

IC Engine

Formual Sheet – Final Term – Spring 2021

$$c_i = \sqrt{kRT} \quad \eta_v = \frac{\dot{m}_a}{\rho_a V_d} \quad \text{bsfc} = \frac{\dot{m}_f}{\dot{W}_b} \quad Q_{\text{in}} = \dot{m}_f Q_{\text{HV}} \eta_c$$

$$\eta_v = \frac{n \dot{m}_a}{\rho_a V_d N}$$

$$U_p / \bar{U}_p = (\pi/2) \sin \theta [1 + (\cos \theta / \sqrt{R^2 - \sin^2 \theta})]$$

$$l_{\text{max}} < d_v/4$$

$$R = r/a$$

$$V_d = N_c (\pi/4) B^2 S$$

$$s = a \cos \theta + \sqrt{r^2 - a^2 \sin^2 \theta}$$

$$\bar{U}_p = 2SN$$

$$\dot{m}_f = C_D A_n \sqrt{2\rho_f \Delta P}$$

$$r_c = V_{\text{BDC}}/V_{\text{TDC}} = (V_c + V_d)/V_c = v_{\text{BDC}}/v_{\text{TDC}}$$

$$A_p = (\pi/4) B^2$$

$$I = mB^2/8$$

$$mep = \frac{W}{V}$$

$$R = \text{gas constant for air} = 0.287 \text{ kJ/kg-K} = 53.33 \text{ ft-lbf/lbm-}^\circ\text{R}$$

$$\dot{W} = WN/n$$

$$SP = \dot{W}_b / A_p$$

$$\dot{m}_f = C_D A_n \sqrt{2\rho_f \Delta P} (\Delta\theta/360N)$$

$$\dot{W} = 2\pi N\tau$$

$$OPD = \dot{W}_b / V_d$$

$$A_{\text{pass}} = \pi d_v l$$

$$\dot{W} = (1/2n)(mep)A_p \bar{U}_p$$

$$SV = V_d / \dot{W}_b$$

$$SW = (\text{engine weight}) / \dot{W}_b$$

$$\Gamma = I\omega$$

$$C_{Dv} = A_{\text{act}} / A_{\text{pass}}$$

$$(SR)_1 = (\text{angular speed}) / (\text{engine speed}) = \omega / N$$

$$(SR)_2 = (\text{swirl tangential speed}) / (\text{average piston speed}) = u_t / \bar{U}_p$$

$$Q = mCdT$$

$$\dot{W}_{\text{sc}} = \dot{m}_a (h_{\text{out}} - h_{\text{in}}) = \dot{m}_a c_p (T_{\text{out}} - T_{\text{in}})$$

$$T_{2s} = T_1 (P_2/P_1)^{(k-1)/k}$$

$$(\eta_s)_{\text{sc}} = \dot{W}_{\text{isen}} / \dot{W}_{\text{act}} = [\dot{m}_a (h_{2s} - h_1)] / [\dot{m}_a (h_{2A} - h_1)]$$

$$= [\dot{m}_a c_p (T_{2s} - T_1)] / [\dot{m}_a c_p (T_{2A} - T_1)] = (T_{2s} - T_1) / (T_{2A} - T_1)$$

$$\eta_m = (\dot{W}_{\text{act}})_{\text{sc}} / \dot{W}_{\text{from engine}}$$

$$\text{Eff} = (T_1 - T_2) / (T_1 - T_{\text{coolant}})$$

$$(\eta_s)_{\text{turb}} = (\dot{W}_t)_{\text{act}} / (\dot{W}_t)_{\text{isen}}$$

$$= [\dot{m}_a (h_1 - h_{2A})] / [\dot{m}_a (h_1 - h_{2S})] = (T_1 - T_{2A}) / (T_1 - T_{2S})$$

$$\eta_{\text{turbo}} = (\eta_s)_{\text{comp}} (\eta_s)_{\text{turb}} \eta_m$$