

Nonlinear Continuum + Finite Elements

① Strains

② Stress

③ Nonlinear FEM weak form

Nonlinear Strains

Why? Strains must vanish for rigid body motion + rotation + translation. This is why linear strains are abandoned in nonlinear theory.

→ main nonlinear strain: $E = \text{Green strain}$ (not Young's modulus)

$$F = \text{deformation gradient} = \frac{dx}{d\bar{x}} = 1 \text{ if no strain}$$

$$ds^2 - dS^2 = 2d\bar{x} \cdot E \cdot d\bar{x} \quad ds = \text{line length in current configuration}$$

$$dS = \text{line length in reference configuration}$$

Key idea: green strain changes with square of lengths (makes it nonlinear)

$$\begin{aligned} ds^2 - dS^2 &= dx \cdot dx - d\bar{x} \cdot d\bar{x} = 2d\bar{x} \cdot E \cdot d\bar{x} \\ dx = F \cdot d\bar{x}, \text{ so } (Fd\bar{x})^T \cdot (Fd\bar{x}) - d\bar{x} \cdot d\bar{x} &= 2d\bar{x} \cdot E \cdot d\bar{x} \\ &= d\bar{x}^T F^T F d\bar{x} - d\bar{x} \cdot d\bar{x} = 2d\bar{x} \cdot E \cdot d\bar{x} \\ &= d\bar{x} \cdot F^T \cdot F \cdot d\bar{x} - d\bar{x} \cdot I \cdot d\bar{x} = 2d\bar{x} \cdot E \cdot d\bar{x} \\ &= d\bar{x} \cdot (F^T F - I - 2E) \cdot d\bar{x} = 0 \end{aligned}$$

$$2E = F^T F - I$$

$$E = \frac{1}{2}(F^T F - I)$$

→ connection to linear strain

$$F^T F = \frac{dx_k}{d\bar{x}_i} \frac{dx_k}{d\bar{x}_j} = F_{ik}^T F_{kj} \quad u = x - \bar{x} \quad \text{so} \quad x = u + \bar{x}$$

$$= \left(\frac{du_k}{d\bar{x}_i} + \frac{dx_k}{d\bar{x}_i} \right) \left(\frac{du_k}{d\bar{x}_j} + \frac{dx_k}{d\bar{x}_j} \right)$$

$$= \left(\frac{du_k}{d\bar{x}_i} + \delta_{ii} \right) \left(\frac{du_k}{d\bar{x}_j} + \delta_{jj} \right)$$

$$= \frac{du_k}{d\bar{x}_i} \frac{du_k}{d\bar{x}_j} + \delta_{ii} \frac{du_k}{d\bar{x}_j} + \frac{du_k}{d\bar{x}_i} \delta_{jj} + \delta_{ii} \delta_{jj}$$

$$= \frac{du_k}{d\bar{x}_i} \frac{du_k}{d\bar{x}_i} + \frac{du_k}{d\bar{x}_i} + \frac{du_i}{d\bar{x}_i} + \delta_{ii}$$

$$\text{if } i, j, k = 1$$

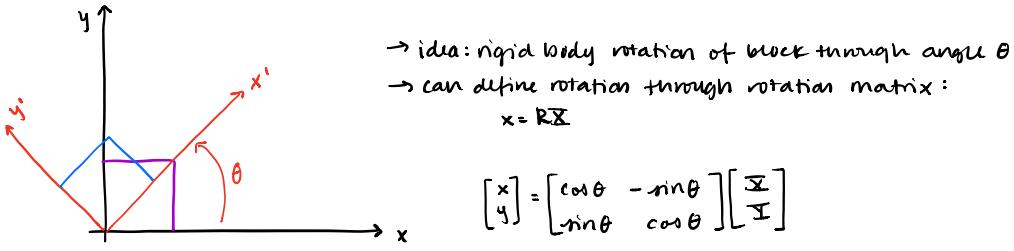
$$= (\varepsilon_{xx} \varepsilon_{xx}) + (\varepsilon_{xx}) + (\varepsilon_{xx}) + 1$$

$$E = \frac{1}{2}(F^T F - I) = \frac{1}{2}(\varepsilon_{xx}^2 + 2\varepsilon_{xx} + 1 - 1)$$

$$E = \underbrace{\varepsilon_{xx}}_{\text{standard linear strain}} + \underbrace{\frac{1}{2}\varepsilon_{xx}^2}_{\text{nonlinear strain}} \quad \text{in 1D}$$

standard linear strain

Why does linear strain fail?



$$u = \bar{x} - \bar{X} = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \begin{bmatrix} \bar{x} \\ \bar{y} \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \bar{x} \\ \bar{y} \end{bmatrix} = \begin{bmatrix} u_x \\ u_y \end{bmatrix}$$

$$\begin{bmatrix} u_x \\ u_y \end{bmatrix} = \begin{bmatrix} \cos\theta - 1 & -\sin\theta \\ \sin\theta & \cos\theta - 1 \end{bmatrix} \begin{bmatrix} \bar{x} \\ \bar{y} \end{bmatrix} \leftarrow \text{now calculate strains}$$

$$u_x = (\cos\theta - 1)\bar{x} - \sin\theta\bar{y}$$

$$u_y = \sin\theta\bar{x} + (\cos\theta - 1)\bar{y}$$

$$\varepsilon_{xx} = \frac{\partial u_x}{\partial \bar{x}} = \cos\theta - 1$$

$$\varepsilon_{yy} = \frac{\partial u_y}{\partial \bar{y}} = \cos\theta - 1$$

$$\gamma_{xy} = \frac{\partial u_x}{\partial \bar{y}} + \frac{\partial u_y}{\partial \bar{x}} = -\sin\theta + \sin\theta$$

$$\gamma_{xy} = 0$$

What does this tell you?

→ only 0 if $\theta = 0$... when there's no rotation at all

→ $\varepsilon_{xx}, \varepsilon_{yy}$ nonzero for any nonzero value of θ

→ So if $\theta \neq 0$, extensional strains do not vanish in linear theory

→ called "geometric nonlinearity"

→ this result does not depend on the material properties!

→ How about nonlinear (green strain)?

$$E = \frac{1}{2}(F^T F - I)$$

$$x = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} \begin{bmatrix} \bar{x} \\ \bar{y} \end{bmatrix}$$

$$F = \begin{bmatrix} \frac{\partial x}{\partial \bar{x}} & \frac{\partial x}{\partial \bar{y}} \\ \frac{\partial y}{\partial \bar{x}} & \frac{\partial y}{\partial \bar{y}} \end{bmatrix} = \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix}$$

$$F^T F = \begin{bmatrix} \cos\theta & \sin\theta \\ -\sin\theta & \cos\theta \end{bmatrix} \begin{bmatrix} \cos\theta & -\sin\theta \\ \sin\theta & \cos\theta \end{bmatrix} = \begin{bmatrix} \cos^2\theta + \sin^2\theta & 0 \\ 0 & \cos^2\theta + \sin^2\theta \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$$

$\frac{1}{2}(F^T F - I) = 0 \leftarrow$ no green strain does not cause spurious strains for any θ .

Nonlinear Stresses → 3 main stress values

① Cauchy Stress: σ

② Nominal Stress: P (aka Piola-Kirchhoff Stress)

③ 2nd Piola-Kirchhoff Stress: S

→ difference between these is whether stress is a reference or current configuration value

→ can convert between all stress values

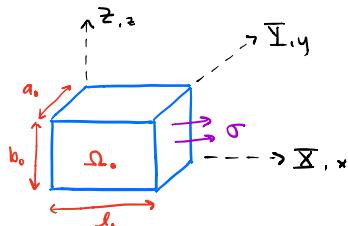
$$P = SF^T$$

$$\sigma = J^{-1}FP$$

$$S = JF^{-1}\sigma F^{-T}$$

} point is, if know one stress, can convert to others

Ex. uniaxial stress



under axial loading:

$$a_0 \rightarrow a$$

$$x = \frac{l}{l_0} X$$

$$b_0 \rightarrow b$$

$$y = \frac{a}{a_0} Y$$

$$l_0 \rightarrow l$$

$$z = \frac{b}{b_0} Z$$

→ goal: relate P and S to σ

$$F = \begin{bmatrix} \frac{\partial x}{\partial X} & \frac{\partial x}{\partial Y} & \frac{\partial x}{\partial Z} \\ \frac{\partial y}{\partial X} & \frac{\partial y}{\partial Y} & \frac{\partial y}{\partial Z} \\ \frac{\partial z}{\partial X} & \frac{\partial z}{\partial Y} & \frac{\partial z}{\partial Z} \end{bmatrix} = \begin{bmatrix} \frac{l}{l_0} & 0 & 0 \\ 0 & \frac{a}{a_0} & 0 \\ 0 & 0 & \frac{b}{b_0} \end{bmatrix} \quad \text{deformation gradient}$$

$$J = \text{Det}(F) = \frac{ab}{a_0 b_0 l_0}$$

$$F^{-1} = \begin{bmatrix} \gamma_1 & 0 & 0 \\ 0 & \gamma_a & 0 \\ 0 & 0 & \gamma_b \end{bmatrix} \quad \sigma = \begin{bmatrix} \sigma_x & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad \text{uniaxial stress}$$

$$P = F^{-1}J\sigma = \begin{bmatrix} \gamma_1 & 0 & 0 \\ 0 & \gamma_a & 0 \\ 0 & 0 & \gamma_b \end{bmatrix} \frac{ab}{a_0 b_0 l_0} \begin{bmatrix} \sigma_x & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} \frac{ab\sigma_x}{a_0 b_0 l_0} & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$$\rightarrow P_{xx} = \frac{ab}{a_0 b_0} \sigma_x = \frac{A}{A_0} \sigma_x \quad \text{... in stress-area is force} \quad \therefore P_{xx} = \frac{F}{A_0} \quad \leftarrow \text{engineering stress}$$

↑ area _{ref}
↑ referring area

$$S = F^{-T}P = \frac{l_0}{l} \frac{A}{A_0} \sigma_x = \boxed{\frac{l_0}{l} \frac{F}{A_0}} \quad \leftarrow \text{physical meaning?}$$

Strains for uniaxial stress:

$$E = \frac{1}{2}(F^T F - I): E_{11} = \frac{l^2 - l_0^2}{2l_0^2}, \quad E_{22} = \frac{a^2 - a_0^2}{2a_0^2}, \quad E_{33} = \frac{b^2 - b_0^2}{2b_0^2}$$

strains are differences of lengths squared \Rightarrow for comparison: linear strain $\varepsilon_{xx} = \frac{l-l_0}{l_0}$
rather than just lengths