

Section 3: Incompressible Fluid Flow

Recall the governing equations:

NSE's: $\vec{u}_t + \vec{u} \cdot \nabla \vec{u} = \nu \Delta \vec{u} - \frac{1}{\rho} \nabla p$ (Conservation of momentum)
(nonlinear)

$$\nabla \cdot \vec{u} = 0$$

(continuity eqn)

Linearize ($Re = \frac{UL}{\nu} \rightarrow 0$):

$$(\nu = \frac{\mu}{\rho})$$

Stokes' eq's

$$\vec{u}_t = \nu \Delta \vec{u} - \frac{1}{\rho} \nabla p$$

Common simplification (ρ constant):

$$\frac{1}{\rho} \nabla p \rightarrow \nabla p \quad \left(\text{rescaled } \tilde{p} = \frac{p}{\rho} \right)$$

Remember: Pressure Poisson Equation

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(PPE)

Take divergence of NSE:

$$\Delta p = \nabla \cdot (\mu \Delta \vec{u} - \rho \vec{u} \cdot \nabla \vec{u})$$

↖ elliptic eqn for pressure.

Notes:

Incompressible

- NSE's are mixed elliptic-parabolic type.

- When convection dominates ($Re \rightarrow \infty$) they take on hyperbolic character.

- Compressible flow has

$$\rho_t + \nabla \cdot (\rho \vec{u}) = 0$$

$$p = p(\rho)$$

which is an evolution equation for p
this is in some ways much easier than the PPE to solve.

Numerical Issues

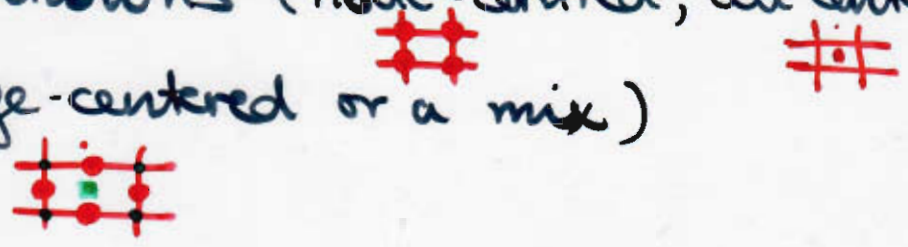
① Dimension must be at least 2, since there are no non-trivial 1D incompressible flows.
(exception: 1D hyperbolic problems)

② Choice of unknowns and equations:

- primitive variables: \vec{u}, p
- streamfunction-vorticity: $\psi, \vec{\omega}$
- velocity-height for SWER: \vec{u}, h
- others

③ Boundary conditions for pressure
(derivation and accurate implementation)

④ Choice of FD/FV grid and location of unknowns (node-centred, cell-centred, edge-centred or a mix)



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⑤ Explicit vs. implicit time-marching:

a) advection terms (hyperbolic in nature):

- can devise good high-order explicit methods
- implicit methods overly diffusive.

b) diffusion terms (parabolic in nature):

- need implicit methods for stability and efficiency.

SOLUTION: mixed implicit-explicit method

OR solve fully implicit (too complicated nonlinear solvers)

⑥ Dealing with incompressibility condition and conserving mass as best as possible

REMEMBER: they're related because

$$\left. \begin{array}{l} \text{incompressible: } \frac{D\rho}{Dt} = 0 \\ \text{conservation of mass: } \frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \vec{u}) = 0 \end{array} \right\} \Rightarrow \nabla \cdot \vec{u} = 0$$

NOTE: solving $\nabla \cdot \vec{u} = 0$ directly is problematic
eg. What are appropriate BC's?

Outline of this Section

- Stream-function - vorticity formulation for 2D flows (adv: eliminates pressure) (Pot. §13.1)
- Primitive variable methods (\vec{u}, p) :
 - artificial compressibility (Pot., p 628)
 - projection methods (Pot. §13.4)
 - ... others? ...

(APOLOGY)

Many ideas from (numerical) linear algebra will be introduced with minimum detail.

Stream-Function-Vorticity Methods

Ref: Pozrikidis, Ch.13 (1997).

In 2D, incompressible Navier-Stokes, introduce a streamfunction Ψ such that

$$u = \frac{\partial \Psi}{\partial y} \quad \text{and} \quad v = -\frac{\partial \Psi}{\partial x}$$

$$\text{Then } \nabla \cdot \vec{u} = \frac{\partial}{\partial x} \left(\frac{\partial \Psi}{\partial y} \right) + \frac{\partial}{\partial y} \left(-\frac{\partial \Psi}{\partial x} \right)$$

$$= 0$$

(automatically satisfied)
DROPS OUT!

Note: Lines of constant Ψ (contours) are streamlines of the flow.

Vorticity is $\vec{\omega} = \nabla \times \vec{u}$ and represents the rotational nature of the flow.

In 2D: $\omega = v_x - u_y$ is a scalar, and

$\omega > 0 \Rightarrow$ clockwise rotation

$\omega < 0 \Rightarrow$ counter-clockwise rotation

Note: $\omega = v_x - u_y = (-\psi_x)_x - (\psi_y)_y = -\Delta\psi$ 9-7

$$\boxed{\omega = -\Delta\psi}$$

①

elliptic eqn
for ψ .

Remember 2D NSE's:

$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) - \frac{1}{\rho} \frac{\partial p}{\partial x} \quad (a)$$

$$\frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = \nu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) - \frac{1}{\rho} \frac{\partial p}{\partial y} \quad (b)$$

(ρ constant) and take $\frac{\partial}{\partial y}(a) - \frac{\partial}{\partial x}(b)$:

$$\boxed{\frac{\partial \omega}{\partial t} + u \frac{\partial \omega}{\partial x} + v \frac{\partial \omega}{\partial y} = \nu \Delta \omega} \quad \textcircled{2} \quad (\text{eliminates pressure})$$

Equations $\textcircled{1}$ and $\textcircled{2}$ (2 eqns, 2 unknowns)
replace ^{inc.} NSE's (3 eqns, 3 unknowns), with
"hidden" conditions $u = \frac{\partial \psi}{\partial y}$, $v = -\frac{\partial \psi}{\partial x}$, $\omega = v_x - u_y$.

Alternate forms of $\textcircled{2}$:

$$\vec{u} \cdot \omega: \quad \frac{\partial \omega}{\partial t} + \vec{u} \cdot \nabla \omega = \nu \Delta \omega \quad \xrightarrow{\text{J}(\omega, \psi) \text{ - Jacobian}}$$

$$\psi \cdot \omega: \quad \frac{\partial \omega}{\partial t} + \frac{\partial \psi}{\partial y} \frac{\partial \omega}{\partial x} - \frac{\partial \psi}{\partial x} \frac{\partial \omega}{\partial y} = \nu \Delta \omega$$

$$\left\{ \begin{array}{l} \frac{\partial \omega}{\partial t} + \frac{\partial \psi}{\partial y} \frac{\partial \omega}{\partial x} - \frac{\partial \psi}{\partial x} \frac{\partial \omega}{\partial y} = \nu \Delta \omega \\ \Delta \psi = -\omega \end{array} \right.$$

Notes:

- Just as the primitive variable eqns have no evolution equation for pressure, there is none for ψ either (elliptic).
- Only 2 unknowns and 2 equations (incompressibility is "built in").
- ψ obeys a fourth order equation:

$$\frac{\partial \Delta \psi}{\partial t} - \frac{\partial \psi}{\partial y} \frac{\partial \Delta \psi}{\partial x} + \frac{\partial \psi}{\partial x} \frac{\partial \Delta \psi}{\partial y} = \nu \Delta^2 \psi$$

Boundary conditions

- No flow through solid walls:

$$\vec{u} \cdot \vec{n} = 0 \Rightarrow \psi = c \quad (\text{boundary is a streamline})$$

and can take $c=0$ since constant is arbitrary

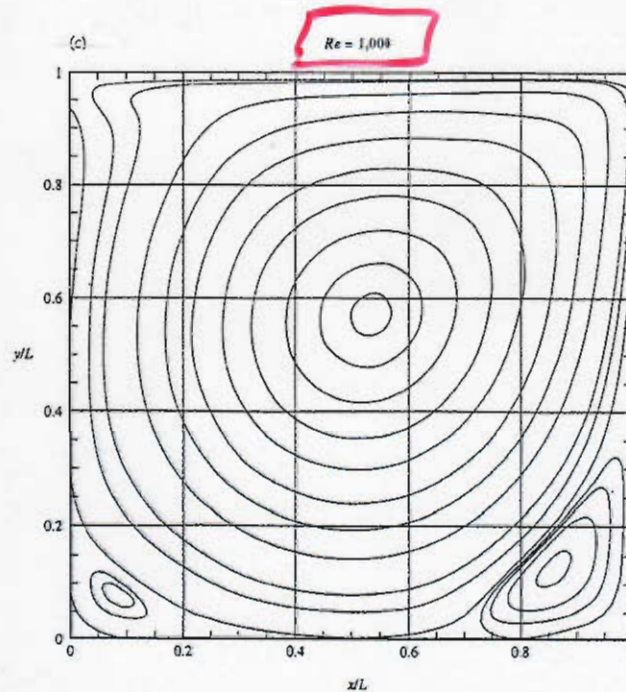
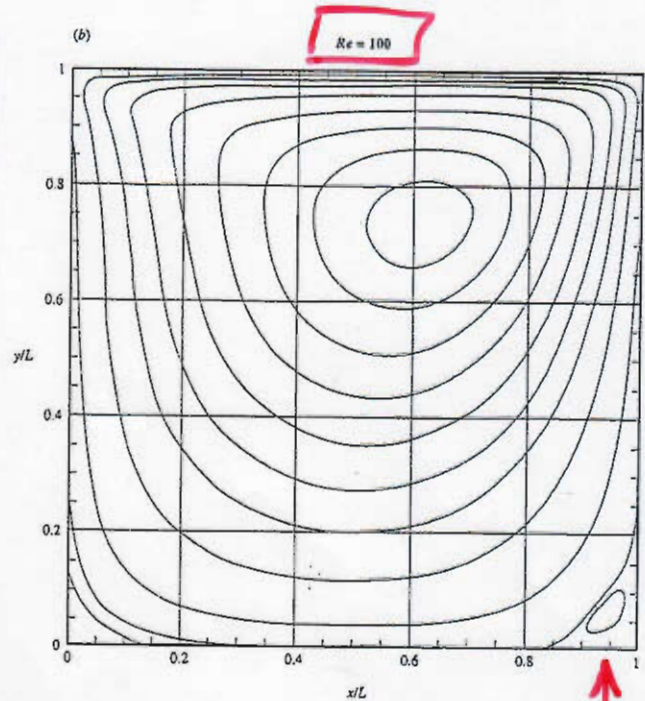
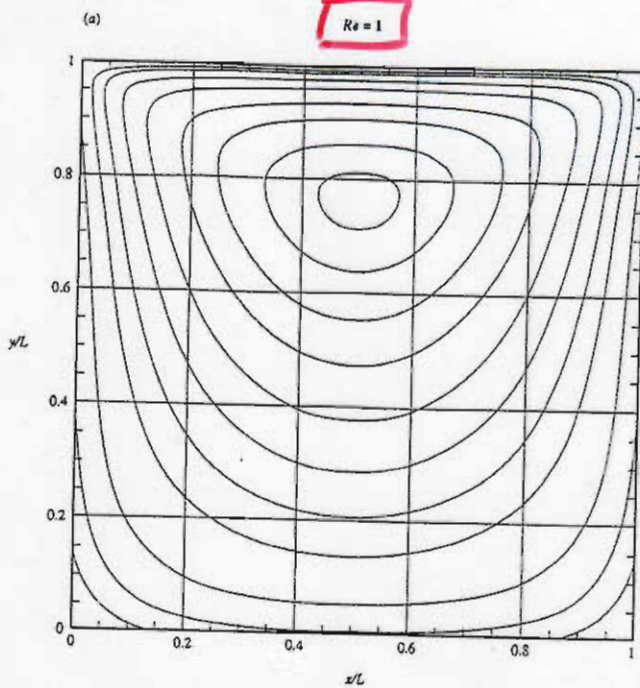
- No ^{tangential} slip on solid walls:

$$\vec{u} \cdot \vec{\tau} = 0 \Rightarrow (\psi_y, -\psi_x) \cdot \vec{\tau} = 0$$

On each solid wall, there are two BC's for ψ . This makes sense because ψ satisfies a 4th order equation.

COMPARE: to primitive variables ✓

Driven Cavity Flow with $Re = \frac{VL}{\nu}$



Source: Shankar & Deshpande

Example

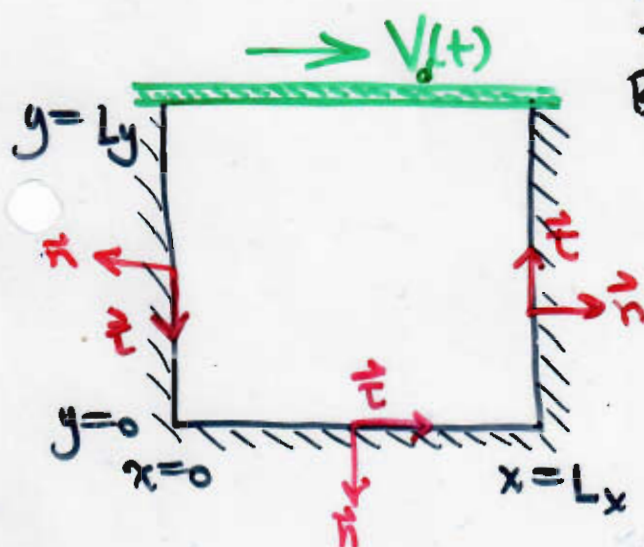
$$\frac{\partial \omega}{\partial t} + \frac{\partial \psi}{\partial y} \frac{\partial \omega}{\partial x} - \frac{\partial \psi}{\partial x} \frac{\partial \omega}{\partial y} = \nu \Delta \omega$$

$$\Delta \psi = -\omega$$

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Impulsively-started, driven-cavity flow:

- $L_x \times L_y$ rectangular box.
- top lid moves with given velocity $V_d(t)$.
- other three walls are stationary.
- starts with zero velocity - "impulsive".



IC's: $\omega = 0$ at $t=0$

BC's: $\psi = 0$ on all 4 sides.

$\psi_x = 0$ on left - $(\psi_y, -\psi_x) \cdot (0, -1) = 0$

$\psi_x = 0$ on right - $(\psi_y, -\psi_x) \cdot (0, 1) = 0$

$\psi_y = 0$ on bottom

$\psi_y = V_d(t)$ on top

Discretize on a regular grid:

$$x_i = i \cdot \Delta x \quad \text{with} \quad \Delta x = \frac{L_x}{N_x}$$

$$y_j = j \cdot \Delta y \quad \text{with} \quad \Delta y = \frac{L_y}{N_y}$$

$$i = 0, 1, \dots, N_x, \quad j = 0, 1, 2, \dots, N_y$$

Solution Algorithm

EXPLICIT, $O(\Delta t, \Delta x^2)$

(Keeps track of u, v as "intermediate" variables)

① Initialize:

$$u_{ij}^0 = v_{ij}^0 = 0$$

(IC/BC) at all interior and boundary points ij

Set except $u_{i,N}^0 = V_d$
 $n = 0$.

② Differentiate velocity to get vorticity at interior points:

$$\omega_{ij}^n = \frac{v_{i+1,j}^n - v_{i-1,j}^n}{2\Delta x} - \frac{u_{i,j+1}^n - u_{i,j-1}^n}{2\Delta y}$$

$$i = 1, 2, \dots, N_x - 1$$

$$j = 1, 2, \dots, N_y - 1$$

At side and bottom boundaries, use second-order, one-sided differences:

• Along $x=0$: $\omega_{0,j}^n = v_x|_{0,j} - u_y|_{0,j}$ because $u=0$ along entire side & $u(0,y,t)=0 \Rightarrow u_y(0,y,t)=0$

$$\approx \frac{-3v_{0,j}^n + 4v_{1,j}^n - v_{2,j}^n}{2\Delta x}$$

$$= \frac{4v_{1,j}^n - v_{2,j}^n}{2\Delta x}$$

• Along $x = L_x$: $\omega_{N,j}^n = \cancel{v_x^n|_{N,j}} - \cancel{u_y^n|_{N,j}} \rightarrow 0$
 $\approx \frac{V_{N-2,j}^n - 4V_{N-1,j}^n + 3V_{N,j}^n}{2\Delta x}$
 $= \frac{V_{N-2,j}^n - 4V_{N-1,j}^n}{2\Delta x}$

• Along $y = 0$: $\omega_{i,0}^n = \cancel{v_x^n|_{i,0}} - u_y^n|_{i,0} \rightarrow 0$
 $= - \left(\frac{-3\cancel{u_{i,0}^n} + 4u_{i,1}^n - u_{i,2}^n}{2\Delta y} \right)$
 $= - \frac{4u_{i,1}^n + u_{i,2}^n}{2\Delta y}$

• Along $y = L_y$: $\omega_{i,N}^n = \cancel{v_x^n|_{i,N}} - \cancel{u_y^n|_{i,N}} \rightarrow 0$
 $= - \left(\frac{u_{i,N-2}^n - 4u_{i,N-1}^n + 3V_d}{2\Delta y} \right)$

\Rightarrow all boundary values of w are known!

③ Solve vorticity equation using FTCS (explicit) and BC's from ② to set ω_{ij}^{n+1} :

NOTE: at high Re, use upwind differences!

$$\omega_{ij}^{n+1} = \omega_{ij}^n - \Delta t \left[U_{ij}^n D_0^x \omega_{ij}^n + V_{ij}^n D_0^y \omega_{ij}^n \right] + \nu \Delta t \left[D_2^x \omega_{ij}^n + D_2^y \omega_{ij}^n \right]$$

for $i = 1, 2, \dots, N_x - 1$
 $j = 1, 2, \dots, N_y - 1$.

④ Solve Poisson eqn $\Delta \psi^{n+1} = -\omega^{n+1}$:

$$D_2^x \psi_{ij}^{n+1} + D_2^y \psi_{ij}^{n+1} = -\omega_{ij}^{n+1}$$

with $\psi_{ij}^{n+1} \big|_{\partial \Omega} = 0$ \leftarrow only use Dirichlet BC.

for $i = 1, 2, \dots, N_x - 1$
 $j = 1, 2, \dots, N_y - 1$.

⑤ Difference Ψ_{ij}^{n+1} to get velocities:

$$U_{ij}^{n+1} = D_0^y \Psi_{ij}^{n+1}$$

$$V_{ij}^{n+1} = -D_0^x \Psi_{ij}^{n+1}$$

for $i=1, 2, \dots, N_x-1$

$j=1, 2, \dots, N_y-1$

(only interior values needed)

⑥ Increment n and goto step ②.

Notes

- Only Dirichlet BC's for ψ are used, although we could have used Neumann BC's also (from $\vec{u} \cdot \vec{n} = 0$) **BUT** then need a compatibility condition

$$\int_{\partial\Omega} \nabla\psi \cdot \vec{n} ds = - \int_{\Omega} \omega d\vec{x}. \quad (\text{problematic in solver})$$

- Use upwinding when Re is large.
- Can easily improve accuracy and stability using a semi-implicit method to update vorticity in step (3) (e.g. ADI).
 - **VERSION 1:** linearize (constant U, V) in both steps.
 - **VERSION 2:**
 - take first ADI substep (in x), and update ψ .
 - calculate an intermediate velocity.
 - use this for second ADI substep (in y).
 - exchange order in alternate steps.

$O(\Delta t)$

Getting $O(\Delta t^2)$ is tricky (see Pozrikidis, p.609)

- **corner** grid points excluded - avoids problems with "corner singularities" in some other methods.
- Extension to 3D - see Potrikidis, Ch. 13.