

# CO2 Estimation Models

## Delta Altitude Reward Models

### Force Model

#### Theory

Given a vehicle moving in a flat surface, there are majorly two forces acting, with other elementary forces:

$$F_T = (F_H + F_V) \cdot J \quad (1)$$

$F_T$  = Total force exerted

$F_H$  = Horizontal Forces; dependent on load momentum

$F_V$  = Vertical Forces; load and gravity only

$$F_H = m \cdot \frac{d^2V}{dT^2} \quad (2)$$

$$F_V = \frac{G \cdot M \cdot m}{r^2} \quad (3)$$

$$J - \text{Calibratable factor calculation, typically 1} \quad (4)$$

Forces exerted due to  $F_H$  are typically overpowered by the truck's engines, which essentially creates the movement and control of the freight in entirety.  $F_V$  is balanced by the structural integrity of the engine.

The demand of the engine is (usually defined in terms of torque requirement), defined as  $T_{dem}$ . In a normalised measurement scenario, the value of  $T_{dem}$  is 1 on flat roads.

However during ascent, the load  $m$  is shifted from  $F_V$  to  $F_H$  partly, effectively increasing net  $F_H$ . This increases  $T_{dem} \geq 1$  finally. The converse is valid; during descent, the  $F_H$  is reduced since  $F_H$  is moved to  $F_V$ .

From a single body propulsion point of view, torque demand is directly proportional to the force exerted by the vehicle to counteract (1):

$$T_{dem} \propto F_T \quad (5)$$

This can be converted to a linear form:

$$F_T = k \cdot T_{dem} \quad (6)$$

These two different simple models complement each other from different directions.

#### Effects

From vehicle combustion point of view, torque demand is directly proportional to the fuel consumed per kilometre:

$$T_{dem} \propto Q_{km} \quad (7)$$

And similarly creating a linear form:

$$Q_{km} = k \cdot T_{dem} \quad (8)$$

With known quantities of average fuel consumption of vehicle based on class  $Q_{km,avg}$ , torque class  $T_{eng}$  to derive  $T_{dem}$  based on average mass in vehicle; it is possible to solve (6) to get  $k$ .

Total consumed  $Q_{km}$  can be calculated from (8) with known  $k$  and  $T_{dem}$ .

On average, CO<sub>2</sub> production from diesel is defined as:

$$CO_{2,avg} = 3.17\text{g per g of diesel} \quad (9)$$

This is the maximum theoretical value derived from combustion of pure diesel, vehicles produce way less that aforementioned value, attributed by the after-treatment parts and active emission control devices, for example EGRs.

This lesser production can be a factor defined similar to  $k$ , since all equations (6), (7) and (9) are linear proportional values.

## Known Approximations

1. Very simple force body model is used throughout the analysis: that is, a vehicle is considered a uni-body. Interacting forces are not considered, for example: momentum imbalance between load and drive-train areas of a truck, sway and other road impacts.
2. The emission systems are dependent to vary between high and low load scenarios, for example: cold-start conditions, any effects due to AECD, full-load operation and limp-mode situations.
3. The factor  $k$  is considered uniform, and approximated heavily. However this can be tuned using factor  $j$  that is controlled by the user, creating a closed loop system that does not affect the calculation chain in the middle, causing and unintended effects.
4. Many external forces are neglected: wind drag, aerodynamics of the vehicle, traffic situations, road smoothness.

## Combustion Model

### Theory

Diesel Engines and all combustion engines depend on considerable amount of O<sub>2</sub> for the proper function and torque delivery. Diesel engines rely on a stoichiometric AFR

(Air-to-fuel ratio) of 14.5 : 1. Essentially this would mean that to combust 1g of diesel, the engines require a minimum of 14.5g of air (in composition, all gases including H<sub>2</sub> and N<sub>2</sub> along with O<sub>2</sub>). This theory uses the mass-flow model to determine rate of change of consumption of air, resulting in proportional change of fuel consumption based on altitude and temperature.

Rewriting the above text mathematically,

$$MF_{air} \propto Q_{act} \quad (10)$$

$$Q_{act} = k \cdot MF_{air}, \text{ where } k = 14.5. \quad (11)$$

From (9), we know that 1g of  $Q_{act} = 3.17\text{g}$  of CO<sub>2</sub>,

$$\implies \frac{dm_{CO_2}}{dQ_{act}} = 3.17\text{g/g} \quad (12)$$

Estimated mass flow can be calculated empirically from ideal gas equation:

$$MF_{in} = \frac{P_{env} \cdot V_{engine}}{R \cdot T_{env}} \quad (13)$$

Lapse rate due to altitude change also has a relation:

$$L = -\frac{dT}{dH} \text{ degC/m}$$

This lapse rate is known for every geographic region and is uniform over continents. This can be used to determine  $dT$ :

$$dT = -L \cdot dH \quad (14)$$

Considering adiabatic compression scenario, (13) can also be written as:

$$P \cdot dV = -\frac{R \cdot dP}{\gamma}; \quad \gamma = c_p/c_v \quad (15)$$

Combine all equations and simplify from (13) to (15) to get:

$$MF_{air} = \frac{M \cdot C_p \cdot dV_{engine}}{R} \quad (16)$$

All values are known, except for  $C_p$ . This can be evaluated using Sutherland's law for mass flow and compression:

$$C_p = \frac{A + B \cdot \left( \frac{\frac{C}{T_{env}}}{\sinh\left(\frac{C}{T_{env}}\right)} \right)^2 + D \cdot \left( \frac{\frac{E}{T_{env}}}{\cosh\left(\frac{E}{T_{env}}\right)} \right)^2}{MW} \quad (17)$$

$$\begin{aligned}A &= 28958 \\B &= 9390 \\C &= 3012 \\D &= 7580 \\E &= 1484 \\MW &= 28.951 \text{ kg/mol}\end{aligned}$$

This estimated mass flow from (16) can be applied to calculate required fuel change for combustion in (11), then re-applied to (12) to determine change of CO<sub>2</sub> emissions.

## Known Approximations

1. The following environmental and vehicle conditions are assumed:
  1. Ideal Gas Conditions:
    1. Temperature = 21 degC
    2. Pressure = 1014 hPa
    3. Volume = 1 litre of O<sub>2</sub>
2. The vehicle is assumed to have one turbo-charger, and is assumed primarily in adiabatic conditions in full load.

## References

1. Specific Heat Capacities: <https://www.grc.nasa.gov/www/k-12/BGP/specheat.html>
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3. Sutherland Law and corresponding values: <https://ubitutors.com/thermal-properties-of-air-at-a-given-temperature/>
4. Sutherland Law in terms of Conductivity: [https://doc.comsol.com/5.5/doc/com.comsol.help.cfd/cfd\\_ug\\_fluidflow\\_high\\_mac\\_h.08.27.html](https://doc.comsol.com/5.5/doc/com.comsol.help.cfd/cfd_ug_fluidflow_high_mac_h.08.27.html)