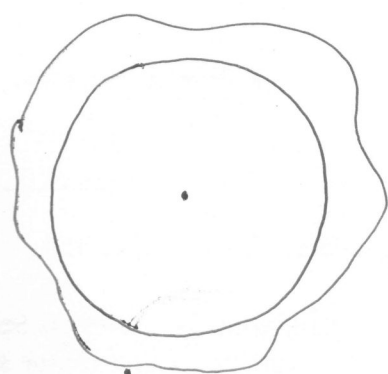


Minor - 2 Solution

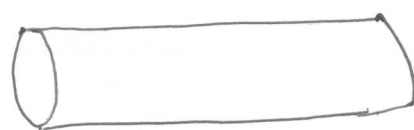
①

1. (a) $u_r(r)$: If u_r were a function of ' θ ', a circular line in cross-section will not remain circular!

② If u_r were a function of ' z ', the radius of tube would change along the tube's axis.



We don't want this!



We don't want this!

- ⑥ $u_z \equiv 0$, u_z is not a function of ' r ' or ' θ ' otherwise
① planar cross-section will become curved!

If, u_z were a function of ' z ', it will lead to $u_z' \neq 0$

but we know u_z' has to be constant, Furthermore, no ~~el~~ from Σ equation.

Furthermore, no axial elongation allowed $\Rightarrow u_z' = 0$

$$\Rightarrow \underline{\underline{u_z = 0}}$$

$$\textcircled{b} \quad \underline{\underline{\epsilon}} = \begin{bmatrix} u_r' & 0 & 0 \\ 0 & \frac{u_r}{r} & \frac{r\theta_0}{2L} \\ 0 & \frac{r\theta_0}{2L} & 0 \end{bmatrix} \quad \textcircled{2}$$

$$\textcircled{3} \quad \sigma_{rr} = \lambda \left(u_r' + \frac{u_r}{r} \right) + 2\mu u_r', \quad \sigma_{\theta\theta} = \lambda \left(u_r' + \frac{u_r}{r} \right) + 2\mu \frac{u_r}{r}$$

$$\sigma_{zz} = \lambda \left(u_r' + \frac{u_r}{r} \right), \quad \sigma_{r\theta} = \sigma_{rz} = 0, \quad \sigma_{\theta z} = \frac{G r \theta_0}{L}$$

③ 'θ' and 'z' equation get satisfied automatically.

$$\text{radial eq.} \Rightarrow \frac{d\sigma_{rr}}{dr} + \frac{\sigma_{rr} - \sigma_{\theta\theta}}{r} = 0 \quad \textcircled{2}$$

$$\Rightarrow (\lambda + 2\mu) \left(u_r'' + \frac{u_r'}{r} - \frac{u_r}{r^2} \right) = 0 \quad \textcircled{3}$$

$$\Rightarrow \left(u_r' + \frac{u_r}{r} \right)' = C$$

$$\text{or } \boxed{u_r'' + \frac{u_r'}{r} - \frac{u_r}{r^2} = 0}$$

$$\textcircled{d} \quad \text{From } \textcircled{c} \quad u_r' + \frac{u_r}{r} = \text{constant} = C \quad \textcircled{2}$$

$$\Rightarrow \sigma_{rr} = \lambda C + 2\mu u_r', \quad \sigma_{\theta\theta} = \lambda C + 2\mu \frac{u_r}{r}$$

$$\text{or } \boxed{\sigma_{rr} + \sigma_{\theta\theta} = 2(\lambda + \mu)C}$$

Again, looking at radial eq.:

$$\frac{d\sigma_{rr}}{dr} + 2\frac{\sigma_{rr}}{r} = \frac{\sigma_{rr} + \sigma_{\theta\theta}}{r} = \frac{2(\lambda + \mu)C}{r} \quad \textcircled{1}$$

$$\Rightarrow \frac{1}{r^2} \frac{d}{dr} (\sigma_{rr} r^2) = \frac{2(\lambda + \mu)c}{r} \quad (3)$$

$$\Rightarrow \boxed{\sigma_{rr} = (\lambda + \mu)c + \frac{D}{r^2}}$$

(1)

Let us use boundary condition

$$\sigma_{rr}(r_1) = -p, \quad \sigma_{rr}(r_2) = 0$$

\Downarrow

$$D = -(\lambda + \mu)c r_2^2$$

\Downarrow

$$\sigma_{rr} = (\lambda + \mu)c \left[1 - \frac{r_2^2}{r^2} \right]$$

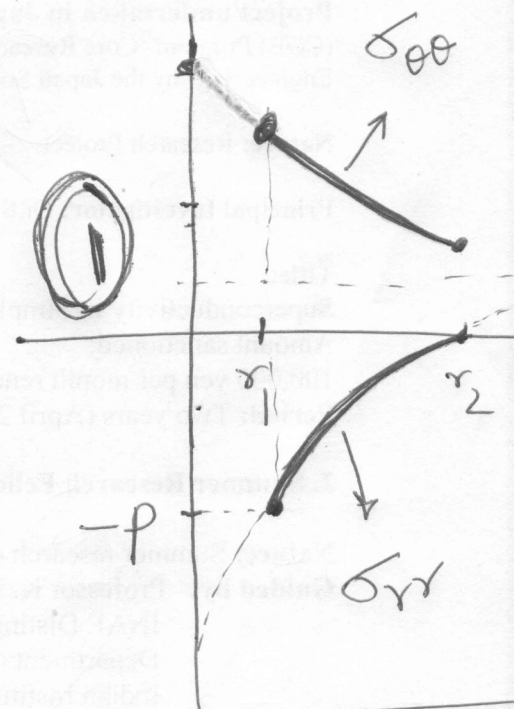
(1)

$$-p = (\lambda + \mu)c \left[1 - \frac{r_2^2}{r_1^2} \right]$$

\Downarrow

$$(\lambda + \mu)c = \frac{p r_1^2}{r_2^2 - r_1^2}$$

$$S_0 \quad \boxed{\begin{aligned} \sigma_{rr} &= \frac{p r_1^2}{(r_2^2 - r_1^2)} \left(1 - \frac{r_2^2}{r^2} \right) \\ \sigma_{\theta\theta} &= \frac{p r_1^2}{(r_2^2 - r_1^2)} \left(1 + \frac{r_2^2}{r^2} \right) \end{aligned}}$$



(e) Axial force: $\sigma_{zz} \cdot \text{Area}$

$$\textcircled{1} = \lambda \left(u_r' + \frac{u_r}{r} \right) \pi (r_2^2 - r_1^2)$$

$$= \lambda c \pi (r_2^2 - r_1^2)$$

$$\textcircled{2} = \frac{\lambda}{(\lambda + \mu)} \frac{p r_1^2}{(r_2^2 - r_1^2)} \pi (r_2^2 - r_1^2) = \frac{\lambda}{\lambda + \mu} \cdot \pi r_1^2 p$$

$$\begin{aligned} &\equiv -\nu (\sigma_{rr} + \sigma_{\theta\theta}) \pi (r_2^2 - r_1^2) \\ &\equiv -2\nu p \pi r_1^2 \end{aligned}$$

(f) To get displacement,

$$u_r' + \frac{u_r}{r} = C$$

$$\Rightarrow \frac{1}{r} \frac{d}{dr} (r u_r) = C$$

$$\Rightarrow \boxed{u_r = \frac{C}{2} r + \frac{B}{r}}$$

(1)

To obtain B,

We see, $\sigma_{\theta\theta} = \lambda C + 2\mu \frac{u_r}{r} = \frac{p r_1^2}{(r_2^2 - r_1^2)} \left(1 + \frac{r_2^2}{r^2} \right)$

Comparing coefficient of $\frac{1}{r^2}$

$$\Rightarrow B = \frac{p r_1^2 r_2^2}{2\mu(r_2^2 - r_1^2)}$$

(2)

$$S_o, u_r = \frac{p r_1^2}{(r_2^2 - r_1^2)} \left[\frac{1}{2(\lambda + \mu)} r + \frac{r_2^2}{2\mu r} \right]$$

$$S_o, u_r(r_1) = \frac{p r_1^2}{2(r_2^2 - r_1^2)} \left[\frac{r_1^2}{\lambda + \mu} + \frac{r_2^2}{\mu} \right]$$

Now, $r_1 + u_r(r_1) = \alpha r_1$, ($\alpha \equiv$ inflation)

$$\Rightarrow \alpha = 1 + \frac{p}{2(r_2^2 - r_1^2)} \left[\frac{r_1^2}{\lambda + \mu} + \frac{r_2^2}{\mu} \right]$$

(11)

$$\Rightarrow \text{inflation strain } (\alpha - 1) = \frac{p}{2(r_2^2 - r_1^2)} \left[\frac{r_1^2}{\lambda + \mu} + \frac{r_2^2}{\mu} \right]$$

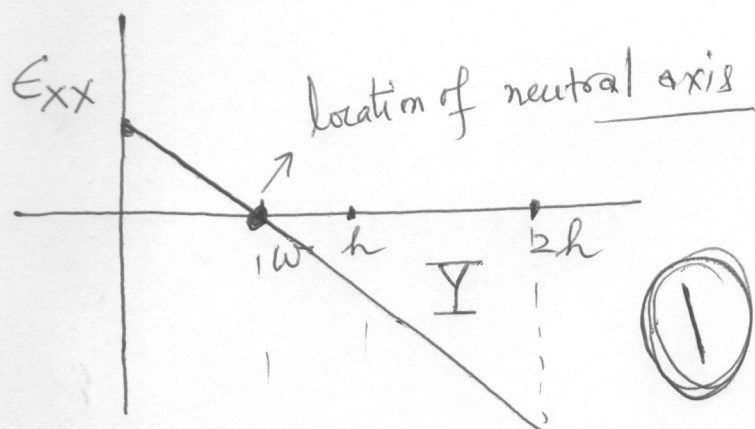
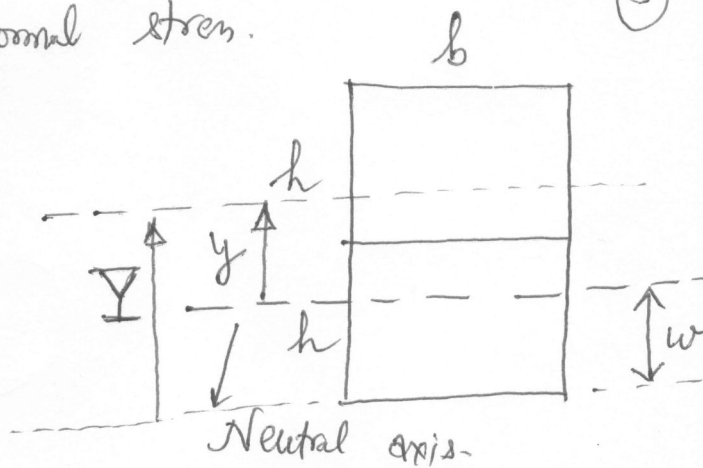
$$S_o, \boxed{p = \frac{2(r_2^2 - r_1^2)(\alpha - 1)}{\left(\frac{r_1^2}{\lambda + \mu} + \frac{r_2^2}{\mu} \right)}}$$

Problem 2

(a) Variation in normal strain and normal stress.

(5)

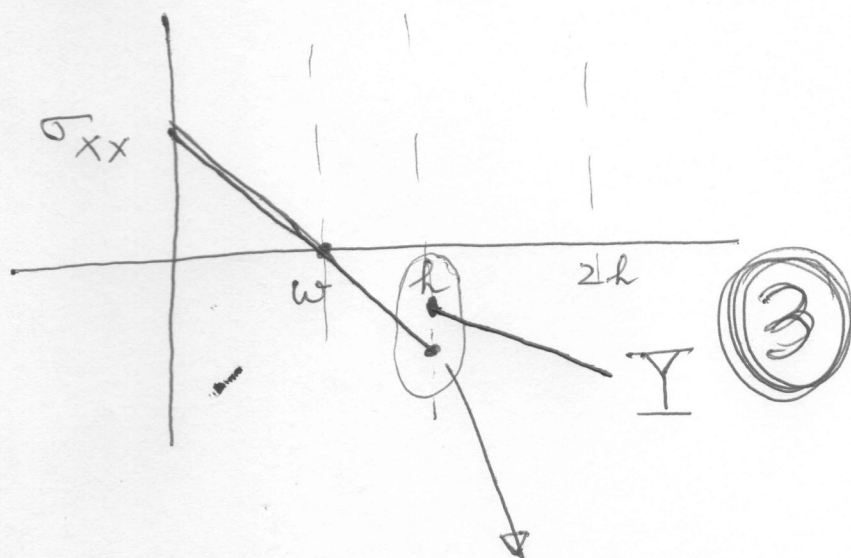
$$\epsilon_{xx} = -\frac{y}{R} = -\frac{(Y-w)}{R}$$



$$Y = w + y$$

↓
distance from bottom of cross-section

$$E_A < E_s$$

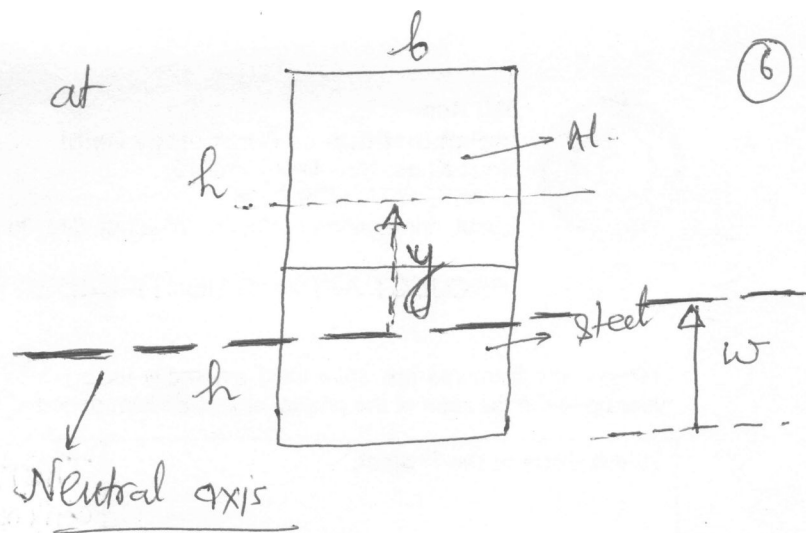


$$\sigma_{xx} = -E_s \frac{y}{R} \quad (-w < y < h-w)$$

$$= -E_A \frac{y}{R} \quad (h-w < y < 2h-w)$$

- * Notice the jump in σ_{xx}
- * Also the slope of the curve varies in Aluminium and steel!

⑤ Let the neutral axis lies at a distance w from the bottom (see figure).



then, $\epsilon_{xx} = -y/R$ (y is the distance from neutral axis)

$$\sigma_{xx} = E \epsilon_{xx} = -\frac{E_s y}{R} \quad (-w < y < h-w)$$

$$= -E_A \frac{y}{R} \quad (y > h-w)$$

To obtain w , we use the fact that the total axial force must vanish

$$\Rightarrow \int_{-w}^{h-w} -\frac{E_s}{R} y dy - \frac{E_A}{R} \int_{h-w}^{2h-w} y dy = 0$$

$$\Rightarrow \left[(h-w)^2 - w^2 \right] + \frac{E_A}{E_s} \left[(2h-w)^2 - (h-w)^2 \right] = 0$$

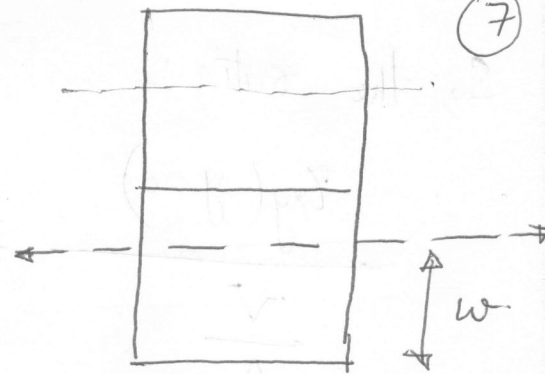
$$\Rightarrow (h^2 - 2hw) + \frac{E_A}{E_s} (3h^2 - 2hw) = 0$$

$$\Rightarrow h^2 \left(1 + 3 \frac{E_A}{E_s} \right) - 2hw \left(1 + \frac{E_A}{E_s} \right) = 0$$

$$\Rightarrow w = \frac{h}{2} \left(\frac{1 + 3 E_A/E_s}{1 + E_A/E_s} \right)$$

(c) I_{zz} about neutral axis.

$$I_{zz} = I_{zz}(A) + I_{zz}(\text{steel})$$



$$= \frac{bh^3}{12} + bh \left(\frac{3h}{2} - w \right)^2 + \frac{bh^3}{12} + \left(\frac{h}{2} - w \right)^2$$

$$= \frac{bh^3}{6} + bh \left(\frac{9h^2}{4} + \frac{h^2}{4} + 2w^2 - 4hw \right)$$

$$= \frac{8}{3} bh^3 + 2bh w (w - 2h)$$

(d) Let us first obtain the bending moment due to above distribution!!

$$M_z = - \iint \sigma_{xx} y dA$$

$$= - \int \sigma_{xx} y dy$$

(y from neutral axis)

$$= \frac{E_s}{R} \iint y^2 dA + \frac{E_{Al}}{R} \iint y^2 dA$$

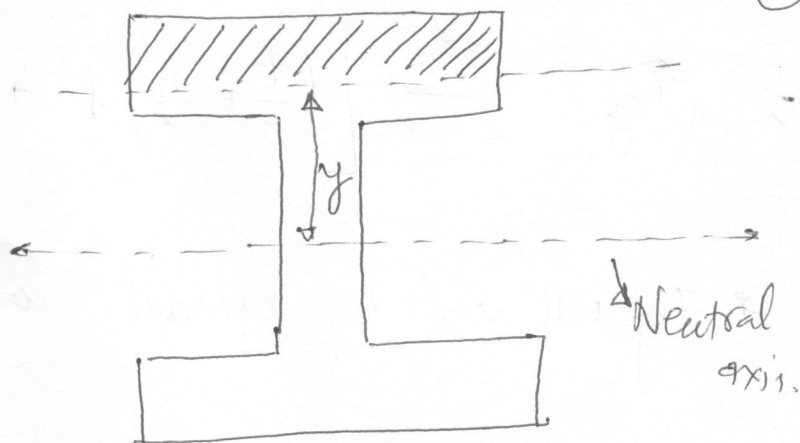
$$= \frac{E_s}{R} I_{zz}(\text{steel about neutral axis}) + \frac{E_{Al}}{R} I_{zz}(Al \text{ about neutral axis})$$

$$\text{So, } M_z = \frac{E_s I_{zz}^s + E_{Al} I_{zz}^{Al}}{R} \rightarrow \text{Effective rigidity!!}$$

③ a

$$I_{xy}(y, z) = I_{xy}(y)$$

does not vary with "z"!



$$I_{xy}(y) = \frac{V \cdot Q_y}{I_{zz} t_y}$$

for $y > \frac{h}{2}$

, $t_y = b$

$$Q_y = \bar{Y} A$$

②

$$= \left(\frac{y}{2} + \frac{1}{4}(t+h) \right) \left(\frac{t}{2} + \frac{h}{2} - y \right) b$$

$$= \frac{1}{2} \left(\frac{t+h}{2} + y \right) \left(\frac{t+h}{2} - y \right) b$$

$$= \frac{1}{2} \left(\frac{(t+h)^2}{4} - y^2 \right) b$$

$$I_{xy} = \frac{V}{2I_{zz}} \left[\frac{(t+h)^2}{4} - y^2 \right], \quad y > h/2$$

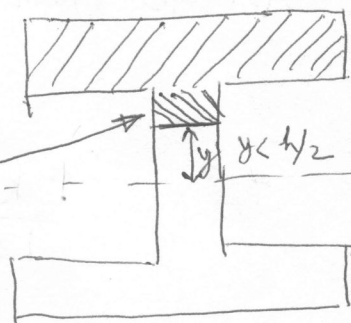
for $y < h/2$

, $t_y = t$

①

$$Q_y = \frac{1}{2} \left(\frac{t^2}{4} + \frac{th}{2} \right) b + \frac{t}{2} \left(\frac{h^2}{4} - y^2 \right)$$

②

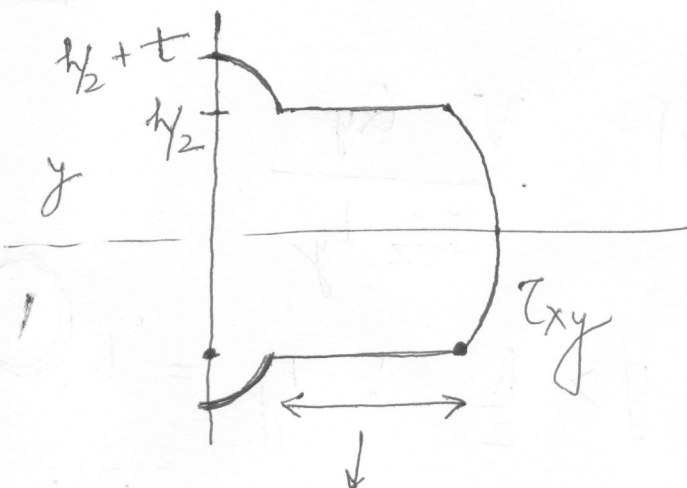


$$\Rightarrow \tau_{xy} = \frac{V}{2I_{zz}} \left[\left(\frac{t}{4} + \frac{h}{2} \right) b + \left(\frac{h^2}{4} - y^2 \right) \right], y < h/2$$

* τ_{xy} will also be symmetric in "y"!

(b)

- (1) → showing jump
- (1) → zero at $y = \pm(h/2 + t)$
- (1) → max at $y = 0$



Notice the jump in τ_{xy} due to abrupt change in the flange width!

(c) shear stress is maximum at centre! ($y = 0$)

$$\tau_{xy}(y=0) = \frac{V}{2I_{zz}} \left[\frac{tb}{4} + \frac{hb}{2} + \frac{h^2}{4} \right]$$

$$\text{Average shear stress} = \frac{V}{A} = \frac{V}{ht + 2bt} = \frac{V}{t(h+2b)}$$

$$I_{zz} = \frac{th^3}{12} + 2 \left(\frac{bt^3}{12} + \frac{bt}{4}(h+t)^2 \right)$$

$$= t \left[\frac{h^3}{12} + \frac{bt^2}{6} + \frac{b}{2}(h+t)^2 \right]$$

(1)

(2)

So, the ratio :-

$$\frac{\tau_{xy}(y=0)}{\frac{V}{A}} = \frac{3 \left(\frac{1}{2}b + 2hb + h^2 \right) (h+2b)}{2 \left(h^3 + 2ht^2 + 6b(h+t)^2 \right)}$$

this is independent of V

Also, note that to compute average shear stress, you don't need to integrate τ_{xy} since, the integration of τ_{xy} will give us 'V' itself!