MAE 275 - Homework 3

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1 Problem 1

The longitudinal linearized aircraft equations of motion can be expressed in state space form, with state variables Δu , Δw , Δq , $\Delta \theta$, Δh , as

$$A = \begin{bmatrix} X_u & X_w & 0 & -g\cos(\theta_0) & 0\\ \frac{Z_u}{1 - Z_{\dot{w}}} & \frac{Z_w}{1 - Z_{\dot{w}}} & \frac{Z_q + u_0}{1 - Z_{\dot{w}}} & \frac{g\sin\theta_0}{1 - Z_{\dot{w}}} & 0\\ M_u + \frac{M_{\dot{w}}Z_u}{1 - Z_{\dot{w}}} & M_w + \frac{M_{\dot{w}}Z_w}{1 - Z_{\dot{w}}} & M_q + \frac{M_{\dot{w}}(Z_q + u_0)}{1 - Z_{\dot{w}}} & -\frac{M_{\dot{w}}g\sin\theta_0}{1 - Z_{\dot{w}}} & 0\\ 0 & 0 & 1 & 0 & 0\\ 0 & -1 & 0 & u_0 & 0 \end{bmatrix}$$

Plugging in the data for the VZ-4 "Doak" aircraft in Appendix A of Aircraft Dynamics and Automatic Control yields

$$A = \begin{bmatrix} -1.3700e - 1 & 0 & 0 & -3.2200e + 1 & 0 \\ 0 & -1.3700e - 1 & 0 & 0 & 0 \\ +1.3600e - 2 & 0 & -4.5200e - 2 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & -1 & 0 & 0 & 0 \end{bmatrix}$$

Matrices B, C, and D can also be formed

$$B = \begin{bmatrix} \frac{X_{\delta_e}}{Z_{\delta_e}} \\ \frac{Z_{\delta_e}}{1 - Z_{\dot{w}}} \\ \frac{M_{\dot{w}} Z_{\delta_e}}{1 - Z_{\dot{w}}} + M_{\delta_e} \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 0 \\ 1.08 \\ 1.00 \\ 0 \\ 0 \end{bmatrix}$$

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 1 \end{bmatrix}$$

$$D = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

We can find the resulting transfer functions, $\frac{q}{\delta e}(s)$ and $\frac{\theta}{\delta e}(s)$, with the following commands

```
1  [n, d] = ss2tf(A, B, C, D);
2  minreal(zpk(tf(n(3, :),d))) % q
3  minreal(zpk(tf(n(4, :),d))) % theta
```

which results in

```
1 s (s+0.137)
2 (s+0.8223) (s^2 - 0.6401s + 0.5326)
```

and

```
1 (s+0.137)
2 (s+0.8223) (s^2 - 0.6401s + 0.5326)
```

2 Problem 2

The following MATLAB command is called to identify the characteristic roots and eigenvector elements

resulting in two real eigenvalues and a complex pair of eigenvalues

$$d_1 = -8.2230e - 01$$

$$d_2 = -1.3700e - 01$$

$$d_3 = +3.2005e - 01 \pm i6.5583e - 1$$

and their associated eigenvectors

```
\begin{split} v_1 &= [-9.9962e - 1,\\ &+ 0.0000e + 0,\\ &+ 1.7494e - 2,\\ &- 2.1275e - 2,\\ &+ 0.0000e + 0],\\ v_2 &= [+0.0000e + 0,\\ &+ 1.3573e - 1,\\ &+ 0.0000e + 0,\\ &+ 0.0000e + 0,\\ &+ 9.9075e - 1],\\ v_3 &= [+9.9953e - 1,\\ &+ 0.0000e + 0,\\ &+ 8.8107e - 3 \mp i1.5820e - 2,\\ &- 1.4187e - 2 \mp i2.0358e - 2,\\ &+ 0.0000e + 0] \end{split}
```

3 Problem 2