Space-Time Ambiguity Function in 3-D ISR

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- By leveraging electronically steerable phased array antenna technology, in-
- 3 coherent scatter radars (ISR) have now become full three-dimensional remote
- sensors for ionosphere plasmas. Currently these systems are operating in the
- 5 high latitude region where the ionosphere is highly dynamic in both space
- and time. Because of the highly dynamic nature of the ionosphere in this re-
- ₇ gion it is import to differentiate between artifacts and the true behavior of
- the plasma. Often the three dimension data is fitted in a spherical coordi-
- 9 nate space and then the parameters are interpolated to a Cartesian grid. This
- and other sources of error could be impacting the reconstructions of the plasma
- 11 parameters.
- In this study we present a new way of analyzing 3-D dimension ISR through
- use of the space-time ambiguity function. This concept is similar to the range
- ambiguity function that is used in traditional ISR for scanning antenna sys-
- tems but has been extended to all spatial dimension along with time as well.
- The use of this new ambiguity function allow us to pose this problem in
- terms of a linear inverse problem for the lags of the intrinsic plasma auto-
- correlation function. From this we can explore the impact of non-stationarity
- in the plasma parameters in both time and space. Along with showing pos-
- 20 sible artifacts we will begin to explore ways of reducing these artifacts.

1. Introduction

Incoherent scatter radar (ISR) systems have enabled researchers since the 1950's to explore the ionosphere [Gordon, 1958]. Using methodology developed by Dougherty, Farley and others these systems can give measurements of electron density N_e , Ion 23 temperature T_i , electron temperature T_e , Ion velocity V_i and other plasma parameters [Dougherty and Farley, 1960], [Farley et al., 1961], [Hagfors, 1961], [Dougherty and Farley, 1963. These parameters are measured by fitting a nonlinear theoretic autocorrelation function (ACF) model derived from first principles physics to an es-27 timated intrapulse time autocorrelation of the scattered radar signal [Lehtinen and Huuskonen, 1996]. As with any real world measurement method there is limit to the spatial and temporal resolution of ISR systems. This resolution is determined by the ability of the sensor to create independent measurements of the ACF in time and space of 32 what is essentially a zero-mean complex Gaussian process. At space and time extents smaller then these resolutions the random process that the ACF is derived from is 34 assumed to be stationary. In ISR literature the ambiguity function encapsulates the spatial resolution in the range dimension. This ambiguity emerges as a function of

range and correlation lag due to the pulse modulation and the finite bandwidth of

the receiver [Farley, 1969]. The time resolution is determined by how long it takes

to average the autocorrelation function to reduce the variance of the estimate to a

40 desired level.

- Because the assumption of stationarity at the level of the ambiguity function constrains the measurement and creates a blurring and other forms of artifacts due to the
 non-linear fitting of plasma parameters, a number techniques have been developed
 to reduce the impact of these ambiguities. Some techniques include trying to reduce
 the extent of the ambiguity function by adjusting the waveform. Barker code wave
 forms are used in ISR but mainly for low altitude studies of the E-region due to the
 inability to form a spectrum from them. Another method for reducing the ambiguity
 in range is using alternating codes [Sulzer, 1993].
- Another set of techniques often shown in the literate focus on reducing the impact of the ambiguity function through post processing. The first method full profile analysis consists of fitting value to the entire range gate of values [Holt et al., 1992], [Lehtinen, 1989], [Lehtinen et al., 1997], [Hysell et al., 2008]. This technique often places constraints on the physical parameters as in Hysell et al. [2008]. Because the the relationship between the autocorrelation function and the lags are non-linear this method often yields algorithms that require a large amount computation.
- The other set of post processing methods is deconvolution methods, such as in Nikoukar et al. [2008], treat the problem as a linear inverse problem of the lags.

 Method such as total-variations or Tikihonov regularization can be applied to the lags to try to fill in the missing information nation from the null space of the operator.
- These methods are easy to compute and can rely on a large body of knowledge from engineering communities that study linear inverse problems.

- Recently phased array technology has started to be leveraged by ISR community.
- The Advance Modular Incoherent Scatter Radar (AMISR) systems have already been
- deployed both at the Poker Flat Alaska (PFISR) and Resolute Bay Canada (RISR)
- [ami, 2014]. The EISCAT-3D project is currently being developed using phased array
- technology as well and will be capable of multistatic processing [eis, 2005]. These
- 67 new systems are already being used to create three dimensional reconstructions of
- plasma parameters [Semeter et al., 2009], [Nicolls and Heinselman, 2007], [Dahlgren
- et al., 2012a],[Dahlgren et al., 2012b].
- There has been no formal derivation of an ambiguity function for all three spatial
- dimensions for these systems, although up until recently ISR systems, which only
- had dish antennas, could not mitigate the finite beam width effects. In a highly
- dynamic region like the high latitude ionosphere this lack of knowledge can be prob-
- 14 lematic. There are numerous phenomena such as polar cap patches that may be
- moving through the field of view at very high speeds [Dahlgren et al., 2012a].
- These three-dimensional reconstructions often consist of taking the fitted param-
- eters and then interpolating them to a Cartesian space from the system's natural
- ₇₈ spherical coordinate space [Butler, 2013]. The step of fitting the autocorrelation
- ₇₉ function to the theoretical functions to find the final plasma parameters is a non-
- 80 linear operation. Because of this it is impossible to exactly predict the impact on the
- ₈₁ parameter values as different plasma populations move through the field of view of
- the radar. Alternatively it is possible to treat the formation of the autocorrelation

- estimates as a linear process with each lag as an independent channel [Nikoukar et al., 2008].
- In this publication we will develop a model for a full three dimensional space-time ambiguity function for 3-D ISR systems. This function can also be modified to show the ambiguity within the rest frame of the moving plasma. In the end this full three dimensional can be represented as kernel in a Fredholm integral equation like in Equation 1 with f(s) being the lag of the autocorrelation function at a specific time and position.

$$g(t) = \int_{a}^{b} K(t,s)f(s)ds \tag{1}$$

- The impact of the three-dimensional ambiguity on moving plasma will be shown through simulation. This ISR simulator fully emulates the ISR data creation process at the IQ level.
- Lastly possible mitigation techniques will be explored. These mitigation techniques
 will borrow heavily from the image and signal processing literature.

2. Space-Time Ambiguity

In the ISR literature the measurement ambiguity along the range dimension is often referred to simply as the ambiguity function [$Hysell\ et\ al.$, 2008]. This only shows the measurement ambiguity along the range r. Due to the three-dimensional imaging capability of phased array ISR systems we will define a new set of terminology to describe this more complicated measurement which is now imaging the ionosphere in three-dimensional coordinates $\mathbf{r} = [x, y, z]^T$. We will represent the sampled coordinates within the radar with s such as \mathbf{r}_s .

Specifically we will refer to the measurement ambiguity in the range dimension as 103 the range ambiguity, $W(\tau, r_s, r)$. The measurement ambiguity in the elevation and az-104 imuth angles will be referred to as the angular or cross range ambiguity $F(\theta_s, \phi_s, \theta, \phi)$, 105 where θ is the physical elevation angle, ϕ is the physical azimuth angle, θ_s is the ele-106 vation angle that the radar is pointing at and ϕ_s is the azimuth angle that the radar is pointing at. The since these two functions are separable they can be multiplied 108 together to form the full spatial ambiguity $K(\tau, \mathbf{r}, \mathbf{r}_s)$. Lastly integration time for 109 each measurement will be referred to the time ambiguity. Again like the full spatial 110 ambiguity function we can multiply the space and time ambiguity functions to get 111 the Space-Time ambiguity function which will be represented as $L(\tau, \mathbf{r}_s, t_s, \mathbf{r}, t)$.

2.1. Derivation

In order to develop the space-time ambiguity we will use two different sets of time commonly known in radar literature as fast-time n and slow-time t. Fast-time looks at time on the order of the radar systems A/D conversion while slow-time time is looking at time on the order of the system's pulse repetition interval [Richards, 2005]. We will use the same reference for fast-time but slow time will be the time it takes the plasma to change state and thus change the ACF.

In order to start we will first develop the range ambiguity which is due to the pulse shape and receiver filter on the ISR systems. If we look at data received by the ISR

system at time n, with a wavenumber \mathbf{k} and pulse shape s(n), noted as x(n) we can see that the received signal can be represented as the following

$$x(n) \propto h(n) * \int_{\mathbf{r}} e^{-j\mathbf{k}\cdot\mathbf{r}} s(n - \frac{2r}{c}) n_e(\mathbf{r}, n) d\mathbf{r},$$
 (2)

where h(n) is the receiver filter, $n_e(\mathbf{r}, n)$ is the electron density fluctuation. Generally it is assumed that the pass band of the filter is larger than the bandwidth of the electron density fluctuation [Kudeki, 2003]. As such it can be assumed that filter will only act on the the shape of the pulse. This will change Equation 2 to the following:

$$x(n) \propto * \int_{\mathbf{r}} e^{-j\mathbf{k}\cdot\mathbf{r}} a(n - \frac{2r}{c}) n_e(\mathbf{r}, n) d\mathbf{r}$$
 (3)

where a(n) = h(n) * s(n).

When the autocorrelation of the signal is done we find the following formula, [Nikoukar et al., 2008].

$$\langle x(n)x^{*}(n+\tau)\rangle = \int_{\mathbf{r}'} \int_{\mathbf{r}} e^{-2j\mathbf{k}\cdot(\mathbf{r}'-\mathbf{r})} a(n-\frac{2r}{c})a^{*}(n+\tau-\frac{2r'}{c})\langle n_{e}(\mathbf{r},n)n_{e}(\mathbf{r}',n+\tau)\rangle d\mathbf{r}d\mathbf{r}'$$
(4)

where r' is the magnitude of the vector \mathbf{r}' . We can make some simplifying assumption at this point that the space-time autocorrelation function of $n_e(\mathbf{r},t)$, $\langle n_e(\mathbf{r},n)n_e(\mathbf{r}',n+\tau)\rangle$, will vanish as the magnitude of $\mathbf{x} \equiv \mathbf{r}' - \mathbf{r}$ increases. Once the
spatial correlation is removed we can rewrite Equation 4 as

$$\langle x(n)x^*(n+\tau)\rangle = \int_{\mathbf{r}} a(n-\frac{2r}{c})a^*(n+\tau-\frac{2r}{c}) \int_{\mathbf{x}} e^{-2j\mathbf{k}\cdot\mathbf{x}} \langle n_e(\mathbf{r},n)n_e(\mathbf{x}+\mathbf{r},n+\tau)\rangle d\mathbf{x}d\mathbf{r}.$$
(5)

The inner integral is a spatial Fourier transform evaluated at the wave number of the radar ${\bf k}$

$$\langle |n_e(\mathbf{k}, r, \tau)|^2 \rangle \equiv \int_{\mathbf{x}} e^{-2j\mathbf{k}\cdot\mathbf{x}} \langle n_e(\mathbf{r}, b)n_e(\mathbf{x} + \mathbf{r}, n + \tau) \rangle d\mathbf{x}.$$
 (6)

Equation 5 becomes

$$\langle x(n)x^*(n+\tau)\rangle = \int_r a(n-\frac{2r}{c})a^*(n+\tau-\frac{2r}{c})\langle |n_e(\tau,\mathbf{k},\mathbf{r})|^2\rangle dr.$$
 (7)

It should be noted that the term $a(n)a^*(n+\tau)$ is the soft target ambiguity function. If n is replaced with $2r_s/c$ we can see that the this function represented as $W(\tau, r_s, r)$ [Nikoukar, 2010]. In order to simplify notation we will represent $\langle |n_e(\tau, \mathbf{k}, \mathbf{r})|^2 \rangle$, as $R(\tau, \mathbf{r})$. Assuming, for the moment, that R only varies across the range dimension rwe can now represent this in the form of a Fredholm integral equation

$$\langle x(2r_s/c)x^*(2r_s/c+\tau)\rangle = \int_r W(\tau, r_s, r)R(\tau, r)dr.$$
 (8)

This function is in a way a lag dependent smoothing along the range dimension of R(au,r).

The spatial ambiguity across angle is determined by the antenna beam pattern.
In phase array radars this beam pattern is ideally the array factor multiplied by
the element pattern [Balanis, 2005]. The array factor is determined by a number of

things including the element spacing in both x and y (dx, dy) and the wave number of the radar, k. Making idealized assumptions with no mutual coupling and that the array elements are cross dipole elements AMISR systems will have the following array pattern for pointing angle (θ_s, ϕ_s) ,

$$F(\theta, \phi, \theta_s, \phi_s) = \frac{1}{2} (1 + \cos(\theta)^2) \left[\frac{1}{MN} \left(1 + e^{j(\psi_y/2 + \psi_x)} \right) \frac{\sin((M/2)\psi_x)}{\sin(\psi_x)} \frac{\sin((N/2)\psi_x)}{\sin(\psi_x/2)} \right]^2, \tag{9}$$

where $\psi_x = -kd_x(\sin\theta\cos\phi - \sin\theta_s\cos\phi_s), \ \psi_y = -kd_y(\sin\theta\sin\phi - \sin\theta_s\sin\phi_s).$

This spatial ambiguity is a separable function made up of the components of $W(\tau, r)$ and $F(\theta, \phi, \theta_s, \phi_s)$. These two functions can be combined by multiplying, creating the spatial ambiguity function $K(\tau, \mathbf{r}_s, \mathbf{r})$, and then doing a volume integration. This will create radar system's estimate of the ACF which will be referred to as $\rho(\tau, \mathbf{r}_s)$,

$$\rho(\tau, \mathbf{r}_s) = \int F(\theta, \phi, \theta_s, \phi_s) W(\tau, r_s, r) R(\tau, \mathbf{r}) dV, \tag{10}$$

$$= \int K(\tau, \mathbf{r}_s, \mathbf{r}) R(\tau, \mathbf{r}) dV. \tag{11}$$

A rendering of an example of this full ambiguity function for an uncoded long pulse and antenna pattern in (9) for four beams can be seen in Figure 1.

Lastly, since the radar requires a finite amount of time to average pulses to create the estimate of the ACF we will add slow-time dependence to R. We will add another separable function $G(t_s,t)$ to the kernel,

$$\rho(\tau, \mathbf{r}_s, t_s) = \int G(t_s, t) K(\tau, \mathbf{r}_s, \mathbf{r}) R(\tau, \mathbf{r}, t) dV dt$$
(12)

$$\rho(\tau, \mathbf{r}_s, t_s) = \int L(\tau, \mathbf{r}_s, t_s, \mathbf{r}, t) R(\tau, \mathbf{r}, t) dV dt.$$
(13)

The final kernel, $L(\tau, \mathbf{r}_s, t_s, \mathbf{r}, t)$ encompasses the full space-time ambiguity.

2.2. Ambiguity after Frame Transformation

We will now focus on the impact of the motion of plasma as it is going through the field of view of the radar. We will assume that the radar is integrating over a length of time T beginning at t_s . The kernel L will be represented as a separable function K and G as in (12). In this case G will be an indicator function of length T and centered at $t_0 + 1/2$. This will change (12) to the following,

$$\rho(\tau, \mathbf{r}_s, t_s) = \int K(\tau, \mathbf{r}_s, \mathbf{r}) \int_{t_s}^{t_s + T} R(\tau, \mathbf{r}, t) dt dV.$$
 (14)

Of specific interest are instances in the high latitude ionosphere where embedded plasma structures are moving due to the electric field of the magnetosphere. Because of this it will be assumed that the plasma is rigid object and will not deform with respect to \mathbf{r} over time period $[t_0, t_0 + T]$. Also it will be assumed that it will be moving with a constant velocity \mathbf{v} . Thus $R(\tau, \mathbf{r}, t) \Rightarrow R(\tau, \mathbf{r} + \mathbf{v}t)$. At this point (14) becomes,

$$\rho(\tau, \mathbf{r}_s, t_s) = \int \int_{t_s}^{t_s + T} K(\tau, \mathbf{r}_s, \mathbf{r}) R(\tau, \mathbf{r} + \mathbf{v}t) dt dV$$
 (15)

A change of variables where $\mathbf{r}' = \mathbf{r} + \mathbf{v}t$ acts as a Galilean transform and applies a warping to the kernel and changing the frame of reference. Then (15) becomes

$$\rho(\tau, \mathbf{r}_s, t_s) = \int \int_{t_s}^{t_s+T} K(\tau, \mathbf{r}_s, \mathbf{r}' - \mathbf{v}t) R(\tau, \mathbf{r}') dt dV.$$
 (16)

Since $R(\tau, \mathbf{r}')$ is no longer dependent on t (16) becomes,

$$\rho(\tau, \mathbf{r}_s, t_s) = \int \left[\int_{t_s}^{t_s + T} K(\tau, \mathbf{r}_s, \mathbf{r}' - \mathbf{v}t) dt \right] R(\tau, \mathbf{r}') dV.$$
 (17)

By performing the integration in t the problem can now be simplified further back to a Fredholm integral equation by simply replacing the terms in the square brackets as a new kernel A,

$$\rho(\tau, \mathbf{r}_s, t_s) = \int A(\tau, \mathbf{r}_s, t_s, \mathbf{r}') R(\tau, \mathbf{r}') dV.$$
(18)

The impact of the plasma velocity on the ambiguity function can be seen in Figure 2. This is the same ambiguity as seen in Figure 1 but with a velocity of 500 m/s in the y direction over a period of 2 minutes. This velocity creates a larger ambiguity function in the frame of reference of the moving plasma.

The operator A can be determined through knowledge of the radar system's beam pattern along with the experiments pulse pattern, integration time and velocity of the plasma. This velocity \mathbf{v} can be estimated by taking taking measurements of the Doppler shift and using a methodology seen in [Butler et al., 2010]. Once the

In order to determine if it is possible to improve the resolution of ISR processing

operator has been determined standard processing techniques can be used as if the plasma is not moving.

3. Simulation

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synthetic data was created using a known condition of a simulated ionosphere. The 188 simulator creates data by deriving filters from the autocorrelation function and applying it to complex white Gaussian noise. In a sense every point in time and space 190 noise plant and filter structure as in Figure 3. The data is then scaled and summed together according to its location in range and angle space to radar. 192 After the IQ data has been created it is processed in to create an estimates of the ACF at desired points of space. This processing follows flow chart seen in Figure 4. The sampled I/Q can be represented as $x(n) \in \mathbb{C}^N$ where N is the number of 195 samples in an inter pulse period. At this point the first step in estimating the autocorrelation function is taken. For each range gate $m \in \{0, 1, ..., M-1\}$ an autocorrelation 197 is estimated for each lag of $l \in \{0, 1, ..., L-1\}$. To get better statistics this operation is performed for each pulse $j \in 0, 1, ... J - 1$ and then summed over the J pulses. The 199

$$\hat{R}(m,l) = \sum_{j=0}^{J-1} x(m - \lfloor l/2 \rfloor, j) x^*(m + \lceil l/2 \rceil, j).$$
(19)

The case shown in Equation 20 is a centered lag product, other types of lag products calculations are available but generally a centered product is used. In the centered lag

entire operation to form the initial estimate of $\hat{R}(m,l)$ can be seen in Equation 20:

product case range gate index m and sample index n can be related by $m = n - \lfloor L/2 \rfloor$ and the maximum lag and sample relation is $M = N - \lceil L/2 \rceil$.

After the lag products have been formed an estimate of the noise correlation is subtracted out of $\hat{R}(m,l)$, which is defined as $\hat{R}_w(m,l)$:

$$\hat{R}_w(m_w, l) = \sum_{j=0}^{J-1} w(m_w - \lfloor l/2 \rfloor, j) w^*(m_w + \lceil l/2 \rceil, j),$$
(20)

where $w(n_w)$ is the background noise process of the radar. Often the noise process is sampled during a calibration period for the radar when nothing is being emitted.

The final estimate of the autocorrelation function after the noise subtraction and summation rule will be represented by $\hat{R}_f(m,l)$. At this point a summation rule is applied and the data is sent off to be fit.

In order to demonstrate the blurring taking place from the motion of plasma a 212 phantom ionosphere is created where a small plasma enhancement moves through 213 the radar field of view. The background electron density varies in altitude as a 214 Chapman function while the electron and ion temperature remains constant. This is 215 done to avoid having to do full fit and thus only try to measure the electron density. 216 Also estimates of the zeroth lag are only necessary. Added to this is a 35 km radius sphere of enhance electron density of about 5×10^{10} m⁻³ centered at 400 km altitude 218 moving at 500 m/s along the y direction. Images from this phantom can be seen in 219 Figure 7. 220

Using the phantom we can see how just simply changing the integration time can impact the reconstruction. In Figure 6 we can see a case were only 10 pulses are used

for the reconstruction. This corresponds to an integration time of about 9 seconds
using the 121 beam pattern. The enhancement can be seen with concentrated energy
as it moves through the field of view. The problem is that there is a high amount
of variance in the reconstruction. Figure 7 shows the reconstruction with 200 pulses,
minute integration time. The variability has been reduced but there is a large
amount of blurring of the enhancement as it moves through the field of view.

4. Possible Mitigation Techniques

There are a number of possible ways to remove the ISR operator function to the
data. A relatively simply way to remove the problem is to process the data in the
frame of reference of the plasma. This technique includes measuring the ion velocity
of the plasma and then integrating the beams that the plasma is present in as it
moves through the field of view.

5. Conclusion

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- 303 **Acknowledgments.** (Text here)

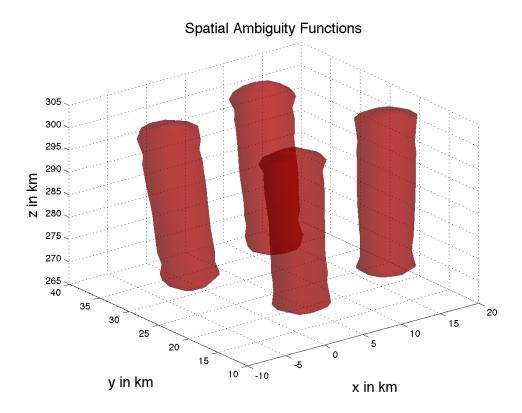


Figure 1. Full Spatial Ambiguity Function

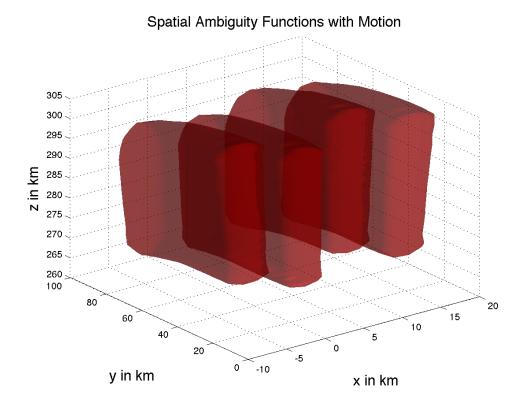


Figure 2. Full Spatial Ambiguity Function With Motion

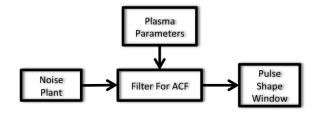


Figure 3. I/Q Simulator Diagram

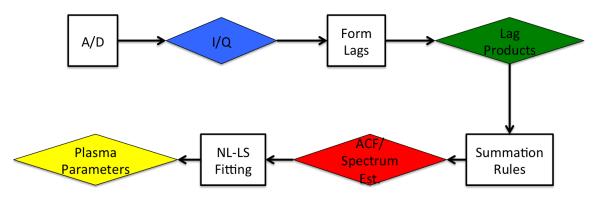


Figure 4. ISR Processing Chain.

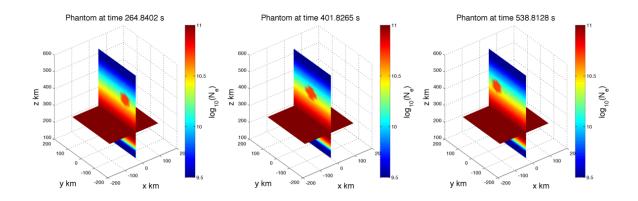


Figure 5. Example images from phantom

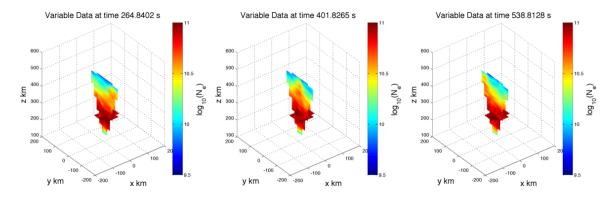


Figure 6. Example images from variable data

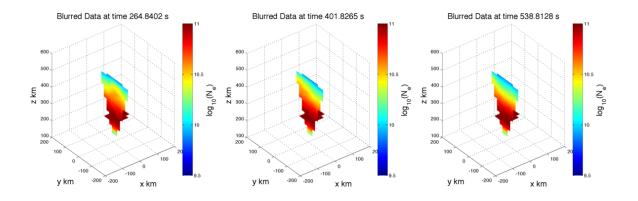


Figure 7. Example images from blurred data