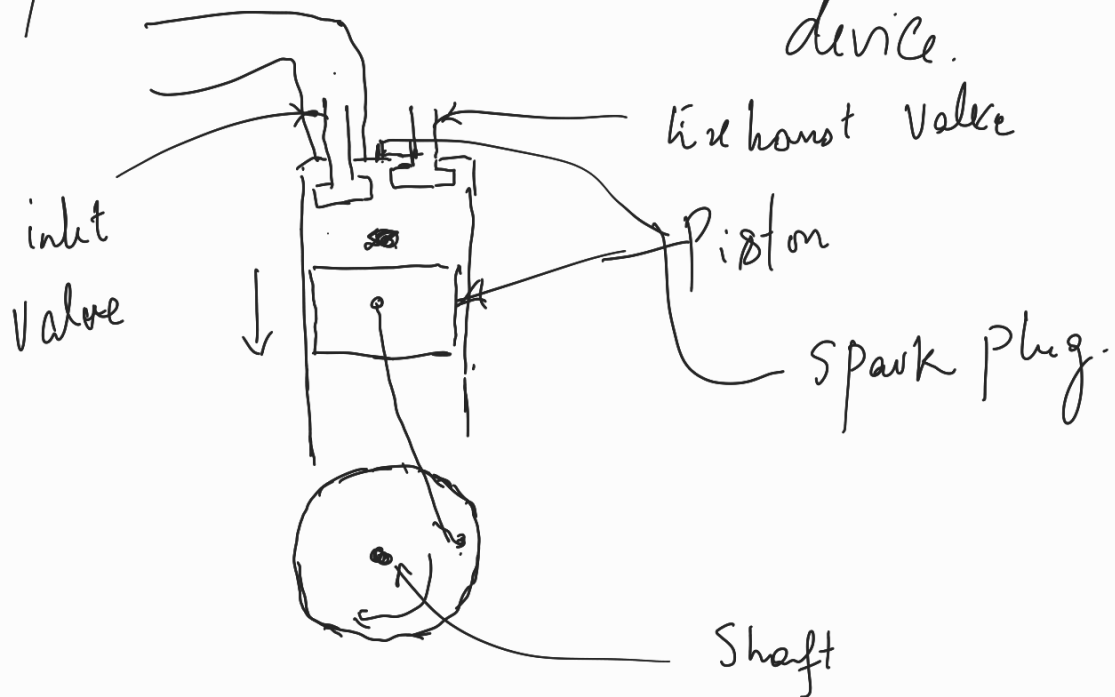


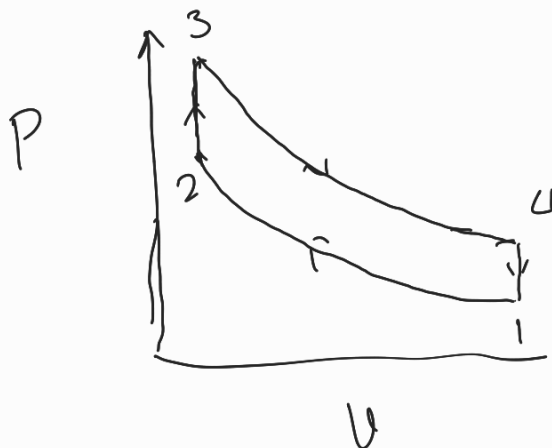
# Piston Engine & Propeller

Piston engine  $\longrightarrow$  Power producing device

Propeller  $\longrightarrow$  thrust producing device.



Top dead Center.



Otto Cycle

- 1-2 isentropic Comp.
- 2-3 Constant Vol. heat addition
- 3-4 Expansion (isentropic)
- 4-1 Constant vol. heat rejection.

1. Mark mechanism.

4 - stroke mechanism

1 power stroke for 2 revolutions of Crankshaft

✓ 2 - Stroke mechanism

1 power stroke for 1 revolution of Crankshaft.

Power developed by the engine.

$$P = \dot{m}_f Q_{cal} \eta_{th.} \quad (25\% - 35\%)$$

Thermal efficiency

Calorific value  
(44 MJ/kg)

mass flow rate of fuel.

$$P = \frac{K (\rho_{air} V_c N)}{60} \dot{m}_f Q_{cal} \eta_{th.}$$

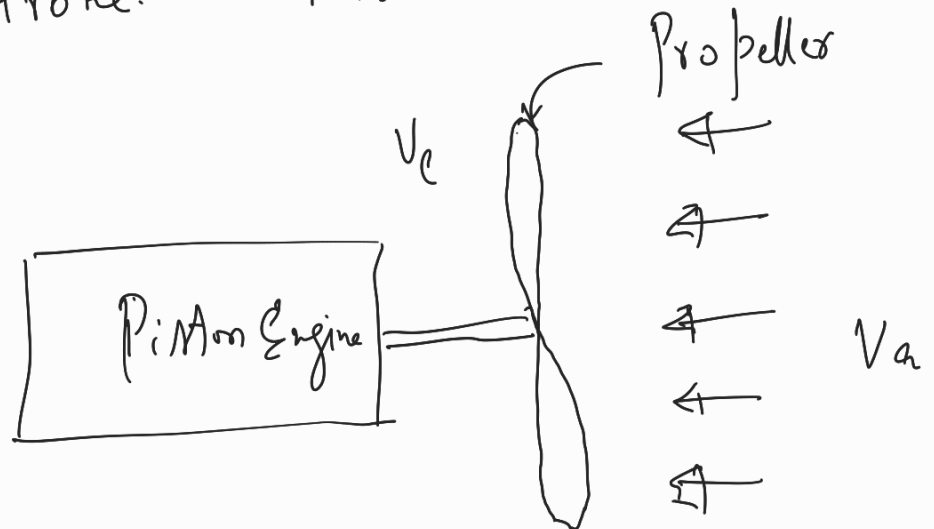
fuel air ratio

$\rho_{air}$  — density of air

$V_c$  — Vol. of the engine

$N$  — RPM of the engine

$K = 1$       0.5  
 $\downarrow$        $\downarrow$   
 2 Stroke      4 stroke



$$P \eta_{prop} = F \times V_a$$

$$F = \frac{P \eta_{prop}}{V_a}$$

Power  $\rightarrow$  Propeller efficiency 70-80%  
 $V_a$   $\rightarrow$   $V_a$  Velocity of Vehicle.

35 CC engine

Thrust      Specific      fuel      Consumption.

$$TSFC = \dot{m}_f / F$$