Abstract

Keywords: Fluid-structure interaction, transverse galloping.

1. Selection of the cross section

The cross section selection process for this study was based to three key criteria

- The cross section should have a bluff front face with sharp upstream corners for the flow to separate at the leading edges;
- As the proximity of shear layers to the body plays a vital role in creating C_u (?), the cross section should have a basic level of streamlining.
- The cross section should consist of a geometric profile in the afterbody, to inhibit or delay the shear layer reattachment.

The square cross section which has been widely used in galloping studies satisfies the first two selection criteria. Thus, a modification to the square cross section was done in order to meet the third criterion. Thus, in order to inhibit the shear layer reattachment, the top and bottom sides of the trailing edges of the square were tapered off and a hybrid cross section of a rectangle and a triangle (illustrated in figure 1), i.e, a pentagon was produced.

The inhibition of the shear layer can be varied systematically by varying one parameter, $\frac{d}{l}$; which was an advantage of this cross section. The ratio $\frac{d}{l}$ was varied from 1 to zero in increments of 0.25 where 1 is the square cross section and 0 is an isosceles triangle.

2. Static body results

Stationary time averaged C_y results were obtained for cross sections where $\frac{d}{l} = 1$, 0.75, 0.5, 0.25 and 0 using DNS at Re = 200. Table 1 shows the coefficients of the 7^{th} order curve fitting for each cross section. To achieve a

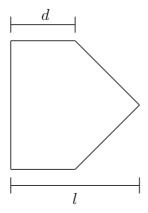


Figure 1: Illustration of the hybrid cross section (combination of a square and a triangle) obtained by tapering the afterbody of the square. The afterbody was changed by changing the ratio of $\frac{d}{l}$. Hence, data were obtained for $\frac{d}{l} = 1, 0.75, 0.5, 0.25$ and 0 were considered in this study.

better fit, piecewise interpolation using multiple 7th order polynomials were incorporated for a single cross section, giving more importance to accurately fitting the positive portion of the C_y curve; as the power transfer from the fluid to the body only occurs in this region.

The C_y vs. θ curves in figure 2 show the resultant of the piecewise curvefits obtained for each cross section. A shift of the peak value of C_y to the right can be observed as the $\frac{d}{l}$ decreases therefore, the peak C_y occurs at higher induced angles. The overall trends of this behaviour agrees with trends of ? where the peak C_y value was shifted to higher induced angles when reattachment was delayed on a trapezoidal body. The peak value of C_y occurs at high induced velocities as $\frac{d}{l}$ is decreased because θ is proportional to the transverse velocity of the body via $\tan \theta = \frac{\dot{y}}{U}$. Thus, these bodies with a short straight section, or small $\frac{d}{l}$, satisfy one of the three conditions required to optimize the power transfer.

A negative region could be observed on the C_y vs. θ curves where $\frac{d}{l} \leq 0.25$. In this region C_y decreases as θ is increased and the increases after reaching a minimum, non-zero value of θ . The presence of this region indicates an unfavourable power transfer, i.e. power transferred from body to the fluid as the direction of the force and velocity vectors are out of phase. This implies that at low induced angles (when the \dot{y} is low), power transfer is from the body to the fluid and as the transverse velocity increases power transfers from fluid to the body. Thus, it means that the direction of power transfer

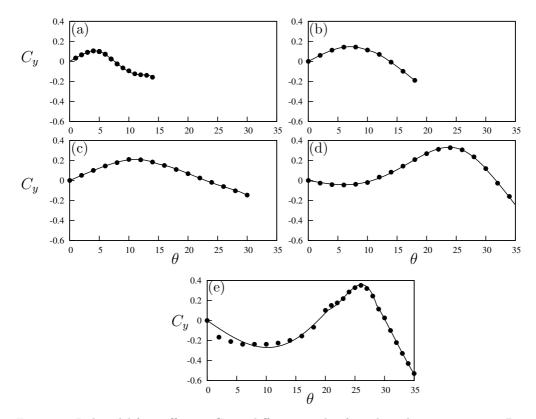


Figure 2: Induced lift coefficient C_y at different angles for selected cross sections. Data presented for cross sections, (a) square, (b) $\frac{d}{l} = 0.75$, (c) $\frac{d}{l} = 0.5$, (d) $\frac{d}{l} = 0.25$ and (e) triangle. Points (\bullet) are predicted from the static body simulations and the curves are the compound 7^{th} order polynomials.

$\frac{d}{l}$	a_1	a_3	a_5	a_7	Overlap range
0	-2.30617	-269.075	-59.2929	4.74389	
	-5.08342	-56.5390	-160.505	-105.773	$20.5^{\circ} - 23.5^{\circ}$
	4.40685	19.9213	22.8894	7.68556	$28.6^{\circ} - 28.7^{\circ}$
0.25	-0.605146	-19.4346	-82.4463	-94.4226	
	2.50538	9.91021	10.2712	3.94112	$30.1^{\circ} - 30.2^{\circ}$
0.5	1.44734	4.83885	-166.900	-983.072	
0.0					$14^{\circ} - 16^{\circ}$
	1.51455e	15.8476	52.5465	62.8067	
0.75	1.76938	35.2630	-345.562	-10072.7	11.03° – 11.11°
	1.77553	43.0120	262.983	638.484	11.00 11.11

Table 1: Coefficient values used in the 7th order interpolation polynomial at Re = 200. Data present for $\frac{d}{l} = 0 - 0.75$ at increments of 0.25. Multiple polynomials were used to attain a better fit. The plot of the compound fit is presented in figure 2. The "Overlap" range refers to the range of angles which the transition of the two polynomials take place.

can be different at different points in a particular oscillation cycle, which will be further discussed in section Section.

3. QSS Mean power output

Predictions of the mean power output of these different cross sections were obtained using the QSS model using the curve fits presented in figure 2 as inputs. Figure 3 shows the mean power vs. Π_2 for different cross sections namely $\frac{d}{l} = 1, 0.75, 0.5, 0.25$ and 0. The cross sections are divided into two classes; high $(\frac{d}{l} > 0.25)$ and low $(\frac{d}{l} \le 0.25)$. As Π_2 increases the mean power increases, peaks and reduces. For high $\frac{d}{l}$, the overall shape of the curves is similar, however as $\frac{d}{l}$ is decreased, the amount of power increases. For low $\frac{d}{l}$,

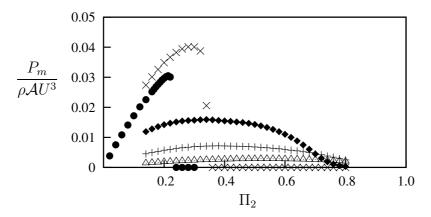


Figure 3: Dimensionless mean power obtained using QSS model as a function of Π_2 . Data presented for five selected cross sections, square (\triangle), $\frac{d}{l} = 0.75$ (+), $\frac{d}{l} = 0.5$ (\spadesuit), $\frac{d}{l} = 0.25$ (\times) and triangle (\blacksquare) at Re = 200, $\Pi_1 = 100$.

the overall curve shape is markedly different; power first increases with Π_2 , then peaks, and then drops dramatically. The power extracted also appears to decrease with a decrease in $\frac{d}{l}$. Furthermore, negative regions of the C_y vs. θ curves in figure 2 appear in the low $\frac{d}{l}$ cases. The change in the trend of power, and the appearance of a negative region in the C_y , for the low $\frac{d}{l}$ cases clearly indicates that there is a distinct change in the flow structure for these cases which is discussed in section .

4. Flow characteristics at low $\frac{d}{l}$ cases

The analysis of the mean power and the static body results shows an indication of a significant change in the flow structure at low $\frac{d}{l}$ cases. As the variation of the mean C_y is the sole input of the fluid dynamics to the QSS model, the distinct features in the C_y vs. θ curves provides a good indication of the change in flow structure which results in the change in mean power discussed in section 3. The main distinct feature between the high and low $\frac{d}{l}$ cases is the negative region present at low $\frac{d}{l}$ in the C_y vs. θ curves. Therefore, it was of interest to investigate the cause of this region. The isosceles triangle $(\frac{d}{l} = 0)$ is taken as the cross section of investigation as it produced the largest negative region out of the cross section considered in this study.

4.1. Surface pressure

The driving force of galloping F_y , which is the induced force is created as a result of the pressure difference created between the upper and lower sides of the body, due to the induced angle of attack occurred as a result of the horizontal movement of the flow and the transverse movement of the body. These pressure fields are created as a result of the relative proximities of the shear layer of the respective sides as discussed in **KJ: Put the reference of the theory section **. Hence, the pressure data of the time averaged flows of the stationary cross section $(\frac{d}{l} = 0)$ is analysed here.

The driving force of galloping is the induced force F_y created as a result of the freestream velocity of the fluid and the transverse velocity of the body. As discussed in section ?? the pressure difference of the upper and lower sides of the body (figure ??) creates this induced force as a result of the relative proximities of the shear layers to the respective sides. Thus, here, surface pressure data on the time averaged flows on the stationary cross section is analysed.

Time averaged (to filter out the influence of vortex shedding) surface pressure data on the top and bottom surfaces of the cross sections at $\theta = 4^{\circ}$, $\theta = 16^{\circ}$ and $\theta = 21^{\circ}$ were obtained for the isosceles triangle. These angles correspond to the regions of the C_y vs θ curve of the triangle where: C_y is negative, but increasing in magnitude; C_y is negative, but decreasing in magnitude and C_y is significantly positive.

Figure 5 shows the surface pressure of the top and bottom surfaces of the body $(\frac{d}{l} = 0)$ as a function of the distance from the leading edge. At $\theta = 4^{\circ}$, the pressure on the bottom of the body is greater than the top at practically all distances. As a result, a pressure difference is created and a force is generated in the upward direction which according to the sign convention in figure ??, is against the velocity of the body, hence giving a negative C_y .

As θ is increased to 16°, (figure 5 (b)) the gap between the surface pressure at the leading edge of the top and the bottom sides reduces. For small distances downstream from the leading edge, the pressure on the top surface is greater than that on the bottom. This effect results in a reduction of the magnitude of C_y (although it is still negative).

As θ is further increased to 21°, (figure 5 (c)) the surface pressure on the top side becomes greater than the bottom over the majority of the body. Therefore, the net effect of the pressure difference is a positive C_y which is the driving force F_y , now in phase with the velocity of the body.

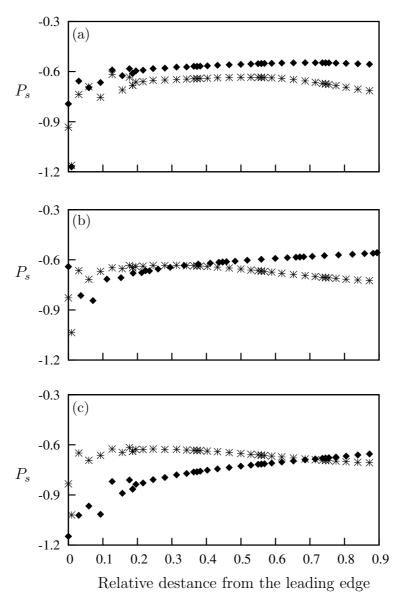


Figure 4: Surface pressure of top (**) and bottom (\spadesuit) surfaces of the static triangular cross section at (a) $\theta = 4^{\circ}$, (b) $\theta = 16^{\circ}$ and (c) $\theta = 21^{\circ}$ A clear pressure difference is visible between the top and bottom surfaces. The top surface comparatively has more negative pressure compared to the bottom surface and reduces as θ is increased and vice versa occurs at the bottom surface. Thus, initially the effective force is upwards which results in a negative C_y . The effective C_y becomes positive as θ is increased.

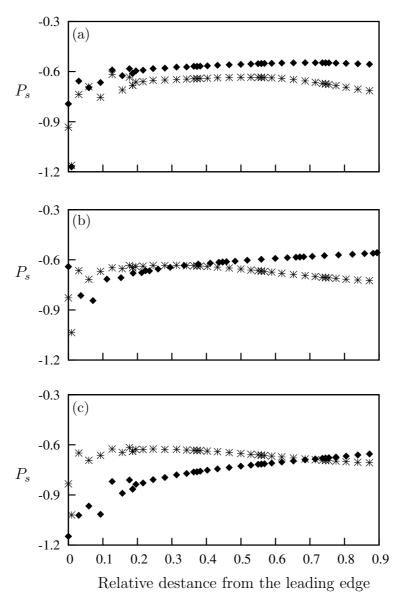


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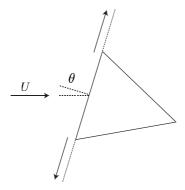


Figure 6: Illustration of the lines along which the flow velocity magnitudes have been extracted. The data have been extracted along a line starting from the separation points in the outward direction (shown with arrows) for the top and bottom surfaces.

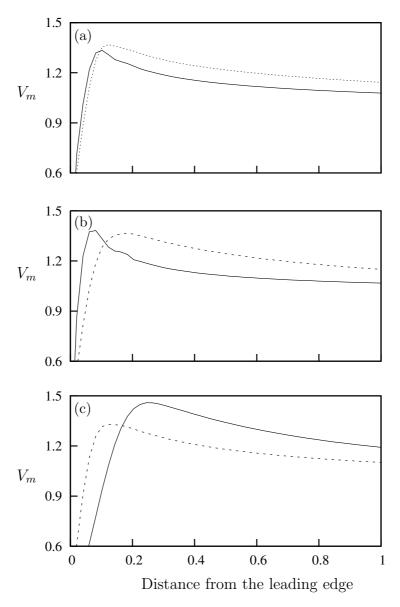


Figure 7: Velocity magnitudes of the flow along a line parallel to the front surface spreading towards top (---) and bottom (——) boundaries (figure 6). These two lines (for the top and bottom surfaces) start from the top and bottom leading edges of the triangular cross section. Data present (a) $\alpha = 4^{\circ}$, (b) $\alpha = 16^{\circ}$ and (c) $\alpha = 21^{\circ}$.

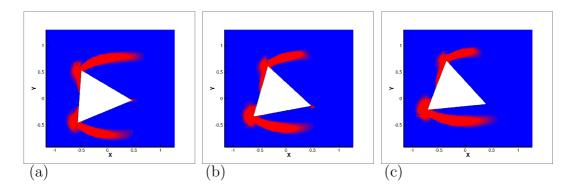


Figure 8: Contours of the magnitude of the shear strain rate of time averaged flow field on the stationary isosceles triangle $(\frac{d}{l}=0)$ at Re=200 at different incidence angles. (a) 4° (negative value of C_y that is further decreasing with increasing θ), (b) 16° (negative value of C_y that is increasing with increasing θ) and (c) 21° (a significantly positive value of C_y). The bottom shear layer comes closer to the bottom wall and as the angle of incidence increases.

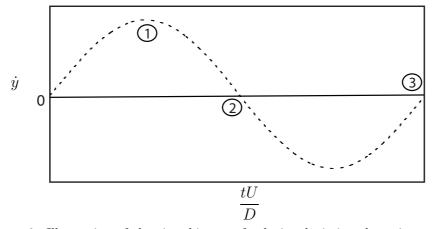


Figure 9: Illustration of the time history of velocity depicting the points considered to obtained time averaged stream traces. The points considered are: point 1 where \dot{y} is maximum, point 2 where \dot{y} is close to zero with a negative gradient and point 3 where \dot{y} is close to zero with a positive gradient.

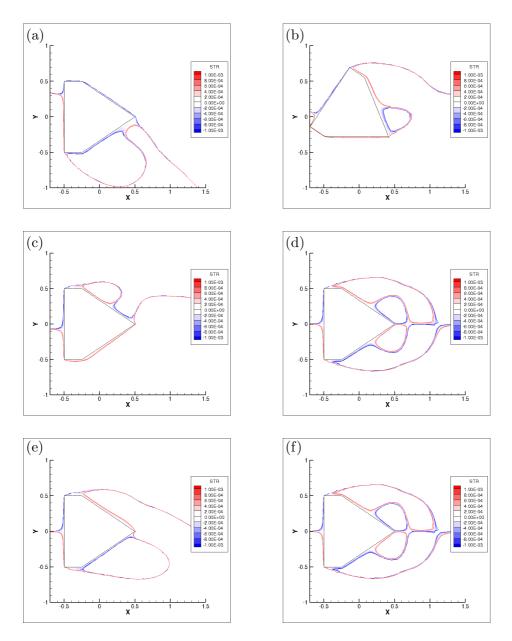


Figure 10: Time averaged stream functions of stationary and oscillating flow-fields of the hybrid cross section ($\frac{d}{l}=0.25$), averaged over a vortex shedding cycle. (a), (c) and (e) are the averaged stream functions of the oscillating case at $\frac{tU}{D}=2295.763$ (point 1), $\frac{tU}{D}=2305.897$ (point 2) and $\frac{tU}{D}=2325.870$ (point 3) . (b), (d) and (f) are the stream functions of the flow field of the stationary body corresponding to the induced angles of (a), (c) and (e).