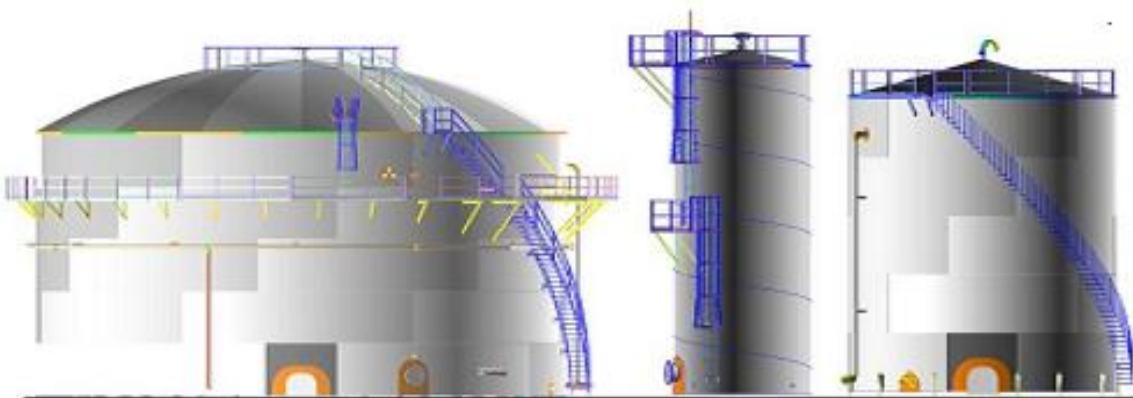


AMETANK REPORT



AMETank

**Field Erected and Shop Built Storage Tanks
Engineering Application Software**

The following report is subject to the disclaimer statement as stated in the Disclaimer and Special Notes section at the end.

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Project Design Data and Summary

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Project Data

Job : 2025-06-19-00-40
Date of Calcs. : 20-Jun-2025
Mfg. or Insp. Date :
Designer : Melior
Project :
Tag ID : Q9270 API
Plant :
Plant Location :
Site :
Design Basis : API-650 13th Edition Errata 1, 2021
Annexes Used : E, F, J, M, S

Design Parameters and Operating Conditions

Design Parameters

Design Internal Pressure = 0.1084 psi or 3 inh2o
Design External Pressure = -0.0361 psi or -1 inh2o

D of Tank = 10 ft
OD of Tank = 10.0313 ft
ID of Tank = 10 ft
CL of Tank = 10.0156 ft
Shell Height = 30 ft
S.G of Contents = 1.1
S.G of Hydrotest = 1
Hydrotest Liquid Level = 30 ft
Max Design Liq. Level = 30 ft
Max Operating Liq. Level = 30 ft
Min Liq. Level = 1 ft
Design Temperature = 375 °F
MDMT (Minimum Design Metal Temperature) = -20 °F
Tank Joint Efficiency = 0.7
Ground Snow Load = 0 psf
Roof Live Load = 20 psf
Additional Roof Dead Load = 0 psf

Appendix F Data

Failure pressure (Pf) = 3.6067 psi
Maximum design pressure (P_max) = 2.2841 psi

Wind Load Basis: ASCE 7-16
3 Second Gust Wind Speed (entered), Vg = 105 mph

Design Wind Speed, $V = V_g = 105$ mph

Seismic Method: API-650 - ASCE7 Mapped(S_s & S_1)

Seismic Use Group = II

Site Class = C

T_L (sec) = 12

S_s (g) = 0.24

S_1 (g) = 0.093

A_v (g) = 0.0896

$Q = 0.6667$

Importance Factor = 1.25

Design Remarks

Summary Results Shell

Shell #	Width (in)	Material	CA (in)	JE	Min Yield Strength (psi)	Tensile Strength (psi)	Reduction Factor	S_d (psi)	S_t (psi)
1	60	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000
2	60	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000
3	60	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000
4	60	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000
5	60	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000
6	59.375	A240-316	0	0.7000	21,875	75,000	0.8575	19,725	27,000

(continued)

Shell #	Weight (lbf)	Weight CA (lbf)	t-min Erection (in)	t-Des (in)	t-Test (in)	t-min Seismic (in)	t-min Ext-Pe (in)	t-min (in)	t-Actual (in)
1	1,228	1,228	0.1875	0.0601	0.0399	0.0646	NA	0.1875	0.1875
2	1,228	1,228	0.1875	0.0497	0.033	0.0539	NA	0.1875	0.1875
3	1,228	1,228	0.1875	0.0394	0.0261	0.0431	NA	0.1875	0.1875
4	1,228	1,228	0.1875	0.029	0.0193	0.0324	NA	0.1875	0.1875
5	1,228	1,228	0.1875	0.0186	0.0124	0.0218	NA	0.1875	0.1875
6	1,215	1,215	0.1875	0.0083	0.0055	0.0111	NA	0.1875	0.1875

(continued)

Shell #	Status
1	OK

2	OK
3	OK
4	OK
5	OK
6	OK

Total Weight of Shell = 7,378.2999 lbf

Roof

Type = Self Supported Conical Roof

Plates Material = A240-316

t.required = 0.1875 in

t.actual = 0.1875 in

Roof corrosion allowance = 0 in

Roof Joint Efficiency = 0.7

Plates Overlap Weight = 0 lbf

Plates Weight = 639.6276 lbf

Bottom

Type : Flat Bottom Non Annular

Bottom Material = A240-316

t.required = 0.1875 in

t.actual = 0.1875 in

Bottom corrosion allowance = 0 in

Bottom Joint Efficiency = 0.7

Total Weight of Bottom = 639.55 lbf

Top Member

Type = Detail B

Size = L2x2x1/4

Material = A240-316

Weight = 100.397 lbf

Anchors

Quantity = 4

Size = 1 in

Material = A36

Bolt Hole Circle Radius = 5.1822 ft

Nameplate Information

Pressure Combination Factor	0.4
Design Standard	API-650 13th Edition Errata 1, 2021
Appendices Used	E, F, J, M, S
Roof	A240-316 : 0.1875 in
Shell (1)	A240-316 : 0.1875 in
Shell (2)	A240-316 : 0.1875 in
Shell (3)	A240-316 : 0.1875 in

Shell (4)	A240-316 : 0.1875 in
Shell (5)	A240-316 : 0.1875 in
Shell (6)	A240-316 : 0.1875 in
Bottom	A240-316 : 0.1875 in

Roof Design Details [Back](#)

Roof Type = Cone

Structure Support Type = None (Self Supported)

Material Properties

Material = A240-316

Minimum Tensile Strength (Sut) = 75.0e3 psi

As per API-650 S.5.b, Minimum Yield Strength (Sy) = 21.8750e3 psi

As per API-650 S.2b, Allowable Design Stress (Sd) = 19.7250e3 psi

Density (d) = 0.290 lb/in³

Modulus of Elasticity at Design Temperature (E) = 26.550e6 psi

Geometry

Rh = Horizontal Radius (in)

slope = Slope (Rise / Run)

Rh = 60.7735 in

slope = 0.166667

Description	Variable	Equation	Value	Unit
Slope Angle	Theta	ARCTAN(slope)	9.46232	deg
Angle With Vertical Line	Alpha	90 - Theta	80.5377	deg
Height	h	Rh * TAN(Theta)	10.1289	in
Surface Area	A	(pi * (Rh ²)) / COS(Theta)	11.7633e3	in ²
Center of Gravity from Base	CG	h / 3	3.37631	in
Vertical Projected Area	Av	Rh * h	615.570	in ²
Horizontal Projected Area	Ah	pi * (Rh ²)	11.6032e3	in ²
Volume	V	(pi * (Rh ²) * h) / 3	39.1760e3	in ³

Weights

d-ins = Insulation Density (lb/ft³)

DL-add = Added dead load (psf)

t = Plates Thickness (in)

t-ins = Insulation Thickness (in)

Wr-pl-add = Additional Weight (lb)

d-ins = 8.0 lb/ft³

DL-add = 0.0 psf

t = 0.18750 in

t-ins = 0.0 in

Wr-pl-add = 0.0 lb

Description	Variable	Equation	Value	Unit
Plates Nominal Weight	Wr-pl	(A * d * t) + Wr-pl-add	639.628	lb

Plates Corroded Weight	Wr-pl-corr	$(A * d * (t - CA)) + Wr-pl-add$	639.628	lb
New Plates Dead Load Pressure	DL-pl	$Wr-pl / Ah$	7.93801	psf
Corroded Plates Dead Load Pressure	DL-pl-corr	$Wr-pl-corr / Ah$	7.93801	psf
Insulation Weight	Wr-ins	$t-ins * d-ins * A$	0.0	lb
Insulation Dead Load Pressure	DL-ins	$Wr-ins / Ah$	0.0	psf
Dead Load	DL	$DL-pl + DL-ins + DL-add$	7.93801	psf
Total Nominal Dead Weight	Wr-DL	$DL * Ah$	639.628	lb
Additional Dead Weight	Wr-DL-add	$DL-add * Ah$	0.0	lb

Loads

B = Maximum Gravity Load Combination Based on Balanced Snow Load (psf)

e.1b = Gravity Loads Combination 1 Based on Balanced Snow Load per *API 650 Section 5.2.2* (psf)

e.1u = Gravity Loads Combination 1 Based on Unbalanced Snow Load per *API 650 Section 5.2.2* (psf)

e.2b = Gravity Loads Combination 2 Based on Balanced Snow Load per *API 650 Section 5.2.2* (psf)

e.2u = Gravity Loads Combination 2 Based on Unbalanced Snow Load per *API 650 Section 5.2.2* (psf)

Fpe = External Pressure Combination Factor

Lr = Minimum Roof Live Load (psf)

max-gravity-load = Maximum Gravity Load (psf)

Pe = Design External Pressure (psf)

S = Ground Snow Load (psf)

Sb = Balanced Snow load per *API 650 Section 5.2.1 (h)* (psf)

Su = Unbalanced Snow load per *API 650 Section 5.2.1 (h)* (psf)

U = Maximum Gravity Load Combination Based on Unbalanced Snow Load (psf)

W-max-gravity-load = Maximum Gravity Load Weight (lb)

Fpe = 0.40

Lr = 20.0 psf

Pe = 5.20219 psf

S = 0.0 psf

Sb = $0.84 * S$

Sb = $0.84 * 0.0$

Sb = 0.0 psf

Su = Sb

Su = 0.0

Su = 0.0 psf

e.1b = $DL + \text{MAX}(Lr, Sb) + (Fpe * Pe)$

e.1b = $7.93801 + \text{MAX}(20.0, 0.0) + (0.40 * 5.20219)$

e.1b = 30.0189 psf

e.2b = $DL + Pe + (0.4 * \text{MAX}(Lr, Sb))$

e.2b = $7.93801 + 5.20219 + (0.4 * \text{MAX}(20.0, 0.0))$

e.2b = 21.1402 psf

B = $\text{MAX}(e.1b, e.2b)$

B = $\text{MAX}(30.0189, 21.1402)$

B = 30.0189 psf

e.1u = $DL + \text{MAX}(Lr, Su) + (Fpe * Pe)$

$$e.1u = 7.93801 + \text{MAX}(20.0, 0.0) + (0.40 * 5.20219)$$

$$e.1u = 30.0189 \text{ psf}$$

$$e.2u = DL + Pe + (0.4 * \text{MAX}(Lr, Su))$$

$$e.2u = 7.93801 + 5.20219 + (0.4 * \text{MAX}(20.0, 0.0))$$

$$e.2u = 21.1402 \text{ psf}$$

$$U = \text{MAX}(e.1u, e.2u)$$

$$U = \text{MAX}(30.0189, 21.1402)$$

$$U = 30.0189 \text{ psf}$$

$$\text{max-gravity-load} = \text{MAX}(B, U)$$

$$\text{max-gravity-load} = \text{MAX}(30.0189, 30.0189)$$

$$\text{max-gravity-load} = 30.0189 \text{ psf}$$

$$W\text{-max-gravity-load} = \text{max-gravity-load} * Ah$$

$$W\text{-max-gravity-load} = 30.0189 * 80.5779$$

$$W\text{-max-gravity-load} = 2.41886e3 \text{ lb}$$

Erection Requirements

t-erec-req = Minimum Erection Thickness Including Corrosion Allowance (in)

As per API-650 5.10.2.2, Minimum Erection Thickness (t-erec) = 0.18750 in

$$t\text{-erec-req} = t\text{-erec} + CA$$

$$t\text{-erec-req} = 0.18750 + 0.0$$

$$t\text{-erec-req} = 0.18750 \text{ in}$$

Gravity Loads Thickness Calculation

B-max = Max Gravity Load Based on Roof Actual Thickness (Balanced Snow Load) (psf)
 Pe-max-1 = MAWV Based on Gravity Loads Combination 1 per *API 650 Section 5.2.2* (psf)
 t-calc-1 = Calculated Thickness Based on Balanced Snow Load (in)

$$t\text{-calc-1} = (((2 * 12 * D) / \text{SIN}(\text{Theta})) * \text{SQRT}((B / (144 * E)))) + CA$$

$$t\text{-calc-1} = (((2 * 12 * 10.0) / \text{SIN}(9.46232)) * \text{SQRT}((30.0189 / (144 * 26.550e6)))) + 0.0$$

$$t\text{-calc-1} = 0.129359 \text{ in}$$

$$B\text{-max} = (((t\text{-actual} - CA) * (\text{SIN}(\text{Theta}) / (2 * 12 * D)))^2 * 144 * E$$

$$B\text{-max} = (((0.18750 - 0.0) * (\text{SIN}(9.46232) / (2 * 12 * 10.0)))^2 * 144 * 26.550e6$$

$$B\text{-max} = 63.0675 \text{ psf}$$

$$Pe\text{-max-1} = \text{MAX}(((L_{\text{max}} - DL - \text{MAX}(Lr, S)) / Fpe), 0)$$

$$Pe\text{-max-1} = \text{MAX}(((63.0675 - 7.93801 - \text{MAX}(20.0, 0.0)) / 0.40), 0)$$

$$Pe\text{-max-1} = 87.8236 \text{ psf}$$

Pe-max-2 = MAWV Based on Gravity Loads Combination 1 per *API 650 Section 5.2.2* (psf)
 t-calc-2 = Calculated Thickness Based on Unbalanced Snow Load (in)
 U-max = Max Gravity Load Based on Roof Actual Thickness (Unbalanced Snow Load) (psf)

$$t\text{-calc-2} = (((2 * 12 * D) / \text{SIN}(\text{Theta})) * \text{SQRT}((U / (1.33 * 144 * E)))) + CA$$

$$t\text{-calc-2} = (((2 * 12 * 10.0) / \text{SIN}(9.46232)) * \text{SQRT}((30.0189 / (1.33 * 144 * 26.550e6)))) + 0.0$$

$$t\text{-calc-2} = 0.112168 \text{ in}$$

$$U\text{-max} = (((t\text{-actual} - CA) * (\text{SIN}(\text{Theta}) / (2 * 12 * D)))^2 * 1.33 * 144 * E$$

$$U\text{-max} = (((0.18750 - 0.0) * (\text{SIN}(9.46232) / (2 * 12 * 10.0)))^2 * 1.33 * 144 * 26.550e6$$

U-max = 83.8797 psf

Pe-max-2 = MAX(((L_max - DL - MAX(Lr , S)) / Fpe) , 0)

Pe-max-2 = MAX(((83.8797 - 7.93801 - MAX(20.0 , 0.0)) / 0.40) , 0)

Pe-max-2 = 139.854 psf

API-650 Appendix F Internal Pressure Thickness Calculation

Alpha = Half Apex Angle of Cone Roof (deg)

Ca = Corrosion Allowance (in)

DL-add = Additional Dead Load (psf)

DL-plates = Plates Dead Load (psf)

E = Joint Efficiency

L-max-F = Maximum Roof Load Based on Installed Thickness (psi)

P = Internal Design Pressure (psi)

P-max-F = Maximum Design Pressure (psi)

P-net = Net Internal Pressure (psi)

Rt = Nominal Tank Radius (in)

Sd = Allowable Stress for the Design Condition (psi)

t = Roof Actual Thickness (in)

t-calc-F = Minimum Roof Thickness Required for Internal Pressure per *API-650 F.6* (in)

Alpha = 80.5377 deg

Ca = 0.0 in

DL-add = 0.0 psf

DL-plates = 7.93801 psf

E = 0.70

P = 0.108379 psi

Rt = 60.0 in

Sd = 19.7250e3 psi

t = 0.18750 in

P-net = MAX((P - DL-plates - DL-add) , 0)

P-net = MAX((0.108379 - 0.0551250 - 0.0) , 0)

P-net = 0.0532538 psi

t-calc-F = ((P-net * Rt) / (COS(Alpha) * Sd * E)) + Ca

t-calc-F = ((0.0532538 * 60.0) / (COS(80.5377) * 19.7250e3 * 0.70)) + 0.0

t-calc-F = 0.00140763 in

L-max-F = ((t - Ca) * (COS(Alpha) * Sd * E)) / Rt

L-max-F = ((0.18750 - 0.0) * (COS(80.5377) * 19.7250e3 * 0.70)) / 60.0

L-max-F = 7.09356 psi

P-max-F = L-max-F + DL-plates + DL-add

P-max-F = 7.09356 + 0.0551250 + 0.0

P-max-F = 7.14868 psi

Required Thickness

MAWP-Roof = Maximum Allowable Working Pressure (psi)

MAWV-Roof = Maximum Allowable Working Vacuum (psf)

t-act = Installed Thickness (in)

t-req = Required Thickness (in)

t-act = 0.18750 in

$t_{req} = \text{MAX}(t_{erec-req}, t_{calc-1}, t_{calc-2}, t_{calc-F})$
 $t_{req} = \text{MAX}(0.18750, 0.129359, 0.112168, 0.00140763)$
 $t_{req} = 0.18750 \text{ in}$

$t \geq t_{req} \implies \text{PASS}$

MAWP-Roof = P-max-F
MAWP-Roof = 7.14868
MAWP-Roof = 7.14868 psi

MAWV-Roof = MIN(Pe-max-1, Pe-max-2)
MAWV-Roof = MIN(87.8236, 139.854)
MAWV-Roof = 87.8236 psf

Top Member Detail B Design [Back](#)

DLR = Nominal Weight of Roof Plates and Attached Structural (lbf)

DLS = Nominal Weight of Shell Plates and Framing (lbf)

DLS = Ws + W_framing

DLS = 7.37830e3 + 103.391

DLS = 7.48169e3 lbf

DLR = Wr + W_structural

DLR = 639.628 + 260.865

DLR = 900.493 lbf

Material Properties

Material = A240-316

Minimum Tensile Strength (Sut) = 75.0e3 psi

As per API-650 S.5.b, Minimum Yield Strength (Sy) = 21.8750e3 psi

Compression Ring Detail b Properties

A_detail = Detail Total Area (in^2)

A_roof = Contributing Roof Area (in^2)

A_shell = Contributing Shell Area (in^2)

A_sum = Total area (in^2)

c_combined = Combined centroid (in)

d_shell = Shell centroid (in)

d_stiff = Stiffener centroid (in)

e1 = Distance from neutral axis to edge 1 (inside) (in)

e2 = Distance from neutral axis to edge 2 (outside) (in)

I_1 = moment of inertia of first body (in^4)

I_2 = moment of inertia of second body (in^4)

I_combined = Combined moment of inertia (in^4)

I_shell = Contributing Shell Moment Of Inertia (in^4)

I_sum = Sum of moments of inertia's (in^4)

R2 = Length of Normal to Head (in)

S = Combined stiffener shell section modulus (in^3)

Wc = Maximum Width of Participating Shell per *API-650 Figure F-2* (in)

Wh = Maximum Width of Participating Head per *API-650 Figure F-2* (in)

$R2 = (ID / 2) / \sin(\theta)$

$R2 = (120.0 / 2) / \sin(9.46232)$

R2 = 364.966 in

$Wh = 0.3 * \sqrt{R2 * (th - CA-head)}$

$Wh = 0.3 * \sqrt{364.966 * (0.18750 - 0.0)}$

Wh = 2.48169 in

$Wc = 0.6 * \sqrt{((ID / 2) * (tc-nominal - CA-shell))}$

$Wc = 0.6 * \sqrt{((120.0 / 2) * (0.18750 - 0.0))}$

Wc = 2.01246 in

Angle Size L2X2X1/4 Section Properties

Description	Variable	New	Corroded	Unit
-------------	----------	-----	----------	------

Weight	W	3.190	3.190	lbf/ft
Cross Sectional Area	A	0.9440	0.9440	in^2
Moment Of Inertia About X Axis	Ix	0.3460	0.3460	in^4
Moment Of Inertia About Y Axis	Iy	0.3460	0.3460	in^4
Section Modulus About X Axis	Sx	0.2440	0.2440	in^3
Section Modulus About Y Axis	Sy	0.2440	0.2440	in^3
Centroid X Coords	cx	0.5860	0.5860	in
Centroid Y Coords	cy	0.5860	0.5860	in
Angle Long Leg Length	L1-angle	2.0	2.0	in
Angle Short Leg Length	L2-angle	2.0	2.0	in
Angle Thickness	t-angle	0.250	0.250	in

$$I_{shell} = ((Wc - h) * ((tc-nominal - CA-shell)^3)) / 12$$

$$I_{shell} = ((2.01246 - 0.250) * ((0.18750 - 0.0)^3)) / 12$$

$$I_{shell} = 968.149e-6 \text{ in}^4$$

$$A_{shell} = (Wc - h) * (tc-nominal - CA-shell)$$

$$A_{shell} = (2.01246 - 0.250) * (0.18750 - 0.0)$$

$$A_{shell} = 0.330461 \text{ in}^2$$

$$A_{roof} = Wh * (th - CA-head)$$

$$A_{roof} = 2.48169 * (0.18750 - 0.0)$$

$$A_{roof} = 0.465317 \text{ in}^2$$

$$A_{detail} = A_{shell} + A_{roof} + A_{corr}$$

$$A_{detail} = 0.330461 + 0.465317 + 0.9440$$

$$A_{detail} = 1.73978 \text{ in}^2$$

Stiffener and Shell Combined Section Properties

Description	Variable	Equation	Value	Unit
Shell centroid	d_shell	(tc-nominal - CA-shell) / 2	0.093750	in
Stiffener centroid	d_stiff	cy + (tc-nominal - CA-shell)	0.77350	in
moment of inertia of first body	I_1	Ic + (Area * (Distance^2))	0.910797	in^4
moment of inertia of second body	I_2	Ic + (Area * (Distance^2))	0.00387260	in^4
Total area	A_sum	A_1 + A_2	1.27446	in^2
Sum of moments of inertia's	I_sum	I_1 + I_2	0.914670	in^4
Combined centroid	c_combined	((Centroid-1 * Area-1) + (Centroid-2 * Area-2)) / (Area-1 + Area-2)	0.597244	in
Combined moment of inertia	I_combined	Ic - (Area * (Distance^2))	0.460069	in^4
Distance from neutral axis to edge 1 (inside)	e1	c_combined	0.597244	in
Distance from neutral axis to edge 2 (outside)	e2	((tc-nominal - CA-shell) + L1-angle) - e1	1.59026	in

Combined stiffener shell section modulus	S	$I / \text{MAX}(d-1, d-2)$	0.289305	in^3
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$$d_{\text{shell}} = (tc - \text{nominal} - CA - \text{shell}) / 2$$

$$d_{\text{shell}} = (0.18750 - 0.0) / 2$$

$$d_{\text{shell}} = 0.093750 \text{ in}$$

$$d_{\text{stiff}} = cy + (tc - \text{nominal} - CA - \text{shell})$$

$$d_{\text{stiff}} = 0.5860 + (0.18750 - 0.0)$$

$$d_{\text{stiff}} = 0.77350 \text{ in}$$

$$I_1 = I_c + (\text{Area} * (\text{Distance}^2))$$

$$I_1 = 0.3460 + (0.9440 * (0.77350^2))$$

$$I_1 = 0.910797 \text{ in}^4$$

$$I_2 = I_c + (\text{Area} * (\text{Distance}^2))$$

$$I_2 = 968.149\text{e-}6 + (0.330461 * (0.093750^2))$$

$$I_2 = 0.00387260 \text{ in}^4$$

$$A_{\text{sum}} = A_1 + A_2$$

$$A_{\text{sum}} = 0.9440 + 0.330461$$

$$A_{\text{sum}} = 1.27446 \text{ in}^2$$

$$I_{\text{sum}} = I_1 + I_2$$

$$I_{\text{sum}} = 0.910797 + 0.00387260$$

$$I_{\text{sum}} = 0.914670 \text{ in}^4$$

$$c_{\text{combined}} = ((\text{Centroid-1} * \text{Area-1}) + (\text{Centroid-2} * \text{Area-2})) / (\text{Area-1} + \text{Area-2})$$

$$c_{\text{combined}} = ((0.77350 * 0.9440) + (0.093750 * 0.330461)) / (0.9440 + 0.330461)$$

$$c_{\text{combined}} = 0.597244 \text{ in}$$

$$I_{\text{combined}} = I_c - (\text{Area} * (\text{Distance}^2))$$

$$I_{\text{combined}} = 0.914670 - (1.27446 * (0.597244^2))$$

$$I_{\text{combined}} = 0.460069 \text{ in}^4$$

$$e1 = c_{\text{combined}}$$

$$e1 = 0.597244$$

$$e1 = 0.597244 \text{ in}$$

$$e2 = ((tc - \text{nominal} - CA - \text{shell}) + L1 - \text{angle}) - e1$$

$$e2 = ((0.18750 - 0.0) + 2.0) - 0.597244$$

$$e2 = 1.59026 \text{ in}$$

$$S = I / \text{MAX}(d-1, d-2)$$

$$S = 0.460069 / \text{MAX}(0.597244, 1.59026)$$

$$S = 0.289305 \text{ in}^3$$

Erection Requirement

As per API-650 5.1.5.9, Minimum Size of Top Corner Ring (Size-min) = L2x2x3/16

Minimum Section Modulus per Erection Requirement ($S_{x-\text{min}}$) = 0.1880 in^3

Section Modulus (S_x) = 0.289305 in^3

$S_x \geq S_{x-\text{min}} \Rightarrow \text{PASS}$

Design Requirements per API-650 Section 5

A_{roof} = Compression Region Required Area for Self Supported Cone Roof per *API-650 5.10.5.2* (in²)

F_a = Least Allowable Tensile Stress for the Materials in the Roof-To-Shell Joint (psi)

F_y = Minimum Specified Yield Strength (psi)

Max-p = Maximum Allowable Load for the Actual Resisting Area per *API-650 5.10.5.2* (psf)

p = Max Gravity Load per *API-650 5.2.2 e*) (psf)

F_y = 21.8750e3 psi

p = 30.0189 psf

F_a = 0.6 * F_y

F_a = 0.6 * 21.8750e3

F_a = 13.1250e3 psi

A_{roof} = (p * (D²)) / (8 * F_a * TAN(theta))

A_{roof} = (30.0189 * (10.0²)) / (8 * 13.1250e3 * TAN(9.46232))

A_{roof} = 0.171536 in²

A_{detail} >= A_{roof} ==> PASS

Max-p = (A_{detail} / (D²)) * 8 * F_a * TAN(theta)

Max-p = (1.73978 / (10.0²)) * 8 * 13.1250e3 * TAN(9.46232)

Max-p = 304.461 psf

Internal Pressure - Appendix F Requirements

A_{actual} = Area resisting compressive force (in²)

D = Tank nominal diameter (ft)

DLR = Nominal weight of roof plates and attached structural (lbf)

DLS = Nominal weight of shell plates and framing (lbf)

F_p = Internal Pressure Combination Factor

F_y = Minimum specified yield-strength of the materials in the roof-to-shell junction (psi)

ID = Tank inside diameter (ft)

MDL = Moment About the Shell-to-Bottom Joint from the Nominal Weight of the Shell and Roof Structural Supported by the Shell that is not Attached to the Roof Plate (ft.lbf)

MDLR = Moment About the Shell-to-Bottom Joint from the Nominal Weight of the Roof Plate Plus any Structural Components Attached to the Roof (ft.lbf)

MF = Moment About the Shell-to-Bottom Joint from Liquid Weight (ft.lbf)

M_w = Wind Moment From Horizontal Plus Vertical Wind Pressures (ft.lbf)

M_{ws} = Wind Moment From Horizontal Wind Pressure (ft.lbf)

Net_{uplift} = Net uplift due to internal pressure (lbf)

P = Design pressure (psi)

P_{uplift} = Uplift due to internal pressure per *API-650 F.1.2* (lbf)

Theta angle = Angle between the roof and a horizontal plane at the roof-to-shell junction (deg)

W_{add_DL} = Additional dead load weight (lbf)

W_{framing} = Weight of framing supported by the shell and roof (lbf)

W_r = Roof plates weight (lbf)

W_s = Shell plates weight (lbf)

W_{structural} = Weight of roof attached structural (lbf)

W_{total} = Weight of roof shell and attached-framing (lbf)

A_{actual} = 1.73978 in²

D = 10.0 ft

DLR = 900.493 lbf

DLS = 7.48169e3 lbf

F_p = 0.40

F_y = 21.8750e3 psi

ID = 10.0 ft
 MDL = 37.4085e3 ft.lbf
 MDLR = 4.50247e3 ft.lbf
 MF = 21.2058e3 ft.lbf
 Mw = 44.9275e3 ft.lbf
 Mws = 39.1098e3 ft.lbf
 P = 0.108379 psi
 Theta angle = 9.46232 deg
 W_add_DL = 0.0 lbf
 W_framing = 103.391 lbf
 Wr = 639.628 lbf
 Ws = 7.37830e3 lbf
 W_structural = 260.865 lbf

 $P_{uplift} = P * \pi * ((ID^2) / 4)$
 $P_{uplift} = 0.108379 * \pi * ((120.0^2) / 4)$
 $P_{uplift} = 1.22574e3 \text{ lbf}$

$W_{total} = W_r + W_s + W_{framing} + W_{structural} + W_{add_DL}$
 $W_{total} = 639.628 + 7.37830e3 + 103.391 + 260.865 + 0.0$
 $W_{total} = 8.38218e3 \text{ lbf}$

$Net_uplift = MAX((P_{uplift} - W_{total}), 0)$
 $Net_uplift = MAX((1.22574e3 - 8.38218e3), 0)$
 $Net_uplift = 0.0 \text{ lbf}$

$W_r < P_{uplift} \leq W_{total}$, Tank design should meet F.2 to F.7 requirements.

$P_{max_internal}$ = Maximum allowable internal pressure (psi)

P_f = Calculated failure pressure for frangible roof per *API 650, Section F.7* (inH2O)

P_{max} = Maximum design pressure (inH2O)

P_{max_F41} = Maximum design pressure per *API 650, Section F.4.1* (inH2O)

$P_{max_F41} = ((0.962 * A_{actual} * F_y * \tan(\theta)) / (D^2)) + ((0.245 * D_{LR}) / (D^2))$
 $P_{max_F41} = ((0.962 * 1.73978 * 21.8750e3 * \tan(9.46232)) / (10.0^2)) + ((0.245 * 900.493) / (10.0^2))$
 $P_{max_F41} = 63.2253 \text{ inH2O [2.28410 psi]}$

$P_f = (1.6 * P) - ((0.147 * D_{LR}) / (D^2))$
 $P_f = (1.6 * 63.2253) - ((0.147 * 900.493) / (10.0^2))$
 $P_f = 99.8368 \text{ inH2O [3.60673 psi]}$

$P_{max} = P_{max_F41}$
 $P_{max} = 63.2253$
 $P_{max} = 63.2253 \text{ inH2O [2.28410 psi]}$

$P \leq P_{max} \Rightarrow$ Design internal pressure is less than maximum allowable pressure per API-650 F.4

A_{F51} = Required area per *API 650 F.5.1* (in²)

$A_{F51} = ((D^2) * (P - ((0.245 * DLR) / (D^2)))) / (0.962 * F_y * \tan(\theta \text{ angle}))$
 $A_{F51} = ((10.0^2) * (3.0 - ((0.245 * 900.493) / (10.0^2)))) / (0.962 * 21.8750e3 * \tan(9.46232))$
 $A_{F51} = 0.0226326 \text{ in}^2$

$A_{actual} \geq A_{F51} \Rightarrow$ PASS

As per API-650 5.2.1 c), Maximum design internal pressure (P_{std}) = 2.50 psi

$P_{max_internal} = \text{MIN}(P_{std}, P_{max})$

$P_{max_internal} = \text{MIN}(2.50, 2.28410)$

$P_{max_internal} = 2.28410$ psi

Agitator Bridge Design [Back](#)

A-es-req = End Support Required Cross Sectional Area (in²)
A-platform = Platform Total Area (ft²)
As-available = End Support Compressive Stress Shell Available Area (in²)
As-required = Shell Required Compressive Area Around End Supports (in²)
CA-es = End Support Corrosion Allowance (in)
CA-ps = Primary Support Corrosion Allowance (in)
delta-d = Primary Support Static + Dynamic Loads Deflection (in)
delta-d-allowable = Primary Support Dynamic Load Allowable Deflection (in)
delta-sd = Primary Support Static + Dynamic Actual Deflection (in)
delta-sd-allowable = Primary Support Static + Dynamic Allowable Deflection (in)
d-ps = Primary Supports Distance (platform width) (in)
Fa-leg = Total Downward Force per Leg including Dead Load, Live Load, and Mixer Dynamic Loads, and Leg Weight (lbf)
Fa-total = Total Downward Force including Dead Load, Live Load, and Mixer Dynamic Loads (lbf)
fb-es = End Support Total Bending Stress (psi)
fb-es-req = End Support Allowable Bending Stress (psi)
fbx-es = End Support Bending Stress About X Axis (psi)
fby-es = End Support Bending Stress About Y Axis (psi)
fc-es = End Support Compressive Stress (psi)
fc-es-req = End Support Allowable Compressive Stress (psi)
k = End Support Effective Length Factor
L-es = End Support Length (in)
L-es-min = End Support Required Length (in)
L-es-min-overlap = End Support Shell Minimum Overlap (in)
L-es-shell-overlap = End Support Shell Overlap (in)
LL-platform = Platform Live Load (psf)
L-ps = Primary Support Length (ft)
L-ps-unbraced = Platform Primary Unbraced Length (in)
Ma-es = End Support Material
Ma-ps = Primary Support Material
M-max-support = Primary Support Maximum Bending Moment (ft.lbf)
Mx-es = End Support Bending Moment About X Axis (in.lbf)
My-es = End Support Bending Moment About Y Axis (in.lbf)
outside-proj-1-ps = Platform Outside Projection Side 2 (in)
outside-proj-2-ps = Platform Outside Projection Side 1 (in)
Pd = Primary Support Agitator Dynamic Point Load (lbf)
Ps = Primary Support Agitator Static Point Load (lbf)
Psd = Primary Support Agitator Center Load (lbf)
Sb-allowable-support = Primary Support Allowable Bending Stress (psi)
Scs = Maximum Allowable Compressive Stress per *API-620 5.5.4.2* (psi)
size-es = End Support Size
size-ps = Primary Support Size
Sx-reqd-support = Primary Support Required Section Modulus (in³)
t/R = Thickness to Tank Radius Ratio
t-repad-es = End Support Repad Thickness (in)
t-req-repad = End Support Repad Required Thickness (in)
V-max-support = Primary Support Maximum Shear (lbf)
W-add-platform = Platform Additional Weight (lbf)
W-platform = Platform Weight (lbf)
w-repad-es = End Support Repad Width (in)
W-support = Primary Support Uniform Load (lbf/ft)
W-total-platform = Platform Total Weight (lbf)

CA-es = 0.0 in
 CA-ps = 0.0 in
 d-ps = 36.0 in
 k = 2.10
 L-es = 22.0 in
 L-es-shell-overlap = 11.0 in
 LL-platform = 20.0 psf
 L-ps-unbraced = 48.50 in
 Ma-es = A240-316
 Ma-ps = A240-316
 outside-proj-1-ps = 15.0 in
 outside-proj-2-ps = 15.0 in
 size-es = w6x20
 size-ps = w6x20
 t-repad-es = 0.3750 in
 W-add-platform = 0.0 lbf
 W-platform = 213.988 lbf
 w-repad-es = 8.020 in

L-ps = OD
 L-ps = 10.0313
 L-ps = 10.0313 ft

A-platform = L-ps * d-ps
 A-platform = 10.0313 * 3.0
 A-platform = 30.0937 ft^2

W-total-platform = W-platform + W-add-platform
 W-total-platform = 213.988 + 0.0
 W-total-platform = 213.988 lbf

Agitator Bridge Primary Support Design **Agitator Bridge Primary Support Material Properties**

Material = A240-316
 Minimum Tensile Strength (Sut) = 75.0e3 psi
 As per API-650 S.5.b, Minimum Yield Strength (Sy) = 21.8750e3 psi
 Modulus of Elasticity at Design Temperature (E) = 26.550e6 psi

Agitator Bridge Primary Support I-Beam Size W6X20 Section Properties

Description	Variable	New	Corroded	Unit
Weight	W	20.0	20.0	lbf/ft
Cross Sectional Area	A	5.870	5.870	in^2
Radius of Gyration About X Axis	rx	2.660	2.660	in
Radius of Gyration About Y Axis	ry	1.50	1.50	in
Moment Of Inertia About X Axis	Ix	41.40	41.40	in^4
Moment Of Inertia About Y Axis	Iy	13.30	13.30	in^4
Section Modulus About X Axis	Sx	13.40	13.40	in^3
Section Modulus About Y Axis	Sy	4.410	4.410	in^3
Plastic Section Modulus About X Axis	Zx	15.0	15.0	in^3
Plastic Section Modulus About Y Axis	Zy	6.720	6.720	in^3

Warping Constant	cw	113.0	113.0	in ⁶
Torsional Constant	j	0.240	0.240	in ⁴
Centroid to Edge Max x Distance	ex	3.010	3.010	in
Centroid to Edge Max y Distance	ey	3.10	3.10	in
I-Beam Flange Width	wf	6.020	6.020	in
I-Beam Flange Thickness	tf	0.3650	0.3650	in
I-Beam Depth	d	6.20	6.20	in
I-Beam Web Thickness	tw	0.260	0.260	in

Agitator Bridge Primary Support Allowable Flexural Strength per AISC-360

b = Flange Outstand Length per *AISC-360 table B4.1b* (in)

h_w = Web height (in)

lambda = Slenderness parameter

lambda_pf = Limiting slenderness parameter for compact flange

lambda_pw = Limiting slenderness parameter for compact web

lambda_rf = Limiting slenderness parameter for noncompact flange

lambda_rw = Limiting slenderness parameter for noncompact web

L_p = Limiting laterally unbraced length for the limit state of yielding per *AISC-360 Section F2, Eq F2-5* (in)

M_allow = Allowable Flexural Strength (lbf.in)

M_n = Nominal flexural strength per *AISC-360 Section F2* (lbf.in)

M_n_Y = Nominal Flexural Strength due to Yielding per *AISC-360 Section F2, Eq F2-1* (lbf.in)

M_n_Y_allow = Allowable Flexural Strength Assuming the Member is Braced (Yield Only) (lbf.in)

w_f = Flange Total width (in)

$$w_f = wf = 6.020 \text{ in}$$

$$b = w_f / 2 = 3.010 \text{ in}$$

$$h_w = d - (2 * tf) = 5.470 \text{ in}$$

$$\lambda = b / tf$$

$$\lambda = 3.010 / 0.3650$$

$$\lambda = 8.24658$$

$$\lambda_{pf} = 0.38 * \sqrt{(E / F_y)}$$

$$\lambda_{pf} = 0.38 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{pf} = 13.2386$$

$$\lambda_{rf} = 1 * \sqrt{(E / F_y)}$$

$$\lambda_{rf} = 1 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rf} = 34.8384$$

$$\lambda_{pw} = 3.76 * \sqrt{(E / F_y)}$$

$$\lambda_{pw} = 3.76 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{pw} = 130.992$$

$$\lambda_{rw} = 5.7 * \sqrt{(E / F_y)}$$

$$\lambda_{rw} = 5.7 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rw} = 198.579$$

As per AISC-360 table B4.1b Flange width to thickness ratio check :
 $\lambda \leq \lambda_{pf}$

==> Flange is compact

As per AISC-360 table B4.1b Web height to thickness ratio check :
 $(h_w / t_w) \leq \lambda_{pw}$

==> Web is compact

$$\begin{aligned} M_{n_Y} &= F_y * Z_x \\ M_{n_Y} &= 21.8750e3 * 15.0 \\ M_{n_Y} &= 328.125e3 \text{ lbf.in} \end{aligned}$$

Unbraced length (L_b) = 48.50 in

$$\begin{aligned} L_p &= 1.76 * r_y * \sqrt{E / F_y} \\ L_p &= 1.76 * 1.50 * \sqrt{(26.550e6 / 21.8750e3)} \\ L_p &= 91.9734 \text{ in} \end{aligned}$$

$$L_b \leq L_p$$

$$M_n = (M_{n_Y}) = 328.125e3 \text{ lbf.in}$$

$$\begin{aligned} M_{allow} &= M_n / 1.67 \\ M_{allow} &= 328.125e3 / 1.67 \\ M_{allow} &= 196.482e3 \text{ lbf.in} \end{aligned}$$

$$\begin{aligned} M_{n_Y_{allow}} &= M_{n_Y} / 1.67 \\ M_{n_Y_{allow}} &= 328.125e3 / 1.67 \\ M_{n_Y_{allow}} &= 196.482e3 \text{ lbf.in} \end{aligned}$$

Agitator Bridge Primary Support Loads

$$\begin{aligned} P_s &= W_{\text{agitator}} / 2 \\ P_s &= 500.0 / 2 \\ P_s &= 250.0 \text{ lbf} \end{aligned}$$

$$\begin{aligned} P_d &= (F_a + (M_x / (0.5 * d_{ps}))) / 2 \\ P_d &= (0.0 + (0.0 / (0.5 * 36.0))) / 2 \\ P_d &= 0.0 \text{ lbf} \end{aligned}$$

$$\begin{aligned} P_{sd} &= P_s + P_d \\ P_{sd} &= 250.0 + 0.0 \\ P_{sd} &= 250.0 \text{ lbf} \end{aligned}$$

$$\begin{aligned} W_{\text{support}} &= W_{\text{primary-support}} + ((0.5 * (W_{\text{total-platform}} + (A_{\text{platform}} * LL_{\text{platform}}))) / L_{ps}) \\ W_{\text{support}} &= 20.0 + ((0.5 * (213.988 + (30.0937 * 20.0))) / 10.0313) \\ W_{\text{support}} &= 60.6661 \text{ lbf/ft} \end{aligned}$$

$$\begin{aligned} M_{\text{max-support}} &= ((W_{\text{support}} * (L_{ps}^2)) / 8) + ((P_{sd} * L_{ps}) / 4) \\ M_{\text{max-support}} &= ((60.6661 * (10.0313^2)) / 8) + ((250.0 * 10.0313) / 4) \\ M_{\text{max-support}} &= 1.39003e3 \text{ ft.lbf} \end{aligned}$$

$$\begin{aligned} V_{\text{max-support}} &= ((W_{\text{support}} * (L_{ps} + \text{outside-proj-1-ps} + \text{outside-proj-2-ps})) + (M_y / (L_{ps} / 2)) + P_{sd}) / 2 \\ V_{\text{max-support}} &= ((60.6661 * (10.0313 + 1.250 + 1.250)) + (0.0 / (120.375 / 2)) + 250.0) / 2 \\ V_{\text{max-support}} &= 505.111 \text{ lbf} \end{aligned}$$

Agitator Bridge Primary Support Required Section Modulus

$Sb_allowable_support = M_allow / Sx_primary_support$

$Sb_allowable_support = 196.482e3 / 13.40$

$Sb_allowable_support = 14.6628e3 \text{ psi}$

$Sx_reqd_support = M_max_support / Sb_allowable_support$

$Sx_reqd_support = 16.6803e3 / 14.6628e3$

$Sx_reqd_support = 1.13759 \text{ in}^3$

$Sx_primary_support \geq Sx_reqd_support \implies \text{PASS}$

Agitator Bridge Primary Support Maximum Allowable Deflection

$\Delta_{sd} = ((5 * W_support * (L_ps^4)) / (384 * E * Ix_primary_support)) + ((Psd * (L_ps^3)) / (48 * E * Ix_primary_support))$

$\Delta_{sd} = ((5 * 5.05551 * (120.375^4)) / (384 * 26.550e6 * 41.40)) + ((250.0 * (120.375^3)) / (48 * 26.550e6 * 41.40))$

$\Delta_{sd} = 0.0208393 \text{ in}$

$\Delta_{sd_allowable} = L_ps / 360$

$\Delta_{sd_allowable} = 120.375 / 360.0$

$\Delta_{sd_allowable} = 0.334375 \text{ in}$

$\Delta_{sd} \leq \Delta_{sd_allowable} \implies \text{PASS}$

$\Delta_d = (Pd * (L_ps^3)) / (48 * E * Ix_primary_support)$

$\Delta_d = (0.0 * (120.375^3)) / (48 * 26.550e6 * 41.40)$

$\Delta_d = 0.0 \text{ in}$

$\Delta_{d_allowable} = 0.5 * L_ps * \tan(0.1)$

$\Delta_{d_allowable} = 0.5 * 120.375 * \tan(0.10)$

$\Delta_{d_allowable} = 0.105047 \text{ in}$

$\Delta_d \leq \Delta_{d_allowable} \implies \text{PASS}$

Agitator Bridge End Support Design

Agitator Bridge End Support Material Properties

Material = A240-316

Minimum Tensile Strength (Sut) = 75.0e3 psi

As per API-650 S.5.b, Minimum Yield Strength (Sy) = 21.8750e3 psi

Modulus of Elasticity at Design Temperature (E) = 26.550e6 psi

Agitator Bridge End Support I-Beam Size W6X20 Section Properties

Description	Variable	New	Corroded	Unit
Weight	W	20.0	20.0	lbf/ft
Cross Sectional Area	A	5.870	5.870	in ²
Radius of Gyration About X Axis	rx	2.660	2.660	in
Radius of Gyration About Y Axis	ry	1.50	1.50	in
Moment Of Inertia About X Axis	Ix	41.40	41.40	in ⁴
Moment Of Inertia About Y Axis	Iy	13.30	13.30	in ⁴
Section Modulus About X Axis	Sx	13.40	13.40	in ³
Section Modulus About Y Axis	Sy	4.410	4.410	in ³

Plastic Section Modulus About X Axis	Zx	15.0	15.0	in ³
Plastic Section Modulus About Y Axis	Zy	6.720	6.720	in ³
Warping Constant	cw	113.0	113.0	in ⁶
Torsional Constant	j	0.240	0.240	in ⁴
Centroid to Edge Max x Distance	ex	3.010	3.010	in
Centroid to Edge Max y Distance	ey	3.10	3.10	in
I-Beam Flange Width	wf	6.020	6.020	in
I-Beam Flange Thickness	tf	0.3650	0.3650	in
I-Beam Depth	d	6.20	6.20	in
I-Beam Web Thickness	tw	0.260	0.260	in

Agitator Bridge End Support Allowable Flexural Strength per AISC-360

b = Flange Outstand Length per *AISC-360 table B4.1b* (in)

h_w = Web height (in)

lambda = Slenderness parameter

lambda_pf = Limiting slenderness parameter for compact flange

lambda_pw = Limiting slenderness parameter for compact web

lambda_rf = Limiting slenderness parameter for noncompact flange

lambda_rw = Limiting slenderness parameter for noncompact web

L_p = Limiting laterally unbraced length for the limit state of yielding per *AISC-360 Section F2, Eq F2-5* (in)

M_allow = Allowable Flexural Strength (lbf.in)

M_n = Nominal flexural strength per *AISC-360 Section F2* (lbf.in)

M_n_Y = Nominal Flexural Strength due to Yielding per *AISC-360 Section F2, Eq F2-1* (lbf.in)

M_n_Y_allow = Allowable Flexural Strength Assuming the Member is Braced (Yield Only) (lbf.in)

w_f = Flange Total width (in)

$$w_f = wf = 6.020 \text{ in}$$

$$b = w_f / 2 = 3.010 \text{ in}$$

$$h_w = d - (2 * tf) = 5.470 \text{ in}$$

$$\lambda = b / tf$$

$$\lambda = 3.010 / 0.3650$$

$$\lambda = 8.24658$$

$$\lambda_{pf} = 0.38 * \sqrt{(E / F_y)}$$

$$\lambda_{pf} = 0.38 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{pf} = 13.2386$$

$$\lambda_{rf} = 1 * \sqrt{(E / F_y)}$$

$$\lambda_{rf} = 1 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rf} = 34.8384$$

$$\lambda_{pw} = 3.76 * \sqrt{(E / F_y)}$$

$$\lambda_{pw} = 3.76 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{pw} = 130.992$$

$$\lambda_{rw} = 5.7 * \sqrt{(E / F_y)}$$

$$\lambda_{rw} = 5.7 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rw} = 198.579$$

As per AISC-360 table B4.1b Flange width to thickness ratio check :

$$\lambda \leq \lambda_{pf}$$

==> Flange is compact

As per AISC-360 table B4.1b Web height to thickness ratio check :

$$(h_w / t_w) \leq \lambda_{pw}$$

==> Web is compact

$$M_{n_Y} = F_y \cdot Z_x$$

$$M_{n_Y} = 21.8750e3 \cdot 15.0$$

$$M_{n_Y} = 328.125e3 \text{ lbf.in}$$

$$\text{Unbraced length } (L_b) = 11.0 \text{ in}$$

$$L_p = 1.76 \cdot r_y \cdot \sqrt{E / F_y}$$

$$L_p = 1.76 \cdot 1.50 \cdot \sqrt{(26.550e6 / 21.8750e3)}$$

$$L_p = 91.9734 \text{ in}$$

$$L_b \leq L_p$$

$$M_n = (M_{n_Y}) = 328.125e3 \text{ lbf.in}$$

$$M_{allow} = M_n / 1.67$$

$$M_{allow} = 328.125e3 / 1.67$$

$$M_{allow} = 196.482e3 \text{ lbf.in}$$

$$M_{n_Y_{allow}} = M_{n_Y} / 1.67$$

$$M_{n_Y_{allow}} = 328.125e3 / 1.67$$

$$M_{n_Y_{allow}} = 196.482e3 \text{ lbf.in}$$

Agitator Bridge End Support Allowable Compressive Strength per AISC-360

bf = Flange width (in)

Fe = Elastic Buckling Stress per AISC-360 E3-4 (psi)

h = Web height (in)

λ_{rf} = Limiting slenderness parameter for flanges per AISC 360 Table B4.1a

λ_{rw} = Limiting slenderness parameter for web per AISC 360 Table B4.1a

Pa = Allowable Compressive Strength (lbf)

P_n = Nominal compressive strength per AISC-360 E3-1 (lbf)

Radius of gyration :

$$((K \cdot L_y) / r_y) > ((K \cdot L_x) / r_x) \Rightarrow \text{Radius of gyration about y axis governs}$$

$$F_e = ((\pi^2) \cdot E) / (((K \cdot L_y) / r_y)^2)$$

$$F_e = ((\pi^2) \cdot 26.550e6) / (((2.10 \cdot 22.0) / 1.50)^2)$$

$$F_e = 276.225e3 \text{ psi}$$

$$b_f = w_f / 2$$

$$b_f = 6.020 / 2$$

$$b_f = 3.010 \text{ in}$$

$$h = d - (2 \cdot t_f)$$

$$h = 6.20 - (2 * 0.3650)$$

$$h = 5.470 \text{ in}$$

$$\lambda_{rf} = 0.56 * \sqrt{(E / F_y)}$$

$$\lambda_{rf} = 0.56 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rf} = 19.5095$$

$$\lambda_{rw} = 1.49 * \sqrt{(E / F_y)}$$

$$\lambda_{rw} = 1.49 * \sqrt{(26.550e6 / 21.8750e3)}$$

$$\lambda_{rw} = 51.9092$$

As per AISC-360 table B4.1a Flange width to thickness ratio check :
 $(b_f / t_f) \leq \lambda_{rf} \Rightarrow$ Flange is not slender

As per AISC-360 table B4.1a Web height to thickness ratio check :
 $(h / t_w) \leq \lambda_{rw} \Rightarrow$ Web is not slender

F_{cr} = Critical stress per AISC-360 Eq. E3-2 (psi)

$$(F_y / F_e) \leq 2.25$$

$$F_{cr} = (0.658^{(F_y / F_e)}) * F_y$$

$$F_{cr} = (0.658^{(21.8750e3 / 276.225e3)}) * 21.8750e3$$

$$F_{cr} = 21.1618e3 \text{ psi}$$

$$P_n = F_{cr} * A_g$$

$$P_n = 21.1618e3 * 5.870$$

$$P_n = 124.220e3 \text{ lbf}$$

$$P_a = P_n / 1.67$$

$$P_a = 124.220e3 / 1.67$$

$$P_a = 74.3831e3 \text{ lbf}$$

Agitator Bridge End Support Required Length

$$F_{a\text{-leg}} = V_{\text{max-support}} + (L_{\text{es}} * W_{\text{end-support}})$$

$$F_{a\text{-leg}} = 505.111 + (22.0 * 1.66667)$$

$$F_{a\text{-leg}} = 541.777 \text{ lbf}$$

$$t/R = (t_s - C_A) / (D / 2)$$

$$t/R = (0.18750 - 0.0) / (120.0 / 2)$$

$$t/R = 0.0031250$$

$$S_{cs} = 1800000 * \text{ratio}$$

$$S_{cs} = 1800000 * 0.0031250$$

$$S_{cs} = 5.6250e3 \text{ psi}$$

$$A_{s\text{-required}} = F_{a\text{-leg}} / S_{cs}$$

$$A_{s\text{-required}} = 541.777 / 5.6250e3$$

$$A_{s\text{-required}} = 0.0963160 \text{ in}^2$$

$$L_{\text{es-min-overlap}} = A_{s\text{-required}} / (2 * (t_s - C_A) * \tan(30))$$

$$L_{\text{es-min-overlap}} = 0.0963160 / (2 * (0.18750 - 0.0) * \tan(30))$$

$$L_{\text{es-min-overlap}} = 0.444865 \text{ in}$$

$$L_{\text{es-min}} = L_{\text{es-min-overlap}} + (L_{\text{es}} - L_{\text{es-shell-overlap}})$$

$$L_{\text{es-min}} = 0.444865 + (22.0 - 11.0)$$

L-es-min = 11.4449 in

L-es >= L-es-min ==> PASS

Agitator Bridge End Support Shell Reinforcement Requirements

$(2 * L\text{-es-min-overlap} * \text{TAN}(30)) < d\text{-ps} ==>$ Supports compression zones are not overlapping

As-available = $(2 * L\text{-es-shell-overlap} * \text{TAN}(30) * (ts - CA)) + (w\text{-repad-es} * t\text{-repad-es})$

As-available = $(2 * 11.0 * \text{TAN}(30) * (0.18750 - 0.0)) + (8.020 * 0.3750)$

As-available = 5.38907 in²

t-req-repad = $(As\text{-required} - (2 * L\text{-es-shell-overlap} * \text{TAN}(30) * (ts - CA))) / w\text{-repad-es}$

t-req-repad = $(0.0963160 - (2 * 11.0 * \text{TAN}(30) * (0.18750 - 0.0))) / 8.020$

t-req-repad = -0.284944 (Set to 0.0 in since it cannot be less than 0.0)

t-repad-es >= t-req-repad ==> PASS

Agitator Bridge End Support Required Section Modulus

Mx-es = V-max-support * ey

Mx-es = 505.111 * 3.10

Mx-es = 1.56584e3 in.lbf

fbx-es = Mx-es / Sx-end-support

fbx-es = 1.56584e3 / 13.40

fbx-es = 116.854 psi

My-es = $(Mz / (0.5 * L\text{-ps})) * (L\text{-es} - L\text{-es-shell-overlap})$

My-es = $(0.0 / (0.5 * 120.375)) * (22.0 - 11.0)$

My-es = 0.0 in.lbf

fby-es = My-es / Sy-end-support

fby-es = 0.0 / 4.410

fby-es = 0.0 psi

fb-es = fbx-es + fby-es

fb-es = 116.854 + 0.0

fb-es = 116.854 psi

fb-es-req = M_allow / Sx-end-support

fb-es-req = 196.482e3 / 13.40

fb-es-req = 14.6628e3 psi

fb-es <= fb-es-req ==> PASS

Agitator Bridge End Support Required Cross Sectional Area

fc-es = Fa-leg / A-end-support

fc-es = 541.777 / 5.870

fc-es = 92.2960 psi

fc-es-req = Pa / A-end-support

fc-es-req = 74.3831e3 / 5.870

fc-es-req = 12.6717e3 psi

A-es-req = Fa-leg / fc-es-req

A-es-req = 541.777 / 12.6717e3

A-es-req = 0.0427548 in²

A-end-support >= A-es-req ==> PASS

Agitator Bridge Total Downward Force (Including Dynamic Loads)

Fa-total = Fa-leg * 4

Fa-total = 541.777 * 4

Fa-total = 2.16711e3 lbf

Shell Design [Back](#)

Ac = Convective Design Response Spectrum Acceleration Coefficient
Ai = Impulsive Design Response Spectrum Acceleration Coefficient
Av = Vertical ground acceleration coefficient description
CG-shell = Shell center of gravity (ft)
D = Tank Nominal Diameter per *API-650 5.6.1.1 Note 1* (ft)
d-ins = Insulation Density (lb/ft³)
G = Product Design Specific Gravity
Gt = Hydrotest Specific Gravity
H = Shell height (ft)
H-Hydrotest-L = Max Hydrotest Liquid Level (ft)
HL = Max Liquid Level (ft)
h-min = Minimum Shell Course Height per *API-650 5.6.1.2* (in)
Pe = Design External Pressure (psf)
Pi = Design Internal Pressure (psi)
Rwi = Impulsive Force Reduction Factor
t-ins = Insulation Thickness (in)
V = Wind velocity (mile/hr)
W-ins = Shell Insulation Weight (lb)
W-shell = Shell Nominal Weight (lb)
W-shell-add = Shell Additional Weight (lb)
W-shell-corr = Shell Corroded Weight (lb)

Ac = 0.05410
Ai = 0.060
Av = 0.08960
D = 10.0 ft
d-ins = 8.0 lb/ft³
G = 1.10
Gt = 1.0
H = 30.0 ft
H-Hydrotest-L = 30.0 ft
HL = 30.0 ft
h-min = 48.0 in
Pe = 5.20219 psf
Pi = 0.108379 psi
Rwi = 4.0
t-ins = 0.0 in
V = 81.90 mile/hr
W-shell-add = 0.0 lb

API-650 Design Method: One Foot (1ft)

Rwi = Impulsive Force Reduction Factor

Rwi = 4.0

Course # 1 (bottom course) Design

A1 = Shell Course Cross Sectional Area (in²)
CA = Corrosion allowance per *API-650 5.3.2* (in)
D1 = Shell Course Centerline Diameter (in)
H' = Effective Design Liquid Level per *API-650 Section F.2* (ft)
H = Design Liquid Level per *API-650 5.6.3.2* (ft)

h_1 = Course Height (ft)
 $H_{\text{Hydrotest}}$ = Hydrotest Liquid Level per *API-650* 5.6.3.2 (ft)
 H_{max} = Maximum Liquid Level for the Installed Thickness (ft)
 $H_{\text{max-@-Pi}}$ = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 H_t' = Effective Hydrostatic Test Liquid Level per *API-650* F.2 (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 $Pi_{\text{max-@-H}}$ = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH₂O)
 R_{wi} = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650* Section S.3.2.2.3 (in)
 t_{min} = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per S.3.2.2.3 (in)
 $W-1$ = Shell Course Nominal Weight (lb)
 $W-1\text{-corr}$ = Shell Course Nominal Weight (lb)
 $W_{s\text{-tot-top}}$ = Top Weight Total (lbf)

$CA = 0.0$ in
 $H = 30.0$ ft
 $h_1 = 5.0$ ft
 $H_{\text{Hydrotest}} = 30.0$ ft
 $JE = 0.70$
 $loc = 0.0$ ft
 $Ma = A240-316$
 $R_{wi} = 4.0$
 $t = 0.18750$ in

Shell Course Center of Gravity (CG-1) = 2.50 ft

$D_1 = ID + t$
 $D_1 = 120.0 + 0.18750$
 $D_1 = 120.188$ in

$W-1 = \pi * D_c * t * h_1 * d$
 $W-1 = \pi * 120.188 * 0.18750 * 60.0 * 0.290$
 $W-1 = 1.23186e3$ lb

$W-1\text{-corr} = \pi * D_c * (t - CA) * h_1 * d$
 $W-1\text{-corr} = \pi * 120.188 * (0.18750 - 0.0) * 60.0 * 0.290$
 $W-1\text{-corr} = 1.23186e3$ lb

Material Properties

Material = A240-316
 Minimum Tensile Strength (S_{ut}) = 75.0e3 psi
 As per *API-650* S.5.b, Minimum Yield Strength (S_y) = 21.8750e3 psi
 As per *API-650* S.2b, Allowable Design Stress (S_d) = 19.7250e3 psi
 As per *API-650* S.2b, Allowable Hydrostatic Test Stress (S_t) = 27.0e3 psi

Thickness Required by Erection

As per *API-650* S.3.2.1.1 and 5.6.1.1, Thickness Required by Erection (t_{erec}) = 0.18750 in

Thickness Required by Design

$H' = H$
 $H' = 30.0$

$$H' = 30.0 \text{ ft}$$

$$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$$

$$t_d = ((2.6 * 10.0 * (30.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$$

$$t_d = 0.0600688 \text{ in}$$

Hydrostatic Test Required Thickness

$$H_t' = H\text{-Hydrotest}$$

$$H_t' = 30.0$$

$$H_t' = 30.0 \text{ ft}$$

$$t_t = (2.6 * D * (H_t' - 1) * SG_t) / (JE * S_t)$$

$$t_t = (2.6 * 10.0 * (30.0 - 1) * 1.0) / (0.70 * 27.0e3)$$

$$t_t = 0.0398942 \text{ in}$$

Seismic Design Required Thickness

$$N_c = \text{Convective Hoop Membrane Unit Force per API 650 Section E.6.1.4 (lbf/in)}$$

$$N_h = \text{Product Hydrostatic Membrane Force per API 650 Section E.6.1.4 and Section 5.6.3.2 (lbf/in)}$$

$$N_i = \text{Impulsive Hoop Membrane Unit Force per API 650 Section E.6.1.4 (lbf/in)}$$

$$S_d\text{-seismic} = \text{Maximum Allowable Hoop Tension Membrane Stress per API-650 E.6.2.4 (psi)}$$

$$t_s = \text{Seismic Minimum Thickness per API 650 Section E.6.2.4 (in)}$$

$$\text{As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 30.0 ft}$$

$$N_i = 1.39 * A_i * G * (D^2)$$

$$N_i = 1.39 * 0.060 * 1.10 * (10.0^2)$$

$$N_i = 9.1740 \text{ lbf/in}$$

$$N_c = (0.98 * A_c * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D))$$

$$N_c = (0.98 * 0.05410 * 1.10 * (10.0^2) * \cosh(((3.68 * (30.0 - 30.0)) / 10.0))) / \cosh(((3.68 * 30.0) / 10.0))$$

$$N_c = 187.169e-6 \text{ lbf/in}$$

$$N_h = 2.6 * (H - H_{\text{offset}}) * D * G$$

$$N_h = 2.6 * (30.0 - 0.0) * 10.0 * 1.10$$

$$N_h = 858.0 \text{ lbf/in}$$

$$\sigma_{T-} = \text{Total Combined Hoop Stress per API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3 (psi)}$$

$$\sigma_{T+} = \text{Total Combined Hoop Stress per API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3 (psi)}$$

$$\sigma_{T+} = (N_h + \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T+} = (858.0 + \sqrt{((9.1740^2) + (187.169e-6^2) + (((0.08960 * 858.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T+} = 4.74715e3 \text{ psi}$$

$$\sigma_{T-} = (N_h - \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T-} = (858.0 - \sqrt{((9.1740^2) + (187.169e-6^2) + (((0.08960 * 858.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T-} = 4.40485e3 \text{ psi}$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * S_d), (0.9 * F_y * E))$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))$$

$$S_d\text{-seismic} = 13.7813e3 \text{ psi}$$

$$t_s = ((\sigma_{T+} * (t_n - CA)) / S_{\text{membrane}}) + CA$$

$$t_s = ((4.74715e3 * (0.18750 - 0.0)) / 13.7813e3) + 0.0$$

$$t_s = 0.0645870 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

$$\text{Roof Total Weight (W}_{r\text{-tot}}) = 2.41886\text{e}3 \text{ lbf}$$

$$\text{Upper Courses Weight (W}_{s\text{-pl-top}}) = 6.14644\text{e}3 \text{ lbf}$$

$$W_{s\text{-tot-top}} = W_{r\text{-tot}} + W_{ss} + W_{s\text{-pl-top}} + W_{-1} + F_{d_agitator}$$

$$W_{s\text{-tot-top}} = 2.41886\text{e}3 + 1.49951\text{e}3 + 6.14644\text{e}3 + 1.23186\text{e}3 + 2.16711\text{e}3$$

$$W_{s\text{-tot-top}} = 13.4638\text{e}3 \text{ lbf}$$

$$A_1 = \pi * ((D_1 / 2)^2)$$

$$A_1 = \pi * ((120.188 / 2)^2)$$

$$A_1 = 11.3451\text{e}3 \text{ in}^2 [78.7854 \text{ ft}^2]$$

$$A_t = \text{Cross Section Area (in}^2)$$

$$c = \text{Corrosion Allowance (in)}$$

$$E = \text{Joint Efficiency}$$

$$F = \text{Summation of Vertical Forces (lb)}$$

$$P = \text{Pressure (lbf/in}^2)$$

$$P_{\text{Other_Loads}} = \text{Pressure Other Loads (lbf/in}^2)$$

$$R_c = \text{Radius (in)}$$

$$S_{ts} = \text{Maximum Allowable Tensile Stress (lbf/in}^2)$$

$$t_{\text{Actual}} = \text{Assumed Thickness (in)}$$

$$W = \text{Total Weight (lb)}$$

$$A_t = 11.3451\text{e}3 \text{ in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other_Loads}} = \text{nil}$$

$$R_c = 60.0938 \text{ in}$$

$$S_{ts} = 19.7250\text{e}3 \text{ lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -13.4638\text{e}3 \text{ lb}$$

$$T_1 = \text{Meridional unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 10 (lbf/in)}$$

$$T_2 = \text{Latitudinal (circumferential) unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 11 (lbf/in)}$$

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-13.4638\text{e}3 / 11.3451\text{e}3))$$

$$T_1 = -35.6581 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T1

$$M = \text{Compression factor}$$

$$S_{cc} = \text{Computed Compressive Stress (psi)}$$

$$S_{ta} = \text{Allowable Tensile Stress per API-620 5.5.3.3 (psi)}$$

Stc = Computed Tensile Stress (psi)
t_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

t_req = (Ratio * R) + c
t_req = (0.00312012 * 60.0938) + 0.0
t_req = 0.18750 in

Scs = ABS(T1) / (t-installed - c)
Scs = ABS(-35.6581) / (0.18750 - 0.0)
Scs = 190.176 psi

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 5.61622e3 psi

Scs <= Scs ==> PASS

Stc = T2 / (t-installed - c)
Stc = 0.0 / (0.18750 - 0.0)
Stc = 0.0 psi

M = Scs / 15000
M = 190.176 / 15000
M = 0.0126784

As per API-620 Figure F-1, Tension factor (N) = 0.993601

Sta = MIN((Sts * N) , (Sts * E))
Sta = MIN((19.7250e3 * 0.993601) , (19.7250e3 * 0.70))
Sta = 13.8075e3 psi

Stc <= Sta ==> PASS

Minimum Required Thickness

t-min = MAX(t_erec , t_d , t_t , ts , t-axial-load)
t-min = MAX(0.18750 , 0.0600688 , 0.0398942 , 0.0645870 , 0.18750)
t-min = 0.18750 in

Rating of Installed Thickness

H-max = (((t - CA) * Sd * JE) / (2.6 * D * SG)) + 1 + loc
H-max = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 0.0
H-max = 91.5212 ft

H-max-@-Pi = MAX(H-max , 0)
H-max-@-Pi = MAX(91.5212 , 0)
H-max-@-Pi = 91.5212 ft

Pi-max-@-H = MAX((((H-max - (H + loc)) * (12 * SG)) + P) , 0)
Pi-max-@-H = MAX((((91.5212 - (30.0 + 0.0)) * (12 * 1.10)) + 3.0) , 0)
Pi-max-@-H = 815.080 inH2O

Course # 2 Design

A2 = Shell Course Cross Sectional Area (in²)
CA = Corrosion allowance per *API-650* 5.3.2 (in)
D2 = Shell Course Centerline Diameter (in)
H' = Effective Design Liquid Level per *API-650 Section F.2* (ft)

H = Design Liquid Level per *API-650 5.6.3.2* (ft)
 h2 = Course Height (ft)
 H-Hydrotest = Hydrotest Liquid Level per *API-650 5.6.3.2* (ft)
 H-max = Maximum Liquid Level for the Installed Thickness (ft)
 H-max-@-Pi = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 Ht' = Effective Hydrostatic Test Liquid Level per *API-650 F.2* (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 Pi-max-@-H = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH2O)
 Rwi = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650 Section S.3.2.2.3* (in)
 t-min = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per *S.3.2.2.3* (in)
 W-2 = Shell Course Nominal Weight (lb)
 W-2-corr = Shell Course Nominal Weight (lb)
 Ws-tot-top = Top Weight Total (lbf)

CA = 0.0 in
 H = 25.0 ft
 h2 = 5.0 ft
 H-Hydrotest = 25.0 ft
 JE = 0.70
 loc = 5.0 ft
 Ma = A240-316
 Rwi = 4.0
 t = 0.18750 in

Shell Course Center of Gravity (CG-2) = 7.50 ft

D2 = ID + t
 D2 = 120.0 + 0.18750
 D2 = 120.188 in

W-2 = $\pi * D_c * t * h_2 * d$
 W-2 = $\pi * 120.188 * 0.18750 * 60.0 * 0.290$
 W-2 = 1.23186e3 lb

W-2-corr = $\pi * D_c * (t - CA) * h_2 * d$
 W-2-corr = $\pi * 120.188 * (0.18750 - 0.0) * 60.0 * 0.290$
 W-2-corr = 1.23186e3 lb

Material Properties

Material = A240-316
 Minimum Tensile Strength (Sut) = 75.0e3 psi
 As per *API-650 S.5.b*, Minimum Yield Strength (Sy) = 21.8750e3 psi
 As per *API-650 S.2b*, Allowable Design Stress (Sd) = 19.7250e3 psi
 As per *API-650 S.2b*, Allowable Hydrostatic Test Stress (St) = 27.0e3 psi

Thickness Required by Erection

As per *API-650 S.3.2.1.1* and *5.6.1.1*, Thickness Required by Erection (t_erec) = 0.18750 in

Thickness Required by Design

H' = H

$$H' = 25.0$$

$$H' = 25.0 \text{ ft}$$

$$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$$

$$t_d = ((2.6 * 10.0 * (25.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$$

$$t_d = 0.0497121 \text{ in}$$

Hydrostatic Test Required Thickness

$$H_t' = H\text{-Hydrotest}$$

$$H_t' = 25.0$$

$$H_t' = 25.0 \text{ ft}$$

$$t_t = (2.6 * D * (H_t' - 1) * SG_t) / (JE * S_t)$$

$$t_t = (2.6 * 10.0 * (25.0 - 1) * 1.0) / (0.70 * 27.0e3)$$

$$t_t = 0.0330159 \text{ in}$$

Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)
 Nh = Product Hydrostatic Membrane Force per *API 650 Section E.6.1.4 and Section 5.6.3.2* (lbf/in)
 Ni = Impulsive Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)
 Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per *API-650 E.6.2.4* (psi)
 ts = Seismic Minimum Thickness per *API 650 Section E.6.2.4* (in)

As per *API 650 Section E.6.1.4*, Shell Course Liquid Surface to Analysis Point Distance (Y) = 25.0 ft

$$N_i = 1.39 * A_i * G * (D^2)$$

$$N_i = 1.39 * 0.060 * 1.10 * (10.0^2)$$

$$N_i = 9.1740 \text{ lbf/in}$$

$$N_c = (0.98 * A_c * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D))$$

$$N_c = (0.98 * 0.05410 * 1.10 * (10.0^2) * \cosh(((3.68 * (30.0 - 25.0)) / 10.0))) / \cosh(((3.68 * 30.0) / 10.0))$$

$$N_c = 604.123e-6 \text{ lbf/in}$$

$$N_h = 2.6 * (H - H_{\text{offset}}) * D * G$$

$$N_h = 2.6 * (25.0 - 0.0) * 10.0 * 1.10$$

$$N_h = 715.0 \text{ lbf/in}$$

sigma_T- = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)
 sigma_T+ = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

$$\sigma_{T+} = (N_h + \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T+} = (715.0 + \sqrt{((9.1740^2) + (604.123e-6^2) + (((0.08960 * 715.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T+} = 3.95850e3 \text{ psi}$$

$$\sigma_{T-} = (N_h - \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T-} = (715.0 - \sqrt{((9.1740^2) + (604.123e-6^2) + (((0.08960 * 715.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T-} = 3.66817e3 \text{ psi}$$

$$S_{d\text{-seismic}} = \text{MIN}((1.33 * S_d), (0.9 * F_y * E))$$

$$S_{d\text{-seismic}} = \text{MIN}((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))$$

$$S_{d\text{-seismic}} = 13.7813e3 \text{ psi}$$

$$t_s = ((\sigma_{T+} * (t_n - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((3.95850e3 * (0.18750 - 0.0)) / 13.7813e3) + 0.0$$

$$ts = 0.0538571 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

Roof Total Weight (W_{r-tot}) = 2.41886e3 lbf

Upper Courses Weight ($W_{s-pl-top}$) = 4.91459e3 lbf

$W_{s-tot-top} = W_{r-tot} + W_{ss} + W_{s-pl-top} + W_2 + F_{d_agitator}$

$W_{s-tot-top} = 2.41886e3 + 1.49951e3 + 4.91459e3 + 1.23186e3 + 2.16711e3$

$W_{s-tot-top} = 12.2319e3 \text{ lbf}$

$$A_2 = \pi * ((D_2 / 2)^2)$$

$$A_2 = \pi * ((120.188 / 2)^2)$$

$$A_2 = 11.3451e3 \text{ in}^2 [78.7854 \text{ ft}^2]$$

A_t = Cross Section Area (in^2)

c = Corrosion Allowance (in)

E = Joint Efficiency

F = Summation of Vertical Forces (lb)

P = Pressure (lbf/in^2)

$P_{\text{Other_Loads}}$ = Pressure Other Loads (lbf/in^2)

R_c = Radius (in)

S_{ts} = Maximum Allowable Tensile Stress (lbf/in^2)

t_{Actual} = Assumed Thickness (in)

W = Total Weight (lb)

$$A_t = 11.3451e3 \text{ in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other_Loads}} = \text{nil}$$

$$R_c = 60.0938 \text{ in}$$

$$S_{ts} = 19.7250e3 \text{ lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -12.2319e3 \text{ lb}$$

T_1 = Meridional unit force for cylindrical walls per *API-620 5.10.2.5.c, Equation 10* (lbf/in)

T_2 = Latitudinal (circumferential) unit force for cylindrical walls per *API-620 5.10.2.5.c, Equation 11* (lbf/in)

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-12.2319e3 / 11.3451e3))$$

$$T_1 = -32.3956 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T_1

M = Compression factor

S_{cc} = Computed Compressive Stress (psi)

Sta = Allowable Tensile Stress per *API-620* 5.5.3.3 (psi)
Stc = Computed Tensile Stress (psi)
t_req = Minimum Required Thickness (in)

As per *API-620* Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

t_req = (Ratio * R) + c
t_req = (0.00312012 * 60.0938) + 0.0
t_req = 0.18750 in

Scc = ABS(T1) / (t-installed - c)
Scc = ABS(-32.3956) / (0.18750 - 0.0)
Scc = 172.776 psi

As per *API-620* Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 5.61622e3 psi

Scc <= Scs ==> PASS

Stc = T2 / (t-installed - c)
Stc = 0.0 / (0.18750 - 0.0)
Stc = 0.0 psi

M = Scc / 15000
M = 172.776 / 15000
M = 0.0115184

As per *API-620* Figure F-1, Tension factor (N) = 0.994191

Sta = MIN((Sts * N) , (Sts * E))
Sta = MIN((19.7250e3 * 0.994191) , (19.7250e3 * 0.70))
Sta = 13.8075e3 psi

Stc <= Sta ==> PASS

Minimum Required Thickness

t-min = MAX(t_erec , t_d , t_t , ts , t-axial-load)
t-min = MAX(0.18750 , 0.0497121 , 0.0330159 , 0.0538571 , 0.18750)
t-min = 0.18750 in

Rating of Installed Thickness

H-max = (((t - CA) * Sd * JE) / (2.6 * D * SG)) + 1 + loc
H-max = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 5.0
H-max = 96.5212 ft

H-max-@-Pi = MAX(H-max , 0)
H-max-@-Pi = MAX(96.5212 , 0)
H-max-@-Pi = 96.5212 ft

Pi-max-@-H = MAX((((H-max - (H + loc)) * (12 * SG)) + P) , 0)
Pi-max-@-H = MAX((((96.5212 - (25.0 + 5.0)) * (12 * 1.10)) + 3.0) , 0)
Pi-max-@-H = 881.080 inH2O

Course # 3 Design

A3 = Shell Course Cross Sectional Area (in²)
CA = Corrosion allowance per *API-650* 5.3.2 (in)
D3 = Shell Course Centerline Diameter (in)

H' = Effective Design Liquid Level per *API-650 Section F.2* (ft)
 H = Design Liquid Level per *API-650 5.6.3.2* (ft)
 h_3 = Course Height (ft)
 $H_{\text{Hydrotest}}$ = Hydrotest Liquid Level per *API-650 5.6.3.2* (ft)
 H_{max} = Maximum Liquid Level for the Installed Thickness (ft)
 $H_{\text{max-@-Pi}}$ = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 H_t' = Effective Hydrostatic Test Liquid Level per *API-650 F.2* (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 $Pi_{\text{max-@-H}}$ = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH₂O)
 R_{wi} = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650 Section S.3.2.2.3* (in)
 t_{min} = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per *S.3.2.2.3* (in)
 $W-3$ = Shell Course Nominal Weight (lb)
 $W-3\text{-corr}$ = Shell Course Nominal Weight (lb)
 $W_{\text{tot-top}}$ = Top Weight Total (lbf)

$CA = 0.0$ in
 $H = 20.0$ ft
 $h_3 = 5.0$ ft
 $H_{\text{Hydrotest}} = 20.0$ ft
 $JE = 0.70$
 $loc = 10.0$ ft
 $Ma = A240-316$
 $R_{wi} = 4.0$
 $t = 0.18750$ in

Shell Course Center of Gravity (CG-3) = 12.50 ft

$D_3 = ID + t$
 $D_3 = 120.0 + 0.18750$
 $D_3 = 120.188$ in

$W-3 = \pi * D_c * t * h_3 * d$
 $W-3 = \pi * 120.188 * 0.18750 * 60.0 * 0.290$
 $W-3 = 1.23186e3$ lb

$W-3\text{-corr} = \pi * D_c * (t - CA) * h_3 * d$
 $W-3\text{-corr} = \pi * 120.188 * (0.18750 - 0.0) * 60.0 * 0.290$
 $W-3\text{-corr} = 1.23186e3$ lb

Material Properties

Material = A240-316
 Minimum Tensile Strength (S_{ut}) = 75.0e3 psi
 As per *API-650 S.5.b*, Minimum Yield Strength (S_y) = 21.8750e3 psi
 As per *API-650 S.2b*, Allowable Design Stress (S_d) = 19.7250e3 psi
 As per *API-650 S.2b*, Allowable Hydrostatic Test Stress (S_t) = 27.0e3 psi

Thickness Required by Erection

As per *API-650 S.3.2.1.1* and *5.6.1.1*, Thickness Required by Erection (t_{erect}) = 0.18750 in

Thickness Required by Design

H' = H
H' = 20.0
H' = 20.0 ft

$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$
 $t_d = ((2.6 * 10.0 * (20.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$
 $t_d = 0.0393554$ in

Hydrostatic Test Required Thickness

Ht' = H-Hydrotest
Ht' = 20.0
Ht' = 20.0 ft

$t_t = (2.6 * D * (Ht' - 1) * SGt) / (JE * St)$
 $t_t = (2.6 * 10.0 * (20.0 - 1) * 1.0) / (0.70 * 27.0e3)$
 $t_t = 0.0261376$ in

Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)
Nh = Product Hydrostatic Membrane Force per *API 650 Section E.6.1.4 and Section 5.6.3.2* (lbf/in)
Ni = Impulsive Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)
Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per *API-650 E.6.2.4* (psi)
ts = Seismic Minimum Thickness per *API 650 Section E.6.2.4* (in)

As per *API 650 Section E.6.1.4*, Shell Course Liquid Surface to Analysis Point Distance (Y) = 20.0 ft

$Ni = 1.39 * Ai * G * (D^2)$
 $Ni = 1.39 * 0.060 * 1.10 * (10.0^2)$
 $Ni = 9.1740$ lbf/in

$Nc = (0.98 * Ac * G * (D^2) * COSH(((3.68 * (H - Y)) / D))) / COSH(((3.68 * H) / D))$
 $Nc = (0.98 * 0.05410 * 1.10 * (10.0^2) * COSH(((3.68 * (30.0 - 20.0)) / 10.0))) / COSH(((3.68 * 30.0) / 10.0))$
 $Nc = 0.00371266$ lbf/in

$Nh = 2.6 * (H - H_{offset}) * D * G$
 $Nh = 2.6 * (20.0 - 0.0) * 10.0 * 1.10$
 $Nh = 572.0$ lbf/in

sigma_T- = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)
sigma_T+ = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

$sigma_{T+} = (Nh + SQRT(((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)))) / MAX((t - CA), 0.0001)$
 $sigma_{T+} = (572.0 + SQRT(((9.1740^2) + (0.00371266^2) + (((0.08960 * 572.0) / 2.5)^2)))) / MAX((0.18750 - 0.0), 0.0001)$
 $sigma_{T+} = 3.17045e3$ psi

$sigma_{T-} = (Nh - SQRT(((Ni^2) + (Nc^2) + (((Av * Nh) / 2.5)^2)))) / MAX((t - CA), 0.0001)$
 $sigma_{T-} = (572.0 - SQRT(((9.1740^2) + (0.00371266^2) + (((0.08960 * 572.0) / 2.5)^2)))) / MAX((0.18750 - 0.0), 0.0001)$
 $sigma_{T-} = 2.93088e3$ psi

Sd-seismic = MIN((1.33 * Sd), (0.9 * Fy * E))
Sd-seismic = MIN((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))
Sd-seismic = 13.7813e3 psi

$$ts = ((\sigma_{T+} * (tn - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((3.17045e3 * (0.18750 - 0.0)) / 13.7813e3) + 0.0$$

$$ts = 0.0431354 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

$$\text{Roof Total Weight (Wr-tot)} = 2.41886e3 \text{ lbf}$$

$$\text{Upper Courses Weight (Ws-pl-top)} = 3.68273e3 \text{ lbf}$$

$$Ws\text{-tot-top} = Wr\text{-tot} + Wss + Ws\text{-pl-top} + W\text{-3} + Fd_{\text{agitator}}$$

$$Ws\text{-tot-top} = 2.41886e3 + 1.49951e3 + 3.68273e3 + 1.23186e3 + 2.16711e3$$

$$Ws\text{-tot-top} = 11.0001e3 \text{ lbf}$$

$$A3 = \pi * ((D3 / 2)^2)$$

$$A3 = \pi * ((120.188 / 2)^2)$$

$$A3 = 11.3451e3 \text{ in}^2 [78.7854 \text{ ft}^2]$$

$$At = \text{Cross Section Area (in}^2\text{)}$$

$$c = \text{Corrosion Allowance (in)}$$

$$E = \text{Joint Efficiency}$$

$$F = \text{Summation of Vertical Forces (lb)}$$

$$P = \text{Pressure (lbf/in}^2\text{)}$$

$$P_{\text{Other_Loads}} = \text{Pressure Other Loads (lbf/in}^2\text{)}$$

$$Rc = \text{Radius (in)}$$

$$Sts = \text{Maximum Allowable Tensile Stress (lbf/in}^2\text{)}$$

$$t_{\text{Actual}} = \text{Assumed Thickness (in)}$$

$$W = \text{Total Weight (lb)}$$

$$At = 11.3451e3 \text{ in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other_Loads}} = \text{nil}$$

$$Rc = 60.0938 \text{ in}$$

$$Sts = 19.7250e3 \text{ lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -11.0001e3 \text{ lb}$$

$$T_1 = \text{Meridional unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 10 (lbf/in)}$$

$$T_2 = \text{Latitudinal (circumferential) unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 11 (lbf/in)}$$

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-11.0001e3 / 11.3451e3))$$

$$T_1 = -29.1331 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T1

M = Compression factor

Sc_c = Computed Compressive Stress (psi)
St_a = Allowable Tensile Stress per *API-620* 5.5.3.3 (psi)
St_c = Computed Tensile Stress (psi)
t_{req} = Minimum Required Thickness (in)

As per *API-620* Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

t_{req} = (Ratio * R) + c
t_{req} = (0.00312012 * 60.0938) + 0.0
t_{req} = 0.18750 in

Sc_c = ABS(T₁) / (t_{installed} - c)
Sc_c = ABS(-29.1331) / (0.18750 - 0.0)
Sc_c = 155.376 psi

As per *API-620* Section 5.5.4.2, Maximum Allowable Compressive Stress (Sc_s) = 5.61622e3 psi

Sc_c <= Sc_s ==> PASS

St_c = T₂ / (t_{installed} - c)
St_c = 0.0 / (0.18750 - 0.0)
St_c = 0.0 psi

M = Sc_c / 15000
M = 155.376 / 15000
M = 0.0103584

As per *API-620* Figure F-1, Tension factor (N) = 0.994781

St_a = MIN((St_s * N) , (St_s * E))
St_a = MIN((19.7250e3 * 0.994781) , (19.7250e3 * 0.70))
St_a = 13.8075e3 psi

St_c <= St_a ==> PASS

Minimum Required Thickness

t_{min} = MAX(t_{erect} , t_d , t_t , t_s , t_{axial-load})
t_{min} = MAX(0.18750 , 0.0393554 , 0.0261376 , 0.0431354 , 0.18750)
t_{min} = 0.18750 in

Rating of Installed Thickness

H_{max} = (((t - CA) * S_d * J_E) / (2.6 * D * S_G)) + 1 + loc
H_{max} = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 10.0
H_{max} = 101.521 ft

H_{max}-@-Pi = MAX(H_{max} , 0)
H_{max}-@-Pi = MAX(101.521 , 0)
H_{max}-@-Pi = 101.521 ft

Pi_{max}-@-H = MAX((((H_{max} - (H + loc)) * (12 * S_G)) + P) , 0)
Pi_{max}-@-H = MAX((((101.521 - (20.0 + 10.0)) * (12 * 1.10)) + 3.0) , 0)
Pi_{max}-@-H = 947.080 inH₂O

Course # 4 Design

A₄ = Shell Course Cross Sectional Area (in²)
CA = Corrosion allowance per *API-650* 5.3.2 (in)

D4 = Shell Course Centerline Diameter (in)
 H' = Effective Design Liquid Level per *API-650 Section F.2* (ft)
 H = Design Liquid Level per *API-650 5.6.3.2* (ft)
 h4 = Course Height (ft)
 H-Hydrotest = Hydrotest Liquid Level per *API-650 5.6.3.2* (ft)
 H-max = Maximum Liquid Level for the Installed Thickness (ft)
 H-max-@-Pi = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 Ht' = Effective Hydrostatic Test Liquid Level per *API-650 F.2* (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 Pi-max-@-H = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH2O)
 Rwi = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650 Section S.3.2.2.3* (in)
 t-min = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per *S.3.2.2.3* (in)
 W-4 = Shell Course Nominal Weight (lb)
 W-4-corr = Shell Course Nominal Weight (lb)
 Ws-tot-top = Top Weight Total (lbf)

CA = 0.0 in
 H = 15.0 ft
 h4 = 5.0 ft
 H-Hydrotest = 15.0 ft
 JE = 0.70
 loc = 15.0 ft
 Ma = A240-316
 Rwi = 4.0
 t = 0.18750 in

Shell Course Center of Gravity (CG-4) = 17.50 ft

D4 = ID + t
 D4 = 120.0 + 0.18750
 D4 = 120.188 in

$W-4 = \pi * D_c * t * h_4 * d$
 $W-4 = \pi * 120.188 * 0.18750 * 60.0 * 0.290$
 $W-4 = 1.23186e3 \text{ lb}$

$W-4\text{-corr} = \pi * D_c * (t - CA) * h_4 * d$
 $W-4\text{-corr} = \pi * 120.188 * (0.18750 - 0.0) * 60.0 * 0.290$
 $W-4\text{-corr} = 1.23186e3 \text{ lb}$

Material Properties

Material = A240-316
 Minimum Tensile Strength (Sut) = 75.0e3 psi
 As per *API-650 S.5.b*, Minimum Yield Strength (Sy) = 21.8750e3 psi
 As per *API-650 S.2b*, Allowable Design Stress (Sd) = 19.7250e3 psi
 As per *API-650 S.2b*, Allowable Hydrostatic Test Stress (St) = 27.0e3 psi

Thickness Required by Erection

As per *API-650 S.3.2.1.1* and *5.6.1.1*, Thickness Required by Erection (t_erec) = 0.18750 in

Thickness Required by Design

$$H' = H$$

$$H' = 15.0$$

$$H' = 15.0 \text{ ft}$$

$$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$$

$$t_d = ((2.6 * 10.0 * (15.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$$

$$t_d = 0.0289987 \text{ in}$$

Hydrostatic Test Required Thickness

$$H_t' = H\text{-Hydrotest}$$

$$H_t' = 15.0$$

$$H_t' = 15.0 \text{ ft}$$

$$t_t = (2.6 * D * (H_t' - 1) * SG_t) / (JE * S_t)$$

$$t_t = (2.6 * 10.0 * (15.0 - 1) * 1.0) / (0.70 * 27.0e3)$$

$$t_t = 0.0192593 \text{ in}$$

Seismic Design Required Thickness

Nc = Convective Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)

Nh = Product Hydrostatic Membrane Force per *API 650 Section E.6.1.4 and Section 5.6.3.2* (lbf/in)

Ni = Impulsive Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)

Sd-seismic = Maximum Allowable Hoop Tension Membrane Stress per *API-650 E.6.2.4* (psi)

ts = Seismic Minimum Thickness per *API 650 Section E.6.2.4* (in)

As per *API 650 Section E.6.1.4*, Shell Course Liquid Surface to Analysis Point Distance (Y) = 15.0 ft

$$N_i = 1.39 * A_i * G * (D^2)$$

$$N_i = 1.39 * 0.060 * 1.10 * (10.0^2)$$

$$N_i = 9.1740 \text{ lbf/in}$$

$$N_c = (0.98 * A_c * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D))$$

$$N_c = (0.98 * 0.05410 * 1.10 * (10.0^2) * \cosh(((3.68 * (30.0 - 15.0)) / 10.0))) / \cosh(((3.68 * 30.0) / 10.0))$$

$$N_c = 0.0233624 \text{ lbf/in}$$

$$N_h = 2.6 * (H - H_{\text{offset}}) * D * G$$

$$N_h = 2.6 * (15.0 - 0.0) * 10.0 * 1.10$$

$$N_h = 429.0 \text{ lbf/in}$$

sigma_T- = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

sigma_T+ = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

$$\sigma_{T+} = (N_h + \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T+} = (429.0 + \sqrt{((9.1740^2) + (0.0233624^2) + (((0.08960 * 429.0) / 2.5)^2))}) /$$

$$\text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T+} = 2.38349e3 \text{ psi}$$

$$\sigma_{T-} = (N_h - \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T-} = (429.0 - \sqrt{((9.1740^2) + (0.0233624^2) + (((0.08960 * 429.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T-} = 2.19251e3 \text{ psi}$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * S_d), (0.9 * F_y * E))$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))$$

$$S_d\text{-seismic} = 13.7813e3 \text{ psi}$$

$$ts = ((\sigma_{T+} * (tn - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((2.38349e3 * (0.18750 - 0.0)) / 13.7813e3) + 0.0$$

$$ts = 0.0324284 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

$$\text{Roof Total Weight (Wr-tot)} = 2.41886e3 \text{ lbf}$$

$$\text{Upper Courses Weight (Ws-pl-top)} = 2.45088e3 \text{ lbf}$$

$$Ws\text{-tot-top} = Wr\text{-tot} + Wss + Ws\text{-pl-top} + W\text{-4} + Fd_{\text{agitator}}$$

$$Ws\text{-tot-top} = 2.41886e3 + 1.49951e3 + 2.45088e3 + 1.23186e3 + 2.16711e3$$

$$Ws\text{-tot-top} = 9.76821e3 \text{ lbf}$$

$$A4 = \pi * ((D4 / 2)^2)$$

$$A4 = \pi * ((120.188 / 2)^2)$$

$$A4 = 11.3451e3 \text{ in}^2 [78.7854 \text{ ft}^2]$$

$$At = \text{Cross Section Area (in}^2\text{)}$$

$$c = \text{Corrosion Allowance (in)}$$

$$E = \text{Joint Efficiency}$$

$$F = \text{Summation of Vertical Forces (lb)}$$

$$P = \text{Pressure (lbf/in}^2\text{)}$$

$$P_{\text{Other Loads}} = \text{Pressure Other Loads (lbf/in}^2\text{)}$$

$$Rc = \text{Radius (in)}$$

$$Sts = \text{Maximum Allowable Tensile Stress (lbf/in}^2\text{)}$$

$$t_{\text{Actual}} = \text{Assumed Thickness (in)}$$

$$W = \text{Total Weight (lb)}$$

$$At = 11.3451e3 \text{ in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other Loads}} = \text{nil}$$

$$Rc = 60.0938 \text{ in}$$

$$Sts = 19.7250e3 \text{ lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -9.76821e3 \text{ lb}$$

$$T_1 = \text{Meridional unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 10 (lbf/in)}$$

$$T_2 = \text{Latitudinal (circumferential) unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 11 (lbf/in)}$$

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-9.76821e3 / 11.3451e3))$$

$$T_1 = -25.8706 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T1

M = Compression factor
 Scc = Computed Compressive Stress (psi)
 Sta = Allowable Tensile Stress per *API-620 5.5.3.3* (psi)
 Stc = Computed Tensile Stress (psi)
 t_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

$t_{req} = (\text{Ratio} * R) + c$
 $t_{req} = (0.00312012 * 60.0938) + 0.0$
 $t_{req} = 0.18750 \text{ in}$

$S_{cc} = \text{ABS}(T1) / (t_{\text{installed}} - c)$
 $S_{cc} = \text{ABS}(-25.8706) / (0.18750 - 0.0)$
 $S_{cc} = 137.976 \text{ psi}$

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 5.61622e3 psi

$S_{cc} \leq S_{cs} \Rightarrow \text{PASS}$

$Stc = T2 / (t_{\text{installed}} - c)$
 $Stc = 0.0 / (0.18750 - 0.0)$
 $Stc = 0.0 \text{ psi}$

$M = S_{cc} / 15000$
 $M = 137.976 / 15000$
 $M = 0.00919842$

As per API-620 Figure F-1, Tension factor (N) = 0.995369

$Sta = \text{MIN}((Sts * N), (Sts * E))$
 $Sta = \text{MIN}((19.7250e3 * 0.995369), (19.7250e3 * 0.70))$
 $Sta = 13.8075e3 \text{ psi}$

$Stc \leq Sta \Rightarrow \text{PASS}$

Minimum Required Thickness

$t_{\text{min}} = \text{MAX}(t_{\text{erect}}, t_d, t_t, t_s, t_{\text{axial-load}})$
 $t_{\text{min}} = \text{MAX}(0.18750, 0.0289987, 0.0192593, 0.0324284, 0.18750)$
 $t_{\text{min}} = 0.18750 \text{ in}$

Rating of Installed Thickness

$H_{\text{max}} = (((t - CA) * S_d * J_E) / (2.6 * D * S_G)) + 1 + loc$
 $H_{\text{max}} = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 15.0$
 $H_{\text{max}} = 106.521 \text{ ft}$

$H_{\text{max-@-Pi}} = \text{MAX}(H_{\text{max}}, 0)$
 $H_{\text{max-@-Pi}} = \text{MAX}(106.521, 0)$
 $H_{\text{max-@-Pi}} = 106.521 \text{ ft}$

$Pi_{\text{max-@-H}} = \text{MAX}((((H_{\text{max}} - (H + loc)) * (12 * S_G)) + P), 0)$
 $Pi_{\text{max-@-H}} = \text{MAX}((((106.521 - (15.0 + 15.0)) * (12 * 1.10)) + 3.0), 0)$
 $Pi_{\text{max-@-H}} = 1.01308e3 \text{ inH}_2\text{O}$

Course # 5 Design

A5 = Shell Course Cross Sectional Area (in²)

CA = Corrosion allowance per *API-650* 5.3.2 (in)
 D5 = Shell Course Centerline Diameter (in)
 H' = Effective Design Liquid Level per *API-650 Section F.2* (ft)
 H = Design Liquid Level per *API-650* 5.6.3.2 (ft)
 h5 = Course Height (ft)
 H-Hydrotest = Hydrotest Liquid Level per *API-650* 5.6.3.2 (ft)
 H-max = Maximum Liquid Level for the Installed Thickness (ft)
 H-max-@-Pi = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 Ht' = Effective Hydrostatic Test Liquid Level per *API-650 F.2* (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 Pi-max-@-H = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH2O)
 Rwi = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650 Section S.3.2.2.3* (in)
 t-min = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per *S.3.2.2.3* (in)
 W-5 = Shell Course Nominal Weight (lb)
 W-5-corr = Shell Course Nominal Weight (lb)
 Ws-tot-top = Top Weight Total (lbf)

CA = 0.0 in
 H = 10.0 ft
 h5 = 5.0 ft
 H-Hydrotest = 10.0 ft
 JE = 0.70
 loc = 20.0 ft
 Ma = A240-316
 Rwi = 4.0
 t = 0.18750 in

Shell Course Center of Gravity (CG-5) = 22.50 ft

D5 = ID + t
 D5 = 120.0 + 0.18750
 D5 = 120.188 in

W-5 = $\pi * D_c * t * h_5 * d$
 W-5 = $\pi * 120.188 * 0.18750 * 60.0 * 0.290$
 W-5 = 1.23186e3 lb

W-5-corr = $\pi * D_c * (t - CA) * h_5 * d$
 W-5-corr = $\pi * 120.188 * (0.18750 - 0.0) * 60.0 * 0.290$
 W-5-corr = 1.23186e3 lb

Material Properties

Material = A240-316
 Minimum Tensile Strength (Sut) = 75.0e3 psi
 As per *API-650* S.5.b, Minimum Yield Strength (Sy) = 21.8750e3 psi
 As per *API-650* S.2b, Allowable Design Stress (Sd) = 19.7250e3 psi
 As per *API-650* S.2b, Allowable Hydrostatic Test Stress (St) = 27.0e3 psi

Thickness Required by Erection

As per *API-650* S.3.2.1.1 and 5.6.1.1, Thickness Required by Erection (t_erec) = 0.18750 in

Thickness Required by Design

$$H' = H$$

$$H' = 10.0$$

$$H' = 10.0 \text{ ft}$$

$$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$$

$$t_d = ((2.6 * 10.0 * (10.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$$

$$t_d = 0.0186420 \text{ in}$$

Hydrostatic Test Required Thickness

$$H_t' = H\text{-Hydrotest}$$

$$H_t' = 10.0$$

$$H_t' = 10.0 \text{ ft}$$

$$t_t = (2.6 * D * (H_t' - 1) * SG_t) / (JE * S_t)$$

$$t_t = (2.6 * 10.0 * (10.0 - 1) * 1.0) / (0.70 * 27.0e3)$$

$$t_t = 0.0123810 \text{ in}$$

Seismic Design Required Thickness

$$N_c = \text{Convective Hoop Membrane Unit Force per API 650 Section E.6.1.4 (lbf/in)}$$

$$N_h = \text{Product Hydrostatic Membrane Force per API 650 Section E.6.1.4 and Section 5.6.3.2 (lbf/in)}$$

$$N_i = \text{Impulsive Hoop Membrane Unit Force per API 650 Section E.6.1.4 (lbf/in)}$$

$$S_d\text{-seismic} = \text{Maximum Allowable Hoop Tension Membrane Stress per API-650 E.6.2.4 (psi)}$$

$$t_s = \text{Seismic Minimum Thickness per API 650 Section E.6.2.4 (in)}$$

As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 10.0 ft

$$N_i = 1.39 * A_i * G * (D^2)$$

$$N_i = 1.39 * 0.060 * 1.10 * (10.0^2)$$

$$N_i = 9.1740 \text{ lbf/in}$$

$$N_c = (0.98 * A_c * G * (D^2) * \cosh(((3.68 * (H - Y)) / D))) / \cosh(((3.68 * H) / D))$$

$$N_c = (0.98 * 0.05410 * 1.10 * (10.0^2) * \cosh(((3.68 * (30.0 - 10.0)) / 10.0))) / \cosh(((3.68 * 30.0) / 10.0))$$

$$N_c = 0.14710 \text{ lbf/in}$$

$$N_h = 2.6 * (H - H_{\text{offset}}) * D * G$$

$$N_h = 2.6 * (10.0 - 0.0) * 10.0 * 1.10$$

$$N_h = 286.0 \text{ lbf/in}$$

$$\sigma_{T-} = \text{Total Combined Hoop Stress per API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3 (psi)}$$

$$\sigma_{T+} = \text{Total Combined Hoop Stress per API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3 (psi)}$$

$$\sigma_{T+} = (N_h + \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T+} = (286.0 + \sqrt{((9.1740^2) + (0.14710^2) + (((0.08960 * 286.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T+} = 1.59870e3 \text{ psi}$$

$$\sigma_{T-} = (N_h - \sqrt{((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2))}) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T-} = (286.0 - \sqrt{((9.1740^2) + (0.14710^2) + (((0.08960 * 286.0) / 2.5)^2))}) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T-} = 1.45196e3 \text{ psi}$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * S_d), (0.9 * F_y * E))$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))$$

$$S_d\text{-seismic} = 13.7813e3 \text{ psi}$$

$$ts = ((\sigma_{T+} * (tn - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((1.59870e3 * (0.18750 - 0.0)) / 13.7813e3) + 0.0$$

$$ts = 0.0217511 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

$$\text{Roof Total Weight (W}_{r\text{-tot}}) = 2.41886e3 \text{ lbf}$$

$$\text{Upper Courses Weight (W}_{s\text{-pl-top}}) = 1.21902e3 \text{ lbf}$$

$$W_{s\text{-tot-top}} = W_{r\text{-tot}} + W_{ss} + W_{s\text{-pl-top}} + W_5 + F_{d_agitator}$$

$$W_{s\text{-tot-top}} = 2.41886e3 + 1.49951e3 + 1.21902e3 + 1.23186e3 + 2.16711e3$$

$$W_{s\text{-tot-top}} = 8.53636e3 \text{ lbf}$$

$$A_5 = \pi * ((D_5 / 2)^2)$$

$$A_5 = \pi * ((120.188 / 2)^2)$$

$$A_5 = 11.3451e3 \text{ in}^2 [78.7854 \text{ ft}^2]$$

$$A_t = \text{Cross Section Area (in}^2)$$

$$c = \text{Corrosion Allowance (in)}$$

$$E = \text{Joint Efficiency}$$

$$F = \text{Summation of Vertical Forces (lb)}$$

$$P = \text{Pressure (lbf/in}^2)$$

$$P_{\text{Other_Loads}} = \text{Pressure Other Loads (lbf/in}^2)$$

$$R_c = \text{Radius (in)}$$

$$S_{ts} = \text{Maximum Allowable Tensile Stress (lbf/in}^2)$$

$$t_{\text{Actual}} = \text{Assumed Thickness (in)}$$

$$W = \text{Total Weight (lb)}$$

$$A_t = 11.3451e3 \text{ in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other_Loads}} = \text{nil}$$

$$R_c = 60.0938 \text{ in}$$

$$S_{ts} = 19.7250e3 \text{ lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -8.53636e3 \text{ lb}$$

$$T_1 = \text{Meridional unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 10 (lbf/in)}$$

$$T_2 = \text{Latitudinal (circumferential) unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 11 (lbf/in)}$$

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-8.53636e3 / 11.3451e3))$$

$$T_1 = -22.6081 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T1

M = Compression factor
Scc = Computed Compressive Stress (psi)
Sta = Allowable Tensile Stress per *API-620* 5.5.3.3 (psi)
Stc = Computed Tensile Stress (psi)
t_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

t_req = (Ratio * R) + c
t_req = (0.00312012 * 60.0938) + 0.0
t_req = 0.18750 in

Scc = ABS(T1) / (t-installed - c)
Scc = ABS(-22.6081) / (0.18750 - 0.0)
Scc = 120.576 psi

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 5.61622e3 psi

Scc <= Scs ==> PASS

Stc = T2 / (t-installed - c)
Stc = 0.0 / (0.18750 - 0.0)
Stc = 0.0 psi

M = Scc / 15000
M = 120.576 / 15000
M = 0.00803842

As per API-620 Figure F-1, Tension factor (N) = 0.995957

Sta = MIN((Sts * N) , (Sts * E))
Sta = MIN((19.7250e3 * 0.995957) , (19.7250e3 * 0.70))
Sta = 13.8075e3 psi

Stc <= Sta ==> PASS

Minimum Required Thickness

t-min = MAX(t_erec , t_d , t_t , ts , t-axial-load)
t-min = MAX(0.18750 , 0.0186420 , 0.0123810 , 0.0217511 , 0.18750)
t-min = 0.18750 in

Rating of Installed Thickness

H-max = (((t - CA) * Sd * JE) / (2.6 * D * SG)) + 1 + loc
H-max = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 20.0
H-max = 111.521 ft

H-max-@-Pi = MAX(H-max , 0)
H-max-@-Pi = MAX(111.521 , 0)
H-max-@-Pi = 111.521 ft

Pi-max-@-H = MAX((((H-max - (H + loc)) * (12 * SG)) + P) , 0)
Pi-max-@-H = MAX((((111.521 - (10.0 + 20.0)) * (12 * 1.10)) + 3.0) , 0)
Pi-max-@-H = 1.07908e3 inH2O

Course # 6 Design

A_6 = Shell Course Cross Sectional Area (in²)
 CA = Corrosion allowance per *API-650* 5.3.2 (in)
 D_6 = Shell Course Centerline Diameter (in)
 H' = Effective Design Liquid Level per *API-650* Section F.2 (ft)
 H = Design Liquid Level per *API-650* 5.6.3.2 (ft)
 h_6 = Course Height (ft)
 $H_{Hydrotest}$ = Hydrotest Liquid Level per *API-650* 5.6.3.2 (ft)
 H_{max} = Maximum Liquid Level for the Installed Thickness (ft)
 $H_{max-@-Pi}$ = Maximum Liquid Level for the Installed Thickness @ Design Internal Pressure (ft)
 H_t' = Effective Hydrostatic Test Liquid Level per *API-650* F.2 (ft)
 JE = Joint efficiency
 loc = Course Location (ft)
 Ma = Course Material
 $Pi_{max-@-H}$ = Maximum Allowable Internal Pressure for the Installed Thickness @ Design Liquid Level (inH₂O)
 R_{wi} = Impulsive Force Reduction Factor
 t = Installed Thickness (in)
 t_d = Course Design Thickness per *API-650* Section S.3.2.2.3 (in)
 t_{min} = Minimum Required Thickness (in)
 t_t = Course Hydrostatic Test Thickness per S.3.2.2.3 (in)
 W_6 = Shell Course Nominal Weight (lb)
 W_6-corr = Shell Course Nominal Weight (lb)
 $W_{s-tot-top}$ = Top Weight Total (lbf)

$CA = 0.0$ in
 $H = 5.0$ ft
 $h_6 = 4.94792$ ft
 $H_{Hydrotest} = 5.0$ ft
 $JE = 0.70$
 $loc = 25.0$ ft
 $Ma = A240-316$
 $R_{wi} = 4.0$
 $t = 0.18750$ in

Shell Course Center of Gravity (CG-6) = 27.4740 ft

$D_6 = ID + t$
 $D_6 = 120.0 + 0.18750$
 $D_6 = 120.188$ in

$W_6 = \pi * D_c * t * h_6 * d$
 $W_6 = \pi * 120.188 * 0.18750 * 59.3750 * 0.290$
 $W_6 = 1.21902e3$ lb

$W_6-corr = \pi * D_c * (t - CA) * h_6 * d$
 $W_6-corr = \pi * 120.188 * (0.18750 - 0.0) * 59.3750 * 0.290$
 $W_6-corr = 1.21902e3$ lb

Material Properties

Material = A240-316
 Minimum Tensile Strength (S_{ut}) = 75.0e3 psi
 As per *API-650* S.5.b, Minimum Yield Strength (S_y) = 21.8750e3 psi
 As per *API-650* S.2b, Allowable Design Stress (S_d) = 19.7250e3 psi
 As per *API-650* S.2b, Allowable Hydrostatic Test Stress (S_t) = 27.0e3 psi

Thickness Required by Erection

As per API-650 S.3.2.1.1 and 5.6.1.1, Thickness Required by Erection (t_{erec}) = 0.18750 in

Thickness Required by Design

$$H' = H$$

$$H' = 5.0$$

$$H' = 5.0 \text{ ft}$$

$$t_d = ((2.6 * D * (H' - 1) * SG) / (JE * Sd)) + CA$$

$$t_d = ((2.6 * 10.0 * (5.0 - 1) * 1.10) / (0.70 * 19.7250e3)) + 0.0$$

$$t_d = 0.00828535 \text{ in}$$

Hydrostatic Test Required Thickness

$$H_t' = H\text{-Hydrotest}$$

$$H_t' = 5.0$$

$$H_t' = 5.0 \text{ ft}$$

$$t_t = (2.6 * D * (H_t' - 1) * SG_t) / (JE * S_t)$$

$$t_t = (2.6 * 10.0 * (5.0 - 1) * 1.0) / (0.70 * 27.0e3)$$

$$t_t = 0.00550265 \text{ in}$$

Seismic Design Required Thickness

N_c = Convective Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)

N_h = Product Hydrostatic Membrane Force per *API 650 Section E.6.1.4 and Section 5.6.3.2* (lbf/in)

N_i = Impulsive Hoop Membrane Unit Force per *API 650 Section E.6.1.4* (lbf/in)

$S_d\text{-seismic}$ = Maximum Allowable Hoop Tension Membrane Stress per *API-650 E.6.2.4* (psi)

t_s = Seismic Minimum Thickness per *API 650 Section E.6.2.4* (in)

As per API 650 Section E.6.1.4, Shell Course Liquid Surface to Analysis Point Distance (Y) = 5.0 ft

$$N_i = 2.77 * A_i * G * (D^2) * ((Y / (0.75 * D)) - (0.5 * ((Y / (0.75 * D))^2)))$$

$$N_i = 2.77 * 0.060 * 1.10 * (10.0^2) * ((5.0 / (0.75 * 10.0)) - (0.5 * ((5.0 / (0.75 * 10.0))^2)))$$

$$N_i = 8.12533 \text{ lbf/in}$$

$$N_c = (0.98 * A_c * G * (D^2) * \text{COSH}(((3.68 * (H - Y)) / D))) / \text{COSH}(((3.68 * H) / D))$$

$$N_c = (0.98 * 0.05410 * 1.10 * (10.0^2) * \text{COSH}(((3.68 * (30.0 - 5.0)) / 10.0))) / \text{COSH}(((3.68 * 30.0) / 10.0))$$

$$N_c = 0.926220 \text{ lbf/in}$$

$$N_h = 2.6 * (H - H_{\text{offset}}) * D * G$$

$$N_h = 2.6 * (5.0 - 0.0) * 10.0 * 1.10$$

$$N_h = 143.0 \text{ lbf/in}$$

σ_{T-} = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

σ_{T+} = Total Combined Hoop Stress per *API 650 Eq E.6.1.4-6, Sections E.6.1.4, EC.6.1.3* (psi)

$$\sigma_{T+} = (N_h + \text{SQRT}(((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2)))) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T+} = (143.0 + \text{SQRT}(((8.12533^2) + (0.926220^2) + (((0.08960 * 143.0) / 2.5)^2)))) /$$

$$\text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T+} = 814.140 \text{ psi}$$

$$\sigma_{T-} = (N_h - \text{SQRT}(((N_i^2) + (N_c^2) + (((A_v * N_h) / 2.5)^2)))) / \text{MAX}((t - CA), 0.0001)$$

$$\sigma_{T-} = (143.0 - \text{SQRT}(((8.12533^2) + (0.926220^2) + (((0.08960 * 143.0) / 2.5)^2)))) / \text{MAX}((0.18750 - 0.0), 0.0001)$$

$$\sigma_{T-} = 711.194 \text{ psi}$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * S_d), (0.9 * F_y * E))$$

$$S_d\text{-seismic} = \text{MIN}((1.33 * 19.7250e3), (0.9 * 21.8750e3 * 0.70))$$

$$S_d\text{-seismic} = 13.7813\text{e3 psi}$$

$$ts = ((\sigma_{T+} * (tn - CA)) / S_{\text{membrane}}) + CA$$

$$ts = ((814.140 * (0.18750 - 0.0)) / 13.7813\text{e3}) + 0.0$$

$$ts = 0.0110767 \text{ in}$$

Vertical Axial Load Design (Empty Tank)

$$\text{Roof Total Weight (Wr-tot)} = 2.41886\text{e3 lbf}$$

$$\text{Upper Courses Weight (Ws-pl-top)} = 0.0 \text{ lbf}$$

$$Ws\text{-tot-top} = Wr\text{-tot} + Wss + Ws\text{-pl-top} + W\text{-6} + Fd_{\text{agitator}}$$

$$Ws\text{-tot-top} = 2.41886\text{e3} + 1.49951\text{e3} + 0.0 + 1.21902\text{e3} + 2.16711\text{e3}$$

$$Ws\text{-tot-top} = 7.30450\text{e3 lbf}$$

$$A6 = \pi * ((D6 / 2)^2)$$

$$A6 = \pi * ((120.188 / 2)^2)$$

$$A6 = 11.3451\text{e3 in}^2 [78.7854 \text{ ft}^2]$$

$$At = \text{Cross Section Area (in}^2)$$

$$c = \text{Corrosion Allowance (in)}$$

$$E = \text{Joint Efficiency}$$

$$F = \text{Summation of Vertical Forces (lb)}$$

$$P = \text{Pressure (lbf/in}^2)$$

$$P_{\text{Other_Loads}} = \text{Pressure Other Loads (lbf/in}^2)$$

$$Rc = \text{Radius (in)}$$

$$Sts = \text{Maximum Allowable Tensile Stress (lbf/in}^2)$$

$$t_{\text{Actual}} = \text{Assumed Thickness (in)}$$

$$W = \text{Total Weight (lb)}$$

$$At = 11.3451\text{e3 in}^2$$

$$c = 0.0 \text{ in}$$

$$E = 0.70$$

$$F = \text{nil}$$

$$P = 0.0 \text{ lbf/in}^2$$

$$P_{\text{Other_Loads}} = \text{nil}$$

$$Rc = 60.0938 \text{ in}$$

$$Sts = 19.7250\text{e3 lbf/in}^2$$

$$t_{\text{Actual}} = 0.18750 \text{ in}$$

$$W = -7.30450\text{e3 lb}$$

$$T_1 = \text{Meridional unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 10 (lbf/in)}$$

$$T_2 = \text{Latitudinal (circumferential) unit force for cylindrical walls per API-620 5.10.2.5.c, Equation 11 (lbf/in)}$$

$$T_1 = (R_c / 2) * (P + (W / A_t))$$

$$T_1 = (60.0938 / 2) * (0.0 + (-7.30450\text{e3} / 11.3451\text{e3}))$$

$$T_1 = -19.3456 \text{ lbf/in}$$

$$T_2 = P * R_c$$

$$T_2 = 0.0 * 60.0938$$

$$T_2 = 0.0 \text{ lbf/in}$$

$$(T_1 < 0) \text{ AND } (T_2 = 0)$$

Thickness calculation based on T1

M = Compression factor
 Scc = Computed Compressive Stress (psi)
 Sta = Allowable Tensile Stress per *API-620* 5.5.3.3 (psi)
 Stc = Computed Tensile Stress (psi)
 t_req = Minimum Required Thickness (in)

As per API-620 Section 5.5.4.5, Figure 5-1 and Figure F-1, Graphical solution (Ratio) = 0.00312012

$t_{req} = (\text{Ratio} * R) + c$
 $t_{req} = (0.00312012 * 60.0938) + 0.0$
 $t_{req} = 0.18750 \text{ in}$

$S_{cc} = \text{ABS}(T_1) / (t_{\text{installed}} - c)$
 $S_{cc} = \text{ABS}(-19.3456) / (0.18750 - 0.0)$
 $S_{cc} = 103.176 \text{ psi}$

As per API-620 Section 5.5.4.2, Maximum Allowable Compressive Stress (Scs) = 5.61622e3 psi

$S_{cc} \leq S_{cs} \Rightarrow \text{PASS}$

$Stc = T_2 / (t_{\text{installed}} - c)$
 $Stc = 0.0 / (0.18750 - 0.0)$
 $Stc = 0.0 \text{ psi}$

$M = S_{cc} / 15000$
 $M = 103.176 / 15000$
 $M = 0.00687842$

As per API-620 Figure F-1, Tension factor (N) = 0.996543

$Sta = \text{MIN}((Sts * N), (Sts * E))$
 $Sta = \text{MIN}((19.7250e3 * 0.996543), (19.7250e3 * 0.70))$
 $Sta = 13.8075e3 \text{ psi}$

$Stc \leq Sta \Rightarrow \text{PASS}$

Minimum Required Thickness

$t_{\text{min}} = \text{MAX}(t_{\text{erect}}, t_d, t_t, t_s, t_{\text{axial-load}})$
 $t_{\text{min}} = \text{MAX}(0.18750, 0.00828535, 0.00550265, 0.0110767, 0.18750)$
 $t_{\text{min}} = 0.18750 \text{ in}$

Rating of Installed Thickness

$H_{\text{max}} = (((t - CA) * S_d * J_E) / (2.6 * D * S_G)) + 1 + loc$
 $H_{\text{max}} = (((0.18750 - 0.0) * 19.7250e3 * 0.70) / (2.6 * 10.0 * 1.10)) + 1 + 25.0$
 $H_{\text{max}} = 116.521 \text{ ft}$

$H_{\text{max-@-Pi}} = \text{MAX}(H_{\text{max}}, 0)$
 $H_{\text{max-@-Pi}} = \text{MAX}(116.521, 0)$
 $H_{\text{max-@-Pi}} = 116.521 \text{ ft}$

$Pi_{\text{max-@-H}} = \text{MAX}((((H_{\text{max}} - (H + loc)) * (12 * S_G)) + P), 0)$
 $Pi_{\text{max-@-H}} = \text{MAX}((((116.521 - (5.0 + 25.0)) * (12 * 1.10)) + 3.0), 0)$
 $Pi_{\text{max-@-H}} = 1.14508e3 \text{ inH}_2\text{O}$

Shell Design Summary Results

$W_{ins} = t_{ins} * d_{ins} * \pi * (OD + t_{ins}) * H$
 $W_{ins} = 0.0 * 8.0 * \pi * (10.0313 + 0.0) * 30.0$
 $W_{ins} = 0.0 \text{ lbf}$

$W_{shell-corr} = (W-1-corr + W-2-corr + W-3-corr + W-4-corr + W-5-corr + W-6-corr) + W_{shell-add}$
 $W_{shell-corr} = (1.23186e3 + 1.23186e3 + 1.23186e3 + 1.23186e3 + 1.23186e3 + 1.21902e3) + 0.0$
 $W_{shell-corr} = 7.37830e3 \text{ lb}$

$W_{shell} = (W-1 + W-2 + W-3 + W-4 + W-5 + W-6) + W_{shell-add}$
 $W_{shell} = (1.23186e3 + 1.23186e3 + 1.23186e3 + 1.23186e3 + 1.23186e3 + 1.21902e3) + 0.0$
 $W_{shell} = 7.37830e3 \text{ lb}$

$CG_{shell} = ((CG-1 * W-1) + (CG-2 * W-2) + (CG-3 * W-3) + (CG-4 * W-4) + (CG-5 * W-5) + (CG-6 * W-6)) / W_{shell}$
 $CG_{shell} = ((2.50 * 1.23186e3) + (7.50 * 1.23186e3) + (12.50 * 1.23186e3) + (17.50 * 1.23186e3) + (22.50 * 1.23186e3) + (27.4740 * 1.21902e3)) / 7.37830e3$
 $CG_{shell} = 14.9740 \text{ ft}$

Shell Design Summary

Course	Height (ft)	Material	CA (in)	JE	Sy (psi)	Sut (psi)	Sd (psi)	St (psi)	t_erec (in)
6	4.94792	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750
5	5.0	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750
4	5.0	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750
3	5.0	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750
2	5.0	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750
1	5.0	A240-316	0.0	0.70	21.8750e3	75.0e3	19.7250e3	27.0e3	0.18750

Shell Design Summary (continued)

Course	t-design (in)	t-test (in)	t-seismic (in)	t-ext (in)	t-min (in)	t-installed (in)	Status	H-max-@-Pi (ft)	Pi-max-@-H (psi)
6	0.00828535	0.00550265	0.0110767	N/A	0.18750	0.18750	PASS	116.521	41.3675
5	0.0186420	0.0123810	0.0217511	N/A	0.18750	0.18750	PASS	111.521	38.9831
4	0.0289987	0.0192593	0.0324284	N/A	0.18750	0.18750	PASS	106.521	36.5988
3	0.0393554	0.0261376	0.0431354	N/A	0.18750	0.18750	PASS	101.521	34.2145
2	0.0497121	0.0330159	0.0538571	N/A	0.18750	0.18750	PASS	96.5212	31.8301
1	0.0600688	0.0398942	0.0645870	N/A	0.18750	0.18750	PASS	91.5212	29.4458

Intermediate Stiffeners Design

Stiffeners Design For Wind Loading

D = Nominal Tank Diameter (ft)

E = Modulus Of Elasticity (psi)

E_ambient = Modulus Of Elasticity at Ambient Temperature (psi)

H1 = Maximum Unstiffened Transformed Shell Height per *API-650 5.9.6.1* (ft)

ME = Reduction Factor per *API-650 S.3.6.5*

N = Actual Wind Girders Quantity

Ns = Required Number of Girders per *API 650 5.9.6.3 and 5.9.6.4*

Pwd = Design Wind Pressure Including Inward Drag per *API-650 5.9.6.1* (psf)

Pwv = Wind Pressure where Design Wind Speed V is Used per *API-650 5.9.6.1* (psf)
ts_min = Thickness of the Thinnest Shell Course (in)
V = Wind velocity (mile/hr)

D = 10.0 ft
E = 26.550e6 psi
E_ambient = 28.10e6 psi
N = 0.0
V = 81.90 mile/hr

Shell Courses Heights (W) = [5.0 5.0 5.0 5.0 5.0 4.94792] ft

ts_min = MIN(ts_1 , ts_2 , ts_3 , ts_4 , ts_5 , ts_6)
ts_min = MIN(0.18750 , 0.18750 , 0.18750 , 0.18750 , 0.18750 , 0.18750)
ts_min = 0.18750 in

ME = E / E_ambient
ME = 26.550e6 / 28.10e6
ME = 0.944840

Stiffeners Required Quantity

HTS = Height of Transformed Shell per *API 650 5.9.6.2* (ft)

Transformed shell courses heights

Variable	Equation	Value	Unit		
Wtr_1	$W_1 * \text{SQRT}(((t_{\min} / ts_1)^5))$	5.0000	ft	N/A	N/A
Wtr_2	$W_2 * \text{SQRT}(((t_{\min} / ts_2)^5))$	5.0000	ft	N/A	N/A
Wtr_3	$W_3 * \text{SQRT}(((t_{\min} / ts_3)^5))$	5.0000	ft	N/A	N/A
Wtr_4	$W_4 * \text{SQRT}(((t_{\min} / ts_4)^5))$	5.0000	ft	N/A	N/A
Wtr_5	$W_5 * \text{SQRT}(((t_{\min} / ts_5)^5))$	5.0000	ft	N/A	N/A
Wtr_6	$W_6 * \text{SQRT}(((t_{\min} / ts_6)^5))$	4.9479	ft	N/A	N/A

HTS = Wtr_1 + Wtr_2 + Wtr_3 + Wtr_4 + Wtr_5 + Wtr_6
HTS = 5.0 + 5.0 + 5.0 + 5.0 + 5.0 + 4.94792
HTS = 29.9479 ft

Pwv = $31 * ((V / 120)^2)$
Pwv = $31 * ((81.90 / 120)^2)$
Pwv = 14.440 psf

Pwd = Pwv + 5
Pwd = 14.440 + 5
Pwd = 19.440 psf

H1 = $(600000 * ts_{\min} * \text{SQRT}(((ts_{\min} / D)^3)) * (36 / Pwd)) * ME$
H1 = $(600000 * 0.18750 * \text{SQRT}(((0.18750 / 10.0)^3)) * (36 / 19.440)) * 0.944840$
H1 = 505.381 ft

Ns = CEILING(((HTS / Hsafe) - 1))
Ns = CEILING(((29.9479 / 505.381) - 1))
Ns = 0.0

$N \geq N_s \implies \text{PASS}$

Flat Bottom: non Annular Plate Design [Back](#)

Bottom Type = Flat

Bottom Support Type = Continuously Supported on Foundation

CA = Corrosion allowance (in)

CA_1 = Bottom Shell Course Corrosion Allowance (in)

chime = Outside Projection (Chime Distance) (in)

E = Joint efficiency

Ma_1 = Bottom Shell Course Material

Ma-bottom = Material

S = Bottom Shell Course Maximum Stress (psi)

S1 = Bottom Shell Course Product Stress per *API-650 Table 5.1b Note b* (psi)

S2 = Bottom Shell Course Hydrostatic Stress per *API-650 Table 5.1b Note b* (psi)

Sd_1 = Bottom Shell Course Allowable Design Stress (psi)

St_1 = Bottom Shell Course Allowable Hydrostatic Test Stress (psi)

tb = Installed Thickness (in)

tb-req = Bottom Required Thickness (in)

td_1 = Bottom Shell Course Design Thickness (in)

ts_1 = Bottom Shell Course Nominal Thickness (in)

tt_1 = Bottom Shell Course Hydrotest Thickness (in)

CA = 0.0 in

CA_1 = 0.0 in

chime = 1.0 in

E = 0.70

Ma_1 = A240-316

Ma-bottom = A240-316

Sd_1 = 19.7250e3 psi

St_1 = 27.0e3 psi

tb = 0.18750 in

td_1 = 0.0600688 in

ts_1 = 0.18750 in

tt_1 = 0.0398942 in

Bottom Plates Material Properties

Material = A240-316

Minimum Tensile Strength (Sut-btm) = 75.0e3 psi

As per API-650 S.5.b, Minimum Yield Strength (Sy-btm) = 21.8750e3 psi

Density (d-btm) = 0.290 lb/in³

Calculation of Hydrostatic Test Stress & Product Stress per API-650 Section 5.5.1

$S1 = ((td_1 - CA_1) / (ts_1 - CA_1)) * Sd_1$

$S1 = ((0.0600688 - 0.0) / (0.18750 - 0.0)) * 19.7250e3$

$S1 = 6.31924e3$ psi

As per API-650 5.5.1 and S.3.1.3, first shell course material, A240-316, is stainless steel and the stress is less than or equal to 23200; therefore, butt welded annular plates are not required

$S2 = (tt_1 / ts_1) * St_1$

$S2 = (0.0398942 / 0.18750) * 27.0e3$

$S2 = 5.74476e3$ psi

As per API-650 5.5.1 and S.3.1.3, first shell course material, A240-316, is stainless steel and the stress is

less than or equal to 24900; therefore, butt welded annular plates are not required

$$S = \text{MAX}(S1, S2)$$
$$S = \text{MAX}(6.31924\text{e}3, 5.74476\text{e}3)$$
$$S = 6.31924\text{e}3 \text{ psi}$$

As per API-650 M.4.1

$D \leq 100 \implies$ Butt welded annular ring is not required

Bottom Weight

A-btm = Bottom Surface Area (ft²)
CA = Corrosion allowance (in)
chime = Outside Projection (Chime Distance) (in)
OD-btm = Bottom Outer Diameter (ft)
tb = Installed Thickness (in)
Wb-pl = Bottom Plates Weight (lb)
Wb-pl-add = Bottom Additional Weight (lb)
Wb-pl-corr = Bottom Corroded Plates Weight (lb)
Wb-pl-corroded-overlap = Bottom Weight Corroded Overlap (lb)
Wb-pl-overlap = Bottom Overlap Weight (lb)

$$CA = 0.0 \text{ in}$$
$$\text{chime} = 1.0 \text{ in}$$
$$tb = 0.18750 \text{ in}$$
$$Wb\text{-pl-add} = 0.0 \text{ lb}$$
$$Wb\text{-pl-corroded-overlap} = 0.0 \text{ lb}$$
$$Wb\text{-pl-overlap} = 0.0 \text{ lb}$$

$$OD\text{-btm} = OD + (\text{chime} * 2)$$
$$OD\text{-btm} = 10.0313 + (0.0833333 * 2)$$
$$OD\text{-btm} = 10.1979 \text{ ft}$$

$$A\text{-btm} = \pi * ((OD\text{-btm} / 2)^2)$$
$$A\text{-btm} = \pi * ((10.1979 / 2)^2)$$
$$A\text{-btm} = 81.6794 \text{ ft}^2$$

$$Wb\text{-pl} = (A\text{-btm} * tb * d\text{-btm}) + Wb\text{-pl-overlap} + Wb\text{-pl-add}$$
$$Wb\text{-pl} = (11.7618\text{e}3 * 0.18750 * 0.290) + 0.0 + 0.0$$
$$Wb\text{-pl} = 639.550 \text{ lb}$$

$$Wb\text{-pl-corr} = (A\text{-btm} * (tb - CA) * d\text{-btm}) + Wb\text{-pl-corroded-overlap} + Wb\text{-pl-add}$$
$$Wb\text{-pl-corr} = (11.7618\text{e}3 * (0.18750 - 0.0) * 0.290) + 0.0 + 0.0$$
$$Wb\text{-pl-corr} = 639.550 \text{ lb}$$

Bottom Design due to External Pressure

P-btm = Downward Pressure (psi)

Liquid Height to Pressure Conversion Factor (f) = 0.433515

$$P\text{-btm} = (d\text{-btm} * (tb - CA\text{-btm})) + (Lmin * f * SG)$$
$$P\text{-btm} = (0.290 * (0.18750 - 0.0)) + (1.0 * 0.433515 * 1.10)$$
$$P\text{-btm} = 0.531242 \text{ psi}$$

$P\text{-btm} \geq P_v \implies$ There is no uplift due to external pressure

Bottom Required Thickness

As per API-650 S.3.1.2, Required Thickness by Erection (tb-erec) = 0.18750 in

tb-req = tb-erec

tb-req = 0.18750

tb-req = 0.18750 in

tb >= tb-req ==> PASS

Bottom Outside Projection

As per API-650 J.3.2.3, Minimum Required Outside Projection (chime) = 1.0 in

chime >= chime ==> PASS

Wind Moment (Per API-650 Section 5.11)

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Wind Pressures per API-650 & ASCE7-16

P_{WR} = Roof Design Wind Pressure per *API-650 5.2.1.k* (psf)

P_{WS} = Shell Design Wind Pressure per *API-650 5.2.1.k* (psf)

V = Design Wind Velocity (3-sec gust) per *ASCE 7-16* (mph)

V_s = Adjusted Design Wind Velocity per *API 650 Section 5.2.1.k* (mph)

V = 105.0 mph

Wind Velocity per API-650 and ASCE7-16

V_s = 0.78 * V

V_s = 0.78 * 105.0

V_s = 81.90 mph

Roof Wind Pressure

P_{WR} = 31 * ((V_s / 120)²)

P_{WR} = 31 * ((81.90 / 120)²)

P_{WR} = 14.440 psf

Shell Wind Pressure

P_{WS} = 18.6 * ((V_s / 120)²)

P_{WS} = 18.6 * ((81.90 / 120)²)

P_{WS} = 8.6640 psf

Wind Overturning and Sliding Stability

A_h = Roof Horizontal Projected Area (ft²)

A_{h-total} = Roof Horizontal Projected Area Including Insulation (ft²)

A_s = Shell Total Vertical Projected Area (ft²)

A_{v-roof} = Roof Vertical Projected Area (ft²)

CA₁ = Bottom Shell Course Corrosion Allowance (in)

CA_{btm} = Corrosion Allowance of Bottom Plates Under the Shell (in)

CG_{roof} = Roof Center of Gravity (ft)

COF = Maximum Allowable Sliding Friction Coefficient per *API 650, Section 5.11.4*

D_{LR} = Nominal Weight of Roof Plates and Attached Structural (lbf)

D_{LS} = Nominal Weight of Shell Plates and Framing (lbf)

D_{outer} = Tank Max Outer Diameter (ft)

F_{by} = Yield Strength of Bottom Plates Under the Shell (psi)

F_{friction} = Friction Force per *API 650, Section 5.11.4* (lbf)

F_{wind} = Sliding Force (lbf)

M_{DL} = Moment About the Shell-To-Bottom Joint from the Nominal Weight of the Shell (ft.lbf)

M_{D_{LR}} = Moment About the Shell-To-Bottom Joint from the Nominal Weight of the Roof Plate Plus any Attached Structural (ft.lbf)

M_F = Moment About the Shell-To-Bottom Joint From Liquid Weight (ft.lbf)

M_{Pi} = Moment About the Shell-To-Bottom Joint From Design Internal Pressure per *API-650 5.11.2.2* (ft.lbf)

M_w = Overturning Moment About the Shell-To-Bottom Joint from Wind Pressures per *API-650 5.11.2.2* (ft.lbf)

M_{WR} = Roof Wind Overturning Moment per *API-650 5.11.2.2* (ft.lbf)

M_{WS} = Shell Wind Overturning Moment per *API-650 5.11.2.2* (ft.lbf)
 R_h = Roof Horizontal Radius (ft)
 t_b = Thickness of Bottom Plates Under the Shell (in)
 t_{r-ins} = Roof Insulation Thickness (in)
 t_{s_1} = Bottom Shell Course Nominal Thickness (in)
 t_{s-ins} = Shell Insulation Thickness (in)
 $W_{access-corr}$ = Access Corroded Weight (lbf)
 $W_{app-corr}$ = Appurtenances Corroded Weight (lbf)
 $W_{b-pl-corr}$ = Bottom Corroded Plates Weight (lbf)
 $wind-uplift$ = Wind Uplift per *API-650 5.2.1.k* (psf)
 w_L = Tank Content Resisting Weight per *API-650 5.11.2.3* (lbf/ft)
 W_{r-pl} = Roof New Plates Weight (lbf)
 $W_{r-pl-corr}$ = Roof Corroded Plates Weight (lbf)
 $W_{s-framing}$ = Shell New Framing Weight (lbf)
 $W_{s-framing-corr}$ = Shell Corroded Framing Weight (lbf)
 W_{s-pl} = Shell New Plates Weight (lbf)
 $W_{s-pl-corr}$ = Shell Corroded Plates Weight (lbf)
 $W_{s-struct-corr}$ = Roof Corroded Structure Weight Supported by Shell (lbf)
 $W_{stairs-corr}$ = Stairways Corroded Weight (lbf)
 W_{struct} = Roof New Structure Weight (lbf)
 $W_{struct-corr}$ = Roof Corroded Structure Weight (lbf)
 $W_{windgirder-corr}$ = Wind Girder Corroded Weight (lbf)
 X_s = Moment Arm of Wind Force on Shell (ft)
 X_w = Moment Arm of Wind Force on Roof (ft)

$A_h = 80.5779 \text{ ft}^2$
 $A_{v-roof} = 4.27479 \text{ ft}^2$
 $CA_1 = 0.0 \text{ in}$
 $CA_{-btm} = 0.0 \text{ in}$
 $CG_{-roof} = 0.281359 \text{ ft}$
 $COF = 0.40$
 $DLR = 900.493 \text{ lbf}$
 $DLS = 7.48169\text{e}3 \text{ lbf}$
 $F_{by} = 21.8750\text{e}3 \text{ psi}$
 $R_h = 5.06446 \text{ ft}$
 $t_b = 0.18750 \text{ in}$
 $t_{r-ins} = 0.0 \text{ in}$
 $t_{s_1} = 0.18750 \text{ in}$
 $t_{s-ins} = 0.0 \text{ in}$
 $W_{access-corr} = 0.0 \text{ lbf}$
 $W_{app-corr} = 759.415 \text{ lbf}$
 $W_{b-pl-corr} = 639.550 \text{ lbf}$
 $W_{r-pl} = 639.628 \text{ lbf}$
 $W_{r-pl-corr} = 639.628 \text{ lbf}$
 $W_{s-framing} = 103.391 \text{ lbf}$
 $W_{s-framing-corr} = 100.397 \text{ lbf}$
 $W_{s-pl} = 7.37830\text{e}3 \text{ lbf}$
 $W_{s-pl-corr} = 7.37830\text{e}3 \text{ lbf}$
 $W_{s-struct-corr} = 0.0 \text{ lbf}$
 $W_{stairs-corr} = 0.0 \text{ lbf}$
 $W_{struct} = 1.49951\text{e}3 \text{ lbf}$
 $W_{struct-corr} = 0.0 \text{ lbf}$
 $W_{windgirder-corr} = 0.0 \text{ lbf}$

Design Uplift Pressure per *API-650 5.2.1.k*

The wind uplift is limited to the case where the roof to shell junction will fail and overturning is not

possible.

$\text{wind-uplift} = \text{MIN}(P_{WR}, ((1.6 * P_{F41}) - P_g))$
 $\text{wind-uplift} = \text{MIN}(14.440, ((1.6 * 328.910) - 15.6066))$
 $\text{wind-uplift} = 14.440 \text{ psf}$

Overturning Moments

$X_w = D / 2$
 $X_w = 10.0 / 2$
 $X_w = 5.0 \text{ ft}$

$A_{h\text{-total}} = \pi * ((R_h + t_{r\text{-ins}})^2)$
 $A_{h\text{-total}} = \pi * ((5.06446 + 0.0)^2)$
 $A_{h\text{-total}} = 80.5779 \text{ ft}^2$

$M_{Pi} = P_g * A_h * X_w$
 $M_{Pi} = 15.6066 * 80.5779 * 5.0$
 $M_{Pi} = 6.28772\text{e}3 \text{ ft.lbf}$

$M_{WR} = \text{wind-uplift} * A_{h\text{-total}} * X_w$
 $M_{WR} = 14.440 * 80.5779 * 5.0$
 $M_{WR} = 5.81772\text{e}3 \text{ ft.lbf}$

$D_{\text{-outer}} = OD + (2 * (t_{s\text{-ins}} / 12))$
 $D_{\text{-outer}} = 10.0313 + (2 * (0.0 / 12))$
 $D_{\text{-outer}} = 10.0313 \text{ ft}$

$A_s = D_{\text{-outer}} * H$
 $A_s = 10.0313 * 30.0$
 $A_s = 300.938 \text{ ft}^2$

$X_s = H / 2$
 $X_s = 30.0 / 2$
 $X_s = 15.0 \text{ ft}$

$M_{WS} = P_{WS} * A_s * X_s$
 $M_{WS} = 8.6640 * 300.938 * 15.0$
 $M_{WS} = 39.1098\text{e}3 \text{ ft.lbf}$

$M_w = M_{WR} + M_{WS}$
 $M_w = 5.81772\text{e}3 + 39.1098\text{e}3$
 $M_w = 44.9275\text{e}3 \text{ ft.lbf}$

Resistance to Overturning per API-650 5.11.2

$M_{DL} = (D / 2) * DLS$
 $M_{DL} = (10.0 / 2) * 7.48169\text{e}3$
 $M_{DL} = 37.4085\text{e}3 \text{ ft.lbf}$

$M_{DLR} = (D / 2) * DLR$
 $M_{DLR} = (10.0 / 2) * 900.493$
 $M_{DLR} = 4.50247\text{e}3 \text{ ft.lbf}$

$wL = \text{MIN}((0.45 * L_{\text{max}} * D), (4.67 * (t_{b\text{-req}} - CA_{\text{-btm}}) * \text{SQRT}((F_{by} * L_{\text{max}}))))$
 $wL = \text{MIN}((0.45 * 30.0 * 10.0), (4.67 * (0.18750 - 0.0) * \text{SQRT}((21.8750\text{e}3 * 30.0))))$
 $wL = 135.0 \text{ lbf/ft}$

$$M_F = (D / 2) * wL * \pi * D$$

$$M_F = (10.0 / 2) * 135.0 * \pi * 10.0$$

$$M_F = 21.2058e3 \text{ ft.lbf}$$

An unanchored tank must meet the criteria from API-650 5.11.2.1

Criterion 1

$$((0.6 * M_w) + M_{Pi}) < ((M_{DL} / 1.5) + M_{DLR})$$

$$((0.6 * 44.9275e3) + 6.28772e3) < ((37.4085e3 / 1.5) + 4.50247e3)$$

Since $33.2442e3 \geq 29.4414e3 \implies$ Tank must be anchored

Criterion 2

$$(M_w + (F_p * M_{Pi})) < (((M_{DL} + M_F) / 2) + M_{DLR})$$

$$(44.9275e3 + (0.40 * 6.28772e3)) < (((37.4085e3 + 21.2058e3) / 2) + 4.50247e3)$$

Since $47.4426e3 \geq 33.8096e3 \implies$ Tank must be anchored

Criterion 3

$$(M_{WS} + (F_p * M_{Pi})) < ((M_{DL} / 1.5) + M_{DLR})$$

$$(39.1098e3 + (0.40 * 6.28772e3)) < ((37.4085e3 / 1.5) + 4.50247e3)$$

Since $41.6249e3 \geq 29.4414e3 \implies$ Tank must be anchored

Resistance to Sliding per API-650 5.11.4

$$F_{wind} = P_{WS} * A_s$$

$$F_{wind} = 8.6640 * 300.938$$

$$F_{wind} = 2.60732e3 \text{ lbf}$$

$$F_{friction} = COF * (W_{r-pl-corr} + W_{struct-corr} + W_{s-pl-corr} + W_{s-framing-corr} + W_{b-pl-corr} + W_{stairs-corr} + W_{access-corr} + W_{app-corr} + W_{windgirder-corr})$$

$$F_{friction} = 0.40 * (639.628 + 0.0 + 7.37830e3 + 100.397 + 639.550 + 0.0 + 0.0 + 759.415 + 0.0)$$

$$F_{friction} = 3.80692e3 \text{ lbf}$$

$$F_{friction} \geq F_{wind} \implies \text{Tank is stable}$$

Anchorage Requirement

Tank must be anchored per API-650 5.11 by Criterion 1, Criterion 2, and Criterion 3

Seismic Design [Back](#)

Site Ground Motion Design

Seismic Method = ASCE7-MAPPED-SS-AND-S1

Ac = Convective Design Response Spectrum Acceleration Coefficient per *API 650 Sections E.4.6.1*

Ac-min = Adjusted Convective Design Response Spectrum Acceleration Coefficient

Af = Acceleration Coefficient for Sloshing Wave Height per *API 650 Sections E.7.2*

Ai = per *API 650 Sections E.4.6.1*

Ai = Impulsive Design Response Spectrum Acceleration Coefficient per *API 650 Sections E.4.6.1*

Anchorage_System = Anchorage System

Av = Vertical Ground Acceleration Coefficient per *API 650 Sections E.6.1.3 and E.2.2*

D = Nominal Tank Diameter (ft)

Fa = Site Acceleration Coefficient

Fv = Site Velocity Coefficient

H = Maximum Design Product Level (ft)

I = Importance Factor

K = Spectral Acceleration Adjustment Coefficient

Ks = Sloshing Coefficient per *API 650 Section E.4.5.2*

Q = MCE to Design Level Scale Factor

rho_product = Product Mass Density (lb/ft³)

Rwc = Convective Force Reduction Factor

Rwi = Impulsive Force Reduction Factor

S1 = Spectral Response Acceleration at a Period of One Second

SD1 = Design Spectral Response Acceleration at a Period of One Second per *API 650 Sections E.4.6.1 and E.2.2*

SDS = Design Spectral Response Acceleration at Short Period per *API 650 Sections E.4.6.1 and E.2.2*

Seismic_Site_Class = Seismic Site Class

Seismic_Use_Group = Seismic Use Group

shell-course-modulus-of-elasticity-information-list = Shell Course Modulus of Elasticity Information List (psi)

shell-course-thickness-information-list = Shell Course Thickness Information List (in)

Ss = Spectral Response Acceleration Short Period

Tc = Convective Natural Period per *API 650 Section E.4.5.2* (sec)

TL = Regional Dependent Transistion Period for Longer Period Ground Motion (sec)

Anchorage_System = MECHANICALLY-ANCHORED

D = 10.0 ft

Fa = 1.20

Fv = 1.70

H = 30.0 ft

I = 1.250

K = 1.50

Q = 0.666667

rho_product = 68.6707 lb/ft³

Rwc = 2.0

Rwi = 4.0

S1 = 0.0930

Seismic_Site_Class = SEISMIC-SITE-CLASS-C

Seismic_Use_Group = SEISMIC-USE-GROUP-II

shell-course-modulus-of-elasticity-information-list = [26.550e6 26.550e6 26.550e6] psi

shell-course-thickness-information-list = [0.18750 0.18750 0.18750] in

Ss = 0.240

TL = 12.0 sec

$$\begin{aligned} \text{SDS} &= Q * F_a * S_s \\ \text{SDS} &= 0.666667 * 1.20 * 0.240 \\ \text{SDS} &= 0.1920 \end{aligned}$$

$$\begin{aligned} \text{SD1} &= Q * F_v * S_1 \\ \text{SD1} &= 0.666667 * 1.70 * 0.0930 \\ \text{SD1} &= 0.10540 \end{aligned}$$

$$\begin{aligned} K_s &= 0.578 / \text{SQRT}(\text{TANH}(((3.68 * H) / D))) \\ K_s &= 0.578 / \text{SQRT}(\text{TANH}(((3.68 * 30.0) / 10.0))) \\ K_s &= 0.5780 \end{aligned}$$

$$\begin{aligned} T_c &= K_s * \text{SQRT}(D) \\ T_c &= 0.5780 * \text{SQRT}(10.0) \\ T_c &= 1.82780 \text{ sec} \end{aligned}$$

$$\begin{aligned} A_i &= \text{SDS} * (I / R_{wi}) \\ A_i &= 0.1920 * (1.250 / 4.0) \\ A_i &= 0.060 \end{aligned}$$

$$\begin{aligned} A_i &= \text{MAX}(A_i, 0.007) \\ A_i &= \text{MAX}(0.060, 0.007) \\ A_i &= 0.060 \end{aligned}$$

$$T_c \leq T_L$$

$$\begin{aligned} A_c &= K * \text{SD1} * (1 / T_c) * (I / R_{wc}) \\ A_c &= 1.50 * 0.10540 * (1 / 1.82780) * (1.250 / 2.0) \\ A_c &= 0.0540610 \end{aligned}$$

$$\begin{aligned} A_{c\text{-min}} &= \text{MIN}(A_c, A_i) \\ A_{c\text{-min}} &= \text{MIN}(0.0540610, 0.060) \\ A_{c\text{-min}} &= 0.0540610 \end{aligned}$$

$$\begin{aligned} A_v &= (2 / 3) * 0.7 * \text{SDS} \\ A_v &= (2 / 3) * 0.7 * 0.1920 \\ A_v &= 0.08960 \end{aligned}$$

$$\begin{aligned} A_f &= K * \text{SD1} * I * (1 / T_c) \\ A_f &= 1.50 * 0.10540 * 1.250 * (1 / 1.82780) \\ A_f &= 0.108122 \end{aligned}$$

Seismic Design

$$\begin{aligned} h_{\text{bottom}} &= \text{Bottom Elevation (ft)} \\ h_{\text{bottom-ground}} &= \text{Bottom Elevation from Ground (ft)} \end{aligned}$$

$$\begin{aligned} h_{\text{bottom}} &= 0.0 \text{ ft} \\ h_{\text{bottom-ground}} &= 0.0 \text{ ft} \end{aligned}$$

$$\begin{aligned} A &= \text{Roof Surface Area (ft}^2\text{)} \\ A_c &= \text{Convective Design Response Spectrum Acceleration Coefficient} \\ A_f &= \text{Acceleration Coefficient for Sloshing Wave Height} \\ A_{h\text{-shell}} &= \text{Roof Horizontal Projected Area Supported by The Shell (ft}^2\text{)} \\ A_i &= \text{Impulsive Design Response Spectrum Acceleration Coefficient} \end{aligned}$$

Anchorage_System = Anchorage System
 A-rs = Roof Area Supported by The Shell (ft²)
 Av = Vertical Ground Acceleration Coefficient
 ca1 = Bottom Shell Course Corrosion Allowance (in)
 ca_bottom = Bottom Corrosion Allowance (in)
 D = Nominal Tank Diameter (ft)
 DELTAs = Sloshing Wave Height Above Product Design Height per *API 650 Section E.7.2* (ft)
 Event_Type = Event Type
 Fa = Site Acceleration Coefficient
 F_c = Allowable Longitudinal Shell Compression Stress per *API 650 Section E.6.2.2.3, Eq E.6.2.2.3-2b* (lbf/in²)
 Freeboard = Actual Freeboard (ft)
 Freeboard_recommended = Minimum Recommended Freeboard per *API-650 Table E.7* (ft)
 Fv = Site Velocity Coefficient
 Fy = Yield Strength (lb/in²)
 G = Specific Gravity
 Ge = Effective Specific Gravity per *API 650 Section E.2.2*
 H = Maximum Design Product Level (ft)
 Hrcg = Top of Shell to Roof and roof appurtenances Center of Gravity (ft)
 hs = Additional Shell Height Required Above Sloshing Height (ft)
 H_shell = Shell height (ft)
 I = Importance Factor
 J = Anchorage Ratio per *API 650 Section E.6.2.1.1.1*
 K = Spectral Acceleration Adjustment Coefficient
 Ks = Sloshing Coefficient
 Min_Anchor_Quantity = Minimum Anchor Quantity per *API-650 5.12.2*
 Min_Anchor_Spacing = Minimum Anchor Spacing per *API-650 5.12.3* (ft)
 Mrw = Ringwall Overturning Moment per *API 650 Section E.6.1.5* (ft.lb)
 Ms = Slab Overturning Moment per *API 650 Section E.6.1.5* (ft.lb)
 mu = Friction Coefficient per *API 650, Section E.7.6*
 Overturn_Stability_Ratio = Overturning Stability Ratio per *API 650 Section E.6.2.3*
 P = Design Pressure (lbf/in²)
 Q = MCE to Design Level Scale Factor
 S1 = Spectral Response Acceleration at a Period of One Second
 Sb = Roof Balanced Snow Load (psf)
 SD1 = Design Spectral Response Acceleration at a Period of 1 Second
 SDS = Design Spectral Response Acceleration at Short Period
 Seismic_Site_Class = Seismic Site Class
 Seismic_Use_Group = Seismic Use Group
 sigma_c = Mechanically Anchored Maximum Longitudinal Shell Compression Stress per *API 650 Section E.6.2.2.2, Eq E.6.2.2.2-1b* (lbf/in²)
 Ss = Spectral Response Acceleration Short Period
 ta = Thickness, excluding corrosion allowance, of the bottom annulus under the shell required to provide the resisting force for self anchorage per *API-650 E.2.2* (in)
 tb-corr = Bottom Plates Corroded Thickness (in)
 t_bottom = Bottom Plate Thickness (in)
 Tc = Convective Natural Period (sec)
 TL = Regional Dependent Transistion Period for Longer Period Ground Motion (sec)
 ts1 = Bottom Shell Course Thickness (in)
 ts1_c = Shell Course 1 Corroded Thickness (in)
 V = Total Design Base Shear per *API 650 Section E.6.1* (lbf)
 Vc = Design Base Shear for Convective Component per *API 650 Section E.6.1* (lbf)
 Vi = Design Base Shear for Impulsive Component per *API 650 Section E.6.1* (lbf)
 Vmax = Local Shear Transfer per *API 650 Section E.7.7* (lbf/ft)
 Vs = Self Anchored Sliding Resistance Maximum Allowable Base Shear per *API 650 Section E.7.6* (lbf)
 w_a = Force Resisting Uplift - Self Anchored per *API 650 Section E.6.2.1.1* (lbf/ft)

$w_a = (\text{lbf/ft})$
 $w_{a_limit} = \text{Self Anchored Force Resisting Uplift Max Limit per API 650 Section E.6.2.1.1, Eq E.6.2.1.1-1b}$
 (lbf/ft)
 $w_{a_self-anchored} = \text{Self Anchored Force Resisting Uplift per API 650 Section E.6.2.1.1, Eq E.6.2.1.1-1b}$
 (lbf/ft)
 $W_{b-attachments} = \text{Bottom Attachments Weight (lb)}$
 $W_{b-pl} = \text{Bottom Plates Weight (lb)}$
 $W_c = \text{Convective Effective Weight per API 650 Section E.6.1.1 (lbf)}$
 $W_{eff} = \text{Total Effective Weight per API 650 Section E.6.1.1 (lbf)}$
 $W_f = \text{Tank Bottom Total Weight (lbf)}$
 $W_{fd} = \text{Tank Foundation Weight (lbf)}$
 $W_g = \text{Soil Weight (lbf)}$
 $W_i = \text{Impulsive Effective Weight per API 650 Section E.6.1.1 (lbf)}$
 $w_{int} = \text{Calculated Design Uplift Due to Product Pressure (lbf/ft)}$
 $W_p = \text{Tank Contents Total Weight (lbf)}$
 $W_r = \text{Total Weight of Fixed Tank Roof including Framing, Knuckles, any Permanent Attachments and 10}$
 $\% \text{ of the Roof Balanced Design Snow Load (lbf)}$
 $W_{r-attachments} = \text{Roof Attachments Weight (lb)}$
 $W_{r-DL-add} = \text{Roof Additional Dead Weight (lb)}$
 $W_{r-pl} = \text{Roof Plates Nominal Weight (lb)}$
 $w_{rs} = \text{Specified Tank Roof Load Acting on Tank Shell (lbf/ft)}$
 $W_{rs} = \text{Roof Load Acting on The Tank Shell Including 10 \% of the Roof Balanced Design Snow Load (lbf)}$
 $W_s = \text{Total Weight of Tank Shell and Appurtenances (lbf)}$
 $W_{s-attachments} = \text{Shell Attachments Weight (lb)}$
 $W_{s-framing} = \text{Shell Framing Weight (lb)}$
 $W_{s-pl} = \text{Shell Plates Nominal Weight (lb)}$
 $W_{ss} = \text{Roof Structure Weight Supported by The Tank Shell (lb)}$
 $W_{struct} = \text{Roof Structure Weight (lb)}$
 $W_T = \text{Total Weight of Tank Shell, Roof, Framing, Knuckles, Product, Bottom, Attachments,}$
 $\text{Appurtenances, Participating Balanced Snow Load per API-650 Eq E.6.2.3-1 (lbf)}$
 $w_t = \text{Tank and Roof Weight Acting at base of Shell per API 650 Section E.6.2.1.1.1 (lbf/ft)}$
 $X_c = \text{Height from tank shell bottom to the center of action of convective lateral force for computing}$
 $\text{ringwall overturning moment per API 650 Section E.6.1.2.1 (ft)}$
 $X_{cs} = \text{Height from tank shell bottom to the center of action of convective lateral force for computing slab}$
 $\text{overturning moment per API 650 Section E.6.1.2.2 (ft)}$
 $X_i = \text{Height from tank shell bottom to the center of action of impulsive lateral force for computing ringwall}$
 $\text{overturning moment per API 650 Section E.6.1.2.1 (ft)}$
 $X_{is} = \text{Height from tank shell bottom to the center of action of impulsive lateral force for computing slab}$
 $\text{overturning moment per API 650 Section E.6.1.2.2 (ft)}$
 $X_r = \text{Height from tank shell bottom to the center of gravity of roof and roof appurtenances per API 650}$
 $\text{Section E.6.1.2 (ft)}$
 $X_s = \text{Height from tank shell bottom to shell's center of gravity (ft)}$

$A = 81.6894 \text{ ft}^2$
 $A_c = 0.05410$
 $A_f = 0.10810$
 $A_{h-shell} = 80.5779 \text{ ft}^2$
 $A_i = 0.060$
 $\text{Anchorage_System} = \text{MECHANICALLY-ANCHORED}$
 $A_{rs} = 81.6894 \text{ ft}^2$
 $A_v = 0.08960$
 $ca_1 = 0.0 \text{ in}$
 $ca_{bottom} = 0.0 \text{ in}$
 $D = 10.0 \text{ ft}$
 $\text{Event_Type} = \text{MAXIMUM-CONSIDERED-EARTHQUAKE-MCE}$
 $F_a = 1.20$

$F_v = 1.70$
 $F_y = 21.8750 \text{e}3 \text{ lb/in}^2$
 $G = 1.10$
 $H = 30.0 \text{ ft}$
 $H_{rcg} = 0.281359 \text{ ft}$
 $h_s = 0.0 \text{ ft}$
 $H_{\text{shell}} = 30.0 \text{ ft}$
 $I = 1.250$
 $K = 1.50$
 $K_s = 0.5780$
 $\text{Min_Anchor_Quantity} = 4.0$
 $\text{Min_Anchor_Spacing} = 10.0 \text{ ft}$
 $\mu = 0.40$
 $P = 0.108379 \text{ lbf/in}^2$
 $Q = 0.666667$
 $S_1 = 0.0930$
 $S_b = 0.0 \text{ psf}$
 $SD_1 = 0.10540$
 $SDS = 0.1920$
 $\text{Seismic_Site_Class} = \text{SEISMIC-SITE-CLASS-C}$
 $\text{Seismic_Use_Group} = \text{SEISMIC-USE-GROUP-II}$
 $S_s = 0.240$
 $t_{\text{bottom}} = 0.18750 \text{ in}$
 $T_c = 1.82780 \text{ sec}$
 $TL = 12.0 \text{ sec}$
 $ts_1 = 0.18750 \text{ in}$
 $W_{b\text{-attachments}} = 0.0 \text{ lb}$
 $W_{b\text{-pl}} = 639.550 \text{ lb}$
 $W_{fd} = 0.0 \text{ lbf}$
 $W_g = 0.0 \text{ lbf}$
 $W_p = 161.802 \text{e}3 \text{ lbf}$
 $W_{r\text{-attachments}} = 260.865 \text{ lb}$
 $W_{r\text{-DL-add}} = 0.0 \text{ lb}$
 $W_{r\text{-pl}} = 639.628 \text{ lb}$
 $W_{s\text{-attachments}} = 593.550 \text{ lb}$
 $W_{s\text{-framing}} = 103.391 \text{ lb}$
 $W_{s\text{-pl}} = 7.37830 \text{e}3 \text{ lb}$
 $W_{ss} = 1.49951 \text{e}3 \text{ lb}$
 $W_{\text{struct}} = 1.49951 \text{e}3 \text{ lb}$
 $X_s = 14.9740 \text{ ft}$

Seismic Method (seismic-method) = ASCE7-MAPPED-SS-AND-S1

Weights

$W_f = W_{b\text{-pl}}$
 $W_f = 639.550$
 $W_f = 639.550 \text{ lbf}$

$W_r = (W_{r\text{-pl}} + W_{r\text{-attachments}} + W_{\text{struct}} + W_{r\text{-DL-add}}) + (0.1 * S_b * A_h)$
 $W_r = (639.628 + 260.865 + 1.49951 \text{e}3 + 0.0) + (0.1 * 0.0 * 80.5779)$
 $W_r = 2.40 \text{e}3 \text{ lbf}$

$W_{rs} = ((W_{r\text{-pl}} + W_{r\text{-attachments}} + W_{r\text{-DL-add}}) * (A_{rs} / A)) + W_{ss} + (0.1 * S_b * A_{h\text{-shell}})$
 $W_{rs} = ((639.628 + 260.865 + 0.0) * (81.6894 / 81.6894)) + 1.49951 \text{e}3 + (0.1 * 0.0 * 80.5779)$
 $W_{rs} = 2.40 \text{e}3 \text{ lbf}$

$W_s = W_{s-pl} + W_{s-framing} + W_{s-attachments}$
 $W_s = 7.37830e3 + 103.391 + 593.550$
 $W_s = 8.07524e3 \text{ lbf}$

$W_T = W_s + W_r + W_p + W_f$
 $W_T = 8.07524e3 + 2.40e3 + 161.802e3 + 639.550$
 $W_T = 172.916e3 \text{ lbf}$

Effective Weight of Product

$W_i = (1.0 - (0.218 * (D / H))) * W_p$
 $W_i = (1.0 - (0.218 * (10.0 / 30.0))) * 161.802e3$
 $W_i = 150.044e3 \text{ lbf}$

$W_c = 0.23 * (D / H) * \tanh(((3.67 * H) / D)) * W_p$
 $W_c = 0.23 * (10.0 / 30.0) * \tanh(((3.67 * 30.0) / 10.0)) * 161.802e3$
 $W_c = 12.4048e3 \text{ lbf}$

$W_{eff} = W_i + W_c$
 $W_{eff} = 150.044e3 + 12.4048e3$
 $W_{eff} = 162.449e3 \text{ lbf}$

Design Loads

$V_i = A_i * (W_s + W_r + W_f + W_i)$
 $V_i = 0.060 * (8.07524e3 + 2.40e3 + 639.550 + 150.044e3)$
 $V_i = 9.66952e3 \text{ lbf}$

$V_c = A_c * W_c$
 $V_c = 0.05410 * 12.4048e3$
 $V_c = 671.099 \text{ lbf}$

$V = \sqrt{V_i^2 + V_c^2}$
 $V = \sqrt{(9.66952e3)^2 + (671.099)^2}$
 $V = 9.69278e3 \text{ lbf}$

Center of Action for Effective Lateral Forces

$X_r = H_{shell} + H_{rcg}$
 $X_r = 30.0 + 0.281359$
 $X_r = 30.2814 \text{ ft}$

$X_i = (0.5 - (0.094 * (D / H))) * H$
 $X_i = (0.5 - (0.094 * (10.0 / 30.0))) * 30.0$
 $X_i = 14.060 \text{ ft}$

$X_c = (1.0 - ((\cosh(((3.67 * H) / D)) - 1) / (((3.67 * H) / D) * \sinh(((3.67 * H) / D)))) * H$
 $X_c = (1.0 - ((\cosh(((3.67 * 30.0) / 10.0)) - 1) / (((3.67 * 30.0) / 10.0) * \sinh(((3.67 * 30.0) / 10.0)))) * 30.0$
 $X_c = 27.2753 \text{ ft}$

$X_{is} = (0.5 + (0.06 * (D / H))) * H$
 $X_{is} = (0.5 + (0.06 * (10.0 / 30.0))) * 30.0$
 $X_{is} = 15.60 \text{ ft}$

$X_{cs} = (1.0 - ((\cosh(((3.67 * H) / D)) - 1.937) / (((3.67 * H) / D) * \sinh(((3.67 * H) / D)))) * H$
 $X_{cs} = (1.0 - ((\cosh(((3.67 * 30.0) / 10.0)) - 1.937) / (((3.67 * 30.0) / 10.0) * \sinh(((3.67 * 30.0) / 10.0)))) * 30.0$
 $X_{cs} = 27.2754 \text{ ft}$

Overturing Moment

$$Mrw = \text{SQRT}((((Ai * ((Wi * Xi) + (Ws * Xs) + (Wr * Xr)))^2) + ((Ac * (Wc * Xc))^2)))$$

$$Mrw = \text{SQRT}((((0.060 * ((150.044e3 * 14.060) + (8.07524e3 * 14.9740) + (2.40e3 * 30.2814)))^2) + ((0.05410 * (12.4048e3 * 27.2753))^2)))$$

$$Mrw = 139.40e3 \text{ ft.lb}$$

$$Ms = \text{SQRT}((((Ai * ((Wi * Xis) + (Ws * Xs) + (Wr * Xr)))^2) + ((Ac * (Wc * Xcs))^2)))$$

$$Ms = \text{SQRT}((((0.060 * ((150.044e3 * 15.60) + (8.07524e3 * 14.9740) + (2.40e3 * 30.2814)))^2) + ((0.05410 * (12.4048e3 * 27.2754))^2)))$$

$$Ms = 153.155e3 \text{ ft.lb}$$

Resistance to Design Loads

$$Ge = G * (1 - (0.4 * Av))$$

$$Ge = 1.10 * (1 - (0.4 * 0.08960))$$

$$Ge = 1.06058$$

$$wrs = Wrs / (\pi * D)$$

$$wrs = 2.40e3 / (\pi * 10.0)$$

$$wrs = 76.3945 \text{ lbf/ft}$$

$$wt = (Ws / (\pi * D)) + wrs$$

$$wt = (8.07524e3 / (\pi * 10.0)) + 76.3945$$

$$wt = 333.437 \text{ lbf/ft}$$

$$wint = P * 144 * ((\pi * ((D^2) / 4)) / (\pi * D))$$

$$wint = 0.108379 * 144 * ((\pi * ((10.0^2) / 4)) / (\pi * 10.0))$$

$$wint = 39.0164 \text{ lbf/ft}$$

Bottom Annular Plates Requirements

$$tb_corr = t_bottom - ca_bottom$$

$$tb_corr = 0.18750 - 0.0$$

$$tb_corr = 0.18750 \text{ in}$$

$$ts1_c = ts1 - ca1$$

$$ts1_c = 0.18750 - 0.0$$

$$ts1_c = 0.18750 \text{ in}$$

$$ta = \text{MIN}(tb_corr, ts1_c)$$

$$ta = \text{MIN}(0.18750, 0.18750)$$

$$ta = 0.18750 \text{ in}$$

$$wa_limit = 1.28 * H * D * Ge$$

$$wa_limit = 1.28 * 30.0 * 10.0 * 1.06058$$

$$wa_limit = 407.261 \text{ lbf/ft}$$

$$w_a_self\text{-anchored} = 7.9 * ta * \text{SQRT}((Fy * H * Ge))$$

$$w_a_self\text{-anchored} = 7.9 * 0.18750 * \text{SQRT}((21.8750e3 * 30.0 * 1.06058))$$

$$w_a_self\text{-anchored} = 1.23576e3 \text{ lbf/ft}$$

$$w_a = \text{MIN}(w_a_self\text{-anchored}, wa_limit)$$

$$w_a = \text{MIN}(1.23576e3, 407.261)$$

$$w_a = 407.261 \text{ lbf/ft}$$

$$w_a = w_a$$

$$w_a = 407.261$$

$$w_a = 407.261 \text{ lbf/ft}$$

Tank Stability

$$J = Mrw / ((D^2) * (((wt * (1 - (0.4 * Av))) + wa) - (Fp * wint)))$$

$$J = 139.40e3 / ((10.0^2) * (((333.437 * (1 - (0.4 * 0.08960))) + 407.261) - (0.40 * 39.0164)))$$

$$J = 1.95473$$

As per API 650, Section E.6.2.1.1.1, Table E.6

$J > 1.54 \Rightarrow$ Tank is not stable, anchoring is required

Seismic Method (seismic-method) = ASCE7-MAPPED-SS-AND-S1

$$\sigma_c = ((wt * (1 + (0.4 * Av))) + ((1.273 * Mrw) / (D^2))) * (1 / (12 * ts))$$

$$\sigma_c = ((333.437 * (1 + (0.4 * 0.08960))) + ((1.273 * 139.40e3) / (10.0^2))) * (1 / (12 * 0.18750))$$

$$\sigma_c = 942.198 \text{ lbf/in}^2$$

$$F_c = (1.0E6 * (ts / (2.5 * D))) + (600 * \text{SQRT}((G * H)))$$

$$F_c = (1.0E6 * (0.18750 / (2.5 * 10.0))) + (600 * \text{SQRT}((1.10 * 30.0)))$$

$$F_c = 10.9467e3 \text{ lbf/in}^2$$

$$\sigma_c < F_c$$

$$\text{Overturn_Stability_Ratio} = (0.5 * D * (W_T + W_{fd} + W_g)) / M_s$$

$$\text{Overturn_Stability_Ratio} = (0.5 * 10.0 * (172.916e3 + 0.0 + 0.0)) / 153.155e3$$

$$\text{Overturn_Stability_Ratio} = 5.64516$$

$\text{Overturn_Stability_Ratio} \geq 2.0 \Rightarrow$ PASS

Freeboard

$$\text{DELTA}s = 0.42 * D * A_f$$

$$\text{DELTA}s = 0.42 * 10.0 * 0.10810$$

$$\text{DELTA}s = 0.454020 \text{ ft}$$

$$\text{Freeboard} = H_{\text{shell}} - L_{\text{max-operating}}$$

$$\text{Freeboard} = 30.0 - 30.0$$

$$\text{Freeboard} = 3.55271e-15 \text{ ft} [42.6326e-15 \text{ in}]$$

$$\text{Freeboard_recommended} = 0.7 * \text{DELTA}s$$

$$\text{Freeboard_recommended} = 0.7 * 0.454020$$

$$\text{Freeboard_recommended} = 0.317814 \text{ ft} [3.81377 \text{ in}]$$

As per API-650 E.7.2 and Table E.7, freeboard is recommended but not required

Sliding Resistance

$$V_s = MU * (W_s + W_r + W_f + W_p) * (1.0 - (0.4 * A_v))$$

$$V_s = 0.40 * (8.07524e3 + 2.40e3 + 639.550 + 161.802e3) * (1.0 - (0.4 * 0.08960))$$

$$V_s = 66.6876e3 \text{ lbf}$$

$$V \leq V_s$$

Local Shear Transfer

$$V_{\text{max}} = (2 * V) / (\pi * D)$$

$$V_{\text{max}} = (2 * 9.69278e3) / (\pi * 10.0)$$

$$V_{\text{max}} = 617.062 \text{ lbf/ft}$$

Anchor Bolt Design [Back](#)

A-s = Installed Bolt Nominal Root Area (in²)
A-s-r = Anchor Required Root Area (in²)
Av = Seismic Vertical Earthquake Acceleration Coefficient (g)
Ca-anchor = Anchor Corrosion Allowance (in)
d = Anchor Bolt Diameter (in)
D = Tank nominal diameter (ft)
Dac = Bolt Circle Diameter (ft)
d-req = Bolt Required Diameter per *ANSI B1.1* (in)
Fp = Design Pressure Operating Ratio
Fty = Minimum Yield Strength of the Bottom Shell Course (psi)
Fy = Anchor Yield Strength per *API-650 Table 5.21b* (psi)
Fy-ambient = Anchor Yield Strength at Ambient Temperature per *API-650 Table 5.21b* (psi)
H = Tank Height (ft)
Ma-anchor = Anchor Material
Mrw = Seismic Overturning Moment (ft.lbf)
MWS = Shell Wind Overturning Moment (ft.lbf)
N = Anchors Quantity
n = Number of threads per unit length (in)
N-min = Minimum Required Number of Anchors per *API-650 5.12.3*
OD = Tank Outer diameter (ft)
P = Internal Pressure (inH2O)
p = Bolt Thread Pitch
P_attachment = Anchor Attachment Design Load per *API-650 Section 5.12.13 and Steel Plate Engineering Data-Volume 2 Part V* (lbf)
position_angles = Anchors Position Angles (deg)
Pt = Test Pressure (inH2O)
PWR = Roof Wind Pressure (inH2O)
S_d = Allowable Anchor Stress per *API-650 Table 5.20b* (psi)
S_d_shell = Allowable Shell Stress at Anchor Attachment per *API-650 Table 5.20b* (psi)
T_b = Load per Anchor per *API-650 5.12.2* (lbf)
U = Net Uplift Load per *API-650 Section 5.12.2 and Table 5.20b* (lbf)
W1 = Corroded Weight of the Roof Plates Plus the Corroded Weight of the Shell and any Other Corroded Permanent Attachments Acting on the Shell (lbf)
W2 = Corroded Weight of the Shell and any Corroded Permanent Attachments Acting on the Shell Including the Portion of the Roof Plates and Framing Acting on The Shell (lbf)
W3 = Nominal Weight of the Roof Plates Plus the Nominal Weight of the Shell and any Other Permanent Attachments Acting on the Shell (lbf)
Wr-pl = Roof Plates Nominal Weight (lb)
Wr-pl-corr = Roof Corroded Plates Weight (lb)
Wrs-pl-corr = Roof Plates Corroded Weight Acting on The Shell (lb)
Ws-framing = Shell New Framing Weight (stiffeners) (lbf)
Ws-framing-corr = Shell Corroded Framing Weight (stiffeners) (lbf)
Ws-pl = Shell Plates Nominal Weight (lb)
Ws-pl-corr = Shell Corroded Plates Weight (lb)
Wss = Roof Structure Nominal Weight Acting on The Shell (lb)
Wss-corr = Roof Structure Corroded Weight Acting on The Shell (lb)
Y-bolt = Anchor Yield Load (lbf)

Av = 0.08960 g
Ca-anchor = 0.0 in
D = 10.0 ft
d = 1.0 in

Dac = 10.3646 ft
 Fp = 0.40
 Fty = 21.8750e3 psi
 H = 30.0 ft
 Ma-anchor = A36
 Mrw = 139.40e3 ft.lbf
 MWS = 39.1098e3 ft.lbf
 n = 8.0 in
 N = 4.0
 OD = 10.0313 ft
 p = 0.130
 P = 3.0 inH2O
 position_angles = [22.50 112.50 202.50 292.50] deg
 Pt = 3.750 inH2O
 PWR = 2.77576 inH2O
 Wr-pl = 639.628 lb
 Wr-pl-corr = 639.628 lb
 Wrs-pl-corr = 639.628 lb
 Ws-framing = 103.391 lbf
 Ws-framing-corr = 100.397 lbf
 Ws-pl = 7.37830e3 lb
 Ws-pl-corr = 7.37830e3 lb
 Wss = 1.49951e3 lb
 Wss-corr = 0.0 lb

Anchors Spacing Requirements

Max Allowable Spacing Between Anchors at Shell Outer Diameter per API-650 5.12.3

Max Allowable Spacing (max_allowable_spacing) = 10.0 ft
 Actual Spacing (actual_spacing) = 7.87853 ft

actual_spacing <= max_allowable_spacing ==> PASS

N-min = CEILING(((pi * OD) / 10))
 N-min = CEILING(((pi * 10.0313) / 10))
 N-min = 4.0

N >= N-min ==> PASS

Anchors meet spacing requirements.

Anchors Average Spacing (half the span on each side of the anchor) at Bolt Circle

Anchors are equally spaced.
 Average Spacing (average_spacing) = 8.140 ft
 Bolt loads will be based on equally spaced anchors.

Anchor Material Properties

Material = A36
 Minimum Tensile Strength (Sut-anchor) = 58.0e3 psi
 As per API-650 S.5.b, Minimum Yield Strength at Ambient Temperature (Sy-ambient-anchor) = 36.0e3 psi
 As per API-650 Table M.1.b, Minimum Yield Strength (Sy-anchor) = 30.870e3 psi

Fy = MIN(Sy-anchor , 55000)
 Fy = MIN(30.870e3 , 55000)
 Fy = 30.870e3 psi

$F_{y-ambient} = \text{MIN}(S_{y-ambient-anchor}, 55000)$
 $F_{y-ambient} = \text{MIN}(36.0e3, 55000)$
 $F_{y-ambient} = 36.0e3 \text{ psi}$

Uplift Load Cases per API-650 Table 5.20b

$W1 = W_{s-pl-corr} + W_{s-framing-corr} + W_{r-pl-corr}$
 $W1 = 7.37830e3 + 100.397 + 639.628$
 $W1 = 8.11832e3 \text{ lbf}$

$W2 = W_{s-pl-corr} + W_{s-framing-corr} + W_{rs-pl-corr} + W_{ss-corr}$
 $W2 = 7.37830e3 + 100.397 + 639.628 + 0.0$
 $W2 = 8.11832e3 \text{ lbf}$

$W3 = W_{s-pl} + W_{s-framing} + W_{r-pl} + W_{ss}$
 $W3 = 7.37830e3 + 103.391 + 639.628 + 1.49951e3$
 $W3 = 9.62083e3 \text{ lbf}$

As per API-650 Section 5.12.13, Seismic Attachment Load Multiplier ($f_{seismic}$) = 3.0
Seismic anchoring is required and API-650 Section 5.12.13 requires a multiplier of 3.

Uplift Case 1: Design Pressure Only

$U = (P * (D^2) * 4.08) - W1$
 $U = (3.0 * (10.0^2) * 4.08) - 8.11832e3$
 $U = -6.89432e3 \text{ (Set to 0.0 lbf since it cannot be less than 0.0)}$

$T_b = U / N$
 $T_b = 0.0 / 4.0$
 $T_b = 0.0 \text{ lbf}$

$S_d = (5 / 12) * F_y$
 $S_d = (5 / 12) * 30.870e3$
 $S_d = 12.8625e3 \text{ psi}$

$A_{s-r} = T_b / S_d$
 $A_{s-r} = 0.0 / 12.8625e3$
 $A_{s-r} = 0.0 \text{ in}^2$

$P_{attachment} = 1.5 * T_b$
 $P_{attachment} = 1.5 * 0.0$
 $P_{attachment} = 0.0 \text{ lbf}$

$S_{d_shell} = (2 / 3) * F_{ty}$
 $S_{d_shell} = (2 / 3) * 21.8750e3$
 $S_{d_shell} = 14.5833e3 \text{ psi}$

Uplift Case 2: Test Pressure Only

$U = (P_t * (D^2) * 4.08) - W3$
 $U = (3.750 * (10.0^2) * 4.08) - 9.62083e3$
 $U = -8.09083e3 \text{ (Set to 0.0 lbf since it cannot be less than 0.0)}$

$T_b = U / N$
 $T_b = 0.0 / 4.0$
 $T_b = 0.0 \text{ lbf}$

$S_d = (5 / 9) * F_{y-ambient}$
 $S_d = (5 / 9) * 36.0e3$

$$S_d = 20.0e3 \text{ psi}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 0.0 / 20.0e3$$

$$A-s-r = 0.0 \text{ in}^2$$

$$P_{\text{attachment}} = 1.5 * T_b$$

$$P_{\text{attachment}} = 1.5 * 0.0$$

$$P_{\text{attachment}} = 0.0 \text{ lbf}$$

$$S_{d_shell} = (5 / 6) * F_{ty}$$

$$S_{d_shell} = (5 / 6) * 21.8750e3$$

$$S_{d_shell} = 18.2292e3 \text{ psi}$$

Uplift Case 3: Wind Load Only

$$U = ((PWR * (D^2) * 4.08) + ((4 * MWS) / D)) - W2$$

$$U = ((2.77576 * (10.0^2) * 4.08) + ((4 * 39.1098e3) / 10.0)) - 8.11832e3$$

$$U = 8.65811e3 \text{ lbf}$$

$$T_b = U / N$$

$$T_b = 8.65811e3 / 4.0$$

$$T_b = 2.16453e3 \text{ lbf}$$

$$S_d = 0.8 * F_y$$

$$S_d = 0.8 * 30.870e3$$

$$S_d = 24.6960e3 \text{ psi}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 2.16453e3 / 24.6960e3$$

$$A-s-r = 0.0876469 \text{ in}^2$$

$$P_{\text{attachment}} = 1.5 * T_b$$

$$P_{\text{attachment}} = 1.5 * 2.16453e3$$

$$P_{\text{attachment}} = 3.24679e3 \text{ lbf}$$

$$S_{d_shell} = (5 / 6) * F_{ty}$$

$$S_{d_shell} = (5 / 6) * 21.8750e3$$

$$S_{d_shell} = 18.2292e3 \text{ psi}$$

Uplift Case 4: Seismic Load Only

$$U = ((4 * Mrw) / D) - (W2 * (1 - (0.4 * A_v)))$$

$$U = ((4 * 139.40e3) / 10.0) - (8.11832e3 * (1 - (0.4 * 0.08960)))$$

$$U = 47.9325e3 \text{ lbf}$$

$$T_b = U / N$$

$$T_b = 47.9325e3 / 4.0$$

$$T_b = 11.9831e3 \text{ lbf}$$

$$S_d = 0.8 * F_y$$

$$S_d = 0.8 * 30.870e3$$

$$S_d = 24.6960e3 \text{ psi}$$

$$A-s-r = T_b / S_d$$

$$A-s-r = 11.9831e3 / 24.6960e3$$

$$A-s-r = 0.485225 \text{ in}^2$$

$P_{\text{attachment}} = f_{\text{seismic}} * T_b$
 $P_{\text{attachment}} = 3.0 * 11.9831\text{e}3$
 $P_{\text{attachment}} = 35.9494\text{e}3 \text{ lbf}$

$S_{d_shell} = (5 / 6) * F_{ty}$
 $S_{d_shell} = (5 / 6) * 21.8750\text{e}3$
 $S_{d_shell} = 18.2292\text{e}3 \text{ psi}$

Uplift Case 5: Design Pressure + Wind Load

$U = (((F_p * P) + PWR) * (D^2) * 4.08) + ((4 * MWS) / D)) - W1$
 $U = (((0.40 * 3.0) + 2.77576) * (10.0^2) * 4.08) + ((4 * 39.1098\text{e}3) / 10.0)) - 8.11832\text{e}3$
 $U = 9.14771\text{e}3 \text{ lbf}$

$T_b = U / N$
 $T_b = 9.14771\text{e}3 / 4.0$
 $T_b = 2.28693\text{e}3 \text{ lbf}$

$S_d = (5 / 9) * F_y$
 $S_d = (5 / 9) * 30.870\text{e}3$
 $S_d = 17.150\text{e}3 \text{ psi}$

$A-s-r = T_b / S_d$
 $A-s-r = 2.28693\text{e}3 / 17.150\text{e}3$
 $A-s-r = 0.133349 \text{ in}^2$

$P_{\text{attachment}} = 1.5 * T_b$
 $P_{\text{attachment}} = 1.5 * 2.28693\text{e}3$
 $P_{\text{attachment}} = 3.43039\text{e}3 \text{ lbf}$

$S_{d_shell} = (5 / 6) * F_{ty}$
 $S_{d_shell} = (5 / 6) * 21.8750\text{e}3$
 $S_{d_shell} = 18.2292\text{e}3 \text{ psi}$

Uplift Case 6: Design Pressure + Seismic Load

$U = ((F_p * P * (D^2) * 4.08) + ((4 * Mrw) / D)) - (W1 * (1 - (0.4 * A_v)))$
 $U = ((0.40 * 3.0 * (10.0^2) * 4.08) + ((4 * 139.40\text{e}3) / 10.0)) - (8.11832\text{e}3 * (1 - (0.4 * 0.08960)))$
 $U = 48.4221\text{e}3 \text{ lbf}$

$T_b = U / N$
 $T_b = 48.4221\text{e}3 / 4.0$
 $T_b = 12.1055\text{e}3 \text{ lbf}$

$S_d = 0.8 * F_y$
 $S_d = 0.8 * 30.870\text{e}3$
 $S_d = 24.6960\text{e}3 \text{ psi}$

$A-s-r = T_b / S_d$
 $A-s-r = 12.1055\text{e}3 / 24.6960\text{e}3$
 $A-s-r = 0.490182 \text{ in}^2$

$P_{\text{attachment}} = f_{\text{seismic}} * T_b$
 $P_{\text{attachment}} = 3.0 * 12.1055\text{e}3$
 $P_{\text{attachment}} = 36.3166\text{e}3 \text{ lbf}$

$S_{d_shell} = (5 / 6) * F_{ty}$
 $S_{d_shell} = (5 / 6) * 21.8750\text{e}3$

S_d_shell = 18.2292e3 psi

Uplift Case 7: Frangibility Pressure

Not applicable. It is applied if the roof to shell joint is frangible.

Summary of Uplift Cases

Uplift Cases	Total Uplift Load (lbf)	Load per Anchor (lbf)	Anchor Allowable Stress (psi)	Anchor Required Area (in^2)	Anchor Bolt Required Diameter (in)	Attachment Design Load (lbf)	Allowable Shell Stress at Anchor Attachment (psi)
Design Pressure	0.0	0.0	12.8625e3	0.0	0.16250	0.0	14.5833e3
Test Pressure	0.0	0.0	20.0e3	0.0	0.16250	0.0	18.2292e3
Wind Load	8.65811e3	2.16453e3	24.6960e3	0.0876469	0.496559	3.24679e3	18.2292e3
Seismic Load	47.9325e3	11.9831e3	24.6960e3	0.485225	0.948508	35.9494e3	18.2292e3
Design Pressure + Wind	9.14771e3	2.28693e3	17.150e3	0.133349	0.574549	3.43039e3	18.2292e3
Design Pressure + Seismic	48.4221e3	12.1055e3	24.6960e3	0.490182	0.952512	36.3166e3	18.2292e3
<ul style="list-style-type: none">Anchor Bolt Required Diameter = $\text{SQRT}((A-s-r * (4 / \pi))) + (1.3 / n) + (Ca\text{-anchor} * 2)$Governing Uplift Case = Design Pressure + SeismicAnchor Bolt Minimum Required Area = 0.490182 in^2							

Bolt Required Diameter per ANSI B1.1

$d\text{-req} = \text{SQRT}((A * (4 / \pi))) + (1.3 / n) + (Ca * 2)$
 $d\text{-req} = \text{SQRT}((0.490182 * (4 / \pi))) + (1.3 / 8.0) + (0.0 * 2)$
 $d\text{-req} = 0.952512 \text{ in}$

$d \geq d\text{-req} \Rightarrow \text{PASS}$

$A-s = (\pi / 4) * ((d - (1.3 / n))^2)$
 $A-s = (\pi / 4) * ((1.0 - (1.3 / 8.0))^2)$
 $A-s = 0.550883 \text{ in}^2$

$Y\text{-bolt} = A-s * S_y\text{-ambient-anchor}$
 $Y\text{-bolt} = 0.550883 * 36.0e3$
 $Y\text{-bolt} = 19.8318e3 \text{ lbf}$

Anchorage Summary

Required Number of Anchors = 4.0
Actual Number of Anchors = 4.0
Bolt Hole Circle Radius = 5.18229 ft

Required Bolt Diameter = 0.952512 in
Actual Bolt Diameter = 1.0 in
Bolt Thread Pitch = 0.130
Bolt Embedded Depth = 10.0 in
Bolt Length = 30.1875 in

Anchor Chair Design [Back](#)

Anchor Chair Design per AISI T-192 Part V

a = Top Plate Width Along Shell (in)

b = Top Plate Length (in)

bmin = Top Plate Minimum Length (in)

c = Top Plate Thickness (in)

CA = Chair Corrosion Allowance (in)

c_corr = Top Plate Corroded Thickness (in)

D = Tank Nominal Diameter (ft)

d = Anchor Bolt Diameter (in)

e = Anchor Bolt Eccentricity (in)

Earthquakes-Considered = Earthquakes Considered

emin = Minimum Calculated Eccentricity (in)

emin-btm = Minimum Eccentricity Based on Bolt Clearance From Bottom Plates per *API-650 5.12.4* (in)

emin-req = Minimum Required Eccentricity (in)

Et = Bottom Plates Thermal Expansion Coefficient per *API-650 Table P.1b* (in/in.fdeg)

f = Top Plate Outside To Hole Edge Distance (in)

f_min = Distance from Outside of Top Plate to Edge of Hole per *AISI T-192 Part V, Notation* (in)

g = Vertical Plates Distance (in)

g_min = Minimum Distance Between Vertical Plates per *AISI T-192, Part V, Notation* (in)

h = Chair Height (in)

h-eff = Effective Chair Height (in)

hmax = Chair Maximum Height (in)

j = Vertical Plate Thickness (in)

j_corr = Vertical Plate Corroded Thickness (in)

j_min = Vertical Plate Minimum Thickness per *AISI T-192 Part V, Vertical Side Plates* (in)

k = Vertical Plates Average Width (in)

m = Base or Bottom Plate Thickness (in)

Ma-chair = Chair Material

outside-projection = Bottom Outside Projection (in)

R = Nominal Shell Radius (in)

Ssw-chair = Chair Allowable Stress for Seismic or Wind Design per *API-650 5.12.9* (psi)

T = Difference between ambient and design temperature per *API 650 5.12.4* (°F)

t = Shell Thickness (in)

T_ambient = Ambient Temperature (°F)

T_design = Design Temperature (°F)

V = Wind Velocity (mph)

Y-bolt = Anchor Bolt Yield Load (lbf)

a = 10.0 in

b = 10.0 in

c = 0.50 in

CA = 0.0 in
d = 1.0 in
D = 10.0 ft
e = 2.0 in
Earthquakes-Considered = ASCE7-MAPPED-SS-AND-S1
Et = 7.070e-6 in/in.fdeg
f = 6.0 in
g = 4.250 in
h = 14.0 in
j = 0.56250 in
k = 5.47902 in
m = 0.18750 in
Ma-chair = A240-316
outside-projection = 1.0 in
R = 60.0 in
t = 0.18750 in
T_ambient = 70.0 °F
T_design = 375.0 °F
V = 81.90 mph
Y-bolt = 19.8318e3 lbf



Anchor Chair Material Properties

Material = A240-316
Minimum Tensile Strength (Sut-chair) = 75.0e3 psi
As per API-650 S.5.b, Minimum Yield Strength (Sy-chair) = 21.8750e3 psi
As per API-650 S.2b, Allowable Design Stress (Sd-chair) = 19.7250e3 psi
As per API-650 S.2b, Allowable Hydrostatic Test Stress (St-chair) = 27.0e3 psi

Ssw-chair = 1.33 * Sd-chair
Ssw-chair = 1.33 * 19.7250e3
Ssw-chair = 26.2342e3 psi

Size Requirements

c_corr = c - (2 * CA)
c_corr = 0.50 - (2 * 0.0)
c_corr = 0.50 in

j_corr = j - (2 * CA)
j_corr = 0.56250 - (2 * 0.0)
j_corr = 0.56250 in

Chair Minimum Height (hmin) = 12.0 in

h >= hmin ==> PASS

$h_{max} = 3 * a$
 $h_{max} = 3 * 10.0$
 $h_{max} = 30.0 \text{ in}$

$h_{eff} = \text{MIN}(h_{max}, h)$
 $h_{eff} = \text{MIN}(30.0, 14.0)$
 $h_{eff} = 14.0 \text{ in}$

$e_{min} = (0.886 * d) + 0.572$
 $e_{min} = (0.886 * 1.0) + 0.572$
 $e_{min} = 1.4580 \text{ in}$

$T = T_{design} - T_{ambient}$
 $T = 375.0 - 70.0$
 $T = 305.0 \text{ }^{\circ}\text{F}$

$e_{min-btm} = (d / 2) + \text{outside-projection} + 0.125 + (6 * E_t * D * T)$
 $e_{min-btm} = (1.0 / 2) + 1.0 + 0.125 + (6 * 7.070\text{e-}6 * 10.0 * 305.0)$
 $e_{min-btm} = 1.75438 \text{ in}$

$e_{min-req} = \text{MAX}(e_{min}, e_{min-btm})$
 $e_{min-req} = \text{MAX}(1.4580, 1.75438)$
 $e_{min-req} = 1.75438 \text{ in}$

$e \geq e_{min-req} \implies \text{PASS}$

$g_{min} = d + 1$
 $g_{min} = 1.0 + 1$
 $g_{min} = 2.0 \text{ in}$

$g \geq g_{min} \implies \text{PASS}$

$f_{min} = (d / 2) + 0.125$
 $f_{min} = (1.0 / 2) + 0.125$
 $f_{min} = 0.6250 \text{ in}$

$f \geq f_{min} \implies \text{PASS}$

$j_{min} = \text{MAX}(0.5, (0.04 * (h_{eff} - c_{corr})), ((P_{design} / (25000 * k)) + (2 * CA)))$
 $j_{min} = \text{MAX}(0.5, (0.04 * (14.0 - 0.50)), ((19.8318\text{e}3 / (25000 * 5.47902)) + (2 * 0.0)))$
 $j_{min} = 0.540 \text{ in}$

$j \geq j_{min} \implies \text{PASS}$

$b_{min} = e_{min} + d + 0.25$
 $b_{min} = 1.4580 + 1.0 + 0.25$
 $b_{min} = 2.7080 \text{ in}$

$b \geq b_{min} \implies \text{PASS}$

Top Plate Minimum Required Thickness

Uplift Cases	P-chair (lbf)	P-design (lbf)	Sd-chair (psi)	c_min (in)	Status
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Design Pressure	0.0	0.0	19.7250e3	0.0	PASS
Test Pressure	0.0	0.0	27.0e3	0.0	PASS
Wind Load	3.24679e3	3.24679e3	26.2342e3	0.168334	PASS
Seismic Load	35.9494e3	19.8318e3	26.2342e3	0.416030	PASS
Design Pressure + Wind	3.43039e3	3.43039e3	26.2342e3	0.173028	PASS
Design Pressure + Seismic	36.3166e3	19.8318e3	26.2342e3	0.416030	PASS
<ul style="list-style-type: none"> P-chair = Anchor Chair Uplift Load P-design = Anchor Chair Design Load = min(P-chair, Y-bolt) Sd-chair = Anchor Chair Allowable Stress c_min = Top Plate Minimum Required Thickness c_min = $\text{SQRT}(((P\text{-design} / (Sd\text{-chair} * f)) * ((0.375 * g) - (0.22 * d)))) + (2 * CA)$ Governing Uplift Case = Seismic Load Governing Thickness (c_min) = 0.416030 in 					

Top Plate Stress

Uplift Cases	P-chair (lbf)	P-design (lbf)	S_top-plate (psi)	Sd-chair (psi)	Stress Ratio	Status
Design Pressure	0.0	0.0	0.0	19.7250e3	0.0%	PASS
Test Pressure	0.0	0.0	0.0	27.0e3	0.0%	PASS
Wind Load	3.24679e3	3.24679e3	2.97352e3	26.2342e3	11.3345%	PASS
Seismic Load	35.9494e3	19.8318e3	18.1626e3	26.2342e3	69.2325%	PASS
Design Pressure + Wind	3.43039e3	3.43039e3	3.14167e3	26.2342e3	11.9754%	PASS
Design Pressure + Seismic	36.3166e3	19.8318e3	18.1626e3	26.2342e3	69.2325%	PASS
<ul style="list-style-type: none"> P-chair = Anchor Chair Uplift Load P-design = Anchor Chair Design Load = min(P-chair, Y-bolt) S_top-plate = Top Plate Stress $S_{\text{top-plate}} = (P\text{-design} / (f * (c_{\text{corr}}^2))) * ((0.375 * g) - (0.22 * d))$ Sd-chair = Anchor Chair Allowable Stress Governing Uplift Case = Seismic Load Governing Stress (S_top-plate) = 18.1626e3 psi 						

Z = Chair Reduction Factor per *AISI T-192 Part V, Eq 5-4*

Shell Stress at Anchor Attachment

$$Z = 1 / (((0.177 * a * m) / \text{SQRT}((R * t))) * ((m / t)^2) + 1)$$

$$Z = 1 / (((0.177 * 10.0 * 0.18750) / \text{SQRT}((60.0 * 0.18750))) * ((0.18750 / 0.18750)^2) + 1)$$

$$Z = 0.909963$$

Uplift Cases	P-chair (lbf)	P-design (lbf)	S_Shell (psi)	Sd-shell (psi)	Stress Ratio	Status
Design Pressure	0.0	0.0	0.0	14.5833e3	0.0%	PASS
Test Pressure	0.0	0.0	0.0	18.2292e3	0.0%	PASS

Wind Load	3.24679e3	3.24679e3	2.53206e3	18.2292e3	13.8902%	PASS
Seismic Load	35.9494e3	19.8318e3	15.4661e3	18.2292e3	84.8429%	PASS
Design Pressure + Wind	3.43039e3	3.43039e3	2.67525e3	18.2292e3	14.6756%	PASS
Design Pressure + Seismic	36.3166e3	19.8318e3	15.4661e3	18.2292e3	84.8429%	PASS
<ul style="list-style-type: none"> • P-chair = Anchor Chair Uplift Load • P-design = Anchor Chair Design Load = min(P-chair, Y-bolt) • S_Shell = Stress at Attachment • $S_Shell = ((P\text{-design} * e) / (t^2)) * (((1.32 * Z) / (((1.43 * a * (h^2)) / (R * t)) + ((4 * a * (h^2))^{0.333})) + (0.031 / \text{SQRT}((R * t))))$ • Sd-shell = Allowable Stress at Anchor Attachment • Governing Uplift Case = Seismic Load • Governing Stress (S_Shell) = 15.4661e3 psi • Governing Allowable Stress (Sd-Shell) = 18.2292e3 psi 						

Appurtenances Design [Back](#)

Plan View

LABEL	MARK	CUST. MARK	DESCRIPTION	OUTSIDE PROJ (in)	INSIDE PROJ (in)	ORIENT	RADIUS (in)	REMARKS	REF DWG
Agitator-Bridge	AB01		AGITATOR BRIDGE	--	--	0 '	--		
M1	RM01A		24" ROOF MANWAY	10"	1"	270 '	3'-4"		
N1	RN01A		6" ROOF NOZZLE	6"	1"	0 '	3'-9"		
N2	RN01A		6" ROOF NOZZLE	6"	1"	45 '	3'-9"		
N3	RN01A		6" ROOF NOZZLE	6"	1"	90 '	3'-9"		
N4	RN01A		6" ROOF NOZZLE	6"	1"	135 '	3'-9"		

Elevation View

LABEL	MARK	CUST. MARK	DESCRIPTION	OUTSIDE PROJ (in)	INSIDE PROJ (in)	ORIENT	ELEVATION (in)	REMARKS	REF DWG
Agitator-Bridge	AB01		AGITATOR BRIDGE	--	--	0 '	2'-5"		
Anchor-Chair-Bolts	AC01A		ANCHOR CHAIRS	--	--	SEE TABLE	--		
M2	SM01A		24" SHELL MANWAY	10"	1"	325 '	2'-6"	W/ DAVIT	
N5	SN01A		6" SHELL NOZZLE	8"	1"	0 '	1'-0 1/8"		
N6	SN01A		6" SHELL NOZZLE	8"	1"	45 '	1'-0 1/8"		
N7	SN01A		6" SHELL NOZZLE	8"	1"	90 '	1'-0 1/8"		
N8	SN01A		6" SHELL NOZZLE	8"	1"	135 '	1'-0 1/8"		

N9	SN02A		3" SHELL NOZZLE	7"	1"	180 '	9 1/2"	
N10	SN02A		3" SHELL NOZZLE	7"	1"	160 '	9 1/2"	
N11	SN02A		3" SHELL NOZZLE	7"	1"	220 '	9 1/2"	
N12	SN02A		3" SHELL NOZZLE	7"	1"	240 '	9 1/2"	
Name- Plate	NP01A		STD API	--	--	0 '	3'-4"	

Shell Nozzle: N5

Repad Design

NOZZLE Description : 6 in SCH 40S TYPE RFSO
Material: A312-TP316

t_rpr = (Repad Required Thickness)
t_n = (Thickness of Neck)
Sd_n = (Stress of Neck Material)
Sd_s = (Stress of Shell Course Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 1.0104 ft

COURSE PARAMETERS:

t-calc = 0.0646 in
t_cr = 0.0646 in (Course t-calc less C.A)
t_c = 0.1875 in (Course t less C.A.)
t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t_Basis * D
Required Area = 0.0646 * 6.625
Required Area = 0.4279 in²

Available Shell Area = (t_c - t_Basis) * D
Available Shell Area = (0.1875 - 0.0646) * 6.625
Available Shell Area = 0.8143 in²

Available Nozzle Neck Area = 2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * MIN((Sd_n/Sd_s) 1)
Available Nozzle Neck Area = 2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * MIN((19,725/19,725) 1)
Available Nozzle Neck Area = 0.7322 in²

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)
A-rpr = 0.4279 - 0.8143 - 0.7322

A-rpr = 0 in²

Since A-rpr <= 0, t_rpr = 0

No Reinforcement Pad required.

t_shell_PWHT = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate

t_shell_PWHT = 0.18750

t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_shell = Shell Plate Thickness (in)

D = 6.0 in

Group = None

t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N6

Repad Design

NOZZLE Description : 6 in SCH 40S TYPE RFSO

Material: A312-TP316

t_rpr = (Repad Required Thickness)

t_n = (Thickness of Neck)

Sd_n = (Stress of Neck Material)

Sd_s = (Stress of Shell Course Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 1.0104 ft

COURSE PARAMETERS:

t-calc = 0.0646 in

t_cr = 0.0646 in (Course t-calc less C.A)

t_c = 0.1875 in (Course t less C.A.)

t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = $t_{Basis} * D$
Required Area = $0.0646 * 6.625$
Required Area = 0.4279 in²

Available Shell Area = $(t_c - t_{Basis}) * D$
Available Shell Area = $(0.1875 - 0.0646) * 6.625$
Available Shell Area = 0.8143 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.7322 in²

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)
A-rpr = 0.4279 - 0.8143 - 0.7322
A-rpr = 0 in²

Since A-rpr <= 0, $t_{rpr} = 0$

No Reinforcement Pad required.

t_{shell_PWHT} = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_{shell_PWHT} = t-plate
 t_{shell_PWHT} = 0.18750
 t_{shell_PWHT} = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_{shell} = Shell Plate Thickness (in)

D = 6.0 in
Group = None
 t_{shell} = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N7

Repad Design

NOZZLE Description : 6 in SCH 40S TYPE RFSO
Material: A312-TP316

t_{rpr} = (Repad Required Thickness)
 t_n = (Thickness of Neck)

Sd_n = (Stress of Neck Material)
Sd_s = (Stress of Shell Course Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 1.0104 ft

COURSE PARAMETERS:

t_calc = 0.0646 in
t_cr = 0.0646 in (Course t_calc less C.A.)
t_c = 0.1875 in (Course t less C.A.)
t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t_Basis * D
Required Area = 0.0646 * 6.625
Required Area = 0.4279 in²

Available Shell Area = (t_c - t_Basis) * D
Available Shell Area = (0.1875 - 0.0646) * 6.625
Available Shell Area = 0.8143 in²

Available Nozzle Neck Area = 2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * MIN((Sd_n/Sd_s) 1)
Available Nozzle Neck Area = 2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * MIN((19,725/19,725) 1)
Available Nozzle Neck Area = 0.7322 in²

A_rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)
A_rpr = 0.4279 - 0.8143 - 0.7322
A_rpr = 0 in²

Since A_rpr <= 0, t_rpr = 0

No Reinforcement Pad required.

t_shell_PWHT = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316
As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate
t_shell_PWHT = 0.18750
t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)
Group = Shell Material Group
t_shell = Shell Plate Thickness (in)

D = 6.0 in
Group = None
t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for

Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N8

Repad Design

NOZZLE Description : 6 in SCH 40S TYPE RFSO
Material: A312-TP316

t_{rpr} = (Repad Required Thickness)
 t_n = (Thickness of Neck)
 Sd_n = (Stress of Neck Material)
 Sd_s = (Stress of Shell Course Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 1.0104 ft

COURSE PARAMETERS:

t_{calc} = 0.0646 in
 t_{cr} = 0.0646 in (Course t_{calc} less C.A.)
 t_c = 0.1875 in (Course t less C.A.)
 t_{Basis} = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = $t_{Basis} * D$
Required Area = 0.0646 * 6.625
Required Area = 0.4279 in²

Available Shell Area = $(t_c - t_{Basis}) * D$
Available Shell Area = (0.1875 - 0.0646) * 6.625
Available Shell Area = 0.8143 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.7322 in²

A_{rpr} = (Required Area - Available Shell Area - Available Nozzle Neck Area)
 A_{rpr} = 0.4279 - 0.8143 - 0.7322
 A_{rpr} = 0 in²

Since $A_{rpr} \leq 0$, $t_{rpr} = 0$

No Reinforcement Pad required.

t_{shell_PWHT} = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316
As per API-650 S.2b, Allowable Design Stress (Sd_{neck}) = 19.7250e3 psi

t_shell_PWHT = t-plate
t_shell_PWHT = 0.18750
t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)
Group = Shell Material Group
t_shell = Shell Plate Thickness (in)

D = 6.0 in
Group = None
t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N9

Repad Design

NOZZLE Description : 3 in SCH 40S TYPE RFSO
Material: A312-TP316

t_rpr = (Repad Required Thickness)
t_n = (Thickness of Neck)
Sd_n = (Stress of Neck Material)
Sd_s = (Stress of Shell Course Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.7917 ft

COURSE PARAMETERS:

t-calc = 0.0646 in
t_cr = 0.0646 in (Course t-calc less C.A.)
t_c = 0.1875 in (Course t less C.A.)
t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t_Basis * D
Required Area = 0.0646 * 3.5
Required Area = 0.2261 in²

Available Shell Area = (t_c - t_Basis) * D
Available Shell Area = (0.1875 - 0.0646) * 3.5
Available Shell Area = 0.4302 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.216 - 0)) + 0.1875] * (0.216 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.4542 in²

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)

A-rpr = 0.2261 - 0.4302 - 0.4542

A-rpr = 0 in²

Since A-rpr <= 0, t_rpr = 0

No Reinforcement Pad required.

t_shell_PWHT = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate

t_shell_PWHT = 0.18750

t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_shell = Shell Plate Thickness (in)

D = 3.0 in

Group = None

t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N10

Repad Design

NOZZLE Description : 3 in SCH 40S TYPE RFSO

Material: A312-TP316

t_rpr = (Repad Required Thickness)

t_n = (Thickness of Neck)

Sd_n = (Stress of Neck Material)

Sd_s = (Stress of Shell Course Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.7917 ft

COURSE PARAMETERS:

t-calc = 0.0646 in

t_cr = 0.0646 in (Course t-calc less C.A.)

t_c = 0.1875 in (Course t less C.A.)

t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = $t_{\text{Basis}} * D$
Required Area = $0.0646 * 3.5$
Required Area = 0.2261 in²

Available Shell Area = $(t_c - t_{\text{Basis}}) * D$
Available Shell Area = $(0.1875 - 0.0646) * 3.5$
Available Shell Area = 0.4302 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.216 - 0)) + 0.1875] * (0.216 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.4542 in²

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)
A-rpr = $0.2261 - 0.4302 - 0.4542$
A-rpr = 0 in²

Since A-rpr <= 0, $t_{\text{rpr}} = 0$

No Reinforcement Pad required.

$t_{\text{shell_PWHT}}$ = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

$t_{\text{shell_PWHT}} = t_{\text{plate}}$
 $t_{\text{shell_PWHT}} = 0.18750$
 $t_{\text{shell_PWHT}} = 0.18750$ in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_{shell} = Shell Plate Thickness (in)

D = 3.0 in
Group = None
 $t_{\text{shell}} = 0.18750$ in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N11

Repad Design

NOZZLE Description : 3 in SCH 40S TYPE RFSO
Material: A312-TP316

t_rpr = (Repad Required Thickness)
t_n = (Thickness of Neck)
Sd_n = (Stress of Neck Material)
Sd_s = (Stress of Shell Course Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.7917 ft

COURSE PARAMETERS:

t-calc = 0.0646 in
t_cr = 0.0646 in (Course t-calc less C.A.)
t_c = 0.1875 in (Course t less C.A.)
t_Basis = 0.0646 in

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t_Basis * D
Required Area = 0.0646 * 3.5
Required Area = 0.2261 in²

Available Shell Area = (t_c - t_Basis) * D
Available Shell Area = (0.1875 - 0.0646) * 3.5
Available Shell Area = 0.4302 in²

Available Nozzle Neck Area = 2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * MIN((Sd_n/Sd_s) 1)
Available Nozzle Neck Area = 2 * [(4 * (0.216 - 0)) + 0.1875] * (0.216 - 0) * MIN((19,725/19,725) 1)
Available Nozzle Neck Area = 0.4542 in²

A-rpr = (Required Area - Available Shell Area - Available Nozzle Neck Area)
A-rpr = 0.2261 - 0.4302 - 0.4542
A-rpr = 0 in²

Since A-rpr <= 0, t_rpr = 0

No Reinforcement Pad required.

t_shell_PWHT = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316
As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate
t_shell_PWHT = 0.18750
t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)
Group = Shell Material Group
t_shell = Shell Plate Thickness (in)

D = 3.0 in
Group = None

$t_{\text{shell}} = 0.18750 \text{ in}$

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Nozzle: N12

Repad Design

NOZZLE Description : 3 in SCH 40S TYPE RFSO

Material: A312-TP316

t_{rpr} = (Repad Required Thickness)

t_{n} = (Thickness of Neck)

Sd_{n} = (Stress of Neck Material)

Sd_{s} = (Stress of Shell Course Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 0.7917 ft

COURSE PARAMETERS:

$t_{\text{calc}} = 0.0646 \text{ in}$

$t_{\text{cr}} = 0.0646 \text{ in}$ (Course t_{calc} less C.A.)

$t_{\text{c}} = 0.1875 \text{ in}$ (Course t less C.A.)

$t_{\text{Basis}} = 0.0646 \text{ in}$

(SHELL NOZZLE REF. API-650 S.3.3.1, AND FOOTNOTE A OF TABLE 5-7)

Required Area = $t_{\text{Basis}} * D$

Required Area = $0.0646 * 3.5$

Required Area = 0.2261 in^2

Available Shell Area = $(t_{\text{c}} - t_{\text{Basis}}) * D$

Available Shell Area = $(0.1875 - 0.0646) * 3.5$

Available Shell Area = 0.4302 in^2

Available Nozzle Neck Area = $2 * [(4 * (t_{\text{n}} - CA)) + t_{\text{c}}] * (t_{\text{n}} - CA) * \text{MIN}((Sd_{\text{n}}/Sd_{\text{s}}) 1)$

Available Nozzle Neck Area = $2 * [(4 * (0.216 - 0)) + 0.1875] * (0.216 - 0) * \text{MIN}((19,725/19,725) 1)$

Available Nozzle Neck Area = 0.4542 in^2

$A_{\text{rpr}} = (\text{Required Area} - \text{Available Shell Area} - \text{Available Nozzle Neck Area})$

$A_{\text{rpr}} = 0.2261 - 0.4302 - 0.4542$

$A_{\text{rpr}} = 0 \text{ in}^2$

Since $A_{\text{rpr}} \leq 0$, $t_{\text{rpr}} = 0$

No Reinforcement Pad required.

$t_{\text{shell_PWHT}}$ = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Nozzle Neck Material Properties

Material = A312-TP316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate

t_shell_PWHT = 0.18750

t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_shell = Shell Plate Thickness (in)

D = 3.0 in

Group = None

t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Shell Manway: M2

Repad Design

MANWAY Description : 24 in Neck Thickness 0.25

Material: A240-316

t_rpr = (Repad Required Thickness)

t_n = (Thickness of Neck)

Sd_n = (Stress of Neck Material)

Sd_s = (Stress of Shell Course Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON SHELL 1 : Elevation = 2.5 ft

COURSE PARAMETERS:

t-calc = 0.0646 in

t_cr = 0.0646 in (Course t-calc less C.A)

t_c = 0.1875 in (Course t less C.A.)

t_Basis = 0.0646 in

(SHELL MANWAY REF. API-650 TABLE 5-6, AND FOOTNOTE A OF TABLE 5-7)

Required Area = t_Basis * D

Required Area = 0.0646 * 24

Required Area = 1.5501 in²

Available Shell Area = (t_c - t_Basis) * D

Available Shell Area = (0.1875 - 0.0646) * 24

Available Shell Area = 2.9499 in²

Available Manway Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - CA) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Manway Neck Area = $2 * [(4 * (0.25 - 0)) + 0.1875] * (0.25 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Manway Neck Area = 0.5938 in²

A-rpr = (Required Area - Available Shell Area - Available Manway Neck Area)
A-rpr = 1.5501 - 2.9499 - 0.5938
A-rpr = 0 in²

Since A_rpr <= 0, t_rpr = 0

No Reinforcement Pad required.

t_shell_PWHT = Thickness of the shell plate, insert plate, or thickened insert plate for PWHT (in)

Manway Neck Material Properties

Material = A240-316

As per API-650 S.2b, Allowable Design Stress (Sd-neck) = 19.7250e3 psi

t_shell_PWHT = t-plate
t_shell_PWHT = 0.18750
t_shell_PWHT = 0.18750 in

Thermal Stress Relief (PWHT) Requirements

D = Nozzle Nominal Diameter (NPS) (in)

Group = Shell Material Group

t_shell = Shell Plate Thickness (in)

D = 24.0 in
Group = None
t_shell = 0.18750 in

Shell material group (None) is not a group specified by API 650, 13th Ed, Section 5.7.4. Requirement for Thermal Stress Relief (PWHT) is unknown.

Cover Plate and Bolting Flange Design

CA-cover = Cover Plate and Bolting Flange Corrosion Allowance (in)
Db = Bolt Circle Diameter (in)
H = Design Liquid Level (ft)
M = Cover Plate Thickness Multiplication Factor per *API-650 S.3.3.3*
M = Bolting Flange Thickness Multiplication Factor per *API-650 S.3.3.3*
Ma-cover = Cover Plate Material
Ma-flange = Bolting Flange Material
Sd = Allowable Stress per *API-650 5.7.5.6* (psi)
SG = Product Specific Gravity
tc = Cover Plate Thickness (in)
tc-design = Cover Plate Required Thickness per *API-650 5.7.5.6* (in)
tc-req = Cover Plate Minimum Required Thickness (in)
tf = Bolting Flange Thickness (in)
tf-design = Cover Plate Required Thickness per *API-650 5.7.5.6* (in)
tf-req = Bolting Flange Minimum Required Thickness (in)
t-neck = Neck Thickness (in)

CA-cover = 0.0 in
Db = 30.250 in

H = 30.0 ft
Ma-cover = A240-316
Ma-flange = A240-316
SG = 1.10
tc = 0.6250 in
tf = 0.50 in
t-neck = 0.250 in

Water Density (Y) = 0.4330 psi/ft
As per API-650 5.7.5.6, Coefficient For Circular Plate (C) = 0.30

Cover Plate Material Properties and Required Thickness

Material = A240-316
As per API-650 S.5.b, Minimum Yield Strength at Ambient Temperature (Sy-ambient-cover) = 30.0e3 psi
As per API-650 S.5.b, Minimum Yield Strength (Sy-cover) = 21.8750e3 psi
Thickness for MDMT-permissible-cover (per API-650 Figure 4.3) = 0.156250 in

Sd = MIN(Sy-ambient-cover , 30000) / 2 = 15.0e3 psi

M = MAX(SQRT((Sy-ambient-cover / Sy-cover)) , SQRT((30000 / Sy-cover)) , 1) = 1.17108

As per API-650 5.7.5.6, Cover Plate Erection Thickness (tc-erec) = 0.31250 in

tc-design = ((Db * SQRT(((C * Y * H * MAX(SG , 1)) / Sd))) + CA-cover) * M
tc-design = ((30.250 * SQRT(((0.30 * 0.4330 * 30.0 * MAX(1.10 , 1)) / 15.0e3))) + 0.0) * 1.17108
tc-design = 0.598864 in

tc-req = MAX(tc-erec , tc-design)
tc-req = MAX(0.31250 , 0.598864)
tc-req = 0.598864 in

t-cover >= tc-req ==> PASS

Bolting Flange Material Properties and Required Thickness

Material = A240-316
As per API-650 S.5.b, Minimum Yield Strength at Ambient Temperature (Sy-ambient-flange) = 30.0e3 psi
As per API-650 S.5.b, Minimum Yield Strength (Sy-flange) = 21.8750e3 psi
Thickness for MDMT-permissible-flange (per API-650 Figure 4.3) = 0.250 in

M = MAX(SQRT((Sy-ambient-flange / Sy-flange)) , SQRT((30000 / Sy-flange)) , 1) = 1.17108

As per API-650 5.7.5.6, Bolting Flange Erection Thickness (tf-erec) = 0.250 in

tf-design = tc-design - 0.125
tf-design = 0.598864 - 0.125
tf-design = 0.473864 in

tf-req = MAX(tf-erec , tf-design)
tf-req = MAX(0.250 , 0.473864)
tf-req = 0.473864 in

t-flange >= tf-req ==> PASS

Roof Nozzle: N1

Repad Design

(Per API-650 and other references below)

NOZZLE Description : 6 in SCH 40 TYPE RFSO

Material: A312-TP316

t_{rpr} = (Repad Required Thickness)

t_n = (Thickness of Neck)

Sd_n = (Stress of Neck Material)

Sd_s = (Stress of Roof Material)

CA = (Corrosion Allowance of Neck)

MOUNTED ON ROOF: Elevation = 30.2558 ft

ROOF PARAMETERS:

t_{calc} = 0.1875 in

t_{cr} = 0.1875 in (Roof t_{act} less C.A)

t_c = 0.1875 in

t_{Basis} = 0.1875 in

(FOR ROOF NOZZLES, REF. API-650 FIG 5-19, TABLE 5-14 AND FOOTNOTE A OF TABLE 5-14, or API-650 FIG 5-20, TABLE 5-15 AND FOOTNOTE A OF TABLE 5-15)

Required Area = $t_{Basis} * D$

Required Area = $0.1875 * 6.625$

Required Area = 1.2422 in²

Available Roof Area = $(t_c - t_{Basis}) * D$

Available Roof Area = $(0.1875 - 0.1875) * 6.625$

Available Roof Area = 0 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - ca) * \text{MIN}((Sd_n/Sd_s) 1)$

Available Nozzle Neck Area = $2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * \text{MIN}((19,725/19,725) 1)$

Available Nozzle Neck Area = 0.7322 in²

A_{rpr} = (Required Area - Available Roof Area - Available Nozzle Neck Area)

A_{rpr} = $1.2422 - 0 - 0.7322$

A_{rpr} = 0.51 in²

As per API-650 J.3.6.3, reinforcement pad is not required since roof loads do not exceed 25 psf.

No Reinforcement Pad required.

Roof Nozzle: N2

Repad Design

(Per API-650 and other references below)

NOZZLE Description : 6 in SCH 40 TYPE RFSO
Material: A312-TP316

t_{rpr} = (Repad Required Thickness)
 t_n = (Thickness of Neck)
 Sd_n = (Stress of Neck Material)
 Sd_s = (Stress of Roof Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON ROOF: Elevation = 30.2558 ft

ROOF PARAMETERS:

t_{calc} = 0.1875 in
 t_{cr} = 0.1875 in (Roof t_{act} less C.A)
 t_c = 0.1875 in
 t_{Basis} = 0.1875 in

(FOR ROOF NOZZLES, REF. API-650 FIG 5-19, TABLE 5-14 AND FOOTNOTE A OF TABLE 5-14, or API-650 FIG 5-20, TABLE 5-15 AND FOOTNOTE A OF TABLE 5-15)

Required Area = $t_{Basis} * D$
Required Area = $0.1875 * 6.625$
Required Area = 1.2422 in²

Available Roof Area = $(t_c - t_{Basis}) * D$
Available Roof Area = $(0.1875 - 0.1875) * 6.625$
Available Roof Area = 0 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - ca) * MIN((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * MIN((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.7322 in²

A_{rpr} = (Required Area - Available Roof Area - Available Nozzle Neck Area)
 A_{rpr} = $1.2422 - 0 - 0.7322$
 A_{rpr} = 0.51 in²

As per API-650 J.3.6.3, reinforcement pad is not required since roof loads do not exceed 25 psf.

No Reinforcement Pad required.

Roof Nozzle: N3

Repad Design

(Per API-650 and other references below)

NOZZLE Description : 6 in SCH 40 TYPE RFSO
Material: A312-TP316

t_{rpr} = (Repad Required Thickness)
 t_n = (Thickness of Neck)
 Sd_n = (Stress of Neck Material)
 Sd_s = (Stress of Roof Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON ROOF: Elevation = 30.2558 ft

ROOF PARAMETERS:

t_{calc} = 0.1875 in
 t_{cr} = 0.1875 in (Roof t-act less C.A)
 t_c = 0.1875 in
 t_{Basis} = 0.1875 in

(FOR ROOF NOZZLES, REF. API-650 FIG 5-19, TABLE 5-14 AND FOOTNOTE A OF TABLE 5-14, or API-650 FIG 5-20, TABLE 5-15 AND FOOTNOTE A OF TABLE 5-15)

Required Area = $t_{Basis} * D$
Required Area = $0.1875 * 6.625$
Required Area = 1.2422 in²

Available Roof Area = $(t_c - t_{Basis}) * D$
Available Roof Area = $(0.1875 - 0.1875) * 6.625$
Available Roof Area = 0 in²

Available Nozzle Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - ca) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Nozzle Neck Area = $2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Nozzle Neck Area = 0.7322 in²

A_{rpr} = (Required Area - Available Roof Area - Available Nozzle Neck Area)
 A_{rpr} = $1.2422 - 0 - 0.7322$
 A_{rpr} = 0.51 in²

As per API-650 J.3.6.3, reinforcement pad is not required since roof loads do not exceed 25 psf.

No Reinforcement Pad required.

Roof Nozzle: N4

Repad Design

(Per API-650 and other references below)

NOZZLE Description : 6 in SCH 40 TYPE RFSO
Material: A312-TP316

t_rpr = (Repad Required Thickness)
t_n = (Thickness of Neck)
Sd_n = (Stress of Neck Material)
Sd_s = (Stress of Roof Material)
CA = (Corrosion Allowance of Neck)

MOUNTED ON ROOF: Elevation = 30.2558 ft

ROOF PARAMETERS:

t_calc = 0.1875 in
t_cr = 0.1875 in (Roof t-act less C.A)
t_c = 0.1875 in
t_Basis = 0.1875 in

(FOR ROOF NOZZLES, REF. API-650 FIG 5-19, TABLE 5-14 AND FOOTNOTE A OF TABLE 5-14, or
API-650 FIG 5-20, TABLE 5-15 AND FOOTNOTE A OF TABLE 5-15)

Required Area = t_Basis * D
Required Area = 0.1875 * 6.625
Required Area = 1.2422 in²

Available Roof Area = (t_c - t_Basis) * D
Available Roof Area = (0.1875 - 0.1875) * 6.625
Available Roof Area = 0 in²

Available Nozzle Neck Area = 2 * [(4 * (t_n - CA)) + t_c] * (t_n - ca) * MIN((Sd_n/Sd_s) 1)
Available Nozzle Neck Area = 2 * [(4 * (0.28 - 0)) + 0.1875] * (0.28 - 0) * MIN((19,725/19,725) 1)
Available Nozzle Neck Area = 0.7322 in²

A_rpr = (Required Area - Available Roof Area - Available Nozzle Neck Area)
A_rpr = 1.2422 - 0 - 0.7322
A_rpr = 0.51 in²

As per API-650 J.3.6.3, reinforcement pad is not required since roof loads do not exceed 25 psf.

No Reinforcement Pad required.

Roof Manway: M1

Repad Design

(Per API-650 Section 5.8.4 and other references below)
MANWAY Description : 24 in Neck Thickness 0.25
Material: A240-316

t_{rpr} = (Repad Required Thickness)
MOUNTED ON ROOF: Elevation = 30.3252 ft

ROOF PARAMETERS:

t_{calc} = 0.1875 in
 t_{cr} = 0.1875 in (Roof t_{act} less C.A)
 t_c = 0.1875 in
 t_{Basis} = 0.1875 in

(FOR ROOF MANWAY, REF. API-650 FIG 5-16, TABLE 5-13)

Required Area = $t_{Basis} * D$
Required Area = $0.1875 * 24$
Required Area = 4.5 in²

Available Roof Area = $(t_c - t_{Basis}) * D$
Available Roof Area = $(0.1875 - 0.1875) * 24$
Available Roof Area = 0 in²

Available Manway Neck Area = $2 * [(4 * (t_n - CA)) + t_c] * (t_n - ca) * \text{MIN}((Sd_n/Sd_s) 1)$
Available Manway Neck Area = $2 * [(4 * (0.25 - 0)) + 0.1875] * (0.25 - 0) * \text{MIN}((19,725/19,725) 1)$
Available Manway Neck Area = 0.5938 in²

A_{rpr} = (Required Area - Available Roof Area - Available Manway Neck Area)
 A_{rpr} = 4.5 - 0 - 0.5938
 A_{rpr} = 3.9063 in²

As per API-650 J.3.6.3, since roof loads does not exceed 25 psf, t_{rpr} = 0

No Reinforcement Pad required.

Normal and Emergency Venting (API-2000 7th Edition) [Back](#)

Normal Venting

insulation_type = Insulation type

latitude = Latitude zone

R_i = Reduction factor for insulation per *API-2000 Sections 3.3.2.3.2 and 3.3.2.3.3*

T = Product storage temperature (°F)

vapor-pressure-type = Vapor pressure type

V_i = Total required in-breathing volumetric flow rate (ft³/hr)

V_ip = Required in-breathing flow rate due to liquid movement per *API-2000 Section 3.3.2.2, Eq 6* (ft³/hr)

V_IT = Required in-breathing flow rate due to thermal effects per *API-2000 Section 3.3.2.3.3, Eq 10* (ft³/hr)

V_o = Total Required out-breathing volumetric flow rate (ft³/hr)

V_op = Required out-breathing flow rate due to liquid movement per *API-2000 Section 3.3.2.2.1-a, Eq 2* (ft³/hr)

V_OT = Required out-breathing flow rate due to thermal effects per *API-2000 Section 3.3.2.3.2, Eq 8* (ft³/hr)

V_pe = Maximum emptying rate (gpm)

V_pf = Maximum filling rate (gpm)

V_tk = Tank capacity (gal)

insulation_type = NO-INSULATION

latitude = BETWEEN-42-AND-58

T = 0.0 °F

vapor-pressure-type = HEXANE-OR-SIMILAR

V_pe = 100.0 gpm

V_pf = 100.0 gpm

V_tk = 17.7952e3 gal

R_i = 1

R_i = 1

R_i = 1.0

In-breathing

V_ip = 8.02 * V_pe

V_ip = 8.02 * 100.0

V_ip = 802.0 ft³/hr

V_ip = 802 ft³/hr of air

As per API-2000 Section 3.3.2.3.3, Table 2, Vapor Pressure Factor (C) = 3.0

V_IT = 3.08 * C * (V_tk^{0.7}) * R_i

V_IT = 3.08 * 3.0 * (2.37887e3^{0.7}) * 1.0

V_IT = 2.13369e3 ft³/hr

V_IT = 2,133.688 ft³/hr of air

V_i = V_ip + V_IT

V_i = 802.0 + 2.13369e3

V_i = 2.93569e3 ft³/hr

$$V_i = 2,935.688 \text{ ft}^3/\text{hr of air}$$

Out-breathing

$$V_{op} = 8.02 * V_{pf}$$

$$V_{op} = 8.02 * 100.0$$

$$V_{op} = 802.0 \text{ ft}^3/\text{hr}$$

$$V_{op} = 802 \text{ ft}^3/\text{hr of air}$$

As per API-2000 Section 3.3.2.3.2, Table 1, Latitude Factor (Y) = 0.250

$$V_{OT} = 1.51 * Y * (V_{tk}^{0.9}) * R_i$$

$$V_{OT} = 1.51 * 0.250 * (2.37887e3^{0.9}) * 1.0$$

$$V_{OT} = 412.715 \text{ ft}^3/\text{hr}$$

$$V_{OT} = 412.7146 \text{ ft}^3/\text{hr of air}$$

$$V_o = V_{op} + V_{OT}$$

$$V_o = 802.0 + 412.715$$

$$V_o = 1.21471e3 \text{ ft}^3/\text{hr}$$

$$V_o = 1,214.7146 \text{ ft}^3/\text{hr of air}$$

Emergency Venting

ATWS = Wetted surface area (ft²)

D = Tank diameter (ft)

F = Environmental factor per *API-2000 Section 3.3.3.3.4*

H = Tank height (ft)

insulation_type = Insulation type

P_g = Design pressure (psi)

q = Required emergency venting capacity per *API-2000 Section 3.3.3.3.4, Table 6* (ft³/hr)

vapor-pressure-type = Vapor pressure type

$$D = 10.0 \text{ ft}$$

$$H = 30.0 \text{ ft}$$

$$\text{insulation_type} = \text{NO-INSULATION}$$

$$P_g = 0.108379 \text{ psi}$$

$$\text{vapor-pressure-type} = \text{HEXANE-OR-SIMILAR}$$

As per API-2000 Table 9, Environmental factor for insulation (F_ins) = 1.0

As per API-2000 Table 9, Environmental factor for drainage (F_drain) = 1.0

$$F = \text{MIN}(F_{ins}, F_{drain})$$

$$F = \text{MIN}(1.0, 1.0)$$

$$F = 1.0$$

$$\text{ATWS} = \pi * D * \text{MIN}(H, 30)$$

$$\text{ATWS} = \pi * 10.0 * \text{MIN}(30.0, 30)$$

$$\text{ATWS} = 942.478 \text{ ft}^2$$

$$q = 506168.1168 * F$$

$$q = 506168.1168 * 1.0$$

$$q = 506.168e3 \text{ ft}^3/\text{hr}$$

Capacities and Weights [Back](#)

Capacity to Top of Shell (to Tank Height) : 17,625 gal
 Capacity to Design Liquid Level : 17,625 gal
 Capacity to Maximum Liquid Level : 17,625 gal
 Working Capacity (to Normal Working Level) : 0 gal
 Net working Capacity (Working Capacity - Min Capacity) : 0 gal
 Minimum Capacity (to Min Liq Level) : 587 gal

Component	New Condition (lbf)	Corroded (lbf)
SHELL	7,379	7,379
ROOF	634	634
RAFTERS	0	0
GIRDERS	0	0
FRAMING	0	0
COLUMNS	0	0
TRUSS	0	0
STRUCTURE COMPONENTS	0	0
BOTTOM	626	626
STAIRWAYS	0	0
ACCESS	0	0
STIFFENERS	101	101
WIND GIRDERS	0	0
AGITATOR BRIDGE	1,000	1,000
ANCHOR CHAIRS	95	95
SHELL APPURTENANCES	498	498
ROOF APPURTENANCES	260	260
BOTTOM APPURTENANCES	0	0
INSULATION	0	0
FLOATING ROOF	0	0
TOTAL	10,594.4152	10,594.4152

Weight of Tank, Empty : 10,594.4152 lbf
 Weight of Tank, Full of Product (Design SG = 1.1) : 172,396.4152 lbf
 Weight of Tank, Full of Water : 157,686.6893 lbf
 Net Working Weight, Full of Product (Design SG = 1.1) : 167,002.5333 lbf
 Net Working Weight Full of Water : 152,783.6135 lbf

Foundation Area Req'd : 81.6794 ft²
 Foundation Loading, Empty : 129.7072 lbf/ft²
 Foundation Loading, Full of Product Design : 2,110.6461 lbf/ft²
 Foundation Loading, Full of Water : 1,930.5552 lbf/ft²

SURFACE AREAS

Roof : 81.6893 ft²
 Shell : 942.4777 ft²

Bottom : 81.6794 ft2

Internal Pressure Moment : 6,128.6806 lbf-ft

Wind Moment : 44,927.5415 lbf-ft

Seismic Moment (Ringwall) : 139,399.6676 lbf-ft

Seismic Moment (Slab) : 153,154.5112 lbf-ft

MISCELLANEOUS ATTACHED ROOF ITEMS

MISCELLANEOUS ATTACHED SHELL ITEMS

Reactions on Foundation [Back](#)

A_{rss} = Area of Tank Roof Supported by the Tank Shell (ft²)
A_v = Vertical Earthquake Acceleration Coefficient (g)
D = Tank Nominal Diameter (ft)
F_{wind} = Wind Horizontal Force (lbf)
gamma_b = Bottom Plate Density (lbf/in³)
gamma_w = Water Density (lb/ft³)
H_s = Tank Height (Shell Only) (ft)
L_{dead_bottom} = Dead Load on Bottom per *API 650, Section 5.13, Table 5.21* (psf)
L_{dead_shell} = Dead Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
L_{hydrostatic_bottom} = Hydrostatic Load on Bottom per *API 650, Section 5.13, Table 5.21* (psf)
L_{internal-pressure_bottom} = Internal Pressure Load on Bottom per *API 650, Section 5.13, Table 5.21* (psf)
L_{max} = Maximum Liquid Level (ft)
L_{minimum-roof-live-load_shell} = Minimum Roof Live Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
L_{pressure-test_bottom} = Pressure Test Load on Bottom per *API 650, Section 5.13, Table 5.21* (psf)
L_r = Minimum Roof Live Load (psf)
L_{seismic_bottom} = Seismic Load on Bottom per *API 650, Section 5.13, Table 5.21* (psf)
L_{seismic_shell} = Seismic Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
L_{snow_shell} = Snow Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
L_{stored-liquid_bottom} = Stored Liquid Load on Shell per *API 650, Section 5.13, Table 5.21* (psf)
L_{vacuum_shell} = Vacuum Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
L_{wind_shell} = Wind Load on Shell per *API 650, Section 5.13, Table 5.21* (lbf/ft)
M_{rw} = Ringwall Foundation Seismic Overturning Moment (ft.lbf)
M_s = Slab Foundation Seismic Overturning Moment (ft.lbf)
P = Design Internal Pressure (psi)
P_t = Test pressure (psi)
P_v = Design External Pressure (psi)
P_{WS} = Shell Wind Pressure (psi)
S = Design Snow Load (psf)
SG = Product Specific Gravity
t_b = Bottom Plate Thickness (in)
W_{r-attachments} = Roof Attachments Weight (lb)
W_{r-ins} = Roof Insulation Weight (lb)
W_{r-pl} = Roof Plates Nominal Weight (lb)
W_{s-attachments} = Shell Attachments Weight (lb)
W_{s-framing} = Shell Framing Weight (lb)
W_{s-ins} = Shell Insulation Weight (lb)
W_{s-pl} = Shell Plates Nominal Weight (lb)
W_{wind} = Wind Overturning Moment (ft.lbf)

A_{rss} = 81.6894 ft²
A_v = 0.08960 g
D = 10.0 ft
F_{wind} = 2.60732e3 lbf
gamma_b = 0.290 lbf/in³ [501.120 lbf/ft³]
gamma_w = 62.4279 lb/ft³
H_s = 30.0 ft
L_{max} = 30.0 ft
L_r = 20.0 psf
M_{rw} = 139.40e3 ft.lbf
M_s = 153.155e3 ft.lbf

P = 0.108379 psi
 Pt = 0.135474 psi
 Pv = 0.0361263 psi
 P_WS = 0.0601666 psi
 S = 0.0 psf
 SG = 1.10
 tb = 0.18750 in [0.0156250 ft]
 Wr-attachments = 260.865 lb
 Wr-ins = 0.0 lb
 Wr-pl = 639.628 lb
 Ws-attachments = 593.550 lb
 Ws-framing = 103.391 lb
 Ws-ins = 0.0 lb
 Ws-pl = 7.37830e3 lb
 W-wind = 44.9275e3 ft.lbf

W_rss = Wr-pl + Wr-ins + Wr-attachments
 W_rss = 639.628 + 0.0 + 260.865
 W_rss = 900.493 lbf

Ws = Ws-pl + Ws-ins + Ws-attachments + Ws-framing
 Ws = 7.37830e3 + 0.0 + 593.550 + 103.391
 Ws = 8.07524e3 lbf

Unfactored (Working Stress) Downward Reactions on Foundations

Load Case	Location	Equation	Value	Unit
Dead Load	Shell	$(W_s + W_{rss}) / (\pi * D)$	285.706	lbf/ft
Dead Load	Bottom	$t_b * \gamma_b$	7.830	psf
Internal Pressure	Bottom	P	15.6066	psf
Vacuum	Shell	$(P_v * A_{rss}) / (\pi * D)$	13.5270	lbf/ft
Hydrostatic Test	bottom	$L_{max} * \gamma_w$	1.87284e3	psf
Minimum Roof Live Load	Shell	$(L_r * A_{rss}) / (\pi * D)$	52.0051	lbf/ft
Seismic	Shell	$((4 * (M_{rw} / D)) + (0.4 * (W_s + W_{rss}) * A_v)) / (\pi * D)$	1.78513e3	lbf/ft
Seismic	Bottom	$(32 * M_s) / (\pi * (D^3))$	1.56002e3	psf
Snow	Shell	$(S * A_{rss}) / (\pi * D)$	0.0	lbf/ft
Stored Liquid	Bottom	$SG * L_{max} * \gamma_w$	2.06012e3	psf
Pressure Test	Bottom	Pt	19.5082	psf
Wind	Shell	$(2 * (H_s^2) * P_{WS}) / (\pi * D)$	496.410	lbf/ft
<ul style="list-style-type: none"> Seismic bottom reaction varies linearly from $32 * M_s / (\pi * D^3)$ at the tank shell to zero at the center of the tank API 650, Section 5.13, Table 5.21 				

L_dead_shell = $(W_s + W_{rss}) / (\pi * D)$
 L_dead_shell = $(8.07524e3 + 900.493) / (\pi * 10.0)$
 L_dead_shell = 285.706 lbf/ft

$L_{dead_bottom} = t_b * \gamma_b$
 $L_{dead_bottom} = 0.0156250 * 501.120$
 $L_{dead_bottom} = 7.830 \text{ psf}$

$L_{internal-pressure_bottom} = P$
 $L_{internal-pressure_bottom} = 15.6066$
 $L_{internal-pressure_bottom} = 15.6066 \text{ psf}$

$L_{vacuum_shell} = (P_v * A_{rss}) / (\pi * D)$
 $L_{vacuum_shell} = (5.20219 * 81.6894) / (\pi * 10.0)$
 $L_{vacuum_shell} = 13.5270 \text{ lbf/ft}$

$L_{hydrostatic_bottom} = L_{max} * \gamma_w$
 $L_{hydrostatic_bottom} = 30.0 * 62.4279$
 $L_{hydrostatic_bottom} = 1.87284e3 \text{ psf}$

$L_{minimum-roof-live-load_shell} = (L_r * A_{rss}) / (\pi * D)$
 $L_{minimum-roof-live-load_shell} = (20.0 * 81.6894) / (\pi * 10.0)$
 $L_{minimum-roof-live-load_shell} = 52.0051 \text{ lbf/ft}$

$L_{seismic_shell} = ((4 * (M_{rw} / D)) + (0.4 * (W_s + W_{rss}) * A_v)) / (\pi * D)$
 $L_{seismic_shell} = ((4 * (139.40e3 / 10.0)) + (0.4 * (8.07524e3 + 900.493) * 0.08960)) / (\pi * 10.0)$
 $L_{seismic_shell} = 1.78513e3 \text{ lbf/ft}$

$L_{seismic_bottom} = (32 * M_s) / (\pi * (D^3))$
 $L_{seismic_bottom} = (32 * 153.155e3) / (\pi * (10.0^3))$
 $L_{seismic_bottom} = 1.56002e3 \text{ psf}$

$L_{snow_shell} = (S * A_{rss}) / (\pi * D)$
 $L_{snow_shell} = (0.0 * 81.6894) / (\pi * 10.0)$
 $L_{snow_shell} = 0.0 \text{ lbf/ft}$

$L_{stored-liquid_bottom} = SG * L_{max} * \gamma_w$
 $L_{stored-liquid_bottom} = 1.10 * 30.0 * 62.4279$
 $L_{stored-liquid_bottom} = 2.06012e3 \text{ psf}$

$L_{pressure-test_bottom} = P_t$
 $L_{pressure-test_bottom} = 19.5082$
 $L_{pressure-test_bottom} = 19.5082 \text{ psf}$

$L_{wind_shell} = (2 * (H_s^2) * P_{WS}) / (\pi * D)$
 $L_{wind_shell} = (2 * (30.0^2) * 8.6640) / (\pi * 10.0)$
 $L_{wind_shell} = 496.410 \text{ lbf/ft}$

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