

FINITE VOLUME METHOD

Degroot's Notes

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1 INTRODUCTION

1.1 Generic Conservation Equation

Consider the following generic conservation equation:

$$\frac{\partial \phi}{\partial t} + \nabla \cdot (\mathbf{u}\phi) + \nabla \cdot \mathbf{J}_\phi = S_\phi \quad (1)$$

where the variables are defined as:

Variable	Description
ϕ	Generic variable
t	Time
\mathbf{u}	Velocity vector
\mathbf{J}_ϕ	Diffusive flux of ϕ
S_ϕ	Volumetric source/sink of ϕ

1.2 Mass Conservation Equation

Setting $\phi = \rho$, where ρ is the density. Also, mass conservation of a continuous substance does not have diffusive flux => $\mathbf{J}_\rho = 0$.

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\mathbf{u}\rho) = S_\rho \quad (2)$$

Notes:

- For incompressible, constant density flow:
 - $\frac{\partial \rho}{\partial t} = 0$
 - $\nabla \cdot (\mathbf{u}\rho) = \rho \nabla \cdot \mathbf{u}$
 - Result in: $\nabla \cdot \mathbf{u} = \frac{S_\rho}{\rho}$
 - And if no source/sink => $\nabla \cdot \mathbf{u} = 0$

1.3 Momentum Conservation Equation

Setting $\phi = \rho\mathbf{u}$. Diffusive flux term $\mathbf{J}_u = -\nabla \cdot \sigma$, where σ is the fluid stress tensor.

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \nabla \cdot (\rho\mathbf{u}\mathbf{u}) = \nabla \cdot \sigma + S_u \quad (3)$$

The stress tensor, σ can be expressed in terms of pressure (p) and viscous stress tensor (τ) and identity matrix, I :

$$\sigma = -p\mathbf{I} + \tau \quad (4)$$

After substituting, we get the following form of the momentum conservation equation:

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \nabla \cdot (\rho\mathbf{u}\mathbf{u}) = -\nabla p + \nabla \cdot \tau + S_u \quad (5)$$

For incompressible Newtonian fluid, we can rewrite τ in terms of the dynamic viscosity, μ $\tau = \mu(\nabla\mathbf{u} + \nabla\mathbf{u}^T)$. Thus, a momentum conservation equation for incompressible, Newtonian fluid, constant velocity:

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \nabla \cdot (\rho\mathbf{u}\mathbf{u}) = -\nabla p + \mu\nabla^2\mathbf{u} + S_u \quad (6)$$

1.4 Energy Conservation Equation

Setting $\phi = \rho h$ with h being the specific enthalpy of a substance at a given state. Thus, the unit for ϕ is energy per unit volume. The diffusive flux is given by Fourier's law: $J = -k\nabla\mathbf{T}$ with k is the thermal conductivity.

$$\frac{\partial(\rho h)}{\partial t} + \nabla \cdot (\rho \mathbf{u} h) = \nabla \cdot (k \nabla \mathbf{T}) + S_h \quad (7)$$

If we assume:

- incompressible flow
- constant specific heat capacity, $h = c_p \mathbf{T}$
- constant thermophysical properties (k and ρ)
- no source term

$$\frac{\partial(\mathbf{T})}{\partial t} + \nabla \cdot (\mathbf{u} \mathbf{T}) = \alpha \nabla^2 \mathbf{T} \quad (8)$$

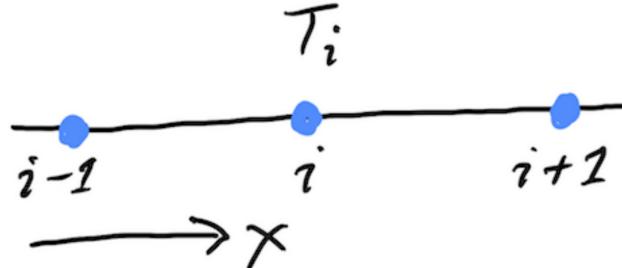
where $\alpha = \frac{k}{\rho c_p}$ is the thermal diffusivity.

1.5 Discretization of the Generic Conservation Equation

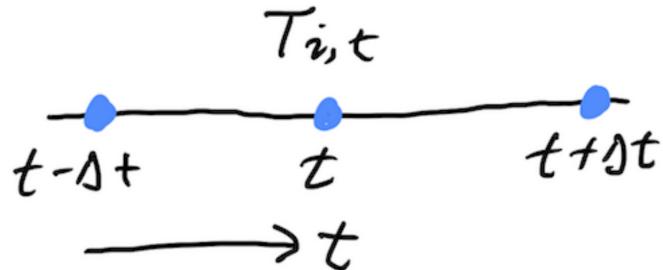
Our generic variable, ϕ is a function of spatial and time: $\phi = \phi(\mathbf{x}, t)$, where $\mathbf{x} = (x, y, z)$. Note that spatial variable can be influenced "one way" or "two way", i.e.

- One way: changes in ϕ only occur due to change on one side of that location
- Two way: changes in ϕ occur due to changes on both sides of that location.

For example, heat conduction in the image below at cell i is influenced by cell $i-1$ and $i+1$. Here, \mathbf{x} is a two way coordinate for heat conduction



Now consider transient heat convection/conduction. The temperature at any given time is influenced by existing conditions before that point in time. Here, t is a one way coordinate for transient heat conduction/convection



Recall our generic conservation equation:

$$\frac{\partial \phi}{\partial t} + \nabla \cdot (\mathbf{u} \phi) + \nabla \cdot \mathbf{J}_\phi = S_\phi$$

We consider the diffusion term or **elliptic PDE** : $\boxed{\nabla \cdot \mathbf{J}_\phi}$ to be two-way in space

Likewise, the convection term or **parabolic PDE** : $\boxed{\nabla \cdot (\mathbf{u} \phi)}$ to be one-way in space

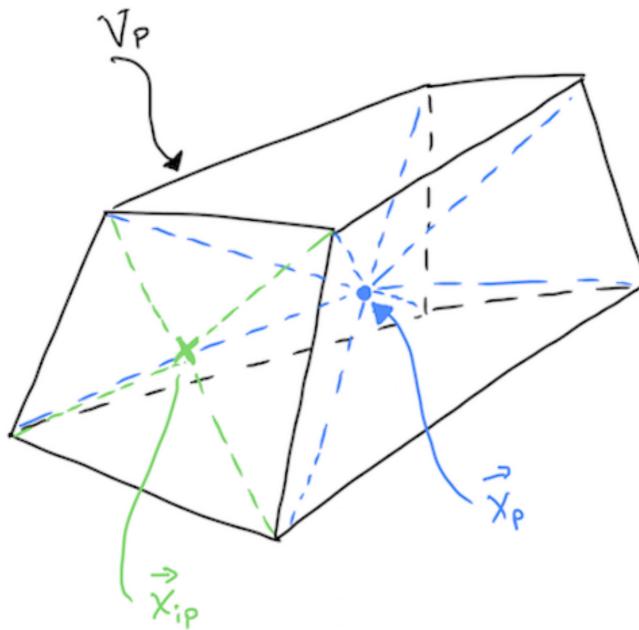
1.6 Main idea behind Discretization

Our goal is to:

- replace the PDEs' continuous solution with discrete solution, at specific location that approximates the continuous solution suitably.

For finite volume:

1. domain is split into non overlapping finite regions that fill the domain
2. the discrete point is at the centroid of each control volume with volume V_p , at position \mathbf{x}_p
3. surround these cells, we have the "faces". At the center of these "faces", we have the integration point at position \mathbf{x}_{ip}
4. the governing equations are then integrated over a control volume, where surface flux terms and volume source terms are balanced.



1.7 Determine Cell Centre + Face Integration Points

Cell centre => location of solution variables.

Points on face => fluxes are evaluated.

Consider a volume integral of a quantity ϕ , we may express this integral in discrete form as follow:

$$\int_V \phi dV \approx \phi_P V_p \quad (9)$$

where ϕ_P is the value of ϕ at some internal within V and V_p is total volume of the cell:

$$V_p = \int_V dV \quad (10)$$

To prove the above result, we expand ϕ in a Taylor series about the point P .

$$\phi \approx \phi_P + \nabla \phi_P (\mathbf{x} - \mathbf{x}_P) + \nabla^2 \phi_P (\mathbf{x} - \mathbf{x}_P)(\mathbf{x} - \mathbf{x}_P) + \dots O(\delta^3) \quad (11)$$

with δ being the characteristic grid spacing. Substitute this into our assumed expression for V_p :

$$\int_V \phi dV \approx \int_V [\phi_P + \nabla \phi_P (\mathbf{x} - \mathbf{x}_P) + \nabla^2 \phi_P (\mathbf{x} - \mathbf{x}_P)(\mathbf{x} - \mathbf{x}_P) + \dots O(\delta^3)] dV \quad (12)$$

We note that ϕ_P and its derivatives are constants:

$$\int_V \phi dV \approx \phi_P dV + \nabla \phi_P \int_V (\mathbf{x} - \mathbf{x}_P) dV + \nabla^2 \phi_P \int_V (\mathbf{x} - \mathbf{x}_P)(\mathbf{x} - \mathbf{x}_P) dV + \dots O(\delta^3) \quad (13)$$

Because our \mathbf{x}_P point is at centroid, so $\int_V (\mathbf{x} - \mathbf{x}_P) dV = 0$. Likewise, the last term is also neglected, resulting in:

$$\int_V \phi dV \approx [\phi_V + O(\delta^2)] V_P \quad (14)$$

This means that there is a second order error when approximating the cell volume in this way. This is OK because the accuracy of the method is also second order.

Note: If our $\{\mathbf{x}_P\}$ does not lie at the centroid of the cell. The second term, $\int_V (\mathbf{x} - \mathbf{x}_P) dV$ does not go to zero, making our approximation to be 1st order, which is worse.

1.8 Transient term

Here we deal with the transient term, $\frac{\partial \phi}{\partial t}$. Discretization of this term relies on:

- order of accuracy
- implicit vs explicit

The idea is to integrate this term over control volume V_P and some time step $\Delta t = t_1 - t_0$ to get the formula for the discretization.

$$\int_{t_0}^{t_1} \int_V \frac{\partial \phi}{\partial t} dV dt \approx (\phi V_P)^{t_1} - (\phi V_P)^{t_0} \quad (15)$$

1.9 Advection term

Here we deal with the advection term, $\nabla \cdot (\mathbf{u}\phi)$. Similar to the transient term, the formula for the discretization can be obtained by integrating over the control volume V_P . We also employ Gauss' theorem to convert volume integral to surface integral:

$$\int_V \nabla \cdot (\mathbf{u}\phi) dV = \int_S (\mathbf{u}\phi) \cdot \mathbf{n} dS \quad (16)$$

For the surface integral, we approximate by summing up over the faces surrounding the cell, each with area A_{ip} .

$$\int_S (\mathbf{u}\phi) \cdot \mathbf{n}_{ip} dS \approx \sum_{i=0}^{N_{ip}-1} \mathbf{u}_{ip} \cdot \mathbf{n}_{ip} \phi_{ip} A_{ip} \quad (17)$$

Note:

- using C program notation, so we sum from 0 till $N_{ip} - 1$
- approximate \mathbf{u}_{ip} by many interpolation methods
- interpolating ϕ_{ip} carefully to obtain stable numerical method.

1.10 Diffusion term

Now, we deal with the diffusion term, $\nabla \cdot \mathbf{J}_\phi$. Similar to the advection term, we integrate over a control volume, then apply Gauss' theorem

$$\int_V \nabla \cdot \mathbf{J}_\phi dV = \int_S \mathbf{J}_\phi \cdot \mathbf{n} dS \quad (18)$$

Again, the surface integral is approximated as discrete sum over the faces surrounding the cell:

$$\int_S \mathbf{J}_\phi \cdot \mathbf{n} dS \approx \sum_{i=0}^{N_{ip}-1} \mathbf{J}_{\phi,ip} \cdot \mathbf{n}_{ip} \mathbf{A}_{ip} \quad (19)$$

where the flux, $\mathbf{J}_{\phi,ip}$ is interpolated from neighboring cell values.

1.11 Source term

Recall our source term: S_ϕ , we assume that the source term is piecewise continuous, with one specific value, S_ϕ , being represented by each cell. We can then write:

$$\int_V S_\phi dV \approx S_\phi V_P \quad (20)$$

Generally, the source term may depend on ϕ so linearization is needed to obtain stable numerical method.

1.12 Linearization

With regard to our last point about J_ϕ , the discretized terms depend non linearly on the solution. This non-linearity is caused by:

- source term depend non linearly on primitive variable, e.g. J_ϕ .
- non linearities in the governing equation, e.g. advection term $\nabla \cdot (\mathbf{u}\phi)$
- on non-orthogonal grid, gradient correction terms are needed <= these are non linear.

To linearize, we assume the governing PDE is represented by the following general differential operator

$$L(\phi^*) = 0 \quad (21)$$

where:

- ϕ^* = the continuous solution to the PDE
- Note that to solve a PDE using finite volume, the continuous solution ϕ^* is approximated by the discrete solution vector $\phi \in \mathbb{R}^N$ on N number of control volume. Our PDE is then integrated over each control volume and each term in the governing equation is approximated using the discrete solution ϕ
- Of course, the numerical solution will not satisfy the discretized equation exactly; rather we will have a residual, $\mathbf{r} \in \mathbb{R}^N$.

We expand the residual about the solution ϕ_i at iteration i , and find the solution where $r = 0$:

$$\mathbf{r}(\phi_i) + \left. \frac{\partial \mathbf{r}}{\partial \phi} \right|_{\phi_i} (\phi - \phi_i) = 0 \quad (22)$$

We define the **Jacobian of the residual vector** as:

$$\mathbf{J}(\phi) = \frac{\partial \mathbf{r}}{\partial \phi} \quad (23)$$

We use this to update according to fix point iteration:

$$\phi = \phi_i + \Delta\phi \quad (24)$$

where:

$$\Delta\phi = (\phi - \phi_i) \quad (25)$$

and:

$$\mathbf{J}(\phi_i)\Delta\phi = -\mathbf{r}(\phi_i) \quad (26)$$

The remaining unknowns are: the residual vector \mathbf{r} and Jacobian matrix $\mathbf{J}(\phi_i)$.

Note: we can express the linear system for a control volume P as:

$$a_P \delta\phi_P + \sum_{nb} a_{nb} \delta\phi_{nb} = -r_P \quad (27)$$

where nb is sum over all neighboring cells. The coefficients are defined as:

$$a_P = \frac{\partial r_P}{\partial \phi_P} \quad (28)$$

$$a_{nb} = \frac{\partial r_P}{\partial \phi_{nb}} \quad (29)$$

1.13 Four Basic Rules

Outlined by Patankar(1980), these 4 rules are:

- **Rule 1: Consistency at control volume faces**

For common faces between cells, the flux through those common faces must be the same when evaluated at each cell. If this is not the case, then it means there is an artificial source of the energy at the face.

- **Rule 2:** $\alpha_p > 0$ and $\alpha_{nb} < 0$

Consider situations involving only convection and diffusion and all other conditions unchanged: if ϕ in 1 cell increases, then we can expect ϕ in the neighboring cells to increase as well. The only way that this could happen is in this equation

$$a_P \delta\phi_P + \sum_{nb} a_{nb} \delta\phi_{nb} = -r_P$$

a_p must have opposite sign from each of its a_{nb} coefficients, just so that $\delta\phi_p$ and $\delta\phi_{nb}$ have the same signs and r_p is unchanged.

- **Rule 3: Negative slope linearization of source terms**

Suppose we have a source term in the form: $S_\phi = a + b\phi_P$. If this is moved to the LHS, the coefficient a_P can be negative if b is positive. So we require that $b < 0$, or negative slope linearization. The idea is that a positive slope linearization would be unstable because the source would cause the variables to increase, which would then increase the source term. This would continue indefinitely and without bounds. In terms of heat transfer, we can have a heat source that grows with temperature and also a heat sink for removal of temperature. This is done to avoid an uncontrolled increase in temperature.

- **Rule 4: Sum of neighboring coefficients**

Our governing equations contain derivatives of dependent variables, i.e. both ϕ and $\phi + c$ will satisfy the same governing equations. Thinking practically, this means that temperature field in both Kelvin and Celcius would both satisfy the same discretized equations, because Celcius and Kelvin scale are related via a constant. For this to be true, we require:

$$a_p = - \sum_{nb} a_{nb}$$

Note that in the case of the linearization of the source term above, it indicates that same equation cannot be used for both ϕ and $\phi + c$. So in this case, make sure to modify the source term coefficients appropriately.

2 STEADY DIFFUSION EQUATIONS

2.1 Problem Definition

We consider the solution of a steady, 1D heat diffusion equation

$$-k\nabla^2T - S = 0 \quad (30)$$

2.2 Discretization

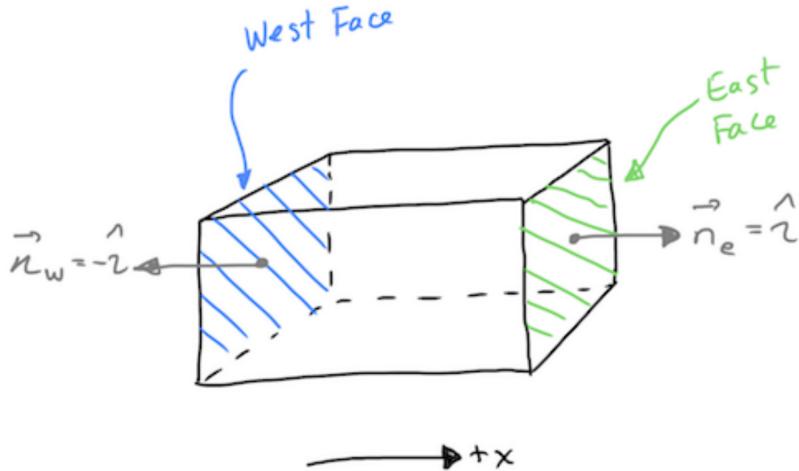
Recall our diffusion term can be discretized as:

$$\int_S \mathbf{J} \cdot \mathbf{n} dS \approx \sum_{i=0}^{N_{ip}-1} \mathbf{J}_{ip} \cdot \mathbf{n}_{ip} A_{ip} \quad (31)$$

Our flux \mathbf{J} here is the diffusive flux, so: $\mathbf{J} = -k\nabla T$. Thus:

$$\int_S \mathbf{J} \cdot \mathbf{n} dS \approx - \sum_{i=0}^{N_{ip}-1} k_{ip} \nabla T_{ip} \cdot \mathbf{n}_{ip} A_{ip} \quad (32)$$

We assume constant thermal conductivity, $k_{ip} = k$. A 1D control volume, with West/East faces and unit vectors drawn, is shown below:



Since we are in 1D, our unit vector is in the \mathbf{i} only.

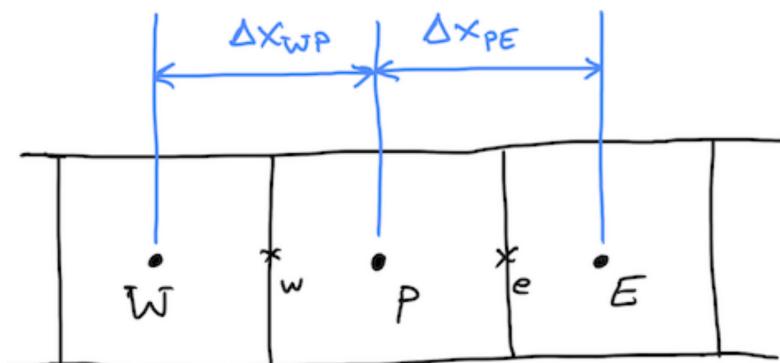
Thus, $\nabla T \cdot \mathbf{n} = \nabla T \cdot \mathbf{i}$.

But, $\nabla T \cdot \mathbf{i} = \left\langle \frac{\partial T}{\partial x} \mathbf{i} + \frac{\partial T}{\partial y} \mathbf{j} + \frac{\partial T}{\partial z} \mathbf{k} \right\rangle \cdot (1\mathbf{i} + 0\mathbf{j} + 0\mathbf{k}) = \frac{\partial T}{\partial x}$.

With these points in mind, the discretization for the diffusion term is simplified to:

$$\int_S \mathbf{J} \cdot \mathbf{n} dS \approx k \left. \frac{\partial T}{\partial x} \right|_w A_w - k \left. \frac{\partial T}{\partial x} \right|_e A_e \quad (33)$$

The diagram below shows the cell locations and the nomenclature for the distance between them, note how Δx is center-center



We apply finite differences to the derivatives in the diffusion term, i.e.:

$$k \frac{\partial T}{\partial x} \Big|_w A_w - k \frac{\partial T}{\partial x} \Big|_e A_e = k \frac{T_p - T_w}{\Delta x_{WP}} A_w - k \frac{T_E - T_p}{\Delta x_{PE}} A_e \quad (34)$$

Our discretized source term is simply:

$$\int_V S dV \approx S_p V_p \quad (35)$$

where S_p = value of source term **within** the cell, and V_p = cell volume.

Put everything on one side, we can form the residual equation for the cell **P** as:

$$r_p = -k \frac{T_E - T_p}{\Delta x_{PE}} A_e + k \frac{T_p - T_w}{\Delta x_{WP}} A_w - S_p V_p \quad (36)$$

or expressing in terms of the diffusive fluxes, \mathbf{F}^d , through each face:

$$r_p = F_e^d - F_w^d - S_p V_p \quad (37)$$

where:

$$F_e^d = -k \frac{T_E - T_p}{\Delta x_{PE}} A_e = -D_e (T_E - T_p) \quad (38)$$

$$F_w^d = -k \frac{T_p - T_w}{\Delta x_{WP}} A_w = -D_w (T_p - T_w) \quad (39)$$

$$D_e = \frac{k A_e}{\Delta x_{PE}} \quad (40)$$

$$D_w = \frac{k A_w}{\Delta x_{WP}} \quad (41)$$

Our cell residual equation is then:

$$r_p = D_w (T_p - T_w) - D_e (T_E - T_p) - S_p V_p \quad (42)$$

The linearized coefficients are then calculated as:

$$a_p = \frac{\partial r_p}{\partial T_p} = D_w + D_e - \frac{\partial S_p}{\partial T_p} V_p \quad (43)$$

$$a_w = \frac{\partial r_p}{\partial T_w} = -D_w \quad (44)$$

$$a_E = \frac{\partial r_p}{\partial T_E} = -D_e \quad (45)$$

Recall that we can form an algebraic system of equation for each control volume like this:

$$a_p \delta \phi_p + \sum_{nb} a_{nb} \delta \phi_{nb} = -r_p \quad (46)$$

$$a_p \delta T_p + a_w \delta T_w + a_E \delta T_E = -r_p \quad (47)$$

The above linear system of equations can be written as as tridiagonal matrix, like this:

Equation for one cell

$$\begin{bmatrix} a_{p,0} & a_{E,0} & 0 & \cdots & \cdots & 0 \\ a_{w,1} & a_{p,1} & a_{E,1} & & & \\ 0 & a_{w,2} & a_{p,2} & a_{E,2} & & \\ \vdots & & & & \ddots & \\ 0 & \cdots & \cdots & 0 & a_{w,N} & a_{p,N} \end{bmatrix} \begin{pmatrix} \delta T_0 \\ \delta T_1 \\ \vdots \\ \delta T_N \end{pmatrix} = \begin{pmatrix} -r_0 \\ -r_1 \\ \vdots \\ -r_N \end{pmatrix}$$

Note: The first and last row only has 2 non zero elements each. This is because these are the left most/right most side and they are adjacent to the domain boundary. Therefore, special boundary conditions are needed to be set.

In matrix notation, we are solving:

$$\mathbf{Ax} = \mathbf{b} \quad (48)$$

where \mathbf{A} is the Jacobian matrix, $\mathbf{b} = -\mathbf{r}$ is the residual vector, $\mathbf{x} = \delta\mathbf{T}$ is the solution correction. At each current iteration i , the solution is updated according to:

$$\mathbf{T} = \mathbf{T}_i + \delta\mathbf{T}_i \quad (49)$$

2.3 Source Terms

Our source term can have many forms, depending on the type of heat source. We will assume *external convection* and *radiation exchange*:

- For external convection:

$$\frac{S_{conv,P}}{V_P} = -hA_0(T_P - T_{\infty,c}) \quad (50)$$

where:

- h is the convective coefficient.
- A_0 is external surface area of the cell P .
- T_P is temperature at the centroid of cell P .
- $T_{\infty,c}$ is the ambient temperature for the convection process.

- For radiation exchange:

$$\frac{S_{rad}}{V_P} = -\epsilon\sigma A_0(T_P^4 - T_{\infty,r}^4) \quad (51)$$

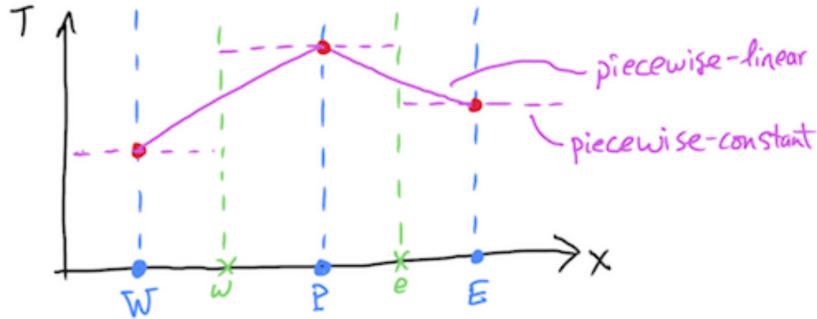
where:

- ϵ is the surface emissivity.
- σ is the Stefan-Boltzmann constant.
- $T_{\infty,r}$ is the surrounding temperature for radiation exchange.

2.4 Discussion of Discretization Procedure

2.4.1 Temperature Profile Assumptions

When computing the diffusive fluxes through the faces, we assumed a **piecewise-linear profile** for the temperature. This ensures that the derivatives are defined at the integration points and provides consistency for flux at control-volume faces. For the source term, **piece-wise constant profile** is used, implying a single value of the source term in each cell. Note that for piece-wise constant profile, the derivatives are not defined at integration points, due to jump discontinuity. So if fluxes will be inconsistent if piecewise-constant profile is used for temperature.



2.4.2 Implementation of Linearization

In Patankar's method, the solution of the linear system **is** the solution for the variables at the control volume center. In our method, the solution of the linear system is the **correction** to apply to the previous iteration of the solution. The correction method is preferred because:

- at convergence, the solution for the correction goes to zero \rightarrow zero a good initial guess for the linear solver.
- linear system involves the residual vector. In Patankar's, there are more work to calculate the residual vector.

2.4.3 Properties of the Discrete Algebraic Equations

Recall our algebraic equation for the linear system

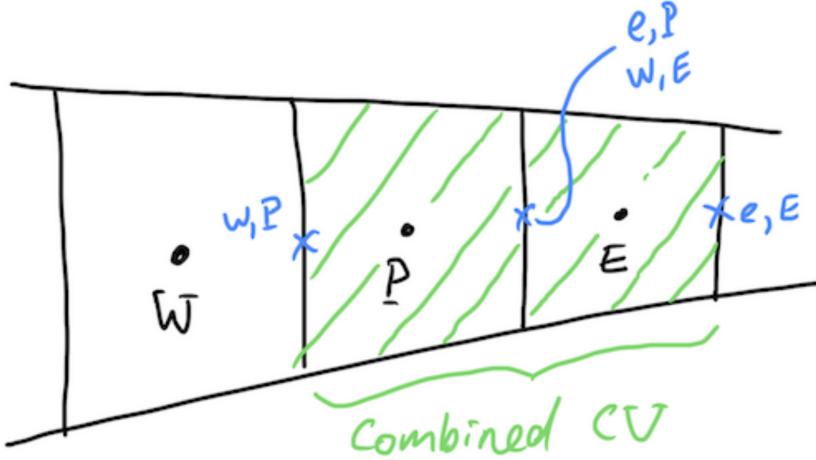
$$a_P \delta T_P + a_W \delta T_W + a_E \delta T_E = -r_P \quad (52)$$

In Rule 2, we require that $a_P > 0$ and $a_W, a_E < 0$. The reason for this is if we consider the case with no source, and the solution converge, $r_P \rightarrow 0$:

$$a_P \delta T_P = -a_W \delta T_W - a_E \delta T_E \quad (53)$$

Now, suppose both T_P and T_E are perturbed. If either of these temperatures were to rise, then T_P would also rise. Similarly, if either temperatures were to drop, T_P should also drop. Therefore, to ensure correct physical effect, if $a_P > 0$ then $a_W, a_E > 0$.

Consider the two cells (P and E) below:



At convergence, $r_P = 0$, the equation for the control volume P is:

$$F_{e,P}^d - F_{w,P}^d - S_P V_P = 0 \quad (54)$$

For the control volume E :

$$F_{e,E}^d - F_{w,E}^d - S_E V_E = 0 \quad (55)$$

Adding these equations together gives:

$$F_{e,P}^d - F_{w,P}^d + F_{e,E}^d - F_{w,E}^d - S_P V_P - S_E V_E = 0 \quad (56)$$

Note that $F_{e,P}^d = F_{w,E}^d$ by continuity, i.e. the flux at cell P going eastward should be the same flux going from westward at cell E . If these are not equal, then it implies that there is a fictitious force at the face, which is not reasonable. Therefore, our algebraic equation for control volume P and E becomes:

$$F_{e,E}^d - F_{w,P}^d - S_P V_P - S_E V_E = 0 \quad (57)$$

The above equation demonstrates integral conservation: a balance of the total source term within the combined control volume with the net diffusive flux from that same control volume. In addition, recall the definition of the diffusive flux:

$$F_{e,P}^d = -k \frac{T_E - T_P}{\Delta x_{PE}} A_{e,P} \quad (58)$$

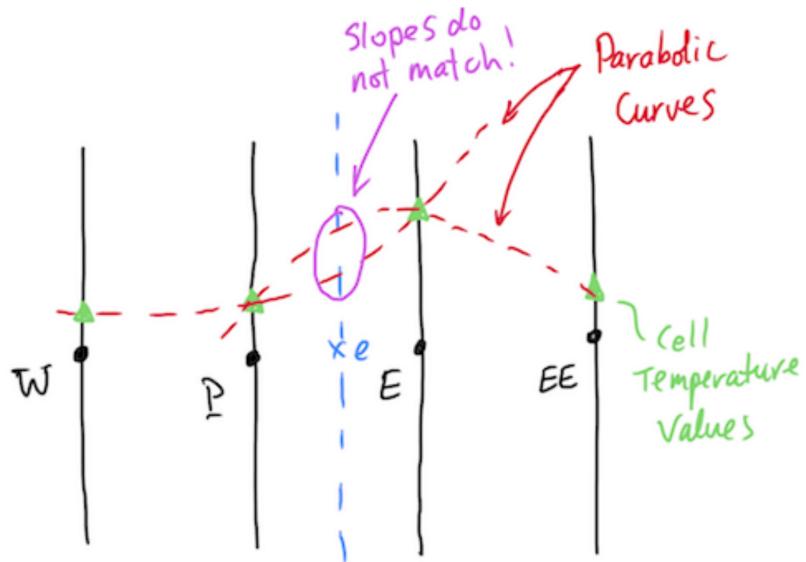
$$F_{w,E}^d = -k \frac{T_E - T_P}{\Delta x_{PE}} A_{w,E} \quad (59)$$

From the geometry of the grid, $A_{e,P} = A_{w,E}$; therefore, it is in fact the two-point finite difference estimation of the derivative that cause the fluxes to be equal. This is also due to the piecewise-linear profile that we assume. If we assume a **parabolic profile** instead, there is no guarantee that the fluxes would be equal. Instead, we would have:

$$F_{e,P}^d = f(T_W, T_P, T_E) \quad (60)$$

$$F_{w,E}^d = f(T_P, T_E, T_{EE}) \quad (61)$$

This means that the flux through the common face depends on different temperature, so we cannot be sure that the derivative from either side is consistent.



2.5 Implementation

```
# Solving the 1D Diffusion Equation: -k del_sq T = S = 0
import numpy as np
from enum import Enum
from scipy.sparse.linalg import spsolve
from scipy.sparse import csr_matrix
from numpy.linalg import norm
import sys
import matplotlib.pyplot as plt

# class Grid: defining a 1D cartesian grid
class Grid:

    def __init__(self, lx, ly, lz, ncv):
        # # constructor
        # lx = total length of domain in x direction [m]
        # ly = total length of domain in y direction [m]
        # lz = total length of domain in z direction [m]

        # store the number of control volumes
        self._ncv = ncv

        # calculate the control volume length
        dx = lx/float(ncv)

        # array of x location of face integration points
        # generate array of control volume element with length dx each
        self._xf = np.array([i*dx for i in range(ncv+1)])

        # array of x location of the cell centroid
        # Note: figure out why add a xf[-1] last item
        self._xP = np.array([self._xf[0]] +
                            [0.5*(self._xf[i]+self._xf[i+1]) for i in range(ncv)] +
                            [self._xf[-1]])

        # calculate face areas
        self._Af = ly*lz*np.ones(ncv+1)

        # calculate the outer surface area of each cell
        self._Ao = (2.0*dx*ly + 2.0*dx*lz)*np.ones(ncv)

        # calculate cell volumes
        self._vol = dx*ly*lz*np.ones(ncv)

    @property
    def ncv(self):
        # number of control volumes in domain
        return self._ncv

    @property
    def xf(self):
        # face location array
        return self._xf

    @property
    def xP(self):
        # cell centroid array
        return self._xP
```

```

@property
def dx_WP(self):
    return self.xP[1:-1]-self.xP[0:-2]

@property
def dx_PE(self):
    return self.xP[2:]-self.xP[1:-1]

@property
def Af(self):
    # Face area array
    return self._Af

@property
def Aw(self):
    # West face area array
    return self._Af[0:-1]

@property
def Ae(self):
    # East face area array
    return self._Af[1:]

@property
def Ao(self):
    # Outer face area array
    return self._Ao

@property
def vol(self):
    # Cell volume array
    return self._vol

# class ScalarCoeffs defining coefficients for the linear system
class ScalarCoeffs:
    def __init__(self,ncv):
        # # constructor:
        self._ncv = ncv
        self._aP = np.zeros(ncv)
        self._aW = np.zeros(ncv)
        self._aE = np.zeros(ncv)
        self._rP = np.zeros(ncv)

    def zero(self):
        # zero out the coefficient arrays
        self._aP.fill(0.0)
        self._aW.fill(0.0)
        self._aE.fill(0.0)
        self._rP.fill(0.0)

    def accumulate_aP(self,aP):
        # accumulate values onto aP
        self._aP += aP

    def accumulate_aW(self,aW):
        # accumulate values onto aW
        self._aW += aW

    def accumulate_aE(self,aE):
        # accumulate values onto aE
        self._aE += aE

    def accumulate_rP(self,rP):
        # accumulate values onto rP
        self._rP += rP

    @property
    def ncv(self):
        # number of control volume in domain
        return self._ncv

    @property
    def aP(self):
        # cell coefficient
        return self._aP

    @property
    def aW(self):
        # West cell coefficient
        return self._aW

    @property
    def aE(self):
        # East cell coefficient
        return self._aE

    @property

```

```

def rP(self):
    # cell coefficient
    return self._rP

# class BoundaryLocation defining boundary condition locations
class BoundaryLocation(Enum):
    WEST = 1
    EAST = 2

# Implementing Boundary Conditions
class DirichletBC:
    def __init__(self, phi, grid, value, loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # value = boundary value
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._value = value
        self._loc = loc

    def value(self):
        # return the boundary condition value
        return self._value

    def coeff(self):
        # return the linearization coefficient
        return 0

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = self._value
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = self._value
        else:
            raise ValueError("Unknown boundary location")

class NeumannBC:
    def __init__(self, phi, grid, gradient, loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # gradient = gradient at cell adjacent to boundary
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._gradient = gradient
        self._loc = loc

    def value(self):
        # return the boundary condition value
        if self._loc is BoundaryLocation.WEST:
            return self._phi[1]-self._gradient*self._grid.dx_WP[0] # T(0) = T(1) - gb*dx
        elif self._loc is BoundaryLocation.EAST:
            return self._phi[-2]-self._gradient*self._grid.dx_PE[-1] #T(-1) = T(-2) -gb*dx
        else:
            raise ValueError("Unknown boundary location")

    def coeff(self):
        # return the linearization coefficient
        return 1

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = self._phi[1] - self._gradient*self._grid.dx_WP[0]
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = self._phi[-2] + self._gradient*self._grid.dx_PE[-1]
        else:
            raise ValueError("Unknown boundary location")

class RobinBC:
    def __init__(self, phi, grid, h, k, tinfity, loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # h = convective coefficient
        # k = conductive coefficient
        # tinfity = freestream temperature
        # loc = boundary location

```

```

        self._phi = phi
        self._grid = grid
        self._h = h
        self._k = k
        self._tinfy = tinfy
        self._loc = loc

    def value(self):
        # return the boundary condition value
        if self._loc is BoundaryLocation.WEST:
            return (self._phi[1] + self._grid.dx_WP[0]*(self._h/self._k)*(self._tinfy)) \
                /(1 + self._grid.dx_WP[0]*(self._h/self._k))
        elif self._loc is BoundaryLocation.EAST:
            return (self._phi[-2] - self._grid.dx_PE[-1]*(self._h/self._k)*(self._tinfy))\
                /(1 - self._grid.dx_PE[-1]*(self._h/self._k))
        else:
            raise ValueError("Unknown boundary location")

    def coeff(self):
        # return the linearization coefficient
        return 1

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = (self._phi[1] + self._grid.dx_WP[0]*(self._h/self._k)*(self._tinfy))\
                /(1 + self._grid.dx_WP[0]*(self._h/self._k))
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = (self._phi[-2] - self._grid.dx_PE[-1]*(self._h/self._k)*(self._tinfy))\
                /(1 - self._grid.dx_PE[-1]*(self._h/self._k))
        else:
            raise ValueError("Unknown boundary location")

# class DiffusionModel defining a diffusion model
class DiffusionModel:
    def __init__(self, grid, phi, gamma, west_bc, east_bc):
        # constructor
        self._grid = grid
        self._phi = phi
        self._gamma = gamma      # gamma here is "k" in the heat model
        self._west_bc = west_bc
        self._east_bc = east_bc

    # add diffusion terms to coefficient arrays
    def add(self, coeffs):

        # calculate west/east diffusion flux for each face
        flux_w = -self._gamma * self._grid.Aw * (self._phi[1:-1] - self._phi[0:-2]) \
            / self._grid.dx_WP # Fw = k*Aw*(Tp-Tw)/dx_WP, each in [1:-1] minus each in [0:-2] aka Tp - Tw (on the left)
        flux_e = -self._gamma * self._grid.Ae * (self._phi[2:] - self._phi[1:-1]) \
            / self._grid.dx_PE # Fe = k*Ae*(Te-Tp)/dx_PE, each in [2:] minus each in [1:-1] aka Te - Tp (on the right)

        # calculate the linearized coefficient [aW, aP, aE]
        coeffW = -self._gamma * self._grid.Aw / self._grid.dx_WP # -Dw or -kAw/dxWP
        coeffE = -self._gamma * self._grid.Ae / self._grid.dx_PE # -De or -kAe/dxPE
        coeffP = -coeffW - coeffE                                # double negative to get plus

        # modified the linearized coefficient on the boundaries
        coeffP[0] = coeffP[0] + coeffW[0]*self._west_bc.coeff()
        coeffP[-1] = coeffP[-1] + coeffE[-1]*self._east_bc.coeff()

        # zero the coefficients that are not used (on the boundary)
        coeffW[0] = 0.0
        coeffE[-1] = 0.0

        # calculate the net flux from each cell (out -in)
        flux = flux_e - flux_w

        # add to coefficient arrays
        coeffs.accumulate_aP(coeffP)
        coeffs.accumulate_aW(coeffW)
        coeffs.accumulate_aE(coeffE)
        coeffs.accumulate_rP(flux)

        # return the coefficient arrays
        return coeffs

    # function to return a sparse matrix representation of a set of scalar coefficients
    def get_sparse_matrix(coeffs):
        ncv = coeffs.ncv
        data = np.zeros(3*ncv-2)
        rows = np.zeros(3*ncv-2, dtype = int)
        cols = np.zeros(3*ncv-2, dtype = int)

```

```

data[0] = coeffs.aP[0]
rows[0] = 0
cols[0] = 0

if ncv > 1:
    data[1] = coeffs.aE[0]
    rows[1] = 0
    cols[1] = 1

for i in range(ncv-2):
    data[3*i+2] = coeffs.aW[i+1]
    data[3*i+3] = coeffs.aP[i+1]
    data[3*i+4] = coeffs.aE[i+1]

    rows[3*i+2:3*i+5] = i+1

    cols[3*i+2] = i
    cols[3*i+3] = i+1
    cols[3*i+4] = i+2

if ncv > 1:
    data[3*ncv-4] = coeffs.aW[-1]
    data[3*ncv-3] = coeffs.aP[-1]

    rows[3*ncv-4:3*ncv-1] = ncv-1

    cols[3*ncv-4] = ncv-2
    cols[3*ncv-3] = ncv-1

return csr_matrix((data, (rows,cols)))

# solve the linear system and return the field variables
def solve(coeffs):
    # get the sparse matrix
    A = get_sparse_matrix(coeffs)
    # solve the linear system
    return spsolve(A, -coeffs.rP)

# Solving the 1D steady conduction with Dirichlet BC
# Initially , east = 300K, west = 300K
# define the grid
lx = 1.0
ly = 0.1
lz = 0.1
ncv = 10
mygrid = Grid(lx,ly,lz,ncv)

#set the max iterations and convergence criterion
maxiter = 100
converged=1e-6

# thermal properties
k = 0.1

# coefficients
coeffs = ScalarCoeffs(mygrid.ncv)

# # initial condition
T0 = 300

# # initialize field variable arrays
T = T0*np.ones(mygrid.ncv+2)

# # # boundary condition
west_bc = DirichletBC(T, mygrid, 400, BoundaryLocation.WEST)
east_bc = DirichletBC(T, mygrid, 0, BoundaryLocation.EAST)

# # # apply boundary conditions
west_bc.apply()
east_bc.apply()

# # # list to store the solution at each iteration
T_solns = [np.copy(T)]

# # # define the diffusion model
diffusion = DiffusionModel(mygrid, T, k, west_bc, east_bc)

avgResidList = []
iterList = []

# # # iterate until the solution is converged
for i in range(maxiter):

```

```

# zero the coeff and add
coeffs.zero()
coeffs = diffusion.add(coeffs)

# compute residual and check for convergence
maxResid = norm(coeffs.rP, np.inf)
avgResid = np.mean(np.absolute(coeffs.rP))
print("Iteration = {} ; Resid = {} ; Avg. Resid = {}".format(i,maxResid, avgResid) )
if maxResid < converged:
    break

# solve the sparse matrix system
dT = solve(coeffs)

# update the solution and boundary conditions
T[1:-1] = T[1:-1] + dT
west_bc.apply()
east_bc.apply()

# store the solution
T_solns.append(np.copy(T))

avgResidList.append(avgResid)
iterList.append(i)

# plotting
for Ti in T_solns:
    plt.plot(mygrid.xP, Ti, label = str(i))
    i += 1

# plt.title('Explicit')
# plt.xlabel('X')
# plt.ylabel('T')
# plt.show()
# plt.savefig('pic/1d_transient_explicit.png')

# plt.title('Residual')
# plt.plot(avgResidList, iterList)
plt.show()

```

3 TRANSIENT 1D HEAT DIFFUSION

3.1 Problem Definition

In contrast to the steady case, here we are solving the following transient 1D heat diffusion equation

$$\frac{\partial(\rho c_p T)}{\partial t} = k \nabla^2 T + S \quad (62)$$

assuming constant ρ and c_p

3.2 Discretization

Integrating the governing equation through space and time yields:

$$\int_{t_0}^{t_1} \int_V \frac{(\partial \rho c_p T)}{\partial t} dt dV = \int_{t_0}^{t_1} \int_V k \nabla^2 T dt dV + \int_{t_0}^{t_1} \int_V S dt dV \quad (63)$$

By assuming a timestep $\Delta t = t_1 - t_0$, we also assume that the solution is stored at time levels t and later at $t + \Delta t$. Thus, we can assume various profiles for the integrands as functions of time. Here, we will examine the following:

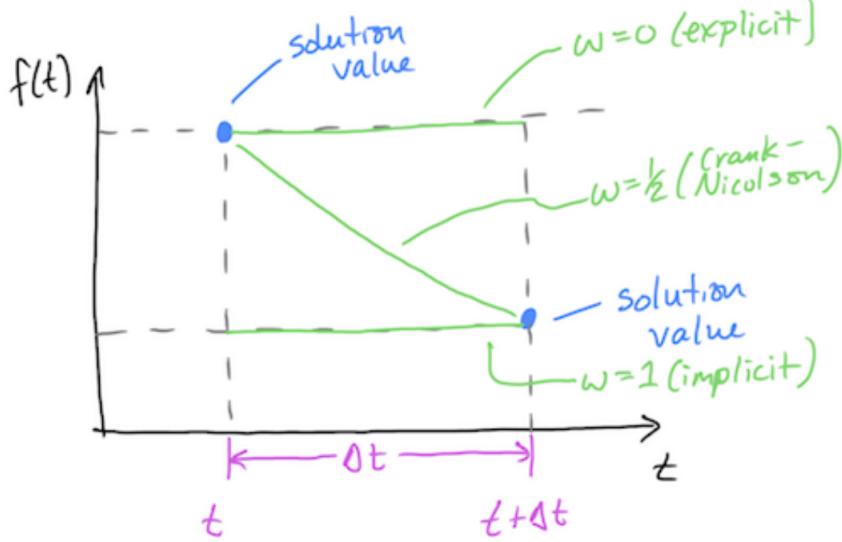
- **Fully explicit:** evaluate the integrands on RHS at initial time level, $t_0 = t$.
- **Fully implicit:** evaluate the integrands on RHS at final time level, $t_1 = t_0 + \Delta t$
- **Crank Nicolson:** assume a linear profile of the RHS over the interval Δt .

For the LHS, we interchange the order of integration, which results in this, for a control volume \mathbf{P} :

$$\int_{t_0}^{t_1} \int_V \frac{(\partial \rho c_p T)}{\partial t} dt dV = (\rho c_p T_p)^{t+\Delta t} - (\rho c_p T_p)^t \quad (64)$$

Recall that for the steady case, the diffusive term is the difference in flux. Here, we add a weighting function w to control the assumed variation of the integrand over the timestep.

$$\int_{t_0}^{t_1} \int_V k \nabla^2 T dt dV = - [\omega(F_e^d)^{t+\Delta t} + (1-\omega)(F_e^d)^t] \Delta t + [\omega(F_w^d)^{t+\Delta t} + (1-\omega)(F_w^d)^t] \Delta t \quad (65)$$



Grouping the time levels together:

$$\int_{t_0}^{t_1} \int_V k \nabla^2 T dt dV = [\omega [F_w^d - F_e^d]^{t+\Delta t} + (1-\omega) [F_w^d - F_e^d]^t] \Delta t \quad (66)$$

Assuming no source term, $S = 0$, constant properties and divide by Δt , our discretized equation becomes

$$\rho c_p \frac{T_P^{t+\Delta t} - T_P^t}{\Delta t} V_P = \omega [F_w^d - F_e^d]^{t+\Delta t} + (1-\omega) [F_w^d - F_e^d]^t \quad (67)$$

Note

- fully implicit and fully explicit are *1st order* accurate in time.
- Crank-Nicolson are *2nd order* accurate in time.
 - but Crank-Nicolson are *less stable*, cause oscillations.
- Generally, explicit solutions do not require the solution of a system of equations. All diffusive fluxes are calculated using solution values from previous timestep
- Implicit requires solution of a linear system at current time step. The same is true for Crank-Nicolson, or any scheme where $0 < \omega < 1$

3.3 Analysis of Explicit Scheme

Recall explicit scheme means $\omega = 0$. Our discretized equation becomes:

$$\rho c_p \frac{T_P^{t+\Delta t} - T_P^t}{\Delta t} V_P = [F_w^d - F_e^d]^t + \rho c_p \frac{T_P^t}{\Delta t} V_P \quad (68)$$

Recall that the fluxes can be defined as:

$$F_e^d = -k \frac{T_E - T_P}{\Delta x_{PE}} A_e = -D_e (T_E - T_P) \quad (69)$$

$$F_w^d = -k \frac{T_P - T_W}{\Delta x_{WP}} A_w = -D_w (T_P - T_W) \quad (70)$$

$$D_e = \frac{k A_e}{\Delta x_{PE}} \quad (71)$$

$$D_w = \frac{k A_w}{\Delta x_{WP}} \quad (72)$$

Thus, our explicit formulation becomes:

$$\rho c_p \frac{T_P^{t+\Delta t} - T_P^t}{\Delta t} V_P = D_e T_E^t + D_w T_W^t + \left(\frac{\rho c_p V_P}{\Delta t} - D_e - D_w \right) T_P^t \quad (73)$$

To get correct physical influence, the coefficient of T_P^t must be positive, to ensure $\uparrow T_P^t$ leads to $\uparrow T_P^{t+\Delta t}$. Therefore, our timestep must be selected such that:

$$\frac{\rho c_p V_P}{\Delta t} \geq D_e + D_w$$

or

$$\Delta t \leq \frac{\rho c_p V_P}{D_e + D_w} = \frac{1}{\frac{D_e}{\rho c_p V_P} + \frac{D_w}{\rho c_p V_P}}$$

Simplifying further, we assume $V_P = A\Delta x$ where A is the cross-sectional area of the domain at P , and Δx is the grid spacing. Also assuming $A_e, A_w \approx A$

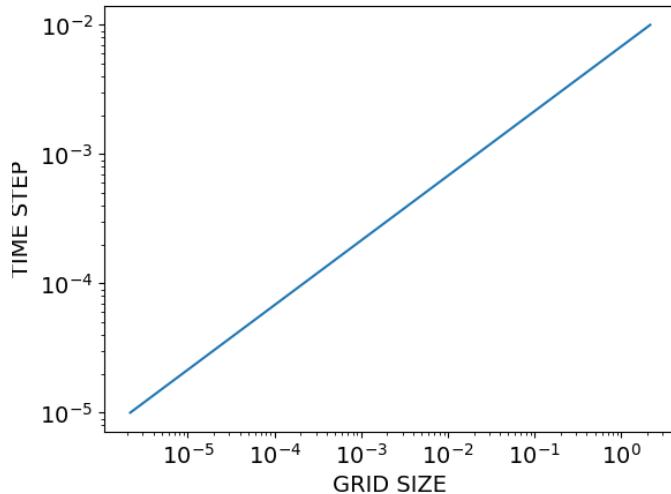
$$\frac{D_e}{\rho c_p V_P} \approx \frac{\frac{k A_e}{\Delta x}}{\rho c_p A \Delta x} = \frac{k}{\rho c_p} \frac{1}{\Delta x^2} = \frac{\alpha}{\Delta x^2}$$

The quantity $\frac{\Delta x^2}{\alpha}$ may be interpreted as the timescale associated with conduction through the face. For uniform grid, $A_e = A_w = A$, the timestep restriction is:

$$\Delta t \leq \frac{1}{\frac{\alpha}{\Delta x^2} + \frac{\alpha}{\Delta x^2}} = \frac{\Delta x^2}{2\alpha} \quad (74)$$

Note how our timestep is related to the square of the grid size, so a fine grid will have a very small " Δx^2 ". To study this, we consider an iron bar with $\alpha = 23.1 \times 10^{-6} [m^2/s]$ at various grid sizes:

GRID SIZE [m]	TIME STEP [sec]
0.01	2.1645022
0.001	0.021645022
0.0001	2.1645022e-4
0.00001	2.1645022e-6



We can quickly see how the timestep restriction gets worse with increasing grid refinement. As a result, implicit methods are more commonly used in practice. Exception would be calculation of turbulent flow using direct numerical simulation (DNS). In this case, explicit method are a good choice because they are less expensive per timestep, since no linear system must be solved.

3.4 Analysis of Fully Implicit Scheme

Setting $\omega = 1$ for implicit scheme. Our discretized equation becomes:

$$\rho c_p \frac{T_P^{t+\Delta t} - T_P^t}{\Delta t} V_P = [-D_w(T_P - T_W) + D_e(T_E - T_P)]^{t+\Delta t} \quad (75)$$

Drop the superscript $t + \Delta t$ and denotes old value as "o", after rearranging, we have:

$$\left(\frac{\rho c_p V_P}{\Delta t} + D_w + D_e \right) T_P = D_w T_W + D_e T_E + \frac{\rho c_p V_P}{\Delta t} T_P^o \quad (76)$$

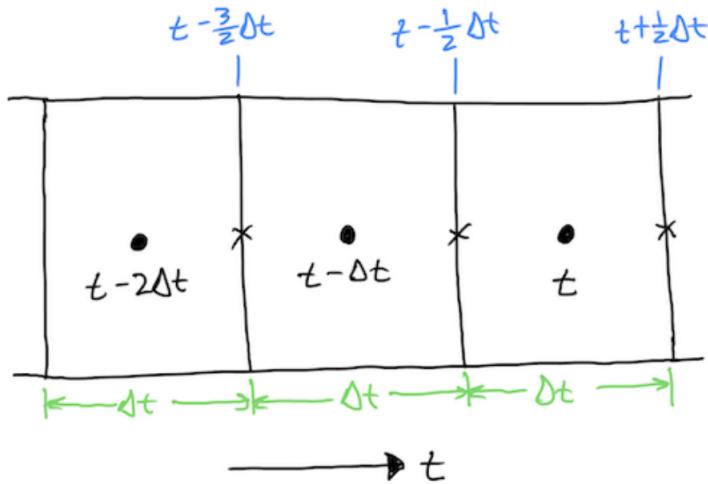
We can see that none of the coefficients can be come negative when written out this way. Still, we must ensure the timestep is small enough to resolve all transient phenomena. In contrast to this fully implicit scheme, the Crank-Nicholson scheme has no formal restriction on Δt , but still produces **oscillatory** at large Δt .

3.5 Derivation of Second Order Implicit Scheme

3.5.1 General idea

We consider integration over a special **space-time control volume**, with:

- the time faces being located at $t - \Delta t/2, t + \Delta t/2$
- the solution values are stored at $t, t - \Delta t, t - 2\Delta t$.



By using the space-time control volume, the RHS of the discretized equation evaluated at time t , can be considered as a representative of the entire timesweep. The advantages are:

- we do not need to assume a profile in time, e.g. piecewise constant for fully implicit/explicit, piecewise linear for Crank-Nicholson.
- no need to store old flux value
- interpolation depends on the face values:
 - if piecewise constant \rightarrow 1st order scheme
 - if piecewise linear \rightarrow 2nd order scheme

3.5.2 Derivation

Integrate the governing equation over the space-time control volume

$$\int_{t-\Delta t/2}^{t+\Delta t/2} \int_V \frac{(\partial \rho c_p T)}{\partial t} dt dV = \int_{t-\Delta t/2}^{t+\Delta t/2} \int_V k \nabla^2 T dt dV + \int_{t-\Delta t/2}^{t+\Delta t/2} \int_V S dt dV \quad (77)$$

Resulting in:

$$(\rho c_p T_p V_p)^{t+\Delta t/2} - (\rho c_p T_p V_p)^{t-\Delta t/2} = [F_w^d - F_e^d]^t \Delta t + S_P^t \Delta t V_P \quad (78)$$

Divide by Δt , express the diffusive fluxes in terms of D_w and D_e , dropping superscripts t for current time:

$$\frac{(\rho c_p T_p V_p)^{t+\Delta t/2} - (\rho c_p T_p V_p)^{t-\Delta t/2}}{\Delta t} = -D_w (T_P - T_W) + D_e (T_E - T_P) + S_P V_P \quad (79)$$

LHS is known, for RHS \rightarrow need to specify values for times $t - \Delta t/2$ and $t + \Delta t/2$

- 1st order time integration scheme We assume a **piecewise constant** distribution over each timestep between the face values, resulting in:

$$T_P^{t-\Delta t/2} = T_P^{t-\Delta t}$$

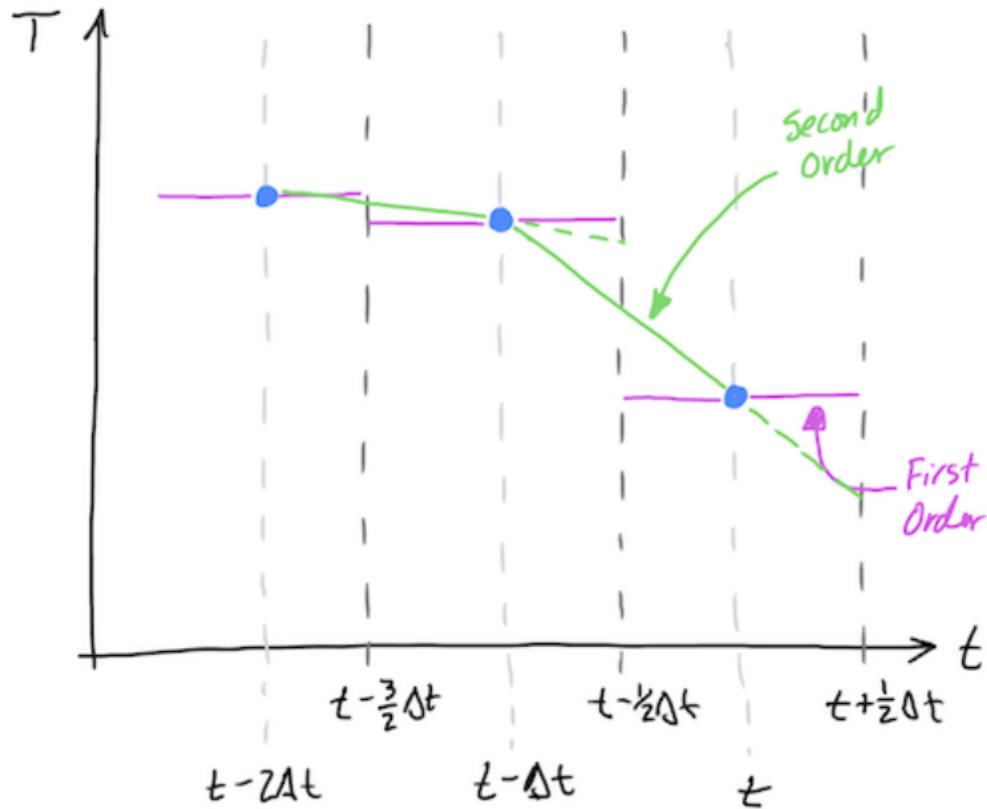
$$T_P^{t+\Delta t/2} = T_P^t$$

- 2nd order time integration scheme We assume a **piecewise linear** distribution over each timestep between the face values, resulting in:

$$T_P^{t-\Delta t/2} = T_P^{t-\Delta t} + \frac{1}{2}(T_P^{t-\Delta t} - T_P^{t-2\Delta t})$$

$$T_P^{t+\Delta t/2} = T_P^t + \frac{1}{2}(T_P^t - T_P^{t-\Delta t})$$

This is achieved by doing backward interpolation and then forward interpolation on the face values. The schematic below shows these interpolations:



Substituting these relations to the integrated governing equation's LHS:

- For 1st order scheme:

$$\frac{(\rho c_p T_P V_P)^{t+\Delta t/2} - (\rho c_p T_P V_P)^{t-\Delta t/2}}{\Delta t} = \rho c_p V_P \frac{T_P - T_P^o}{\Delta t} \quad (80)$$

Note:

- * the superscript for current time is dropped, and superscript for $t - \Delta t$ is replaced by o for "old value".
- * also that this is exactly the same as the result for the fully implicit scheme.

- For 2nd order scheme:

$$\frac{(\rho c_p T_P V_P)^{t+\Delta t/2} - (\rho c_p T_P V_P)^{t-\Delta t/2}}{\Delta t} = \rho c_p V_P \frac{T_P + \frac{1}{2}(T_P - T_P^o) - T_P^o - \frac{1}{2}(T_P^o - T_P^{oo})}{\Delta t}$$

or a more simplified version....

$$\frac{(\rho c_p T_P V_P)^{t+\Delta t/2} - (\rho c_p T_P V_P)^{t-\Delta t/2}}{\Delta t} = \rho c_p V_P \frac{\frac{3}{2}T_P - 2T_P^o + \frac{1}{2}T_P^{oo}}{\Delta t} \quad (81)$$

Note:

- * superscript oo is used for time value $t - 2\Delta t$.
- * unlike Crank-Nicholson's, flux values at previous timestep **do not need to be solved**. Instead, only temperature values at the **previous two time step** need to be retained.

3.5.3 Other Transient Discretization Schemes

Some higher order schemes are also used such as:

- Adams-Bashforth (explicit)
- Adams-Moulton (implicit)
- Runge-Kutta (implicit or explicit)

3.5.4 Linearization

Recall the cell residual for steady conduction:

$$r_P = D_w(T_P - T_W) - D_e(T_E - T_P) - S_P V_P$$

where $D_e = \frac{kA_e}{\Delta x_{PE}}$, and $D_w = \frac{kA_w}{\Delta x_{WP}}$. If we apply 1st order implicit to the transient term:

$$r_P = \rho c_p V_P \frac{T_P - T_P^o}{\Delta t} D_w(T_P - T_W) - D_e(T_E - T_P) - S_P V_P$$

This makes the linearization coefficients to be as follow:

$$\begin{aligned} a_P &= \frac{\partial r_P}{\partial T_P} = \frac{\rho c_p V_P}{\Delta t} + D_w + D_e - \frac{\partial S_P}{\partial T_P} V_P \\ a_W &= \frac{\partial r_P}{\partial T_W} = -D_w \\ a_E &= \frac{\partial r_P}{\partial T_E} = -D_e \end{aligned}$$

Similar to before, we can form an algebraic equation for each control volume like this:

$$a_P \delta T_P + a_W \delta T_W + a_E \delta T_E = -r_P \quad (82)$$

If we apply 2nd order implicit temporal scheme instead, then a_P term would look like this:

$$a_P = \frac{\partial r_P}{\partial T_P} = \frac{3}{2} \frac{\rho c_p V_P}{\Delta t} + D_w + D_e - \frac{\partial S_P}{\partial T_P} V_P$$

3.6 Implementation

```
# Solving the 1D Diffusion Equation-TRANSIENT
import numpy as np
from enum import Enum
from scipy.sparse.linalg import spsolve
from scipy.sparse import csr_matrix
from numpy.linalg import norm
import sys
import matplotlib.pyplot as plt

# class Grid: defining a 1D cartesian grid
class Grid:

    def __init__(self, lx, ly, lz, ncv):
        # # constructor
        # lx = total length of domain in x direction [m]
        # ly = total length of domain in y direction [m]
        # lz = total length of domain in z direction [m]

        # store the number of control volumes
        self._ncv = ncv

        # calculate the control volume length
        dx = lx/float(ncv)

        # calculate the face locations
```

```

# generate array of control volume element with length dx each
self._xf = np.array([i*dx for i in range(ncv+1)])

# calculate the cell centroid locations
# Note: figure out why add a xf[-1] last item
self._xP = np.array([self._xf[0]] +
                    [0.5*(self._xf[i]+self._xf[i+1]) for i in range(ncv)] +
                    [self._xf[-1]])
# calculate face areas
self._Af = ly*lz*np.ones(ncv+1)

# calculate the outer surface area of each cell
self._Ao = (2.0*dx*ly + 2.0*dx*lz)*np.ones(ncv)

# calculate cell volumes
self._vol = dx*ly*lz*np.ones(ncv)

@property
def ncv(self):
    # number of control volumes in domain
    return self._ncv

@property
def xf(self):
    # face location array
    return self._xf

@property
def xP(self):
    # cell centroid array
    return self._xP

@property
def dx_WP(self):
    return self.xP[1:-1]-self.xP[0:-2]

@property
def dx_PE(self):
    return self.xP[2:]-self.xP[1:-1]

@property
def Af(self):
    # Face area array
    return self._Af

@property
def Aw(self):
    # West face area array
    return self._Af[0:-1]

@property
def Ae(self):
    # East face area array
    return self._Af[1:]

@property
def Ao(self):
    # Outer face area array
    return self._Ao

@property
def vol(self):
    # Cell volume array
    return self._vol

# class ScalarCoeffs defining coefficients for the linear system
class ScalarCoeffs:
    def __init__(self, ncv):
        # # constructor:
        # ncv = number of control volume in domain

        self._ncv = ncv
        self._aP = np.zeros(ncv)
        self._aW = np.zeros(ncv)
        self._aE = np.zeros(ncv)
        self._rP = np.zeros(ncv)

    def zero(self):
        # zero out the coefficient arrays
        self._aP.fill(0.0)
        self._aW.fill(0.0)
        self._aE.fill(0.0)
        self._rP.fill(0.0)

    def accumulate_aP(self, aP):
        # accumulate values onto aP
        self._aP += aP

```

```

def accumulate_aW(self ,aW):
    # accumulate values onto aW
    self._aW += aW

def accumulate_aE(self ,aE):
    # accumulate values onto aE
    self._aE += aE

def accumulate_rP(self ,rP):
    # accumulate values onto rP
    self._rP += rP

@property
def ncv(self):
    # number of control volume in domain
    return self._ncv

@property
def aP(self):
    # cell coefficient
    return self._aP

@property
def aW(self):
    # West cell coefficient
    return self._aW

@property
def aE(self):
    # East cell coefficient
    return self._aE

@property
def rP(self):
    # cell coefficient
    return self._rP

# class BoundaryLocation defining boundary condition locations
class BoundaryLocation(Enum):
    WEST = 1
    EAST = 2

# Implementing Boundary Conditions
class DirichletBC:
    def __init__(self ,phi ,grid ,value ,loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # value = boundary value
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._value = value
        self._loc = loc

    def value(self):
        # return the boundary condition value
        return self._value

    def coeff(self):
        # return the linearization coefficient
        return 0

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = self._value
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = self._value
        else:
            raise ValueError("Unknown boundary location")

class NeumannBC:
    def __init__(self ,phi ,grid ,gradient ,loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # gradient = gradient at cell adjacent to boundary
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._gradient = gradient
        self._loc = loc

```

```

def value(self):
    # return the boundary condition value
    if self._loc is BoundaryLocation.WEST:
        return self._phi[1]-self._gradient*self._grid.dx_WP[0] \
            # T(0) = T(1) - gb*dx
    elif self._loc is BoundaryLocation.EAST:
        return self._phi[-2]-self._gradient*self._grid.dx_PE[-1] \
            #T(-1) = T(-2) -gb*dx
    else:
        raise ValueError("Unknown boundary location")

def coeff(self):
    # return the linearization coefficient
    return 1

def apply(self):
    # applies the boundary condition in the referenced field variable array
    if self._loc is BoundaryLocation.WEST:
        self._phi[0] = self._phi[1] - self._gradient*self._grid.dx_WP[0]
    elif self._loc is BoundaryLocation.EAST:
        self._phi[-1] = self._phi[-2] + self._gradient*self._grid.dx_PE[-1]
    else:
        raise ValueError("Unknown boundary location")

class RobinBC:
    def __init__(self ,phi ,grid ,h,k,tinfty , loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # h = convective coefficient
        # k = conductive coefficient
        # tinfty = freestream temperature
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._h = h
        self._k = k
        self._tinfty = tinfty
        self._loc = loc

    def value(self):
        # return the boundary condition value
        if self._loc is BoundaryLocation.WEST:
            return (self._phi[1] + self._grid.dx_WP[0]*(self._h/self._k)*(self._tinfty))\
                /(1 + self._grid.dx_WP[0]*(self._h/self._k))
        elif self._loc is BoundaryLocation.EAST:
            return (self._phi[-2] - self._grid.dx_PE[-1]*(self._h/self._k)*(self._tinfty))\
                /(1 - self._grid.dx_PE[-1]*(self._h/self._k))
        else:
            raise ValueError("Unknown boundary location")

    def coeff(self):
        # return the linearization coefficient
        return 1

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = (self._phi[1] + self._grid.dx_WP[0]*(self._h/self._k)*(self._tinfty))\
                /(1 + self._grid.dx_WP[0]*(self._h/self._k))
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = (self._phi[-2] - self._grid.dx_PE[-1]*(self._h/self._k)*(self._tinfty))\
                /(1 - self._grid.dx_PE[-1]*(self._h/self._k))
        else:
            raise ValueError("Unknown boundary location")

# class DiffusionModel defining a defusion model
class DiffusionModel:
    def __init__(self ,grid ,phi ,gamma,west_bc ,east_bc):
        # constructor
        self._grid = grid
        self._phi = phi
        self._gamma = gamma      # gamma here is "k" in the heat model
        self._west_bc = west_bc
        self._east_bc = east_bc

    # add diffusion terms to coefficient arrays
    def add(self ,coeffs):

        # calculate west/east diffusion flux for each face
        flux_w = -self._gamma* self._grid.Aw*(self._phi[1:-1]-self._phi[0:-2])\
            /self._grid.dx_WP # Fw = k*(Tp-Tw)*Aw/ dx_WP
        flux_e = -self._gamma* self._grid.Ae*(self._phi[2:]-self._phi[1:-1])\

```

```

    /self._grid.dx_PE # Fe = k*(Te-Tp)*Ae/dx_PE

    # calculate the linearized coefficient [aW, aP, aE]
    coeffW = -self._gamma* self._grid.Aw/ self._grid.dx_WP # Dw
    coeffE = -self._gamma* self._grid.Ae/ self._grid.dx_PE # De
    coeffP = -coeffW - coeffE # Dw + De in notes , but due to minus sign so -Dw-De in code

    # modified the linearized coefficient on the boundaries
    coeffP[0] += coeffW[0]* self._west_bc.coeff()
    coeffP[-1] += coeffE[-1]* self._east_bc.coeff()

    # zero the coefficients that are not used (on the boundary)
    coeffW[0] = 0.0
    coeffE[-1] = 0.0

    # calculate the net flux from each cell (out -in)
    flux = flux_e - flux_w

    # add to coefficient arrays
    coeffs.accumulate_aP(coeffP)
    coeffs.accumulate_aW(coeffW)
    coeffs.accumulate_aE(coeffE)
    coeffs.accumulate_rP(flux)

    # return the coefficient arrays
    return coeffs

# function to return a sparse matrix representation of a set of scalar coefficients
def get_sparse_matrix(coeffs):
    ncv = coeffs.ncv
    data = np.zeros(3*ncv-2)
    rows = np.zeros(3*ncv-2, dtype = int)
    cols = np.zeros(3*ncv-2, dtype = int)
    data[0] = coeffs.aP[0]
    rows[0] = 0
    cols[0] = 0

    if ncv > 1:
        data[1] = coeffs.aE[0]
        rows[1] = 0
        cols[1] = 1

    for i in range(ncv-2):
        data[3*i+2] = coeffs.aW[i+1]
        data[3*i+3] = coeffs.aP[i+1]
        data[3*i+4] = coeffs.aE[i+1]

        rows[3*i+2:3*i+5] = i+1

        cols[3*i+2] = i
        cols[3*i+3] = i+1
        cols[3*i+4] = i+2

    if ncv > 1:
        data[3*ncv-4] = coeffs.aW[-1]
        data[3*ncv-3] = coeffs.aP[-1]

        rows[3*ncv-4:3*ncv-2] = ncv-1

        cols[3*ncv-4] = ncv-2
        cols[3*ncv-3] = ncv-1

    return csr_matrix((data, (rows,cols)))

# solve the linear system and return the field variables
def solve(coeffs):
    # get the sparse matrix
    A = get_sparse_matrix(coeffs)
    # solve the linear system
    return spsolve(A, -coeffs.rP)

# defining a first order implicit transient model
class FirstOrderTransientModel:
    # constructor
    def __init__(self, grid, T, Told, rho, cp, dt):
        self._grid = grid
        self._T = T
        self._Told = Told
        self._rho = rho
        self._cp = cp
        self._dt = dt

    def add(self, coeffs):
        # calculate the transient term
        # rp = rho cp Vp (TP - TPold)/ dt
        transient = self._rho* self._cp* self._grid.vol*(self._T[1:-1]-self._Told[1:-1])/ self._dt

        # calculate the linearization coefficients

```

```

coeff = self._rho * self._cp * self._grid.vol / self._dt # ap = rho*cp*vp/dt

# add to coefficient arrays
coeffs.accumulate_aP(coeff)
coeffs.accumulate_rP(transient)

return coeffs

# class defining a surface convection model
class SurfaceConvectionModel:
    # constructor
    def __init__(self, grid, T, ho, To):
        self._grid = grid
        self._T = T
        self._ho = ho
        self._To = To

    # add surface convection terms to coefficient arrays
    def add(self, coeffs):
        # calculate the source term q = hA(T-Tinfty)
        source = self._ho * self._grid.Ao * (self._T[1:-1] - self._To)

        # calculate linearization coefficients
        coeffP = self._ho * self._grid.Ao

        # add to coefficient arrays
        coeffs.accumulate_aP(coeffP)
        coeffs.accumulate_rP(source)

    return coeffs

# Solving the 1D steady conduction with Dirichlet BC
# Initially, east = 300K, west = 300K
# define the grid
lx = 1.0
ly = 0.1
lz = 0.1
ncv = 10
mygrid = Grid(lx, ly, lz, ncv)

# set the timestep information
nTime = 10
dt = 1
time = 0

# set the max iterations and convergence criterion
maxIter = 100
converged=1e-6

# thermal properties
k = 0.1
rho = 1000
cp = 1000
k = 100

# convection parameters
ho = 25
To = 200

# coefficients
coeffs = ScalarCoeffs(mygrid.ncv)

# initial condition
T0 = 300

# initialize field variable arrays
T = T0*np.ones(mygrid.ncv+2)

# boundary condition
west_bc = DirichletBC(T, mygrid, 400, BoundaryLocation.WEST)
east_bc = DirichletBC(T, mygrid, 0, BoundaryLocation.EAST)

# apply boundary conditions
west_bc.apply()
east_bc.apply()

# list to store the solution at each iteration
T_solns = [np.copy(T)]

# define the transient model
Told = np.copy(T)
transient = FirstOrderTransientModel(mygrid, T, Told, rho, cp, dt)

# define the diffusion model
diffusion = DiffusionModel(mygrid, T, k, west_bc, east_bc)

```

```

# define the surface convection model
surfaceConvection = SurfaceConvectionModel(mygrid,T,ho,To)

resList = []
iterList = []

# # # iterate until the solution is converged
for tStep in range(nTime):

    # update the time information
    time += dt

    # print the timestep information
    print("Timestep = {} ; Time = {}".format(tStep,time))

    # store the old temperature field
    Told[:] = T[:]

    # iterate until converge
    for i in range(maxIter):

        # zero the coeff and add
        coeffs.zero()
        coeffs = diffusion.add(coeffs)
        coeffs = surfaceConvection.add(coeffs)
        coeffs = transient.add(coeffs)

        # compute residual and check for convergence
        maxResid = norm(coeffs.rP, np.inf)
        avgResid = np.mean(np.absolute(coeffs.rP))
        print("Iteration = {} ; Resid = {} ; Avg. Resid = {}".format(i,maxResid, avgResid) )
        if maxResid < converged:
            break

        # solve the sparse matrix system
        dT = solve(coeffs)

        # update the solution and boundary conditions
        T[1:-1] += dT
        west_bc.apply()
        east_bc.apply()

        # store the solution
        T_solns.append(np.copy(T))
        resList.append(avgResid)
        iterList.append(i)

    # plotting
    # i = 0
    # for Ti in T_solns:
    #     plt.plot(mygrid.xP, Ti, label = str(i))
    #     i += 1

    # plt.title('Implicit')
    # plt.xlabel('X')
    # plt.ylabel('T')
    # plt.savefig('pic/1d_transient_implicit.png')
    # plt.show()

plt.plot(resList, iterList)
plt.show()

```

4 ONE-DIMENSIONAL CONVECTION OF A SCALAR

4.1 Problem Definition

For thermal convection, we need the advection-diffusion equation. So far, we only dealt with diffusion, now we add the advection term which results in:

$$\frac{\partial(\rho c_p T)}{\partial t} + \nabla \cdot (\rho c_p \mathbf{u} T) = k \nabla^2 T + S \quad (83)$$

assuming constant ρ and c_p . We also assume that the flow field \mathbf{u} is known, and we only use it to advect and solve for the temperature field. To preserve continuity across cells, we also define a mass conservation equation without mass source.

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0 \quad (84)$$

4.2 Discretization

Just like we did before, we now integrate the advection-diffusion equation through space and time:

$$\int_{t_0}^{t_1} \int_V \frac{\partial(\rho c_p T)}{\partial t} dt dV + \int_{t_0}^{t_1} \int_V \nabla \cdot (\rho c_p \mathbf{u} T) dt dV = \int_{t_0}^{t_1} \int_V k \nabla^2 T dV dt + \int_{t_0}^{t_1} \int_V S dV dt \quad (85)$$

Integration of the transient diffusion equation is covered. Here, we deal with the advection term. Using the Gauss' divergence theorem to convert volume integral to surface integral:

$$\int_V \nabla \cdot (\rho c_p \mathbf{u} T) dV = \int_S (\rho c_p \mathbf{u} T) \cdot \mathbf{n} dS \quad (86)$$

The surface integral is then approximated as discrete sum over the integration points

$$\int_S (\rho c_p \mathbf{u} T) \cdot \mathbf{n} dS = \sum_{i=0}^{N_{ip}-1} (\rho c_p \mathbf{u} T) \cdot \mathbf{n}_{ip} \mathbf{A}_{ip} \quad (87)$$

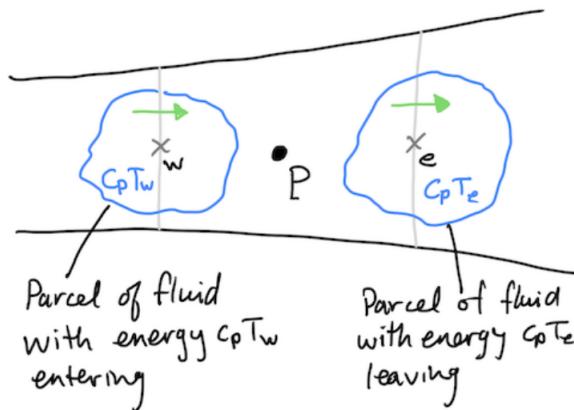
For 1D flow across control volume P , this results in:

$$\sum_{i=0}^{N_{ip}-1} (\rho c_p \mathbf{u} T) \cdot \mathbf{n}_{ip} \mathbf{A}_{ip} = \rho c_p u_e T_e A_e - \rho c_p u_w T_w A_w \quad (88)$$

Or in terms of the mass flux, $\dot{m} = \rho u A$:

$$\sum_{i=0}^{N_{ip}-1} (\rho c_p \mathbf{u} T) \cdot \mathbf{n}_{ip} \mathbf{A}_{ip} = \dot{m}_e c_p T_e - \dot{m}_w c_p T_w \quad (89)$$

Note how the $c_p T$ terms are similar to some forms of internal energy (enthalpy). Thus, we can think off the above equation as the difference in energy between 2 parcels of fluids, one with internal energy $c_p T_e$ and one with internal energy $c_p T_w$



As a result, our discretized energy equation becomes:

$$\begin{aligned} & \frac{(\rho c_p T_P V_P)^{t+\Delta t/2} - (\rho c_p T_P V_P)^{t-\Delta t/2}}{\Delta t} + \dot{m}_e c_p T_e - \dot{m}_w c_p T_w \\ &= -D_w(T_P - T_W) + D_e(T_E - T_P) + S_P V_P \end{aligned} \quad (90)$$

Note:

- transient term is like before, evaluated at time $t + \Delta t/2$ and $t - \Delta t/2$.
- our discretization scheme is **NOT** completed, because we still do not know how to calculate mass flux and temperature at integration points, namely \dot{m}_e, \dot{m}_w and T_e, T_w .
- we must consider whether the given equation is independent of temperature level according to Rule 4. One can say that it has to be independent because transient, diffusion, advection terms involve only derivative of temperature. This is true, if mass is conserved ($\dot{m}_e = \dot{m}_w$). For 1D, this is easy to ensure. For multidimensional problems, this is difficult. In other words, we cannot assure that the numerical mass fluxes will always be conserved. This can lead to major problem, because if mass is not conserved, one may think that there is an energy source (or sink) within the domain.
- to get around the mass conservation problem, we subtract the discretized mass equation from the energy equation. Assuming constant density:

$$\dot{m}_e - \dot{m}_w = 0$$

multiply this by T_P and C_P and subtracting from the discretized equation:

$$\begin{aligned} & \frac{(\rho c_p T_P V_P)^{t+\Delta t/2} - (\rho c_p T_P V_P)^{t-\Delta t/2}}{\Delta t} + \dot{m}_e c_p (T_e - T_P) - \dot{m}_w c_p (T_w - T_P) \\ &= -D_w(T_P - T_W) + D_e(T_E - T_P) + S_P V_P \end{aligned} \quad (91)$$

This means that if there is a positive imbalance of mass ($\dot{m}_e > \dot{m}_w$), there will be a negative source in the energy equation to counter balance. If there is a negative imbalance, the opposite is true. This step helps with the stability of the numerical method such that the equations are again independent of the temperature level.

4.3 Advection term with Explicit Time Integration

Assume we can interpolate the integration point values in the advection term using a piecewise linear approximation:

$$\begin{aligned} T_e &= \frac{1}{2}(T_P + T_E) \\ T_w &= \frac{1}{2}(T_W + T_P) \end{aligned}$$

Assuming no source term and use an explicit time integration scheme, and keeping the T_P term arising from subtracting the mass conservation from the energy equation as implicit (i.e. at current timestep). We get the following discretized equation:

$$\begin{aligned} & \frac{\rho c_p V_P (T_P - T_P^o)}{\Delta t} + \dot{m}_e c_p \left[\frac{1}{2}(T_P^o + T_E^o) - T_P \right] - \dot{m}_w c_p \left[\frac{1}{2}(T_W^o + T_P^o) - T_P \right] \\ &= -D_w(T_P^o - T_W^o) + D_e(T_E^o - T_P^o) \end{aligned}$$

where ' o ' denotes values at previous timestep, and those without superscripts are for current timestep (i.e. those being solved). We can then group the terms according to their temperature:

$$\begin{aligned} \left(\frac{\rho c_p V_P}{\Delta t} + c_p \dot{m}_w - c_p \dot{m}_e \right) T_P &= \left(\frac{\rho c_p V_P}{\Delta t} + \frac{c_p \dot{m}_w}{2} - \frac{c_p \dot{m}_e}{2} - D_e - D_w \right) T_P^o \\ &+ \left(D_e - \frac{c_p \dot{m}_e}{2} \right) T_E^o + \left(D_w - \frac{c_p \dot{m}_w}{2} \right) T_W^o \end{aligned}$$

If mass is conserved, i.e. $\dot{m}_e = \dot{m}_w$, then:

$$\begin{aligned} \frac{\rho c_p V_P}{\Delta t} T_P - \left(\frac{\rho c_p V_P}{\Delta t} - D_e - D_w \right) T_P^o - \left(D_e - \frac{c_p \dot{m}_e}{2} \right) T_E^o \\ + \left(D_w - \frac{c_p \dot{m}_w}{2} \right) T_W^o = 0 \end{aligned}$$

From Rule 2, we need :

- the coefficient on T_P to be positive
- the coefficients on remaining terms, T_P^o , T_E^o , and T_W^o to be negative.

For T_P^o , this requires:

$$D_e + D_w \leq \frac{\rho c_p V_P}{\Delta t}$$

or:

$$\Delta t \leq \frac{\rho c_p V_P}{D_e + D_w}$$

Refer to chapter 3, we see that this is the same timestep restriction in the form:

$$\boxed{\frac{\alpha \Delta t}{\Delta x^2} \leq \frac{1}{2}}$$

where $\alpha = \frac{k}{\rho c_p}$. We can say that the addition of the advection term does not change the timestep restriction. Note how the coefficients for T_P^o and T_E^o **can be positive** for certain mass flow rates. Thus, for T_E^o , we need the following condition:

$$D_e \geq \frac{c_p \dot{m}_e}{2}$$

simplifying...

$$\begin{aligned} D_e &\geq \frac{c_p \dot{m}_e}{2} \\ \frac{kA}{\Delta x} &\geq \frac{c_p \rho u A}{2} \\ \Delta x &\leq \frac{2k}{\rho c_p u} \end{aligned}$$

or in terms of the thermal diffusivity, α :

$$\boxed{\frac{u \Delta x}{\alpha} < 2}$$

To sum up, we have both the spatial and temporal conditions on Δx and Δt . Multiplying them together:

$$\begin{aligned} \frac{\alpha \Delta t}{\Delta x^2} \cdot \frac{u \Delta x}{\alpha} &< \frac{1}{2} \cdot 2 \\ \frac{u \Delta t}{\Delta x} &< 1 \end{aligned}$$

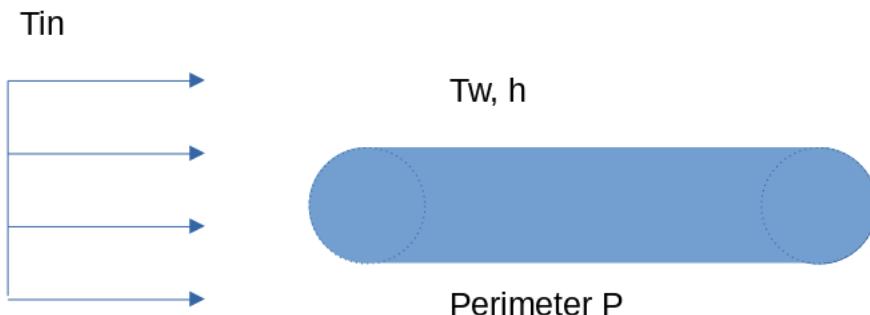
The LHS is known as the Courant number, thus the space-time restriction can be written as:

$$\boxed{Co < 1} \quad (92)$$

To see whether such restriction is serious.

Consider flow in a tube with constant wall temperature, T_w , this problem has the exact solution as:

$$\frac{T_w - T(x)}{T_w - T_{in}} = \exp \left(-\frac{hP}{\dot{m}c_p} x \right)$$



Let us consider the solution for this problem until some point x_L where the bulk temperature difference, $(T_w - T(x))$ reached 5% the difference in inlet and wall, $(T_w - T_{in})$, i.e.

$$\frac{T_w - T(x_L)}{T_w - T_{in}} = 0.05$$

We further assume the Nusselt number is defined as:

$$Nu = \frac{hD}{k}$$

Then the general solution for this particular geometry is as follow:

$$\frac{T_w - T(x)}{T_w - T_{in}} = \exp\left(-\frac{\frac{Nu k \pi D}{D} x_L}{\rho u \pi \frac{D^2}{4} c_p}\right) = \exp\left(-\frac{4Nu \alpha}{uD^2} x_L\right)$$

By definition, $uD/\alpha = RePr$, $R_e = uD/\nu$ and $Pr = \nu/\alpha$:

$$\frac{T_w - T(x)}{T_w - T_{in}} = \exp\left(-\frac{4Nu}{RePr} \frac{x_L}{D}\right)$$

Recall we want 5% between the temperature difference

$$\begin{aligned} \frac{4Nu}{RePr} \frac{x_L}{D} &= 3 \\ \frac{x_L}{D} &= \frac{3}{4} \frac{RePr}{Nu} \end{aligned}$$

Now, we use the restriction from above, $\Delta x \leq 2\alpha/u$ and the definition of the number of control volume, $N_{cv} = x_L/\Delta x$:

$$N_{cv} \leq \frac{x_L}{\Delta x} = \frac{3}{8} \frac{Re^2 Pr^2}{Nu}$$

Using some numbers to test:

PARAMETERS	NO. OF CONTROL VOLUME
------------	-----------------------

$$Nu = 5, Re = 1000, Pr = 1 \quad 10^5 = 100,000$$

$$Nu = 5, Re = 1000, Pr = 10 \quad 10^7 = 10,000,000$$

Note: a 10 times increase in Pr results in 10^7 number of control volume. This is impractical because it means that we need to solve 10^7 equations. So what is the required minimum number of timestep? We can calculate this by taking the ratio between the time it take for a fluid to exit the pipe to the time step restriction we derived above:

$$\begin{aligned} N_t &= \frac{x_L/u}{\Delta t} \\ &= \frac{\frac{3}{4} \frac{RePr}{Nu} \frac{D}{u}}{\frac{1}{2} \frac{\Delta x^2}{\alpha}} \\ &= \frac{3}{8} \frac{Re^2 Pr^2}{Nu} \end{aligned}$$

Again, using some numbers, we face similar problem: too impractical.

PARAMETERS	NO. OF CONTROL VOLUME
------------	-----------------------

$$Nu = 5, Re = 1000, Pr = 1 \quad 10^5 = 100,000$$

$$Nu = 5, Re = 1000, Pr = 10 \quad 10^7 = 10,000,000$$

Next, we will discuss in more details about these restrictions.

4.4 Discussion of the Restrictions on Timestep

We see how explicit scheme results in timestep restriction. Next, let's consider a 1st order implicit scheme:

$$\begin{aligned} & \left(\frac{\rho c_p V_p}{\Delta t} + D_e + D_w + c_p \dot{m}_w - c_p \dot{m}_e \right) T_P - \left(D_e - \frac{c_p \dot{m}_e}{2} \right) T_E \\ & - \left(D_w + \frac{c_p \dot{m}_e}{2} \right) T_W - \frac{\rho c_p V_p}{\Delta t} T_P^o = 0 \end{aligned}$$

Rule 2 requires coefficient on T_p to be positive. From above equation, there is no time restriction because coefficient on T_p is always positive, assuming mass is conserved.

On the other hand, T_W and T_E ' coefficients need to be negative:

$$\begin{aligned} D_e - \frac{c_p \dot{m}_e}{2} &\leq 0 \\ \frac{u \Delta x}{\alpha} &\leq 2 \end{aligned}$$

Previous analysis still applies so the number of control volume is still large (even if there is no restriction on the timestep size)

4.5 Discussion of the Restriction on Spatial Resolution

By moving to an implicit time integration scheme, we encountered restriction on time step.

Follow the same logic, we can also say that the restriction on grid size must come from the interpolation method chosen for the integration point in the advection term.

Again, we consider flow in a duct where analytical solution is known and can be applied between P and E locations. Our temperature profile between cell centers is:

$$\frac{T - T_P}{T_E - T_P} = \frac{\exp \left[Pe_\Delta \left(\frac{x - x_P}{x - x_E} \right) \right] - 1}{\exp(Pe_\Delta) - 1}$$

where Pe_Δ is the Pectet number representing ratio of convection to diffusion

$$\begin{aligned} Pe_\Delta &= \frac{u \Delta x}{\alpha} \\ &= \frac{u \Delta x}{\alpha} \cdot \frac{\mu/\rho}{\mu/\rho} \\ &= \frac{(\mu/\rho) u \Delta x}{(\mu/\rho) \alpha} \\ &= \frac{\nu}{\alpha} \left(\frac{u \Delta x}{\mu/\rho} \right) \\ &= \frac{\nu}{\alpha} \left(\frac{\rho u \Delta x}{\mu} \right) \\ &= Re_\Delta Pr \end{aligned}$$

with the following flow regimes:

- $Pe_\Delta \approx 0$: diffusion dominates
- $|Pe_\Delta| \approx 1$: convection and diffusion
- $|Pe_\Delta| \gg 1$: convection dominates

Visualization for different values of Pe_Δ : Python code

```
import matplotlib.pyplot as plt
import numpy as np

# set range for Pe
Pe_vals = [-50, -5, 1e-6, 5, 50]

# assign x values
x = np.linspace(0, 1)
```

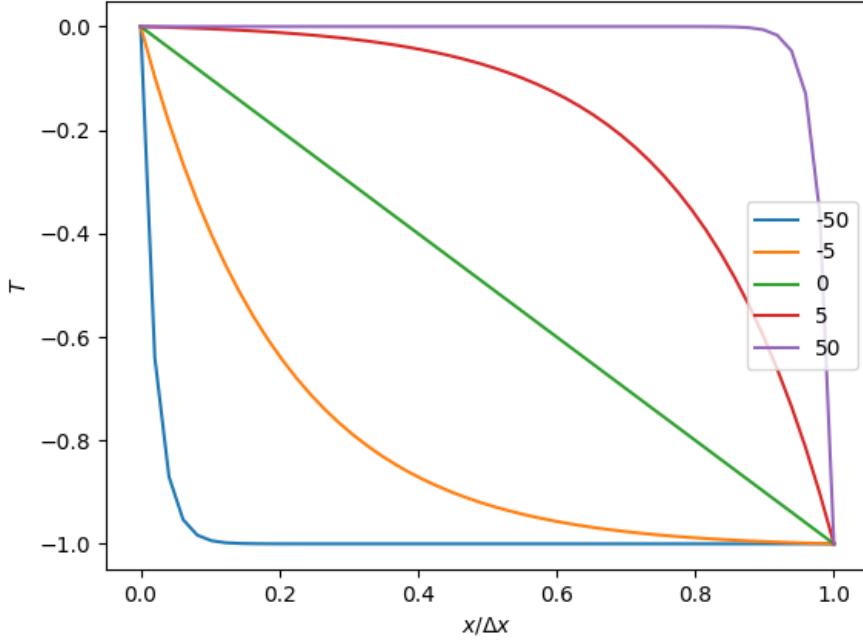
```

# set TP and TE
TP = 1
TE = 0

for Pe in Pe_vals:
    T = (TE-TP)*(np.exp(Pe*(x-x[0])/(x[-1]-x[0]))-1) \
        /(np.exp(Pe)-1)
    plt.plot(x, T, label=str(int(Pe)))

plt.xlabel(r"$x/\Delta x$")
plt.ylabel("$T$")
plt.legend()
plt.savefig("pic/pectet_visual_pipe.png")

```



We note from the figure above that our assumption piecewise linear temperature profile, using the central difference scheme (CDS), is only valid for $Pe_\Delta \approx 0$.

In practice, Pe_Δ will be large and thus we need different interpolation scheme.

4.6 The Upwind Difference Scheme (UDS)

We attempt a new interpolation scheme, upwind difference scheme. For the east integration point, we have:

$$T_e = \frac{1 + \alpha_e}{2} T_P + \frac{1 - \alpha_e}{2} T_E \quad (93)$$

with α being the weighting factor, i.e.

- $Pe_\Delta \approx 0, \alpha_e = 0$: diffusion dominates, CDS recovered
- $|Pe_\Delta| \approx 1, \alpha_e = 1$: convection and diffusion, $T_e = T_P$
- $|Pe_\Delta| \gg 1, \alpha_e = -1$: convection dominates, $T_e = T_E$

Using this new interpolation scheme, the discrete equation in terms of the cell residual becomes:

$$\begin{aligned} r_P = & \left(\frac{\rho c_p v_P}{\Delta t} + D_e + D_w + \frac{1}{2} c_p \dot{m}_w (1 + \alpha_w) - \frac{1}{2} c_p \dot{m}_e (1 - \alpha_e) \right) T_P \\ & - \left[D_e - \frac{1}{2} c_p \dot{m}_e (1 - \alpha_e) \right] T_E - \left[D_w + \frac{1}{2} c_p \dot{m}_w (1 + \alpha_w) \right] T_W - \frac{\rho c_p V_P}{\Delta t} T_P^o \end{aligned}$$

with the following linearization coefficients:

$$\begin{aligned} a_W &= -D_w - \frac{1}{2} c_p \dot{m}_w (1 + \alpha_w) \\ a_E &= -D_e + \frac{1}{2} c_p \dot{m}_e (1 - \alpha_e) \\ a_P &= \frac{\rho c_p v_P}{\Delta t} - a_W - a_E \end{aligned} \quad (94)$$

For fast flowing fluid in the positive direction, $\alpha_w = \alpha_e = 1$. Our east/west coefficients become:

$$\begin{aligned} a_W &= -D_w - c_p \dot{m}_w \\ a_E &= D_e \end{aligned}$$

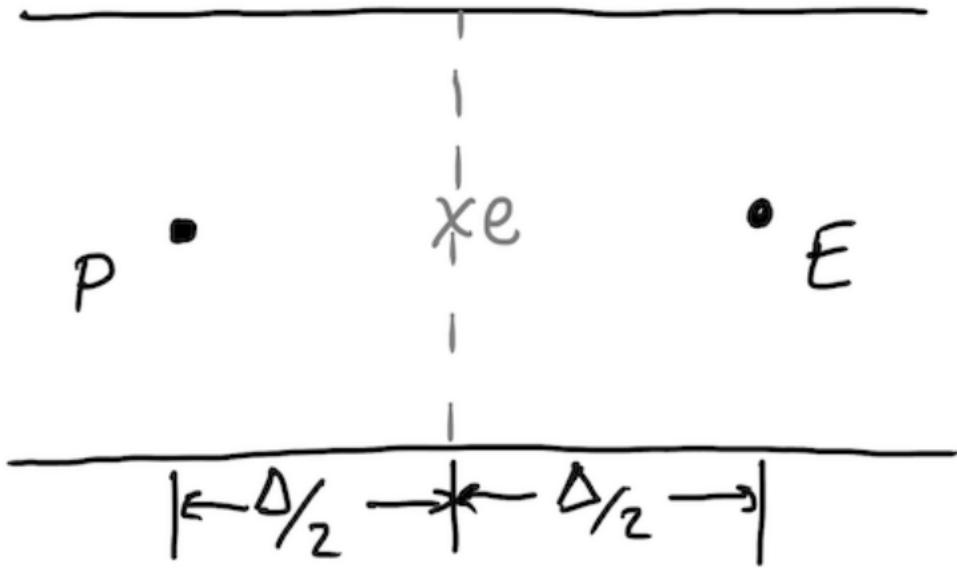
To satisfy Rule 2, here the coefficients cannot become positive, because the fluid is flowing in the positive direction and \dot{m}_w is positive. In contrast, for a fluid flowing in the negative direction, $\alpha_w = \alpha_e = -1$ and the coefficients become:

$$\begin{aligned} a_W &= -D_w \\ a_E &= -D_e + c_p \dot{m}_e \end{aligned}$$

Again, these cannot be positive because \dot{m}_e is negative in this case. As a result, we can see that using Upwind Difference Scheme ensures the solution is stable for any Δx . Combining this with an implicit time integration scheme means that there are no formal restrictions on timestep or grid size.

4.7 False Diffusion

UDS is only 1st order and only when $\alpha_e = \pm 1$. Here, we try to estimate the accuracy of UDS against CDS using Taylor series about east face integration point.



Expanding about this point gives the following cell values:

$$\begin{aligned} T_E &= T_e + \frac{\Delta}{2} dT dx \Big|_e + \frac{(\Delta/2)^2}{2} \frac{d^2 T}{dx^2} \Big|_e + \dots \\ T_P &= T_e - \frac{\Delta}{2} dT dx \Big|_e + \frac{(\Delta/2)^2}{2} \frac{d^2 T}{dx^2} \Big|_e - \dots \end{aligned}$$

Recall the CDS interpolation, $T_e = 1/2(T_P + T_E)$. We sub this into the above estimates:

$$T_e^{CDS} = T_e + \frac{(\Delta/2)^2}{2} \frac{d^2 T}{dx^2} \Big|_e + O(\Delta^4)$$

Likewise, for UDS, we assume flow in the positive direction:

$$T_e^{UDS} = T_P = T_e - \frac{\Delta}{2} \frac{dT}{dx} \Big|_e + O(\Delta^2) \quad (95)$$

The error is based on the first truncated term. For both cases (UDS and CDS), the leading term is T_e and the next term is the truncated term.

$$e^{CDS} \sim \dot{m}c_p \frac{(\Delta/2)^2}{2} \frac{d^2T}{dx^2} \Big|_e \sim O(\Delta^2)$$

$$e^{UDS} \sim \dot{m}c_p \frac{\Delta}{2} \frac{dT}{dx} \Big|_e \sim O(\Delta)$$

Note how we have the $\dot{m}c_p$ term, this is because the integration point temperatures are multiplied by this value in the energy equation. Thus, this is the full error for that term, not just the error for the interpolated values. We can conclude that CDS is 2nd order accurate in space, while UDS is only 1st order in space. Another way to think is: if we half the grid size, the UDS error will reduce by factor of 2, while the CDS will reduce by factor of 4.

Note also that the error for UDS is proportional to the temperature gradient in the energy equation, this behaves very much like a diffusion term: $k \frac{dT}{dx}$. We call this ‘false diffusion’.

$$e^{UDS} = -\dot{m}c_p \frac{\Delta}{2} \frac{dT}{dx} \Big|_e = -\frac{\rho c_p u_e A_e \Delta}{2} \frac{dT}{dx} \Big|_e = -\Gamma^{false} \frac{dT}{dx} \Big|_e A_e$$

with $\Gamma^{false} = \frac{\rho c_p u_e \Delta}{2}$. Obviously, our real diffusion involves k and ∇T . Taking the ratio between these:

$$\frac{\Gamma^{false}}{\Gamma^{real}} = \frac{\rho c_p u_e \Delta}{2k} = \frac{1}{2} \frac{u \Delta}{\nu} \frac{\rho c_p}{k} = \frac{1}{2} \frac{u \Delta}{\nu} \frac{\nu}{\alpha} = \frac{1}{2} Re_\Delta Pr = \frac{1}{2} Pe_\Delta$$

Note: for large Pe , false diffusion dominates real diffusion. This is bad because we can't model real diffusion. However, note that in our analysis, we assume that the leading term is a good estimation of the error. For convection problem, this may not be the case. To test this, let us consider the exact solution between points P and E :

$$\frac{T - T_p}{T_E - T_p} = \frac{\exp(Pe(x^*)) - 1}{\exp(Pe) - 1}$$

where:

$$Pe = Pe_\Delta$$

$$x^* = \left(\frac{x - x_P}{x_E - x_P} \right)$$

or we can think of it like this:

$$T - T_p = (T_E - T_p) \frac{\exp(Pe(x^*)) - 1}{\exp(Pe) - 1} = A[\exp(Pe(x^*)) - 1]$$

Recall the Taylor series expansion for T_P up to 4 terms is:

$$T_P = T_e - \frac{\Delta}{2} \frac{dT}{dx} \Big|_e + \frac{(\Delta/2)^2}{2} \frac{d^2T}{dx^2} \Big|_e - \frac{(\Delta/2)^3}{6} \frac{d^3T}{dx^3} \Big|_e$$

The first derivative term in the Taylor series, for this particular solution is:

$$\frac{dT}{dx} \Big|_e = \frac{dT}{dx^*} \Big|_e \frac{dx^*}{dx}$$

$$\frac{dx^*}{dx} = \frac{1}{x_E - x_P} = \frac{1}{\Delta}$$

$$\frac{dT}{dx^*} \Big|_e = APe \exp(Pe(x^*)) = APe \exp\left(\frac{Pe}{2}\right)$$

Following the same procedure, we can find out expressions for the 2nd and 3rd derivatives. Plugging into the Taylor series for T_P :

$$T_P = \dots = T_e - \frac{APe \exp\left(\frac{Pe}{2}\right)}{2} \left[1 - \frac{Pe}{4} + \frac{Pe}{24} \right]$$

Let:

$$S = \left[1 - \frac{Pe}{4} + \frac{Pe}{24} \right]$$

Then:

Pe	S
0.01	1-0.0025+0.00004-
1	1-0.025+0.0416-
100	1-25+416.6
1000	1-250+4166.6-

We can see that the series only converge for $Pe \leq 1$. Note the first term is only representative of error. If profile is linear, then series converge and we have a good estimate.

Else if the profile is non linear, the Taylor series does not give any useful information. So, at high Pe , the false diffusion by UDS is not as bad. But UDS is a first order scheme and therefore, its accuracy is limited. So, we need to look for ways to improve accuracy of UDS while preserving stability.

4.8 Improvements to Advection Scheme

4.8.1 Power Law Scheme

By choosing the appropriate weighting coefficient, α_e , we can prevent the linearization coefficients from taking the incorrect signs. From the generalized UDS

$$T_e = \frac{1 + \alpha_e}{2} T_P + \frac{1 - \alpha_e}{2} T_E$$

with:

$$\alpha_e = \frac{Pe^2}{5 + Pe^2}$$

At $Pe \approx 1$, $\alpha_e \approx 1/2$, so the scheme is 2nd order accurate.

At large Pe , the scheme approaches UDS and can only be 1st order accurate. This is only a partial solution.

4.8.2 Deferred Correction Approach

The idea is use UDS as main advection scheme, then linearized accordingly. Note here that the UDS term are subtracted from the discretized equation.

Also, the higher order terms for higher order schemes are added explicitly.

These two are not linearized. As a result, the linearization process only maintains the stability of the UDS.

The advective flux through east face of a control volume can be written as:

$$F_e = F_e^{UDS} + (F_e^{HOS} - F_e^{UDS})$$

with F_e^{UDS} and F_e^{HOS} are the flux from UDS and flux from higher order scheme respectively.

Linearization is only carried out in the 1st term, because this will guarantee stability. This can be not exact because it is UDS-based, not higher order schemes. Iteration is needed to arrive at the solution.

4.8.3 Central Difference Scheme (CDS)

We can also use the CDS scheme and implement it with the deferred correction approach. Recall the CDS scheme:

$$T_e = \frac{1}{2}(T_P + T_E)$$

4.8.4 Quadratic Upwind Interpolation for Convective Kinematics (QUICK)

This scheme is derived by passing a parabola through cell values in the upwind direction.

Flow in positive direction, interpolation for east integration point involves cells W , P , and E .

Flow in negative direction, interpolation for east integration point involves cells P , E , and $EE\$$

This results in in the following expression for $T(x)$:

$$T(x) = \frac{(x - x_P)(x - x_E)}{(x_W - x_P)(x_W - x_E)} T_W + \frac{(x - x_W)(x - x_E)}{(x_P - x_W)(x_P - x_E)} T_P \\ + \frac{(x - x_W)(x - x_P)}{(x_E - x_W)(x_E - x_P)} T_E$$

for uniform grid spacing Δ :

$$T_e = -\frac{1}{8}T_W + \frac{3}{4}T_P + \frac{3}{8}T_E$$

$$T_w = -\frac{1}{8}T_W W + \frac{3}{4}T_W + \frac{3}{8}T_P$$

Note how the negative sign could cause the coefficients having the wrong signs if implement directly.

Nevertheless, if implement using deferred correction approach, the QUICK scheme is an effective higher order scheme.

4.9 Implementation

```
import numpy as np
from enum import Enum
from scipy.sparse.linalg import spsolve
from scipy.sparse import csr_matrix
from numpy.linalg import norm
import sys
import matplotlib.pyplot as plt

# class Grid: defining a 1D cartesian grid
class Grid:

    def __init__(self, lx, ly, lz, ncv):
        # # constructor
        # lx = total length of domain in x direction [m]
        # ly = total length of domain in y direction [m]
        # lz = total length of domain in z direction [m]

        # store the number of control volumes
        self._ncv = ncv

        # calculate the control volume length
        dx = lx/float(ncv)

        # calculate the face locations
        # generate array of control volume element with length dx each
        self._xf = np.array([i*dx for i in range(ncv+1)])

        # calculate the cell centroid locations
        # Note: figure out why add a xf[-1] last item
        self._xP = np.array([self._xf[0]] +
                            [0.5*(self._xf[i]+self._xf[i+1]) for i in range(ncv)] +
                            [self._xf[-1]])

        # calculate face areas
        self._Af = ly*lz*np.ones(ncv+1)

        # calculate the outer surface area of each cell
        self._Ao = (2.0*dx*ly + 2.0*dx*lz)*np.ones(ncv)

        # calculate cell volumes
        self._vol = dx*ly*lz*np.ones(ncv)

    @property
    def ncv(self):
        # number of control volumes in domain
        return self._ncv

    @property
    def xf(self):
        # face location array
        return self._xf

    @property
    def xP(self):
        # cell centroid array
        return self._xP

    @property
    def dx_WP(self):
        return self.xP[1:-1]-self.xP[0:-2]

    @property
    def dx_PE(self):
        return self.xP[2:]-self.xP[1:-1]

    @property
    def Af(self):
        # Face area array
        return self._Af

    @property
    def Aw(self):
        # West face area array
        return self._Af[0:-1]
```

```

@property
def Ae(self):
    # East face area array
    return self._Af[1:]

@property
def Ao(self):
    # Outer face area array
    return self._Ao

@property
def vol(self):
    # Cell volume array
    return self._vol

# class ScalarCoeffs defining coefficients for the linear system
class ScalarCoeffs:
    def __init__(self ,ncv):
        # # constructor:
        # ncv = number of control volume in domain

        self._ncv = ncv
        self._aP = np.zeros(ncv)
        self._aW = np.zeros(ncv)
        self._aE = np.zeros(ncv)
        self._rP = np.zeros(ncv)

    def zero(self):
        # zero out the coefficient arrays
        self._aP.fill(0.0)
        self._aW.fill(0.0)
        self._aE.fill(0.0)
        self._rP.fill(0.0)

    def accumulate_aP(self ,aP):
        # accumulate values onto aP
        self._aP += aP

    def accumulate_aW(self ,aW):
        # accumulate values onto aW
        self._aW += aW

    def accumulate_aE(self ,aE):
        # accumulate values onto aE
        self._aE += aE

    def accumulate_rP(self ,rP):
        # accumulate values onto rP
        self._rP += rP

    @property
    def ncv(self):
        # number of control volume in domain
        return self._ncv

    @property
    def aP(self):
        # cell coefficient
        return self._aP

    @property
    def aW(self):
        # West cell coefficient
        return self._aW

    @property
    def aE(self):
        # East cell coefficient
        return self._aE

    @property
    def rP(self):
        # cell coefficient
        return self._rP

# class BoundaryLocation defining boundary condition locations
class BoundaryLocation(Enum):
    WEST = 1
    EAST = 2

# Implementing Boundary Conditions
class DirichletBC:
    def __init__(self ,phi ,grid ,value ,loc):
        # # constructor
        # phi = field variable array

```

```

# grid = grid
# value = boundary value
# loc = boundary location

    self._phi = phi
    self._grid = grid
    self._value = value
    self._loc = loc

def value(self):
    # return the boundary condition value
    return self._value

def coeff(self):
    # return the linearization coefficient
    return 0

def apply(self):
    # applies the boundary condition in the referenced field variable array
    if self._loc is BoundaryLocation.WEST:
        self._phi[0] = self._value
    elif self._loc is BoundaryLocation.EAST:
        self._phi[-1] = self._value
    else:
        raise ValueError("Unknown boundary location")

class NeumannBC:
    def __init__(self, phi, grid, gradient, loc):
        # # constructor
        # phi = field variable array
        # grid = grid
        # gradient = gradient at cell adjacent to boundary
        # loc = boundary location

        self._phi = phi
        self._grid = grid
        self._gradient = gradient
        self._loc = loc

    def value(self):
        # return the boundary condition value
        if self._loc is BoundaryLocation.WEST:
            return self._phi[1]-self._gradient*self._grid.dx_WP[0] # T(0) = T(1) - gb*dx
        elif self._loc is BoundaryLocation.EAST:
            return self._phi[-2]-self._gradient*self._grid.dx_PE[-1] #T(-1) = T(-2) -gb*dx
        else:
            raise ValueError("Unknown boundary location")

    def coeff(self):
        # return the linearization coefficient
        return 1

    def apply(self):
        # applies the boundary condition in the referenced field variable array
        if self._loc is BoundaryLocation.WEST:
            self._phi[0] = self._phi[1] - self._gradient*self._grid.dx_WP[0]
        elif self._loc is BoundaryLocation.EAST:
            self._phi[-1] = self._phi[-2] + self._gradient*self._grid.dx_PE[-1]
        else:
            raise ValueError("Unknown boundary location")

# class DiffusionModel defining a defusion model
class DiffusionModel:
    def __init__(self, grid, phi, gamma, west_bc, east_bc):
        # constructor
        self._grid = grid
        self._phi = phi
        self._gamma = gamma      # gamma here is "k" in the heat model
        self._west_bc = west_bc
        self._east_bc = east_bc

    # add diffusion terms to coefficient arrays
    def add(self, coeffs):

        # calculate west/east diffusion flux for each face
        flux_w = -self._gamma* self._grid.Aw*(self._phi[1:-1]-self._phi[0:-2]) \
            /self._grid.dx_WP # Fw = k*(Tp-Tw)*Aw/ dx_WP
        flux_e = -self._gamma* self._grid.Ae*(self._phi[2:]-self._phi[1:-1]) \
            /self._grid.dx_PE # Fe = k*(Te-Tp)*Ae/dx_PE

        # calculate the linearized coefficient [aW, aP, aE]
        coeffW = -self._gamma* self._grid.Aw/ self._grid.dx_WP
        coeffE = -self._gamma* self._grid.Ae/ self._grid.dx_PE
        coeffP = -coeffW - coeffE # should be +?

```

```

# modified the linearized coefficient on the boundaries
coeffP[0] += coeffW[0]*self._west_bc.coeff()
coeffP[-1] += coeffE[-1]*self._east_bc.coeff()

# zero the coefficients that are not used (on the boundary)
coeffW[0] = 0.0
coeffE[-1] = 0.0

# calculate the net flux from each cell (out -in)
flux = flux_e - flux_w

# add to coefficient arrays
coeffs.accumulate_aP(coeffP)
coeffs.accumulate_aW(coeffW)
coeffs.accumulate_aE(coeffE)
coeffs.accumulate_rP(flux)

# return the coefficient arrays
return coeffs

# class defining a surface convection model
class SurfaceConvectionModel:
    # constructor
    def __init__(self, grid, T, ho, To):
        self._grid = grid
        self._T = T
        self._ho = ho
        self._To = To

    # add surface convection terms to coefficient arrays
    def add(self, coeffs):
        # calculate the source term
        source = self._ho * self._grid.Ao*(self._T[1:-1]-self._To)

        # calculate linearization coefficients
        coeffP = self._ho * self._grid.Ao

        # add to coefficient arrays
        coeffs.accumulate_aP(coeffP)
        coeffs.accumulate_rP(source)

        return coeffs

# defining a first order implicit transient model
class FirstOrderTransientModel:
    # constructor
    def __init__(self, grid, T, Told, rho, cp, dt):
        self._grid = grid
        self._T = T
        self._Told = Told
        self._rho = rho
        self._cp = cp
        self._dt = dt

    def add(self, coeffs):
        # calculate the transient term
        # rp = rho cp Vp (TP - TPold) / dt
        transient = self._rho * self._cp * self._grid.vol * (self._T[1:-1]-self._Told[1:-1]) / self._dt

        # calculate the linearization coefficients
        coeff = self._rho * self._cp * self._grid.vol / self._dt # ap = rho*cp*vp/dt

        # add to coefficient arrays
        coeffs.accumulate_aP(coeff)
        coeffs.accumulate_rP(transient)

        return coeffs

# function to return a sparse matrix representation of a set of scalar coefficients
def get_sparse_matrix(coeffs):
    ncv = coeffs.ncv
    data = np.zeros(3*ncv-2)
    rows = np.zeros(3*ncv-2, dtype = int)
    cols = np.zeros(3*ncv-2, dtype = int)
    data[0] = coeffs.aP[0]
    rows[0] = 0
    cols[0] = 0

    if ncv > 1:
        data[1] = coeffs.aE[0]
        rows[1] = 0
        cols[1] = 1

    for i in range(ncv-2):
        data[3*i+2] = coeffs.aW[i+1]
        data[3*i+3] = coeffs.aP[i+1]
        data[3*i+4] = coeffs.aE[i+1]

```

```

rows[3*i+2:3*i+5] = i+1
cols[3*i+2] = i
cols[3*i+3] = i+1
cols[3*i+4] = i+2

if ncv > 1:
    data[3*ncv-4] = coeffs.aW[-1]
    data[3*ncv-3] = coeffs.aP[-1]

    rows[3*ncv-4:3*ncv-2] = ncv-1
    cols[3*ncv-4] = ncv-2
    cols[3*ncv-3] = ncv-1

return csr_matrix((data, (rows,cols)))

# solve the linear system and return the field variables
def solve(coeffs):
    # get the sparse matrix
    A = get_sparse_matrix(coeffs)
    # solve the linear system
    return spsolve(A, -coeffs.rP)

# class defining an upwind advection model
class UpwindAdvectionModel:

    # constructor
    def __init__(self,grid,phi,Uhe, rho, cp, west_bc, east_bc):
        self._grid = grid
        self._phi = phi
        self._Uhe = Uhe
        self._rho = rho
        self._cp = cp
        self._west_bc = west_bc
        self._east_bc = east_bc
        self._alphae = np.zeros(self._grid.ncv+1)
        self._phie = np.zeros(self._grid.ncv+1)

    # add diffusion terms to coefficient arrays
    def add(self,coeffs):

        # calculate the weighting factor
        for i in range(self._grid.ncv+1):
            if self._Uhe[i] >= 0:
                self._alphae[i] = 1
            else:
                self._alphae[i] = -1

        # calculate the east integration point (including both boundaries)
        self._phie= (1 + self._alphae)/2*self._phi[0:-1] + (1 - self._alphae)/2*self._phi[1:]

        # calculate the face mass fluxes
        mdote = self._rho * self._Uhe * self._grid.Af

        # calculate west/east advection flux
        flux_w = self._cp*mdote[:-1]*self._phie[:-1]
        flux_e = self._cp*mdote[1:]*self._phie[:-1]

        # calculate mass imbalance
        imbalance = -self._cp*mdote[1:]*self._phi[1:-1] + self._cp*mdote[:-1]*self._phi[1:-1]

        # calculate linearization coefficients
        coeffW = -self._cp*mdote[:-1]*(1 + self._alphae[:-1])/2
        coeffE = self._cp*mdote[1:]*(1 + self._alphae[1:])/2
        coeffP = -coeffW - coeffE

        # modify linearization coefficients on the boundaries
        coeffP[0] += coeffW[0] * self._west_bc.coeff()
        coeffP[-1] += coeffE[-1] * self._east_bc.coeff()

        # zero the bc that are not used
        coeffW[0] = 0.0
        coeffE[-1] = 0.0

        # calculate the net flux from each cell
        flux = flux_e -flux_w

        # add to coefficient arrays
        coeffs.accumulate_aE(coeffP)
        coeffs.accumulate_aW(coeffW)
        coeffs.accumulate_aE(coeffE)
        coeffs.accumulate_rP(flux)
        coeffs.accumulate_rP(imbalance)

    # return modified coeff arrays

```

```

    return coeffs

# Define the grid
lx = 1.0
ly = 0.1
lz = 0.1
ncv = 50
grid = Grid(lx, ly, lz, ncv)

# timestep information
nTime = 1
dt = 1e9
time = 0

# set iteration and convergence criterion
maxIter = 10
converged = 1e-6

# thermophysical properties
rho = 1000
cp = 4000
k = 0.5

# surface convection parameters
ho = 50
To = 200

# define coefficients
coeffs = ScalarCoeffs(grid.ncv)

# Initial conditions
T0 = 300
U0 = 0.01

# Initialize field variable arrays
T = T0*np.ones(grid.ncv+2)
Uhe = U0*np.ones(grid.ncv+1)

# define boundary conditions
west_bc = DirichletBC(T, grid, 400, BoundaryLocation.WEST)
east_bc = DirichletBC(T, grid, 0, BoundaryLocation.EAST)

# apply boundary conditions
west_bc.apply()
east_bc.apply()

# define the transient model
Told = np.copy(T)
transient = FirstOrderTransientModel(grid, T, Told, rho, cp, dt)

# define the diffusion model
diffusion = DiffusionModel(grid, T, k, west_bc, east_bc)

# define the surface convection model
surfaceConvection = SurfaceConvectionModel(grid, T, ho, To)

# define the advection model
advection = UpwindAdvectionModel(grid, T, Uhe, rho, cp, west_bc, east_bc)

iterList = []
resList = []

# loop through all timesteps
for tStep in range(nTime):
    # update time
    time += dt

    # print timestep information
    print("Timestep = {} ; Time = {}".format(tStep, time))

    # store the old temperature field
    Told[:] = T[:]

    # iterate until solution is converged
    for i in range(maxIter):
        # zero out the coefficients and add
        coeffs.zero()
        coeffs = diffusion.add(coeffs)
        coeffs = surfaceConvection.add(coeffs)
        coeffs = advection.add(coeffs)
        coeffs = transient.add(coeffs)

        # compute residual and check for convergence
        maxResid = norm(coeffs.rP, np.inf)
        avgResid = np.mean(np.absolute(coeffs.rP))
        print("Iteration = {} ; MaxResid = {} ; AvgResid = {}".format(i, maxResid, avgResid))

```

```
if maxResid < converged:  
    break  
  
# solve the sparse matrix system  
dT = solve(coeffs)  
  
# update the solution and boundary conditions  
T[1:-1] += dT  
west_bc.apply()  
east_bc.apply()  
  
resList.append(avgResid)  
iterList.append(i)  
  
plt.plot(iterList, resList)  
plt.show()
```

5 SOLUTION OF MASS AND MOMENTUM EQUATIONS

5.1 Problem Definition

Here, we will solve a system of coupled mass and momentum equations using the finite volume method. For compressible flow, we use the **density-based method**; however, this method is not applicable to incompressible flow. Here, we consider the **pressure-based methods**, which can be applied to both compressible and incompressible flow.

- Conservation of mass with constant density and no mass sources/sinks:

$$\nabla \cdot (\rho \mathbf{u}) = 0$$

- Conservation of momentum in x-direction, f_x is body force per unit volume

$$\frac{\partial \rho u}{\partial t} + \nabla \cdot (\rho \mathbf{u} u) = -\frac{dp}{dx} + \nabla \cdot (\mu \nabla u) + f_x$$

5.2 Discretization

Mass equation is discretized just like previous chapter, resulting in

$$\dot{m}_e - \dot{m}_w = 0$$

For the momentum equation, the procedure is similar: we integrate over both space and time.

For the transient term:

$$\int_{t-\Delta t}^{t+\Delta t} \int_V \frac{\partial(\rho u)}{\partial t} dV dt = (\rho u_P V_P)^{t+\Delta t/2} - (\rho u_P V_P)^{t-\Delta t/2}$$

Interpolation at values $t - \Delta t/2$ and $t + \Delta t/2$ is conducted like in previous chapter (i.e. first order implicit, second order implicit). Assuming constant density and constant volume, the transient term divided by Δt is:

$$\rho v_P \frac{(u_P)^{t+\Delta t/2} - (u_P)^{t-\Delta t/2}}{\Delta t}$$

For the convection term:

$$\int_V \nabla \cdot (\rho \mathbf{u} u) dV \xrightarrow{\text{Gauss}} \int_S \rho \mathbf{u} u \cdot \mathbf{n} dS \approx \sum_{i=0}^{N_{ip}-1} \rho u \mathbf{u} \cdot \mathbf{n}_{ip} A_{ip}$$

For 1D grid, this becomes: $\approx \dot{m}_e u_e - \dot{m}_w u_w$

For the pressure term, we treat it similar to a source term:

$$-\int_V \frac{dp}{dx} dV \approx -\left. \frac{dp}{dx} \right|_P V_P$$

For the viscous term, in the 1D case, we separate the the normal stresses (East/West) and the viscous shear stresses (North/South)

$$\begin{aligned} \int_V \nabla \cdot (\mu \nabla u) dV &= \int_S \mu \nabla u \cdot \mathbf{n} dS \approx \sum_{i=0}^{N_{ip}-1} \mu \nabla u \cdot \mathbf{n}_{ip} A_{ip} \\ &= \mu \left. \frac{\partial u}{\partial x} \right|_e A_e - \mu \left. \frac{\partial u}{\partial x} \right|_w A_w + \mu \left. \frac{\partial u}{\partial y} \right|_n A_n - \mu \left. \frac{\partial u}{\partial y} \right|_s A_s \\ &= \mu \left. \frac{\partial u}{\partial x} \right|_e A_e - \mu \left. \frac{\partial u}{\partial x} \right|_w A_w + F_u \end{aligned}$$

Note that all of our discretized terms are result of volume integration. For the time integration, each term will be multiplied by an additional Δt term. This is why our transient term has a $1/\Delta t$ factor, just so that we can get it in the form below. Note that F_u is the net viscous shear stress acting on the control volume.

Neglecting any body forces, our discretized momentum equation is as follow.

$$\rho v_P \frac{(u_P)^{t+\Delta t/2} - (u_P)^{t-\Delta t/2}}{\Delta t} + \dot{m}_e u_e - \dot{m}_w u_w = -\left. \frac{dp}{dx} \right|_P V_P + \mu \left. \frac{\partial u}{\partial x} \right|_e A_e - \mu \left. \frac{\partial u}{\partial x} \right|_w A_w + F_u$$

Note also how this equation is similar to the transport equation derived for convection of scalar, only with the addition of the pressure and viscous term. On the same note, the diffusion coefficients are defined similar to the energy equation, only replace k with μ :

$$D_e = \frac{\mu A_e}{\Delta x_{PE}}$$

$$D_w = \frac{\mu A_w}{\Delta x_{WP}}$$

With this in mind, we can estimate the diffusive terms using piecewise linear approximation, i.e:

$$\left. \mu \frac{\partial u}{\partial x} \right|_e A_e = D_e(u_E - u_P)$$

$$\left. \mu \frac{\partial u}{\partial x} \right|_w A_w = D_w(u_P - u_W)$$

We then subtract the mass equation multiplied by u_P from the momentum equation. Also substitute in the estimations for the diffusive terms:

$$\rho v_P \frac{(u_P)^{t+\Delta t/2} - (u_P)^{t-\Delta t/2}}{\Delta t} + \dot{m}_e(u_e - u_P) - \dot{m}_w(u_w - u_P) = -\left. \frac{dp}{dx} \right|_P V_P + D_e(u_E - u_P) - D_w(u_P - u_W) + F_u$$

A first order implicit time integration and UDS advection scheme will give the following cell residual:

$$r_P = \rho V_P \frac{u_P - u_P^o}{\Delta t} + \dot{m}_e \left[\left(\frac{1 + \alpha_e}{2} \right) u_P + \left(\frac{1 - \alpha_e}{2} \right) u_E - u_P \right]$$

$$\dot{m}_w \left[\left(\frac{1 + \alpha_w}{2} \right) u_W + \left(\frac{1 - \alpha_w}{2} \right) u_P - u_P \right] + \left. \frac{dp}{dx} \right|_P V_P$$

$$+ D_w(u_P - u_W) - D_e(u_E - u_P) - F_u$$

where the coefficients are as follow:

$$a_W = -D_w - \frac{\dot{m}_w}{2}(1 + \alpha_w)$$

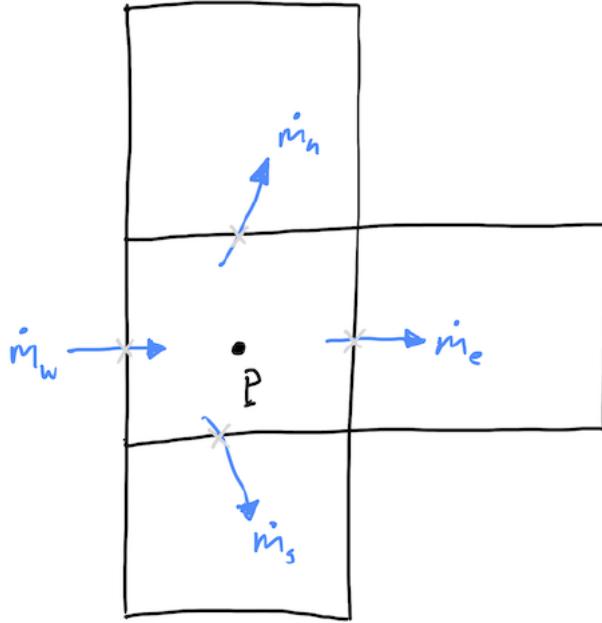
$$a_E = -D_e + \frac{\dot{m}_e}{2}(1 - \alpha_e)$$

$$a_P = \frac{\rho V_P}{\Delta t} - a_W - a_E$$

5.3 Pressure-Velocity Coupling

For incompressible flow problem, the pressure does not appear in the conservation of mass equation → because ρ is constant. As a result, the idea of "pressure-velocity coupling" for incompressible flow, is that we need the correct pressure that drives the velocities in the momentum equation such that mass is conserved. We can also think of the conservation of mass equation as a constraint so that the correct pressure can be determined.

- in 1D, knowing velocities at both inlet and outlet can help us to back out the pressure in the momentum equation.
- in 2D/3D, a mass flux through a west face is not certain to come out at the east face. In fact, we cannot be sure about how the mass flux will be splitted among the remaining faces. Fortunately, it is the pressure at the surrounding control volumes that determine how the mass is split.



Consider a steady, inviscid flow in a duct with uniform cross sectional area, the exact solution is that both velocities and pressure will be constant. For a control volume P , the discrete mass equation is:

$$\dot{m}_e - \dot{m}_w = 0$$

Likewise, the discrete momentum equation is:

$$\dot{m}_e u_e - \dot{m}_w u_w = -\frac{dp}{dx} \Big|_P V_P$$

Say we use central differences to calculate the mass fluxes base on the integration point velocities:

$$\begin{aligned}\dot{m}_e &= \rho A_e \left(\frac{u_P + u_E}{2} \right) \\ \dot{m}_w &= \rho A_w \left(\frac{u_W + u_P}{2} \right)\end{aligned}$$

The pressure gradient can be computed using the surrounding pressure, with $\Delta x = x_e - x_w$.

$$\frac{dp}{dx} \Big|_P = \frac{P_E - P_W}{2\Delta x}$$

Substituting the above expressions into the conservation of mass equation, we get:

$$\rho A_e \left(\frac{u_P + u_E}{2} \right) - \rho A_w \left(\frac{u_W + u_P}{2} \right) = 0$$

Assuming constant density and cross sectional area

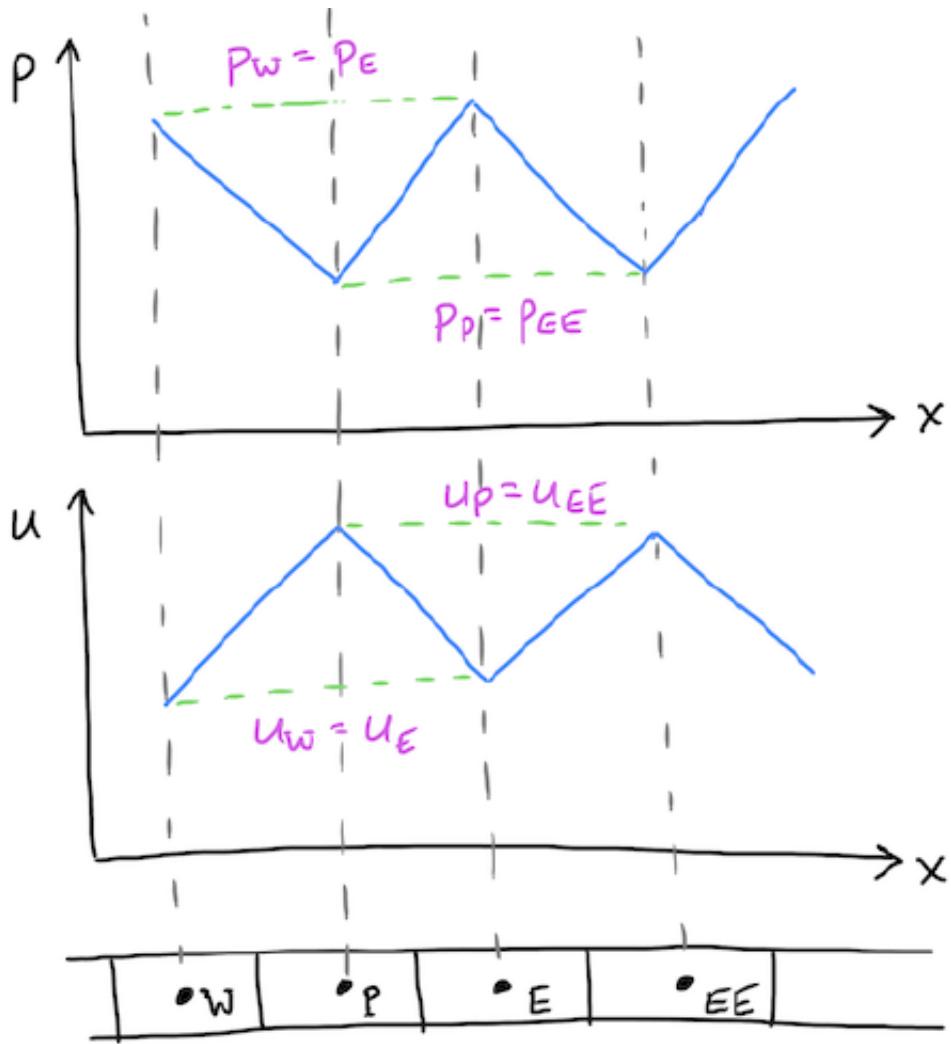
$$\left(\frac{u_P + u_E}{2} \right) - \left(\frac{u_W + u_P}{2} \right) = 0$$

$$[u_E = u_W]$$

Likewise, a substitution into the momentum equation gives:

$$\begin{aligned}\frac{\dot{m}_e^2}{\rho A_e} - \frac{\dot{m}_w^2}{\rho A_w} &= -\frac{p_E - p_W}{2\Delta x} V_P \\ 0 &= -\frac{p_E - p_W}{2\Delta x} V_P \\ [p_E &= p_W]\end{aligned}$$

These results make sense because it says that the velocity/pressure on the side of cell P must be equal. The problem is that it does not say anything about the pressure/velocity within the cell P . This allows for oscillating pressure/velocity field:



This suggests that there is an unconstrained mode in p and u that can grow without bound. This is still accepted as solution to the problem ! In other words, the solution to the side of cell P , cell W and E , can grow to any value, as long as they are both equal.

To solve this, we explore the idea that the CDS approximation could lead to this problem. Thus, we implemented an upwind scheme for evaluation of the mass fluxes (assuming flow in the positive direction):

$$\begin{aligned}\dot{m}_e &= \rho A_e u_P \\ \dot{m}_e &= \rho A_e u_P\end{aligned}$$

Our mass equation is then:

$$\begin{aligned}\rho A_e u_P - \rho A_w u_W &= 0 \\ u_P &= u_W\end{aligned}$$

The equation above will no longer allow the unconstrained velocity mode to exist. Here, only solution where u is constant everywhere is now allowed. In the momentum equation, the left side is still zero (since mass is conserved), resulting in:

$$p_E = p_W$$

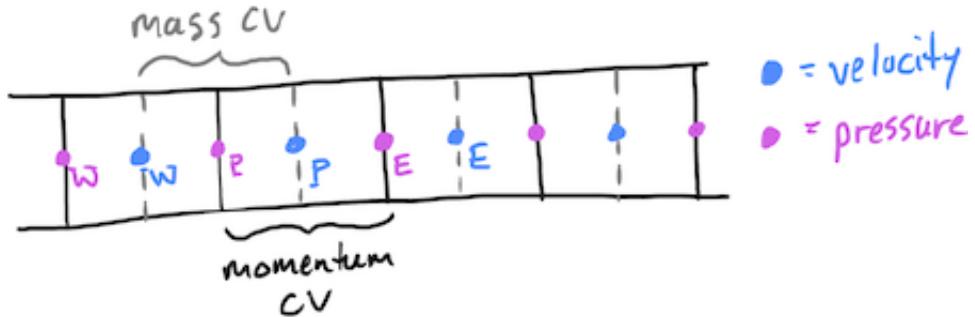
So the same problem still exists for the pressure field. In fact, no matter what approximation method is used for the integration point velocities, the same problem always exists for the pressure field if the same approximation is used to calculate the mass fluxes (because the left side of the momentum equation will cancel out exactly)

There are 2 main methods to address this problem with the pressure field:

- Staggered grids
- Collocated grids

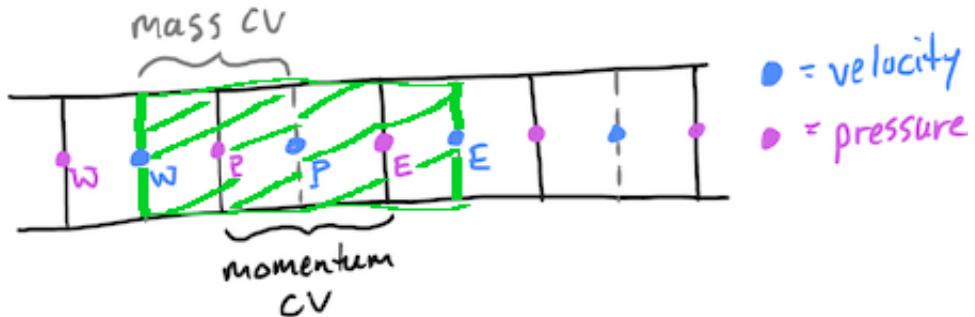
5.4 Staggered Grid Methods

This method uses 2 overlapping grids: one stores velocities at its cell center, the other stores pressure at its cell center. The grids are shifted by one half of the grid spacing relative to one another.



As we can see, there are 2 control volumes: mass and momentum. In the diagram above, the mass equation is evaluated over the mass control volume. Staggered grid stores velocities at the integration point of the mass control volume, so no interpolation is required.

A mass control volume P is as followed (shaded in green):

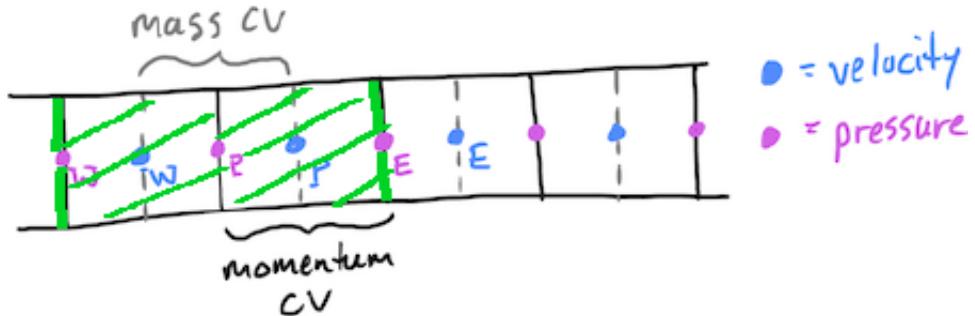


As in the diagram above, we can write a discrete mass equation for control volume P :

$$\begin{aligned}\dot{m}_e - \dot{m}_w &= 0 \\ \rho A_e u_P - \rho A_w u_W &= 0 \\ u_W &= u_P\end{aligned}$$

Note: the subscripts on the velocities are in reference to the momentum control volume labeling. The equation above implies that there can be no decoupling of the velocity field.

A momentum control volume P is as followed (shaded in green):



The momentum equation can be written out based on the diagram above:

$$\begin{aligned}\dot{m}_e u_e - \dot{m}_w u_w &= -\frac{dp}{dx} \Big|_P V_P \\ \frac{\dot{m}_e^2}{\rho A_e} - \frac{\dot{m}_w^2}{\rho A_w} &= -\frac{p_E - p_P}{\Delta x} V_P \\ 0 &= -\frac{p_E - p_P}{\Delta x} V_P \\ p_E &= p_P\end{aligned}$$

This shows that staggered grid is effective in removing possibility of pressure oscillations, which prevents the decoupling of velocity and pressure fields. One thing to note is that staggered grid becomes impractical in arbitrary unstructured meshes, which is the standard for CFD involving complex geometries.

5.5 Collocated Grid Methods

Developed by Rhee and Chow (1983), collocated grid is a method where velocity and pressure share the same grid. This was successful in maintaining coupling between velocity and pressure fields. The main ideas are:

- adverted velocity is obtained from a different equation from the advecting velocity
- advecting velocity: used in the calculation of mass flux
- adverted velocity: the one that is multiplied by the mass flux in the advection term

For the East face, the mass flux is defined as:

$$\dot{m}_e = \rho A_e \hat{u}_e$$

- here, \hat{u}_e is the advecting velocity. This term is obtained from a special momentum equation, deriving later.
- Our advected velocity is the same as before, u_e . It is calculated using the deferred correction approach like before.
- We desire $\hat{u}_e \approx u_e$, but in reality, the expressions contain different influences and oscillating pressure and velocity fields will be damped out of the solution.

To derive the special momentum equation and the corresponding advecting velocity, we recall the cell residual that was derived previously:

$$r_P = \rho V_P \frac{(u_P)^{t+\Delta t/2} - (u_P)^{t-\Delta t/2}}{\Delta t} + \dot{m}_e u_e - \dot{m}_w u_w + \left. \frac{dp}{dx} \right|_P V_P - \mu \left. \frac{\partial u}{\partial x} \right|_e A_e + \mu \left. \frac{\partial u}{\partial x} \right|_w A_w - F_u$$

Assume a converged, steady state solution $\rightarrow r_P = 0$ and transient term vanishes, the above equation becomes:

$$\dot{m}_e u_e - \dot{m}_w u_w + \left. \frac{dp}{dx} \right|_P V_P - \mu \left. \frac{\partial u}{\partial x} \right|_e A_e + \mu \left. \frac{\partial u}{\partial x} \right|_w A_w - F_u = 0$$

We implement piecewise profile for the velocity derivatives and UDS for the advection term, this becomes:

$$\begin{aligned} & \dot{m}_e \left[\left(\frac{1 + \alpha_e}{2} \right) u_P + \left(\frac{1 + \alpha_e}{2} \right) u_E - u_P \right] \\ & - \dot{m}_w \left[\left(\frac{1 + \alpha_w}{2} \right) u_W + \left(\frac{1 - \alpha_w}{2} \right) u_P - u_P \right] + \left. \frac{dp}{dx} \right|_P V_P + D_w(u_P - u_W) \\ & - D_e(u_E - u_P) - F_u = 0 \end{aligned}$$

or:

$$\begin{aligned} & \left[D_w + \frac{\dot{m}_w}{2} (1 + \alpha_w) + D_e - \frac{\dot{m}_e}{2} (1 - \alpha_e) \right] u_P + \left[-D_w - \frac{\dot{m}_w}{2} (1 + \alpha_w) \right] \\ & + \left[-D_e + \frac{\dot{m}_e}{2} (1 - \alpha_e) \right] u_E - F_u + \left. \frac{dp}{dx} \right|_P V_P = 0 \end{aligned}$$

where the linearization coefficients are recalled as follow:

$$\begin{aligned} a_W &= -D_w - \frac{\dot{m}_w}{2} (1 + \alpha_w) \\ a_E &= -D_e - \frac{\dot{m}_e}{2} (1 - \alpha_e) \\ a_P &= D_w - \frac{\rho V_P}{\Delta t} - a_W - a_E \end{aligned}$$

We can then define:

$$\bar{a}_P = -a_W - a_E$$

Note how this definition removes the timestep dependence from the linearization coefficient P . This is done to make the pressure-velocity coupling independent of timestep. Our momentum equation thus becomes:

$$\bar{a}_P u_P + a_W u_W + a_E u_E - b_P + \left. \frac{dp}{dx} \right|_P V_P = 0$$

with b_P containing all body force terms. Next, we define:

$$\bar{u}_P = -a_W u_W - a_E u_E + b_P$$

or more generally:

$$\bar{u}_P = - \sum_{nb} a_{nb} u_{nb} + b_P$$

With this, our momentum equation becomes:

$$\bar{a}_P u_P = \bar{u}_P - \frac{dp}{dx} \Big|_P V_P$$

For an east control volume:

$$\bar{a}_E u_E = \bar{u}_E - \frac{dp}{dx} \Big|_E V_E$$

By analogy, for a virtual control volume at the east integration point:

$$\bar{a}_e \hat{u}_e = \bar{u}_e - \frac{dp}{dx} \Big|_e V_e$$

The equation above defines the advecting velocity. The quantity \bar{u}_e is obtained by central differencing from the P and E values:

$$\begin{aligned} \bar{u}_e &= \frac{1}{2}(\bar{u}_P + \bar{u}_E) \\ &= \frac{1}{2} \left(\bar{a}_P u_P + \bar{a}_E u_E + \frac{dp}{dx} \Big|_P V_P + \frac{dp}{dx} \Big|_E V_E \right) \end{aligned}$$

We make further approximations within the \bar{u}_e term:

$$\begin{aligned} \bar{a}_P &\approx \bar{a}_E \approx \bar{a}_e \\ V_P &\approx V_E \approx V_e \end{aligned}$$

With these approximations, our quantity \bar{u}_e is then:

$$\bar{u}_e = \frac{\bar{a}_e}{2}(u_P + u_E) + \frac{V_e}{2} \left(\frac{dp}{dx} \Big|_P + \frac{dp}{dx} \Big|_E \right)$$

Thus, the expression for the advecting velocity is then:

$$\bar{a}_e \hat{u}_e = \frac{\bar{a}_e}{2}(u_P + u_E) + \frac{V_e}{2} \left(\frac{dp}{dx} \Big|_P + \frac{dp}{dx} \Big|_E \right) - \frac{dp}{dx} \Big|_e V_e$$

or:

$$\hat{u}_e = \frac{1}{2}(u_P + u_E) - \frac{V_e}{\bar{a}_e} \left[\frac{dp}{dx} \Big|_e - \frac{1}{2} \left(\frac{dp}{dx} \Big|_P + \frac{dp}{dx} \Big|_E \right) \right]$$

If we go ahead and define:

$$\begin{aligned} \bar{a}_e &\approx \frac{1}{2}(\bar{a}_P + \bar{a}_E) \\ V_e &\approx \frac{1}{2}(V_P + V_E) \\ \hat{d}_e &= \frac{V_e}{\bar{a}_e} \end{aligned}$$

Note: for a cell wth index i , $\bar{a}_E = \bar{a}_P$ at index $i + 1$. With these new definitions, the advecting velocity is written as:

$$\hat{u}_e = \frac{1}{2}(u_P + u_E) - \hat{d}_e \bar{a}_e \left[\frac{dp}{dx} \Big|_e - \frac{1}{2} \left(\frac{dp}{dx} \Big|_P + \frac{dp}{dx} \Big|_E \right) \right]$$

Note:

- the 1st term on RHS is the CDS approximation of the east face velocity. We used this to calculate the mass flux, leading to unconstrained velocity and pressure modes.
- the 2nd term on RHS is a 4th order pressure correction term. It smooths out any oscillations from the pressure field.

- it is 4th order because evaluations of the pressure gradient involve 4 location: W , P , E and EE .
- coefficient \hat{d}_e can be interpreted as a relaxation parameter. Generally, the pressure term should be small, which keeps the advecting velocity close to the CDS approximation. Else, if the pressure oscillations emerge, the pressure terms will be activated to damp out.

For the east face, the pressure gradient is calculated as:

$$\left. \frac{dp}{dx} \right|_e = \frac{p_E - p_P}{\Delta x_{PE}}$$

For the cell at E and P , the pressure gradient is calculated as:

$$\begin{aligned} \left. \frac{dp}{dx} \right|_P &= \frac{p_E - p_W}{\Delta x_{WP} + \Delta x_{PE}} \\ \left. \frac{dp}{dx} \right|_E &= \frac{p_{EE} - p_P}{\Delta x_{PE} + \Delta x_{E,EE}} \end{aligned}$$

5.6 Coupled (Direct) Solution Method

We need to compute the mass fluxes using the advecting velocity above. In terms of the advecting velocity, the discretized conservation of mass equation looks like this:

$$\rho A_e \hat{u}_e - \rho A_w \hat{u}_w = 0$$

or:

$$\begin{aligned} \rho A_e \frac{1}{2}(u_P + u_E) - \rho A_e \hat{d}_e \left[\frac{p_E - p_P}{\Delta x_{PE}} - \frac{1}{2} \left(\left. \frac{dp}{dx} \right|_P + \left. \frac{dp}{dx} \right|_E \right) \right] - \rho A_w \frac{1}{2}(u_W + u_P) \\ + \rho A_w \hat{d}_w \left[\frac{p_P - p_W}{\Delta x_{WP}} - \frac{1}{2} \left(\left. \frac{dp}{dx} \right|_W + \left. \frac{dp}{dx} \right|_P \right) \right] = 0 \end{aligned}$$

Collecting all the pressure and velocity terms together, we then rewrite the result as a residual for the mass equation:

$$\begin{aligned} r_P = & \left[\frac{\rho A_e \hat{d}_e}{\Delta x_{PE}} + \frac{\rho A_w \hat{d}_w}{\Delta x_{WP}} \right] p_P + \left[\frac{\rho A_e}{2} - \frac{\rho A_w}{2} \right] u_P - \left[\frac{\rho A_w \hat{d}_w}{\Delta x_{WP}} \right] p_W \\ & - \left[\frac{\rho A_w}{2} \right] u_W - \left[\frac{\rho A_e \hat{d}_e}{\Delta x_{PE}} \right] p_E + \left[\frac{\rho A_e}{2} \right] u_E \\ & - \frac{\rho A_w \hat{d}_w}{2} \left[\left. \frac{dp}{dx} \right|_W + \left. \frac{dp}{dx} \right|_P \right] + \frac{\rho A_e \hat{d}_e}{2} \left[\left. \frac{dp}{dx} \right|_P + \left. \frac{dp}{dx} \right|_E \right] \end{aligned}$$

We see that our mass equation now contains pressure. This allows us to solve for both velocity and pressure simultaneously, because we have 2 equations with 2 unknowns. We will linearize w.r.t both pressure and velocity. However, we leave the final 2 terms as explicit "lagged" correction terms to eliminate decoupling. Let the solution variable be the vector $[p, u]$, the linearized problem is as follows:

$$\begin{bmatrix} a_P^{pp} & a_P^{pu} \\ a_P^{up} & a_P^{uu} \end{bmatrix} \begin{Bmatrix} \delta p_P \\ \delta u_P \end{Bmatrix} + \begin{bmatrix} a_W^{pp} & a_W^{pu} \\ a_W^{up} & a_W^{uu} \end{bmatrix} \begin{Bmatrix} \delta p_W \\ \delta u_W \end{Bmatrix} + \begin{bmatrix} a_E^{pp} & a_E^{pu} \\ a_E^{up} & a_E^{uu} \end{bmatrix} \begin{Bmatrix} \delta p_E \\ \delta u_E \end{Bmatrix} = - \begin{Bmatrix} r_P^p \\ r_P^u \end{Bmatrix}$$

Note:

- 1st row in matrix: mass equation denoted by p
- 2nd row in matrix: momentum equation denoted by u
- superscripts: equation and variable to which the coefficient is associated.
 - 1st letter: row or the equation
 - 2nd letter: column or the variable
 - Eg. pu represent the velocity in the momentum equation.

5.6.1 Linearization of Momentum Equation

For the linearized coefficients for velocity in the momentum equation, we have already established them:

$$\begin{aligned} a_W^{uu} &= -D_w - \frac{\dot{m}_w}{2}(1 + \alpha_w) \\ a_E^{uu} &= -D_e + \frac{\dot{m}_e}{2}(1 - \alpha_e) \\ a_P^{uu} &= \frac{\rho V_p}{\Delta t} - a_W^{uu} - a_E^{uu} \end{aligned}$$

The pressure term can be discretized as follow:

$$\left. \frac{dp}{dx} \right|_P V_P = \frac{1}{2} \left[\frac{p_P - p_W}{\Delta x_{WP}} + \frac{p_E - p_P}{\Delta x_{PE}} \right] V_P$$

Linearization coefficients for pressure are as follow:

$$\begin{aligned} a_W^{up} &= -\frac{V_p}{2\Delta x_{WP}} \\ a_E^{up} &= \frac{V_p}{2\Delta x_{PE}} \\ a_P^{up} &= -a_W^{up} - a_E^{up} \end{aligned}$$

5.6.2 Linearization of Mass Equation

This is based on the residual form of the mass equation above.

- For linearization w.r.t pressure:

$$\begin{aligned} a_W^{pp} &= -\frac{\rho A_w \hat{d}_w}{\Delta x_{WP}} \\ a_E^{pp} &= -\frac{\rho A_e \hat{d}_e}{\Delta x_{PE}} \\ a_P^{pp} &= -a_W^{pp} - a_E^{pp} \end{aligned}$$

- For linearization w.r.t velocity:

$$\begin{aligned} a_W^{pu} &= -\frac{\rho A_w}{2} \\ a_E^{pu} &= \frac{\rho A_e}{2} \\ a_P^{pu} &= a_W^{pu} + a_E^{pu} \end{aligned}$$

Note: these equations are only valid in control volume where advecting velocity is computed based on the special momentum equation (i.e. interior faces only). At the boundary, the special momentum is not needed, so our mass equation needs to be modified.

Consider the **LEFT BOUNDARY** control volume, here $u_w = u_W$ is specified through the boundary condition. The mass equation is then:

$$\rho A_e \hat{u}_e - \rho A_w u_W = 0$$

where the advecting velocity is calculated using the special momentum equation. We carry the procedure in a similar manner as before, resulting in:

$$\begin{aligned} \rho A_e \frac{1}{2}(u_P + u_E) - \rho A_e \hat{d}_e \left[\frac{p_E - p_P}{\Delta x_{PE}} - \frac{1}{2} \left(\left. \frac{dp}{dx} \right|_P + \left. \frac{dp}{dx} \right|_E \right) \right] - \rho A_w u_W &= 0 \\ a_W^{pp} &= 0 \\ a_E^{pp} &= -\frac{\rho A_e \hat{d}_e}{\Delta x_{PE}} \\ a_P^{pp} &= -a_E^{pp} \\ a_W^{pu} &= -\rho A_w \\ a_E^{pu} &= \frac{\rho A_e}{2} \\ a_P^{pu} &= a_E^{pu} \end{aligned}$$

5.7 Implementation

```
from Lesson5.Grid import Grid
from Lesson5.ScalarCoeffs import ScalarCoeffs
from Lesson5.BoundaryConditions import BoundaryLocation, DirichletBc, NeumannBc
from Lesson5.Models import DiffusionModel
from Lesson5.LinearSolver import solve
from Lesson5.Models import FirstOrderTransientModel, DiffusionModel, UpwindAdvectionModel, PressureForceModel
from Lesson5.Models import AdvectionVelocityModel, MassConservationEquation

import numpy as np
from numpy.linalg import norm

## MAIN CODE ##
lx = 4.0
ly = 0.02
lz = 0.02
ncv = 10
grid = Grid(lx, ly, lz, ncv)

# set timestep
nTime = 1
dt = 1e-2
time = 0

# iteration info
maxIter = 100
converged = 1e-6

# thermo properties
rho = 1000
mu = 1e-3

# coefficients
PU_coeffs = ScalarCoeffs(grid.ncv)
PP_coeffs = ScalarCoeffs(grid.ncv)
UP_coeffs = ScalarCoeffs(grid.ncv)
UU_coeffs = ScalarCoeffs(grid.ncv)

# initial condition
U0 = 10
P0 = 0

# initialize field variable arrays
U = U0*np.ones(grid._ncv+2)
P = P0*np.ones(grid._ncv+2)

# initialize advecting and damping coefficient arrays
dhat = np.zeros(grid.ncv+1)
Uhe = np.zeros(grid.ncv+1)

# define boundary conditions for velocity
U_west_bc = DirichletBc(U, grid, U0, BoundaryLocation.WEST)
U_east_bc = NeumannBc(U, grid, 0, BoundaryLocation.EAST)

# define boundary conditions for pressure
P_west_bc = NeumannBc(P, grid, 0, BoundaryLocation.WEST)
P_east_bc = DirichletBc(P, grid, 0, BoundaryLocation.EAST)

# apply boundary conditions
U_west_bc.apply()
U_east_bc.apply()
P_west_bc.apply()
P_east_bc.apply()

# define the transient mdel
Uold = np.copy(U)
transient = FirstOrderTransientModel(grid, U, Uold, rho, 1, dt)

# define the diffusion model
diffusion = DiffusionModel(grid, U, mu, U_west_bc, U_east_bc)

#define advection model
advection = UpwindAdvectionModel(grid, U, Uhe, rho, 1, U_west_bc, U_east_bc)

# define the pressure force model
pressure = PressureForceModel(grid, P, P_west_bc, P_east_bc)

# define the advecting velocity model
advection = AdvectionVelocityModel(grid, dhat, Uhe, P, U, UU_coeffs)

# define conservation of mass equation
mass = MassConservationEquation(grid, U, P, dhat, Uhe, rho, P_west_bc,
                                 P_east_bc, U_west_bc, U_east_bc)

# Loop through all timesteps
```

```

for tStep in range(nTime):
    time += dt
    print("Timestep = {}; Time = {}".format(tStep, time))

Uold[:, :] = U[:, :]

for i in range(maxIter):
    # zero out coefficients
    PP_coeffs.zero()
    PU_coeffs.zero()
    UU_coeffs.zero()
    UP_coeffs.zero()

    # assemble the momentum equation
    UU_coeffs = diffusion.add(UU_coeffs)
    PU_coeffs = advection.add(UU_coeffs)
    UU_coeffs = transient.add(UU_coeffs)
    UP_coeffs = pressure.add(UP_coeffs)

    # assemble the mass equation
    adverting.update()
    PP_coeffs, PU_coeffs = mass.add(PP_coeffs, PU_coeffs)

    # compute the residual and check for convergence
    PmaxResid = norm(PU_coeffs.rP + PP_coeffs.rP, np.inf)
    PavgResid = np.mean(np.absolute(PU_coeffs.rP + PP_coeffs.rP))

    UmaxResid = norm(UU_coeffs.rP + UP_coeffs.rP, np.inf)
    UavgResid = np.mean(np.absolute(UU_coeffs.rP + UP_coeffs.rP))

    print("Iteration = {}".format(i))
    print("Mass: Max. Resid = {}; Avg. Resid = {}".format(PmaxResid, PavgResid))
    print("Momentum: Max. Resid = {}; Avg. Resid = {}".format(UmaxResid, UavgResid))

    if PmaxResid < converged and UmaxResid < converged:
        break

    # solve the sparse matrix system
    dP, dU = solve(PP_coeffs, PU_coeffs, UP_coeffs, UU_coeffs)

    # update the solution
    P[1:-1] += dP
    U[1:-1] += dU

    # update boundary condition
    U_west_bc.apply()
    U_east_bc.apply()
    P_west_bc.apply()
    P_east_bc.apply()

    # update the advecting velocities
    adverting.update()

import matplotlib.pyplot as plt
plt.plot(grid.xP, U)
plt.show()

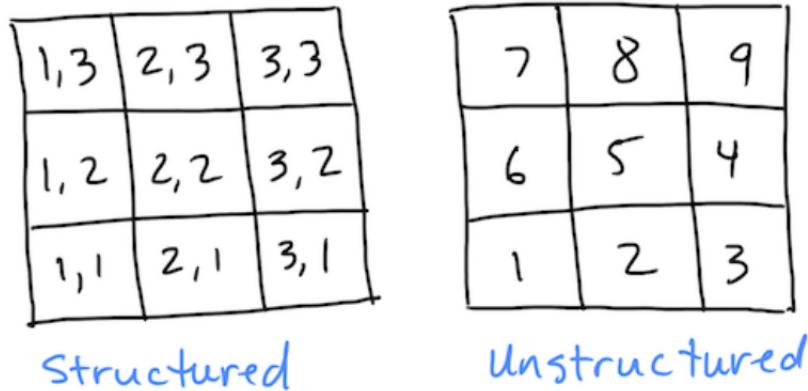
```

6 Two and Three Dimensional Grids

6.1 General introduction

If the grids have an ordered, orthogonal structure, then it is easy to extend what we have done so far to higher dimensional grids.

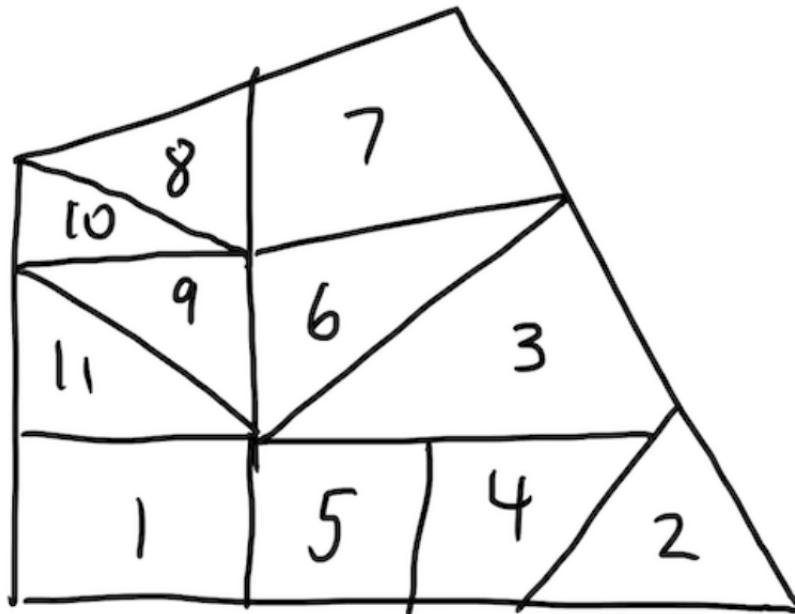
We will discuss **structured** vs **unstructured** grid. The main idea is that the labels are associated with how the grid cells are labeled, and not their topologies.



We can see from the image above that:

- **structured grid** uses **ordered** row and column labelling
- **unstructured grid** uses **arbitrary** cell labelling

As a result, for grid with arbitrary polyhedra with no inherent structure, it is better to store the grid with an unstructured labelling scheme, as shown below.



6.2 Structured Grid

6.2.1 General Discretization

The goal is to discretize a generic transport equation over a 2D structured grid. Then we will extend from 2D to 3D. Recall our generic transport equation:

$$\frac{\partial \phi}{\partial t} + \nabla \cdot (\mathbf{u}\phi) + \nabla \cdot \mathbf{J}_\phi = S_\phi$$

Integrate through space and time, we arrive at:

$$\frac{\phi^{t+\Delta t/2} - \phi^{t-\Delta t/2}}{\Delta t} + \sum_{i=0}^{N_{ip}-1} \mathbf{u}_{ip} \cdot \mathbf{n}_{ip} \phi_{ip} A_{ip} = \sum_{i=0}^{N_{ip}-1} \mathbf{J}_{\phi,ip} \cdot \mathbf{n}_{ip} A_{ip} + S_{\phi} V_P$$

1. Mass Equation

Say we want to make this general form to represent the conservation of mass equation, we set $\phi = \rho$, $\mathbf{J}_\phi = 0$ and $S_\phi = 0$. Then, using the integration points as *w* (west), *e* (east), *s* (south) and *n* (north), we get:

$$\frac{\rho^{t+\Delta t/2} - \rho^{t-\Delta t/2}}{\Delta t} + \dot{m}_e - \dot{m}_w + \dot{m}_n - \dot{m}_s = 0$$

For the case of constant density:

$$\dot{m}_e - \dot{m}_w + \dot{m}_n - \dot{m}_s = 0$$

2. Momentum Equation

Likewise, to make the general form representing the conservation of momentum in the *x*-direction, we simply set $\phi = \rho u$, $\mathbf{J}_\phi = \mu \nabla u$ and $S_\phi = -\partial p / \partial x$.

$$\begin{aligned} \frac{\rho^{t+\Delta t/2} - \rho^{t-\Delta t/2}}{\Delta t} + \dot{m}_e u_e - \dot{m}_w u_w + \dot{m}_n u_n - \dot{m}_s u_s &= \mu \left. \frac{\partial u}{\partial x} \right|_e - \mu \left. \frac{\partial u}{\partial x} \right|_w \\ &+ \mu \left. \frac{\partial u}{\partial y} \right|_n - \mu \left. \frac{\partial u}{\partial y} \right|_s - \frac{\partial p}{\partial y} V_P \end{aligned}$$

The discretization procedure for the momentum equation is as follow:

- appropriate momentum equation - (mass equation * appropriate velocity component at cell P)
- choose time integration scheme for the transient term
- choose advection scheme for the advection term
- approximate the derivatives using finite differences.

The momentum equations in *x* and *y* can be written in terms of their linearization coefficients:

$$\begin{aligned} a_p u_p &= -a_W u_W - a_E u_E - a_S u_S - a_N u_N + b_u - \frac{p_E - p_W}{2\Delta x} V_P \\ a_p v_p &= -a_W v_W - a_E v_E - a_S v_S - a_N v_N + b_v - \frac{p_N - p_S}{2\Delta y} V_P \end{aligned}$$

Recall back to our 1D case, the pressure can field can oscillate if we are not careful. In 2D, this is even worse because the oscillations can happen in both directions. An example of an oscillated pressure field but would be accepted by the solver as a smooth pressure field is shown below:

100	0	100	0	100
50	-100	50	-100	50
100	0	100	0	100
50	-100	50	-100	50

Same as the 1D case, this oscillations can be solved by

- using staggered grid
- using collocated grid with different advected and advecting velocities.

because the staggered grid becomes more complicated at higher dimensions, we only consider collocated grid. The resulting expressions for advecting velocities in each direction are as follow:

$$\begin{aligned}\hat{u}_e &= \frac{1}{2}(u_P + u_E) - \hat{d}_e^u \left[\frac{dp}{dx} \Big|_e - \frac{1}{2} \left(\frac{dp}{dx} \Big|_P + \frac{dp}{dx} \Big|_E \right) \right] \\ \hat{v}_n &= \frac{1}{2}(v_P + v_N) - \hat{d}_n^v \left[\frac{dp}{dy} \Big|_n - \frac{1}{2} \left(\frac{dp}{dy} \Big|_P + \frac{dp}{dy} \Big|_N \right) \right]\end{aligned}$$

with the superscript on \hat{d} represents the equation it is associated with. Similar to 1D, the coupling can be done direct or segregated method, e.g. SIMPLE or SIMPLEC

6.2.2 False Diffusion

Recall in 1D, Taylor series analysis shows some serious problem, but they are not as bad. Using UDS for linearization and correcting the advective fluxes with a higher order method, we can get good results in these cases.

In 2D and 3D, false diffusion comes from a different source than 1D. Usually, this associates with cases where flow streamlines are not well aligned with the grid lines.

Consider a steady advection of scalar quantity with **no sources**, real diffusion is **negligible** compare to advection.

$$\dot{m}_e \phi_e - \dot{m}_w \phi_w + \dot{m}_n \phi_n - \dot{m}_s \phi_s = 0$$

Assume positive flow in x and y and using UDS for advection:

$$\dot{m}_e \phi_P - \dot{m}_w \phi_W + \dot{m}_n \phi_P - \dot{m}_s \phi_S = 0$$

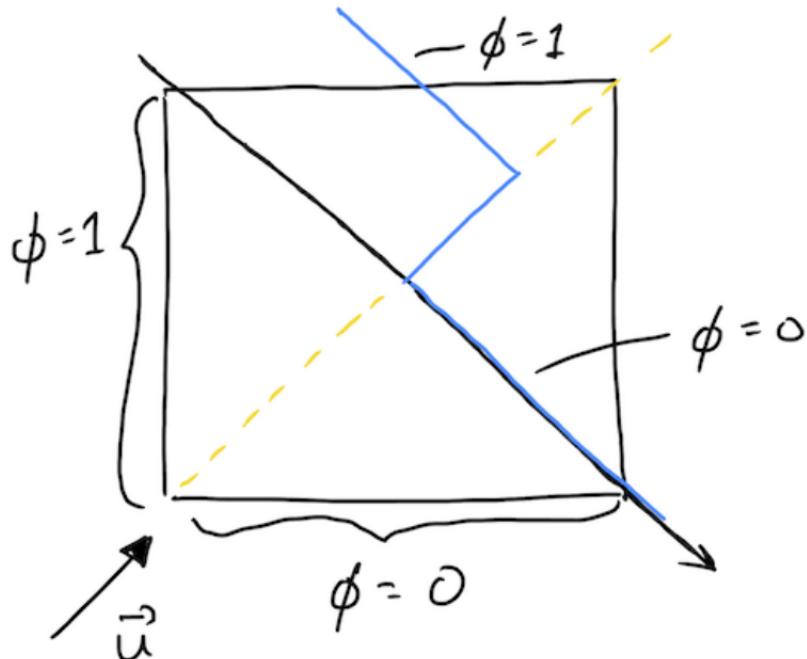
Solving for ϕ_P , we get:

$$\phi_P = \frac{\dot{m}_w}{\dot{m}_e + \dot{m}_n} \phi_W + \frac{\dot{m}_s}{\dot{m}_e + \dot{m}_n} \phi_S$$

Consider flow at 45 degrees to the x axis, meaning $\dot{m}_e = \dot{m}_w = \dot{m}_n = \dot{m}_s$, we have:

$$\phi_P = \frac{1}{2} \phi_W + \frac{1}{2} \phi_S$$

If $\phi = 0$ is advected from the bottom, and $\phi = 1$ is advected from the left. Then the exact solution is a step profile at any cross-section perpendicular to the flow direction:



Below is the code to solve this problem:

```

import matplotlib
import matplotlib.pyplot as plt
import numpy as np
matplotlib.use('Agg')
fname = '../pic/45degreeStep.png'

# create array to hold the solution
phi = np.zeros((7,7))

# set left advected value
phi[:,0] = 1

print('Initial matrix:\n')
print(np.vectorize("%3f".__mod__)(phi))
# compute the solution starting from the bottom left
for j in reversed(range(phi.shape[0]-1)):
    for i in range(1,phi.shape[1]):
        phi[j,i] = 0.5*phi[j,i-1] + 0.5*phi[j+1,i]

# print the solution matrix
print('\nSolution matrix:\n')
print(np.vectorize("%3f".__mod__)(phi))

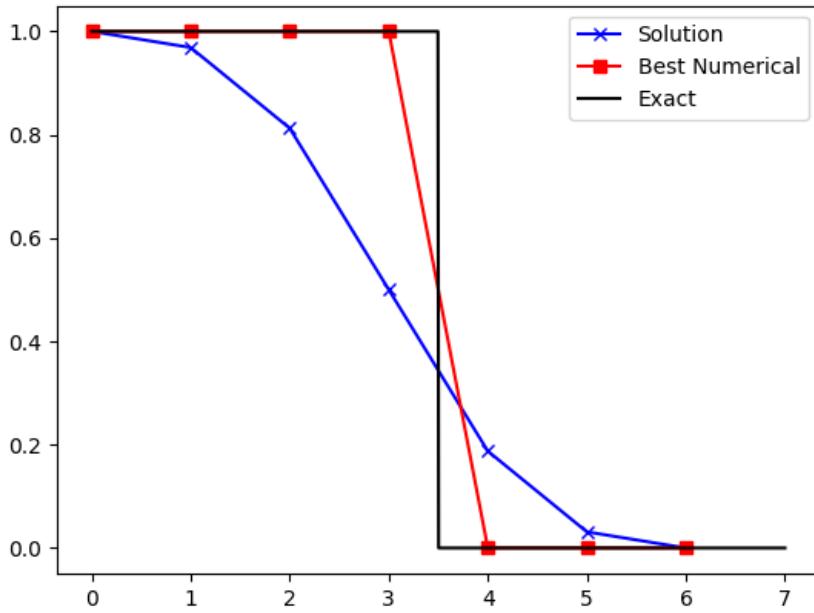
# plot solution along the diagonal cross-section
sol = np.diag(phi)
x = np.array([i for i in range(sol.size)])
plt.plot(x,sol,'-bx', label='Solution')

# plot the best possible numerical solution based on this grid
best = np.where(x < x.size/2.0,1,0)
plt.plot(x , best, '-rs',label = 'Best Numerical')

# plot exact solution on fine grid
x_exact = np.linspace(0,x.size,1000)
exact = np.where(x_exact < x.size/2.0,1,0)
plt.plot(x_exact, exact, 'k-',label = 'Exact')

plt.legend()
plt.savefig(fname)

```



We can see that the solution looks quite diffusive. Because there is no actual diffusion, all of this represents false diffusion. To get a good solution, the false diffusion coefficient Γ_{false} should much less than the real diffusion coefficient Γ_{real} . In 2D, our false diffusion coefficient looks like this:

$$\Gamma_{false} = \frac{\rho|\mathbf{u}|\Delta x \Delta y \sin(2\Theta)}{4(\Delta y \sin^3 \Theta + \Delta x \cos^3 \Theta)}$$

where $\Delta x, \Delta y$ are grid spacing in each direction and Θ is the angle the velocity makes with the x-axis. Here, we plot the values of Γ_{false} against different values of grid spacing and Θ .

```

import matplotlib
import matplotlib.pyplot as plt
import numpy as np
matplotlib.use('Agg')
fname = '../pic/false_diffusion2D.png'
# assume velocity , density magnitudes have unit values (1)
# set parameters of study
delta = [0.01, 0.05, 0.1]
theta = np.linspace(0,np.pi/2,100)

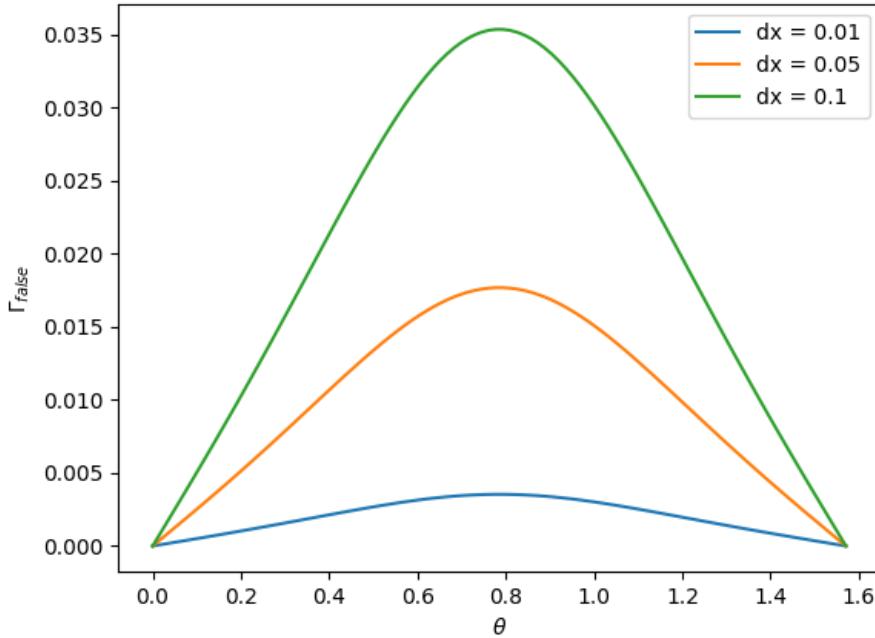
# for tables
headers = ["delta", "theta", "gamma"]

# calculate the false diffusion
for d in delta:
    gamma = d*d*np.sin(2*theta)/4/(d*np.power(np.sin(theta),3) +
                                    d*np.power(np.cos(theta),3))
    plt.plot(theta, gamma, label = "dx = " + str(d))

print("THETA      GAMMA")
print("THETA      GAMMA")
for i,theta_i in enumerate(theta):
    print("%2f | %2f" %(theta_i, gamma[i]) )

# show the plot
plt.xlabel(r"\theta")
plt.ylabel(r"\Gamma_{false}")
plt.legend()
plt.savefig(fname)

```



Note:

- when flow is essentially parallel to the grid lines, the false diffusion is zero.
- False diffusion is most severe (highest) when the flow is at 45 degrees ($\frac{pi}{2} \approx 0.8$)
- Refining the grid spacing reduces the diffusion.
- To improve accuracy: use higher order advection schemes (reduce effects of false diffusion) and ensure grid is fine enough.

6.3 Non-Orthogonal Unstructured Grid

- There is no natural ordering available in an unstructured grid.

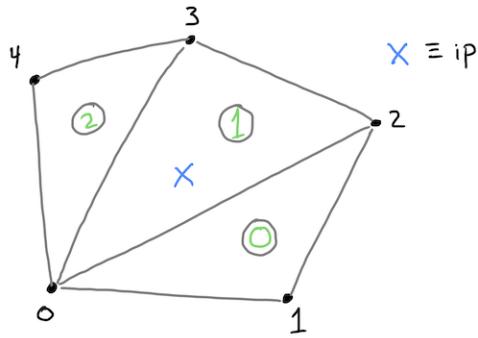
- In structured grid, the next neighboring control volume can be found by shifting the current cell index. In unstructured grid, we need a map that stores the cell connectivity. In other words, each cell needs a list of all of its neighboring cells.
- Problems include grid geometry, interpolation and gradient reconstruction

6.3.1 Grid Geometry

We have a set of points representing the corners of the control volume. These points are connected by edges, defining a set of faces. Each face belongs to 2 control volumes, one on each side.

To calculate the grid geometry, we start with the faces and build the volumes. Assume that the faces are arbitrary polygons, these combine to make arbitrary polyhedral control volumes.

First, we start by choosing an arbitrary corner node, then we connect it with the other corner nodes, creating a set of triangular faces. The area of each triangle can be calculated using cross products.



$$\begin{aligned} A_0 &= \frac{1}{2} \|(\mathbf{x}_1 - \mathbf{x}_0) \times (\mathbf{x}_2 - \mathbf{x}_0)\| \\ A_1 &= \frac{1}{2} \|(\mathbf{x}_2 - \mathbf{x}_0) \times (\mathbf{x}_3 - \mathbf{x}_0)\| \\ A_2 &= \frac{1}{2} \|(\mathbf{x}_3 - \mathbf{x}_0) \times (\mathbf{x}_4 - \mathbf{x}_0)\| \end{aligned}$$

In general, for triangle with index i :

$$A_i = \frac{1}{2} \|(\mathbf{x}_{i+1} - \mathbf{x}_0) \times (\mathbf{x}_{i+2} - \mathbf{x}_0)\|$$

For even more general case with N_c corner nodes, the total area associated with the integration point ip is calculated as:

$$A_{ip} = \sum_{i=0}^{N_c-2} A_i = \frac{1}{2} \sum_{i=0}^{N_c-2} \|(\mathbf{x}_{i+1} - \mathbf{x}_0) \times (\mathbf{x}_{i+2} - \mathbf{x}_0)\|$$

Recall the centroid of the face, which is the location of the integration point. To find this centroid, we use the **area-weighted average** of the centroid for **each sub-divided** triangular faces, defined above. The centroid of a triangle is defined by the average of its corner positions:

$$\mathbf{x}_{c,i} = \frac{1}{3} (\mathbf{x}_i + \mathbf{x}_{i+1} + \mathbf{x}_{i+2})$$

with $i = 0, 1, \dots, N_c - 2$. It follows that the integration point is:

$$\mathbf{x}_{ip} = \frac{1}{A_{ip}} \sum_{i=0}^{N_c-2} A_i \mathbf{x}_{c,i}$$

For the normal vector from the face, we can calculate similarly to the face area. This is because the cross product define vectors normal to each triangular sub-face. The unit normal vector can be obtained by:

$$\mathbf{n}_i = \frac{(\mathbf{x}_{i+1} - \mathbf{x}_0) \times (\mathbf{x}_{i+2} - \mathbf{x}_0)}{\|(\mathbf{x}_{i+1} - \mathbf{x}_0) \times (\mathbf{x}_{i+2} - \mathbf{x}_0)\|}$$

The normal vector at the integration point is calculated using the area-weighted average:

$$\mathbf{n}_{ip} = \frac{1}{A_{ip}} \sum_{i=0}^{N_c-2} A_i \mathbf{n}_i$$

Note: This is assuming the faces are nearly planar. If they are warped, we need to repeat this process (calculate \mathbf{n}_i then calculate \mathbf{n}_{ip}) for different choice of x_0 until we get the best possible estimate of the normal vector.

Next, recall the volume of the cell is calculated as:

$$V_P = \int_V dV$$

We want to relate this integral to the face geometry. Using the following relation:

$$\nabla \cdot \mathbf{x} = \frac{\partial x}{\partial x} + \frac{\partial y}{\partial y} + \frac{\partial z}{\partial z} = 3$$

so the volume integral can be re-written as:

$$V_P = \frac{1}{3} \int_V \nabla \cdot \mathbf{x} dV$$

By doing it this way, we introduce a divergence operator into the volume integral. This can then be transformed into a surface integral by Gauss' theorem:

$$V_P = \frac{1}{3} \int_S \mathbf{x} \cdot \mathbf{n} dS$$

or as a discrete summation over all of the integration points:

$$V_P = \frac{1}{3} \sum_{ip=0}^{N_{ip}-1} \mathbf{x}_{ip,i} \cdot \mathbf{n}_{ip,i} A_{ip,i}$$

Recall the centroid of a volume P is given as:

$$\mathbf{x}_P = \frac{1}{V_P} \int_V \mathbf{x} dV$$

Similar to before, we like to express quantity under integral to involve some divergence operator, so that we can use Gauss' theorem to transform it into an area integral. This time, we rely on this trick:

$$\nabla \cdot (\mathbf{x}\mathbf{x}) = \mathbf{x}\nabla \cdot \mathbf{x} + \mathbf{x} \cdot \nabla \mathbf{x}$$

The first term on the RHS involves $\nabla \cdot \mathbf{x}$ which we already showed to equal to 3. So, we can write:

$$\nabla \cdot (\mathbf{x}\mathbf{x}) = 3\mathbf{x} + \mathbf{x} \cdot \nabla \mathbf{x}$$

The second term on the RHS can be expanded as follow:

$$\begin{aligned} \mathbf{x} \cdot \nabla \mathbf{x} &= \left(x \frac{\partial}{\partial x} + y \frac{\partial}{\partial y} + z \frac{\partial}{\partial z} \right) \mathbf{x} \\ &= \left(x \frac{\partial \mathbf{x}}{\partial x} + y \frac{\partial \mathbf{x}}{\partial y} + z \frac{\partial \mathbf{x}}{\partial z} \right) \\ &= x \begin{pmatrix} \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} + y \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} + z \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \end{pmatrix} = \begin{bmatrix} x \\ y \\ z \end{bmatrix} \\ &= \mathbf{x} \end{aligned}$$

As a result:

$$\nabla \cdot (\mathbf{x}\mathbf{x}) = 4\mathbf{x}$$

Now, we can rewrite the expression for the cell centroid:

$$\mathbf{x}_P = \frac{1}{4V_P} \int_V \nabla \cdot (\mathbf{x}\mathbf{x}) dV = \frac{1}{4V_P} \int_S (\mathbf{x}\mathbf{x}) \cdot \mathbf{n} dS$$

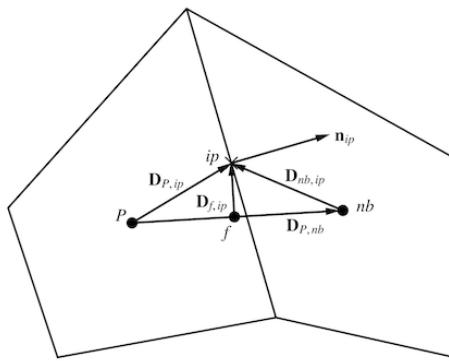
Same as before, we express as a discrete summation over the integration points:

$$\mathbf{x}_P = \frac{1}{4V_P} \sum_{ip=0}^{N_{ip}-1} \mathbf{x}_{ip,i} \mathbf{x}_{ip,i} \cdot \mathbf{n}_{ip,i} A_{ip,i}$$

Thus, we can define all the required face and cell geometry for unstructured grid calculations.

6.3.2 Interpolation

To do interpolations, we define the following variables for a particular control volume face (diagram):



Label	Description
P	control volume under consideration
<i>nb</i>	Neighboring control volume sharing the face containing <i>ip</i>
<i>ip</i>	Integration point location (face centroid)
<i>f</i>	Point along the vector connecting <i>P</i> to <i>nb</i>
$\mathbf{D}_{P,nb}$	Displacement vector from <i>P</i> to <i>nb</i>
$\mathbf{D}_{f,ip}$	Displacement vector from <i>f</i> to <i>ip</i>

Note: *f* could be anywhere along $\mathbf{D}_{P,nb}$ but best to make it so $\mathbf{D}_{P,nb}$ perpendicular to $\mathbf{D}_{f,ip}$, i.e. $\mathbf{D}_{f,ip} \cdot \mathbf{D}_{P,nb} = 0$. This minimizes $\mathbf{D}_{f,ip}$ which will also minimize the gradient correction term. In addition, if the grid is orthogonal, then this makes sure that $\mathbf{D}_{f,ip}$ is exactly zero.

We define the quantity f_{ip} to represent *f* as a function of $\mathbf{D}_{P,nb}$:

$$\mathbf{x}_f = \mathbf{x}_P + f_{ip} \mathbf{D}_{P,nb}$$

A general second order interpolation of a value ϕ to the integration point can be formulated as:

$$\phi_{ip} = (1 - f_{ip})\phi_P + f_{ip}\phi_{nb} + \mathbf{D}_{f,ip} \cdot \left[(1 - f_{ip})\nabla\phi \Big|_P + f_{ip}\nabla\phi \Big|_{nb} \right]$$

Note:

- 1st term on RHS is an **inverse distance interpolation** to the point *f*.
- 2nd term on RHS is **non-orthogonal correction** from *f* to *ip*. Here, *f* is estimated based on an inverse distance interpolation along $\mathbf{D}_{P,nb}$.

6.3.3 Gradient Reconstruction

We focus on the Gauss-based method because they are simple to explain. This method is based on the Gauss' theorem, which allows us to write:

$$\int_V \nabla\phi dV = \int_S \phi \mathbf{n} dS$$

Assume $\nabla\phi$ to be piecewise constant in each cell, we can rewrite the above equation for cell *P* as:

$$\nabla\phi \Big|_P V_P = \sum_{ip=0}^{N_{ip}-1} \phi_{ip} \mathbf{n}_{ip} A_{ip}$$

where we have replaced the surface integral with a discrete summation. Solving for the gradient:

$$\nabla \phi \Big|_P = \frac{1}{V_P} \sum_{ip=0}^{N_{ip}-1} \phi_{ip} \mathbf{n}_{ip} A_{ip}$$

Note: This interpolation of ϕ_{ip} depends on the gradient, making this a non linear system. A few solutions:

- ignore the gradient in the calculation of ϕ_{ip} , only use the inverse distance approximation based on ϕ_P and ϕ_{nb} . Not very accurate because gradients are at most 1st order accurate. ANSYS Fluent "Green Gauss Cell-Based" uses this. In fact, $f_{ip} = 0.5$ is used so values are not even accurate on orthogonal grid.
- Use the most recent value of ϕ_{ip} and solve the system iteratively until it converges. Need to update the gradients many time, so this method is computational expensive. Convergence is also quite poor, so it is not recommended.
- Startout not using the gradient corrections, then calculate the gradient in multiple stages. This method is efficient if the number of gradient updates is low. The accuracy is much better than ignoring the gradient correction altogether.
- Create a linear system which solves all of the gradients simultaneously. This is expensive, requires coding complexity and can be expensive.

6.3.4 Discretization Details

The transient and source terms are **volume-based terms**, so they can be treated just similarly to structured grids. Advection and diffusion terms are **surface-based terms**, so they need to be treated differently for unstructured grids.

1. Advection Terms The discretized advection term is given as:

$$\int_V \nabla \cdot (\mathbf{u}\phi) dV = \sum_{ip=0}^{N_{ip}-1} \mathbf{u}_{ip} \cdot \mathbf{n}_{ip} \phi_{ip} A_{ip}$$

Generally, \mathbf{u}_{ip} is replaced by the advecting velocity $\hat{\mathbf{u}}_{ip}$. A similar definition for $\hat{\mathbf{u}}_{ip}$ can be developed for unstructured grid. We use a 2nd order upwind interpolation:

$$\phi_{ip} = \phi_u + \nabla \phi \Big|_u \cdot \mathbf{D}_{ip,u}$$

where u represents the upwind cell, i.e.:

$$u = \begin{cases} P & \text{if } \dot{m}_{ip} \geq 0 \\ nb & \text{if } \dot{m}_{ip} < 0 \end{cases}$$

Note:

- linearization is carried out with respect to ϕ_u , ensuring the linearization coefficients have correct signs, similar to UDS.
- CDS and QUICK can also be adapted for unstructured grids.
- Sometimes a flux limiter is applied to ensure the integration point value is bounded by the the surrounding cell values. This is important in flow with discontinuities (shocks)

2. Diffusion Terms The discretization diffusion term is given as:

$$\int_V \nabla \cdot \mathbf{J}_\phi dV = \sum_{ip=0}^{N_{ip}-1} \mathbf{J}_{\phi,ip} \cdot \mathbf{n}_{ip} A_{ip}$$

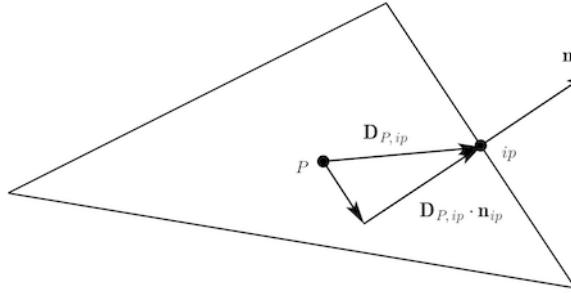
In Fourier's or Fick's law, the diffusive flux \mathbf{J} is proportional to the gradient of ϕ

$$\mathbf{J}_{\phi,ip} = -\Gamma_P \nabla \phi \Big|_{ip}$$

To calculate the gradient at the integration point, we need to enforce a continuous flux across all faces. Mathematically, this continuity of the diffusive flux is expressed as:

$$\Gamma_P \nabla \phi \Big|_{ip} \cdot \mathbf{n}_{ip} = \Gamma_{nb} \nabla \phi \Big|_{ip,nb} \cdot \mathbf{n}_{ip}$$

The derivatives normal to the integration point are calculated by extrapolating from the cell center to a point on a line that intersects ip and is normal to the control surface:



We then use a finite difference approximation to evaluate the normal derivative along this line:

$$\nabla\phi|_{ip,P} \cdot \mathbf{n}_{ip} = \frac{\phi_{ip} - [\phi_P + \nabla\phi|_P \cdot (\mathbf{D}_{P,ip} - (\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip})\mathbf{n}_{ip})]}{\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}}$$

We form a similar expression for control volume nb , then equating the two through flux balance expression. We then get the following expression for the integration point value, satisfying the heat flux from both sides of the control surface:

$$\begin{aligned} \phi_{ip} &= \frac{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip})}{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})}\phi_{nb} - \frac{\Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})}{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})}\phi_P \\ &\quad + \frac{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip})(\mathbf{D}_{nb,ip} - (\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})\mathbf{n}_{ip})}{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})} \cdot \nabla\phi|_{nb} \\ &\quad - \frac{\Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})(\mathbf{D}_{P,ip} - (\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip})\mathbf{n}_{ip})}{\Gamma_{nb}(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \Gamma_P(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})} \cdot \nabla\phi|_P \end{aligned}$$

We then substitute this ϕ_{ip} into the equation for the normal derivative to get an expression in terms of the cell-centered value.

$$\begin{aligned} \nabla\phi|_{ip,P} \cdot \mathbf{n}_{ip} &= \frac{\phi_{nb} - \phi_P}{(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \frac{\Gamma_P}{\Gamma_{nb}}(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})} + \frac{(\mathbf{D}_{nb,ip} - (\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})\mathbf{n}_{ip})}{(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \frac{\Gamma_P}{\Gamma_{nb}}(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})} \cdot \nabla\phi|_{nb} \\ &\quad - \frac{(\mathbf{D}_{P,ip} - (\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip})\mathbf{n}_{ip})}{(\mathbf{D}_{P,ip} \cdot \mathbf{n}_{ip}) - \frac{\Gamma_P}{\Gamma_{nb}}(\mathbf{D}_{nb,ip} \cdot \mathbf{n}_{ip})} \cdot \nabla\phi|_P \end{aligned}$$

This expression can then be used to calculate the diffusive flux from the P cell, with a similar expression being used for nb , which ensures the flux is consistent. The two gradient terms in the equation above account for non-orthogonality in the grid. In the limit of a completely orthogonal grid the equation above reduces to the harmonic mean formulation of Patankar. In the limit of an orthogonal grid with constant diffusivity, Γ , this expression is identical to that used in our one-dimensional codes.