

About

This document is Junwoo’s effort in understanding “On Flying Backwards” paper [1]:

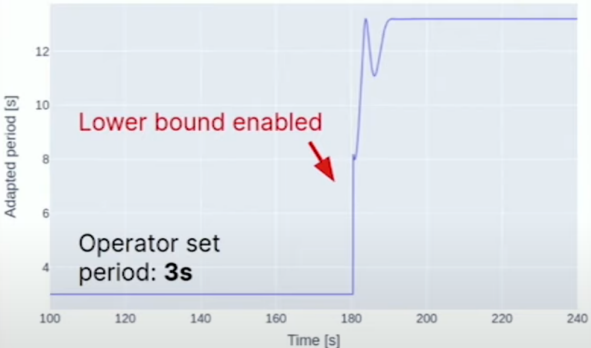
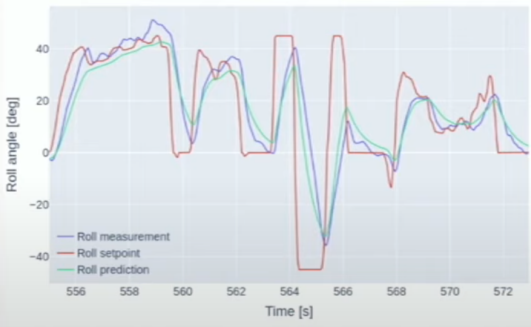
[1] T. Stastny and R. Siegwart, “TJ NPFG: On Flying Backwards: Preventing Run-away of Small, Low-speed, Fixed-wing UAVs in Strong Winds.” arXiv, Aug. 04, 2019. doi: 10.48550/arXiv.1908.01381.

Also, the presentation & PX4’s NPFG library is referenced: <https://www.youtube.com/watch?v=LY6hYBCdy-0>

Presentation Revisited - Jan 11, 2023

While writing down the [new velocity formulation](#) & pondering on how vehicle dynamics comes into play with path following guidance formulation, I wanted to revisit the formulation and find more information.

Adaptive Stability Bounds

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|---|---|
| <div><h4>Automatic stability bounds (advanced)</h4><ul style="list-style-type: none">- What if we are tuned too aggressively?- An adaptive lower bound on period keeps the controller from oscillating or diverging</div> | <p>NPFG_LB_PERIOD: Avoids limit cycling from a too aggressively tuned period/damping combination. If set to false, also disables the upper bound</p> <p>NPFG_UB_PERIOD: Adapts period to maintain track keeping in variable winds and path curvature.</p> |
| <div><h4>Automatic stability bounds (advanced)</h4><p>How to identify the roll time constant NPFG_ROLL_TC</p><ul style="list-style-type: none">- Set logging to high rate- Fly in altitude or stabilized mode- Make several full left/right roll commands, ideally with both short and long steps- Fit a first order transfer function to the roll response and set NPFG_ROLL_TC = the time constant</div> | <p>NPFG_ROLL_TC: Depends on autopilot’s behavior on roll attitude control</p> |

What can be improved

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| <h2>Outlook</h2> <ul style="list-style-type: none">- Independent wind estimator / filter for high level control feedback- Path following interface (move away from waypoints.. use paths)- VTOL / multicopter path following (NPFG can handle zero airspeed!)- Surface vessels? Submarines? | <ol style="list-style-type: none">1. Wind estimate is crucial for a reliable behavior of NPFG (PX4 Improvement)2. Path following interface (PX4 Improvement)3. VTOL / Multicopter application4. Marine Vehicles application (especially in high currents) <p>During QnA</p> <ol style="list-style-type: none">1. For Multicopter: Using the 'jerk limited trajectory' library to construct a track-error boundary is possible (makes sense)<ol style="list-style-type: none">a. Tuning is still question mark2. |
|--|--|

NPFG Parameters

https://docs.px4.io/main/en/advanced_config/parameter_reference.html#fw-npfg-control

- **NPFG_DAMPING**:
- **NPFG_PERIOD** [s]
 - Damping & Period defines:
 - P-gain for lateral acceleration control setpoint
 - Time-constant for track-error-boundary calculation
- **NPFG_PERIOD_SF**: Safety Factor. Multiplied by period for conservative minimum period bounding (when period lower bounding is enabled). 1.0 bounds at marginal stability.
- **NPFG_ROLL_TC** [s]: Time constant of roll controller command/response, modeled as first order delay. Used to determine lower period bound. Setting zero disables automatic period bounding.
 - Only used in 'lower period bound' function:
 - Less Wind: $PI * ROLL_TC / Damping$
 - Strong Wind: $4 * PI * ROLL_TC * Damping$
 - The lower bound varies during runtime by feasibility & wind factor & air-turn-rate of the path (desired)
- **NPFG_SW_DST_MLT**: Multiplied by the track error boundary to determine when the aircraft switches to the next waypoint and/or path segment. Should be less than 1. $1/\pi$ (0.32) sets the switch distance equivalent to that of the L1 controller.

Period and Damping Ratios of the NPFG (and L1)

```

/**
 * L1 period
 *
 * Used to determine the L1 gain and controller time constant. This parameter is
 * proportional to the L1 distance (which points ahead of the aircraft on the path
 * it is following). A value of 18-25 seconds works for most aircraft. Shorten
 * slowly during tuning until response is sharp without oscillation.
 *
 * @unit s
 * @min 7.0
 * @max 50.0
 * @decimal 1
 * @increment 0.5
 * @group FW L1 Control
 */
PARAM_DEFINE_FLOAT(FW_L1_PERIOD, 20.0f);

/**
 * L1 damping
 *
 * Damping factor for L1 control.
 *
 * @min 0.6
 * @max 0.9
 * @decimal 2
 * @increment 0.05
 * @group FW L1 Control
 */
PARAM_DEFINE_FLOAT(FW_L1_DAMPING, 0.75f);

```

Even in the original paper, the period & damping is only referenced for the case of straight path following.

How is the Period & Damping actually defined for NPFG & L1? How come can we actually use them unanimously between two controllers?

What are the theoretical backgrounds behind the calculation of lower period bound & p-gain (for lateral acceleration) and time constant for track-error boundary calculation?

END

Questions - from Dec 2, 2022

Question 1: Reasoning behind the Bearing Feasibility buffer zone formulation

Originally, the feasibility function is proposed as in equation 8:

(see Section III for reference command generation). In [8], the following continuous feasibility function was proposed:

$$\text{feas}(\lambda, \beta) = \frac{\sqrt{1 - (\beta \sin \bar{\lambda})^2}}{\cos \bar{\lambda}} \quad (3)$$

where $\text{feas}(\lambda, \beta) \in [0, 1]$ transitions from a value of 1 at “fully” feasible conditions ($\beta < 1$) to 0 in infeasible conditions (definition in (1)), see Fig. 3 (left). Input $\bar{\lambda} = \text{sat}(|\lambda|, 0, \frac{\pi}{2})$, where operator $\text{sat}(\cdot, \min, \max)$ saturates the input at the bounds min and max.

However, a concept of ‘buffer zone’ is introduced for dealing with multiple factors, but primarily binary jump on feasibility around the boundary: $|\lambda| \geq \frac{\pi}{2} \cap \beta = 1$.

$$\text{feas}(\lambda, \beta) = \begin{cases} 0 & \beta > \beta_+ \\ \cos^2\left(\frac{\pi}{2} \text{sat}\left(\frac{\beta - \beta_-}{\beta_+ - \beta_-}, 0, 1\right)\right) & \beta > \beta_- \\ 1 & \text{else} \end{cases} \quad (4)$$

where the upper limit of the transitioning region β_+ is approximated as a piecewise-continuous function with a linear finite cut-off to avoid singularities, the cut-off angle λ_{co} chosen small such that the regular operational envelope is not affected:

$$\beta_+ = \begin{cases} \beta_{+co} + m_{co}(\lambda_{co} - \bar{\lambda}) & \bar{\lambda} < \lambda_{co} \\ 1/\sin \bar{\lambda} & \text{else} \end{cases} \quad (5)$$

with $\beta_{+co} = 1/\sin \lambda_{co}$ and $m_{co} = \cos \lambda_{co} / \sin^2 \lambda_{co}$. The lower limit of the transitioning region β_- is similarly made piecewise-continuous to correspond with β_+ :

$$\beta_- = \begin{cases} \beta_{-co} + m_{co}(\lambda_{co} - \bar{\lambda})\beta_{buf} & \bar{\lambda} < \lambda_{co} \\ (1/\sin \bar{\lambda} - 2)\beta_{buf} + 1 & \text{else} \end{cases} \quad (6)$$

where $\beta_{-co} = (1/\sin \lambda_{co} - 2)\beta_{buf} + 1$.

However, the formulation of β_+ and β_- isn't so straightforward, because the distance between the two values doesn't equal β_{buf} , which I expected to be the case.

- What are the mathematical significances defining the β_+ and β_- in that specified way (Equation 5 & 6)?
 - It does make the function piecewise continuous, but why wasn't β_- defined to something like: $\beta_{+co} - \beta_{buf}$ instead?
- Which value should λ_{co} take? It is mentioned that it is chosen ‘small’ to not affect regular operational envelope. Does that imply that λ_{co} takes a value above $\frac{\pi}{2}$?
 - If that is the case, since $\bar{\lambda} = \text{sat}(|\lambda|, 0, \pi/2)$, the $\bar{\lambda}$ can't take a value over $\frac{\pi}{2}$. So the $\bar{\lambda} < \lambda_{co}$ will always be true. This wouldn't make sense.
- Should the formulation (Equation 5 & 6) rather compare the λ_{co} to the actual raw bearing to wind vector value λ , instead of $\bar{\lambda}$, due to the reason mentioned above? (I.e, does the paper have a typo?)

>> The PX4 code gives the latest reference, TJ said I shouldn't need to deal with this old logic

Note: an effort to code-ify this function is being done in: <https://colab.research.google.com/drive/1jhuTI5EvrFLxM09NVj4rUQbBFtf0INid?usp=sharing>.

Question 2: