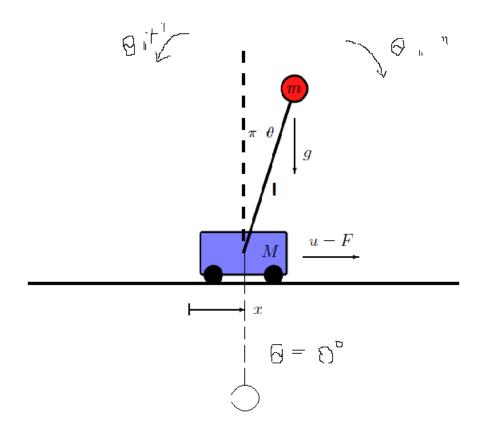
```
clc
clear all
close all
% System definition
```



Equations:

$$\begin{split} x &= \begin{pmatrix} x \\ \dot{x} \\ \dot{\theta} \\ \dot{\theta} \end{pmatrix} = \begin{pmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{pmatrix} \\ \ddot{x} &= \frac{-m^2.l^2.g.\cos(\theta).\sin(\theta) + m.l^2(m.l.\dot{\theta}^2.\sin(\theta) - b.\dot{x}) + m.l^2.F}{m.l^2(M + m(1 - \cos^2(\theta)))} \\ \ddot{\theta} &= \frac{(m + M)m.g.l.\sin(\theta) - m.l.\cos(\theta)(m.l.\dot{\theta}^2.\sin(\theta) - b.\dot{x}) - m.l.\cos(\theta).F}{m.l^2(M + m(1 - \cos^2(\theta)))} \\ \dot{x}_1 &= x_2 \\ \dot{x}_2 &= \frac{-m^2.l^2.g.\cos(x_3).\sin(x_3) + m.l^2(m.l.x_4^2.\sin(x_3) - b.x_2 + m.l^2.F}{m.l^2(M + m(1 - \cos^2(x_3)))} \\ \dot{x}_3 &= x_4 \\ \dot{x}_4 &= \frac{(m + M)m.g.l.\sin(x_3) - m.l.\cos(x_3)(m.l.x_4^2.\sin(x_3) - b.x_2) - m.l.\cos(x_3).F}{m.l^2(M + m(1 - \cos^2(x_3)))} \end{split}$$

Linearization:

```
% Linearizace kolem bodu theta = pi

A = [0 1 0 0;
    0 -d/M b*m*g/M 0;
    0 0 0 1;
    0 -b*d/(M*L) -b*(m+M)*g/(M*L) 0];

B = [0; 1/M; 0; b*1/(M*L)];

C = [[1 0 0 0];[0 0 1 0]];

D = [0; 0];
inv_pend_linsys = ss(A, B, C, D)
```

inv_pend_linsys = A = x1 x2 1 0 x1 0 -0.2 -1.962 x2 x3 0 -0.1 5.886 x4 B = u1 x1 x2 0.2 х3 0 х4 0.1 C = x1 x2 x3 x4 1 0 0 0 у1 y2 D =

```
u1
y1 0
y2 0
```

Continuous-time state-space model.

```
% step(inv_pend_linsys)
 [p, z] = pzmap(inv_pend_linsys)
 p = 4 \times 1
    -2.4077
    -0.2336
     2.4414
   0×1 empty double column vector
 % controllability:
 rank(ctrb(inv_pend_linsys))
 ans = 4
 % observability
 rank(obsv(inv_pend_linsys))
 ans = 4
LQR controller:
 % state cost matrix
 global Q R S
                                                                                      % state cost matrix
 Q = [[1 \ 0 \ 0 \ 0]; [0 \ 1 \ 0 \ 0]; [0 \ 0 \ 10 \ 0]; [0 \ 0 \ 0 \ 1]];
 R = 0.01;
                                           % input cost matrix
 S = [[0]; [0]; [0]; [0]]
                                           % state input weight matrix
 S = 4 \times 1
      0
      0
      0
 global K
 [K, W, ~] = lqr(inv_pend_linsys, Q, R, S)
 K = 1 \times 4
   -10.0000 -22.7468 314.2601 123.4261
 W = 4 \times 4
             1.8596 -12.3426
                                -4.7192
     2.1747
             3.1025 -22.1220
     1.8596
                                -8.4797
   -12.3426 -22.1220 198.7696
                                75.6700
                               29.3021
    -4.7192
            -8.4797 75.6700
 inv_pend_sys_lqr = ss(A-B*K, B, C, D)
 inv_pend_sys_lqr =
```

```
A =
     x1 x2 x3
0 1 0
                    x4
x1
                       0
    2 4.349 -64.81 -24.69
0 0 0 1
x2
x3
     1 2.175 -25.54 -12.34
x4
   u1
x1
   0
x2 0.2
x3 0
x4 0.1
C =
   x1 x2 x3 x4
u1
y1 0
y2
   0
```

Continuous-time state-space model.

```
% stability of new system:
eig(A-B*K)
```

```
ans = 4×1 complex

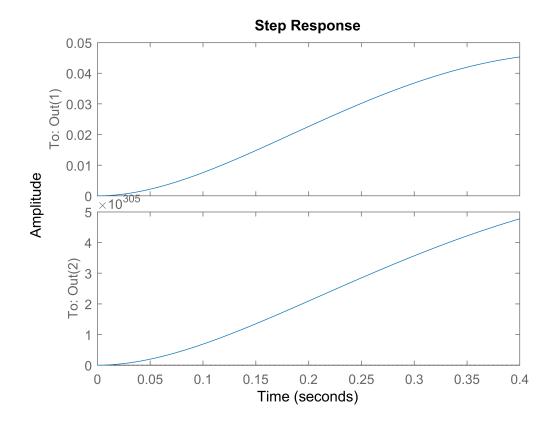
-2.5891 + 0.5460i

-2.5891 - 0.5460i

-1.5473 + 0.0000i

-1.2677 + 0.0000i
```

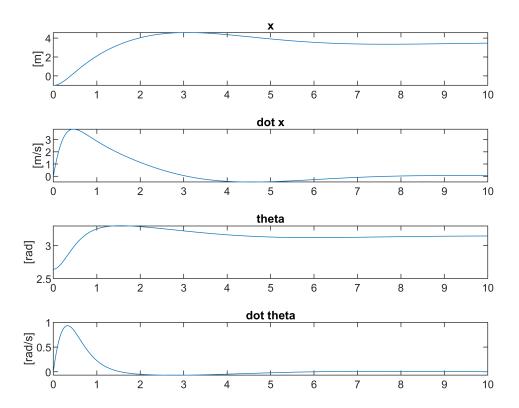
```
% !!! rescaling:
kr = -1./dcgain(inv_pend_sys_lqr);
kr(isinf(kr)) = realmax;  % subs inf for biggest double in matlab
inv_pend_sys_lqr.C = inv_pend_sys_lqr.C.*kr;
step(inv_pend_sys_lqr)
```



simulation with LQR controller:

```
tspan = 0:.001:10;
global x0
x0 = [-1; 0; pi-0.5; 0]; % initial condition
sp = [3.5; 0; pi; 0];
                          % set point
[t,x_1qr] = ode45(@(t,x)inv_pend(x, LQR(x, sp)),tspan,x0);
tiledlayout(4,1)
ax1 = nexttile;
plot(ax1,t,x_lqr(:, 1))
title(ax1,'x')
ylabel(ax1,'[m]')
ax2 = nexttile;
plot(ax2,t,x_lqr(:, 2))
title(ax2,'dot x')
ylabel(ax2,'[m/s]')
ax3 = nexttile;
plot(ax3,t,x_lqr(:, 3))
title(ax3,'theta')
ylabel(ax3,'[rad]')
ax4 = nexttile;
plot(ax4,t,x_lqr(:, 4))
```

title(ax4,'dot theta')
ylabel(ax4,'[rad/s]')



MPC:

discrete time model for MPC:

0

-5e-08

-9.999e-05

```
Ts = .001; % [s]
inv_pend_linsys_d = c2d(inv_pend_linsys, Ts, 'zoh')
inv_pend_linsys_d =
 A =
             x1
                        x2
                                   х3
  x1
                  0.0009999
                            -9.809e-07
                                        -3.27e-10
              0
                     0.9998
                             -0.001962
                                       -9.809e-07
  x2
```

0.001

1

0.005886

х3

х4

Calculation of optimal input for first moment:

```
global A_d B_d W_d
A_d = inv_pend_linsys_d.A;
B_d = inv_pend_linsys_d.B;
W_d = idare(A_d,B_d,Q,R,S)

W_d = 4x4

10<sup>5</sup> x

0.0218  0.0186  -0.1234  -0.0472
0.0186  0.0310  -0.2212  -0.0848
-0.1234  -0.2212  1.9878  0.7567
-0.0472  -0.0848  0.7567  0.2930
```

Simulation with non-linear model:

```
tspan = 0:Ts:10;
[t,x_mpc] = ode45(@(t,x)inv_pend(x, MPC(x, sp)),tspan,x0);
tiledlayout(4,1)

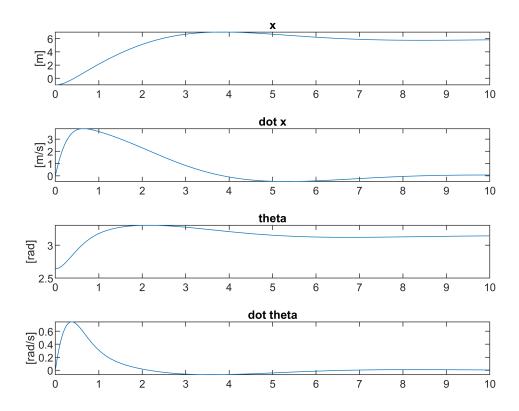
ax1 = nexttile;
plot(ax1,t,x_mpc(:, 1))
title(ax1,'x')
ylabel(ax1,'[m]')

ax2 = nexttile;
plot(ax2,t,x_mpc(:, 2))
title(ax2,'dot x')
ylabel(ax2,'[m/s]')

ax3 = nexttile;
plot(ax3,t,x_mpc(:, 3))
```

```
title(ax3,'theta')
ylabel(ax3,'[rad]')

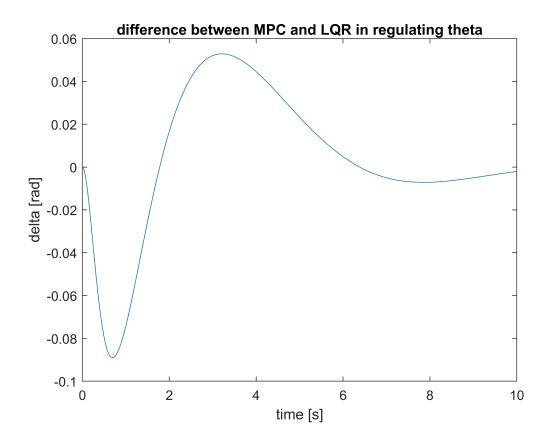
ax4 = nexttile;
plot(ax4,t,x_mpc(:, 4))
title(ax4,'dot theta')
ylabel(ax4,'[rad/s]')
```



Difference between LQR and MPC:

```
tiledlayout(1,1)

plot(t, x_mpc(:,3) - x_lqr(:,3))
title('difference between MPC and LQR in regulating theta')
xlabel('time [s]')
ylabel('delta [rad]')
```



Inverted pendulum function definition (for simulating non-linear model):

```
function dx = inv_pend(x, F)
   M = 5;
              % mass of the cart
              % mass of the pendulum
   m = 1;
   b = 1; % coefficient of the friction rotation
   g = -9.81; % acceleration of gravity
            % length of shoulder of pendulum
   1 = 2;
   Sx = sin(x(3));
   Cx = cos(x(3));
   denom = m*(1^2)*(M+m*(1-Cx^2));
   dx1 = x(2);
   dx2 = (-(m^2)*(1^2)*g*Cx*Sx + m*(1^2)*(m*1*(x(4)^2)*Sx) - b*x(2) + m*(1^2)*F)/denom;
   dx3 = x(4);
   dx4 = ((m+M)*m*g*1*Sx - m*1*Cx*(m*1*(x(4)^2)*Sx-b*x(2)) - m*1*Cx*F)/denom;
   dx = [dx1 dx2 dx3 dx4]';
end
```

LQR regulator function definition:

```
function u = LQR(x, sp)
    global K
    u = -K*(x-sp);
end
```

MPC regulator function definition: