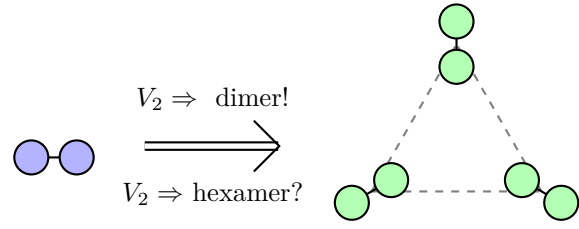


Aperitif

—“The structure of dynamical equations is desceptively simple, obfuscating the marvelous phenomena which they point us to in nature decades and conturies post their conception.”

- The goal is the investigation of N -body systems and how linked/disentangled they behave with respect to their subsystems
- The elementary instance we want to analyze—empirically, at first—is the celestial N -body problem with the trajectories of $M < N$ of its constituents being fixed (we aim from the start to generalize the text-book problem [1, 2] which deals with 3 objects of which 2 are fixed).
- Let us tackle the restricted 5-body problem and commence by formulating the pertinent equations of motion. We chose to fix the orbits of 4 of the 5 objects to circular trajectories about the origin, which is, the centre of mass.
- The coordinates and the velocity of the 5th object comprise the only degrees of freedom of the system, and we choose to start from a Lagrangean formulated in cartesian coordinates as we expect this approach to generalize straight-forwardly to larger N and M .



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- N point masses with $\{(m_i, \mathbf{r}_i) \text{ for } i = 1, \dots, N\}$ and a fixed time dependence for M of those objects: $\mathbf{r}_{i=1, \dots, M}(t) \equiv \boldsymbol{\rho}_i(t)$

- I suggest to use centre-of-mass coordinates:

$$\mathbf{R} = \frac{1}{M} \sum_{i=1}^N m_i \mathbf{r}_i \quad \text{and} \quad \tilde{\mathbf{r}}_i = \mathbf{r}_i - \mathbf{R} \quad \text{for } i = 1, \dots, N, \quad (\text{CM})$$

and

$$\tilde{\boldsymbol{\rho}}_i(t) = \rho_i \begin{pmatrix} \cos(\omega_i t) \\ \sin(\omega_i t) \\ 0 \end{pmatrix} \quad \text{for } i = 1, \dots, M. \quad (\text{FC})$$

$$\text{d) } \mathfrak{L} = T - V = \frac{m_5}{2} (\dot{\tilde{\mathbf{r}}}_5)^2 - \gamma \sum_{i=1}^4 \frac{m_5 m_i}{|\tilde{\mathbf{r}}_5 - \tilde{\boldsymbol{\rho}}_i|}$$



Derive the Hamiltonian as the Legendre transform of Lagrangean

$$\mathfrak{H} = \dot{\tilde{\mathbf{r}}}_5 \cdot \mathbf{p}_5 - \mathfrak{L}$$

and express it in terms of coordinates and conjugate momenta

$$\mathbf{p}_5 = \frac{\partial \mathfrak{L}}{\partial \dot{\tilde{\mathbf{r}}}_5}.$$

Subsequently, derive the equations of motion:

$$\dot{\tilde{\mathbf{r}}}_5 = \frac{\partial \mathfrak{H}}{\partial \mathbf{p}_5} \quad \text{and} \quad \dot{\mathbf{p}}_5 = -\frac{\partial \mathfrak{H}}{\partial \tilde{\mathbf{r}}_5}. \quad (\text{EQM})$$

Ecce, these six equations will look quite ugly but thankfully it will be the computer who will do the heavy lifting.

Primo piatto

For the general $(N - M)$ -body problem, the equations of motion are conveniently(?) written in matrix form. The vectors \mathbf{y} and \mathbf{f} have $2 \times \dim_{\text{space}} \times A$ components and we chose the letters to conform with [1]. Also note, that in the summation over j we need to consider all objects, i.e., the constrained ones, too, because they still exert a force on the unconstrained masses. For $\dim_{\text{space}} = 3$, the equations of motion read explicitly

$$\frac{d\mathbf{y}}{dt} := \frac{d}{dt} \begin{pmatrix} r_{x1} \\ r_{y1} \\ r_{z1} \\ \vdots \\ r_{xA} \\ r_{yA} \\ r_{zA} \\ p_{x1} \\ p_{y1} \\ p_{z1} \\ \vdots \\ p_{xA} \\ p_{yA} \\ p_{zA} \end{pmatrix} = \begin{pmatrix} \frac{\partial \mathfrak{H}}{\partial p_{x1}} \\ \vdots \\ \frac{\partial \mathfrak{H}}{\partial p_{zA}} \\ -\frac{\partial \mathfrak{H}}{\partial r_{x1}} \\ \vdots \\ -\frac{\partial \mathfrak{H}}{\partial r_{zA}} \end{pmatrix} = \begin{pmatrix} p_{x1}/m_1 \\ p_{y1}/m_1 \\ p_{z1}/m_1 \\ \vdots \\ p_{xN}/m_A \\ p_{yN}/m_A \\ p_{zN}/m_A \\ \gamma m_1 \sum_{j \neq 1}^N \frac{(r_{x1} - r_{xj})m_j}{|\mathbf{r}_1 - \mathbf{r}_j|^3} \\ \gamma m_1 \sum_{j \neq 1}^N \frac{(r_{y1} - r_{yj})m_j}{|\mathbf{r}_1 - \mathbf{r}_j|^3} \\ \vdots \\ \gamma m_N \sum_{j \neq A} \frac{(r_{zA} - r_{zj})m_j}{|\mathbf{r}_A - \mathbf{r}_j|^3} \end{pmatrix} = \begin{pmatrix} y_{3A+1}/m_1 \\ y_{3A+2}/m_1 \\ y_{3A+3}/m_1 \\ \vdots \\ \gamma m_1 \sum_{j \neq 1}^N \frac{(y_1 - y_{3j-2})m_j}{|(y_1 - y_{3j-2})^2 + (y_2 - y_{3j-1})^2 + (y_3 - y_{3j})^2|^{3/2}} \\ \vdots \\ \gamma m_1 \sum_{j \neq A} \frac{(y_{3A} - y_{3j})m_j}{|(y_{3A-2} - y_{3j-2})^2 + (y_{3A-1} - y_{3j-1})^2 + (y_{3A} - y_{3j})^2|^{3/2}} \end{pmatrix} =: \mathbf{f}(\mathbf{y}, t) \quad (\text{EQM})$$



Obtain $\mathbf{y}(t)$ for a temporal grid $t \in \{h, 2h, \dots, nh\}$ using a computer program which takes as input:

- the masses m_i of the N objects,
- the initial positions $\mathbf{r}_i(0)$ and velocities $\dot{\mathbf{r}}_i(0)$ of the A unconstrained objects,
- the fixed orbits $\{\mathbf{y}_i(t)\}$ with $i \in \{3A + 1, \dots\}$ of the M constrained objects, and
- the time-grid parameters h and n .

Nomenclature

$\tilde{\mathbf{r}}$	Bold symbols denote, if not stated otherwise, three-dimensional vectors, e.g., $\mathbf{r} = (x, y, z)$, the length/magnitude of a vector $r := \mathbf{r} $, and the tilde refers to vectors in the centre-of-mass frame.
N, M, A	Of N objects, the orbits of M are fixed which leaves $N - M := A$ trajectories for which we seek a solution.
dimer	With <i>dimer</i> , we refer to a pair of particles which are <i>loosely</i> bound, i.e., a small disturbance suffices to induce a transition from a trajectory on which the relative distance between them does not diverge when $t \rightarrow \infty$ to one on which it does.
M	total mass of the system, $M = \sum_{i=1}^N m_i$

References

[1] E. W. Schmid, G. Spitz, and W. Lösch, “The celestial mechanics three-body problem,” in *Theoretical Physics on the Personal Computer* (Springer Berlin Heidelberg, Berlin, Heidelberg, 1990) pp. 81–89.

[2] H. Geiges, “The three-body problem,” in *The Geometry of Celestial Mechanics*, London Mathematical Society Student Texts (Cambridge University Press, 2016) p. 77–100.