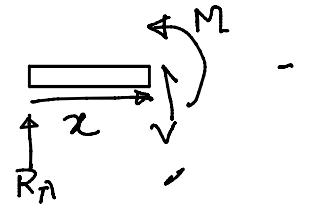


$$\left. \begin{array}{l} R_A + R_B = 0 \\ R_B L = M_0 \end{array} \right\}$$

$$0 < x < L \quad V + R_A = 0 \\ V = -R_A \\ M = -Vx = R_A x = -\frac{M_0 x}{L}$$

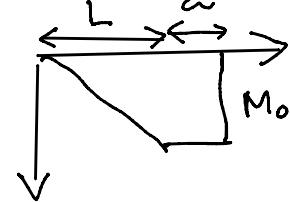


$$L < x < L+a$$

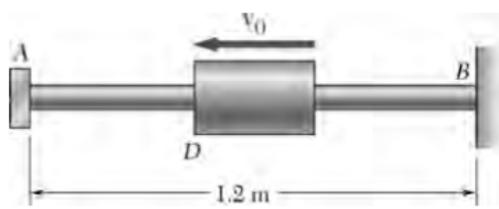
$$V + R_A + R_B = 0 \Rightarrow V = 0$$

$$M + Vx + R_B L = 0 \Rightarrow M = -R_B L = -M_0$$

$$U = \int_0^L \frac{M^2}{2EI} dx + \int_L^{L+a} \frac{M^2}{2EI} dx = \frac{1}{2EI} \int_0^L \frac{M_0^2 x^2}{L^2} dx + \frac{1}{2EI} \int_L^{L+a} \frac{M_0^2 x^2}{L^2} dx \\ = \frac{M_0^2}{2EI} L^2 \frac{x^3}{3} \Big|_0^L + \frac{M_0^2}{2EI} L^2 \frac{x^3}{3} \Big|_L^{L+a} = \frac{M_0^2 L}{6EI} + \frac{M_0^2 a^3}{2EI}$$



$$\frac{1}{2} M_0 / \Delta \theta_D = U = \frac{M_0^2}{2EI} \left( \frac{L}{3} + a \right) \Rightarrow \theta_D = \frac{M_0}{EI} \left( \frac{L}{3} + a \right)$$

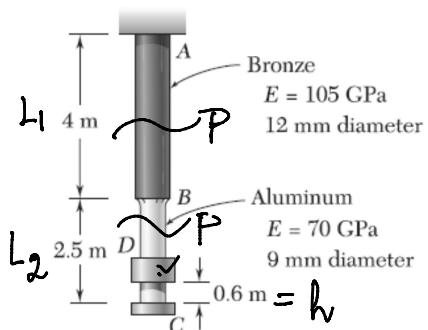


$$\frac{1}{2} m v_0^2 = \int_0^L \frac{P^2}{2EA} dx = \frac{P^2 L}{2EA}$$

$$P^2 = \frac{m v_0^2 EA}{L} \Rightarrow P = \sqrt{\frac{m v_0^2 EA}{L}}$$

$$\nabla_{\cdot} = \frac{P}{A}, \quad \Delta = \sqrt{\frac{m v_0^2 EA}{L}} \times \frac{L}{EA}$$

$$= \frac{1}{A} \sqrt{\frac{m v_0^2 EA}{L}}$$



$$mg(h + \delta_m) = \int_0^L \frac{P_m^2}{2EA} dx = \frac{P_m^2}{2} \int_0^{L_1} \frac{dx}{E_1 A_1} + \frac{P_m^2}{2} \int_{L_1}^{L_1+L_2} \frac{dx}{E_2 A_2}$$

$$= \frac{P_m^2}{2} \left( \frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2} \right)$$

Energy balance at max deflection at C

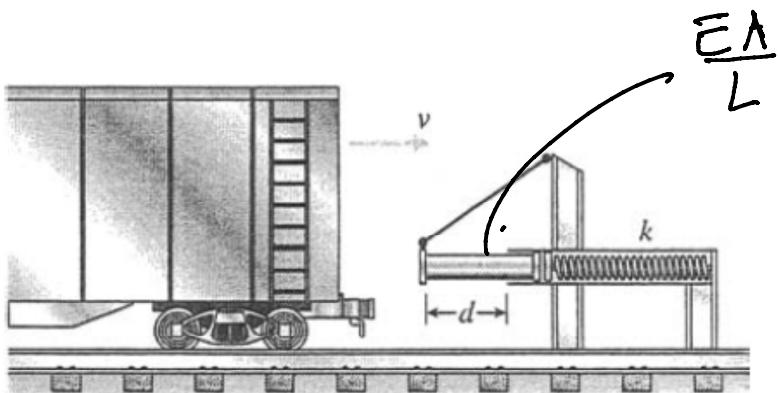
$$\delta = \frac{P}{E_1 A_1} \left( \frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2} \right) \rightarrow \text{force vs displacement at C}$$

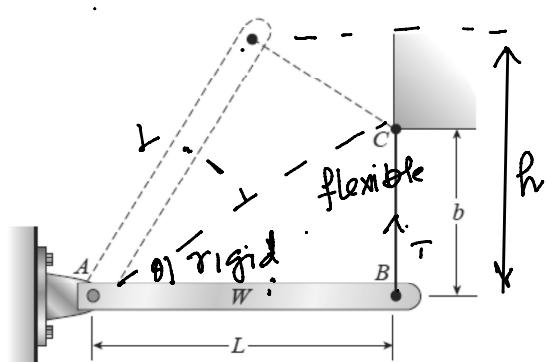
$$mg \left[ h + P_m \left( \frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2} \right) \right] = \frac{P_m^2}{2} \left( \frac{L_1}{E_1 A_1} + \frac{L_2}{E_2 A_2} \right)$$

Solve for  $P_m$  (+ve root)

$$\Delta_{m1} = \frac{P_m}{A_1}, \quad \Delta_{m2} = \frac{P_m}{A_2}$$

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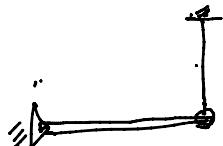
$$\tan \theta = \frac{b}{L}, \quad 2\theta = ? \quad \text{use } \sin 2\theta \text{ formula}$$

$$h = L \sin 2\theta = L \times \frac{2b/L}{1 + (b/L)^2}$$

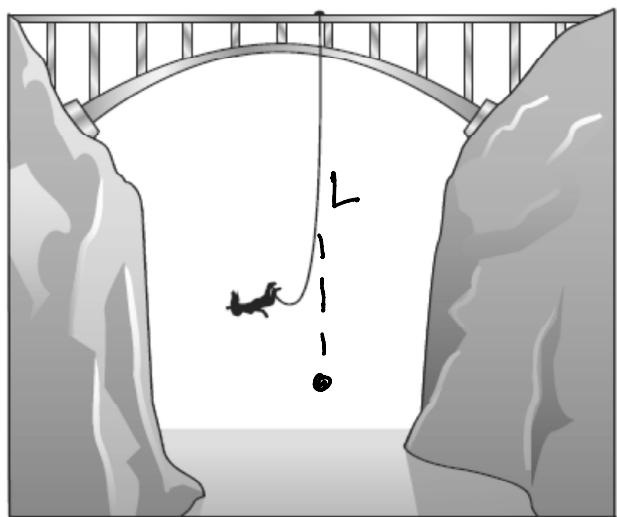
C.m. loses  $\frac{h}{2}$  height

$$mg \left( \frac{h}{2} + \delta_m \right) = \frac{T_m b}{2EA}$$

$$\delta_m = \frac{T_m b}{EA} \Rightarrow T_m = \frac{EA}{b}$$

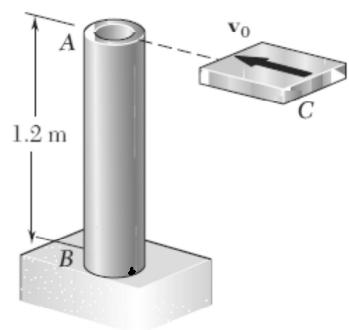


$$mg \left( \frac{h}{2} + \frac{T_m b}{EA} \right) = \frac{T_m^2 b}{2EA}$$



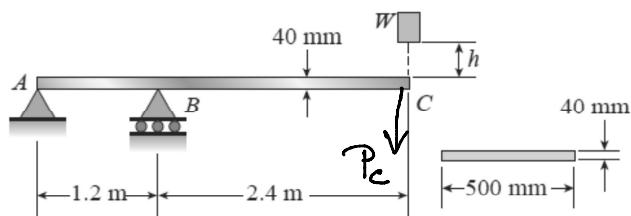
$$mg(L + \delta_m) = \frac{TL^2}{2EA}$$

$$\delta = \frac{TL}{EA}$$



$$\frac{1}{2}mv_0^2 = \int_s^L \frac{Mx}{2EI} dx = \int_0^L \frac{\frac{P(L-x)}{m}}{2EI} dx$$

$$\frac{PL^3}{3EI} = \delta, \quad \Delta_B = \frac{M_y y}{I} = \frac{P_m L y}{EI I}$$

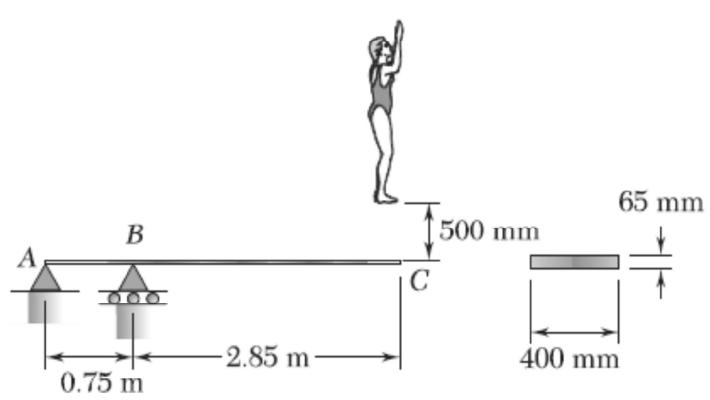


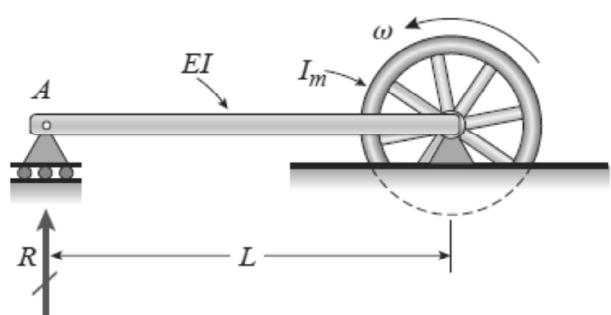
$$mg(h + \delta_{c,\max}) = \int_0^L \frac{M^2}{2EI} dx$$

$$\delta_c \rightarrow P_c.$$

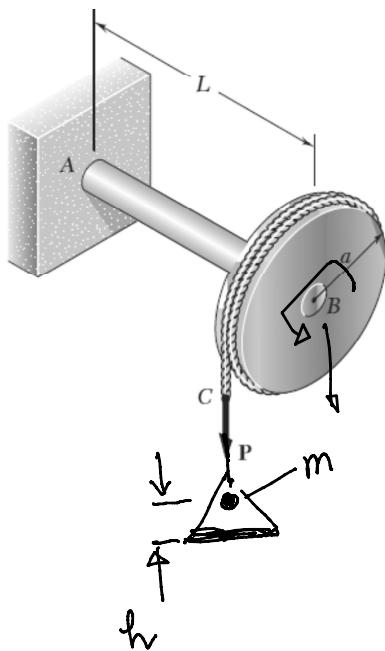
1. Find  $M(x)$  in terms of  $P_c$   $\rightarrow$  Energy balance eqn
2. Find  $\delta_c$  in terms of  $P_c$   $\rightarrow$  Substitute for  $\delta_c$  in above eqn

Real life version





$$\frac{1}{2} I m \omega^2 = \int_0^L \frac{M^2}{2EI} dx.$$



$$\frac{1}{2}P\delta = \frac{T^2}{2GJ} + \int_0^L \frac{kP(L-x)^2}{2EI} dx = U_{\text{total}}$$

$\Rightarrow T = P_n$

$$mg(h + \delta_m) = U_{\text{total}}$$

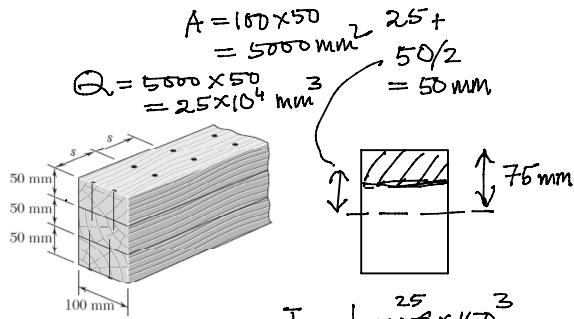
$$\frac{1}{2}P\delta = \frac{Pa^2L}{2GJ} + \frac{PL^3}{2EI} \Rightarrow \delta = P \left( \frac{a^2L}{2GJ} + \frac{L^3}{3EI} \right)$$

$$mg(h + \delta_m) = \frac{Pa^2L}{2GJ} + \frac{P_m L^3}{2EI}$$

## PROBLEM 6.2

- 6.1 Three full-size  $50 \times 100$  mm boards are nailed together to form a beam that is subjected to a vertical shear of 1500 N. Knowing that the allowable shearing force in each nail is 400 N, determine the largest longitudinal spacing  $s$  that can be used between each pair of nails.

- 6.2 For the built-up beam of Prob. 6.1, determine the allowable shear if the spacing between each of nails is  $s = 45$  mm.



a) Each nail pair supports  $400 \text{ N} \times 2$ .  
 $\therefore \text{N of pairs per mm} = \frac{400 \times 2}{2 \times 400} = \frac{1}{2} \text{ mm}^{-1}$ .  
 $\therefore \text{Spacing} = \frac{1}{2} \text{ mm} = 0.5 \text{ mm}$

$$I = \frac{1}{4} \times 150 \times 150^3 = \frac{125}{4} \times 225 \times 10^6 \text{ mm}^4$$

b)  $V$  is unknown

$$\text{Spacing} = 45 \text{ mm}$$

$$\text{Each pair supports } 400 \times 2 = 800 \text{ N.}$$

$$\text{Hence } q_{\max} = \frac{800}{45} \frac{\text{N}}{\text{mm}} = \frac{VQ}{I} \Rightarrow \frac{800}{45} = V \times \frac{25 \times 10^3}{125 \times 225 \times 10^6} \text{ mm}^{-1}$$

$$\Rightarrow V_{\text{allow}} = \frac{25}{125 \times 225 \times 10^6} \times \frac{800}{45} = 2000 \text{ N}$$

$$a) q = \frac{VQ}{I} \quad ?$$

$$b) q = \frac{VQ}{I} \quad ?$$

## PROBLEM 6.3

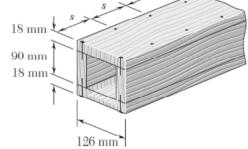
6.3 A square box beam is made of two  $18 \times 90$  mm planks and two  $18 \times 126$  mm planks nailed together as shown. Knowing that the spacing between nails is  $s = 30$  mm and that the vertical shear in the beam is  $V = 1100$  N, determine (a) the shearing force in each nail, (b) the maximum shearing stress in the beam.

$$y_c = \frac{45 + 18}{2} = 54 \text{ mm}$$

$$\Rightarrow A = 18 \times 126 \text{ mm}^2$$

$$Q = 18 \times 126 \times 54 \text{ mm}^3$$

$$I = \frac{1}{12} (126 \times 126^3 - 90 \times 90^3) \text{ mm}^4$$



$$\text{Area to be supported} = 18 \times 126 \text{ mm}^2$$



Supporting width  
= 18 mm each

$$\tau_v = \frac{VQ}{I}$$

$$= \frac{1100 \times 18 \times 126 \times 54}{12(126^4 - 90^4)} \text{ N/mm}$$

a) Shearing force per pair =  $q_v \times s$

$$\text{Shearing force per nail} = \frac{12 \times (1100 \times 18 \times 126 \times 54)}{(126^4 - 90^4)} \times \frac{20}{2} \text{ N} \quad s = 30 \text{ mm}$$

b) Shear stress =  $\left( \frac{12 \times 1100 \times 18 \times 126 \times 54}{126^4 - 90^4} \right) / (2 \times 18) \text{ N/mm}^2$        $t = 2 \times 18 \text{ mm}$   
 at the plane of the nails.  
 (just below the junctions)

Maximum shear is at centroid . . .

$$63 \uparrow \quad A_m = 126 \times 18 + (18 \times 45) \times 2$$

$$Q_{cm} = A_m \times y_{cm}$$

$$C = \frac{V_m \times Q_m}{It}$$

$$y_{cm} = \frac{126 \times 18 \times 54 + 2 \times 18 \times 45 \times 22.5}{126 \times 18 + 2 \times 18 \times 45}$$

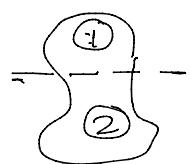
$$, t = 2 \times 18, V_m = 1100$$

$$I = \frac{1}{12} (126^4 - 90^4)$$

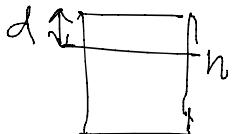
just above

just below

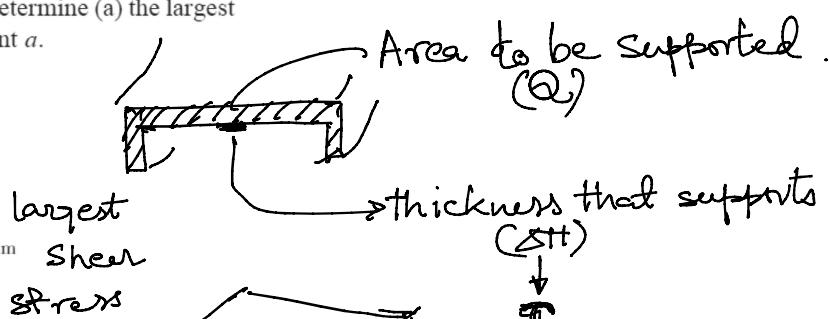
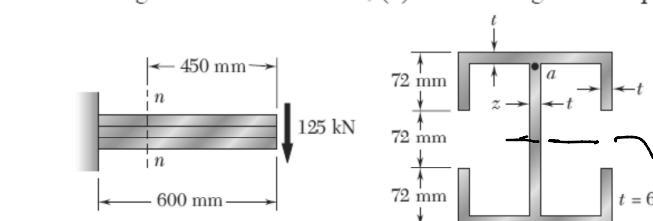
$$A_1 y_1 + A_2 y_2 = 0$$



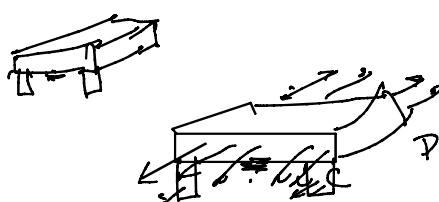
b



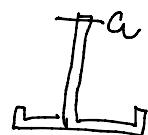
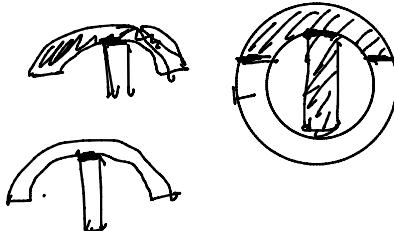
For the beam and loading shown, consider section  $n-n$  and determine (a) the largest shearing stress in that section, (b) the shearing stress at point  $a$ .



largest shear stress

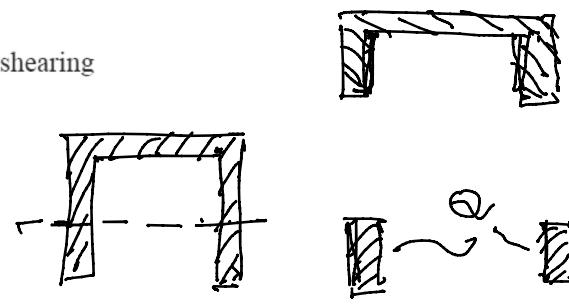
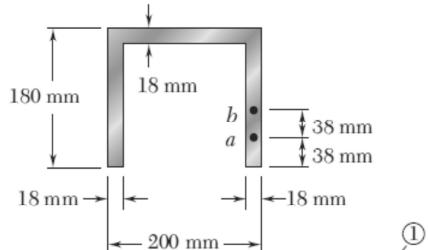
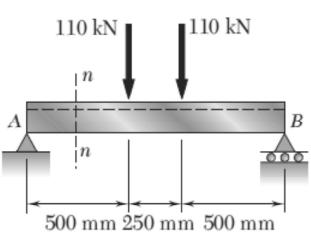


$$\Delta H = \int \sigma_c dA - \int \sigma_b dA$$

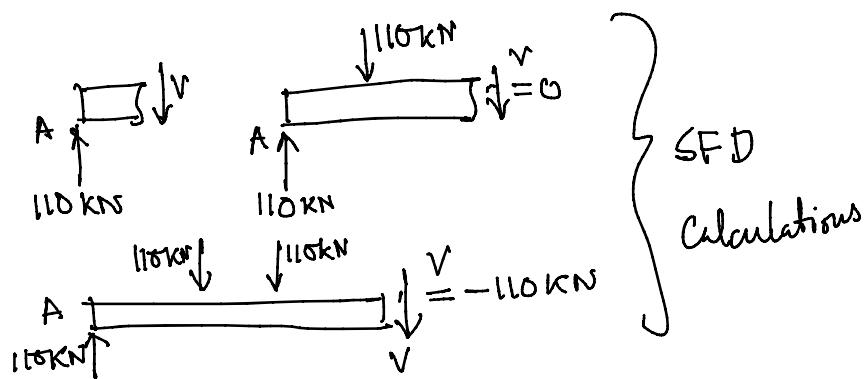


$$\Delta H = 72 + \frac{72}{2} = 108$$

For the beam and loading shown, consider section  $n-n$  and determine the shearing stress at (a) point  $a$ , (b) point  $b$ .



$$V = 110 \text{ kN}$$

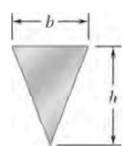


$$\tau = \frac{VQ}{It}$$

A beam having the cross section shown is subjected to a vertical shear  $V$ . Determine (a) the horizontal line along which the shearing stress is maximum, (b) the constant  $k$  in the following expression for the maximum shearing stress

$$\tau_{\max} = k \frac{V}{A}$$

where  $A$  is the cross-sectional area of the beam.

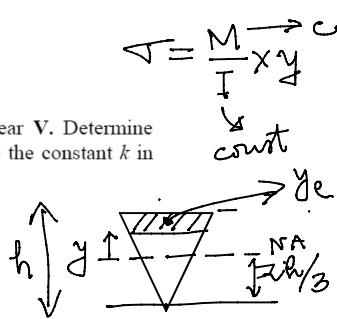


A beam having the cross section shown is subjected to a vertical shear  $V$ . Determine  
(a) the horizontal line along which the shearing stress is maximum, (b) the constant  $k$  in  
the following expression for the maximum shearing stress

$$\tau_{\max} = k \frac{V}{A}$$

where  $A$  is the cross-sectional area of the beam.

$$\begin{aligned} \chi & \left( bh - b \frac{h^2 - 2ah + a^2}{h} \right) \\ &= \frac{1}{3}bh^2 - b \times \frac{h^2 - 2ah + a^2}{h} \times \frac{h-2a}{3} \\ &= \frac{b}{3h} \left[ \cancel{h^3} - \cancel{h^2} + 2ah^2 - a^2h + 2ah^2 - 4a^2h + 2a^3 \right] \\ &= \frac{b}{3h} [4ah^2 - 5a^2h + 2a^3] \\ \frac{1}{h} \chi & \left( \cancel{h^2} - \cancel{h^2} + 2abh - a^2b \right) = \frac{b}{3}h [4ah^2 - 5a^2h + 2a^3] \\ \chi &= \frac{\frac{1}{3} (4ah^2 - 5a^2h + 2a^3)}{h (2ah - a^2)} \end{aligned}$$

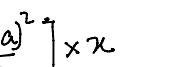
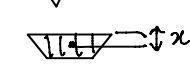
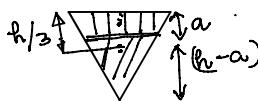
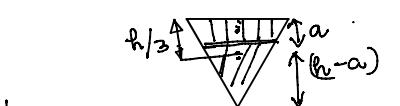


Frr shear  
 $Q$  is  
maximum at  
centroid.  
But  $t$  is at there

$$A = bh - \frac{b(h-a)}{h}$$

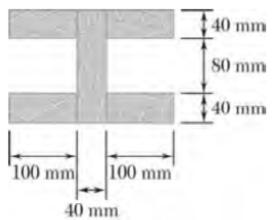
$$Q = y_c \times A$$

$$C = \frac{VQ}{IT} \rightarrow \text{max}$$

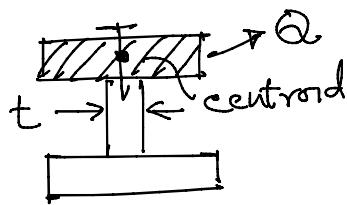


$$h = a + y + \frac{2h}{3} \Rightarrow a = -y + \frac{h}{3}$$

$$y_c = (h - a) - \frac{2h}{3}$$

**PROBLEM 6.29**

The built-up beam shown is made by gluing together five planks. Knowing that in the glued joints the average allowable shearing stress is 350 kPa, determine the largest permissible vertical shear in the beam.



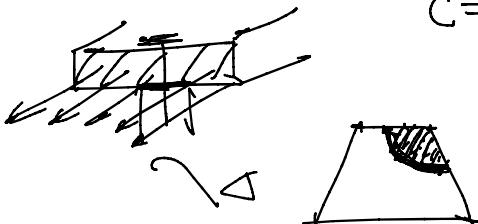
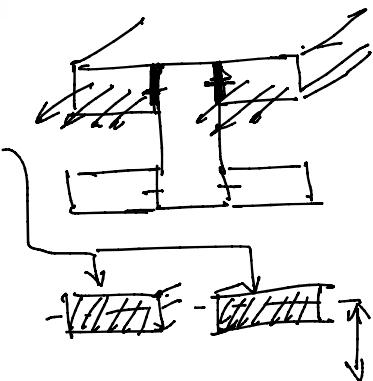
$$\bar{C} = \frac{VQ}{It}$$

$$Q = (40 \times 100 \times 2) \times 60$$

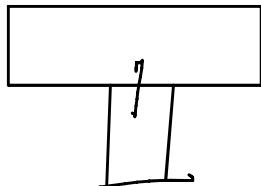
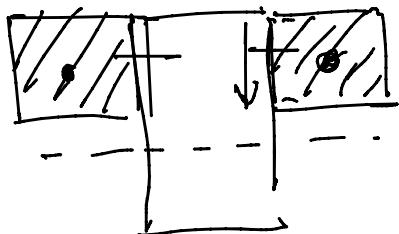
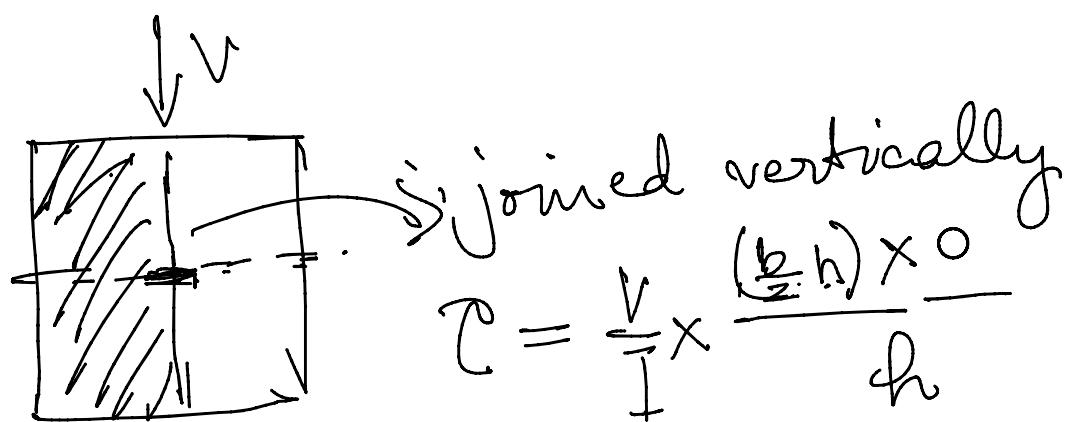
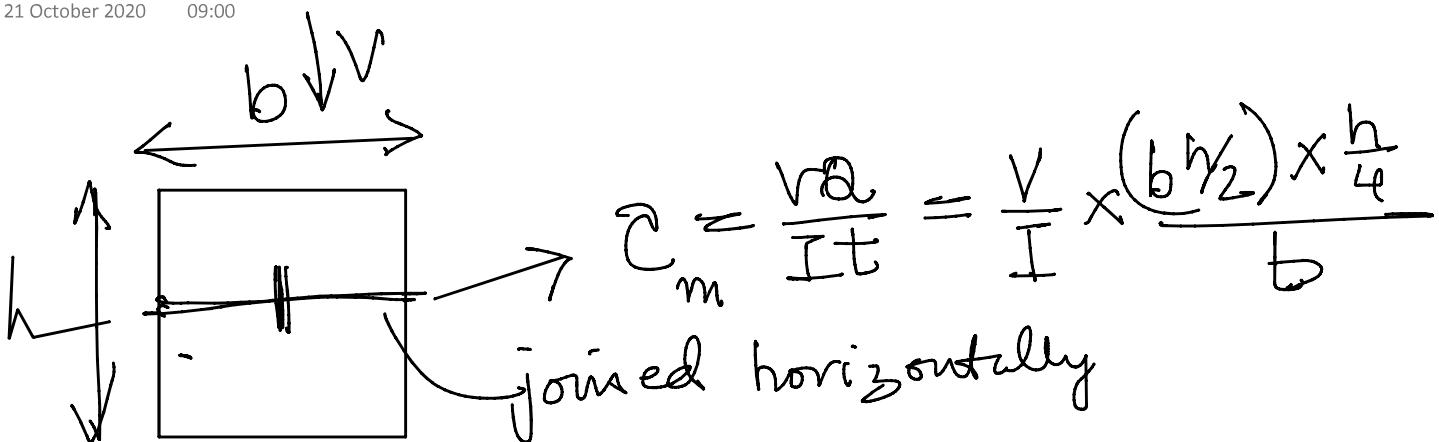
$$t = 40 + 40$$

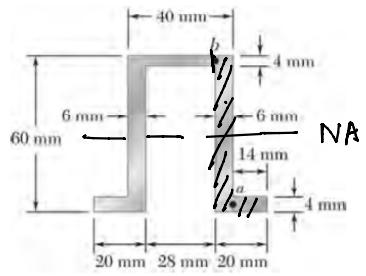
$$q_f = \frac{VQ}{I}$$

$$\bar{C} = \frac{V \times 48 \times 10^4 \text{ mm}^3}{I \times 80 \text{ mm}} = 350 \text{ kPa} \Rightarrow V = \frac{350 \text{ kPa} \times I \times 80 \text{ mm}}{48 \times 10^4 \text{ mm}^3}$$



$$\bar{y} = 40 + 20 = 60$$



**PROBLEM 6.36**

Knowing that a given vertical shear  $V$  causes a maximum shearing stress of 75 MPa in the hat-shaped extrusion shown, determine the corresponding shearing stress (a) at point  $a$ , (b) at point  $b$ .

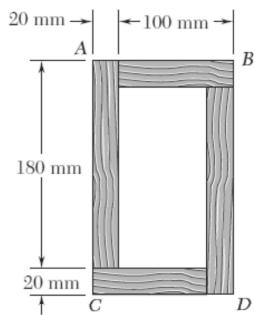
At  $a \rightarrow$

$$A = 4 \times (20 - 6)$$

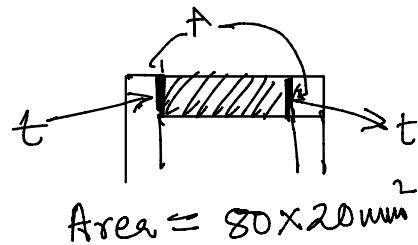
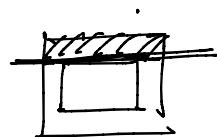


$$A = (6 \times 60 + 14 \times 4) \text{ mm}^2$$

$$t = 4 \text{ m}$$



Two  $20 \times 100$  mm and two  $20 \times 180$  mm boards are glued together as shown to form a  $120 \times 200$  mm box beam. Knowing that the beam is subjected to a vertical shear of  $3.5$  kN, determine the average shearing stress in the glued joint (a) at  $A$  (b) at  $B$ .

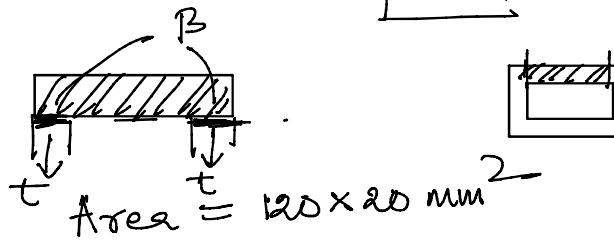


$$\text{Area} = 80 \times 20 \text{ mm}^2$$

$$\bar{y} = 90 \text{ mm}$$

$$Q = (80 \times 20) \times 90$$

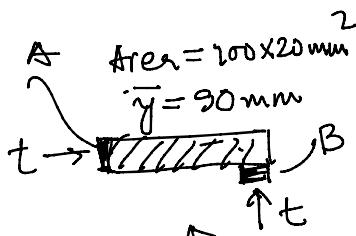
$$t = 20 + 20$$



$$\bar{y} = 90 \text{ mm}$$

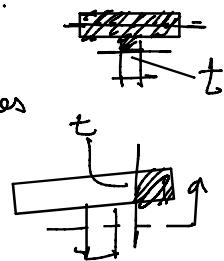
$$Q = (120 \times 20 \times 90)$$

$$t = 20 + 20$$



What is the average stress for joints A & B taken together

$$Q = (100 \times 20) \times 90 \quad t = 20 + 20 = 40$$



- 7-35. The boards are glued together to form the built-up beam. If the wood has an allowable shear stress of  $\tau_{allow} = 3 \text{ MPa}$ , and the glue seam at D can withstand a maximum shear stress of 1.5 MPa, determine the maximum allowable shear  $V$  that can be developed in the beam.

$$\tau_{glue} = \frac{V_1 \times 25 \times 10^4}{I \times 50} Q_1$$

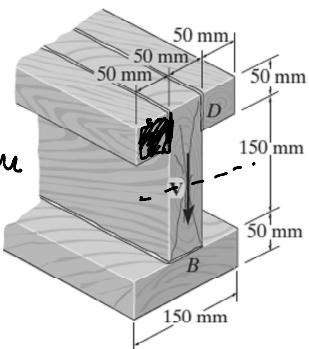
↓  
1.5 MPa.

$$\bar{y}_1 = 75 + \frac{50}{2} = 100 \text{ mm}$$

$$A_1 = 50 \times 50 = 2500 \text{ mm}^2$$

$$Q_1 = 2500 \times 100$$

$$= 25 \times 10^4 \text{ mm}^3$$



$$V_1 = 1.5 \text{ MPa} \times I \times \frac{50 \text{ mm}}{25 \times 10^4 \text{ mm}^3}$$

$$I = \left( \frac{1}{12} \times 150 \times 50^3 + 150 \times 50 \times 150^2 \right) \times 2 + \frac{1}{12} \times 50 \times 150^3$$

$\tau_w$  ~~MPa~~  $= \frac{V_2 \times Q_2}{I \times 50}$

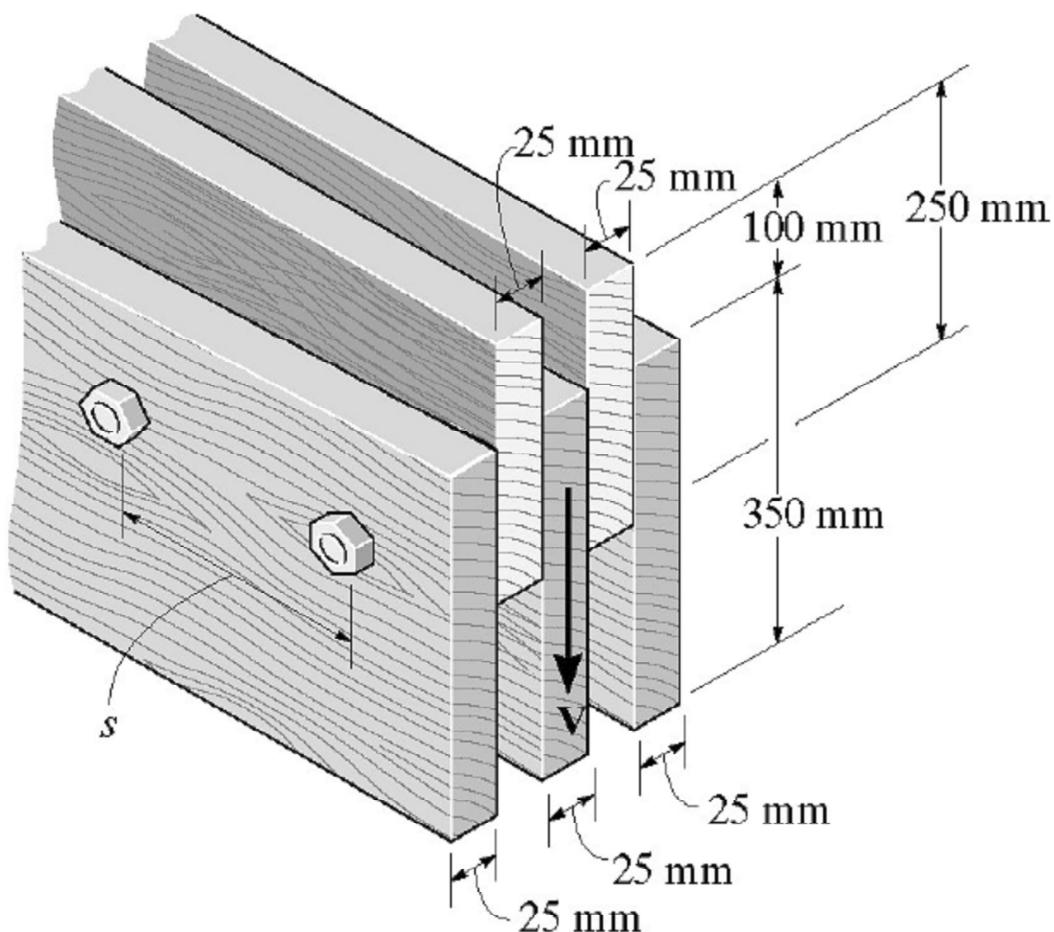
$$V_2 = \frac{I \times 50 \text{ mm} \times 3 \text{ MPa}}{Q_2}$$

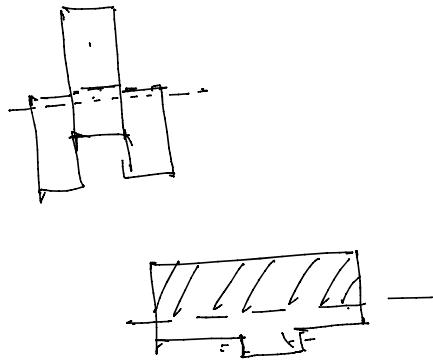
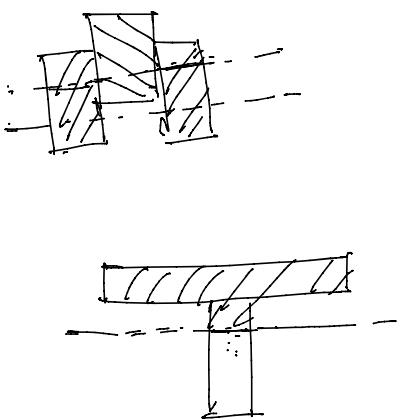
$\tau_2 = 50$

$$Q_2 = \frac{50 \times 150 + 75 \times 50}{50 \times 150 + 75 \times 50 + 50 \times 150 \times 150 + 75 \times 50 \times 27.5}$$

$$\text{Min of } V_1, V_2 \quad Q_2 = A_2 \times \bar{y}_2$$

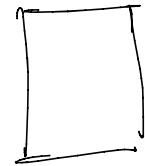
**7-38.** A beam is constructed from five boards bolted together as shown. Determine the maximum shear force developed in each bolt if the bolts are spaced  $s = 250$  mm apart and the applied shear is  $V = 35$  kN.

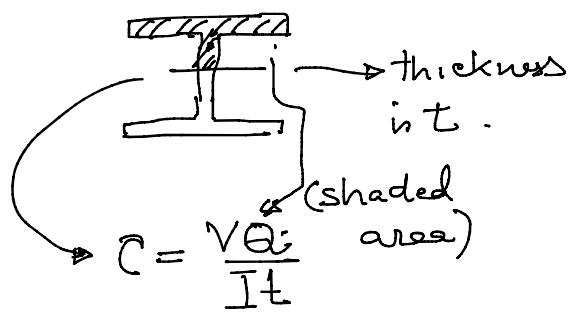
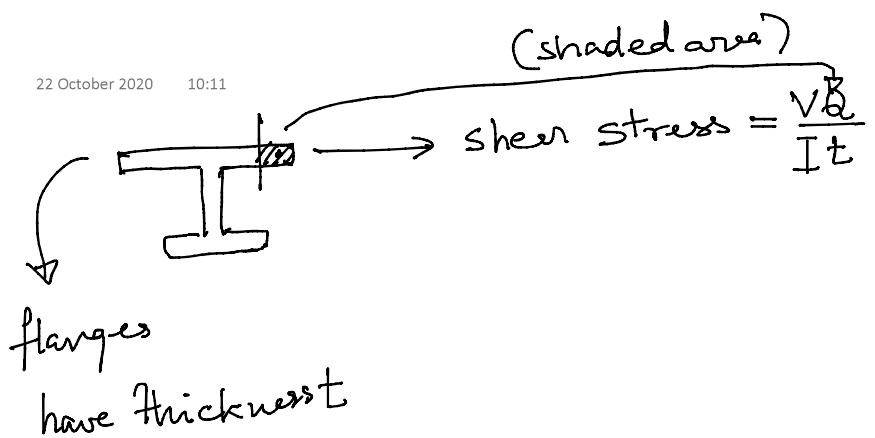


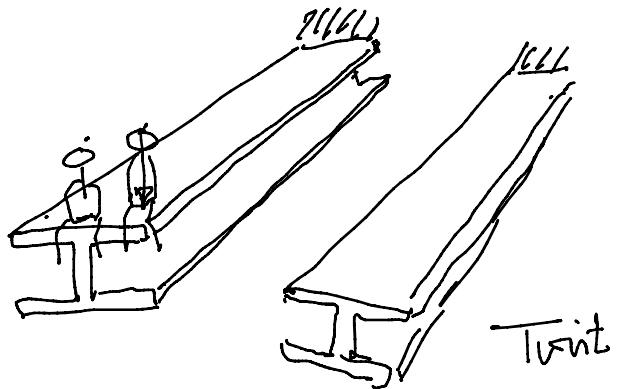


$$\Delta = \frac{M}{I} y$$

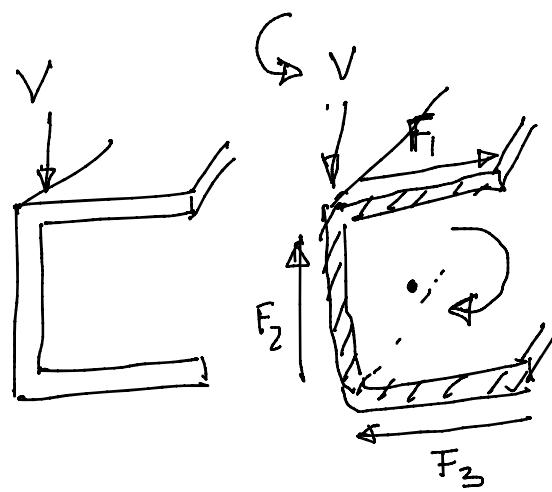
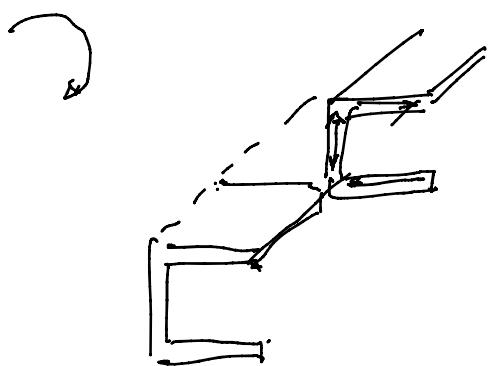
$$C = \frac{V}{I} \left( \frac{Q}{t} \right)_{\max}$$

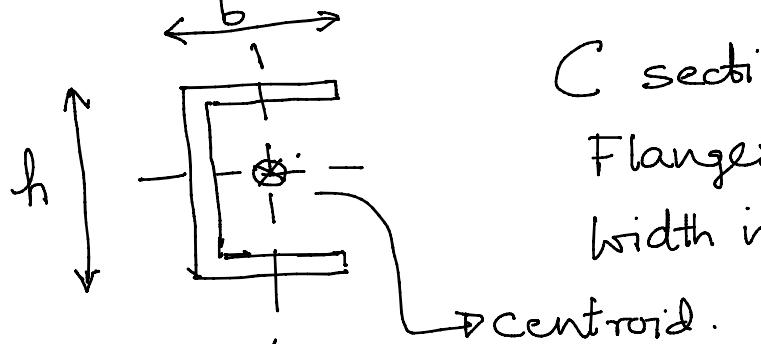






How to avoid  
twisting a beam?

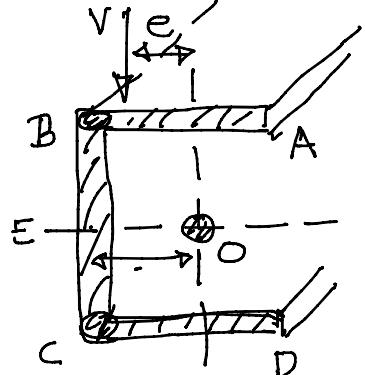




C section

Flanges have thickness  $t$   
width is  $b$ , height is  $h$ .

Centroid.



$$I \approx \left( \frac{1}{12} bt^3 + bt \times \left(\frac{h}{2}\right)^2 \right) \times 2 + \frac{1}{12} bt^3$$

$\downarrow$   
 $AB, CD$

$\downarrow$   
 $BC$ .

Location (horizontal of centroid) from BC.

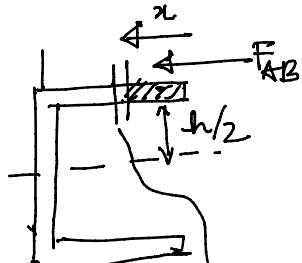
$$OE \rightarrow \frac{(bt) \times \frac{b}{2} \times d}{bt \times 2 + ht} = \frac{b^2}{2b + h}$$

$$A(x) = x \times t, \bar{y}(x) = \frac{h}{2} + \frac{t}{2} \approx \frac{h}{2}$$

$$Q(x)$$

$$= (t-x) \frac{h}{2} = \frac{1}{2} ht x.$$

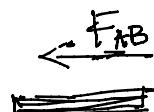
$$\text{Part AB: } C(x) = \frac{VQ}{It}$$



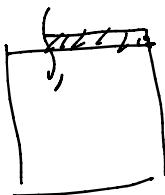
$$= \frac{V}{It} \times \frac{htx}{2} =$$

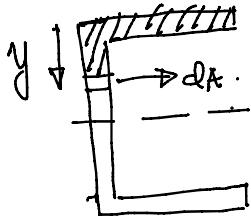
What is the force in AB?

$$dF_{AB} = C(x) dA = \frac{Vht}{2I} \times t dx.$$



$$F_{AB} = \int_0^b dF_{AB} = \int_0^b \frac{Vht}{2I} t dx = \frac{Vht}{2} \frac{b^2}{2} = \frac{1}{4} Vhtb^2$$





$$Q(y) = (bt + yt)\bar{y}$$

$$= \frac{1}{2}bth + \frac{1}{2}y^2h - \frac{1}{2}y^2t.$$

$$\dot{C}(y) = \frac{V}{2It} (bt^2h + t^2hy - ty^2).$$

$$A(y) = bt + yt$$

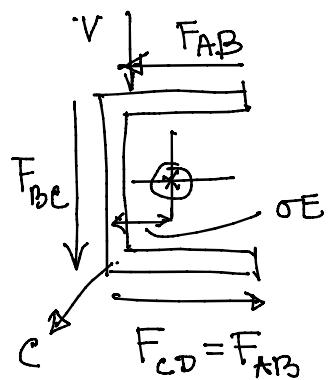
$$\bar{y} = \frac{bt \times \frac{h}{2} + yt \times (\frac{h}{2} - \frac{y}{2})}{bt + yt}.$$

$$dF_{BC} = \dot{C}(y) dA = \frac{V}{2It} t(bh + hy - y^2) t dy$$

$$F_{BC} = \int_B^C dF_{BC} = \int_{y=0}^{y=h} \frac{1}{2} \frac{Vt}{I} (bh + hy - y^2) dy$$

$$= \frac{Vt}{2I} \left[ bhx + \frac{hy^2}{2} - \frac{y^3}{3} \right]_0^h = \frac{Vt}{2I} \left( bh^2 + \frac{1}{2}h^3 - \frac{h^3}{3} \right)$$

$$= \frac{Vt}{2I} \left( bh^2 + \frac{h^3}{6} \right) = \frac{Vt h^2}{2I} \left( b + \frac{h}{6} \right)$$



$$F_{AB} = \frac{1}{4I} V h t b^2$$

$$F_{BC} = \frac{1}{2I} V h^2 t \left( b + \frac{h}{6} \right) = F_{CD}.$$

For no twist, moment about C = 0.

Let V be at distance d from segment BC.

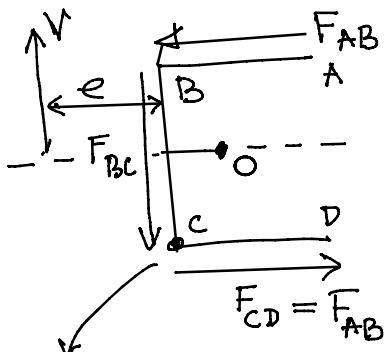
$$V_e = F_{AB} \times \frac{h}{2} + F_{CD} \times \frac{h}{2} + F_{BC} \times OE$$

$$\text{Then } Vd = F_{AB} \times h.$$

$$Xd = \frac{1}{4I} \sqrt{htb^2}$$

Shear center.

$$d = \frac{htb^2}{4I}$$



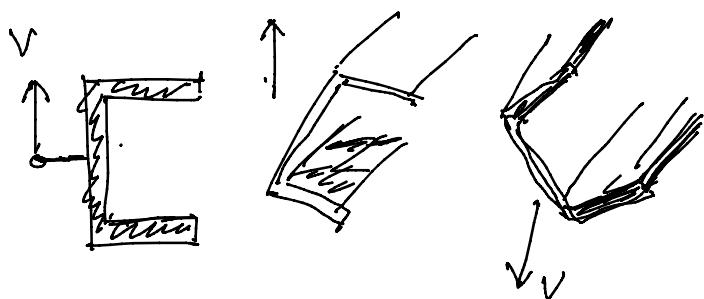
Take moments about C  
to avoid using  $F_{BC}, F_{CD}$

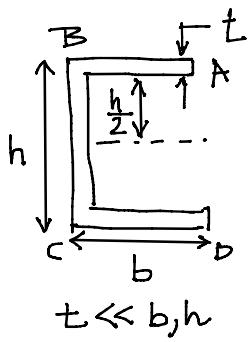
$$\sum \text{Forces} = 0$$

$$\sum \text{Moments} = 0$$

$$\Rightarrow Ve = hF_{AB} = h \frac{Vthb^2}{4I}$$

$$\Rightarrow e = \frac{b^2 h^2 t}{4I} = \frac{b^2 h^2}{4} \times \frac{3/2}{t/h(h+6b)} = \frac{3b^2}{6b+h}$$





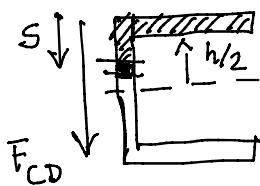
① For horizontal parts

$$\text{At } s \rightarrow 0: Q(s) = (ts) \times \left(\frac{h}{2} + \frac{t}{2}\right) \approx ts \frac{h}{2}$$

$$\text{At } s \rightarrow b: C(s) = \frac{VQ}{It} = \frac{Vts \frac{h}{2}}{It} = \frac{Vhs}{2I}$$

$$F_{AB} = \int_0^b C(s) ds = \int_0^b \frac{Vhs}{2I} \times ts ds = \frac{Vth}{4I} s^2 \Big|_0^b = \frac{Vthb^2}{4I}$$

② For vertical part

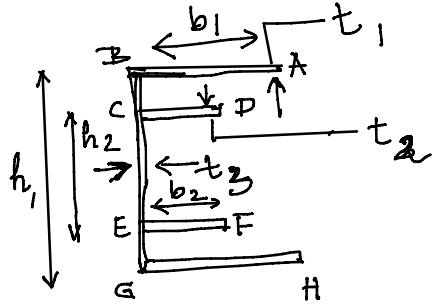


$$Q(s) = (tb) \times \frac{h}{2} + (ts) \times \left(\frac{h}{2} - \frac{s}{2}\right) = \frac{t}{2} [bh + sh - s^2]$$

$$C(s) = \frac{VQ}{It} = \frac{V}{It} \times \frac{1}{2} t (bh + sh - s^2)$$

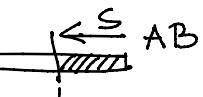
$$F_{CD} = \int_0^h \frac{V}{2I} (bh + sh - s^2) \times ts ds = \frac{Vh^2}{2I} \left(b + \frac{h}{c}\right) = V$$

$$I = \frac{th^3}{12} (b^2 + h^2)$$

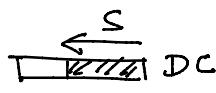


$$I = \frac{1}{12} t_1^3 h_1^3 + \frac{1}{2} b_1 t_1 h_1^2$$

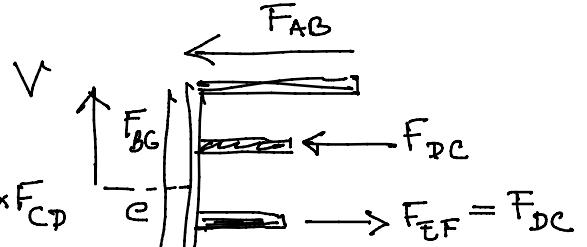
$$+ \frac{1}{2} b_2 t_2 h_2^2$$



$$F_{AB} = \frac{\sqrt{t_1 h_1 b_1}^2}{4I} = F_{GH}$$



$$F_{DC} = \frac{\sqrt{t_2 h_2 b_2}^2}{4I} = F_{EF}$$

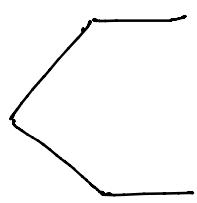


$$\sum M_G = 0$$

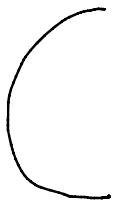
$$Ve = h_1 \times F_{AB} + \left( h_1 - \frac{h_1 - h_2}{2} \right) \times F_{CD} - \frac{h_1 - h_2}{2} \times F_{EF}$$

$$\Rightarrow X_C = h_1 \times \frac{\sqrt{t_1 h_1 b_1}^2}{4I} + \frac{h_1 + h_2}{2} \times \frac{\sqrt{t_2 h_2 b_2}^2}{4I} - \frac{h_1 - h_2}{2} \times \frac{\sqrt{t_2 h_2 b_2}^2}{4I}$$

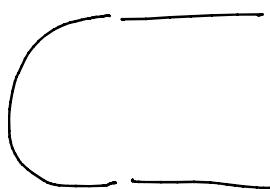
$$e = \frac{1}{4I} \left[ t_1 h_1^2 b_1^2 + \frac{t_2 h_2^2 b_2^2}{2} (h_1 + h_2 - h_1 + h_2) \right] = \frac{1}{4I} (t_1^2 h_1^2 b_1^2 + t_2^2 h_2^2 b_2^2)$$



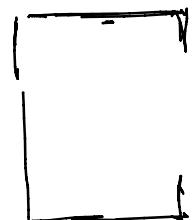
?



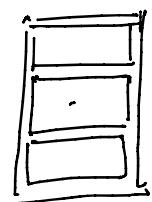
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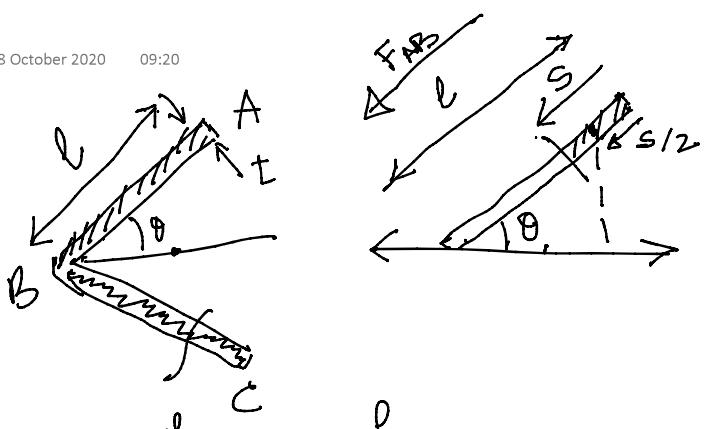


?



?





$$Q = A y_c$$

$$A = t \times s.$$

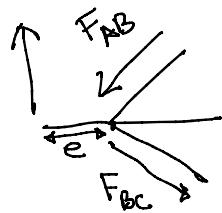
$$y_c = \left(l - \frac{s}{2}\right) \sin\theta.$$

$$Q(s) = ts \left(l - \frac{s}{2}\right) \sin\theta.$$

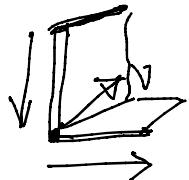
$$\begin{aligned} C(s) &= \frac{VQ}{It} = \frac{V}{It} ts \left(l - \frac{s}{2}\right) \sin\theta \\ &= \frac{V}{I} \sin\theta \left(ls - \frac{1}{2}s^2\right) \end{aligned}$$

$$F_{AB} = \int_0^l C ds = \int_0^l \frac{V}{I} \sin\theta \left(ls - \frac{s^2}{2}\right) ts ds$$

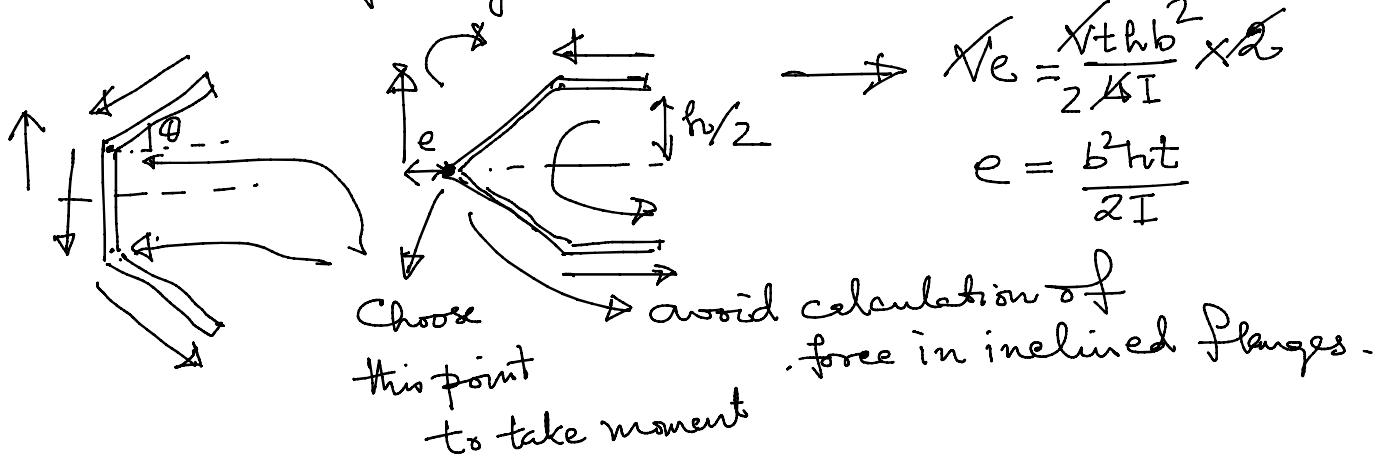
$$= \frac{Vt}{I} \sin\theta \left[\frac{ls^2}{2} - \frac{s^3}{6}\right]_0^l = \frac{Vts \sin\theta}{I} \times \frac{l^3}{6} = \frac{Vtl^3 \sin\theta}{6}$$

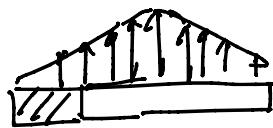
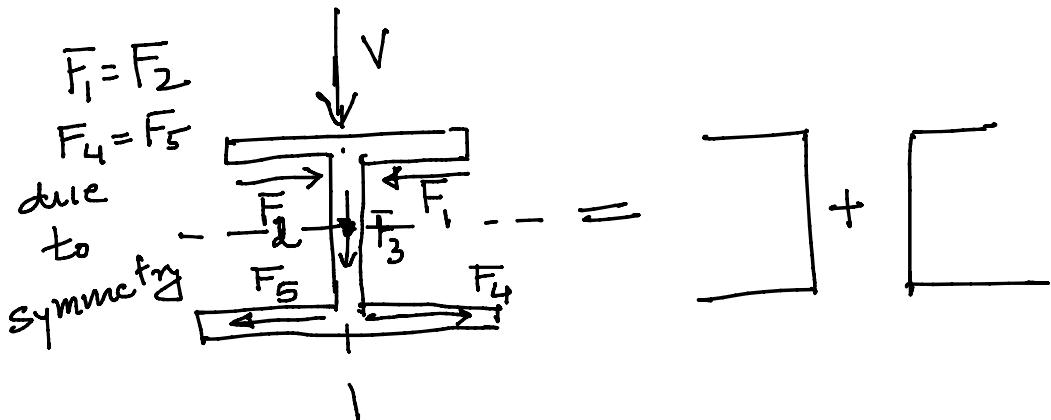


$$\sum M_C = 0 \Rightarrow Ve = 0$$

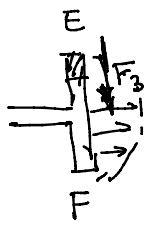
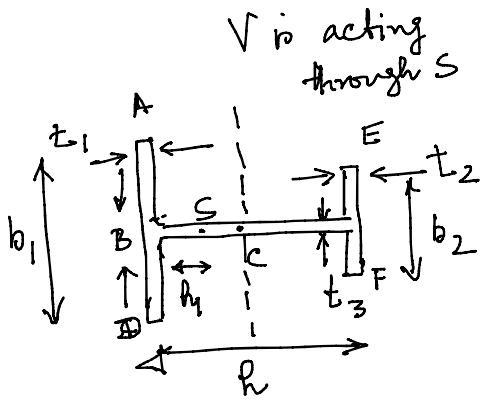


Although the shear center is intuitively located, the derivation is needed for shapes involving inclined flanges





Because of symmetry  
shear center passes through centroid



$$F_3 = \frac{V t_2 b_2^3}{24 I}$$

Diagram shows a rectangular cross-section with width  $b_1$  and height  $b_2$ . A coordinate system is centered at the centroid  $C$ , with  $s$  being the distance from the left edge to a fiber at distance  $s$  from the neutral axis.

Equations derived:

$$Q(s) = (t_1 s) \left( \frac{b_1 - s}{2} \right)$$

$$\bar{y}(s) = \frac{b_1 - s}{2}$$

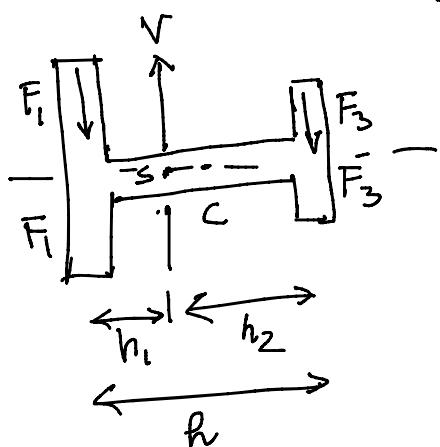
$$C(s) = \frac{V Q}{I t} = \frac{\frac{1}{2} (b_1 - s) t_1 s}{I t_1}$$

$$F_1 = \int_0^{b_1} C dA = \int_0^{b_1} \frac{V \times \frac{1}{2} t_1 (b_1 - s) s}{I t_1} \times \frac{b_2}{2} ds$$

$$= \frac{1}{2} \frac{V t_1}{I} \int_0^{b_1} (b_1 s - s^2) ds = \frac{V t_1}{2 I} \left( \frac{b_1 s^2}{2} - \frac{s^3}{3} \right) \Big|_0^{b_1}$$

$$= \frac{V t_1}{2 I} \left( b_1 \times \frac{b_1^2}{2} - \frac{b_1^3}{3} \right)$$

$$= \frac{V t_1}{2 I} \left( \frac{b_1^3}{2} - \frac{b_1^3}{3} \right) = \frac{V t_1 b_1^3}{12 I}$$



Taking moments about S

$$F_1 \times h_1 = F_3 \times h_2$$

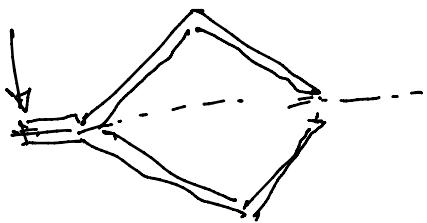
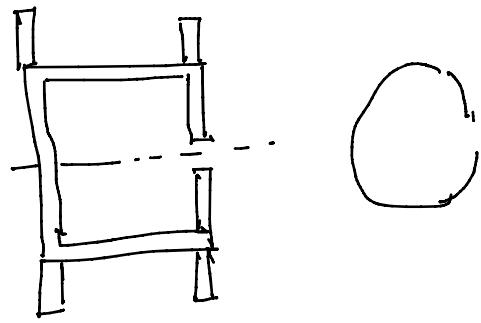
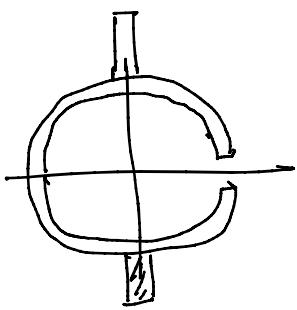
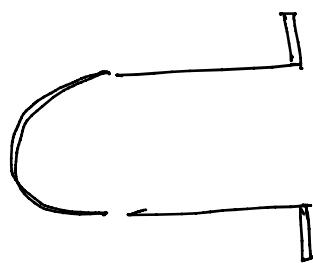
$$\frac{\sqrt{t_1 b_1^3}}{12 I} h_1 = \frac{\sqrt{t_2 b_2^3}}{12 I} h_2$$

$$\frac{h_1}{h_2} = \frac{t_2 b_2^3}{t_1 b_1^3}$$

$$h_1 + h_2 = h$$

$$I \approx \frac{1}{12} (b_1^3 t_1 + b_2^3 t_2)$$

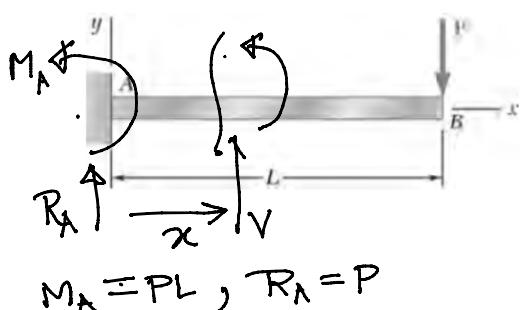
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## Section

$$M_A + Vx + M = 0, \quad V + R_A = 0$$

$$M = -Vx - M_A \quad V = -R_A = -Px \\ = Px - PL.$$



$$U = \int_0^L \frac{P^2(x-L)^2}{2EI} dx = \int_0^L \frac{M(x)^2}{2EI} dx$$

① Find deflection at B. (at point of application of force)

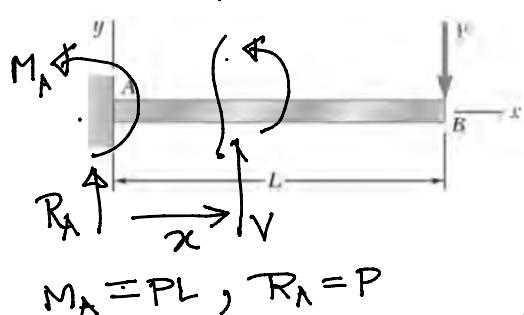
$$\Delta_B = \frac{\partial U}{\partial P}$$

Two ways → Integrate & differentiate wrt P  
→ Differentiate under integral sign & integrate.

## Section

$$M_A + Vx + M = 0, \quad V + R_A = 0$$

$$M = -Vx - M_A \quad V = -R_A = -P_x \\ = P_x - PL$$



$$M_A = PL, \quad R_A = P$$

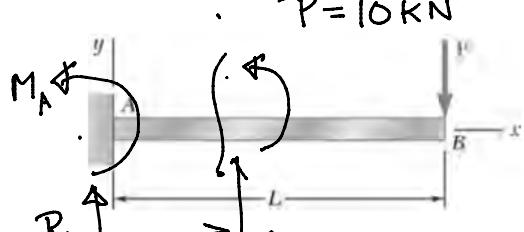
$$U = \frac{P^2}{2EI} \left[ \frac{(x-L)^3}{3} \right] \Big|_0^L = +\frac{P^2}{6EI} L^3 \quad , \quad \frac{\partial U}{\partial P} = \cancel{\frac{2PL^3}{3}} = \frac{PL^3}{3EI} = \Delta_B$$

$$U = \int_0^L \frac{P^2(x-L)^2}{2EI} dx = \int_0^L \frac{M(x)^2}{2EI} dx$$

(e)

2.

## Section



$$M_A = PL, R_A = P$$

$$\frac{\partial U}{\partial P} = \frac{1}{2EI} \frac{\partial}{\partial P} \int_0^L M(x)^2 dx = \frac{1}{2EI} \int_0^L \frac{\partial}{\partial P} (M^2) dx = \frac{1}{2EI} \int_0^L 2M \frac{\partial M}{\partial P} dx$$

$$\Delta_B = \frac{1}{EI} \int_0^L M \frac{\partial M}{\partial P} dx = \frac{1}{EI} \int_0^L P(x-L) \times (x-L) dx = \frac{P}{EI} \int_0^L (x-L)^2 dx$$

$$= \frac{PL^3}{3EI} = 10 \times \frac{L^3}{EI}$$

$$M_A + Vx + M = 0, V + R_A = 0$$

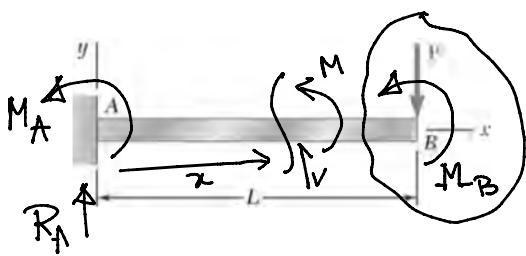
$$M = -Vx - M_A \quad V = -R_A = -Px$$

$$= Px - PL, \frac{\partial M}{\partial P} = (x - L)$$

$$U = \int_0^L \frac{P^2(x-L)^2}{2EI} dx = \int_0^L \frac{M(x)^2}{2EI} dx$$

(e)

2.



$$\frac{1}{2}M\theta = \int \frac{M^2}{2EI} dx$$

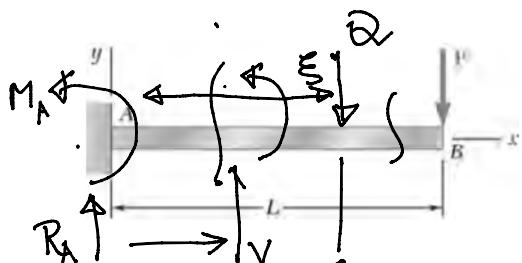
$$\begin{aligned}
 +M_B &= PL, \quad R_A = P \\
 V &= -R_A = -P \quad \rightarrow M_A = PL - M_B \\
 M &= -M_A - Vx = -PL + M_B + Px \\
 \frac{\partial M}{\partial P} &= -L + x, \quad \frac{\partial M}{\partial M_B} = 1
 \end{aligned}$$

Find the slope at B

$$\begin{aligned}
 \Theta_B &= \frac{\partial U}{\partial M_B} = \frac{\partial}{\partial M_B} \int_0^L \frac{M^2 dx}{2EI} = \int_0^L \frac{M}{EI} \frac{\partial M}{\partial M_B} dx \\
 &= \frac{1}{EI} \int_0^L (-PL + M_B + Px) 1 dx
 \end{aligned}$$

$$\begin{aligned}
 \text{Put } M_B = 0 \quad \Theta_B &= \frac{1}{EI} \int_0^L P(x-L) dx = \frac{P}{EI} \frac{(x-L)^2}{2} \Big|_0^L = \frac{P}{EI} \left( -\frac{L^2}{2} \right)
 \end{aligned}$$

Energy  $\rightarrow F(GF, GD)$   
 $F, x$   
 $M, \theta, T, \phi$



$$M_A = PL + Q\xi$$

$$\therefore R_A = P + Q$$

$$\Delta_C = \frac{1}{EI} \int_0^L M \frac{\partial M}{\partial \alpha} dx = \frac{1}{EI} \int_0^\xi M \cdot \frac{\partial M}{\partial \alpha} dx + \frac{1}{EI} \int_\xi^L M \cdot \frac{\partial M}{\partial \alpha} dx \\ = \frac{1}{EI} \int_0^\xi [-(PL + Q\xi) + (P + Q)x] Q(x - \xi) dx.$$

$\Delta_C = f(\xi) \rightarrow v(\xi)$ : deflection

$$AC: V = -R_A = -(P+Q)$$

$$M = -M_A - Vx = -(PL + Q\xi) + (P+Q)x$$

$$\frac{\partial M}{\partial Q} = Q(x - \xi)$$

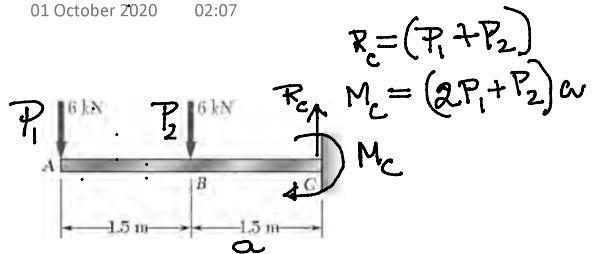
$$CB: V = -R_A + Q = -P$$

$$M = -M_A - Vx + Q\xi = -PL + Px$$

$$\frac{\partial M}{\partial Q} = 0$$

)

2.



$$\frac{\partial M}{\partial P_1} = -x \quad V = P_1 \quad M = -Vx = -P_1x \quad \frac{\partial M}{\partial P_2} = 0$$

$$\frac{\partial M}{\partial P_1} = -x, \quad V = P_1 + P_2 \quad M = -Vx \neq P_2a$$

$$= -P_1x - P_2x + P_2a.$$

$$S_A = \frac{\partial U}{\partial P_1} = \frac{\partial}{\partial P_1} \int_{0}^{2a} \frac{M^2}{2EI} dx = \frac{1}{EI} \int_{0}^{2a} M \frac{\partial M}{\partial P_1} dx$$

$$EI S_A = \int_{0}^{2a} (-P_1x)(-x) dx + \int_{2a}^{2a} (-P_1x - P_2x + P_2a)(-x) dx$$

$$= \int_{0}^{2a} P_1x^2 dx + \int_{2a}^{2a} (P_1x^2 + P_2x^2 - P_2ax) dx$$

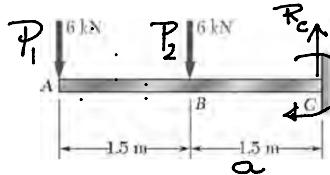
$$= \frac{P_1a^3}{3} + \left( \frac{P_1x^3}{3} + \frac{P_2x^3}{3} - \frac{P_2ax^2}{2} \right) \Big|_{0}^{2a}$$

$$= \frac{P_1a^3}{3} + \frac{7P_1a^3}{3} + \frac{7P_2a^3}{3} - \frac{3P_2a^3}{2}$$

$$S_A \rightarrow P_1 = 6 \text{ kN}$$

when  $P_1 = 0$

$$EI S_A = \frac{7P_2a^3}{3} - \frac{3P_2a^3}{2} \rightarrow \text{deflection of overhang}$$



$$R_c = (P_1 + P_2)$$

$$M_c = (2P_1 + P_2)a$$

$$\delta_B = \frac{\partial}{\partial P_2} \int_0^{2a} \frac{M^2}{2EI} dx = \frac{1}{EI} \int_0^{2a} M \frac{\partial M}{\partial P_2} dx$$

$$EI\delta_B = \int_0^a (-P_1x) 0 dx + \int_a^{2a} (-P_1x - P_2x + P_2a)(-x+a) dx$$

$$= \int_a^{2a} [P_1x^2 + P_2x^2 - P_2ax - P_1ax - P_2ax + P_2a^2] dx$$

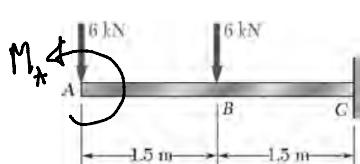
$$= \left[ \frac{P_1x^3}{3} + \frac{P_2x^3}{3} - \frac{P_1ax^2}{2} - \frac{P_1ax^2}{2} + P_2a^2x \right]_a^{2a}$$

$$\begin{aligned} P_1 &\downarrow & V = P_1 & \frac{\partial M}{\partial P_1} = -x \\ \text{Free Body Diagram} & & M = -Vx = -P_1x & \frac{\partial M}{\partial P_2} = 0 \\ P_1 &\downarrow & V = P_1 + P_2 & \\ \text{Free Body Diagram} & & M = -Vx - P_2a & \\ && = -P_1x - P_2x + P_2a. & \\ \frac{\partial M}{\partial P_1} &= -x, \quad \frac{\partial M}{\partial P_2} &= -x+a \end{aligned}$$

Suppose we wish to find deflection at B when  $P_2 = 0$  i.e. no  $P_2$

$$EI\delta_B = \left[ \frac{P_1x^3}{3} - \frac{P_1ax^2}{2} \right]_a^{2a} \text{ when } P_2 = 0$$

$$\theta_A = \frac{\partial V}{\partial M_A} = \int_0^{2a} \frac{M^2}{2EI} dx = \frac{1}{EI} \int_0^{2a} M \cdot \frac{\partial M}{\partial M_A} dx$$



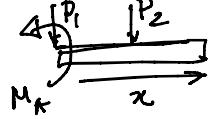
$$0 < x < a, \quad a < x < 2a$$

$$M = -P_1x - M_A$$



$$\frac{\partial M}{\partial M_A} = -1$$

$$M = -P_1x - P_2x + P_2a - M_A$$



$$\frac{\partial M}{\partial M_A} = -1$$

$$EI\theta_A = \int_0^a (-P_1x - M_A)(-1)dx + \int_a^{2a} (-P_1x - P_2x + P_2a - M_A)(-1)dx$$

$$= \left[ \frac{P_1x^2}{2} + M_Ax \right]_0^a + \left[ \frac{P_1x^2}{2} + \frac{P_2x^2}{2} - P_2ax + M_Ax \right]_a^{2a}$$

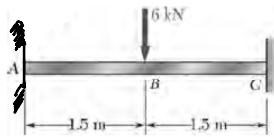
If there is no  $M_A$

$$EI\theta_A = \left[ \frac{P_1x^2}{2} \right]_0^a + \left[ \frac{P_1x^2}{2} + \frac{P_2x^2}{2} - P_2ax \right]_a^{2a}$$

$$= \frac{P_1a^2}{2} + \frac{P_1}{2}3a^2 + \frac{P_2}{2}3a^2 - P_2a^2$$

$$= \frac{5}{2}P_1a^2 + \frac{1}{2}P_2a^2$$

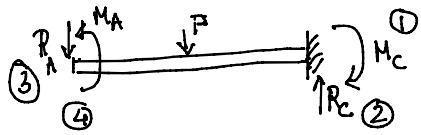
Double cantilever



$$R_A + P = R_c \quad M_c = (2R_A + P)a + M_A$$

$$EI\delta_A = \int_0^a (-R_A x - M_A)(-x) dx + \int_a^{2a} (-R_A x - P x + P a - M_A)(-x) dx$$

$$= \left[ \frac{R_A x^3}{3} + \frac{M_A x^2}{2} \right]_0^a + \left[ \frac{R_A x^3}{3} + \frac{P x^3}{3} - \frac{P a x^2}{2} + \frac{M_A x^2}{2} \right]_a^{2a}$$



$$0 < x < a$$

$$M = -R_A x - M_A, \quad \frac{\partial M}{\partial R_A} = -x$$

$$a < x < 2a$$

$$M = -R_A x - P x + P a - M_A, \quad \frac{\partial M}{\partial R_A} = -x$$

$$\frac{\partial M}{\partial M_A} = -1$$

By solving for  $R_A = 0, M_A = 0$   
we can obtain  $R_A, M_A$

$$\delta_A = 0 \Rightarrow 0 = \frac{R_A a^3}{3} + \frac{M_A a^2}{2} + \frac{7a^3 R_A}{3} + \frac{7a^3 P}{3} - \frac{3Pa^3}{2} + \frac{3}{2} M_A a^2$$

If at A there is a fixed joint there is no deflection

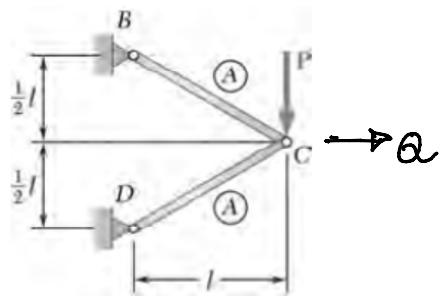
Also slope is 0

$$\therefore \theta_A = 0 \quad EI\theta_A = \int_0^a (-R_A x - M_A)(-1) dx + \int_a^{2a} (-R_A x - P x + P a - M_A)(-1) dx$$

$$= \left[ \frac{R_A x^2}{2} + M_A x \right]_0^a + \left[ \frac{R_A x^2}{2} + \frac{P x^2}{2} - P a x + M_A x \right]_a^{2a}$$

$$0 = \frac{R_A a^2}{2} + M_A a + \frac{3R_A a^2}{2} + \frac{3}{2} Pa^2 - Pa^2 + Ma^2$$

Find horizontal deflection at C due to P.

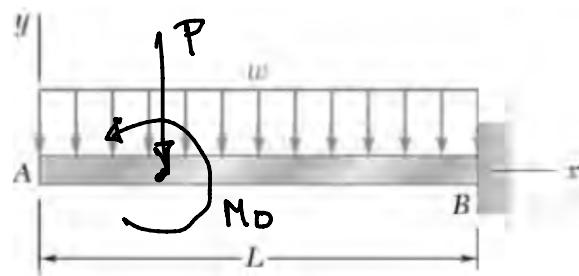


$$\delta_{C,V} = \frac{\partial U}{\partial P} \quad \text{vertically downward +ve}$$

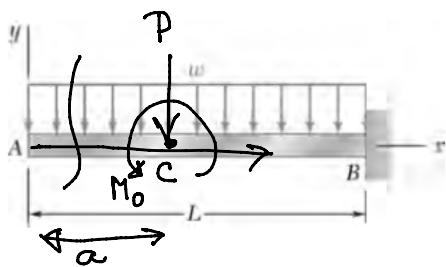
$$\delta_{C,H} = \left. \frac{\partial U}{\partial \alpha} \right|_{\alpha=0} \quad L \rightarrow R +ve.$$

$$U = \sum \frac{F^2 L}{2EA}$$

$$\frac{\partial U}{\partial \alpha} = \sum \frac{L}{EA} F \frac{\partial F}{\partial \alpha}$$



Find slope & deflection at C



AC.  $0 < x < a$

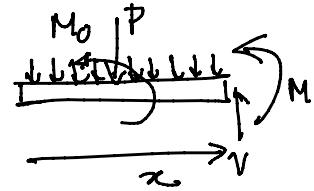
$$\begin{aligned} \text{Free Body Diagram: } & V = \int_0^x \omega d\xi = \omega x \\ & M = -Vx + \int_0^x (\omega d\xi) \xi \\ & = -Vx + \frac{\omega x^2}{2} \\ & = -\frac{\omega x^2}{2} \end{aligned}$$

$$\frac{\partial M}{\partial P} = 0, \quad \frac{\partial M}{\partial M_0} = 0$$

CB  $a < x < L$

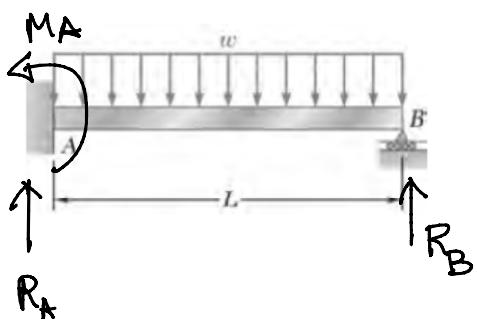
$$\begin{aligned} V &= \int_0^x \omega d\xi + P \\ &= \omega x + P \end{aligned}$$

$$\begin{aligned} M &= -Vx + \int_0^x \omega \xi d\xi + Pa - M_0 \\ &= -\omega x^2 - Px + \frac{\omega x^2}{2} + Pa - M_0 \\ &= -\frac{\omega x^2}{2} - Px + Pa - M_0 \\ \frac{\partial M}{\partial P} &= -x + a, \quad \frac{\partial M}{\partial M_0} = -1 \end{aligned}$$



$$U = \int_0^L \frac{M^2}{2EI} dx, \quad S_c = \int_0^L \frac{M}{EI} \frac{\partial M}{\partial P} dx, \quad \Theta_c = \int_0^L \frac{M}{EI} \frac{\partial M}{\partial M_0} dx$$

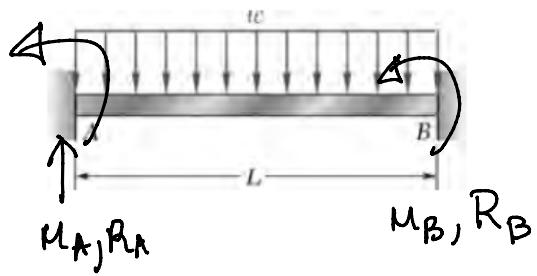
Find reactions



Replace constraint at B. with reaction  $R_B$ .

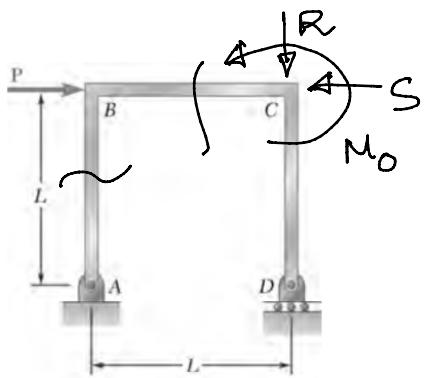
$$\text{Put } \delta_B = \int_0^L \frac{M}{EI} \frac{\partial M}{\partial R_B} dx = 0$$

Write  $M$  in terms of  $R_B$  only.  
 (No  $M_A$  or  $R_A$  should be present and  
 be replaced by  $R_B$ ).



$$\theta_B = 0, \delta_B = 0$$

Write M in terms of  $M_B$  &  $R_B$  only.

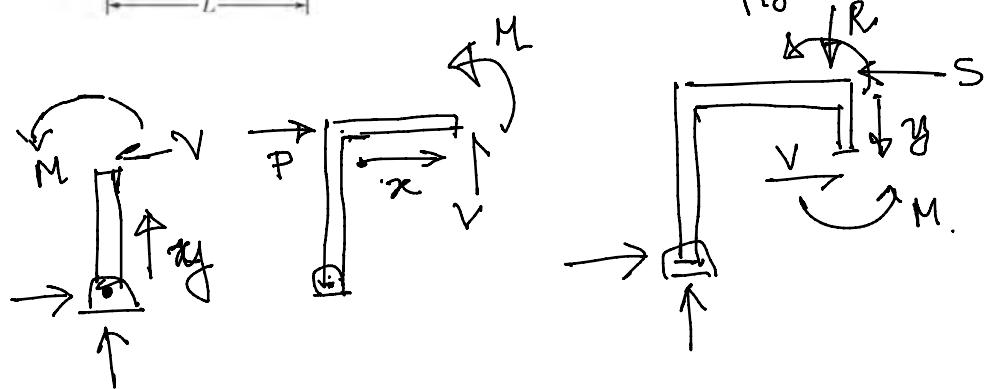


Lope & deflections at C?

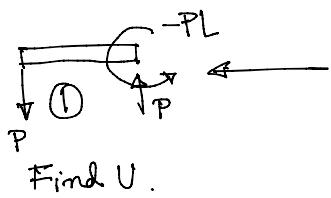
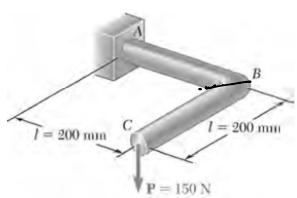
$$\delta_r = \int \frac{1}{EI} M \frac{\partial M}{\partial R} dx$$

$$\delta_H = \int \frac{1}{EI} M \cdot \frac{\partial M}{\partial S} dx$$

$$\theta_c = \int \frac{1}{EI} N \frac{\partial M}{\partial M_0} dx$$

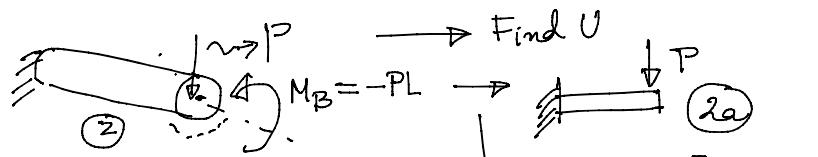


Put  $R, S, M_0 = 0$   
once derivatives  
are obtained.

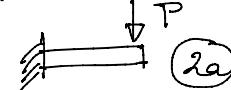


Find  $U_1$ .

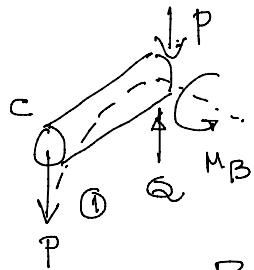
$$U_1 = \int_0^l \frac{M^2}{2EI} dx$$



Find  $U$

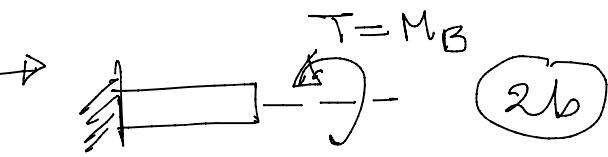


$$U_{2a} = \int \frac{M^2}{2EI} dx$$



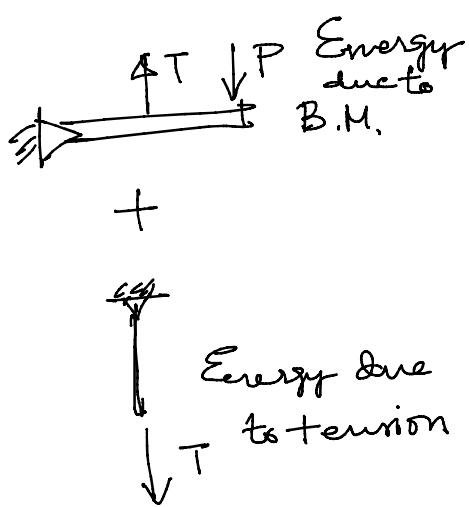
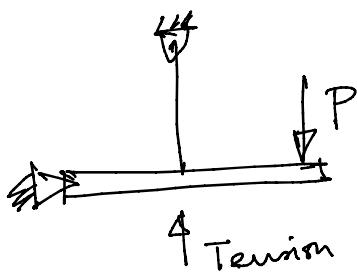
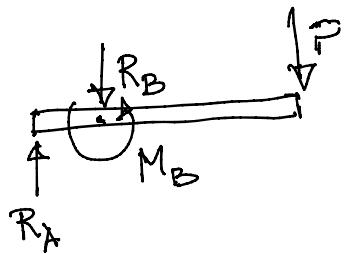
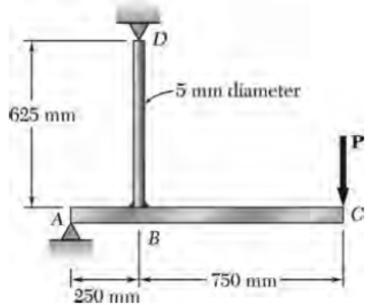
$$Q = P$$

$$M_B = -PL$$



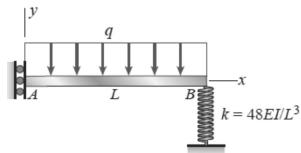
(2b)

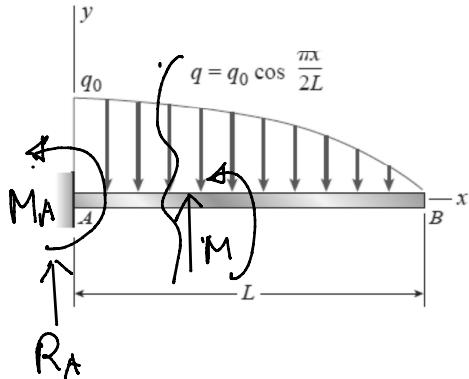
$$U_{2b} = \int \frac{T^2}{2GJ} dx$$



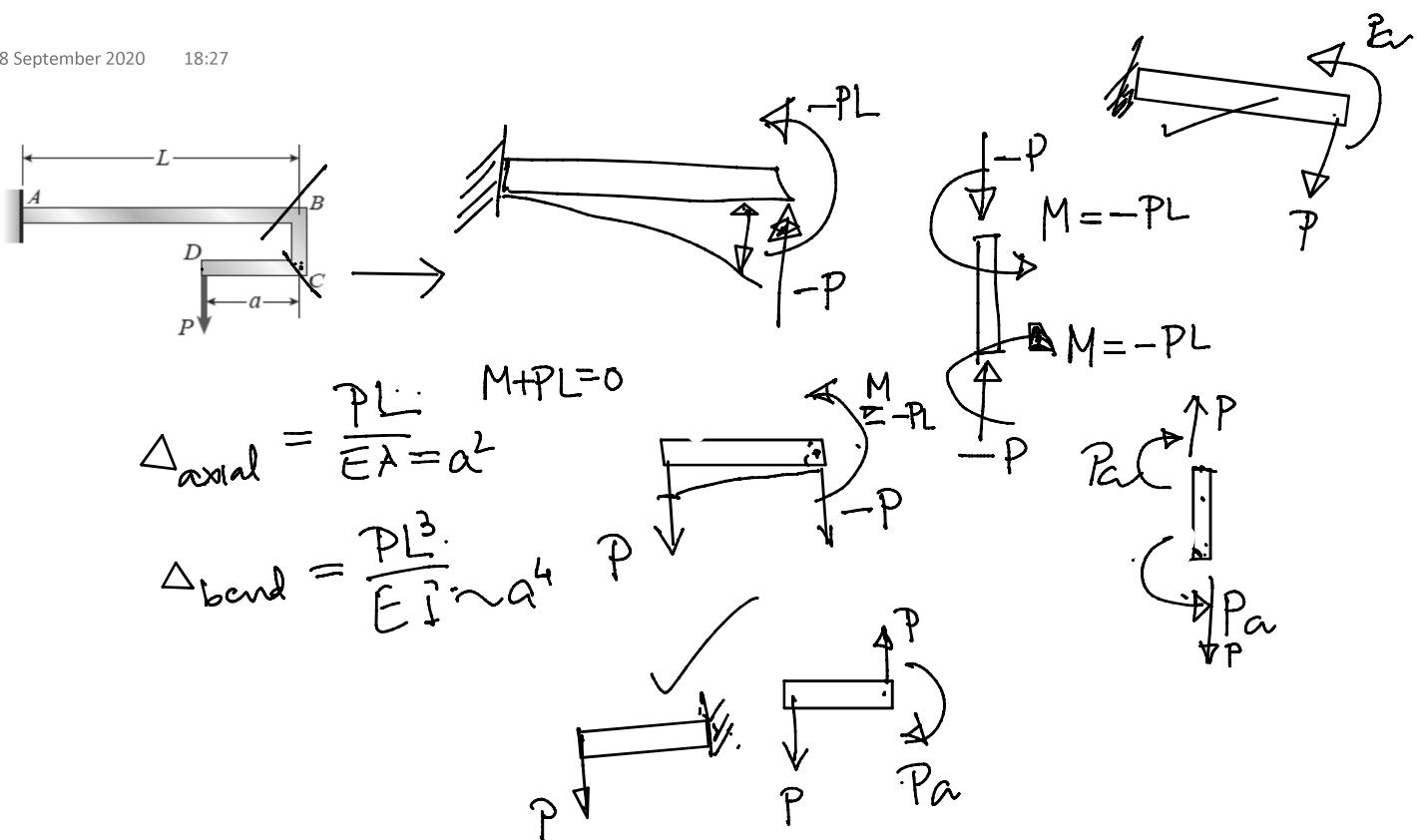
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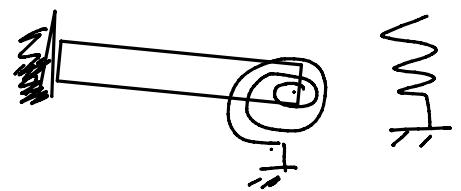
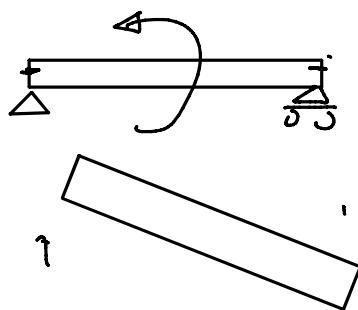
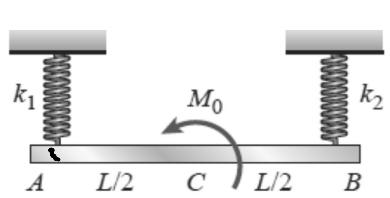
$$\begin{aligned} P &= k\Delta \quad \Delta = -v(L) \\ k &= \frac{48EI}{L^3} \\ P &= qL, \quad M_A + PL = \frac{qL^2}{2} \\ M_A &= -\frac{qL^2}{2} \\ -\frac{qL^2}{2} + M_A + Vx + M &= 0, \quad v = Vx \\ EJv'' = M &= \frac{qL^2}{2} + \frac{qx^2}{2} - Vx = \frac{qL^2}{2} - \frac{qx^2}{2} \\ v' &= \frac{qL^2}{2}x - \frac{qx^3}{6} + C_1 \rightarrow 0 \\ v &= \frac{qL^2x^2}{4} - \frac{qx^4}{24} + C_1x + C_2 \\ -kv(L) &= \left( \frac{qL^4}{4} - \frac{qL^4}{24} + C_2 \right) k \Rightarrow P \end{aligned}$$

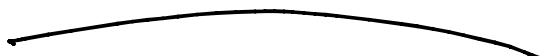
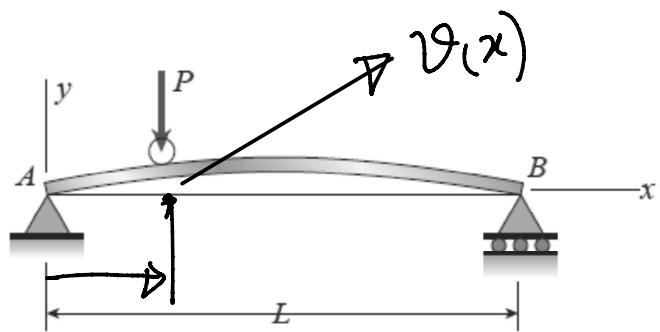


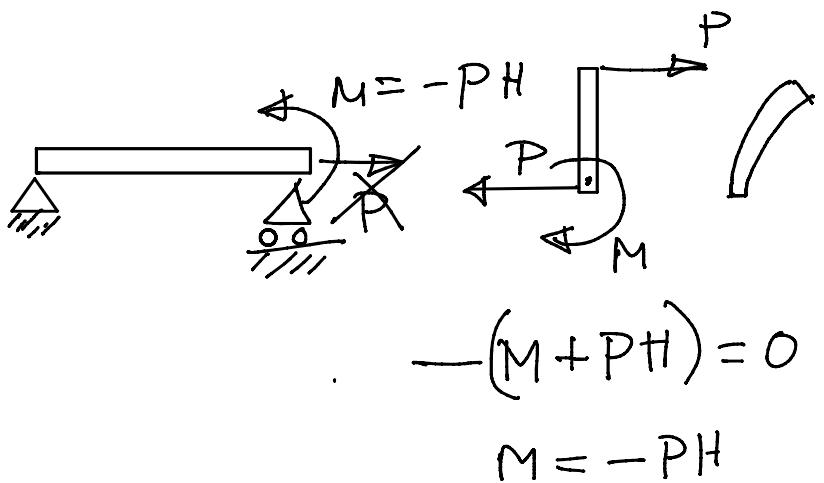
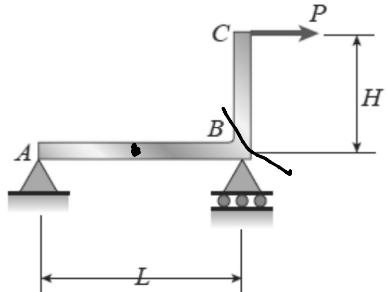


$$\begin{aligned}
 V + R_A &= \int_0^x q_0 \cos \frac{\pi x}{2L} dx = q_0 \left[ \sin \frac{\pi x}{2L} \right]_0^x / \frac{\pi}{2L} \\
 M_x + Vx + M &= \int_0^x (q_0 \cos \frac{\pi \xi}{2L}) \xi d\xi \quad \text{using } \int \sin u du = -\cos u + C \\
 M &= -M_A - Vx + \int_0^x q_0 \cos \frac{\pi \xi}{2L} d\xi \\
 &= -M_A - Vx + \left[ \xi \int_{\xi=0}^x \frac{\cos \pi \xi}{2L} d\xi - \int q_0 \times \frac{2L}{\pi} \sin \frac{\pi \xi}{2L} d\xi \right]_0^x
 \end{aligned}$$



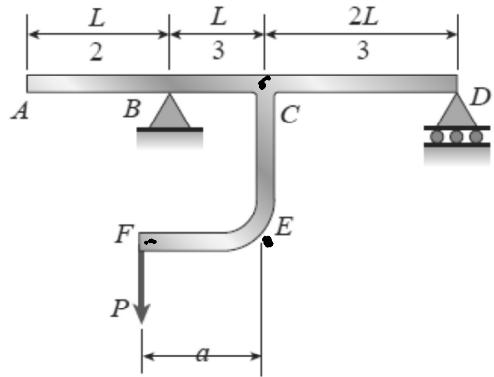




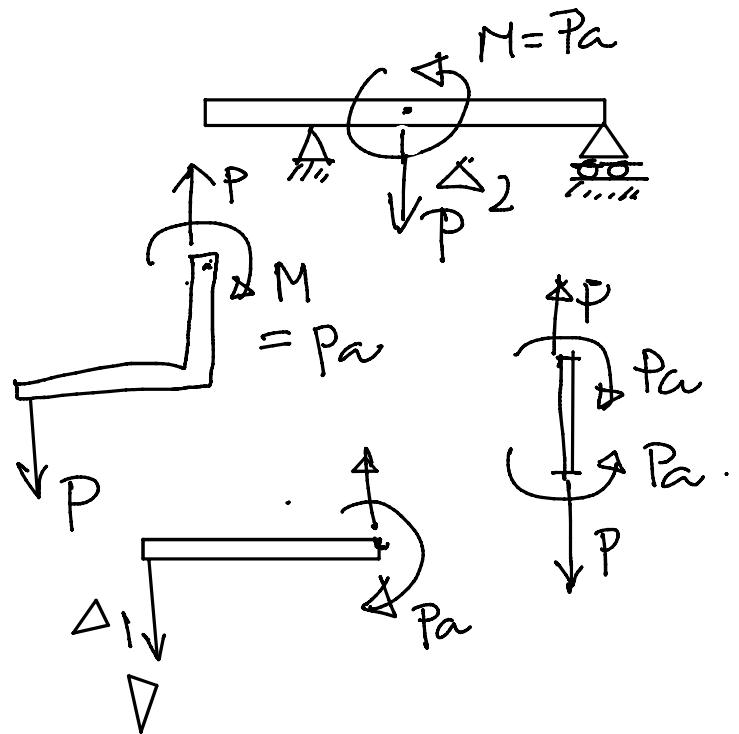


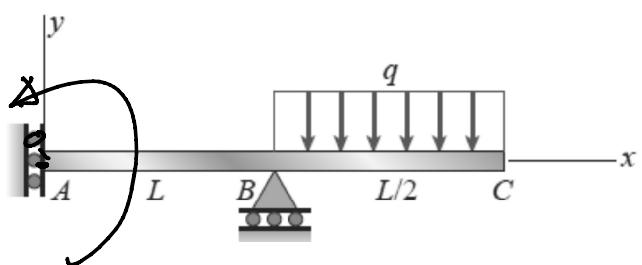
$$-(M + PH) = 0$$

$$M = -PH$$

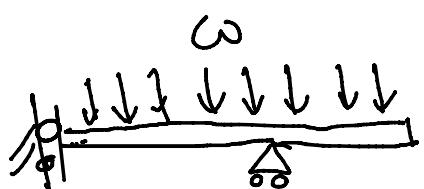


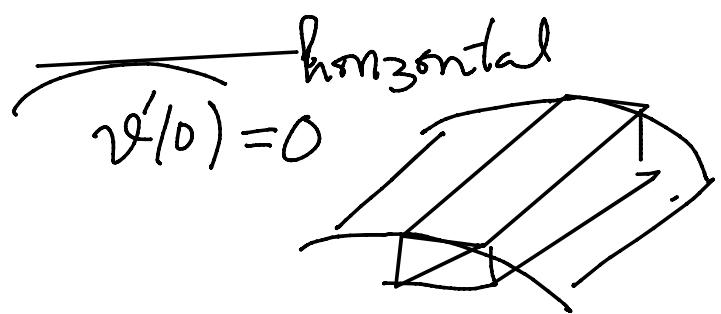
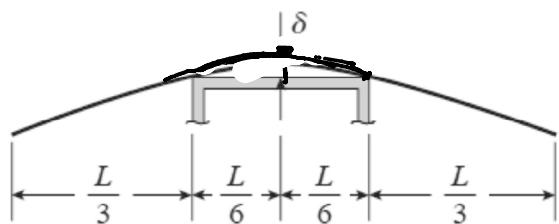
$$\Delta_1 + \Delta_2$$

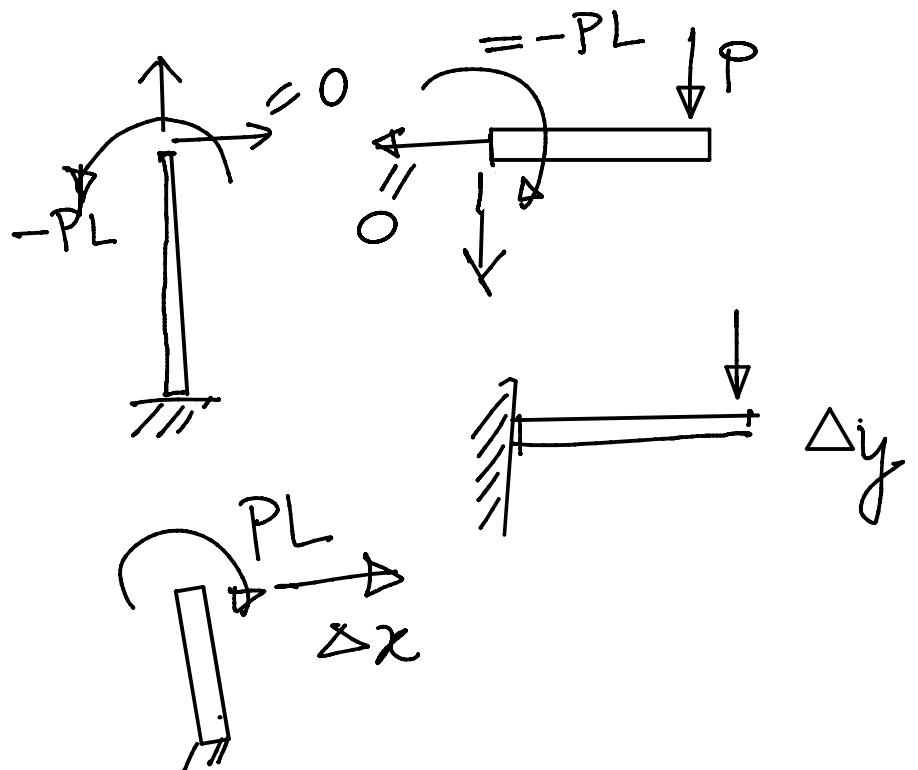
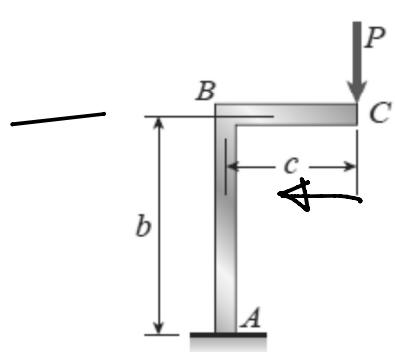


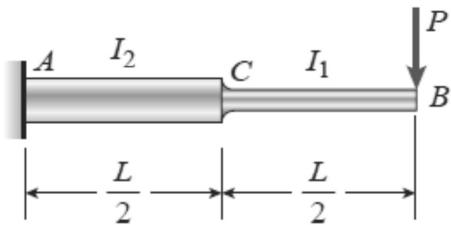


$$\varphi'(0) = 0$$
$$\varphi(L) = 0$$

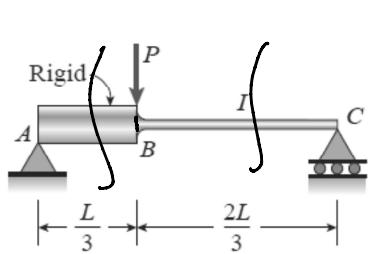








$$\varphi'' = \frac{M(x)}{EI_2} \quad AC$$
$$= \frac{M(x)}{EI_1} \quad CB$$



$$V(x) \quad M(x)$$

~~$E \int v'' = M(x) \times AB$~~

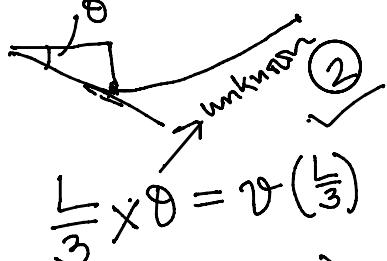
$$E \int v'' = M(x) \text{ BC.}$$

$$v' = g(x) + c_1$$

$$v = f(x) + c_1 x + c_2$$

$$x = \frac{L}{3} \quad v'\left(\frac{L}{3}\right)$$

$$v\left(\frac{L}{3}\right)$$



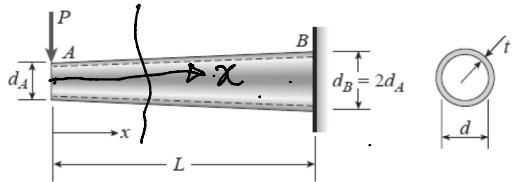
$$v(L) = 0 \quad (1)$$

v

$$\frac{L}{3} \times \theta = v\left(\frac{L}{3}\right)$$

$$\frac{\theta L}{3} = f\left(\frac{L}{3}\right) + c_1 \frac{L}{3} + c_2 \quad (2)$$

$$\theta = g\left(\frac{L}{3}\right) + c_1 \quad (3)$$



$$d = \frac{d_A}{L} (L + x)$$

$$I = \frac{\pi t d^3}{8} = \frac{\pi t d_A^3}{8L^3} (L + x)^3 = \frac{I_A}{L^3} (L + x)^3$$

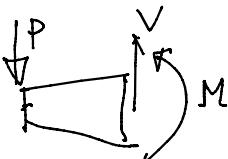
$$v'(L) = 0$$

$$v(L) = 0$$

$$\textcircled{E} I v'' = M(x)$$

$$v'' = \frac{M(x)}{E I(x)}$$

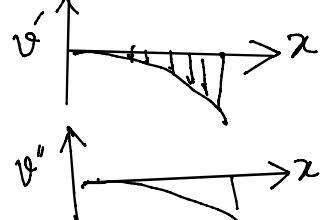
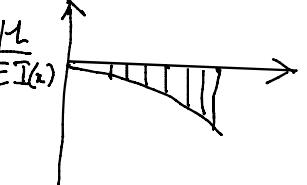
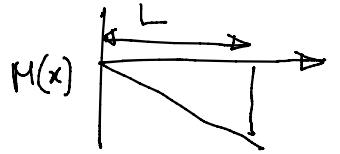
$$v = P$$

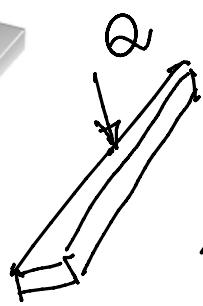
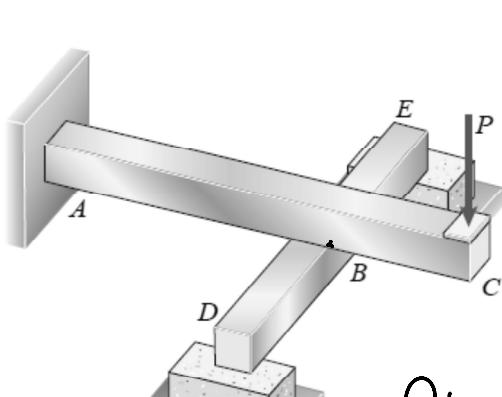


$$\begin{aligned} v_x + M &= 0 \\ M &= -v_x \\ &= -P_x \end{aligned}$$

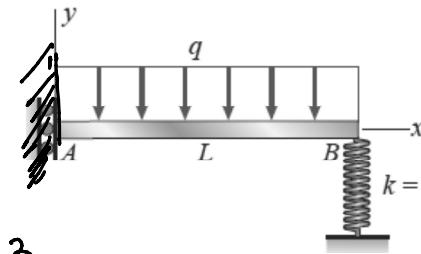
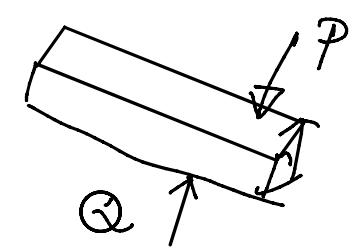
$$v'' = -\frac{P_x}{E I_A / L^3} (L + x)^3$$

$$\begin{matrix} v' \\ v \end{matrix}$$

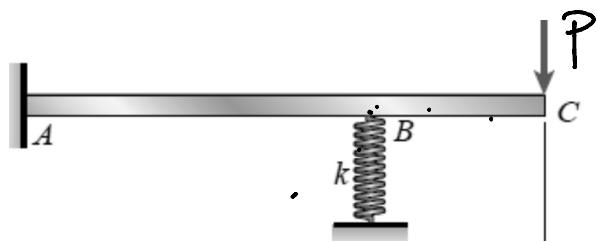




$$\Delta = \frac{Q L^3}{\alpha E I}$$

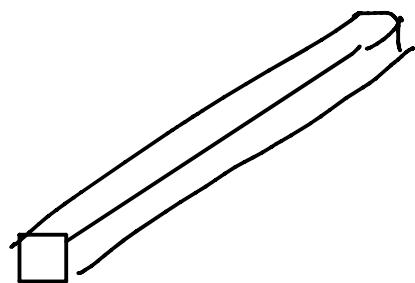


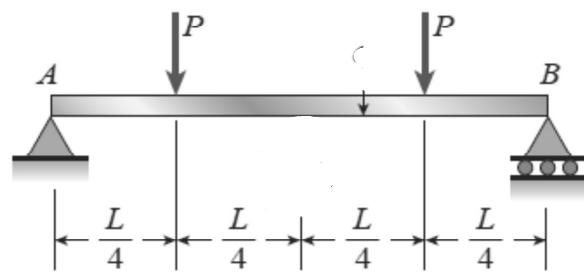
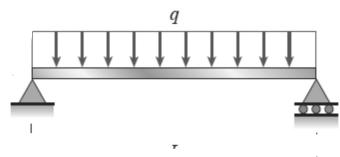
$$Q = \frac{\alpha E I}{L^3} \Delta = K \Delta$$



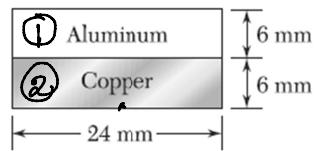
$$\Delta_c = \frac{PL^3}{3EI_1}$$

$$K = \frac{\alpha EI_1}{L^3}$$





A copper strip ( $E_c = 105 \text{ GPa}$ ) and an aluminum strip ( $E_a = 75 \text{ GPa}$ ) are bonded together to form the composite bar shown. Knowing that the bar is bent about a horizontal axis by a couple of moment 35 N m, determine the maximum stress in (a) the aluminum strip, (b) the copper strip. For the composite bar indicated, determine the radius of curvature caused by the couple of moment 35 Nm.



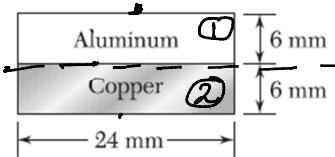
$$y_1 = 9 \text{ mm}, y_2 = 3 \text{ mm}$$

$$A_1 = 6 \times 24 = 144 \text{ mm}^2 = A_2$$

$$E_1 A_1 = 75 \text{ GPa} \times 144 \text{ mm}^2 =$$

$$E_2 A_2 = 105 \text{ GPa} \times 144 \text{ mm}^2 =$$

$$\begin{aligned} j_N &= \frac{E_1 A_1 y_1 + E_2 A_2 y_2}{E_1 A_1 + E_2 A_2} \\ &= \frac{75 \times 144 \times 9 + 105 \times 144 \times 3}{75 \times 144 + 105 \times 144} \\ &\stackrel{15}{=} \frac{75 \times 9 + 105 \times 3}{75 + 105} = \frac{15 \times 3 + 21 \times 1}{5+7} = \frac{\cancel{45}^{15}}{\cancel{12}^2} = 5.5 \text{ mm} \end{aligned}$$



$$I_{N1} = \frac{1}{12} \times 24 \times 6^3 + (24 \times 6)(9 - 5.5)^2$$

$$I_{N2} = \frac{1}{12} \times 24 \times 6^3 + (24 \times 6)(3 - 5.5)^2$$

$$\frac{l}{P} = \frac{M}{E_1 I_{N1} + E_2 I_{N2}}$$

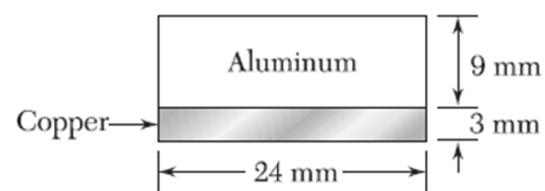
$$y_{\max} \text{ for Al} = 6 + 0.5 = 6.5 \text{ mm.}$$

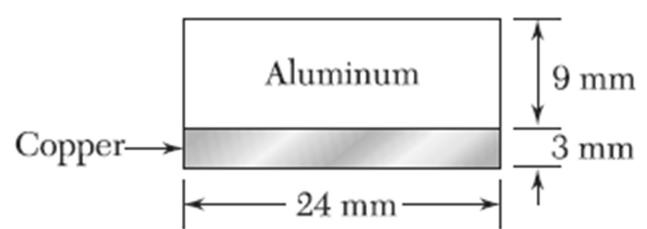
$$y_{\min} \text{ for Al} = 0.5 \text{ mm}$$

$$y_{\max} \text{ for Cu} = 0.5 \text{ mm}$$

$$y_{\min} \text{ for Cu} = -5.5 \text{ mm}$$

A cooper strip ( $E_c = 105 \text{ GPa}$ ) and an aluminum strip ( $E_a = 75 \text{ GPa}$ ) are bonded together to form the composite bar shown. Knowing that the bar is bent about a horizontal axis by a couple of moment  $35 \text{ N m}$ , determine the maximum stress in (a) the aluminum strip, (b) the copper strip. For the composite bar indicated, determine the radius of curvature caused by the couple of moment  $35 \text{ Nm}$ .



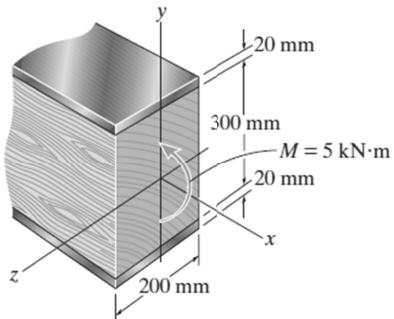


A wood beam is reinforced with steel straps at its top and bottom as shown. Determine the maximum bending stress developed in the wood and steel if the beam is subjected to a bending moment of  $M = 5 \text{ kN m}$ . Take  $E_w = 11 \text{ GPa}$ ,  $E_{st} = 200 \text{ GPa}$ .

$$\gamma_N = \frac{E_1 A_1 y_1 + E_2 A_2 y_2 + E_3 A_3 y_3}{E_1 A_1 + E_2 A_2 + E_3 A_3}$$

$$\begin{aligned} E_2 A_2 &= 11 \times 6 \times 10^4 \text{ GPa mm}^2 \\ &= 66 \times 10^4 \times 10^3 \text{ N} \\ &= 660 \text{ MN}. \end{aligned}$$

$$\gamma_N = \frac{E_1 A_1 y_1 + 0 + E_3 A_3 (-y_3)}{2E_1 A_1 + E_2 A_2} = 0$$



a

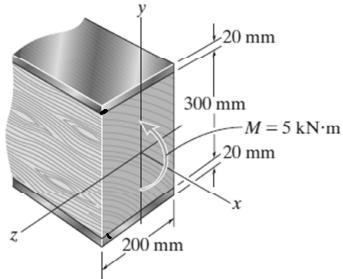
$$y_1 = -150 - 10 = -160 \text{ mm}$$

$$y_2 = 0, y_3 = 150 + 10 = 160 \text{ mm}$$

$$A_1 = 200 \times 20 = 4 \times 10^3 \text{ mm}^2 = A_3$$

$$A_2 = 300 \times 20 = 6 \times 10^3 \text{ mm}^2$$

$$\begin{aligned} E_1 A_1 &= E_3 A_3 = 200 \times 4 \times 10^3 \text{ GPa mm}^2 \\ &= 800 \times 10^3 \times 10^5 \times 10^{-6} \text{ N} \\ &= 800 \text{ MN} \end{aligned}$$

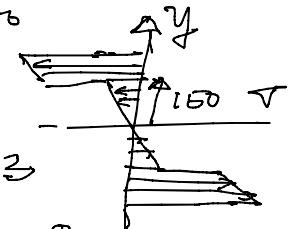


$$\sigma_{m,s} = M \times \frac{E_s y_1}{E_g I_{N1} + E_{\omega} I_{N2} + E_s I_{N3}}$$

$$I_{N1} = \frac{1}{12} \times 200 \times 20^3 + (200 \times 20) \times 160^2$$

$$= \frac{1}{12} \times 8 \times 10^5 + 4000 \times 256 \times 10^6$$

$$= \frac{4}{3} \times 10^5 + 1024 \times 10^5 = I_{N3}$$



$$E_1 I_{N1} = \left(1024 + \frac{4}{3}\right) \times 10^5 \text{ mm}^4$$

$\times 200 \text{ GPa}$

$$= E_3 I_{N3}$$

$$I_{N2} = \frac{1}{12} \times 200 \times 300^3 = \frac{1}{12} \times 27 \times 10^8$$

$$= \frac{9}{2} \times 10^8$$

$$E_2 I_{N2} = \frac{9}{2} \times 10^8 \times 11 \text{ GPa m}$$

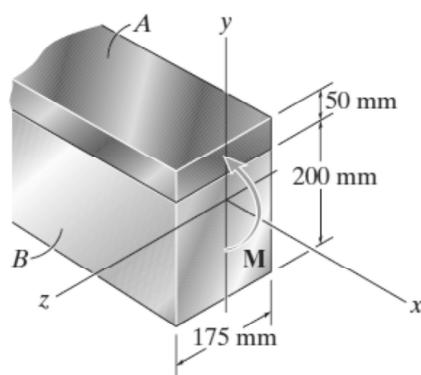
$$\sigma_{min,\omega=0}$$

$$\sigma_{min,s} = 5 \text{ kN m} \times \frac{200 \text{ GPa} \times 150 \text{ mm}}{\sum E I_N}$$

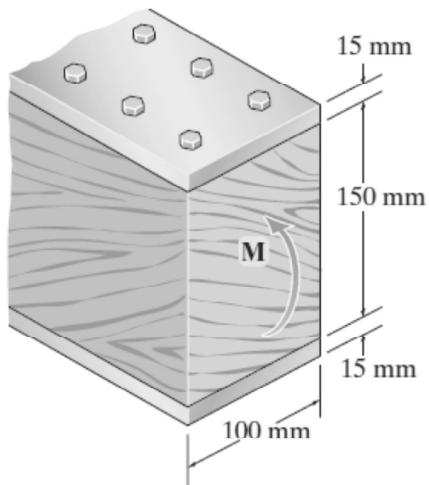
$$\sigma_{m,s} = 5 \text{ kN m} \times \frac{200 \text{ GPa} \times 170 \text{ mm}}{\sum E I_N}$$

$$y_m = 150 + 20$$

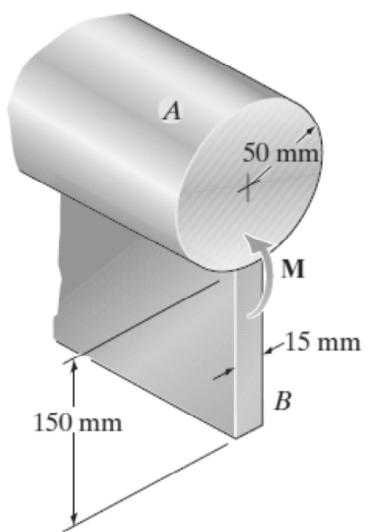
$$\sigma_{m,\omega} = 5 \text{ kN m} \times \frac{11 \text{ GPa} \times 150 \text{ mm}}{\sum E I_N}$$



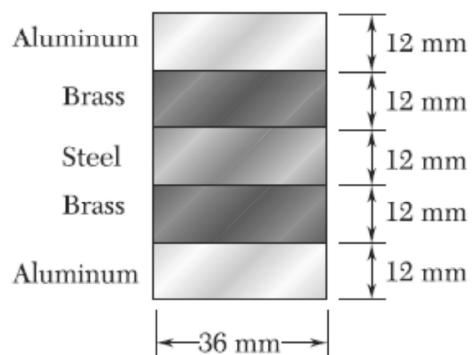
**6–121.** The composite beam is made of steel (A) bonded to brass (B) and has the cross section shown. If the allowable bending stress for the steel is  $(\sigma_{\text{allow}})_{\text{st}} = 180 \text{ MPa}$ , and for the brass  $(\sigma_{\text{allow}})_{\text{br}} = 60 \text{ MPa}$ , determine the maximum moment  $M$  that can be applied to the beam.  $E_{\text{br}} = 100 \text{ GPa}$ ,  $E_{\text{st}} = 200 \text{ GPa}$ .



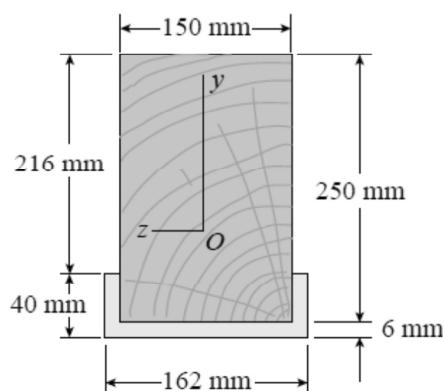
**6–126.** The wooden section of the beam is reinforced with two steel plates as shown. If the beam is subjected to an internal moment of  $M = 30 \text{ kN} \cdot \text{m}$ , determine the maximum bending stresses developed in the steel and wood. Sketch the stress distribution over the cross section. Take  $E_w = 10 \text{ GPa}$  and  $E_{st} = 200 \text{ GPa}$ .



**6-134.** If the beam is subjected to an internal moment of  $M = 45 \text{ kN}\cdot\text{m}$ , determine the maximum bending stress developed in the A-36 steel section A and the 2014-T6 aluminum alloy section B.

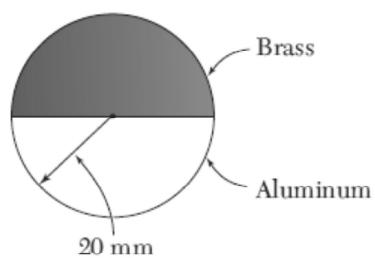


Five metal strips, each of  $12 \times 36$  mm cross section, are bonded together to form the composite beam shown. The modulus of elasticity is 200 GPa for the steel, 100 GPa for the brass, and 70 GPa for the aluminum. Knowing that the beam is bent about a horizontal axis by couples of moment 1350 N·m., determine (a) the maximum stress in each of the three metals, (b) the radius of curvature of the composite beam.



**Problem 6.3-12** A wood beam reinforced by an aluminum channel section is shown in the figure. The beam has a cross section of dimensions 150 mm by 250 mm, and the channel has a uniform thickness of 6 mm.

If the allowable stresses in the wood and aluminum are 8.0 MPa and 38 MPa, respectively, and if their moduli of elasticity are in the ratio 1 to 6, what is the maximum allowable bending moment for the beam?



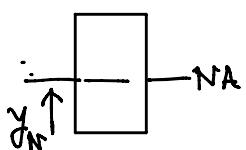
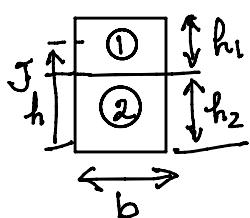
The composite beam shown is formed by bonding together a brass rod and an aluminum rod of semicircular cross sections. The modulus of elasticity is 100 GPa for the brass and 70 GPa for the aluminum. Knowing that the composite beam is bent about a horizontal axis by couples of moment 900 N·m, determine the maximum stress (a) in the brass, (b) in the aluminum.

A rectangular beam is made of a material for which the modulus of elasticity is  $E_t$  in tension and  $E_c$  in compression. Show that the curvature of the beam in pure bending is

$$\frac{1}{\rho} = \frac{M}{E_r I}$$

where  $E_r = \frac{4E_t E_c}{(\sqrt{E_t} + \sqrt{E_c})^2}$

- Consider a composite beam made of materials ① & ② subjected to a moment  $M$ . The neutral axis coincides with the junction line. What is the ratio of the thickness of the layers ① & ②?



$$y_1 = h_2 + \frac{h_1}{2}$$

$$y_2 = \frac{h_2}{2}$$

$$A_1 = bh_1, A_2 = bh_2$$

$$\int \sigma dA = 0 \quad \sigma = \frac{h - y_N}{P} x E_i$$

$$E_i = E_t, E_x = E_c$$

$$\int \sigma dA + \int \sigma dA = 0$$

(1)

$$y_N = \frac{E_t A_1 y_1 + E_c A_2 y_2}{E_t A_1 + E_c A_2}$$

$y_1$  in terms of  $b, h_1, h_2, E_t, E_c$

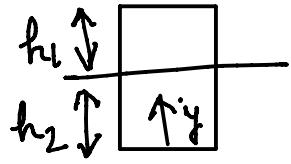
$$= \frac{E_t(bh_1)(h_2 + \frac{h_1}{2}) + E_c(bh_2)\frac{h_2}{2}}{E_t bh_1 + E_c bh_2}$$

$$= \frac{E_t y_1 (2h_2 + h_1) + E_c y_2 h_2^2}{2(E_t h_1 + E_c h_2) b}$$

$$= \frac{E_t (2h_2 h_1 + h_1^2) + E_c h_2^2}{2(E_t h_1 + E_c h_2)}$$

"

$$\gamma_N = \frac{E_t(2h_1h_2 + h_1^2) + E_c h_2^2}{2(E_t h_1 + E_c h_2)}$$



But  $\gamma_N = h_2$

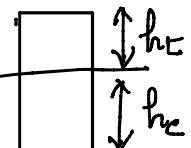
$$\frac{E_t(2h_1h_2 + h_1^2) + E_c h_2^2}{2(E_t h_1 + E_c h_2)} = h_2$$

$$E_t(2h_1h_2 + h_1^2) + E_c h_2^2 = 2E_t h_1h_2 + 2E_c h_2^2$$

$$E_t h_1^2 = E_c h_2^2$$

$$\frac{h_1}{h_2} = \sqrt{\frac{E_c}{E_t}} = \frac{f_t}{f_c}$$

$$\begin{aligned}
 \frac{I}{P} &= \frac{M}{E_t I_{Nt} + E_c I_{Nc}} \\
 &= \frac{M}{E_t \frac{1}{3} b h_t^3 + E_c \frac{1}{3} b h_t^3 \left(\sqrt{\frac{E_t}{E_c}}\right)^3} \\
 &= \frac{3M E_c^{3/2} / (b h_t^3)}{E_t E_c^{3/2} + E_c E_t^{3/2}} \\
 &= \frac{3M E_c^{3/2}}{b h_t^3 E_t} \times \frac{\sqrt{E_c}}{\sqrt{E_c} + \sqrt{E_t}}
 \end{aligned}$$



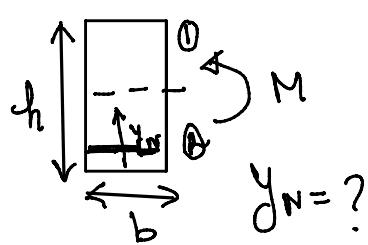
$$h_c = h_t \sqrt{\frac{E_t}{E_c}}$$

$$I_{Nt} = \frac{1}{3} b h_t^3$$

$$I_{Nc} = \frac{1}{3} b h_c^3$$

$$= \frac{1}{3} b h_t^3 \left(\frac{E_t}{E_c}\right)^{3/2}$$

$$I = \frac{1}{3} b h_c^3 + \frac{1}{3} b h_t^3 \left(\frac{E_t}{E_c}\right)^{3/2}$$



$$E_1 = E, E_2 = 0$$

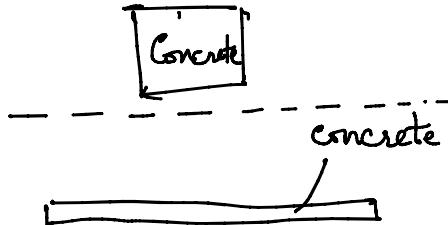
$$\begin{aligned} y_N &= \frac{E_1 A_1 y_1 + E_2 A_2 y_2}{E_1 A_1 + E_2 A_2} \\ &= \frac{E (b - y_N) (y_N + \frac{h - y_N}{2})}{E X (h - y_N)} \end{aligned}$$

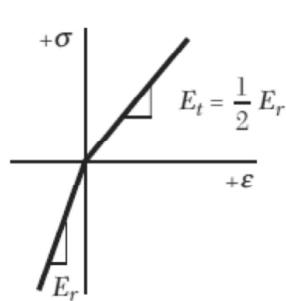
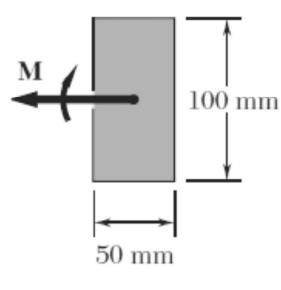
$y_N = y_N + \frac{h}{2} - \frac{y_N}{2}$

$y_N = h$

$A_2$  = Area of c.s of all rods

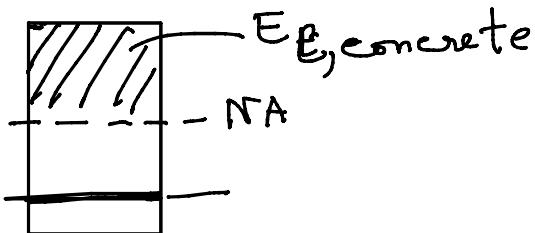
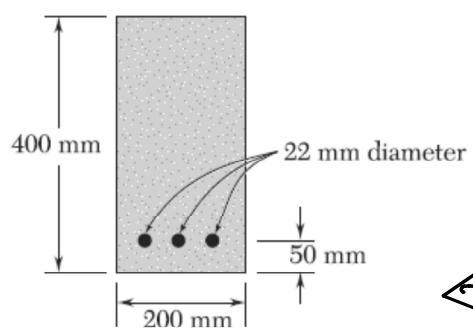
$y_2$  = distance from bottom



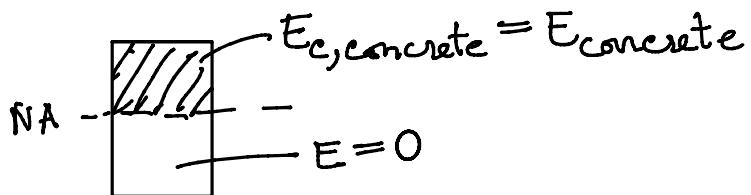


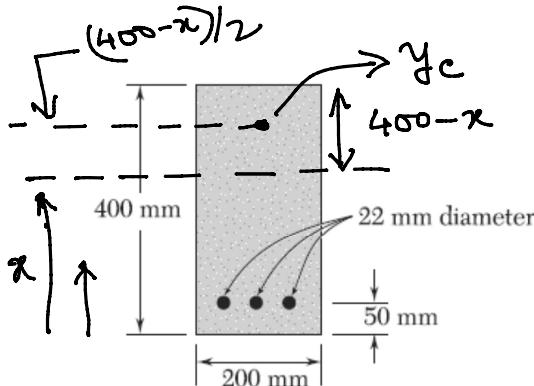
The rectangular beam shown is made of a plastic for which the value of the modulus of elasticity in tension is one half of its value in compression. For a bending moment  $M = 600 \text{ N}\cdot\text{m}$ , determine the maximum (a) tensile stress, (b) compressive stress.

A concrete beam is reinforced by three steel rods placed as shown. The modulus of elasticity is 20 GPa for the concrete and 200 GPa for the steel. Using an allowable stress 9 MPa for the concrete and 140 MPa for the steel, determine the largest allowable positive bending moment in the beam.



without reinforcement





A concrete beam is reinforced by three steel rods placed as shown. The modulus of elasticity is 20 GPa for the concrete and 200 GPa for the steel. Using an allowable stress 9 MPa for the concrete and 140 MPa for the steel, determine the largest allowable positive bending moment in the beam.

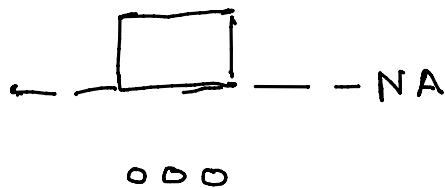
$$A_s = \pi \times 11^2 = 380 \text{ mm}^2$$

$$A_s = 3 \times 380 = 1140 \text{ mm}^2, y_s = 50 \text{ mm}$$

$$A_c = 400 \times 200 = 8 \times 10^4 \text{ mm}^2, y_c = 200 \text{ mm}$$

$$y_N = \frac{E_s A_s y_s + E_c A_c y_c}{E_s A_s + E_c A_c} = \frac{200 \times 1140 \times 50 + 20 \times 8 \times 10^4 \times 200}{200 \times 1140 + 20 \times 8 \times 10^4}$$

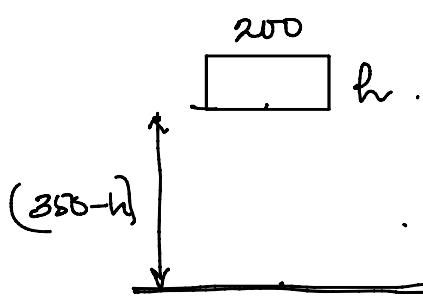
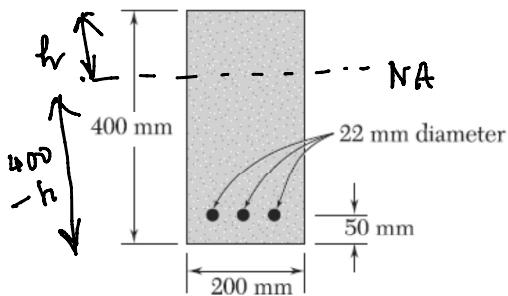
$$x = \frac{200 \times 1140 \times 50 + 20(400-x) \times 200 \times \left\{ x + \frac{(400-x)}{2} \right\}}{200 \times 1140 + 20(400-x) \times 200}$$



$$I_{Nc,c} = \frac{1}{3} \times 200 x (400-x)^3$$

$$I_{Ns,s} = A_s \times (x - 50)^2$$

$$\frac{1}{P} = \frac{M}{E_c I_{Nc} + E_s I_{Ns}}$$



Concrete  
is reference.

$$I_N = \frac{1}{3} 200 \times h^3 + 11400 \times (350-h)^2$$

$1140 \times 10 \text{ mm}^2$

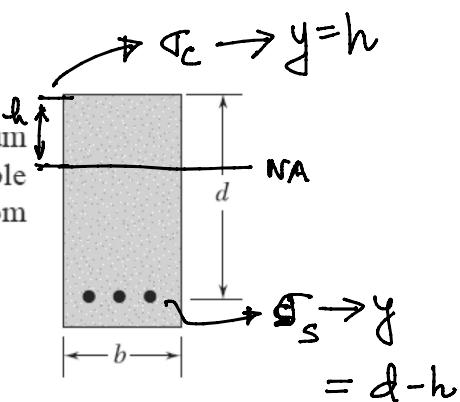
$$(200 \times h) \times \frac{h}{2} = (11400) \times (350-h)$$

$$\Delta_c = \frac{M_y}{E_c I_N} = \frac{M_y}{200 \text{ GPa} \times I_N}$$

$$\sigma_s = -\frac{M_y}{E_c I_N} \times n = \frac{M_y}{200 \text{ GPa} \times I_N} \times 10$$

The design of a reinforced concrete beam is said to be *balanced* if the maximum stresses in the steel and concrete are equal, respectively, to the allowable stresses,  $\sigma_s$  and  $\sigma_c$ . Show that to achieve a balanced design the distance  $x$  from the top of the beam to the neutral axis must be

$$x = \frac{d}{1 + \frac{\sigma_s E_c}{\sigma_c E_s}}$$



where  $E_c$  and  $E_s$  are the moduli of elasticity of concrete and steel, respectively, and  $d$  is the distance from the top of the beam to the reinforcing steel.

