

# “Implementation of the Pushover Analysis in MatLab for 3D frames”

Luis Fernando Verduzco Martínez [1]

[1] luiz.verduz@gmail.com

Faculty of Engineering, Autonomous University of Quretaro, Santiago de Querétaro, Mexico

## Overview

This software is the implementation of the Pushover analysis for structural 3D frames, fully developed in MatLab as a function. Such function is based entirely on flexure hinges formations at the ends of an element with respect to their local y axis. A uniform seismic load incrementation is carried out on each step until a collapse or stiffness degradation criteria is reached. Such function has proved to have a great potential for its implementation in teaching the Pushover method in structural mechanics and applied sciences through the simulation of collapse mechanisms. Not only  $P - \Delta$  collapse graphics are obtained, but also the evolution of the collapse mechanism deformation of the structure and the seismic Collapse Safety Factors (CSF), including as well the evaluation of some basic Damage Indices (DI) such as the Inter-story Drift DI, the Inter-story Plastic Drift DI and the Deformation Based DI.

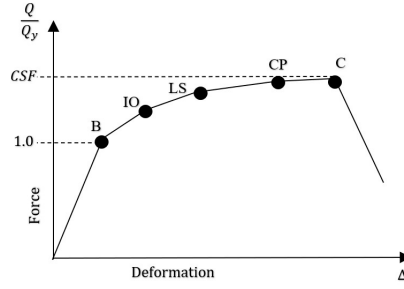
**Keywords:** Pushover, 3D Frames, Teaching, Pushover, Collapse Mechanisms , Damage Indices

## 1 The Pushover analysis

The *Pushover* analysis is used in structural engineering for many tasks when is required to evaluate the response of a structure in the plastic range behaviour, for instance, when evaluating performance of structures and estimating potential Damage States under certain load conditions. It consists of analysing a structure by incremental lateral loads imposed, defining step by step plastic-formations on the element's cross-section. The analysis is terminated when the collapse mechanism is reached, either by stating a certain state of damage for the structure or a degree of stiffness degradation. This way, seismic Collapse Safety Factors can be found **Fig. 1**. A condition of deformation based damage can be imposed as (1) to stop the analysis process, where  $d_u$  is the last or max floor deformation in the plastic range,  $d_y$  is the max elastic floor deformation and  $deg$  is the max allowed damage by floor deflection (for which values between 0.1 – 0.8 are recommended).

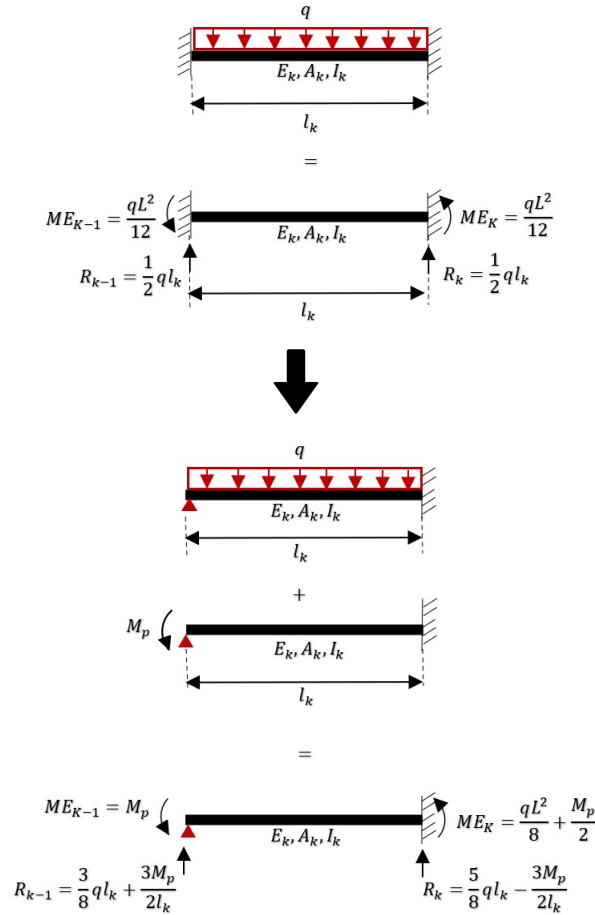
$$\frac{d_u - d_y}{d_y} > deg \quad (1)$$

During the analysis process, when a plastic formation is detected over one or various elements' cross-sections, then such elements are transformed to an equivalent structure in which such plastic formations are considered as articulations. Such articulations are simulated through the liberation of the DOF in such plastic formation location and its corresponding proper distribution of the loads and bending moments to the element's ends according to the new generated equivalent structural system due to such DOF liberation **Fig. 2**. This way, when both ends of an element have suffered plastic formations its equivalent structural system will correspond to the one shown in **Fig. 3**. The external equivalent plastic bending moments at each plastic formation location will remain constant through out the subsequent analysis



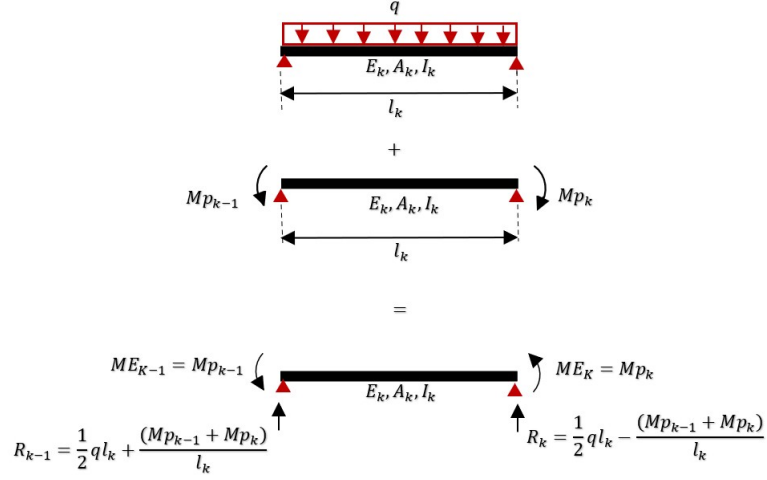
**Figure 1.** Force-Deformation diagram representing the damage states (IO, LS, CP) through the collapse mechanism (C) of ductile structural frame. IO stands for Immediate Occupancy, LS for Life Safety, CP for Collapse Prevention according to [FEMA 273, 1997].

steps such that the structure's stiffness and force-displacement capacity will be eventually degraded.



**Figure 2.** Equivalent structural system transformation when plastic formations at the ends of an element take place.

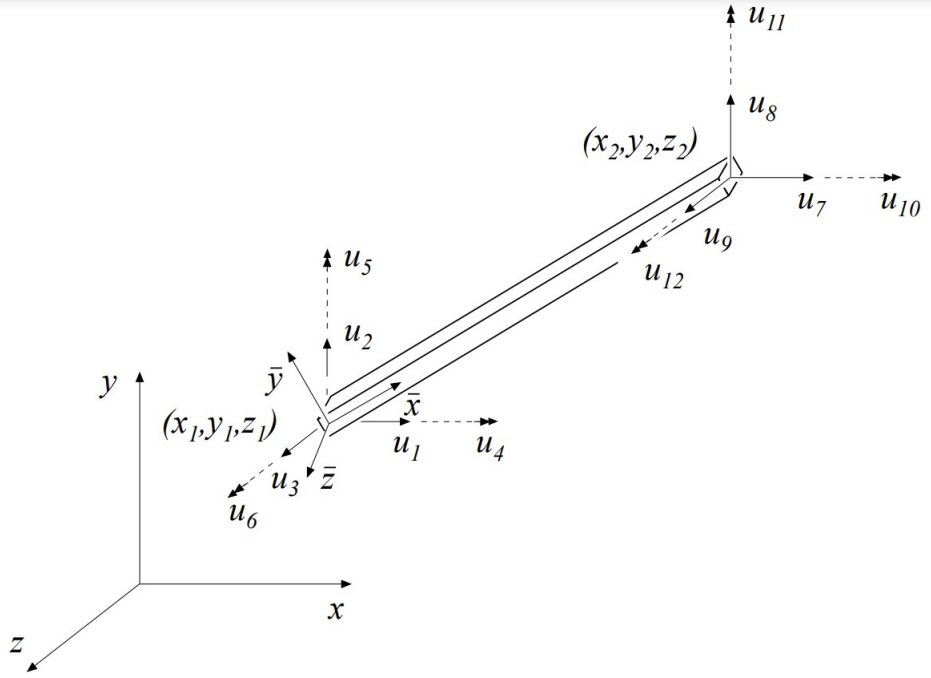
When dealing with 3D frame structures, however, more degrees of freedom than just 3 are involved,



**Figure 3.** Equivalent structural system transformation when plastic formations at each of an element's ends takes place.

therefore, it is important to establish properly the local axes of each element. The stiffness matrix of a 3D structural element that has not yet suffered any plastic formation at any of its ends is represented as following, where  $k_1 = EA/L$  and  $k_2 = GK_v/L$ , and in which the element's local axis  $(\bar{x}, \bar{y}, \bar{z})$  are depicted in **Fig. 4**:

$$\bar{\mathbf{K}}^e = \begin{bmatrix} k_1 & 0 & 0 & 0 & 0 & 0 & -k_1 & 0 & 0 & 0 & 0 & 0 \\ 0 & \frac{12EI_{\bar{z}}}{L^3} & 0 & 0 & 0 & \frac{6EI_{\bar{z}}}{L^2} & 0 & -\frac{12EI_{\bar{z}}}{L^3} & 0 & 0 & 0 & \frac{6EI_{\bar{z}}}{L^2} \\ 0 & 0 & \frac{12EI_{\bar{y}}}{L^3} & 0 & -\frac{6EI_{\bar{y}}}{L^2} & 0 & 0 & 0 & -\frac{12EI_{\bar{y}}}{L^3} & 0 & -\frac{6EI_{\bar{y}}}{L^2} & 0 \\ 0 & 0 & 0 & k_2 & 0 & 0 & 0 & 0 & 0 & -k_2 & 0 & 0 \\ 0 & 0 & -\frac{6EI_{\bar{y}}}{L^2} & 0 & \frac{4EI_{\bar{y}}}{L} & 0 & 0 & 0 & \frac{6EI_{\bar{y}}}{L^2} & 0 & \frac{2EI_{\bar{y}}}{L} & 0 \\ 0 & \frac{6EI_{\bar{z}}}{L^2} & 0 & 0 & 0 & \frac{4EI_{\bar{z}}}{L} & 0 & -\frac{6EI_{\bar{z}}}{L^2} & 0 & 0 & 0 & \frac{2EI_{\bar{z}}}{L} \\ -k_1 & 0 & 0 & 0 & 0 & 0 & k_1 & 0 & 0 & 0 & 0 & 0 \\ 0 & -\frac{12EI_{\bar{z}}}{L^3} & 0 & 0 & 0 & -\frac{6EI_{\bar{z}}}{L^2} & 0 & \frac{12EI_{\bar{z}}}{L^3} & 0 & 0 & 0 & -\frac{6EI_{\bar{z}}}{L^2} \\ 0 & 0 & -\frac{12EI_{\bar{y}}}{L^3} & 0 & \frac{6EI_{\bar{y}}}{L^2} & 0 & 0 & 0 & \frac{12EI_{\bar{y}}}{L^3} & 0 & \frac{6EI_{\bar{y}}}{L^2} & 0 \\ 0 & 0 & 0 & -k_2 & 0 & 0 & 0 & 0 & 0 & k_2 & 0 & 0 \\ 0 & 0 & -\frac{6EI_{\bar{y}}}{L^2} & 0 & \frac{2EI_{\bar{y}}}{L} & 0 & 0 & 0 & \frac{6EI_{\bar{y}}}{L^2} & 0 & \frac{4EI_{\bar{y}}}{L} & 0 \\ 0 & \frac{6EI_{\bar{z}}}{L^2} & 0 & 0 & 0 & \frac{2EI_{\bar{z}}}{L} & 0 & -\frac{6EI_{\bar{z}}}{L^2} & 0 & 0 & 0 & \frac{4EI_{\bar{z}}}{L} \end{bmatrix}$$



**Figure 4.** Global and local system of reference considered in this project for the beam elements. The negative direction of the global Z axis corresponds to the direction of gravity.

## 1.1 Damage Assessment

In order to evaluate peak dynamic deformation demand of structures at various performance levels, non-linear static analysis are required, so that a good estimation of damage may be determined for the structure life-cycle. The assessment of damage performance in structures is of great importance in the design stage of a structure, since it can be related with potential economical losses (cost of repairing, rehabilitation, etc.).

One of the most effective tools to estimate damage performance is through *Damage Indices (DI)*. A vast number of Damage Indices have been proposed and developed based on structural properties (response) or dynamical properties of the structure. Such indices may predict the level of degradation state of a structure and therefore its vulnerability. For this function, the following three non-cumulative DI were considered: *the Inter-story Drift DI* (2), *the Plastic Inter-story Drift DI* (3) and *the Deformation Based DI* (4). Such DI can reflect quite well the state of a structure at his last stage of collapse and are among the most detailed DI used and of quite simplicity for application [Makhloof et al., 2021].

$$DI_{drift} = \frac{\Delta_{max}}{H} \quad (2)$$

$$DI_{p-drift} = \frac{\Delta_{max} - \Delta_y}{H} \quad (3)$$

$$DI_{\mu} = \frac{\Delta_{max} - \Delta_y}{\Delta_y} \quad (4)$$

Where  $H$  is the floor height,  $\Delta_y$  is the yielding lateral deformation of the floor,  $\Delta_{max}$  represents the maximum lateral deformation of the floor,  $\Delta_u$  denotes the ultimate displacement at failure. Classification of the damage state for the first two inter-story drift DI can be made according to [FEMA-356, 2000], as **Table 1** and for Deformation Based DI a classification can be made according to Park and Ang as **Table 2**.

Performance Level	Damage State	DI %
Immediate Occupancy (IO)	No damage	< 0.2
Damage Control (DC)	Minor Damage	< 0.5
Life Safety (LS)	Moderate Damage	< 1.5
Collapse Prevention (CP)	Severe Damage	< 2.5
Collapse	Collapse	> 2.5

**Table 1.** Interpretation of Inter-Story drift DI.

Damage State	State of the structure	DI
Minor Damage	Serviceable	0 – 0.2
Moderate Damage	Repairable	0.2 – 0.5
Severe Damage	Irreparable	0.5 – 1.0
Collapse Damage	Total loss	> 1.0

**Table 2.** Interpretation of Deformation Based DI.

## 1.2 Teaching the Pushover analysis

Teaching the Pushover analysis in Postgraduate program courses of engineering is still a challenging task by many Professors. The inherent dualism of the method similar to the Finite Element method requires a balance between the mathematical theory and physical understanding, and can only be applied through a computer. Even though many textbooks related with the subject contain didactic exercises treating theoretical issues and simple sample problems, they do not contain full sections that may guide the student on how to compute the method by oneself, thus lacking the motivating factor for students to fully appreciate the intimate relationship between the theoretical issues of the method, its physical interpretation and its computer implementation.

On the other hand, there is a tendency by educational practitioners in a course related with this such method to leave the students to operate the method by themselves in any way possible, not really paying attention on the programming skills of each individual. Therefore, those students who lack the good programming knowledge base requirement tend to get frustrated by the so many operations and code required, putting in risk their own learning capabilities.

It has been recommended by some authors [Matti, et al., 2000] that in order to make a computed aided program pedagogical effective such program should be written in such a manner that all the usual subroutines related to the method in question are present; the student should be able to assemble all of such functions and operations so that a calculation is possible, without requiring a large amount of programming effort or computational skills. This way, a much more time-efficient learning environment is enhanced, requiring only for each student to build their own code for every problem to solve, instead of coding and programming the whole computational process with the risk of losing track in the run. Thus, the physical interpretation along with the solution strategy of any problem are continuously connected to the mathematical and numerical treatment of such problem at every moment.

When learning the Pushover analysis method or any other engineering process it is required to review whether or not the calculations are correct, this task is usually observable by the collapse mechanism. It is thus advantageous for students to be able to visualize the evolution of the deformed structure as it gets closer to the collapse mechanism. Even though software like Microsoft Excel could also provide good mean for the computation of the Pushover method, such package is limited in graphic tools, as they lack of dynamism compared to MatLab graphic functions for instance.

## 2 Algorithmic process

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**Algorithm 2.1:** Pseudo-code for the Pushover analysis method.

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1. Initialize load incremental factor as  $\lambda = 1.0$
  2. Apply incremental load factor to the lateral loads to perform a static linear analysis:
    - Check for support conditions at the ends of members and build the corresponding stiffness matrix for each member for their assembly globally
  3. Compute mechanic elements of bars and check if any of the members' ends have overpassed their plasticity flexure limit  $M \leq M_p$ 
    - If any end of a member has overpassed such plasticity limit, then change its supports conditions (substitute "*Fixed*" condition for "*Art*") and recollect displacements and forces for each DOF. Then apply increment  $di$  in the  $\lambda$  factor as  $\lambda = \lambda + di$  (recommendation  $di = 0.01$ )
    - If no member has suffered overpassed such plasticity limit then apply increment  $di$  in the  $\lambda$  factor as  $\lambda = \lambda + di$  (recommendation  $di = 0.01$ )
  4. Check the terminal criteria of equation (1). If such terminal condition is complied stop the iteration, otherwise go back to step 2
  5. Compute Damage Indices
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