A mass (body) experiences au inortal & load proportional to it's acceleration and apposite to the direction of acceleration

D'Alembert & Poinciple.

$$\sum F_x + ma_x = 0$$
 Eqn. of dynamic equilibres.

Load factor The load factor (in a particular direction) is the total force.

(acting in that dire.) less the gravitational and inertial force (forces associated with maps) divided by useight.

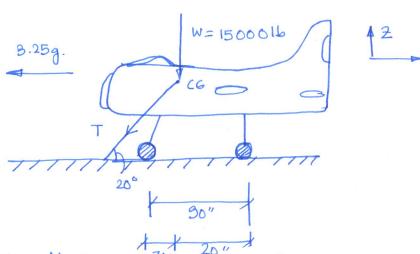
ZFy - may =0 Eqn. of dynamic equilm.

(total force not containing not & inertial force)

Load factor in y-dire =  $\sum_{w}^{F_{y}} / \omega$ ,

$$=\frac{\partial n(q+\alpha y)}{w}=\frac{w(1+\alpha y)}{w}=\frac{1+\alpha y}{q}.$$

Example

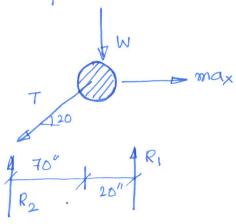


The aircraft has a gross weight of w= 1500016. It is being lausched from an aircraft carrier. The trusum o T is the cable acts through the airplane C.G. and is sufficient to the give the airplane a forward acceleration 3.25g.

Find:

- (a) tension in the cable
- (b) Reaction bouds on the landing gear R, & R2
- (c) Load factor mx and m2

FBD



Egns of dynamic equilm

R1+R2 - T070 -W = 0

=> 
$$T(an20) = \frac{W}{a} \cdot ax = 15000 \times 3.25$$

Load factors

$$\mathfrak{N}_{\mathcal{X}} = -\frac{T(\mathfrak{m}_{20})}{W} = -\frac{W}{9} \cdot \frac{3.25g}{W} \Rightarrow R_{1} = 25467 \text{ lb}$$

$$= -3.25 \qquad R_{2} = 7276 \text{ lb}$$

$$\mathfrak{N}_2 = \frac{R_1 + R_2 - T(\mathfrak{O}_{70})}{W} = \frac{W}{W} = 1$$

## Procedure for computing apparant at using load factors:

An airplane is maneuvering so that following load factors are compared on x = -1.5 and  $m_2 = 2$ . An item weighing 101b is suspended from a wore. If the load factors genain constant for a long enough time. find the tension in the wore