

Thermo Final Exam Review

ISENTROPIC PROCESSES → constant entropy

Ideal Gas Model: $\Delta = s^o(T_2) - s^o(T_1) - R \ln\left(\frac{P_2}{P_1}\right)$ 4 property values:
 T_1, T_2, P_1, P_2

$$P_2 = P_1 \exp\left[\frac{s^o(T_2) - s^o(T_1)}{R}\right]$$

AIR modelled as an ideal gas, 2 states having same specific entropy

$$\frac{P_2}{P_1} = \frac{\exp[s^o(T_2)/R]}{\exp[s^o(T_1)/R]} \Rightarrow \frac{P_2}{P_1} = \frac{P_{r2}}{P_{r1}} \quad (s_1 = s_2, \text{ air only})$$

where $P_{r1} = P_r(T_1)$, $P_{r2} = P_r(T_2)$

relative pressure

$$V = \frac{RT}{P} \rightarrow \text{take ratio of specific volumes}$$

$$\frac{V_2}{V_1} = \left(\frac{RT_2}{P_2}\right) \left(\frac{P_1}{RT_1}\right) \Leftrightarrow \left[\frac{RT_2}{P_r(T_2)}\right] \left[\frac{P_r(T_1)}{RT_1}\right]$$

for 2 states
having same
specific entropy

relative volume

$$\frac{V_2}{V_1} = \frac{V_{r2}}{V_{r1}} \quad \text{for } s_1 = s_2$$

$V_r(T_2)$ $1/V_r(T_1)$

ISENTROPIC PROCESSES OF IDEAL GAS WHEN SPECIFIC HEATS CONSTANT

$$\Delta = C_p \ln \frac{T_2}{T_1} - R \ln \frac{P_2}{P_1} \quad \Delta = C_v \ln \frac{T_2}{T_1} + R \ln \frac{V_2}{V_1}$$

$$\rightarrow \text{ideal gas relations} \quad C_p = \frac{kR}{k-1} \quad C_v = \frac{R}{k-1}$$

$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}} \quad \text{for } s_1 = s_2, \text{ constant } k$$

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{k-1} \quad \Leftrightarrow \quad \frac{P_2}{P_1} = \left(\frac{V_1}{V_2}\right)^k \quad \text{for } s_1 = s_2, \text{ constant } k$$

↙ The polytropic process $PV^k = C$ of an ideal gas w/ constant specific heat ratio k is an isentropic process.
 \Rightarrow from $PV^n = C$, for ideal gas $n=1$ == isothermal, for any fluid $n=0$ == isobaric

ISENTROPIC EFFICIENCIES - comparison between actual performance of a device and performance that would be achieved under idealized circumstances for the same inlet state and same exit pressure.

ISENTROPIC TURBINE EFFICIENCY

$$\frac{W_{\text{act}}}{m} = h_1 - h_2 \quad \begin{aligned} &\bullet \text{heat xfer w/ surroundings ignored} \\ &\bullet \text{KE/PE effects ignored} \end{aligned}$$

W_{act} as $h_2 \downarrow \Rightarrow$ max value of turbine work corresponds to smallest allowed value for the specific enthalpy at exit

$$\frac{\dot{S}_{\text{gen}}}{m} = s_2 - s_1 \geq 0 \quad \text{bc entropy production } \frac{\dot{S}_{\text{gen}}}{m} \text{ can't be } \ominus, \text{ states with } s_2 < s_1 \text{ are not accessible in an adiabatic expansion}$$

\Rightarrow the only states that actually can be attained adiabatically are those with $s_2 > s_1$

\Rightarrow smallest allowed value for h_2 corresponds to state 2s and then the maximum value for turbine work is $(\dot{W}_{cv})_{s} = h_1 - h_{2s}$

\rightarrow in actual expansion thru turbine $h_2 > h_{2s}$ and thus less work than max developed

$$\eta_t = \frac{\dot{W}_{cv}/\dot{m}}{(\dot{W}_{cv}/\dot{m})_s} = \frac{h_1 - h_2}{h_1 - h_{2s}} \quad \text{typically } 0.7 - 0.9$$

ISENTROPIC NOZZLE EFFICIENCIES

\rightarrow ratio of actual specific KE of gas leaving nozzle to KE at exit that would be achieved in an isentropic expansion

$$\eta_{noz} = \frac{V_2^2/2}{(V_2^2/2)_s} \quad 95\% + \dots \text{ well designed, nearly free of internal irreversibilities}$$

ISENTROPIC COMPRESSOR & PUMP EFFICIENCIES

work input per unit mass flowing through compressor $(-\frac{\dot{W}_{cv}}{\dot{m}}) = h_2 - h_1$ $\begin{aligned} &\text{- neg. heat & fr w/noz} \\ &\text{- neg. KE/PE effects} \end{aligned}$

\rightarrow value of work input depends on specific enthalpy at exit h_2

\rightarrow magnitude of work input \downarrow as $h_2 \downarrow$

\rightarrow minimum work input corresponds to smallest allowed value for the specific enthalpy at the compressor exit

\therefore smallest allowed enthalpy at exit state would be achieved in an isentropic compression from the specific inlet state to the specified exit pressure

$$(-\frac{\dot{W}_{cv}}{\dot{m}})_s = h_{2s} - h_1 \leftarrow \text{minimum work input}$$

\Rightarrow in actual compression, $h_2 > h_{2s}$ thus more work than minimum required

$$\eta_c = \frac{(-\dot{W}_{cv}/\dot{m})_s}{(-\dot{W}_{cv}/\dot{m})} = \frac{h_{2s} - h_1}{h_2 - h_1} \quad \text{typically } 75\% \text{ to } 85\%$$

η_p defined similarly

In thermal, internally reversible $Q = \frac{\dot{Q}_{cv}}{T} + \dot{m}(s_1 - s_2) + \dot{S}_{cv}^0$

$$\frac{\dot{Q}_{cv}}{\dot{m}} = T(s_2 - s_1) \Leftrightarrow (\frac{\dot{Q}_{cv}}{\dot{m}})_{rev}^{int} = \int_1^2 T ds \quad \begin{aligned} &\text{[Area under} \\ &\text{T-s curve]} \end{aligned}$$

$$(\frac{\dot{W}_{cv}}{\dot{m}})_{rev}^{int} = \int_1^2 T ds + (h_1 - h_2) + \left(\frac{V_1^2 - V_2^2}{2} \right) + g(z_1 - z_2)$$

$$\Rightarrow T ds = dh - v dp \Rightarrow \int_1^2 T ds = (h_2 - h_1) - \int_1^2 v dp$$

$$\Rightarrow (\frac{\dot{W}_{cv}}{\dot{m}})_{rev}^{int} = - \int_1^2 v dp + \left(\frac{V_1^2 - V_2^2}{2} \right) + g(z_1 - z_2) \quad \begin{aligned} &\text{[magnitude of integral is} \\ &\text{area shaded behind curve} \\ &\text{on PV diagram]} \end{aligned}$$

applicable to turbines, compressors, pumps

$$-\int_1^2 \dot{v} dp + \left(\frac{v_1^2 - v_2^2}{2} \right) + g(z_1 - z_2) \rightarrow \text{if no KE/PE} \Rightarrow \left(\frac{\dot{W}_{cv}}{\dot{m}} \right)_{rev} = -\int_1^2 \dot{v} dp$$

shows that work value is related to magnitude of v of gas or liquid as flow from inlet to exit

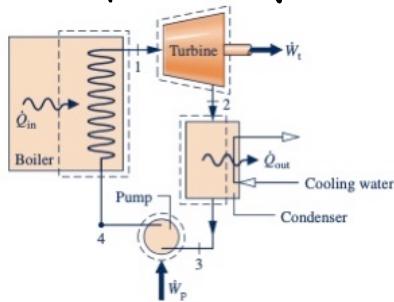
if v constant, as in many applications w/liquids

$$\left(\frac{\dot{W}_{cv}}{\dot{m}} \right)_{rev} = -v(p_2 - p_1) \quad v = C, \Delta KE, \Delta PE = 0$$

*ex: for the same pressure rise, pump requires smaller work input per unit of mass flowing than compressor bc liquid v is much smaller than that of vapor (also correct for actual pumps & compressors where irreversibilities are present during operation)

THE RANKINE CYCLE VAPOR POWER PLANTS

- 1st law of TD requires that net work developed by a system undergoing a power cycle must equal the net energy added by heat xfer to the system
- 2nd law of TD requires that the thermal efficiency of a power cycle must be less than 100%



* 4 principal components: turbine, condenser, pump, boiler

* most large scale plants use water as the working fluid

* each component regarded as operating at steady-state for our purposes

① **TURBINE** vapor from the boiler @ state 1 w/its elevated temp & press, expands thru turbine to produce work, then is discharged to condenser at state 2 w/relatively low pressure

$$\frac{\dot{W}_t}{\dot{m}} = h_1 - h_2$$

② **CONDENSER** here thereby heat xfer from working fluid to cooling H₂O flowing in separate stream. The working fluid condenses and temp of cooling water \uparrow

$$\frac{\dot{Q}_{out}}{\dot{m}} = h_2 - h_3 \quad \dots \text{energy xfer by heat from working fluid to cooling H}_2\text{O}$$

③ **PUMP** liquid leaving condenser is pumped to into high press boiler $\frac{\dot{W}_p}{\dot{m}} = h_3 - h_4$... \dot{W}_p/\dot{m} is work input

④ **BOILER** working fluid completes cycle as liquid leaving pump @ 4 (the "boiler feedwater") is heated to saturation & evaporated in boiler

$$\frac{\dot{Q}_{in}}{\dot{m}} = h_1 - h_4 \quad \dots \frac{\dot{Q}_{in}/\dot{m}}{\dot{m}} = \text{energy xfer by heat from energy source into working fluid per unit mass}$$

THERMAL EFFICIENCY gauges extent to which energy input to working fluid passing thru boiler is converted to net work output

→ net work output = net heat input

$$\eta = \frac{\dot{W}_t/m - \dot{W}_p/m}{\dot{Q}_{in}/m} = \frac{(h_1 - h_2) - (h_3 - h_4)}{h_1 - h_4}$$

$$\eta = \frac{\dot{Q}_{in}/m - \dot{Q}_{out}/m}{\dot{Q}_{in}/m} = 1 - \frac{\dot{Q}_{out}/m}{\dot{Q}_{in}/m} = 1 - \frac{(h_2 - h_3)}{(h_1 - h_4)}$$

HEAT RATE amount of energy added by heat xfer to cycle to produce a unit of net work output

→ heat rate is $1/\alpha$ to η

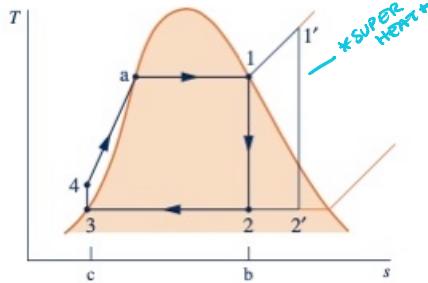
BACK WORK RATIO (BWR) ratio of pump work input to work developed by turbine

$$bwr = \frac{\dot{W}_p/m}{\dot{W}_t/m} = \frac{(h_4 - h_3)}{(h_1 - h_2)}$$

→ typically quite low since change in specific enthalpy for expansion of vapor through turbine is normally many times greater than the increase in enthalpy for the liquid passing through the pump

IDEAL RANKINE CYCLE

- w/out irreversibilities, no friction in boiler/condenser so working fluid would flow thru these components @ constant pres
- processes thru turbine and pump are isentropic
- ideal Rankine cycle consists of internally reversible processes



$1'-2'-3-4-1'$
* example of *
superheating

1→2 isentropic expansion of working fluid through turbine from sat. vapor at state 1 to the condenser press

2→3 heat xfer from working fluid as it flows @ constant pres thru condenser exiting as sat. liq. at state 3

3→4 isentropic compression in pump to state 4 in compressed liquid region

4→1 heat xfer to working fluid as it flows at constant pres through boiler to complete cycle

Area 1-b-c-4-a-1 heat xfer to working fluid passing thru boiler

Area 2-b-c-3-2 heat xfer from working fluid passing thru condenser

Enclosed Area 1-2-3-4-a-1 the net heat input per unit mass, or equivalently, net work output per unit mass

$$\left(\frac{\dot{W}_t}{m}\right)_{int} = v_3(P_u - P_3) \rightarrow \text{isentropic, int. rev., adiabatic}$$

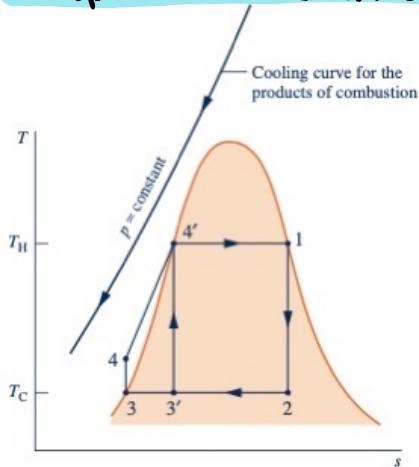
* v_{th} ↑ ave temp @ which energy added by heat xfer ↑ and/or ave temp @ which energy rejected by heat xfer ↓

Effects of Boiler and Condenser Pressures on the Rankine Cycle

↑ boiler pressure, ↑ η_{th}
 ↓ condenser pres., ↑ η_{th}

* including a condenser in which steam side is operated at press below atmospheric, turbine then has lower-press region in which to discharge, resulting in significant ↑ in net work and η_{th}

Comparison w/ Carnot Cycle



Rankine: 1-2-3-4-4'-1

Carnot: 1-2-3'-4'-1

$$(\eta_{th})_{\text{Rankine}} < (\eta_{th})_{\text{Carnot}}$$

but shortcoming of Carnot is that pump as to pump 2-phase mixture and it's better to condense vapor completely as in Rankine

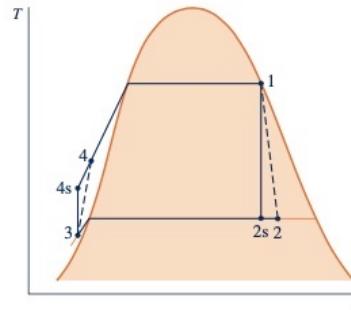
Principal Irreversibilities and Losses

I. Internal Effects

- Turbine - actual adiabatic expansion thru turbine is accompanied by an ↑ in entropy
 → the work developed is less than corresponding isentropic expansion 1→2s

$$\eta_t = \frac{(\dot{W}_t/m)}{(\dot{W}_t/m)_s} = \frac{h_1 - h_2}{h_1 - h_{2s}} \quad \leftarrow \text{actual work}$$

⇒ net result: ↓ power plant power output + η_{th}



- Pump - work input to pump required to overcome irreversibilities reduces net power output of plant
 → actual pumping accompanied by ↑ entropy
 → ∴ $W_{\text{input actual}} > W_{\text{input isentropic}} 3 \rightarrow 4_s$

$$\eta_p = \frac{(\dot{W}_p/m)_s}{(\dot{W}_p/m)} = \frac{h_{4s} - h_3}{h_4 - h_3} \quad \leftarrow \text{actual and larger in magnitude than isentropic}$$

$$\Rightarrow \text{can also say } \eta_p = \frac{(\dot{W}_p/m)_s}{(\dot{W}_p/m)} = \frac{v_3(P_u - P_3)}{h_u - h_3}$$

→ because pump work is much less than turbine work, irreversibilities in pump have smaller effect on η_{th} than turbine

II. External Effects

- energy discharged by heat xfer to cooling H₂O as working fluid condenses

IMPROVING PERFORMANCE - RANKINE

Remember:

- increase in boiler press or decrease in condenser press may result in reduction of steam quality at exit of turbine
- if quality becomes too low, erode turbine blades and cause ↓ in η_{th} (superheat & reheat avoid this)

SUPERHEAT energy added by heat xfer to steam bringing it to a superheated vapor condition at turbine inlet
 Steam generator = boiler + superheater

$$\uparrow \eta_{th} \quad \uparrow \% \text{ of steam}$$

REHEAT steam doesn't expand to condenser pressure in a single stage; instead, steam expands thru first stage turbine (1 → 2) to some pres between steam generator and condenser pressures; steam then reheated in steam generator (2 → 3) and there's no pressure drop

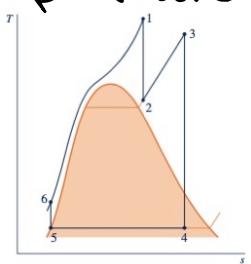
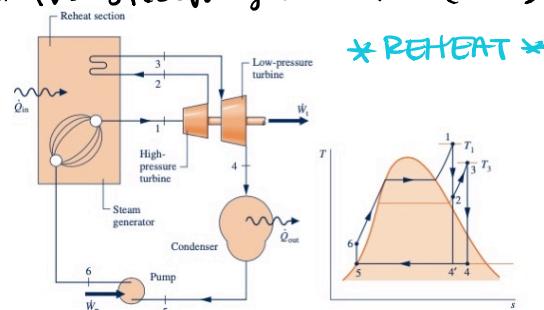
- after reheating, steam expands in 2nd stage turbine to condenser pressure (3 → 4)

$$x \uparrow$$

SUPERCritical

temp of steam entering turbine is restricted by metallurgical limitations imposed by materials used to fabricate the super heater, reheat, and turbine

- high pres in steam generator also requires piping that can withstand great stresses at elevated temperatures
- *this process now allows vapor power plants to operate w/steam generator pressures exceeding the critical pressure of water



Process 6 → 1: steam generation occurs at pres above critical pres (no phase change occur, no boiler is used)

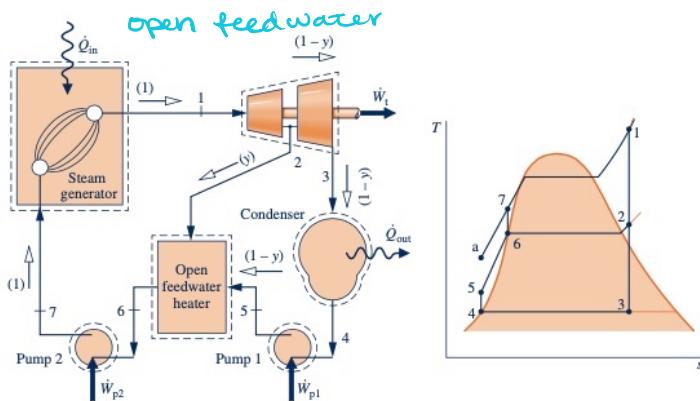
- H_2O flowing thru tubes is gradually heated from liquid to vapor w/out bubbling and w/o boil (heating provided by combustion of coal w/air)

Improving Performance - Regenerative Vapor Power Cycle

$\uparrow \eta_{th}$... ave temp of heat addition ↑ thereby ↑ efficiency

Open Feedwater Heaters

- type of direct-contact heat exchanger in which streams @ different temperatures mix to form a stream at an intermediate temp



- Working fluid passes isentropically thru the turbine stages and pumps, and flow thru the steam generator, condenser, and feedwater heater takes place w/no pressure drop in any of these components

1 → 2: steam enters 1st stage turbine @ 1 and expands to 2 where fraction of total flow is extracted (bleed) into an open feedwater heater operating at the extraction pressure P_2

2 → 3: rest of steam expands thru 2nd stage turbine to 3.

3 → 4: this portion of flow is condensed to sat. i.e. @ 4

4 → 5: then pumped to the extraction pressure and introduced into feedwater heater @ 5

5 → 6: single mixed steam exits feedwater heater @ 6

{ 2 → 6 (6 is sat. i.e.)

6 → 7: liquid then pumped to steam generator pressure and enters 7

7 → 1: working fluid heated

* operating conditions are such that the reduction in heat added more than offsets the decrease in net work developed, resulting in increased η_{th} in regenerative powerplants)

→ looking @ 2 turbines here: (as the ct)

$$\dot{m}_2 + \dot{m}_3 = \dot{m}_1$$

\dot{m}_1 = rate entering turbine, \dot{m}_2 = rate extracted, \dot{m}_3 = rate exits turbine

$$\frac{\dot{m}_2}{\dot{m}_1} + \frac{\dot{m}_3}{\dot{m}_1} = 1$$

$$y = \frac{\dot{m}_2}{\dot{m}_1} \Rightarrow \text{then the fraction of the total flow passing thru 2nd stage turbine} = \frac{\dot{m}_3}{\dot{m}_1} = 1 - y \quad \text{(see parentheses)}$$

$$0 = y h_2 + (1-y) h_5 - h_6 \quad [\text{cons. of mass + energy}]$$

$$y = \frac{h_6 - h_5}{h_2 - h_5} \quad \dots \text{can find fraction } y \text{ when } 2, 5, \text{ & } 6 \text{ fixed}$$

$$\text{total turb. work : } \frac{\dot{W}_t}{\dot{m}_1} = (h_1 - h_2) + (1-y)(h_2 - h_3)$$

$$\text{total pump work : } \frac{\dot{W}_p}{\dot{m}_1} = (h_7 - h_6) + (1-y)(h_5 - h_4)$$

energy added by
heat xfer to working fluid passing thru steam gen. : $\frac{\dot{Q}_{in}}{m_i} = h_1 - h_2$

energy rejected by heat xfer to cooling H₂O : $\frac{\dot{Q}_{out}}{m_i} = (1-\eta)(h_1 - h_2)$

Closed Feedwater Heaters

- shell-and-tube-type recuperators in which the feedwater temp ↑ as extracted steam condenses on outside of tubes carrying feedwater
- since the 2 streams don't mix, they can be @ diff press.

INTERNAL COMBUSTION ENGINES - 2 principal types

1. SPARK-IGNITION ENGINE

mixture of fuel + air ignited by a spark plug

2. COMPRESSION-IGNITION ENGINE

air compressed to high enough press + temp that combustion occurs spontaneously when fuel injected

ENGINE TERMINOLOGY

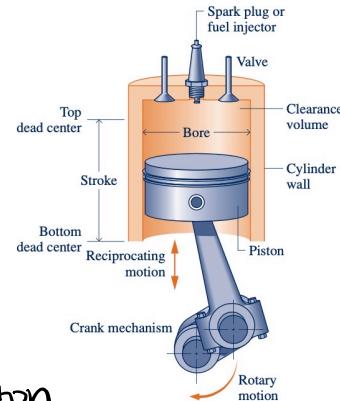
bore = cylinder diameter

stroke = distance piston moves in one direction

piston is @ TDC when it has moved to a position where cylinder volume is a minimum
→ minimum volume = clearance volume

when piston moves to position of max cylinder volume, piston @ BDC

displacement volume = volume swept out by piston as it moves from TDC to BDC



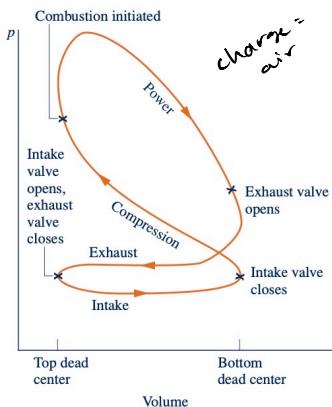
compression ratio, r: $H_{BDC} \div H_{TDC}$

→ the reciprocating motion of the piston is converted to rotary motion by a crank mechanism

mean effective pressure, mep: theoretical constant pressure that, if it acted on the piston during the power stroke, would produce same net work as actually developed in one cycle

$$mep = \frac{\text{net work for one cycle}}{\text{displacement} \times \text{}}$$

⇒ for 2 engines of equal displacement & , one w/ higher mep would produce the greater net work and if the engines run @ same speed, greater power



In a 4-stroke internal combustion engine, the piston executes 4 distinct strokes within the cylinder for every 2 revolutions of the crankshaft.

- (1) w/ intake valve open, piston makes intake stroke to draw fresh charge into cylinder
- (2) w/ both valves closed, piston undergoes compression stroke, raising temp + press of charge... requires work input from piston to cylinder contents... combustion process then initiated, resulting in high press + high temp gas mixture

- (3) Power stroke follows compression stroke during which gas mixture expands and work is done on piston as it returns to BDC
- (4) Piston then executes exhaust stroke, burned gases purged from cylinder through open exhaust valve

Air Standard Analysis

- fixed amount of air modeled as IG
- combustion process is replaced by heat xfer from external source
- no exhaust and intake processes as in an actual engine; cycle completed by constant-T heat xfer process taking place while piston is at BDC
- all processes are internally reversible

Cold Air-Standard Analysis

- specific heats are assumed constant @ their ambient temperature values

IDEAL GAS MODEL REVIEW

$$pV = RT \quad PV = mRT$$

constant sp. heats
Tables A-20, 21

$$u(T_2) - u(T_1) = C_V(T_2 - T_1)$$

$$h(T_2) - h(T_1) = C_p(T_2 - T_1)$$

$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}}$$

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\frac{k-1}{k}}$$

$$\frac{P_2}{P_1} = \left(\frac{V_1}{V_2}\right)^k$$

$$k = \frac{C_p}{C_V}$$

$$u(T_2) - u(T_1) = \int_{T_1}^{T_2} C_V(T) dT$$

$$h(T_2) - h(T_1) = \int_{T_1}^{T_2} C_p(T) dT$$

variable sp. heats
Tables A-22, 23

$$\left[\begin{array}{l} \text{AIR ONLY} \\ \frac{P_2}{P_1} = \frac{P_{r2}}{P_{r1}} \quad \frac{V_2}{V_1} = \frac{V_{r2}}{V_{r1}} \end{array} \right]$$

constant sp. heats

$$s(T_2, V_2) - s(T_1, V_1) = C_V \ln\left(\frac{T_2}{T_1}\right) + R \ln\left(\frac{V_2}{V_1}\right)$$

$$s(T_2, P_2) - s(T_1, P_1) = C_p \ln\left(\frac{T_2}{T_1}\right) - R \ln\left(\frac{P_2}{P_1}\right)$$

variable sp. heats

$$s(T_2, P_2) - s(T_1, P_1) = s^\circ(T_2) - s^\circ(T_1) - R \ln\left(\frac{P_2}{P_1}\right)$$