find_closest_approach.py

November 12, 2017

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In [1]: '''
        Finds the time of closest approach between Mars and Earth. Prints out the closest appr
        Distance is printed out for verification purposes. Time is used in the simulation prog
        import math
        from math import sin, cos
        class Planet:
            Instantiates a planet object for easy calculation and tracking of celestial orbits
            def __init__(self, semimajor_axis, eccentricity, inclination, mean_longitude, l_pe
                All parameters given in Keplerian Elements by E M Standish. Each parameter is
                and the second value being the rate of change in the value with respect to tim
                The primary units for each non-self paramater are as follows: au, none, deq, d
                J2000.0.
                111
                self.semimajor_axis = semimajor_axis
                self.eccentricity = eccentricity
                self.inclination = inclination
                self.l_ascending = l_ascending
                self.l_perihelion = l_perihelion
                self.mean_longitude = mean_longitude
            def GetLocation(self, days):
                Follows algorithm given on pages 1,2 of Keplerian Elements by E M Standish. S
                Days is Julian Days
                111
                time = self.NormalizeTime(days)
                a = (self.semimajor_axis[0] + self.semimajor_axis[1]*time)*149597870.7 # conve
                e = self.eccentricity[0] + self.eccentricity[1]*time
                I = self.inclination[0] + self.inclination[1]*time
                L = self.mean_longitude[0] + self.mean_longitude[1]*time
                ohm_bar = self.l_perihelion[0] + self.l_perihelion[1]*time
                OHM = self.l_ascending[0] + self.l_ascending[1]*time
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ohm = ohm_bar - OHM
        M = L - ohm_bar
        E = math.radians(self.GetEccentricAnomaly(M, e))
        I = math.radians(I)
        ohm = math.radians(ohm)
        OHM = math.radians(OHM)
        x_{prime} = a*(cos(E) - e)
        y_{prime} = a*math.sqrt(1-e**2)*sin(E)
        x = (\cos(ohm)*\cos(OHM)-\sin(ohm)*\sin(OHM)*\cos(I))*x_prime
        x = x + (-\sin(ohm)*\cos(OHM) - \cos(ohm)*\sin(OHM)*\cos(I))*y_prime
        y = (\cos(ohm)*\cos(OHM)+\sin(ohm)*\cos(OHM)*\cos(I))*x_prime
        y = y + (-\sin(ohm)*\sin(OHM) + \cos(ohm)*\cos(OHM)*\cos(I))*y_prime
        z = sin(ohm)*sin(I)*x_prime + cos(ohm)*sin(I)*y_prime
        return (x, y, z)
    def NormalizeTime(self, days):
        Since all time related units in __init__ are given by unit/centuries, this con
        a usable centuries since J2000.0 unit
        return (days - 2451545.0)/36525
    def GetEccentricAnomaly(self, M, e):
        Algorithm given by equations 8-36 and 8-37 of Keplerian Elements by E M Standi
        if M > 180:
            M = 360
        e_star = e
        e = math.radians(e)
        incr = 1
        E = M + e_star*sin(M)
        dE = 1
        tol = 1e-6
        while abs(dE) > tol:
            dM = M - (E - e_star*sin(E))
            dE = dM/(1-e*cos(E))
            E = E + dE
        return E
def lcm(n1, n2):
    Calculates least common multiple in case that ends up being useful
    return int((n1 * n2)/math.gcd(n1,n2))
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def distance(co1, co2):
           Returns the scalar distance between two points/vectors
           temp = 0
           for i in range(len(co1)):
                temp += (co1[i] - co2[i])**2
           return math.sqrt(temp)
       earth = Planet((1.00000261, 0.00000562), (0.01671123, -0.00004392), (-0.00001531, -0.0
                       (100.46457166, 35999.37244981), (102.93768193, 0.32327364), (0.0, 0.0))
       mars = Planet((1.52371034, 0.00001847), (0.09339410, 0.00007882), (1.84969142, -0.0081
                      (-4.55343205, 19140.30268499), (-23.94362959, 0.44441088),
                      (49.55953891, -0.29257343)) # Values from Table 1 of Keplerian Elements
       loc_earth = []
       loc_mars = []
       earth_orbital_period = 36525
       mars_orbital_period = 687
       best = [9e99, 0] # Distance (km), time (days)
       for i in range(245154500, 245154500+earth_orbital_period*100, 1):
           t = i/100
           loc_earth = earth.GetLocation(t)
           loc_mars = mars.GetLocation(t)
           dist = distance(loc_earth, loc_mars)
            if dist < best[0]:</pre>
               best = [dist, t]
       print(best)
[58214181.23314717, 2487971.95]
```