## MCEN90018 Advanced Fluid Dynamics

## Assignment 3 - Signal Analysis

Assignment to be completed in groups of two, and to be submitted by 11:59 pm on Sunday 28th May 2017. Total available marks are 112. The last 2 questions (Bonus questions) do not need to be attempted to obtain an H1 (a maximum of 92/112 are obtainable without attempting this question).

## Part I - Signal analysis

A boundary layer traverse has been performed in a turbulent boundary layer at  $Re_{\tau}=14\,000$ . A single-normal hotwire has been traversed at 40 positions between y=0.27 mm and  $y\approx 450$  mm. Simultaneously a flush mounted hotfilm sensor has also been sampled (which provides the fluctuating friction velocity - or the fluctuating velocity one wall unit from the surface). The experimental set-up can be viewed in slide 38 of lecture 13 (only the central hotfilm, immediately below the traversing hotwire is used here). The data are available in the following zipped directory data with the file names u\_hf\_yposY.bin for the hotfilm and u\_hw\_yposY.bin for the traversing hotwire probe, where 1 < Y < 40 is an integer representing the y measurement station. The data are stored as floating point double precision velocities (all data have been calibrated already). Both signals were sampled at  $f_s=10 \mathrm{kHz}$  for 30 seconds. The boundary layer thickness  $\delta=0.326$  m. The wall normal y locations (in mm) for the 40 measurement stations are given by the file y.txt

- 1. The hot-film sensor is unreliable for frequencies greater than 100Hz. Implement a spectral low-pass cut-off filter on the hotfilm for Y=3. Plot a short time-series comparing the original and low-pass filtered (low frequency) hotfilm data for 1 s < t < 2 s.
- 2. Implement a spectral high-pass cut-off filter on the hotfilm for Y=3. Plot a short time-series comparing the original (with mean subtracted) and high-pass filtered (high frequency) hotfilm data for 1 s < t < 1.4 s.

  2 Marks
- 3. Compute the cross-correlation coefficient between the low pass hot-film signal and the similarly low pass filtered traversing hot-wire sensor at Y = 3. Use a long-hand approach initially,

$$R(\Delta t) = \frac{1}{N} \sum_{m=1}^{N} \frac{u_{hf}(t)u_{hw}(t + \Delta t)}{\sigma_{u_{hf}}\sigma_{u_{hw}}}$$

where  $u_{hf}$  and  $u_{hw}$  are the appropriately filtered fluctuating components from the hotfilm and hotwire respectively, and  $\sigma_{u_{hf}}$  and  $\sigma_{u_{hw}}$  are the standard deviation of the two signals. Taylors hypothesis can be used to convert from the time to the space domain,

$$x = -U_c t$$

where  $U_c$  is the convection velocity of turbulent fluctuations at the Yth position (taken to be the local mean as measured by the hotwire). Plot the cross-correlation coefficient R vs  $\Delta x/\delta$  from  $-3 < \Delta x/\delta < 3$ .

10 Marks

- 4. Calculate the same cross-correlation spectrally (using FFTs) and show that you obtain the same result (do not use xcorr for this I would like you to code your self). Note that the FFT method will return R for values from  $\Delta t = -N\delta t/2 + 1 : N\delta t/2$ , where N is the total length of the signal and  $\delta t$  is the sample interval  $1/f_s$ . Use the tic and toc commands in Matlab to compare the run-times for the long-hand and spectral approach. Comment on this.
- 5. Now you are satisfied that your (faster) spectral code works, calculate the cross-correlation spectrally between the high pass filtered hot-film signal and the similarly high pass filtered traversing hot-wire sensor at Y=3 (the two sensors are very close here, less than 0.5 mm seperating them vertically). Compare the maximum correlation values between the low-pass and high-pass filtered results, comment on the implications of this (and why we might choose to discard frequency content beyond 100Hz from the hot-film sensor).

  5 Marks
- 6. Now compute the cross-correlation coefficient for all Y positions for the low-pass filtered (low frequency) data, and plot isocontours of R as a function of  $\Delta x$  and y. Comment on the what we can learn from this figure in terms of the structural composition of the turbulent boundary layer.

  10 Marks
- 7. Produce a conditional average of the the hot-wire measured velocity for Y=20 based on the occurrence of a positive temporal velocity gradient  $du_{\tau}/dt>0$  indicated by the spectrally low-pass filtered hot-film signal from question 1.

10 Marks

8. Compute the power spectral density for the traversing hot-wire for position Y = 20 (using a single 30 s burst of data). Initially use no overlapping or intervals. Plot the pre-multiplied energy spectrum as a function of frequency and as a sanity check, show that,

$$\int_0^\infty \Phi(f)df = \frac{1}{N} \sum_{m=1}^N u_m^2$$

Comment on the sampling frequency - is it sufficient?

10 Marks

9. Your spectrum is still hairy. There are 10 bursts of 30 second of data at Y = 20 this time sampled at 30kHz in the same directory (named u\_hw\_yposY\_burstB.bin, where B is the burst number). Choose a suitable interval (such that you still capture most of the low frequency information) and attempt to calculate a more converged power spectral density. You can use over-lapping.

10 Marks

## Part II - Resistance of a developing smooth and rough-wall boundary layer

The mean velocity profile for a developing turbulent boundary layer on a smooth or rough flat plat can be written as,

$$\frac{u(y)}{U_{\tau}} = \frac{1}{\kappa} \log(\frac{yU_{\tau}}{\nu}) + A - \Delta U^{+} - \frac{1}{3\kappa} \eta^{3} + \frac{\Pi}{\kappa} 2\eta^{2} (3 - 2\eta)$$

Where  $\eta = y/\delta$ , A = 5,  $\kappa = 0.41$ ,  $\Delta U^+$  is the Hama roughness function which is 0 for the smooth wall or a function of the roughness length  $k_s$  for rough surfaces. Using this

equation we can obtain a prediction for the skin friction coefficient and the boundary layer thickness on the back of a submarine.

We know that the momentum thickness is given by,

$$\theta = \int_0^\delta \frac{U}{U_\infty} \left( 1 - \frac{U}{U_\infty} \right) dz = \frac{\nu}{U_\tau} \int_0^{\delta^+} \left( \frac{u^+}{S} - \frac{u^{+2}}{S^2} \right) dz^+$$

Where  $S = U_{\infty}/U_{\tau}$  and is given for the above functional form (for  $y = \delta$ ) by,

$$S = \frac{1}{\kappa} \log \delta^{+} + A - \Delta U^{+} - \frac{1}{3\kappa} + \frac{2\Pi}{\kappa},$$

Consider the example of a **smooth** submarine with a speed of 30 knots and a kinematic viscosity of  $8.97 \times 10^{-7}$  m<sup>2</sup> s<sup>-1</sup>

- 1. For a range of logarithmically spaced  $\delta^+$  from 100 to 1000000, write a Matlab script to evaluate numerically the momentum thickness  $\theta$  for a smooth surface at each  $\delta^+$ . Plot  $Re_{\tau} = \delta^+$  vs  $Re_{\theta}$ .
- 2. Using the relationship for the local skin friction coefficient,

$$C_f = \frac{\tau_w}{\frac{1}{2}\rho U_{\infty}^2} = \frac{2}{S^2}.$$

Also plot  $Re_{\tau}$  vs.  $C_f$  for the same smooth surface.

5 Marks

3. Bonus Questions At the moment we lack the information about where along the length of the submarine a particular  $\delta^+$  and therefore  $C_f$  is attained. We need to relate the  $Re_{\theta}$  to  $Re_x$ , which is very easy using von Kármán momentum integral equation,

$$\frac{d\theta}{dx} = \frac{dRe_{\theta}}{dRe_{x}} = \frac{C_{f}}{2},$$

Thus we can calculate  $Re_x$  from

$$Re_x = \int \frac{2}{C_f} dR e_\theta$$

Plot the skin friction coefficient as a function of  $Re_x$ , and hence estimate  $C_f$  and the boundary layer thickness (in m) towards the back of a smooth submarine at x = 115 m (you can assume that the flat plate approximates the boundary layer development on the submarine). **10 Marks** 

4. Bonus Questions Assuming that for fully rough surfaces,

$$\Delta U^+ = \frac{1}{\kappa} \log k_s^+ + A - 8.5$$

Plot (on top of the smooth wall result obtained above) the skin friction coefficient as a function of  $Re_x$  for the case where the submarine has a surface roughness with a  $k_s = 325 \ \mu\text{m}$ . Compare the  $C_f$  and the boundary layer thickness (in m) towards the back of a submarine at  $x = 115 \ \text{m}$  for the rough and smooth surfaces.

10 Marks