

# Visual-Servoing Control of a Soft Robot for Maskless Head and Neck Cancer Radiotherapy

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**Abstract**—This work presents an on-going investigation of the control of a pneumatic soft-robot actuator targeted towards accurate patient positioning systems in maskless head and neck cancer radiotherapy. We employ two RGB-D sensors in a sensor fusion scheme to better estimate a patient’s head pitch motion; a system identification prediction error model is used to obtain a linear time invariant state space model. We use the model and linear quadratic Gaussian feedback controller to manipulate the patient head position based on sensed head pitch motion.

## I. INTRODUCTION

This paper presents a continuation of our investigation of an image-guided soft robot patient positioning system for use in head and neck (H&N) cancer radiotherapy (RT). In 2014, over 1.6 million patients developed pharynx and oral cavity cancers in the United States which led to over 580,000 deaths [1]. Typical H&N cancer treatment involves intensity-modulated radiotherapy (IMRT), which delivers high potent dose to tumors while simultaneously minimizing dose to adjacent critical organs such as spinal cord, parotids glands, and optical nerves. Typically, a patient lies on a 6-DOF movable treatment couch, and laser or image-guidance systems are used to ensure the patient is in the proper position. A linear accelerator in conjunction with multileaf collimators adjusts a radiation beam to the shape of the patient’s tumor. The beam is typically directed to a tumor location by accurately positioning the beam generator and/or moving the couch.

IMRT requires accurate patient positioning. In a geometric miss, for instance, highly conformal potent dose increases risk of underdose to tumors or undesirable high dose to critical organs and nearby tissues. An examination of dosimetric effects on patient displacement, collimator and beam angle misalignment during IMRT showed errors as small as 3-mm in anterior-posterior direction caused 38% decrease in minimum target dose or 41% increase in the maximum spinal cord dose [2]. Image-guided radiotherapy (IGRT) has made progress in improving intensity modulated radiotherapy (IMRT) accuracy while reducing set-up times [3], [4], [5]. However, current IGRT practices focus on using images acquired before treatment to confirm beam placement [6]. The discomfort caused by head masks in prolonged IMRT treatment can increase patients voluntary and involuntary

motion. Studies show that translational errors caused by patient motion can be larger than 6mm, and rotational errors can be as high as 2° [7]. Current motion-tracking systems, such as Cyberknife and Novalis are not compatible with conventional linear particle accelerators used at the majority of cancer centers. Moreover, these two systems are limited to assuming the patient’s body is a rigid one during motion tracking and compensation.

In this work, we present extensions to the previous work viz., raising or lowering an example patient’s head, lying in a supine position, to a desired height above a treatment table [8]. This paper contributes better vision tracking and localization methods, and uses an optimal control network that improves the one degree of freedom motion control.

## II. HARDWARE OVERVIEW

The system consists of a single inflatable air bladder (IAB), a mannequin head and a neck/torso motion simulator, two RGB-D sensors, two pneumatic valve actuators controlled by custom-built current regulators, and a National Instruments myRIO microcontroller. The RGB-D sensors are mounted directly above the head for raw head position and velocity measurements while local kalman filters (KFs) estimate the head position and velocity at each sensor site. The sensor estimates are aggregated at a central fusion site using a track-to-track KF-based sensor fusion algorithm. The soft robot actuation mechanism combines a inflatable air bladder (19” x 12”) made of lightweight, durable and deformable polyester and PVC, two current-controlled proportional solenoid valves (Model PVQ33-5G-23-01N, SMC Co., Tokyo, Japan), and a pair of silicone rubber tubes (attached to a T-port connector at the orifice of the IAB) in order to convey air in/out of the IAB. A 1HP air compressor supplied regulated air at 30 psi to the inlet actuating valve, while an interconnection of a 60W micro-diaphragm pump and a PVQ valve removed air from the outlet terminal of the IAB. The diaphragm pump creates the minimal operational differential pressure required by the outlet valve. We apply the fusion result in a new robust control law for the pneumatic actuator valves, thereby regulating air pressure within the IAB and moving the patient’s head as desired. The real-time controller was deployed on a National Instruments myRIO embedded system running LabVIEW 2015. The LabVIEW algorithms were processed within a Windows 7 virtualbox running on the Ubuntu host workstation. The system set-up is described in details in [8, §1]

We use a Kinect Xbox 360, and a Kinect for Windows v2 sensor to estimate head position and velocity. The two

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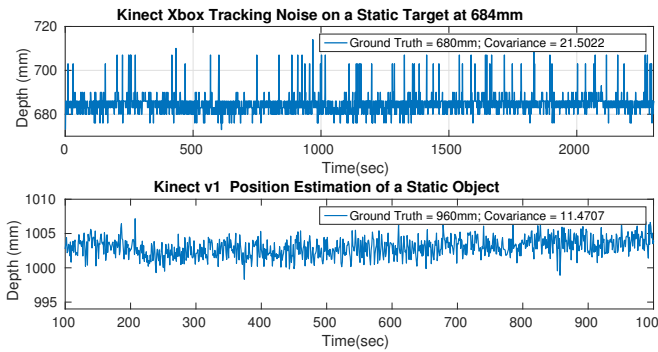


Fig. 1: Noise floor of Kinect Xbox Sensor vs. Kinect v1 Sensor

sensors use different electronic perception technologies to determine distance of an object from the camera origin. They therefore have different lateral and range resolutions as well as different noise characteristics. Image processing for both cameras is executed on a 22GB RAM mobile workstation with Intel Core i7-4800MQ processor running 64-bit Ubuntu Trusty on a Linux 4.04 kernel.

### III. OBJECT DETECTION

We perform recursive filter estimations of the Xbox observations and improve position estimates by using an additional sensor to better localize tracked features. We add the Kinect v2 sensor (henceforth called the v2 sensor), based on the time-of-flight (ToF) electronic perception principle, to the perception system used in [8]. In ToF, light pulses illuminate a scene and depth is calculated by determining the round trip of reflected photons to a photo-sensitive silicon array via the phase shift of the returned light signals. The active infra-red reduces the dependence on ambient lighting [9] and this sensor has a higher spatial depth resolution of  $512 \times 424$  pixels at 30Hz interactive rate, compared to the Xbox's  $320 \times 240$  pixels [10]. To minimize the noise due to the limited sensor resolution, the v2 has in-built noise improvement capabilities including 1) temporal and spatial averaging in the neighborhood of  $N$  pixels to improve the range resolution; 2) minimizing the aliasing effect from “unambiguous distance range”; 3) use of multiple exposure settings to minimize saturation arising from highly reflective objects and non-reflective surfaces; 4) to properly delineate the occlusions on an image’s contours, the foreground image is first binarized before the boundaries of the image are extracted through an erosion morphological operation; the eroded image is then gray-level dilated to construct an output range which reduces motion artifacts; and 5) minimization of ambient lighting effect on the noise floor by eliminating high-frequency energy-bands in the frequency transformation of every local pixel’s neighborhood [11]. These operations result in a higher depth-map accuracy and lower noise floor as can be seen from Fig. 1, where the v2 exhibits a noise auto-covariance of  $11.4707mm^2$  compared with  $22.7057mm^2$  for the Xbox. Despite the improved performance of the v2, noise remains an issue, as is the case for every electronic perception system.

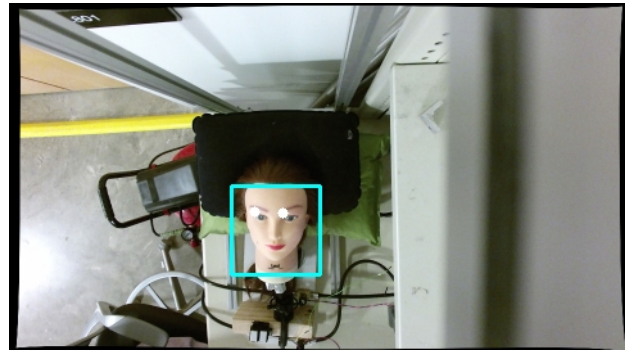


Fig. 2: Original colored image retrieved from the Kinect v1 Sensor.

To alleviate this, we employ a multisensor data fusion of both Kinect sensors’ observations. We achieved this by local Kalman Filter estimates of each sensors observations, and we fuse the estimates via a variance-weighted multisensor kalman filter fusion scheme.

#### A. Face Detection and eye-feature tracking

We approached the face features detection problem with Haar Cascade Classifiers (HCC), originally proposed by Viola and Jones [12]. HCC’s are based on integral image representation, which allow for features evaluation while maintaining high detection rates. The features used resemble Haar basis functions, which are computed from the integral images. A classifier is formed by choosing a small number of crucial features with AdaBoost, and a weighted sum of individual classifiers is used to construct a strong object detection classifier in a cascade manner. This increases the detector’s speed by concentrating on areas within an image with high probability of features of interest.

A drawback of the effectiveness of HCC’s is the memory bandwidth that they consume on computing devices when searching through image pixels for specific regions of interest. Searching through a  $640 \times 480$  pixels grayscale image for specific features as in Fig. 2 caused a 90% reduction in the frame rates of either sensor, when the algorithm is run on a CPU. To overcome this, both sensor’s images were spatially downsampled via linear interpolation before HCCs were applied. Face detection was performed on a single NVIDIA Quadro K1100M GPU. We retrieve each detected face from the GPU, and then detect eyes within detected faces using the same procedure.

To achieve robustness in detection, the number of minimum neighbors in each candidate rectangle feature should have been weighted based on our experience with faces of interest and this gave us more than 90% face detection rate for both sensors. The search area within an image was chosen to be within the range of  $(5 \times 5)$  pixels and  $(20 \times 20)$  pixels. A similar approach was used for the eye classifier. The final implementation achieved a frame rate of 15Hz for each sensor running independently on the Linux host computer. Further improvement in frame rates is an avenue for future work.

### B. Local Kalman Filters

From Fig. 1, we see that both RGB-D sensors suffer from notable associated noise, which is not suitable for our control requirement. To refine the observation, local Kalman Filter (KF) estimates for each sensor was computed. The KF algorithm determines state estimates  $\hat{\mathbf{x}}(i)$  that minimized the mean-squared error to the true state  $\mathbf{x}(i)$ , given a measurement sequence  $z(1), \dots, z(j)$ , that is

$$\hat{\mathbf{x}}(i|j) = \arg \min_{\hat{\mathbf{x}}(i|j) \in \mathbb{R}^n} \mathbb{E}\{(\mathbf{x}(i) - \hat{\mathbf{x}})(\mathbf{x}(i) - \hat{\mathbf{x}})^T | z(1), \dots, z(j)\} \\ \triangleq \mathbb{E}\{\mathbf{x}(i) | z(1), \dots, z(j)\} \triangleq \mathbb{E}\{\mathbf{x}(i) | Z^j\} \quad (1)$$

where the obtained estimate is the expected value of the state at time  $i$  given observations up to time  $j$ . The covariance of the estimation error is given by

$$\mathbf{P}(i|j) \triangleq \mathbb{E}\{(\mathbf{x}(i) - \hat{\mathbf{x}}(i|j))(\mathbf{x}(i) - \hat{\mathbf{x}}(i|j))^T | Z^j\}. \quad (2)$$

Assuming the model of the state is common to both sensors and denoting the distance from the v2 to the head as  $d(k)$ , we define  $\mathbf{x}(k) = [d(k), \dot{d}(k)]^T \in \mathbb{R}^2$  as the state vector of interest, and let  $\Delta T$  be the time between steps  $k-1$  and  $k$ . The model state update equations are given by

$$\mathbf{x}_k = \mathbf{F}_k \mathbf{x}_{k-1} + \mathbf{B}_k \mathbf{u}_k + \mathbf{G}_k \mathbf{w}_k \quad (3)$$

where  $\mathbf{F}(k) \in \mathbb{R}^{2 \times 2}$  is the state transition matrix given by

$$\mathbf{F} = \begin{bmatrix} 1 & \Delta T \\ 0 & 1 \end{bmatrix} \quad (4)$$

$\mathbf{u}(k) \in \mathbb{R}^2$  is the control input,  $\mathbf{B}(k)$  is the control input matrix that maps inputs to system states,  $\mathbf{G}(k) \in \mathbb{R}^{2 \times 2}$  process noise matrix, and  $\mathbf{w}(k) \in \mathbb{R}^2$  is a random variable that models the state uncertainty. In the absence of inputs  $\mathbf{B}_k \mathbf{u}_k = 0$ , and the model becomes

$$\mathbf{x}_k = \mathbf{F}_k \mathbf{x}_{k-1} + \mathbf{G}_k \mathbf{w}_k \quad (5)$$

where  $\mathbf{w}_k$  is the effect of an unknown input and  $\mathbf{G}_k$  applies that effect to the state vector,  $\mathbf{x}_k$ . The process noise is assumed unknown, and is modeled as uncontrolled forces causing an acceleration  $a_k$  in the head position ( $a_k$  is thus a scalar random variable with normal distribution, zero mean and standard deviation  $\sigma_a$ ). We model this into (3) by setting  $\mathbf{G}_k$  to identity and set  $\mathbf{w}(k) \sim \mathcal{N}(0, \mathbf{Q}(k))$  where the covariance matrix  $\mathbf{Q}(k)$  is set to a random walk sequence defined by  $\mathbf{W}_k = [\frac{\Delta T^2}{2}, \Delta T]^T$ . Therefore, we find that

$$\mathbf{Q} = \mathbf{W} \mathbf{W}^T \sigma_a^2 = \begin{bmatrix} \frac{\Delta T^4}{4} & \frac{\Delta T^3}{2} \\ \frac{\Delta T^3}{2} & \Delta T^2 \end{bmatrix} \sigma_a^2. \quad (6)$$

Denoting the head displacement at time  $k$  as measured by the Xbox and v2 as  $z_1(k)$  and  $z_2(k)$  respectively, the sensors' measurements were mapped to the v2 reference frame and modeled as

$$z_s = \mathbf{H}_s(k) \mathbf{x}(k) + v_s(k) \quad s = 1, 2 \quad (7)$$

where  $\mathbf{H}_s(k) = [1 \ 0]^T$  maps the system's state space into the observed space, and  $v_s(k) \in \mathbb{R}$  is a random variable

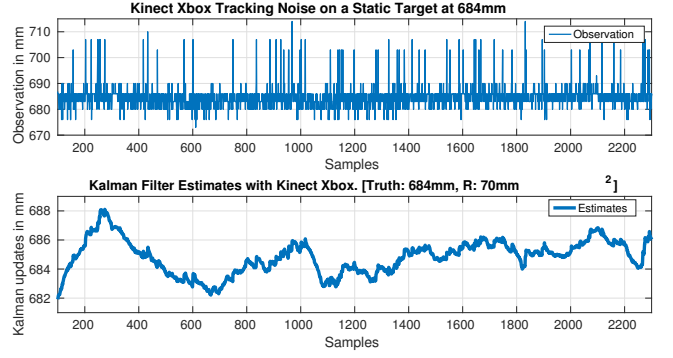


Fig. 3: KF results for the Xbox observation

that models the sensor error. We define  $v_s(k)$  as a normally distributed random variable with zero mean and variance  $\sigma_{rs}^2$ . We assume the random sequences  $v_1(k), v_2(k)$ ,  $\mathbf{w}(k)$  are independent and uncorrelated in time.

At each time step,  $k$ , each local KF's priori and posteriori estimates are computed through the typical prediction and update phases

**Prediction Phase:**

$$\hat{\mathbf{x}}_{k|k-1} = \mathbf{F}_k \hat{\mathbf{x}}_{k-1|k-1} + \mathbf{B}_k \mathbf{u}_k \\ \mathbf{P}_{k|k-1} = \mathbf{F}_k \mathbf{P}_{k-1|k-1} \mathbf{F}_k^T + \mathbf{Q}_k \quad (8)$$

where  $\hat{\mathbf{x}}_{k|k-1}$  and  $\mathbf{P}_{k|k-1}$  are the state prediction vector and the prediction covariance matrix respectively.

**Update Phase:**

$$\mathbf{K}_k = \mathbf{P}_{k|k-1} \mathbf{H}_k^T [\mathbf{H}_k \mathbf{P}_{k|k-1} \mathbf{H}_k^T + \mathbf{R}_k]^{-1} \\ \hat{\mathbf{x}}_{k|k} = \hat{\mathbf{x}}_{k|k-1} + \mathbf{K}_k (z_k - \mathbf{H}_k \hat{\mathbf{x}}_{k|k-1}) \\ \mathbf{P}_{k|k} = (\mathbf{I} - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_{k|k-1} \quad (9)$$

where  $\mathbf{K}_k$ ,  $\hat{\mathbf{x}}_{k|k}$ , and  $\mathbf{P}_{k|k}$  are respectively the KF gain, posteriori state estimate and its state covariance matrix. In implementing the KF of (8) and (9), the variance of the process noise/signal noise of each local KF was informed by our knowledge of the physics of both sensors (electronic perception methods, range resolutions and examining each sensor's depth map to understand the data available to the filter), engineering judgment, and kinematics of the process model. We found these values sufficiently modeled the underlying process dynamics

$$\sigma_a = 2000 \text{mm}^2; \quad \sigma_{r1}^2 = 70 \text{mm}^2 \text{ for the Xbox, and} \\ \sigma_{r2}^2 = 60 \text{mm}^2 \quad \text{for the Kinect v2 sensor.}$$

Figures 3 and 4 show the local filter estimate results of the observation from both the Kinect Xbox and v2 sensors post-filtering. The noise floor becomes noticeably reduced by each sensor after the KF filtering. The steady-state performance of both sensors include a reduction in the variance of the observation sequence by 80.81%, while the Kinect v2 shows an improvement in noise rejection by almost 60%.

### C. Data Fusion

Each local KF estimate was combined at a central fusion site to obtain a track-to-track fused global estimate. To communicate each estimate and associated covariance matrix, we

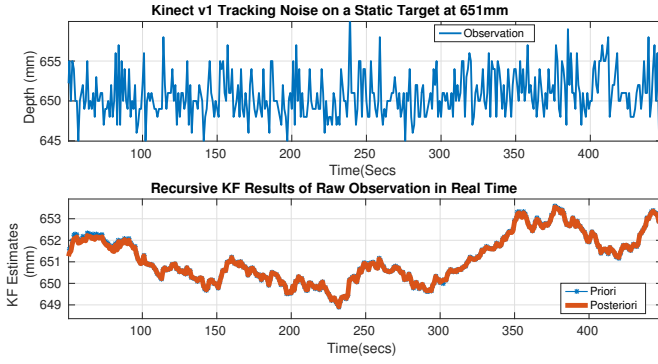


Fig. 4: KF results of Kinect v2's observation

create unix FIFO special files (*i.e. named pipes*) on the kernel file system, write the estimates and covariance matrices to the pipes at each local site and retrieve the values at the central site.

Named pipes are low-level file I/O systems that can be shared by processes with different ancestry. During data exchange through a FIFO, the kernel forwards all data internally without having to write it to the file system. Since they exist within the kernel and the file system is just an entry serving as a reference point for the processes to access the pipe with a filesystem name, there is practically no delay in data communication.

Local tracks are generated at each sensor site according to (8), resulting in two local state predictions from the Kalman filters (3). At the central fusion site, we assume a state model common to both sensors given by (9) and adopt a variance-weighted average of each local track in the global track fusion algorithm [13]

$$\hat{\mathbf{x}}_F(k|k) = \mathbf{P}_F(k|k) \sum_{i=1}^N [\mathbf{P}_s^{-1}(k|k) \hat{\mathbf{x}}_s(k|k)]$$

$$\text{where } \mathbf{P}_F(k|k) = \left[ \sum_{i=1}^N \mathbf{P}_s^{-1}(k|k) \right]^{-1}. \quad (10)$$

Fig. 5 shows the output of the fusion scheme compared against the single Kalman filters. The fusion of the local tracks produces better estimates, with improved signal to noise ratio. The fused estimate assigns more weight to the less noisy signal from Kinect v2. Through the implementation of the local tracks and a global track KF estimator, we improved the accuracy of the effective signal to be used in our control algorithm to no more than a standard deviation of 0.75mm from the true position of an object. The noise spikes in the fused tracks when the process state estimates are yet to converge as noticeable in Fig. 5 can be attributed to the noisy initialization of pixels in the sensors before they attain their steady state values. On average, it takes approximately 30 seconds for the pixel values in the kinect sensor to reach their final steady state values [14]. This can be avoided by running the fusion algorithm for at least 2 minutes before the fused signal is used for any control purposes. The code for the multisensor fusion experiment is available on the git

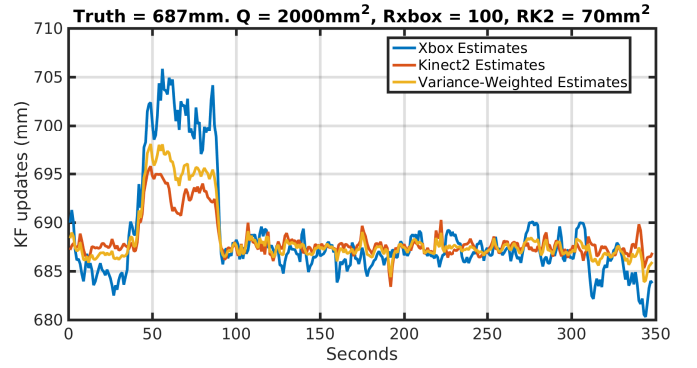


Fig. 5: Kalman filter Track-to-Track fusion of Kinect Xbox and v2's local tracks

repos [15] and [16].

#### IV. IDENTIFICATION AND CONTROL

We approach the modeling procedure with an identification prediction error (PEM) approach, where we estimate a mathematical model,  $G(t)$ , based on the minimization of the sum of squared errors between estimates of the head height above the table,  $y(t)$ , and true head height,  $y(t)$ , from the fusion *i.e.*

$$G(t) = \arg \min_{\theta} V_N(\theta, Z^N)$$

$$\text{where } V_N(\theta, Z^N) = \sum_{k=1}^K \sum_{i=1}^n \frac{1}{2} (\hat{y}_i(k) - y_i(k))^2. \quad (11)$$

$Z^N = \{u(1) \cdots u(N) \ y(1) \cdots y(N)\}$  is the vector of past input and output (fused estimates) measurements over a bounded interval  $[1, N]$  and  $\theta$  is the greedy vector of parameters that approximate the model we seek to build. (11) is a special case of the least squares criterion.

##### A. Model Structure

Following Ljung's formulation in [17, §4.5], we pose the identification problem as determining the “best model” from a set of candidate model sets via an iterative approach that parameterizes the noncountable model sets smoothly over an area with the assumption that the underlying system is linear time-invariant. Here, our model structure is a differentiable mapping from a connected, compact subset  $\mathcal{D}_{\mathcal{M}}$  of  $\mathcal{R}^d$  to a model set  $\mathcal{M}^*$ , such that the gradients of the predictor functions are stable. This procedure comes with the MATLAB system identification toolbox and since the method is well-documented in [18] we omit details here.

External disturbances and stochastic variables are modeled as additive white noise sequence,  $e(k)$ , based on lagged inputs and outputs, and our objective is to estimate a stochastic state space model structure of the form

$$\begin{aligned} x(k+1) &= Ax(k) + Bu(k) + w(k) \\ y(k) &= Cx(k) + Du(k) + v(k) \end{aligned} \quad (12)$$

where the noise terms  $w(k)$  and  $v(k)$  compensate for the effect of disturbances beyond frequencies of interest to system

dynamics and make the model robust to model uncertainties. Since  $u$  and  $y$  alone are measurable in our setup, the states  $x(k)$  are estimated and (12) becomes a linear regression problem, where all the unknown matrix entries are linear combinations of the measured inputs and output variables. This can be written as

$$Y(k) = \Theta\Phi(k) + E(k) \quad (13)$$

where

$$Y(k) = \begin{bmatrix} x(k+1) \\ y(k) \end{bmatrix}, \quad \Theta = \begin{bmatrix} A & B \\ C & D \end{bmatrix}$$

$$\Phi(k) = \begin{bmatrix} x(k) \\ u(k) \end{bmatrix} \quad \text{and} \quad \mathbb{E}(k) = \begin{bmatrix} \mathbb{E}(w(k)) \\ \mathbb{E}(v(k)) \end{bmatrix}.$$

We assume the noise term is white in order to assure an unbiased model. The parameter estimation problem is then to estimate the  $A, B, C$ , and  $D$  matrices by the linear least squares regression of (13) assuming no physical insight into the system (i.e. a black box model).  $\mathbb{E}(w(k))$  and  $\mathbb{E}(v(k))$  are estimated as a sampled sum of squared errors of the residuals.

### B. Parameter Estimation

The input,  $u(k)$ , and output signals,  $y(k)$ , can be characterized by a linear difference equation of the form

$$\begin{aligned} y(k) &= -a_1 y(k-1) - \dots - a_{n_a} y(k-n_a) \\ &\quad - b_1 u(k-1) - \dots - b_{n_b} u(k-n_b) - e(k) \\ &\quad - c_1 e(k-1) - c_{n_c} e(k-n_c) \end{aligned} \quad (14)$$

where  $e(k)$  describes the equation error as a moving average of white noise, and we assume  $e(k)$  has a bias-variance term  $\lambda$ . We can rearrange (14) using the vectors

$$\begin{aligned} \psi(k, \theta) &= [-y(k-1) \dots - y(k-n_a) \quad u(k-1) \dots \\ &\quad u(k-n_b), e(k-1), \dots, e(k-n_c), \theta]^T \end{aligned} \quad (15)$$

$$\theta = [-a_1, \dots, -a_{n_a}, -b_1, \dots, -b_{n_b}, -c_1, \dots, -c_{n_c}]. \quad (16)$$

The adjustable parameters of (15) are elements of  $\theta$ . In our prediction model, it is convenient to write (14) as a one-step-ahead predictor of the form

$$\hat{y}(k) = G(q, \theta)u(k) + H(q, \theta)\hat{e}(k) \quad (17)$$

$$\text{with } G(q, \theta) = \frac{B(q)}{A(q)}, \quad H(q, \theta) = \frac{C(q)}{A(q)}$$

which is a complete autoregressive moving average with exogenous input (ARMAX) model.  $G(q, \theta)$  represents the transfer function from input to output predictions, and  $H(q, \theta)$  denotes the transfer function of prediction errors to the output model,  $\hat{y}(k)$ ;  $q$  is the z-transform,  $z^{-1}$ , while  $A(q)$ ,  $B(q)$ , and  $C(q)$  are polynomials defined as

$$\begin{aligned} A(q) &= 1 + a_1 q^{-1} + \dots + a_{n_a} q^{-n_a}, \\ B(q) &= b_1 q^{-1} + \dots + b_{n_b} q^{-n_b}, \\ C(q) &= 1 + c_1 q^{-1} + \dots + c_{n_c} q^{-n_c} \end{aligned} \quad (18)$$

[19]. The predictor turns out to be a linear filter of the form

$$\hat{y}(k|\theta) = W_y(q, \theta)y(k) + W_u(q, \theta)u(k) \quad (19)$$

$$\text{and } y(k) = G(q, \theta)u(k) + H(q, \theta)[y(k) - \hat{y}(k)] \quad (20)$$

where  $H(q, \theta)$  is the noise model and  $\hat{y}(k)$  above can be regarded as the one-step ahead predictor. After rearranging (19), we find that

$$W_y = 1 - H^{-1}(q, \theta) \quad \text{and} \quad W_u(q, \theta) = G(q, \theta)H^{-1}(q, \theta)$$

such that the residual errors from (19) become

$$e(k) = [y(k) - G(q, \theta)u(k)]H^{-1}(q, \theta). \quad (21)$$

(21) translates to passing the prediction errors through a linear filter that allows extra freedom in dealing with non-momentary properties of the prediction errors. Since the model is that of a linear system, (21) satisfies our objective by approximating the prefilter with the choice of the noise model in (13).

The estimation problem is to predict the estimates,  $\hat{y}(k|\theta)$  so that the errors,  $\varepsilon(t, \theta) = \|y(t) - \hat{y}(t|\theta)\|_p$  are minimized by the choice of an appropriate p-norm criterion function, such as the mean squared error proposed in (11).

1) *Input Signal Design*: The input signal choice for a system identification experiment will determine a system's operating point and model accuracy. Therefore, the input should be rich enough to excite a system and force it to show properties needed for the model's purpose. For the model to be informative across all the desired frequency range, a periodic, persistently exciting uniform Gaussian White noise (UGWN) signal with clipped amplitudes corresponding to the bandwidth of the PVQ valves was designed offline, and its frequency spectrum analyzed to ensure it had as small a crest factor as possible (since the asymptotic properties of the model will be mostly influenced by the spectrum rather than the waveform's time-series shape). Gaussian White Noise signals (GWN) and Pseudo-Random Binary Signals (PRBS) are well-known to achieve virtually any signal spectrum without very narrow pass bands. Therefore, pseudo-random uniform white noise sequences were generated using the very-long cycle random number generator algorithm. Given that the probability density function,  $f(x)$ , of the uniformly distributed uniform white noise is

$$\begin{aligned} f(x) &= \frac{1}{2}A \quad \text{if } x < |A| \text{ and} \\ u(x) &= 0 \quad \text{if } x > |A| \end{aligned} \quad (22)$$

where  $A$  is the specified amplitude. The expected mean,  $\mu$ , and the expected standard deviation,  $\sigma$  of the sequence is [20]

$$\begin{aligned} \mu &= \mathbb{E}(x) = 0 \\ \sigma &= [\mathbb{E}\{(x - \mu)\}^2]^{\frac{1}{2}} = \frac{A}{\sqrt{3}} \end{aligned} \quad (23)$$

The spectrum of the resulting signal in Fig. 6 gives good signal power, which nicely relates to the bandwidth of the pneumatic valves and achieves virtually all signal spectrum with little narrow pass bands.



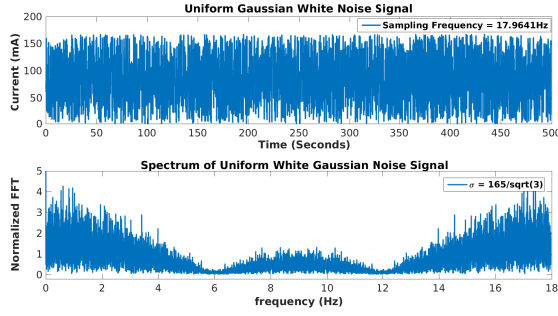


Fig. 6: Time/Frequency Domain Properties of Input Signal

We therefore use the signal of (22) to model the desired asymptotic estimates,  $\hat{y}(t)$ , of (17). We sampled  $y(t)$ , the fused measurement described in (III-C) well-above the system's Nyquist frequency,  $F_s$ , and acquired enough samples to make  $Z^N$  asymptotically approach  $\theta_N$  as  $N \rightarrow \infty$ . The data collection procedure closely follows that described in our previous paper and we refer readers to [8, §IV.A] for a more detailed treatment.

The collected data was separated in a 60:40% ratio for training and testing purposes, respectively, to assure a training model that generalizes well.

2) *State Space Realization*: If we define

$$\hat{Y}_r(k) = [\hat{y}(k|k-1), \dots, \hat{y}(k+r-1|k-1)]^T$$

and

$$\hat{Y} = [\hat{Y}_r(1) \dots \hat{Y}_r(N)],$$

it follows that 1) as  $N \rightarrow \infty$ , there are  $n$ -th order minimal state space descriptions of the system if and only if the rank of the vector of predictions,  $\hat{Y}$ , is equal to  $n$  for all  $r \geq n$ ; and 2) the state vector of any minimal realization in innovations can be chosen as linear combinations of  $\hat{Y}_r$  that form a row basis for  $\hat{Y}$  i.e.,

$$x(t) = L\hat{Y}_r(k)$$

with  $L$  being an  $n \times pr$  matrix ( $p$  is the dimension of  $y(k)$ ) [17, §7.3]. The true prediction is given by (17) with innovations  $e(j)$  written as a linear combination of past input-output data. The predictor can thus be expressed as a linear function of  $u(i)$ ,  $y(i)$ ,  $i \leq k-1$ . In practice, the predictor is approximated so that it depends on a finite amount past data such as  $s_1$  past outputs and  $s_2$  past inputs of the form

$$\begin{aligned} \hat{y}(k|k-1) = & \alpha_1 y(k-1) + \dots + \alpha_{s_1} y(k-s_1) \\ & + \beta_1 u(k-1) + \dots + \beta_{s_2} u(k-s_2). \end{aligned} \quad (24)$$

Piping the identification data through the MATLAB built-in function 'sseset' and testing various model orders based on the ranking of svd values of the Hankel matrix of input-output measurements [17], we obtained the results listed in Table I on training and testing dataset. The MATLAB system identification script is provided on the github repo: [21] and contains the dataset used for the experiment. The model set

TABLE I: Model estimates

| Data Type | Expts | MO <sup>1</sup> | MSE <sup>2</sup> | Fit (%) | FPE <sup>3</sup> |
|-----------|-------|-----------------|------------------|---------|------------------|
| Training  | i     | 2               | 0.001437         | 97.64   | 0.001438         |
|           | ii    | 4               | 0.001454         | 97.62   | 0.00145584       |
|           | iii   | 6               | 0.001333         | 97.72   | 0.001336         |
|           | iv    | 8               | 0.001298         | 97.76   | 0.001298         |
| Testing   | i     | 2               | 0.000963         | 98.47   | 0.000964         |
|           | ii    | 4               | 0.0008574        | 98.56   | 0.0008594        |
|           | iii   | 6               | 0.000846         | 98.57   | 0.000849         |
|           | iv    | 8               | 0.000843         | 98.57   | 0.000848         |

above exhibit a high-fit of estimate to fed data with generally good mean-square errors and final prediction errors for a control experiment. With increasing model order starting from 4, we see that the fits start reaching convergence, as the mean-square errors and final prediction errors become constant. In the frequency-domain, this is the equivalent to having pole-zero cancellations for higher-order models. We therefore conclude there is no useful properties a higher-order model could predict beyond an order of 8. It can be seen that the second-order model sufficiently approximates the system and is not significantly outperformed by the higher order models –which would inadvertently contribute higher complexity to the control design. We therefore picked the 2nd order state space model for the testing data given by

$$\begin{aligned} \mathbf{x}(k+Ts) &= \mathbf{A}\mathbf{x}(k) + \mathbf{B}u(k) + \mathbf{K}e(k) \\ \mathbf{y}(k) &= \mathbf{C}\mathbf{x}(k) + \mathbf{D}u(k) + \mathbf{e}(k) \end{aligned} \quad (25)$$

where  $Ts$  is the sampling period,  $\mathbf{e}(k)$  is the modeled zero-mean Gaussian white noise with non-zero variance,

$$\begin{aligned} \mathbf{A} &= \begin{bmatrix} 0 & 1 \\ -0.9883 & 1.988 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} -3.03e-07 \\ -4.254e-07 \end{bmatrix} \\ \mathbf{C} &= [1 \quad 0], \quad \mathbf{D} = 0, \text{ and } \mathbf{K} = [0.9253 \quad 0.9604]^T. \end{aligned} \quad (26)$$

The pair  $(A, B)$  is stabilizable and the pair  $(A, C)$  is detectable.

## V. LQG CONTROL

We propose to minimize the following cost function subject to the state equation of (26)

$$J = \sum_{k=0}^{\mathcal{K}} x^T(k) Q x(k) + R u(k)^T u(k) + 2x(k)^T N u(k) \quad (27)$$

where  $\mathcal{K}$  is the terminal sampling instant,  $Q$  is a symmetric, positive semi-definite matrix that weights the  $n$ -states of the  $A$  matrix,  $N$  specifies a matrix of appropriate dimensions that penalizes the cross-product between the input and state vectors, while  $R$  is a symmetric, positive definite weighting matrix on the control vector  $u$ . The quadratic cost function in (27) allows us to find an analytical solution (controller

<sup>1</sup>Model Order

<sup>2</sup>Mean Squared Error ( $mm^2$ ).

<sup>3</sup>Akaike Final Prediction Error ([17, Secs 7.4 and 16.4]).

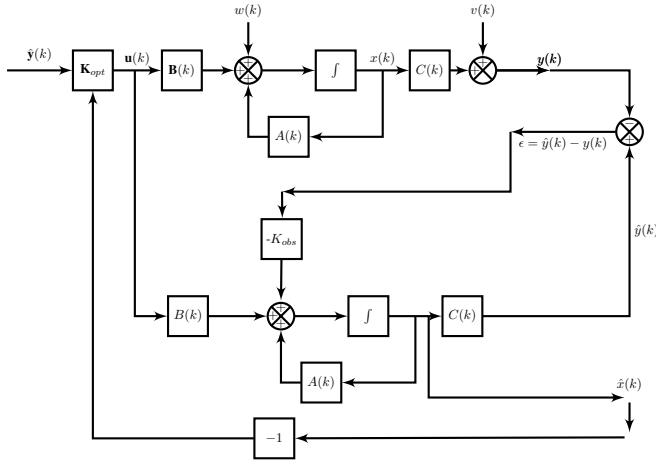


Fig. 7: Full Linear Quadratic Gaussian Plant Estimator

sequence) to the minimization of  $J$  over the prediction horizon,  $n_y$

$$\Delta u = \arg \min_{\Delta u} J \quad (28)$$

where  $\Delta u$  is the future control sequence and the first element in the sequence is used in the control law at every time instant. We model additive white noise disturbances into the discrete estimator's states; therefore the optimization problem becomes a stochastic optimization problem that must be solved. The separation theorem applies such that we can construct a state estimator which asymptotically tracks the internal states from observed outputs,  $y(k)$ , using the algebraic Riccati equation given as

$$A^T P A - (A^T P B + N)(R + B^T P B)^{-1}(B^T P A + N) + Q. \quad (29)$$

where  $P$  is an unknown  $n \times n$  symmetric matrix and  $A$ ,  $B$ ,  $Q$ , and  $R$  are known coefficient matrices as in (27). We find an optimal control law by solving the minimization of the LQ problem, (27) which we then feed into the states.

In practice, it is a good idea to start with an identity matrix,  $Q$ , a zero penalty matrix,  $N$ , and tune  $R$  till one obtains convergence by the state estimator. The following optimal values were used after a heuristic search

$$Q = \begin{bmatrix} 1.0566 & 0 \\ 0 & 1.0566 \end{bmatrix} \quad R = [0.058006] \quad (30)$$

We construct a full online estimator for the identified plant as in Fig. 7 whereby the noise processes are assumed to be independent, white, gaussian, of zero mean and known covariances. The optimal controller gains,  $K_{lqg}$ , are determined from the equation

$$K_{lqg} = R^{-1}(B^T P + N^T) \quad (31)$$

[22] where  $P$  is the solution to the algebraic Riccati equation (29) and  $\mathbb{E}[w(k)w'(\tau)] = R(k)\delta(k - \tau)$ . Therefore, the online optimal estimate,  $\hat{x}(k+1)$  of  $x(k)$  is

$$\hat{x}(k+1) = A(k)\hat{x}(k) + K_{lqg}[C(k)\hat{x}(k) - y(k)] \quad (32)$$

where  $\hat{x}(k_0) = \mathbb{E}[x(k_0)]$ . The observer is equivalent to a discrete stochastic kalman filter that estimates the optimal

state  $\hat{x}(k|k)$  as shown in Fig. 7. The estimator equations are similar to equations (8) and (9) and the online, unbiased estimate is

$$\begin{aligned} \hat{x}(k+1) &= A(k)\hat{x}(k) - K_{obs}[\hat{y}(k) - y(k)] + B(k)u(k) \\ \hat{y}(k) &= C(k)\hat{x}(k) \end{aligned} \quad (33)$$

$\Rightarrow$

$$\begin{aligned} \hat{x}(k+1) &= A(k)\hat{x}(k) - K_{obs}[C(k)\hat{x}(k) - y(k)] \\ &\quad + B(k)u(k). \end{aligned} \quad (34)$$

Through heuristics, we found the following variances of the online estimator to be useful:

$$Q_e = \begin{bmatrix} 0.4511 & 0 \\ 0 & 0.4511 \end{bmatrix}, \quad R_e = [0.01]$$

## VI. RESULTS AND DISCUSSION

The control network was implemented on an NI-myRIO running LabVIEW 2015. We initialized the Kinect sensors to allow for all pixels within the depth cameras to reach steady state under the right amount of light exposure to achieve robust estimation of observations. We perform three experiments to evaluate the developed state space model of IV-B and LQG controller of V. <sup>1</sup>. The input variable is the current that excites the PVQ valve, which in turn actuates the bladder; the head moves in response to bladder actuation. The fused estimate of the kinect sensors was used to estimate the real-time head pitch motion as described in III-C; this is in turn used in a feedback to the LQG controller.

Fig. 8 shows the results from a constant reference trajectory which the head is meant to track. We notice a settling time of approximately 24 seconds before we reach steady state. The delay arises from our design requirements and is not a drawback in clinical trajectory tracking where bumpless control action ensures smooth head motion to desired target. It is also seen that the controller exhibits relatively smooth tracking within a 1.5 mm standard deviation over time after a relative overshoot of 5mm in bottom graph of Fig. 8. The overshoot can be explained by the estimator's search for a steady state region based on the time it takes for the pixel values of the sensors to reach steady state. The controller tracks the reference to within  $\pm 2mm$ . There is however an operating range inconsistency in the current LTI model. We noticed at certain operating ranges of the model that the applied current based on fusion feedback does not guarantee convergence despite adding a proportional gain term in the feedforward path of the control structure as can be seen from Fig. 10. We conjecture this is due to an unmodeled nonlinearity at the inlet PVQ valve that maps input currents to system states. To better approximate the nonlinearity from input to output, we are using a Hammerstein block-structured model Fig. 9 that better approximates the nonlinearity from inputs to states and states to output of the system. In the Hammerstein model shown, the  $g(\cdot)$  block represents the static nonlinearity that maps inputs to states while the  $G(z^{-1})$  block denotes the

<sup>1</sup>The LabVIEW identification and control codes are available on the git repo [23].

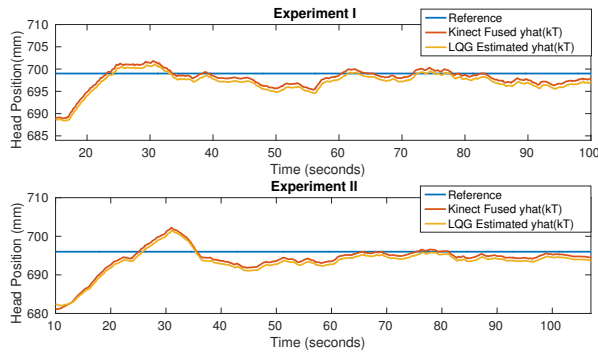


Fig. 8: LQG Controller on Manikin Head

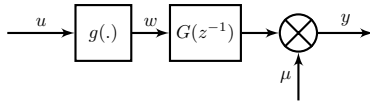


Fig. 9: The Hammerstein model structure

following maps the linear dynamics of the system's states to the sensors' measurements. We reason that this will minimize the mapping errors from input to output signals and provide better offset-free tracking.

The fusion algorithm proved useful enough to cancel jitter in the depth map of the sensor but falls short of 1mm accuracy in head and neck cancer RT. Adding more sensors to the fusion scheme could improve this tracking errors. Or we could achieve better head localization by using sophisticated motion capture systems or laser scanners.

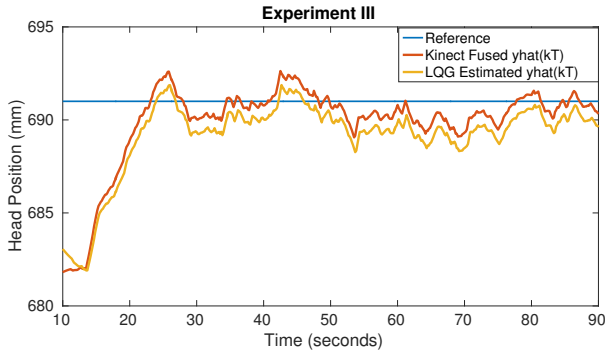


Fig. 10: LQG Controller on Manikin Head

## VII. SUMMARY

Fusion of the observations from depth sensors such as kinect can refine the accuracy of sensor observations. This can extend their use to sensitive robotics research applications. The system identification procedure provides a useful model that can control a PVC soft robot actuator for up to 2.5mm accuracy in pitch motion of a patient head during H&N cancer RT. The linear time invariant assumption does not seem sufficient to describe the dynamical properties of the system as the results show. This is an ongoing investigative research area.

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