2017/2018 SEMESTER ONE EXAMINATION

Diploma in Electrical & Electronic Engineering 3rd Year FT

SATELLITE & OPTICAL COMMUNICATION

Time Allowed: 2 Hours

Instructions to Candidates

- 1. The examination rules set out on the last page of the answer booklet are to be complied with.
- 2. This paper consists of **THREE** sections:

Section A - 20 Multiple Choice Questions, 2 marks each.

Section B - 4 Short Questions, 10 marks each.

Section C - 1 Long Questions, 20 marks each.

- 3. ALL questions are COMPULSORY.
- 4. All questions are to be answered in the answer booklet.
- 5. Start each question in Sections B and C on a new page.
- 6. Fill in the Sections B and C Question Numbers, in the order that they were answered, in the boxes found on the front cover of the answer booklet under the column "Question Answered".
- 7. This paper consists of 8 pages, inclusive of formula sheets.

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SECTION A

MULTIPLE CHOICE QUESTIONS [2 marks each]

- 1. Please **tick** your answers in the **MCQ box** behind the front cover of the answer booklet.
- 2. No marks will be deducted for incorrect answers.
- A1. Single mode fiber is used to:
 - (a) Reduce modal dispersion.
 - (b) Reduce material dispersion.
 - (c) Eliminate modal dispersion.
 - (d) Eliminate material dispersion.
- A2. Which one of the following statements is correct?
 - (a) In graded index multimode fiber, light travels down the fiber along a curved path.
 - (b) In graded index multimode fiber, light travels down the fiber along a zigzag path.
 - (c) In step index multimode fiber, light travels down the fiber along a circular path.
 - (d) In step index multimode fiber, light travels down the fiber along a curved path.
- A3. Rayleigh scattering is the type of fiber attenuation which is:
 - (a) Directly proportional to the fourth power of the wavelength.
 - (b) Directly proportional to the fifth power of the wavelength.
 - (c) Inversely proportional to the fourth power of the wavelength.
 - (d) Inversely proportional to the fifth power of the wavelength.
- A4. In APD, the high carrier multiplication is caused by:
 - (a) High forward biased voltage.
 - (b) High reverse biased voltage.
 - (c) High population inversion.
 - (d) High threshold voltage.
- A5. One of the advantages of ILD source is that:
 - (a) It produces coherent light.
 - (b) It can be used with any detector.
 - (c) It produces less noise.
 - (d) It is not temperature dependent.
- A6. Which one of the following events is not a Fresnel Reflection (FR) event?
 - (a) Dead zone.
 - (b) Splice.
 - (c) Ghost.
 - (d) End of fiber.

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- A7. Identify the incorrect statement:
 - (a) To attain total internal reflection in a fiber, the cladding refractive index must be smaller than the core refractive index.
 - (b) In glass, the reflective index varies with wavelength.
 - (c) A large acceptance angle implies a large numerical aperture.
 - (d) The numerical aperture of a fiber depends on the line width of the source used.
- A8. Photodiodes used as optical detectors are:
 - (a) Unbiased.
 - (b) Forward biased.
 - (c) Reversed biased.
 - (d) Thermoelectrically cooled.
- A9. Responsivity of a photodetector is defined as:
 - (a) The spectral response.
 - (b) The amount of current produced per unit light power input.
 - (c) The amount of time needed to respond to an input.
 - (d) The amount of power output given the input current.
- A10. Losses incurred as light rays travel through fiber are the result of:
 - (a) Reflection and absorption.
 - (b) Refraction and reflection.
 - (c) Scattering and absorption.
 - (d) Total Internal Reflection (TIR) and absorption.
- A11. Which amplifier is used in the receive subsystem of the earth satellite station:
 - (a) Travelling Wave Tube Amplifier.
 - (b) Low Noise Amplifier.
 - (c) Solid State Power Amplifier.
 - (d) Channel Amplifier.
- A12. Which one of the following is not a common IF frequency used in earth satellite station:
 - (a) 770 MHz.
 - (b) 140 MHz.
 - (c) 70 MHz.
 - (d) 90 MHz.
- A13. The modulation normally used with digital data is:
 - (a) AM.
 - (b) FM.
 - (c) SSB.
 - (d) QPSK.

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- A14. The approximate time taken by GPS satellites to orbit around the earth is:
 - (a) More than 24 hours.
 - (b) Less than 24 hours.
 - (c) 12 hours.
 - (d) 24 hours.
- A15. Geostationary satellites use their secondary power supplies:
 - (a) During the hours of darkness.
 - (b) During the solstice times in December and June each year.
 - (c) During the equinox times in March and September each year.
 - (d) Only if and when the solar cells fail.
- A16. Round trip propagation delay between two earth stations through a geostationary satellite is approximately equal to:
 - (a) 0.5 sec.
 - (b) 1.5 sec.
 - (c) 2.0 sec.
 - (d) 2.5 sec.
- A17. A satellite in a circular orbit above the equator takes 5 to 12 hours to complete one orbit.

The satellite is in:

- (a) A geostationary orbit.
- (b) A Helios-synchronous orbit.
- (c) A low earth orbit.
- (d) A medium earth orbit.
- A18. Which one of the following statements is true for geostationary satellites?
 - (a) Earth's polar regions are not covered by geostationary satellites.
 - (b) About 60% of the Earth's surface is covered by a geostationary satellite.
 - (c) The round trip propagation delay through a geostationary satellite is less than 50 ms.
 - (d) The altitude of the geostationary satellite must be 33,786 km.
- A19. The function of Solid State Power Amplifier (SSPA) in the earth satellite station is:
 - (a) To translate the signal frequencies.
 - (b) To reduce the system noise.
 - (c) To boost the signal to a sufficient high power.
 - (d) To modulate the baseband signals.
- A20. In a full transponder TDMA satellite system how many carriers are present in the transponder at any instant in time?
 - (a) 1
 - (b) 24
 - (c) 36
 - (d) 900

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Section B [10 Marks Each]

- B1. An optical communication link consists of the following components:
 - ► ILD with bandgap energy 0.950 eV
 - P Optical fiber with core refractive index 1.45, cladding refractive index 1.35
 - PIN photodiode
 - (a) Determine the operating wavelength of the optical communication link. (2 marks)
 - (b) Determine the numerical aperture & acceptance cone angle of the fiber. (4 marks)
 - (c) If the laser light from ILD enters the fiber at an angle of 34⁰ with the fiber core axis, can the laser light be detected by the PIN photodiode?

 Give reasons to support your answer.

 (4 marks)
- B2. An optical fiber link is set up using the following devices:
 - ➤ ILD source
 - ➤ 10 km length of single mode fiber
 - ► PIN photodiode with a responsivity of 0.8 A/W
 - (a) Give TWO advantages of using the ILD source. (4 marks)
 - (b) Name the dominant dispersion that occurs in the fiber. (1 mark)
 - (c) 9 x 10¹⁰ photons/sec are incident onto the PIN photodiode.

 Determine the operating wavelength in nm, if the current generated at the photodiode output terminal is found to be 9.23nA. (5 marks)
- B3. You have the following components available for setting up an optical fiber link:

ILD source : Output power 5mW; Risetime 1 ns

Step index fiber : Attenuation 2 dB/km; Dispersion 243.2 ps/km.

Optical connectors : Loss 0.5 dB each

Optical splices : Loss 0.2 dB each; Total of 20 splices APD detector : Sensitivity -53 dBm; Risetime 2 ns

Safety margin : 6 dB

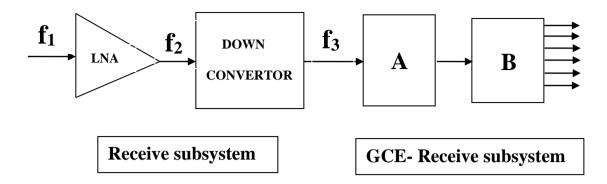
(a) Use **power budget** analysis to determine the maximum link length. (3 marks)

(b) Find the maximum Bit Rate (in Mb/s) of the above link using NRZ signal. (4 marks)

(c) State the THREE wavelengths commonly used in the optical fiber link. (3 marks)

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B4. The block diagram shows Receive & GCE-Receive subsystems of the satellite earth station:



(a) Name Box A and B in the above block diagram.

(2 marks)

(b) State the functions of LNA and Down Converter.

- (2 marks)
- (c) Identify the frequency values of f_1 , f_2 and f_3 for the following frequency bands:
 - (i) Ku-band.
 - (ii) C-band

(6 marks)

Section C [20 Marks]

C1. A satellite communications system employing Ku band has the following parameters:

Up link distance : 38,000 km Down link distance : 42,000 km Earth station transmitting power : 2 kW Earth station transmitting antenna gain: 65 dB Earth station receiving antenna gain : 51 dB Earth station G/T : 27 dB/K Satellite transponder output power : 18 dBW Satellite transmitting antenna gain : 16 dB Satellite G/T $: -10 \, dB/K$ Satellite transponder bandwidth : 36 MHz

(a)	State the RF frequency appeared at the input of the earth station terminal.	(2 marks)
(b)	Calculate the round trip echo delay in ms.	(4 marks)

- (c) Calculate the EIRP values in dBW at the satellite & earth station terminals. (4 marks)
- (d) Determine the received signal level in pW at the court station terminal. (4 marks)
- (d) Determine the received signal level in pW at the earth station terminal. (4 marks)
- (e) Calculate the C/N value at the input of the earth station terminal. (6 marks)

***** End of Paper ******

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Constants & Formulas Sheet

Gravitational constant $G = 6.673 \times 10^{-11} \text{ Nm}^2 \text{ Kg}^{-2}$

Mass of Earth $M_e = 5.975 \times 10^{24} \, \text{Kg}$, Radius of Earth = 6378 km

Boltzmann's constant $k = 1.38 \times 10^{-23} \text{ J/K}$

Speed of light $c = 3 \times 10^8 \text{ m/s}$

Plank's constant $h = 6.626 \times 10^{-34} Js$

Electron charge $e = 1.602 \times 10^{-19} C$

$$v = \sqrt{\frac{Gm_e}{r}}$$

$$S(\Theta) = 52 - 10\log_{10}\left(\frac{D}{\lambda}\right) - 25\log(\Theta)$$
 dBi

For $D < 100\lambda$

$$S(\Theta) = 29 - 25 \log_{10}(\Theta)$$
 dBi

For 2.0⁰ spacing

$$S(\Theta) = 32 - 25\log_{10}(\Theta)$$
 dBi

For 2.9⁰ spacing

$$L_{dB} = 20\log\left(\frac{4\pi d}{\lambda}\right) = 32.44 + 20\log[d] + 20\log[f]$$

$$(C)_{dBW} = (P_T)_{dBW} + (G_T)_{dB} + (G_R)_{dB} - (L)_{dB}$$

$$\left(\frac{C}{N_0}\right)_{dBHz} = (P_T)_{dBW} + (G_T)_{dB} + \left(\frac{G_R}{T}\right)_{dB} - (L)_{dB} - (L_o)_{dB} - 10\log K$$

$$\left(\frac{C}{N}\right)_{dB} = \left(\frac{C}{N_0}\right)_{dBHz} - 10\log B$$

$$\left(\frac{C}{N}\right)_{Total}^{-1} = \left(\frac{C}{N}\right)_{Up}^{-1} + \left(\frac{C}{N}\right)_{Down}^{-1}$$

$$n_1 Sin\Theta_1 = n_2 Sin\Theta_2$$
 $NA = Sin\Theta_a = \sqrt{n_1^2 - n_2^2}$

$$\Delta = \frac{n_1^2 - n_2^2}{2n_1^2} \qquad \Delta \tau = \frac{Ln_1}{c} \left(\frac{n_1 - n_2}{n_2} \right) \qquad v = \frac{c}{n}$$

$$V = \frac{\pi d}{\lambda} (NA) \qquad n(r) = n_1 \sqrt{1 - 2\Delta \left(\frac{r}{a}\right)^{\alpha}} \qquad a \le \frac{2.405\lambda}{2\pi \sqrt{n_1^2 - n_2^2}}$$

For step-index multimode fiber

For graded index multimode fiber

$$M = \frac{V^2}{2} \qquad \sigma = \frac{n_1 L \Delta}{c \sqrt{12}}$$

$$M = \frac{V^2}{4} \qquad \sigma = \frac{n_1 L \Delta^2}{c\sqrt{48}}$$

$$B = \frac{0.35}{\sigma} \qquad B = \frac{0.35}{t_{sys}} \qquad f = \frac{c}{\lambda}$$

$$\lambda = \frac{hc}{E_g}$$
 $\lambda = \frac{1.24}{E_g}$ $D = \frac{ct}{2n}$

$$I_p = (r_e)(e)$$
 $P_o = (r_p)(hc/\lambda)$ $R = \frac{\eta \lambda e}{hc}$

$$t_f = \sqrt{t_{\text{mod}al}^2 + t_{\text{material}}^2}$$
 $t_{\text{sys}} = 1.1\sqrt{t_f^2 + t_s^2 + t_d^2}$