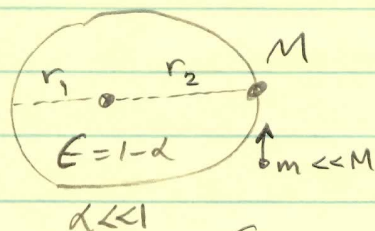


(3.10)


 Minimum KE of m for $(M+m)$ to go from ellipse to parabolic orbit.

 Suppose v - velocity of planet before collision

 v_c - " comet "

 v' - " of planet + comet after collision

 E - energy of the planet before collision that b/c it is an ellipse $E < 0$.

the condition of a parabolic orbit is that

 E' - energy after collision $E' = 0 = \frac{1}{2}(M+m)v'^2 - \frac{K'}{r_2}$

 We used \dot{r} at the aphelion and

 $K = GMM_s$ and $K' = G(M+m)M_s$

$$\dot{r} = 0$$

$$E = \frac{l^2}{2Mr_2^2} - \frac{K}{r_2} \Rightarrow \frac{K}{r_2} = \frac{l^2}{2Mr_2^2} - E$$

$$\text{Substituting } \frac{1}{2}(M+m)v'^2 \approx \left[\frac{1}{2}Mv'^2 = \frac{l^2}{2Mr_2^2} - E \right]$$

 Linear momentum is conserved $Mv + mv_c = (M+m)v' \approx Mv'$

$$l = Mr_2 v \rightarrow v' = \frac{m}{M} v_c + v = \frac{m}{M} v_c + \frac{l}{Mr_2}$$

$$\text{Substitute } v' \text{ above } \frac{1}{2}M \left(\underbrace{\left(\frac{m}{M} v_c \right)^2}_{\approx 0} + \frac{l^2}{M^2 r_2^2} + \frac{2m}{M^2} \cdot \frac{l}{r_2} v_c \right) = \frac{l^2}{2Mr_2^2} - E$$

 Let's get an expression for E :

$$E = \sqrt{1 + \frac{2El^2}{MK^2}} \Rightarrow E^2 = 1 + \frac{2El^2}{MK^2}$$

$$(1-\alpha)^2 =$$

$$1 - 2\alpha \approx 1 + \frac{2El^2}{MK^2} \Rightarrow E = -\frac{\alpha MK^2}{l^2}$$

 Substitute E to get

$$\frac{m}{M} \frac{l}{r_2} v_c = \frac{\alpha MK^2}{l^2} \Rightarrow v_c = \alpha \frac{M^2 r_2 K^2}{m l^3}$$

Now let's get an expression for r_2 and l :

We know $r_2 = a(1+e) = a(1+1-\alpha) \cong 2a$

Semi major axis \leftarrow

And we also know $a = -\frac{k}{2E} = \frac{k}{2} \cdot \frac{l^2}{\alpha M k^2} = \frac{l^2}{2\alpha M k} \Rightarrow l^2 = 2\alpha M k$

Substituting r_2 and l^2 in above:

$$v_c^2 = \alpha^2 \frac{M^4 r_2^2 k^4}{m^2 l^6} = \alpha^2 \frac{M^4 k^4}{m^2} 4a^2 \frac{1}{8a^3 \alpha^3 M^3 k^3}$$

$$= \frac{1}{2} \cdot \frac{1}{2\alpha} \frac{Mk}{m^2}$$

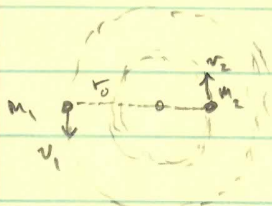
So that the energy of the comet is at least

$$E_c = \frac{1}{2} m v_c^2 = \frac{1}{2} m \frac{1}{2\alpha} \frac{Mk}{m^2}$$

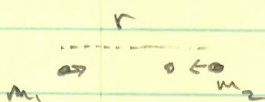
$$E_c = \frac{1}{2\alpha} \cdot \frac{M}{m} \cdot \frac{k}{2a}$$

This is a
big number
due to
that are big.

(3.11)



(orbiting)



(Stopped)

r_0 - initial distance

$$\mu = \frac{m_1 m_2}{m_1 + m_2}$$

When it stops, $E = \frac{1}{2} \mu \dot{r}^2 - \frac{k}{r}$ (note $l=0$)

reduced mass μ
distance between masses

$$k = G m_1 m_2$$

Just after stopping (initial condition) $E = -\frac{k}{r_0}$

Energy conservation $\frac{1}{2} \mu \dot{r}^2 - \frac{k}{r} = -\frac{k}{r_0}$

$$\dot{r}^2 = \frac{2k}{\mu} \left(\frac{r_0 - r}{r_0 r} \right) \Rightarrow \dot{r} = -\sqrt{\frac{2k}{\mu}} \sqrt{\frac{r_0 - r}{r_0 r}}$$

choose negative root b/c $\dot{r} < 0$, r is decreasing

$$\int_0^t dt = \int_{r_0}^0 -\sqrt{\frac{\mu}{2k}} \sqrt{\frac{r_0}{r_0 r}} dr$$

change variables to

$$r = r_0 \sin^2 \theta$$

$$dr = 2r_0 \sin \theta \cos \theta d\theta$$

$$t = -\sqrt{\frac{\mu r_0}{2k}} \int_{\pi/2}^0 \sin^2 \theta d\theta$$

$$t = -r_0 \sqrt{\frac{\mu}{2k}} (\theta - \sin \theta \cos \theta) \Big|_{\pi/2}^0 = r_0 \sqrt{\frac{\mu}{2k}} \left(\frac{\pi}{2} \right) = \sqrt{\frac{\mu}{2k}} r_0^{3/2} \frac{\pi}{2}$$

But we know that $\tau = 2\pi r_0^{3/2} \sqrt{\frac{\mu}{k}}$

So that

$$t = \frac{\tau}{4\sqrt{2}}$$

(3.14)

A) Compare perihelion distance of a parabolic orbit with radius of circular orbit. Both have same l .

$$\text{Use } \frac{1}{r} = \frac{mk}{l^2} (1 + \epsilon \cos \theta)$$

Parabolic: perihelion occurs at $\theta = 0$ with $\epsilon = 1$

$$\frac{1}{r_p} = \frac{mk}{l^2} (1+1) = 2 \frac{mk}{l^2}$$

Circular: $\epsilon = 0$

$$\frac{1}{r_c} = \frac{mk}{l^2}$$

$$\Rightarrow r_p = \frac{1}{2} \frac{l^2}{mk}$$

$$\Rightarrow \boxed{r_p = \frac{1}{2} r_c}$$

B) Starting from $r = \frac{l^2}{mk} \frac{1}{(1 + \epsilon \cos \theta)}$

$$\dot{r} = + \frac{l^2}{mk} \frac{\epsilon \sin \theta}{(1 + \epsilon \cos \theta)^2} \dot{\theta} = r \frac{\epsilon \sin \theta}{1 + \epsilon \cos \theta} \dot{\theta}$$

$$\text{Because } \vec{v} = \dot{r} \hat{r} + r \dot{\theta} \hat{\theta} \Rightarrow v^2 = \dot{r}^2 + r^2 \dot{\theta}^2$$

$$\dot{\theta} = \frac{l}{mr^2}$$

$$\text{So that } v^2 = \left[\left(\frac{\epsilon \sin \theta}{1 + \epsilon \cos \theta} \right)^2 + 1 \right] r^2 \dot{\theta}^2 = \frac{(1 + \epsilon^2 + 2\epsilon \cos \theta)}{(1 + \epsilon \cos \theta)^2} r^2 \left(\frac{l}{mr^2} \right)^2$$

$$\text{Circle: } \epsilon = 0 \quad v_c^2 = \frac{l^2}{m^2 r^2} = \frac{l^2}{m^2 r} \cdot \frac{mk}{l^2} = \frac{k}{mr} \quad (*)$$

$$\text{parabola: } \epsilon = 1 : v_p^2 = \frac{2 + 2 \cos \theta}{(1 + \cos \theta)^2} \cdot \frac{l^2}{m^2 r^2} = \frac{2l^2}{m^2 r^2} \left(\frac{1}{1 + \cos \theta} \right)$$

$$\text{but } r = \frac{l^2}{mk} \left(\frac{1}{1 + \cos \theta} \right) //$$

$$v_p^2 = \frac{2l^2}{m^2 r^2} \cdot \frac{mk r}{l^2} = \frac{2k}{mr} \quad (**)$$

So that

$$\boxed{v_p = \sqrt{2} v_c}$$