, 4, 5, 6, 7, 8, 9]), ...
          {'Global Origin', ...
          'Global Frame x-axis', ...
          'Global Frame y-axis', ...
          'Global Frame z-axis', ...
          'Body Frame x-axis', ...
          'Body Frame y-axis', ...
          'Body Frame z-axis', ...
          'Screw Axis', ...
          'q'})
end
```