

Equation for IMU sensor model:

$$a_{\text{meas}} = a_{\text{true}} + \text{bias}_{\text{acc}} + \text{noise}_{\text{acc}}$$

$$a_{\text{meas}} = a_{\text{true}} + b_a + n_a(t)$$

Equation for a gyroscope:

$$\omega_{\text{meas}} = \omega_{\text{true}} + \text{bias}_{\text{gyro}} + \text{noise}_{\text{gyro}}$$

$$\omega_{\text{meas}} = \omega_{\text{true}} + b_g + n_g(t)$$

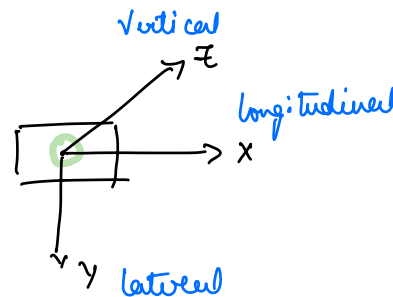
One case:

Let us use the above equations for our hover, which acceleration along longitudinal but zero acceleration along lateral direction.

Accelerometer measures the linear acceleration (x, y and z) directions

$$\text{acceleration}_{\text{true}} (a_{\text{true}}) = \begin{bmatrix} a_{\text{long}} \\ a_{\text{lat}} \\ a_{\text{vertical}} \end{bmatrix}$$

$$= \begin{bmatrix} a_x \\ a_y \\ a_z \end{bmatrix}$$



Accelerometer bias ( $b_{acc}$ ) :

This is a Constant offset along  $x$ ,  $y$  and  $z$  directions

$$b_{acc} = \begin{bmatrix} b_x \\ b_y \\ b_z \end{bmatrix} \quad \text{aka} \quad \begin{bmatrix} b_{ax} \\ b_{ay} \\ b_{az} \end{bmatrix}$$

Accelerometer noise ( $n_{acc}$ ) :

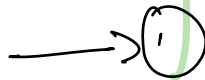
Similar to the bias, noise is affected along each direction

$$n_{acc} = \begin{bmatrix} n_{ax} \\ n_{ay} \\ n_{az} \end{bmatrix}$$

Accelerometer sensor model is given by

$$a_{measured} = \begin{bmatrix} a_x + b_{ax} + n_{ax} \\ a_y + b_{ay} + n_{ay} \\ a_z + b_{az} + n_{az} \end{bmatrix}$$

along Rover's body frame.



## Gyroscope sensor model:

Gyroscope measures the angular velocity for our rover/car

$$\omega_{true} = \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix}$$

$$\text{bias}_{gyro} = \begin{bmatrix} b_{gx} \\ b_{gy} \\ b_{gz} \end{bmatrix}$$

$$\text{noise}_{gyro} = \begin{bmatrix} \wedge_{gx} \\ \wedge_{gy} \\ \wedge_{gz} \end{bmatrix}$$

Similar to accelerometer, gyroscope equations are given by:

$$\omega_{meas} = \omega_{true} + \text{bias}_{gyro} + \text{noise}_{gyro}$$

$$\omega_{meas} = \begin{bmatrix} \omega_x + b_{gx} + \wedge_{gx} \\ \omega_y + b_{gy} + \wedge_{gy} \\ \omega_z + b_{gz} + \wedge_{gz} \end{bmatrix} \rightarrow \textcircled{2}$$

Combining ① and ②, sensor model for IMU is given by:

$$\text{IMU}_{meas} = \text{IMU}_{true} + \text{bias} + \text{noise}$$

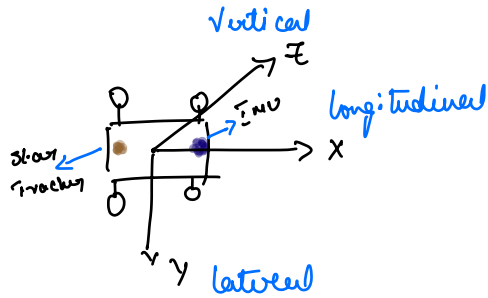
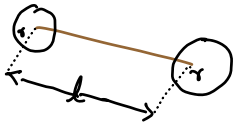
$$= \begin{bmatrix} a_x + b_{ax} + \wedge_{ax} \\ a_y + b_{ay} + \wedge_{ay} \\ a_z + b_{az} + \wedge_{az} \\ \omega_x + b_{gx} + \wedge_{gx} \\ \omega_y + b_{gy} + \wedge_{gy} \\ \omega_z + b_{gz} + \wedge_{gz} \end{bmatrix}$$

→ Ignoring  $a_y$  and  $a_z$ , as we are considering acceleration along x and making it zero in y (lateral) direction

## Star Tracker Sensor Model:

We assume that, the star tracker is placed along the rover wheel axis axle.

Rover wheel and axle:



$r \rightarrow$  Radius of the wheel of our rover

$l \rightarrow$  length of the axle of our rover

$d \rightarrow$  Center of the rover axle to the star tracker.

We assume that the rover moves along longitudinal direction without any slip.

absolute orientation

$$(\theta) = \theta_{true} + b_{st} + n_{st}.$$

absolute orientation provides the Euler angles roll, pitch and yaw i.e.  $(\alpha, \beta, \gamma)$

Position of star tracker is given by:

$$x_{st} = d \cos \theta$$

$$y_{st} = l/2$$

$$z_{st} = r(1 - \cos(\theta)) + d \sin \theta$$

$\theta \rightarrow$  wheel angle.