

$q \in \mathbb{R}$, position of mass

M , mass in kg

u , applied force

$q=0$ at spring equilibrium pos.

e.g. 2.1.1 (mass - spring - damper)

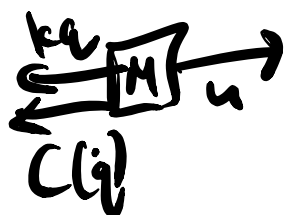
Newton's 2nd Law

$$M\ddot{q} = \sum F$$

spring = kx (linear)

damp: possibly nonlinear, models friction

$$C(\dot{q})$$

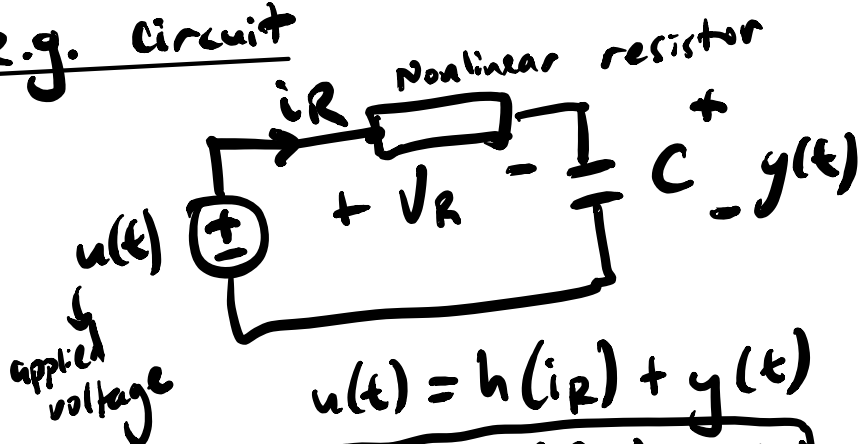


$$M\ddot{q} = u - kq - C(\dot{q})$$

$$M\ddot{q} + kq + C(\dot{q}) = u \rightarrow \text{2nd order ODE}$$

If the damper is linear (i.e. $C(\dot{q}) = b\dot{q}$, b const.), the ODE (and system) becomes linear. \triangle

e.g. circuit



applied voltage

KVL

$$u(t) = h(i_R) + y(t)$$

$$u(t) = h(C\dot{y}) + y(t)$$

$$V_R = h(i_R)$$

$$\hookrightarrow h: \mathbb{R} \rightarrow \mathbb{R}$$

If resistor was linear, 1st order ODE would be linear

$$(h(i) = Ri)$$

Compare w/ e.g. 2.3.4

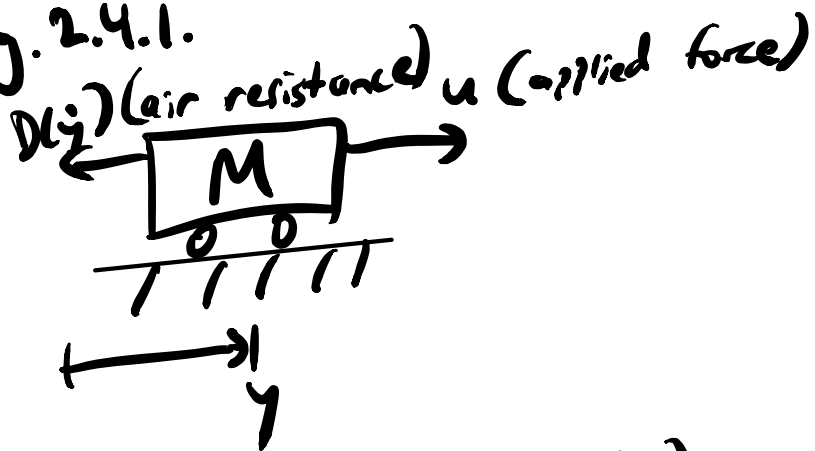
Comments:

- See 2.3 in notes for examples
- expectation is that we can model very simple mechanical and electrical systems. (1Q on MT)

2.4 State Space Models

State-space models are a way of expressing mathematical models in a standard form.

e.g. 2.4.1.



Newton: $M\ddot{y} = u - D(\dot{y})$

Put this model into standard form by defining two so-called state variables:

State vars $\begin{cases} x_1(t) := y(t) & (\text{position}) \\ x_2(t) := \dot{y}(t) & (\text{velocity}) \end{cases}$

We can write this system as:

$$\left. \begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= \frac{1}{M} u - \frac{D(x_2)}{M} \\ y &= x_1 \end{aligned} \right\} \begin{array}{l} 2 \text{ ODEs} = \text{state equation} \\ \text{algebraic part} = \text{output equation} \end{array}$$

These equations have the form $\dot{x} = f(x, u)$
 $y = h(x)$
 non-linear state space model

where $x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \in \mathbb{R}^2$ $f(x, u) = \begin{bmatrix} x_2 \\ \frac{u}{M} - \frac{D(x_2)}{M} \end{bmatrix}$. $h(x) = x_1$

The function f is linear if D is, h is linear.
 As a special case, suppose force due to air resistance were linear.

i.e. $D(x_2) = d x_2$, d constant.

Then f becomes linear and can be written

$$f(x, u) = \underbrace{\begin{bmatrix} 0 & 1 \\ 0 & -\frac{d}{M} \end{bmatrix}}_A \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \underbrace{\begin{bmatrix} 0 \\ \frac{1}{M} \end{bmatrix}}_B u$$

and we get the linear special case

$$\boxed{\begin{aligned} \dot{x} &= Ax + Bu \\ y &= Cx \end{aligned}} \quad C = \begin{bmatrix} 1 & 0 \end{bmatrix}$$

LTI state space model

Generalizing this example, we can say that an

important class of systems have models of the form:

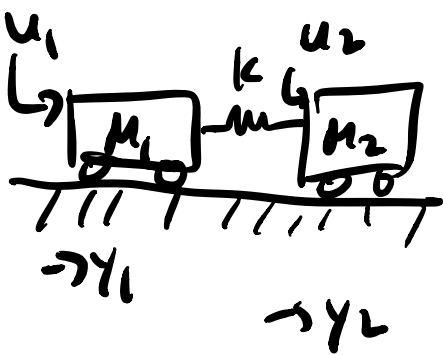
$$\begin{aligned} \dot{x} &= f(x, u) & f: \mathbb{R}^n \times \mathbb{R}^m &\rightarrow \mathbb{R}^n \\ y &= h(x, u) & h: \mathbb{R}^n \times \mathbb{R}^m &\rightarrow \mathbb{R}^p \end{aligned}$$

The model is nonlinear, there are m control inputs
 $u = \begin{bmatrix} u_1 \\ \vdots \\ u_m \end{bmatrix} \in \mathbb{R}^m$. There are p outputs $[y_1, \dots, y_p] \in \mathbb{R}^p$

and there are n state variables, $x = (x_1, \dots, x_n) \in \mathbb{R}^n$

e.g. previous example $n=2$ $m=1$ $p=1$

e.g.



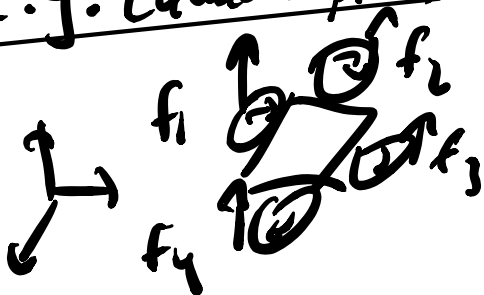
$$u = \begin{bmatrix} u_1 \\ u_2 \end{bmatrix} \in \mathbb{R}^2 \Rightarrow m=2$$

$$y = \begin{bmatrix} y_1 \\ y_2 \end{bmatrix} \in \mathbb{R}^2 \Rightarrow p=2$$

$$x = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} \in \mathbb{R}^4 \Rightarrow n=4$$

$$:= \begin{bmatrix} y_1 \\ \dot{y}_1 \\ y_2 \\ \dot{y}_2 \end{bmatrix}$$

e.g. (Quadcopter)



Claim:

4 inputs (thrusters)

3 outputs

12 states

$$m=4$$

$$p=3$$

$$n=12$$