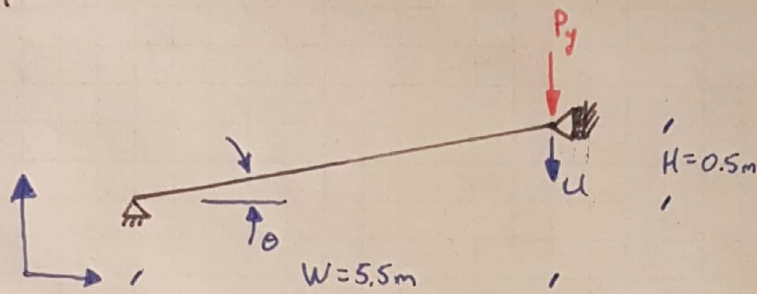


Assignment 1

CESG 506

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1-1



$$EA = 2100 \text{ kN}$$

$$1a \quad \epsilon = \frac{L-L}{L}$$

$$1b \quad \epsilon = \frac{1}{2} \frac{L^2 - L^2}{L^2}$$

$$1c \quad \epsilon = \frac{1}{2} \frac{L^2 - L^2}{L^2}$$

$$1d \quad \tilde{\epsilon} = \frac{1}{2} \ln\left(\frac{L^2}{L^2}\right)$$

1. Derive force-displacement relation for all possible combinations (8 options) of kinematic relations and equilibrium on the undeformed and the deformed configuration and compare them in a single plot for $0 \leq u \leq 2.5 H$. Include the linear solution in the same plot

$$\underline{L} = \begin{Bmatrix} 5.5 \\ 0.5 \end{Bmatrix}$$

$$\underline{L} = \begin{Bmatrix} 5.5 \\ 0.5 - u \end{Bmatrix}$$

$$L^2 = \underline{L} \cdot \underline{L} = 30.5$$

$$L^2 = \underline{L} \cdot \underline{L} = 30.5 + u^2 - u$$

$$f = EA \cdot \epsilon$$

$$P_y = f \cdot \sin(\theta)$$

$$\sin(\theta) = \frac{H-u}{L}$$

Undeformed

Deformed

$$1a \quad P_y = EA \cdot \frac{L-L}{L} \cdot \frac{H}{L}$$

$$P_y = EA \cdot \frac{L-L}{L} \cdot \frac{H-u}{L}$$

$$1b \quad P_y = EA \cdot \frac{L^2 - L^2}{2L^2} \cdot \frac{H}{L}$$

$$P_y = EA \cdot \frac{L^2 - L^2}{2L^2} \cdot \frac{H-u}{L}$$

$$1c \quad P_y = EA \cdot \frac{L^2 - L^2}{2L^2} \cdot \frac{H}{L}$$

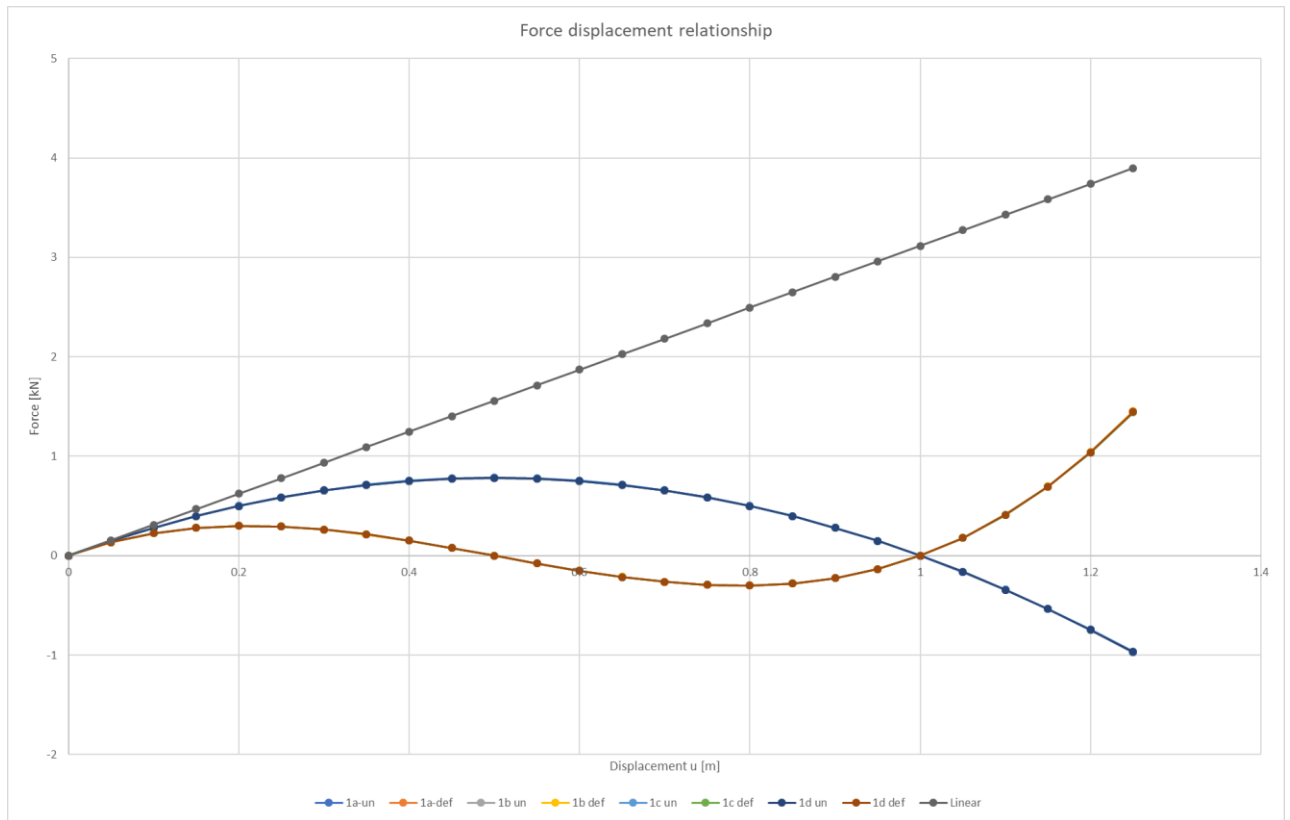
$$P_y = EA \cdot \frac{L^2 - L^2}{2L^2} \cdot \frac{H-u}{L}$$

$$1d \quad P_y = \frac{EA}{2} \ln\left(\frac{L^2}{L^2}\right) \cdot \frac{H}{L}$$

$$P_y = \frac{EA}{2} \ln\left(\frac{L^2}{L^2}\right) \cdot \frac{H-u}{L}$$

Problem 1-1

1.



Problem 1-1

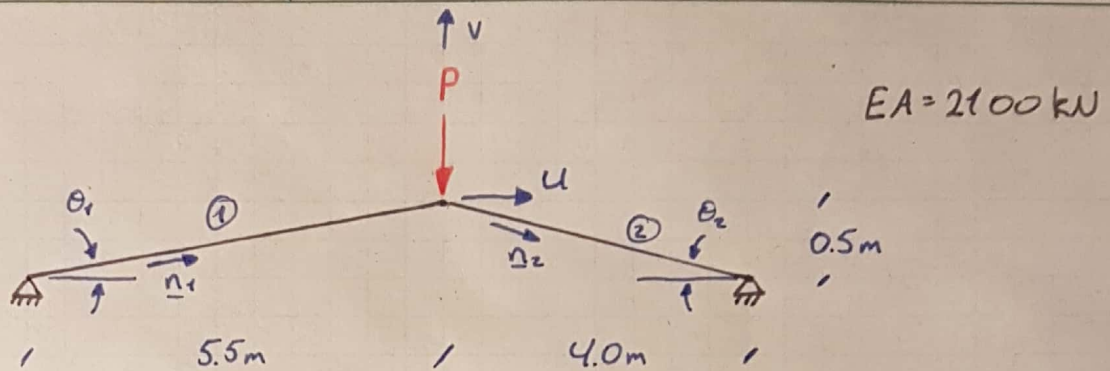
2.

What formulation is better would depend on what range I expect to be working in and how accurate the calculations need to be. If I know that I will be working with low loads that will cause small deflections and be in the range where the system is close to linear. I would choose to use linear formulation with the undeformed equilibrium. That makes the calculations much simpler and saves computational power/time.

On the other hand, if I expect the deflections to be large enough for the system to start behaving in considerably non-linear manner. I would choose to use a formulation based on the deformed equilibrium. When it comes to choosing a strain, I wouldn't worry much about which one I would choose as all of them give really similar results. I would go for strain formulation that would allow for quicker computation.

For larger deformations I would go for a strain that does not blow up to infinity anywhere close to my working range.

1-2



1. Using Henkey strain, a linear relation $\sigma = E \cdot \epsilon$ $A = \text{const.}$ and equilibrium on the deformed system, derive the relationship between displacement and forces on the free node i.e.

$$\begin{Bmatrix} P_x \\ P_y \end{Bmatrix} = \begin{Bmatrix} F_x(u, v) \\ F_y(u, v) \end{Bmatrix}$$

$$\epsilon = \frac{1}{2} \ln(\lambda^2) = \frac{1}{2} \ln\left(\frac{L^2}{L_0^2}\right) \quad (\text{Henkey strain})$$

$$L_1^2 = 5.5^2 + 0.5^2 = 30.5 \text{ m}^2$$

$$L_2^2 = 4.0^2 + 0.5^2 = 16.25 \text{ m}^2$$

$$L_1^2 = (5.5 + u)^2 + (0.5 + v)^2$$

$$L_2^2 = (4.0 - u)^2 + (0.5 + v)^2$$

$$= 30.5 + u^2 + v^2 + 11u + v$$

$$= 16.25 + u^2 + v^2 - 8u + v$$

$$P = -(f_{1,j} + f_{2,i}) = -(f_1 \cdot \underline{n}_1 + (-f_2 \cdot \underline{n}_2)) = -(f_1 \cdot \underline{n}_1 - f_2 \cdot \underline{n}_2) = f_2 \cdot \underline{n}_2 - f_1 \cdot \underline{n}_1$$

same for both elements

$$P = EA \left(\frac{L_2}{\|L_2\|} \epsilon_2 - \frac{L_1}{\|L_1\|} \epsilon_1 \right)$$

$$\underline{n}_k = \frac{\underline{V}_k}{\|\underline{V}_k\|}$$

$$\underline{L}_1 = \begin{Bmatrix} 5.5 \\ 0.5 \end{Bmatrix} + \begin{Bmatrix} u \\ v \end{Bmatrix}$$

$$\underline{L}_2 = \begin{Bmatrix} 4.0 \\ -0.5 \end{Bmatrix} + \begin{Bmatrix} -u \\ -v \end{Bmatrix} = \begin{Bmatrix} 4.0 \\ -0.5 \end{Bmatrix} - \begin{Bmatrix} u \\ v \end{Bmatrix}$$

$$\|\underline{L}_1\| = L_1$$

$$\|\underline{L}_2\| = L_2$$

$$P = EA \left(\frac{\epsilon_2}{L_2} \begin{Bmatrix} 4.0 - u \\ -0.5 - v \end{Bmatrix} - \frac{\epsilon_1}{L_1} \begin{Bmatrix} 5.5 + u \\ 0.5 + v \end{Bmatrix} \right) = \begin{Bmatrix} P_x \\ P_y \end{Bmatrix}$$

Also written as

$$f_2 \underline{n}_2 - f_1 \underline{n}_1 = P$$

1-2

2. Find the tangent stiffness matrix of the system. Keep it simple and don't substitute long expressions into your answer

Tangent stiffness for element is written as

$$[K^e] = \begin{bmatrix} k^e & -k^e \\ -k^e & k^e \end{bmatrix}$$

where

$$k^e = \frac{EA}{L} \underline{n} \otimes \underline{n} + \frac{\tilde{F}}{L} (1 - \underline{n} \otimes \underline{n}) \quad \text{where } \tilde{F} \text{ is force in element}$$

Our global stiffness matrix is then

$$\begin{matrix} & \begin{matrix} 1 & 2 & 3 \end{matrix} \\ \begin{matrix} 1 \\ 2 \\ 3 \end{matrix} & \begin{bmatrix} k^1 & -k^1 & 0 \\ -k^1 & k^1 + k^2 & -k^2 \\ 0 & -k^2 & k^2 \end{bmatrix} \end{matrix}$$

with nodes 1 & 3 being supported

node 2 unsupported

u_1 & $u_3 = 0$, we are therefore only interested in $K_{ff} = k^1 + k^2$

$$k^1 = \frac{EA}{L_1} \underline{n}_1 \otimes \underline{n}_1 + \frac{\tilde{F}_1}{L_1} (1 - \underline{n}_1 \otimes \underline{n}_1)$$

$$\underline{n}_1 = \begin{Bmatrix} \cos(\theta_1) \\ \sin(\theta_1) \end{Bmatrix} = \begin{Bmatrix} c_1 \\ s_1 \end{Bmatrix} \quad \tilde{F}_1 = EAE_1$$

$$= \frac{EA}{L_1} \begin{bmatrix} c_1^2 & c_1 s_1 \\ c_1 s_1 & s_1^2 \end{bmatrix} + \frac{EAE_1}{L_1} \left(\begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} - \begin{bmatrix} c_1^2 & c_1 s_1 \\ c_1 s_1 & s_1^2 \end{bmatrix} \right)$$

$$= \frac{EA(1-E_1)}{L_1} \begin{bmatrix} c_1^2 & c_1 s_1 \\ c_1 s_1 & s_1^2 \end{bmatrix} + \frac{EAE_1}{L_1} \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

1-2 2. continued

$$k^2 = \frac{EA}{l_2} \underline{n}_2 \otimes \underline{n}_2 + \frac{\tilde{f}_2}{l_2} \left(\underline{1} - \underline{n}_2 \otimes \underline{n}_2 \right)$$

$$\underline{n}_2 = \begin{Bmatrix} \cos(\theta_2) \\ -\sin(\theta_2) \end{Bmatrix} = \begin{Bmatrix} c_2 \\ -s_2 \end{Bmatrix} \quad \tilde{f}_2 = EA \cdot \epsilon_2$$

$$= \frac{EA}{l_2} \begin{bmatrix} c_1^2 & -c_1 s_1 \\ -c_1 s_1 & s_1^2 \end{bmatrix} + \frac{EA \epsilon_2}{l_2} \left(\begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} + \begin{bmatrix} c_1^2 & -c_1 s_1 \\ -c_1 s_1 & s_1^2 \end{bmatrix} \right)$$

$$= \frac{EA(1-\epsilon_2)}{l_2} \begin{bmatrix} c_1^2 & -c_1 s_1 \\ -c_1 s_1 & s_1^2 \end{bmatrix} + \frac{EA \epsilon_2}{l_2} \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

K_{ff} then becomes

$$K_{ff} = \begin{bmatrix} K_{11} & K_{12} \\ K_{21} & K_{22} \end{bmatrix}$$

where

$$K_{11} = \frac{EA(1-\epsilon_1)}{l_1} c_1^2 + \frac{EA \epsilon_1}{l_1} + \frac{EA(1-\epsilon_2)}{l_2} c_2^2 + \frac{EA \epsilon_2}{l_2}$$

$$= EA \left(\frac{c_1^2(1-\epsilon_1) + \epsilon_1}{l_1} + \frac{c_2^2(1-\epsilon_2) + \epsilon_2}{l_2} \right)$$

$$K_{21} = K_{12} = \frac{EA(1-\epsilon_1)}{l_1} c_1 s_1 + \frac{EA(1-\epsilon_2)}{l_2} (-c_1 s_2)$$

$$= EA \left(\frac{c_1 s_1(1-\epsilon_1)}{l_1} - \frac{c_1 s_2(1-\epsilon_2)}{l_2} \right)$$

$$K_{22} = \frac{EA(1-\epsilon_1)}{l_1} s_1^2 + \frac{EA \epsilon_1}{l_1} + \frac{EA(1-\epsilon_2)}{l_2} s_2^2 + \frac{EA \epsilon_2}{l_2}$$

$$= EA \left(\frac{s_1^2(1-\epsilon_1) + \epsilon_1}{l_1} + \frac{s_2^2(1-\epsilon_2) + \epsilon_2}{l_2} \right)$$

Terms ϵ, l have been written out on previous pages.

1-2

3. Formulate the condition $P_x = 0$ to the system of equations. Will this allow you to eliminate the horizontal displacement from the system of equations? Why? Why not?

From 1-2.1. we have

$$\frac{EA\epsilon_2}{l_2} \begin{Bmatrix} 4.0 - u \\ -0.5 - v \end{Bmatrix} - \frac{EA\epsilon_1}{l_1} \begin{Bmatrix} 5.5 + u \\ 0.5 + v \end{Bmatrix} = \begin{Bmatrix} P_x \\ P_y \end{Bmatrix}$$

Now setting $P_x = 0$ gives

$$\frac{EA\epsilon_2}{l_2} \begin{Bmatrix} 4.0 - u \\ -0.5 - v \end{Bmatrix} - \frac{EA\epsilon_1}{l_1} \begin{Bmatrix} 5.5 + u \\ 0.5 + v \end{Bmatrix} = \begin{Bmatrix} 0 \\ P_y \end{Bmatrix}$$

Looking only at force in x direction

$$EA \left(\underbrace{\frac{\epsilon_2}{l_2} (4.0 - u) - \frac{\epsilon_1}{l_1} (5.5 + u)}_{= 0 \text{ to fulfill condition}} \right) = 0$$

Solving for u gives

$$u = \frac{5.5 \cdot \frac{\epsilon_1}{l_1} - 4.0 \cdot \frac{\epsilon_2}{l_2}}{-\left(\frac{\epsilon_1}{l_1} + \frac{\epsilon_2}{l_2}\right)}$$

But $l_1, l_2, \epsilon_1, \epsilon_2$ are all functions of u , we can therefore not eliminate u .

Side note, if system was symmetric and $l_1 = \begin{Bmatrix} a \\ b \end{Bmatrix}$ $l_2 = \begin{Bmatrix} a \\ -b \end{Bmatrix}$

we would get

$$\frac{\epsilon_2}{l_2} (a - u) - \frac{\epsilon_1}{l_1} (a + u) \quad \begin{matrix} w/ \epsilon_2 = \epsilon_1 = \epsilon \\ l_2 = l_1 = l \end{matrix}$$

$$\Rightarrow 2u \frac{\epsilon}{l} = P_x = 0$$

which would only be fulfilled at $u = 0$ and we could eliminate u

Problem 1-2

4.

P_{cr} was found by controlling vertical displacement and finding the first local maximum as node 2 is moved down wards. Each step of the way horizontal displacement was iterated on to get total horizontal force equal to zero.

$$P_{cr} = -0.9817134398668483 \text{ kN}$$

Results from each iteration step during the process

Load step = 0

#0 $u=0.0000e+00$ $v=0.0000e+00$ $R_x=0.000000e+00$ $R_y=0.000000e+00$

Load step = 0.25

#0 $u=0.0000e+00$ $v=0.0000e+00$ $R_x=0.000000e+00$ $R_y=2.454284e-01$
#1 $u=-8.1190e-04$ $v=-2.4224e-02$ $R_x=-1.716990e-02$ $R_y=1.787191e-02$
#2 $u=-8.5612e-04$ $v=-2.6212e-02$ $R_x=-1.135897e-04$ $R_y=1.188102e-04$
#3 $u=-8.5642e-04$ $v=-2.6226e-02$ $R_x=-5.160655e-09$ $R_y=5.379573e-09$
#4 $u=-8.5642e-04$ $v=-2.6226e-02$ $R_x=6.616929e-14$ $R_y=-3.969047e-14$

Load step = 0.5

#0 $u=-8.5642e-04$ $v=-2.6226e-02$ $R_x=6.616929e-14$ $R_y=2.454284e-01$
#1 $u=-1.7640e-03$ $v=-5.4771e-02$ $R_x=-2.397555e-02$ $R_y=2.350293e-02$
#2 $u=-1.8342e-03$ $v=-5.8018e-02$ $R_x=-3.059614e-04$ $R_y=2.969055e-04$
#3 $u=-1.8352e-03$ $v=-5.8060e-02$ $R_x=-5.138651e-08$ $R_y=4.960239e-08$
#4 $u=-1.8352e-03$ $v=-5.8060e-02$ $R_x=-7.549517e-14$ $R_y=2.203793e-14$

Load step = 0.75

#0 $u=-1.8352e-03$ $v=-5.8060e-02$ $R_x=-7.549517e-14$ $R_y=2.454284e-01$
#1 $u=-2.9006e-03$ $v=-9.3938e-02$ $R_x=-3.812335e-02$ $R_y=3.452191e-02$
#2 $u=-3.0409e-03$ $v=-1.0063e-01$ $R_x=-1.316545e-03$ $R_y=1.150251e-03$
#3 $u=-3.0458e-03$ $v=-1.0087e-01$ $R_x=-1.674827e-06$ $R_y=1.446128e-06$
#4 $u=-3.0458e-03$ $v=-1.0087e-01$ $R_x=-3.090861e-12$ $R_y=2.350675e-12$
#5 $u=-3.0458e-03$ $v=-1.0087e-01$ $R_x=8.482104e-13$ $R_y=-3.963496e-14$

Load step = 0.99

#0 $u=-3.0458e-03$ $v=-1.0087e-01$ $R_x=8.482104e-13$ $R_y=2.356112e-01$
#1 $u=-4.4038e-03$ $v=-1.5143e-01$ $R_x=-7.633330e-02$ $R_y=6.121922e-02$
#2 $u=-4.9093e-03$ $v=-1.7656e-01$ $R_x=-1.886744e-02$ $R_y=1.366535e-02$
#3 $u=-5.1060e-03$ $v=-1.8654e-01$ $R_x=-2.985580e-03$ $R_y=2.025239e-03$
#4 $u=-5.1469e-03$ $v=-1.8863e-01$ $R_x=-1.312852e-04$ $R_y=8.686342e-05$
#5 $u=-5.1488e-03$ $v=-1.8873e-01$ $R_x=-2.884631e-07$ $R_y=1.899347e-07$
#6 $u=-5.1488e-03$ $v=-1.8873e-01$ $R_x=-1.173284e-12$ $R_y=8.997247e-13$
#7 $u=-5.1488e-03$ $v=-1.8873e-01$ $R_x=-4.494183e-13$ $R_y=1.421085e-14$

Load step = 0.999

#0 u=-5.1488e-03 v=-1.8873e-01 Rx=-4.494183e-13 Ry=8.835421e-03

#1 u=-5.3659e-03 v=-1.9907e-01 Rx=-3.220339e-03 Ry=2.080566e-03

#2 u=-5.4500e-03 v=-2.0338e-01 Rx=-5.605904e-04 Ry=3.537781e-04

#3 u=-5.4716e-03 v=-2.0449e-01 Rx=-3.705937e-05 Ry=2.311145e-05

#4 u=-5.4732e-03 v=-2.0457e-01 Rx=-2.090722e-07 Ry=1.300174e-07

#5 u=-5.4732e-03 v=-2.0457e-01 Rx=-6.854961e-12 Ry=4.194978e-12

#6 u=-5.4732e-03 v=-2.0457e-01 Rx=8.526513e-13 Ry=1.676437e-14

1-2

5.

I could compute to 1×10^{-13} kN error. As soon as I went to 1×10^{-14} kN my code did not converge which was a cut of point for all the load steps.

<p>Load step = 0</p> <p>0, 0.000000e+00</p> <p>Load step = 0.25</p> <p>0, 2.454284e-01</p> <p>1, 2.478327e-02</p> <p>2, 1.643730e-04</p> <p>3, 7.454674e-09</p> <p>4, 7.716028e-14</p> <p>Load step = 0.5</p> <p>0, 2.454284e-01</p> <p>1, 3.357401e-02</p> <p>2, 4.263394e-04</p> <p>3, 7.142108e-08</p> <p>4, 7.864598e-14</p>	<p>Load step = 0.75</p> <p>0, 2.454284e-01</p> <p>1, 5.143104e-02</p> <p>2, 1.748247e-03</p> <p>3, 2.212765e-06</p> <p>4, 3.883181e-12</p> <p>5, 8.491359e-13</p> <p>Load step = 0.99</p> <p>0, 2.356112e-01</p> <p>1, 9.784971e-02</p> <p>2, 2.329640e-02</p> <p>3, 3.607670e-03</p> <p>4, 1.574200e-04</p> <p>5, 3.453783e-07</p> <p>6, 1.478546e-12</p> <p>7, 4.496429e-13</p>	<p>Load step = 0.999</p> <p>0, 8.835421e-03</p> <p>1, 3.833972e-03</p> <p>2, 6.628880e-04</p> <p>3, 4.367535e-05</p> <p>4, 2.462026e-07</p> <p>5, 8.036686e-12</p> <p>6, 8.528161e-13</p>
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