# Homeworks for SM-I course

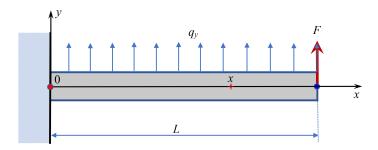
### Important:

- 1). Please download the latest grcodes, images, and related chapters before working on the homework.
- 2). Both pdf files and the source codes must be submitted, or the work will not be marked.

## Homework 6: Solutions for beams

#### Question 1:

Consider a cantilever beam of uniform cross-section, as shown in the following image. The beam is clamped at the left-end. It has length L=1m a square section area of  $A=0.0001m^2$ . It is subjected to a distributed body force  $q_y$  and a concentrated force F at the right end.



A cantilever beam of uniform cross-section, subjected to a distributed body-force, and a concentrated force at the right-end.

- 1. Consider only the body force  $b_y=q$  (N/m) where q is a constant, derive by hand the formulas for computing the deflection, v in the y-direction, cross-section rotation, moment, shear force, and normal stress  $\sigma_{xx}$  in the beam, as functions of x. Compare the solutions with the corresponding ones obtained using the code given in the textbook.
- 2. Given data,  $A=0.0001m^2$ , L=1m, Young's modulus of the material  $E=2.1e^{10}Pa$ , q=500N/m, F=1500N, compute and plot the distributions of the deflection, cross-section rotation, moment, shear force, and the maximum normal stress on the cross-section, along the coordinate x.

1.

$$egin{align*} rac{d^4v}{dx^4} &= rac{b_y}{EI_z} \ rac{d^3v}{dx^3} &= V = xrac{b_y}{EI_z} + c_3 \ rac{d^2v}{dx^2} &= M = x^2rac{b_y}{2EI_z} + xc_3 + c_2 \ rac{dv}{dx} &= heta = x^3rac{b_y}{6EI_z} + rac{1}{2}x^2c_3 + xc_2 + c_1 \ v &= x^4rac{b_y}{24EI_z} + rac{1}{6}x^3c_3 + rac{1}{2}x^2c_2 + xc_1 + c_0 \ \end{pmatrix}$$

$$b_u = q$$

Beam is fixed at x=0 , therefore:

$$v = 0$$

$$\theta = 0$$

at 
$$x = 0$$

$$0 = 0^4 rac{q}{24EI_z} + rac{1}{6}0^3 c_3 + rac{1}{2}0^2 c_2 + 0c_1 + c_0 \ c_0 = 0$$

$$0 = 0^3 rac{q}{6EI_z} + rac{1}{2}0^2 c_3 + 0 c_2 + c_1 \ c_1 = 0$$

$$egin{aligned} v &= x^4 rac{q}{24EI_z} + rac{1}{6} x^3 c_3 + rac{1}{2} x^2 c_2 \ heta &= x^3 rac{q}{6EI} + rac{1}{2} x^2 c_3 + x c_2 \end{aligned}$$

At 
$$x=L$$
:  $V=qL$   $M=0$   $qL=Lrac{q}{EI_z}+c_3$   $c_3=qL(1-rac{1}{EI_z})$ 

$$egin{aligned} 0 &= L^2 rac{q}{2EI_z} + LqL(1 - rac{1}{EI_z}) + c_2 \ c_2 &= -L^2q(rac{1}{2EI_z} + 1 - rac{1}{EI_z}) \ c_2 &= -L^2q(1 - rac{1}{2EI_z}) \end{aligned}$$

$$\frac{3l^2q}{2}-2lqx+\frac{qx^2}{2}$$

$$egin{aligned} v &= x^4 rac{q}{24EI_z} + x^3 rac{qL}{3EI_z} + x^2 rac{3L^2q}{4EI_z} \ & heta &= x^3 rac{q}{6EI_z} + x^2 rac{qL}{EI_z} + x rac{3L^2q}{2EI_z} \ & M &= x^2 rac{q}{2} + 2xqL - rac{3L^2q}{2} \ & V &= 2qL - qx \end{aligned}$$

$$egin{aligned} \sigma_{xx} &= -yrac{M}{I_z} \ \sigma_{xx} &= rac{-y}{I_z}(x^2rac{q}{2} + 2xqL - rac{3L^2q}{2}) \end{aligned}$$

```
In []: def solver1D4(E, I, by, 1, v0, \theta0, v1, \theta1, V0, M0, V1, M1, key='c-c'):
                         '''Solves the Beam Equation for integrable distributed body force:
                        u,x4=-by(x)/EI, with various boundary conditions (BCs):
                        c-c, s-c, c-s, f-c, c-f.
                        Input: EI: bending stiffness factor; by, body force; l, the length
                        of the beam; v0, \theta0, V0, M0, deflection, rotation,
                        shear force, moment at x=0; v1, \theta1, V1, M1, those at x=1.
                        Return: u, v x, v x2, v x3, v x4 up to 4th derivatives of v
                        c0, c1, c2, c3 = symbols('c0, c1, c2, c3') #integration constant
                        EI = E*I # I is the Iz in our chosen coordinates.
                        # Integrate 4 times:
                        v_x3= sp.integrate(by/EI,(x, 0, x))+ c0 \#ci: integration constant
                        v_x2= sp.integrate(v_x3,(x, 0, x))+ c1
                        v_x = sp.integrate(v_x2,(x, 0, x)) + c2
                        v = sp.integrate(v_x,(x, 0, x)) + c3
                        # Solve for the 4 integration constants:
                        if key == "s-s":
                                cs=sp.solve([v.subs(x,0)-v0, v_x2.subs(x,0)-M0/EI,v.subs(x,1)-v1, v_x2.subs(x,0)-M0/EI,v.subs(x,0)-v1, v_x2.subs(x,0)-v1, v_x2.subs(x,0)-v2, v_x2.subs(x,0)-v3, v_x2.subs(x,0)-v4, v_x3, v_x3,
                        elif key == "c-c":
                                cs=sp.solve([v.subs(x,0)-v0, v.x.subs(x,0)-\theta0,v.subs(x,1)-v1, v.x.subs(x,1)]
                        elif key == "s-c":
                                cs=sp.solve([v.subs(x,0)-v0, v x2.subs(x,0)-M0/EI,v.subs(x,1)-v1, v x.subs(
                                #print('solver1D4:',cs[c0],cs[c1],cs[c2],cs[c3])
                        elif key == "c-s":
                                cs=sp.solve([v.subs(x,0)-v0, v.x.subs(x,0)-\theta0,v.subs(x,1)-v1, v.x2.subs(x,1)
                        elif key == "c-f":
                                cs=sp.solve([v.subs(x,0)-v0, v_x.subs(x,0)-\theta0,v_x3.subs(x,1)+V1/EI, v_x2.su
                        elif key == "f-c":
                                cs=sp.solve([v_x3.subs(x,0)-V0/EI, v_x2.subs(x,0)-M0/EI,v.subs(x,1)-v1, v_x]
                        else:
                                print("Please specify boundary condition type.")
                                sys.exit()
                        # Substitute the constants back to the integral solutions
                        v = v.subs({c0:cs[c0],c1:cs[c1],c2:cs[c2],c3:cs[c3]})
                        v = v.expand().simplify().expand()
                        v_x = v_x.subs({c0:cs[c0],c1:cs[c1],c2:cs[c2],c3:cs[c3]})
                        v_x = v_x.expand().simplify().expand()
                        v x2=v x2.subs({c0:cs[c0],c1:cs[c1],c2:cs[c2],c3:cs[c3]})
                        v_x2=v_x2.expand().simplify().expand()
                        v_x3=v_x3.subs({c0:cs[c0],c1:cs[c1],c2:cs[c2],c3:cs[c3]})
                        v_x3=v_x3.expand().simplify().expand()
                        v_x4 = sp.diff(v_x3,x).expand()
                        print("Outputs form solver1D4(): v, \theta, M, V, qy")
                        return v,v x,(EI*v x2).expand(),(-EI*v x3).expand(),(v x4).expand()
```

```
In []: E, I, EI, A, 1, \xi = symbols('E, I, EI, A, 1, \xi ', nonnegative=True) x, q = symbols('x, q ') # geometry & force c0, c1, c2, c3 = symbols('c0, c1, c2, c3') #integration constant # for displacement boundary conditions (DBCs): v0, \theta0, v1, \theta1 = symbols('v0, \theta0, v_1, \theta_1') # DBCs # for force boundary conditions (FBCs): v0, M0, V1, M1 = symbols('V0, M0, V_1, M_1') # FBCs by = q v = \text{solver1D4}(E,I,by,l,0,0,vl,\theta l,V0,M0,by*l,0,key='c-f')
```

Outputs form solver1D4(): v,  $\theta$ , M, V, qy

v:

Out[]: 
$$rac{3l^2qx^2}{4EI} - rac{lqx^3}{3EI} + rac{qx^4}{24EI}$$
  $v = x^4rac{q}{24EI_z} + x^3rac{qL}{3EI_z} + x^2rac{3L^2q}{4EI_z}$ 

 $\theta$ :

Out[]: 
$$rac{3l^2qx}{2EI}-rac{lqx^2}{EI}+rac{qx^3}{6EI}$$
  $heta=x^3rac{q}{6EI_z}+x^2rac{qL}{EI_z}+xrac{3L^2q}{2EI_z}$ 

M:

Out[]: 
$$\frac{3l^2q}{2} - 2lqx + \frac{qx^2}{2}$$

$$M = x^2 rac{q}{2} + 2xqL - rac{3L^2q}{2}$$

V:

Out[ ]: 
$$2lq-qx$$

$$V = 2qL - qx$$

```
In [ ]: E, I, EI, A, 1, ξ = symbols('E, I, EI, A, 1, ξ ', nonnegative=True)
    x, q, F= symbols('x, q, F') # geometry & force
    c0, c1, c2, c3 = symbols('c0, c1, c2, c3') #integration constant
    # for displacement boundary conditions (DBCs):
    v0, θ0, v1, θ1 = symbols('v0, θ0, v_1, θ_1') # DBCs
    # for force boundary conditions (FBCs):
    V0, M0, V1, M1 = symbols('V0, M0, V_1, M_1') # FBCs
    by = q
    V1 = F

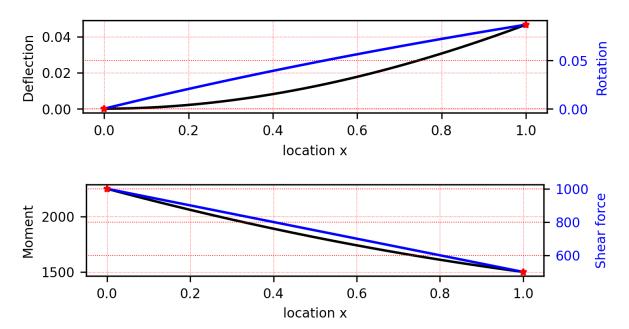
    v = solver1D4(E,I,by,1,0,0,v1,θ1,V0,M0,by*1,F,key='c-f')
    v
```

Outputs form solver1D4(): v, 
$$\theta$$
, M, V, qy 
$$\frac{\text{Out[]}:}{2EI} + \frac{3l^2qx^2}{4EI} - \frac{lqx^3}{3EI} + \frac{qx^4}{24EI}, \frac{Fx}{EI} + \frac{3l^2qx}{2EI} - \frac{lqx^2}{EI} + \frac{qx^3}{6EI}, F + \frac{3l^2q}{2} - 2lqx + \frac{qx^2}{2} + \frac{qx^3}{2} + \frac{qx^3}{2} + \frac{qx^4}{2} + \frac$$

```
In [ ]: def plot2curveS(u, xL=0., xR=1., title="f_title"):
            '''Print out maximum values and loctions, as well as stationary
            points, and the values at the stationary points, and boundaries'''
            x = sp.symbols('x')
            dx = 0.01; dxr = dx*10
                                          # x-step
            xi = np.arange(xL, xR+dx, dx)
            uf = sp.lambdify((x), u[0], 'numpy') #convert Sympy f to numpy f
            yi = uf(xi)
            if type(yi) != np.ndarray: #in case, uf is a constant
                #type(yi) == int or type(yi) == float: # or len(yi)==1:
                xi = np.arange(xL, xR+dxr, dxr)
                yi = float(yi)*np.ones_like(xi)
            fig, ax1 = plt.subplots(figsize=(5.,1.), dpi=300)
            fs = 8
                             # fontsize
            color = 'black'
            ax1.set_xlabel('location x', fontsize=fs)
            ax1.set_ylabel(title[0], color=color, fontsize=fs)
            ax1.plot(xi, yi, color=color)
            ax1.grid(color='r',ls=':',lw=.3, which='both') # Use both tick
            ax1.tick_params(axis='x', labelcolor=color, labelsize=fs)
            ax1.tick_params(axis='y', labelcolor=color, labelsize=fs)
            vmax = yi[yi.argmax()]
            max_l = np.argwhere(yi == vmax)
            ax1.plot(xi[max_l], yi[max_l], 'r*', markersize=4)
            print(f'Maximum {title[0]} value={vmax:.3e}, at x={xi[max_1][0][0]}')
            uf = sp.lambdify((x), u[1], 'numpy') #convert Sympy f to numpy f
            xi = np.arange(xL, xR+dx, dx)
            yi2 = uf(xi)
            if type(yi2) != np.ndarray: # or len(yi2) == 1:
                xi = np.arange(xL, xR+dxr, dxr)
                yi2 = float(yi2)*np.ones_like(xi)
            m1, m2, m3 = np.partition(abs(yi2), 2)[0:3]
            msl=[np.where(abs(yi2)==m1)[0][0], np.where(abs(yi2)==m2)[0][0],
                      np.where(abs(yi2)==m3)[0][0]]
            vmax = yi2[yi2.argmax()]
            max_1 = np.argwhere(yi2 == vmax)
            print(f'Maximum {title[1]} value={vmax:.3e}, at x={xi[max_1][0][0]}')
            if abs(xi[msl[2]]-xi[msl[1]])<2*dx:</pre>
                if abs(yi2[msl[2]]-0.)<abs(yi2[msl[1]]-0.): msl.pop(1)</pre>
                else: msl.pop(2)
            if len(msl) > 2:
                if abs(xi[ms1[2]]-xi[ms1[0]])<2*dx:</pre>
                     if abs(yi2[ms1[2]]-0.)<abs(yi2[ms1[0]]-0.): ms1.pop(0)</pre>
                     else: msl.pop(2)
            if len(msl) > 1:
                if abs(xi[msl[1]]-xi[msl[0]])<2*dx:</pre>
                     if abs(yi2[msl[1]]-0.)<abs(yi2[msl[0]]-0.): msl.pop(0)</pre>
                     else: msl.pop(1)
            ax2 = ax1.twinx() # instantiate second axes sharing the same x-axis
```

```
color = 'blue'
            ax2.set_ylabel(title[1], color=color, fontsize=fs)
            ax2.plot(xi, yi2, color=color)
            ax2.plot(xi[max_1], yi2[max_1], 'r*', markersize=4)
            ax2.plot(xi[msl], yi2[msl], 'r*', markersize=4)
            ax1.plot(xi[msl], yi[msl], 'r*', markersize=4)
            ax2.plot(xi[0], yi2[0], 'ro', markersize=2)
            ax1.plot(xi[0], yi[0], 'ro', markersize=2)
            ax2.plot(xi[-1], yi2[-1], 'ro', markersize=2)
            ax1.plot(xi[-1], yi[-1], 'ro', markersize=2)
            ax2.grid(color='r',ls=':',lw=.5, which='both') # Use both tick
            ax2.tick_params(axis='x', labelcolor=color, labelsize=fs)
            ax2.tick_params(axis='y', labelcolor=color, labelsize=fs)
            np.set_printoptions(formatter={'float': '{: 0.3e}'.format})
            print(f'Extreme {title[0]} values={yi[msl]},\n at x={xi[msl]}')
            print(f'Critical {title[1]} values={yi2[ms1]},\n at x={xi[ms1]}')
            print(f'{title[0]} values at boundary ={yi[0], yi[-1]}')
            print(f'{title[1]} values at boundary ={yi2[0], yi2[-1]}\n')
In [ ]: A = 1*(10**-4)
        # Since its a square cross section, y = sqrt(A)
        y = np.sqrt(A)
        Iact = A*y
        dic = \{E:2.1*(10**10), I:Iact, 1:1, q:500, v0:0, \theta0:0, v1:0, \theta1:0, F:1500\}
        vx=[v[i].subs(dic) for i in range(len(v))]
        vx # Outputs are: v, \theta, Mx, Vx, qy
Out[]: [0.000992063492063492x^4 - 0.00793650793650793x^3 + 0.0535714285714286x^2, 0.00396]
In [ ]: | title = [["Deflection", "Rotation"], ["Moment", "Shear force"]]
        for i in range(len(title)):
            plot2curveS(vx[2*i:2*(i+1)], 0., 1., title=title[i])
        #fig.tight Layout()
        #plt.savefig('images/beam_cq.png', dpi=500) # save the plot to file
        plt.show()
```

```
Maximum Deflection value=4.663e-02, at x=1.0
Maximum Rotation value=8.730e-02, at x=1.0
Extreme Deflection values=[ 0.000e+00],
    at x=[0.000e+00]
Critical Rotation values=[ 0.000e+00],
    at x=[ 0.000e+00]
Deflection values at boundary =(0.0, 0.04662698412698416)
Rotation values at boundary = (0.0, 0.08730158730158716)
Maximum Moment value=2.250e+03, at x=0.0
Maximum Shear force value=1.000e+03, at x=0.0
Extreme Moment values=[ 1.500e+03],
    at x=[1.000e+00]
Critical Shear force values=[ 5.000e+02],
    at x=[ 1.000e+00]
Moment values at boundary =(2250.0, 1500.0)
Shear force values at boundary =(1000.0, 500.0)
```



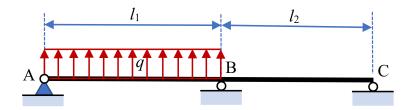
```
In [ ]: def roundS(expr, n_d): # to be used in maxminS()
    '''To limit the number of digits to keep in a variable.
    Usage: roundS(expr, n_d), where n_d: number of digits to keep.
    '''
    return expr.xreplace({n.evalf():round(n,n_d)}
    for n in expr.atoms(sp.Number)})
```

```
In [ ]: def maxminS(f, title="value"):
            '''Find maximum location and values of a symbolic function
            ndg = 8 # number of digits
            df = sp.diff(f, x)
            ddf = sp.diff(df, x)
            df0_points = sp.solve(df, x) #find stationary points
            df0s = [roundS(point.evalf(),ndg) for point in df0 points]
            print(f"Stationary points for {title}: {df0s}")
            for point in df0s:
                fv = roundS((f.subs(x,point)).evalf(), ndg)
                ddfv = ddf.subs({x:point})
                if ddfv < 0:</pre>
                    print(f"At x={point}, local maximum {title}={fv}")
                elif ddfv > 0:
                    print(f"At x={point}, local minimum {title}={fv}")
                else:
                    print(f"At x={point}, max or min {title}={fv}")
            df0s.append(0) # Add in points on the boundaries
            df0s.append(1)
            fs_df0 = [f.subs(x, point).evalf() for point in df0s]
            f_{max} = roundS(max(fs_df0), ndg)
            x max=[pnt for pnt in df0s if ma.isclose(f.subs(x,pnt),max(fs df0))]
            print(f"\nAt x={x_max}, Max {title}={f_max}\n")
            return x_max, f_max
In [ ]: maxMX, maxM = maxminS(vx[2], title="moment")
       Stationary points for moment: [2.00000000000000]
       At x=2.000000000000000, local minimum moment=1250.0000000000000
       At x=[0], Max moment=2250.00000000000
In [ ]: | maxstress = -y*(maxM/Iact)
        print(f"\nAt x={maxMX}, Max Normal Stress={maxstress*(10**-6)}\n")
```

At x=[0], Max Normal Stress=-22.5000000000000

#### **Question 2:**

Consider a beam with two spans. Span-1 has a length  $l_1$ , and span-2 has a length  $l_2$ . The beam is simple-simple supported (s-s-s), as shown in the following image. It is subjected to a distributed force over span-1.



A beam with two spans. It is simple-simple-simple supported.

- 1. Derive formulas for solutions, including deflection, cross-section rotation, moment, and shear force in the beam.
- 2. Set all the variables with unit values, plot the distribution of all these solutions.
- 3. Discuss about the results obtained.

1.

$$l_1$$

$$egin{align*} rac{d^4v}{dx^4} &= rac{b_y}{EI_z} \ rac{d^3v}{dx^3} &= V = xrac{b_y}{EI_z} + c_3 \ rac{d^2v}{dx^2} &= M = x^2rac{b_y}{2EI_z} + xc_3 + c_2 \ rac{dv}{dx} &= heta = x^3rac{b_y}{6EI_z} + rac{1}{2}x^2c_3 + xc_2 + c_1 \ v &= x^4rac{b_y}{24EI_z} + rac{1}{6}x^3c_3 + rac{1}{2}x^2c_2 + xc_1 + c_0 \end{align}$$

$$b_y = q$$

Beam is pinned at x = 0, therefore:

$$v = 0$$

$$M = 0$$

at 
$$x=0$$

$$v = 0 = 0^4 rac{q}{24EI_z} + rac{1}{6}0^3 c_3 + rac{1}{2}0^2 c_2 + 0c_1 + c_0$$
  
 $c_0 = 0$ 

$$M = 0 = 0^2 \frac{q}{2EI_z} + 0c_3 + c_2$$
  
 $c_2 = 0$ 

Beam is pinned at  $x=l_1$ , therefore:

$$v = 0$$

$$M = 0$$

at  $x=l_1$ , however since there is a second span,  $M=M_{l_1}$ 

$$egin{align} rac{d^2v}{dx^2} &= M_{l_1} = {l_1}^2rac{q}{2EI_z} + l_1c_3 \ rac{M_{l_1}}{l_1} &= {l_1}rac{q}{2EI_z} + c_3 \ c_2 &= rac{M_{l_1}}{l_1} - l_2 - rac{q}{l_2} \ \end{array}$$

$$c_3 = rac{M_{l_1}}{l_1} - l_1 rac{q}{2EI_z}$$

$$M = x^2 rac{q}{2EI_z} + x (rac{M_{l_1}}{l_1} - l_1 rac{q}{2EI_z})$$

$$M=x(rac{qx}{2EI_z}+rac{M_{l_1}}{l_1}-rac{ql_1}{2EI_z})$$

$$M=rac{ql_1}{2}-rac{M_{l_1}}{l_1}-qx$$

$$v = 0 = {l_1}^4 rac{q}{24EI_z} + rac{1}{6}{l_1}^3 (rac{M_{l_1}}{l_1} - {l_1}rac{q}{2EI_z}) + l_1 c_1$$

$$0 = {l_1}^4 rac{q}{24EI_z} + rac{1}{6} {M_{l_1}}{l_1}^2 - {l_1}^4 rac{q}{12EI_z} + l_1 c_1$$

$$0=rac{M_{l_1}l_1}{6}-rac{q{l_1}^3}{24EI_z}+c_1$$

$$c_1 = rac{q{l_1}^3}{24EI_z} - rac{M_{l_1}l_1}{6}$$

$$v=x^4rac{q}{24EI_z}-x^3rac{ql_1}{12EI_z}+x^3rac{M_{l_1}}{6l_1EI_z}+xrac{ql_1^3}{24EI_z}-xrac{M_{l_1}l_1}{6EI_z}$$

$$heta = x^3 rac{q}{6EI_z} - x^2 rac{ql_1}{4EI_z} + x^2 rac{M_{l_1}}{2l_1EI_z} + rac{ql_1^3}{24EI_z} - rac{M_{l_1}l_1}{6EI_z}$$

$$M=x^2rac{q}{2}-xrac{ql_1}{2}+xrac{M_{l_1}}{l_1}$$

$$V=qx-rac{ql_1}{2}+rac{M_{l_1}}{l_1}$$

```
In [ ]: # Define variables:
            title = [["Deflection", "Rotation"], ["Moment", "Shear force"]]
            E, I, 11, 12, q = symbols("E, I, 11, 12, q")
            x = symbols("x")
            # for displacement boundary conditions (DBCs):
            v10, \theta10, v11, \theta11 = symbols('v10, \theta10, v_11, \theta11') # DBCs for span-1
            v20, \theta20, v21, \theta21 = symbols('v20, \theta20, v_21, \theta_21') # DBCs for span-2
            # for force boundary conditions (FBCs):
            V10, M10, V11, M11 = symbols('V10, M10, V 11, M 11') # FBCs for span-1
            V20, M20, V21, M21 = symbols('V20, M20, V_21, M_21') # FBCs for span-2
In [ ]: # Distributed force is zero for span-1
            by = q \# BC is s-s
            v1 = solver1D4(E,I,by,l1, v10,\theta10,v11,\theta11, V10, M10, V11, M11,key='s-s')
            print(f'Is solution correct? {by/E/I==v1[4]}; The solution v(x) is:')
            v1[0] # solution of general displacement function
          Outputs form solver1D4(): v, \theta, M, V, qy
          Is solution correct? True; The solution v(x) is:
           v_{10} - rac{v_{10}x}{l_1} + rac{v_{1l}x}{l_1} - rac{M_{10}l_1x}{3EI} + rac{M_{10}x^2}{2EI} - rac{M_{10}x^3}{6EIl_1} - rac{M_{1l}l_1x}{6EI} + rac{M_{1l}x^3}{6EIl_1} + rac{l_1^3qx}{24EI} - rac{l_1qx^3}{12EI}
In [ ]: # Set known BCs for span-1:
            v1x=[v1[i].subs(\{v10:0,M10:0,v11:0\})  for i in range(len(v1))]
            v1x # solution of v, \theta, Mx, Vx, qy
 \frac{\mathsf{Out[\ ]:}}{6EI} \left[ -\frac{M_{1l}l_1x}{6EI} + \frac{M_{1l}x^3}{6EIl_1} + \frac{l_1^3qx}{24EI} - \frac{l_1qx^3}{12EI} + \frac{qx^4}{24EI}, \right. \\ \left. -\frac{M_{1l}l_1}{6EI} + \frac{M_{1l}x^2}{2EIl_1} + \frac{l_1^3q}{24EI} - \frac{l_1qx^2}{4EI} \right] 
            l_2
            \frac{d^4v}{dx^4} = \frac{b_y}{EI_z}
            \frac{d^3v}{dx^3} = V = x\frac{b_y}{EL} + c_3
            rac{d^2 v}{dx^2} = M = x^2 rac{b_y}{2EI_z} + xc_3 + c_2
            rac{dv}{dx} = 	heta = x^3 rac{b_y}{6EL} + rac{1}{2} x^2 c_3 + x c_2 + c_1
            v = x^4 rac{b_y}{24ET} + rac{1}{6}x^3c_3 + rac{1}{2}x^2c_2 + xc_1 + c_0
            b_{\nu}=0
```

$$egin{aligned} rac{d^4v}{dx^4} &= 0 \ rac{d^3v}{dx^3} &= V = c_3 \ rac{d^2v}{dx^2} &= M = xc_3 + c_2 \ rac{dv}{dx} &= heta = rac{1}{2}x^2c_3 + xc_2 + c_1 \ v &= rac{1}{6}x^3c_3 + rac{1}{2}x^2c_2 + xc_1 + c_0 \end{aligned}$$

Conditions from previous deirvation at  $x=l_1$   $\upsilon=0$   $M=M_{l_1}$ 

at  $x=l_1$ 

Treat this as its own function, so  $x=l_1$  is x=0:  $v=0=\frac{1}{6}0^3c_3+\frac{1}{2}0^2c_2+0c_1+c_0$ 

$$c_0 = 0$$

$$M = M_{l_1} = 0c_3 + c_2 \ c_2 = M_{l_1}$$

Beam is pinned at  $x=l_2$ , therefore:

$$v = 0$$

$$M = 0$$

at 
$$x=l_2$$

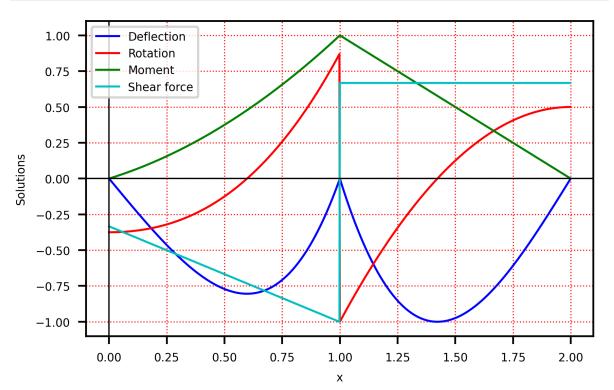
$$M=0=l_2c_3+M_{l_1} \ c_3=M_{l_1}$$

$$egin{aligned} v &= 0 = rac{1}{6} l_2{}^3 M_{l_1} + rac{1}{2} l_2{}^2 M_{l_1} + l_2 c_1 \ c_1 &= -rac{1}{6} l_2{}^3 M_{l_1} - rac{1}{2} l_2{}^2 M_{l_1} \end{aligned}$$

$$egin{align} v &= -x^3rac{M_{l_1}}{6EI_zl_2} + x^2rac{M_{l_1}}{2EI_z} - xrac{M_{l_1}l_2}{3EI_z} \ heta &= -x^2rac{M_{l_1}}{2EI_zl_2} + xrac{M_{l_1}}{EI_z} - rac{M_{l_1}l_2}{3EI_z} \ M &= -xrac{M_{l_1}}{l_2} + M_{l_1} \ V &= rac{M_{l_1}}{l_2} \ \end{pmatrix}$$

```
In [ ]: # General solution for span-2: (constant q)
                                # BC is simple-clamp
                                v2 = solver1D4(E,I,0,12, v20,\theta20,v21, \theta21, V20, M20, V21, M21,key='s-s')
                                 print(f'Is solution correct? {by/E/I==v2[4]}; The solution u(x) is:')
                                v2 # solution of general displacement function
                           Outputs form solver1D4(): v, \theta, M, V, qy
                           Is solution correct? False; The solution u(x) is:
 \begin{array}{c} \texttt{Out[]:} \quad \left( v_{20} - \frac{v_{20}x}{l_2} + \frac{v_{2l}x}{l_2} - \frac{M_{20}l_2x}{3EI} + \frac{M_{20}x^2}{2EI} - \frac{M_{20}x^3}{6EIl_2} - \frac{M_{2l}l_2x}{6EI} + \frac{M_{2l}x^3}{6EIl_2}, \right. \\ \left. - \frac{v_{20}}{l_2} + \frac{v_{2l}}{l_2} - \frac{w_{20}l_2x}{l_2} + \frac{w_{2l}}{l_2} - \frac{w_{20}l_2x}{l_2} + \frac{w_{2l}}{l_2} - \frac{w_{20}l_2x}{l_2} + \frac{w_{2l}}{l_2} - \frac{w_{20}l_2x}{l_2} + \frac{w_{2l}l_2x}{l_2} + \frac{w_{2l}l_2x}{l_2} + \frac{w_{2l}l_2x}{l_2} - \frac{w_{20}l_2x}{l_2} + \frac{w_{2l}l_2x}{l_2} + \frac{w_{2l}l_2x
 In [ ]: # Set known BCs for span-2:
                                v2x=[v2[i].subs(\{v20:0,M20:M11,v21:0,M21:0\})] for i in range(len(v2))]
                                v2x # solution of v, \theta, Mx, Vx, qy
 \left[ -\frac{M_{1l}l_2x}{3EI} + \frac{M_{1l}x^2}{2EI} - \frac{M_{1l}x^3}{6EIl_2}, \right. \\ \left. -\frac{M_{1l}l_2}{3EI} + \frac{M_{1l}x}{EI} - \frac{M_{1l}x^2}{2EIl_2}, \right. \\ \left. M_{1l} - \frac{M_{1l}x}{l_2}, \right. \\ \left. \frac{M_{1l}}{l_2}, \right. \\ \left. \frac{M_{1l}}{l_2} + \frac{M_{1l}x}{2EIl_2}, \right. \\ \left. \frac{M_{1l}x}{l_2} - \frac{M_{1l}x}{l_2}, \right. \\ \left. \frac{M_{1l}x}{l_2} - \frac{M_{1l}x^2}{2EIl_2}, \right. \\ \left. \frac{M_{1l}x}{l_2} - \frac{M_{1l}x}{l_2}, \right. \\ \left. \frac{M_{1l}x}{l_2} - \frac{M_{
                                        2.
                                 Using the following values: l_1=1m, l_2=1m, q=50n/m, E=1*10^{10}Pa, I_z=1.
                                 Using this, M_{l_1} = l_1(ql_1) = 50.
 In []: sln1x=[v1x[i].subs({q:50,11:1,12:1,E:1*(10**10),I:1,M11:50})) for i in range(len(v1x)
                                 sln2x=[v2x[i].subs({11:1,12:1,E:1*(10**10),I:1,M11:50})  for i in range(len(v2x))]
 In [ ]: def plot_fs(x, y, labels, xlbl, ylbl, *p_name):
                                                '''plot multiple x-y curves in one figure.
                                                x: x data, list of 1D numpy array;
                                               y: y data, list of 1D numpy array
                                                labels: labels for these curves
                                                xlbl: for ax.set_xlabel()
                                               ylbl: for ax.set_ylabel()
                                                p_name: string, name of the plot.
                                                colors = ['b', 'r', 'g', 'c', 'm', 'k', 'y', 'w']
                                                plt.rcParams.update({'font.size': 5})  # settings for plots
                                                fig_s = plt.figure(figsize=(4,2.5))
                                                ax = fig_s.add_subplot(1,1,1)
                                                for i in range(len(y)):
                                                                plt.plot(x[i], y[i], label=labels[i], color=colors[i], lw=0.9)
                                                ax.grid(c='r', linestyle=':', linewidth=0.5)
                                                ax.axvline(x=0, c="k", lw=0.6); ax.axhline(y=0, c="k", lw=0.6)
                                                ax.set_xlabel(xlbl); ax.set_ylabel(ylbl)
                                                ax.legend() #loc='center right', bbox_to_anchor=(1, 0.5))
                                                #plt.title('x-y curves'+p name+'')
                                                plt.savefig('images/'+p_name[0]+'.png',dpi=500,bbox_inches='tight')
                                                plt.show()
```

```
In [ ]: np.set printoptions(formatter={'float': '{: 0.4e}'.format})
        11 = 1.0; 12 = 1.0
        dx = 0.002
        x1 = np.arange(0.0, 11, dx) # coordinates in l1, left span
        x2 = np.arange(0.0, 12, dx) # coordinates in l2, right span
        # convert Sympy solution func. to numpy arrays.
        n1 = len(sln1x)-1
        n2 = len(sln1x)-1
        x = symbols("x")
        sln1 = [sp.lambdify(x, sln1x[i], 'numpy') for i in range(n1)]
        sln2 = [sp.lambdify(x, sln2x[i], 'numpy') for i in range(n2)]
        sx1 = [sln1[i](x1) for i in range(n1)]
        sx2 = [sln2[i](x2) for i in range(n2)]
        # The shear force function could be a constant. Need to generate array
        if type(sx1[-1]) != np.ndarray: # in case, func. is a constant
            sx1[-1] = float(sx1[-1])*np.ones_like(x1)
        if type(sx2[-1]) != np.ndarray: # in case, func. is a constant
            sx2[-1] = float(sx2[-1])*np.ones_like(x2)
        X = np.concatenate((x1, 11+x2)) # put the results in a single array
        sx = [np.concatenate((sx1[i], sx2[i])) for i in range(n1)]
        sx_max = [np.max(abs(sx[i])) for i in range(n1)]
        # Plot the distribution curves:
        x_data = [X for i in range(n1)] # put x coordinates to a list
        y_data = [sx[i]/sx_max[i] for i in range(n1)] # func. values to a list
        labels=["Deflection", "Rotation", "Moment", "Shear force"]
        plot_fs(x_data, y_data, labels, 'x', 'Solutions', 'temp')
```



The results obtained seemed to be in line with what is expected. The central pin acts like a fulcrum for the beam, and all 4 diagrams show that the beam reacts around this point.