

Quiz 1

AE220 Aerospace Structural Mechanics

Date: February 3, 2014

Name:

Roll No:

1. An aircraft is flying with a lift, L , equal to the weight of the aircraft, W when a gust with lift $\Delta L = 1.5W$ strikes the aircraft. The resultant of this gust produced lift is 90 inches aft of the C.G. I_{pitch} for the entire plane is 2.25×10^7 lb-in-sec² computed about the aircraft C.G.

Before the gust strikes, a load sensor located on the boom attachment point senses a shear force V of 1000 lb. The weight of the boom is 1000 lb. Its pitch mass moment of inertia about its own G.G. is 3000,000 lb-in-sec².

- Compute the aircraft load factor, n_z .
- Compute the value of V (and direction) at the sensor when the gust strikes.
- Compute the bending moment, M , (and direction) at the sensor when the gust strikes.

Marks 12

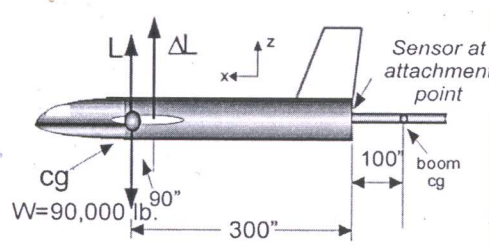


Figure 1:

(a) Load factor = $n_z = \frac{\sum F_{external} \leftarrow \text{(excluding wt \& weight force)}}{W} = \frac{L + \Delta L}{W} = \frac{W + 1.5W}{W} = 2.5$ — ①

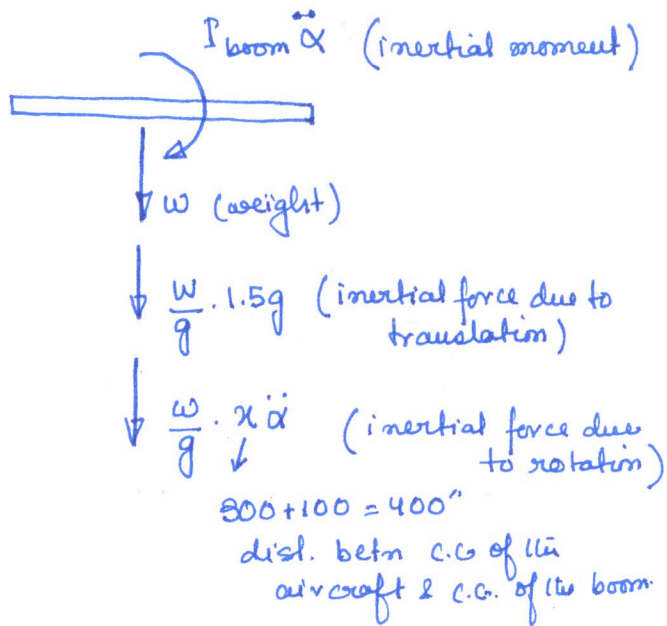
(b) Unbalance moment about c.g. due to gust = $\Delta L \times 90'' = 1.5 \times 90W$ lb-in — ①

Angular acceler about CG due to the gust = $\frac{1.5 \times 90 \times W}{I_{pitch}} = \frac{1.5 \times 90 \times 90000}{2.25 \times 10^7} = 0.54 \text{ rad/sec}^2$ — ①

Translational acceler. due to the gust = $\frac{\Delta L g}{W} = 1.5g \text{ in/sec}^2$ — ①

Weight of the boom = shear force when the gust is not acting = $w = 1000 \text{ lb}$ — ①

Free body diagram of the boom



Shear force at the sensor

$$\text{due to after the gust strikes} = w + \frac{w}{g} \cdot 1.5g + \frac{w}{g} \cdot 400 \ddot{\alpha}$$

$$= 1000 + 1.5 \times 1000 + \frac{1000}{32.2 \times 12} \times 400 \times 0.54$$

$$= 2500 + 559 \text{ lb}$$

$$= 3059 \text{ lb} \quad \text{--- (4)}$$

(c) Bending moment at the sensor

$$= 3059 \times 100 + I_{boom} \ddot{\alpha}$$

$$= 305900 + 3 \times 10^6 \times 0.54 \text{ lb-in}$$

$$= 467900 \text{ lb-in}$$

$$\text{--- (4)}$$

3. Write (maximum two sentences each) the main structural function of (i) *skin*, (ii) *stringers*, and (iii) *ribs*. No marks will be given for answers longer than two sentences. **Marks 4**

Skin: resist shear stress due to torsion — (1)

Stringers: resist normal stress due to bending — (1)

Ribs: Reduces effective length of the panels, increasing buckling stress resistance. Resists hoop stress — (2)