		Buoyant Drone	e Weight Allocation					
Subsections	Part	Number of units	Mass per unit est (g)	Weight Contribution est (g)	Subsection Weight est (g)	Mass per unit (q)	Weight Contribution (g)	Subsection Weight (g)/ percent of estimate
Electronics	Pic32		1 7			1	1	-6
	Raspberry Pi Compute Module 3+		1 9			9	9	
	SODIMM		1			3	3	3
	RC receiver		1 13	13		16	16	3
	IMU sensor		2.5	2.5		1	1	-1.5
	Camera (camera + breakout +			00.7		0.4		00.7
	antenna)		90.7 4 9	90.7		24		
	Ultrasonic sensor Barometeric sensor		1 1.2			8	1	-0.2
	GPS sensor		1 8.5			4	<u> </u>	
	GPS antenna		1	0.0		95		
	Motor		4 30	120		44		
	ESC		1 25			24		
	Battery 1		1 640			637		-3
	Servo	4	4 55			61		24
	5V step down voltage regulator		1			3		3
	Radio telemetry		1			14	14	
	Radio telemetry antenna		1			10	10	
	PCB		1			30	30	30
	Total				1160.9		1324	163.1
Drone Frame	Servo connectors	4	4 64.76	259.04	SLDW	31	124	-135.04
	Ultasonic mount		1 141.68	141.68	SLDW	53	53	-88.68
	Motor mount block	4	13.62	54.48	SLDW	6	24	
	feet	4	1.65			1.2		
	Gondola				SLDW	261		-133.06
	Propellers	4	9.7			10		
	Aluminum sheet		32.3		SLDW	45.7		13.4
	Gondola Plate				SLDW	50		
	Servo plate		9.2		SLDW	10		
	Ultasonic plate		50.29	50.29	SLDW	25		
D. I. d	Total		1000	4000	1066.7		667.5	
Payload	MagAero		1 1000	1000			1000	
laining alamanta	Corbon Stool Deboft/tomp)		4 36.5	146	1000 SLDW			
Joining elements	Carbon Steel Dshaft(temp) Servo Shaft Coupler			12		37		
	Nuts	24			SLDW	3.1		
	Washer	24			SLDW	0.7		
	Screews (3D parts)	28			SLDW	5.8		-0.56
	Ultrasonic SCreews	11			SLDW	0.14		
	Standoff screws					0.5		
	Standoffs	4				0.5		
	Motor mount				SLDW	3		
	Motor shaft propeller attachment		1			5	20	
	Total				438.072		451.28	13.208
Lift bag	Bag itself		350	350		315	315	-35
	Envelope (est)		500	500		696	696	196
	Lift force		-4050	-4050			-4050	0
				C				
	Total				-3200		-3039	
								Change from estimate
				est Effective "weight" g	465.672		403.78	-61.892
				est Effective weight N	4.5635856	actual Effective weight N	3.957044	

if you add items to th	e weight chart add above the bottom	row of each section, so o	calculations update				
				Mass w/o Lift	4.515672kg		
				Weight w/o Lift	44.2535856N		

Version 5.2						State Based Cur				Time in State(hrs	-1	1						
Name	Description	Quantity	Nominal Voltage	Input V Regulation(VIH)	Output V Variation(VOH)			Low Expected	I ow Measured		Low Low	Total Consumption/Cycle(mA*hrs)	Total Energy/Cycle (mW*hrs)	Power%	Datasheet link	Comments	Versioning	Power
LiPo Battery Charger	Charger	1	11.1	100-240V AC (From outlet)		.1-10A adjustable				19		, and the same of	100000 (when using AC input)			With current battery, should take 61 minutes to fully charge from 83% depleted while auto-balancing cel 100W charging power max	,	Microcontroller.
LIPO 11.1V 3S 11000mAh	Battery	1	11.1		9.6-12.6V							-9166.66	101750		https://www.max	83% of 11000mAh discharge for max current 40C discharge rating, 440A max discharge current 5C charge rating, 48.5A max charge current 3.2 per cell cutoff voltage(9.6V total) 4.2 per cell max voltage(12.6V total)	-Verified IMU, Altimeter, RC receiv GPS, and Microcontroler power	
uC32	Microcontroller	1	5	(VDD) 2.3-3.6V (Cannot go lower than 1.75V unless it will lose RAM data) (I/O) 2.64-3.6V VIH 066V VIL	2.4 - 3.6V VOH 0-4V VOL	75.5	60	0	0	1	0	60	300	0.299	https://drive.goo	60mA current tested running servo code	- Added column for tested and verified current for each part	Actuators
Raspberry Pi 3B+	Microprocessor (VBAT)	1	5	2.5-5.25V	N/A	1200	500	0	0	1	0	500	2500	2.46%	https://www.rasp	Using 5V 1.2A maximum current draw specification datasheet	s	
AKK KC03	Camera/Transm	1	11.1	7-20V	Supplies 5V Vout for Camera	340	312	0	0	1	0	312	3463.2			Supply current for transmitter too		
FS-iA6B Receive	RC Receiver	1	5	4-6.5V	N/A	20	34	0	0	1	0	34	170	0.17%		Tested receiving 30mA current constant for receiving	ng	
Serial Telemetry Transmitter	Data Transmitter	1	5	5V	N/A	100	100	0	0	1	0	100	500	0.49%	https://www.spar	100mA current needed for transmitting at 20dBm		
				(VDD)1.95-3.6V (VDDIO)1.62-3.6V														
MPL3115A2	Pressure/Temp erature Sensor	1	3.3	(I/O) 2.475-3.3V VIH 099V VIL	2.97-3.3V VOH 033V VOL	0.2	0.16	0	0	1	0	0.16	0.528	0.00%	https://drive.goog	Typical current needed during Acquisition/Conversion data in high resolution mode Current needed for each 4 sensors running for who	0	
HC-SR04	Ultrasonic	4	5	(VDD) 5V (Trigger)2-5V	2-5V	15	2	0	0	1	0	8	40.844	0.04%	https://drive.goo	cycle, added power for PWM usage by sensors to r data at 16hz (.844mW)	re	
MTK 3339	GPS Module	1	3.3	(VDD)3-4.3V (VDDBackup)2-4.3V (I/O) 2-3.3V VIH 08V VIL	2.4-2.8V VOH 0-4V VOL	25	25	20	20	0.01	0.99	20.05	66.1759	0.07%	https://drive.goog	25mA(Tested) Acquisition of GPS Signal takes 30s 20mA supply current for Tracking, adds 1.09mW for continuous UART power usage	r	
ICM-20948	IMU 9DoF IC		3.3	(VDD)1.71-3.6V (VDDIO) 1.71-1.95V (I/O) 1.35-2.3V VIH 054V VIL	1.62-1.8V VOH 018V VOL	3		0	0				9.9					
MPRLS0001	IMU 9D0F IC	1	3.3	(VDD)1.8-3.6V (I/O) 066V VIL	018V VOL	3	3	U	0	1	0	3	9.9	0.01%	https://drive.goog	Typical current during data acquisition in 9-axis more	a .	
PG00001C	Pressure sensor	1	3.3	2.64-3.3V VIH	2.64-3.3V VOH	4	4	0	0	1	0	4	13.2	0.01%	https://sensing.h	Tested to receive 3mA max while on		
2600Kv	Motor	4	11.1	6.4-12.6V	N/A	4086	4086	1787	1787	1	0	16344	181418.4	178.30%	B08KRRWM7F?	Depends on how much time drone is rising, 2x how at full throttle, .125g of thrust per motor to hover Also took into account 85% motor efficiency(Brushli See motor Power section	ie	
RC Sail Winch Servo																Depends on how much time drone is turning, Current ranges from 3-350mA, depending on how r the drone is turning	m	
	Servo	4	5 11.1	4.8-6V	N/A	350 Total Difference	350 742.54	3	1	0	1	12 17397.21	60 188542.2479	0.06%	https://www.ebay			
			Max Voltage			Total Dillerence	742.04					Total mAh	Total mWh					
Power Rails			Average		Max at Dissipation(r Due to Regulat	nV Max						Insert Flight Time						
	Items	Voltage(V) 11.1 Nominal)ropout Voltage(V	Max Current(mA) Max Regu		nt) Dissipation(mW)	Heat comes from	Comments 7.5ft wires to				Here(Hours)	Final Battery Co	alculation	Results			
Rail 1	3	(Ranges from 12.6-9.6V)	0	12906	N/A 661	661	and from each n	otor 1.588 ohms IAWG at 4.342A				1	Total Energy Ne	eeded(mWh)	189,9	145.53		
Rail 2	1	5	6.1	1400	5000 11	1235	5V switching reg 85% minimum e	ulator for servos, ficiency				Insert Percentag of time at high throttle here(0-1)		scharge limit(mWh) 101	750		
Rail 3	5	3.3	1.7	41	800 70	70	3.3V Regulator u microcontroller					1	Battery Condition	on	Out of	Battery		
Rail 6	5	5	6.1	750	5000 662	662	Same 5V Switch used for every o 85% efficient	ing Regulator her 5V part				Insert Turning time percentage here(0-1)	Power Used by	Motors(mWh)	181	,418 95.51%		
					1403							0		Servos(mWh)	(0.03%		
					Maximum Tota lost to Heat(m)								Power Used by Else(mWh)		70	3.72%		
													Power Lost to F Else(mWh)	reat	1,0	403 0.74%		

													Power/Passive
lame	Description	Designator	Quantity	Manufacturer	Supplier	Total mass[g]	Total Price [\$]	CAD file	EagleCAD library	Shop link	Ordered	Recieved	Components
Po Battery Charger	Charger	CHA BAT1	1	Hitec MaxAmps	MaxAmps MaxAmps	0 640	\$119.99 \$249.99		H	https://www.maxamps.com/lipo-battery-charger-hitec-x2-ac-plus-black-edition-ac-	✓	✓	
1V 3S 11000mAh LiPo battery T 10474	XT60 connectors male/female pair	BAIT	1	MaxAmps	MaxAmps	640	\$249.99	<u>~</u>		https://www.maxamps.com/lipo-11000-3s-11-1v-battery-pack	<u> </u>	~	Otll
(1 104/4	X160 connectors male/female pair								$\overline{\mathbf{v}}$		\checkmark	\checkmark	Controllers, Processors
			1		Digikey		\$1.50		_	https://www.digikey.com/en/products/detail/PRT-10474/1568-1816-ND/8258064?			and Oscillators
	680 ohm resistor SMD 0805		1		Distress		\$0.10			https://www.digikey.com/en/products/detail/AC0805JR-07680RL/YAG3800CT-ND		П	Actuators and Receivers/
00805JR-07680RL	680 Onm resistor SMD 0805		'		Digikey		\$0.10		_	https://www.digikey.com/en/products/detail/AC0805JR-07680RL/TAG3800CT-ND	<u>~</u>	L	Transmitters
RA-6APB471V	470 ohm resistor SMD 0805		1		Digikey		\$0.70		~	https://www.digikey.com/en/products/detail/panasonic-electronic-components/ER	▽		
00805FR-0710KL	10K ohm resistor SMD 0805		9		Digikey		\$0.42	n	\overline{v}	ey.com/en/products/detail/RC0805FR-0710KL/311-10.0KCRCT-ND/730482?item		n	Sensors
0805X103K5RAC7210	0.01uF capacitor SMD 0805		1		Digikey		\$0.35		\overline{v}	products/detail/C0805X103K5RAC7210/399-C0805X103K5RAC7210CT-ND/1356			
053C104KAB2A	0.1uF capacitor SMD 0805		12		Digikey		\$5.32		~	n/en/products/detail/LD053C104KAB2A/478-LD053C104KAB2ACT-ND/12153435			
													External
RM21BC81C106ME15L	40Fit CMD0005			Murata Electronic	Distress		\$2.00		\checkmark	https://www.disilyeverset.com/com/com/com/com/com/com/com/com/com/	~		Hadrware (New Parent)
N1117DT18CTR	10uF capacitor SMD0805		1	STMicroelectroni	Digikey		\$2.00			https://www.digikey.com/en/products/detail/GRM21BC81C106ME15L/490-10499-			(Not Powered) Plug Accessories
1117DT33CTR	1.8V Voltage Regulator		1	STMicroelectroni	Digikey Digikey		\$0.60	H		https://www.digikey.com/en/products/detail/stmicroelectronics/LD1117DT18CTR/1		H	Plug Accessories
	3.3V Voltage Regulator		1	STMicroelectroni	- ,		\$0.51	H	<u> </u>	https://www.digikey.com/en/products/detail/LD1117DT33CTR/497-1235-1-ND/586		H	
1117DT50CTR	5.0V Voltage Regulator Linear				Digikey		\$0.54 \$16.95	H	<u> </u>	https://www.digikey.com/en/products/detail/LD1117DT50TR/497-1238-1-ND/5862	~		
4V50F5	5.0V Voltage Regulator Switching		1	Pololu	Pololu	3	*	Ш		https://www.pololu.com/product/2851	<u> </u>		
32MX340F512H-80V	Microcontroller	M1	1	Microchip Technology	Digikey	2.3	\$6.77		\checkmark	https://www.digikey.com/en/products/detail/PIC32MX340F512H-80V%2fPT/PIC32	\checkmark		
232RO-REFI	USB to UART ETDLIC		1	leamology	Newark		\$4.95			https://www.newark.com/webapp/wcs/stores/servlet/EnhancedCheckoutConfirma	V		
32RQ-REEL	USB to UART FTDI IC		1		Digikey		\$4.60	H		https://www.digikey.com/en/products/detail/FT232RQ-REEL/768-1008-1-ND/1836		H	
and there	Raspberry Pi Compute Module M3+				gc,		\$ 7.00						
	8GB	M2	1	Raspberry Pi	Digikey	9	\$30.00	~	$\overline{\mathbf{v}}$	https://www.digikey.com/en/products/detail/CM3%2b%2f8GB/1690-1030-ND/986	~		
M2837B0	(Microprocessor)												
07S016JA1R1500	USB-C header	J2	2	JAE Electronics	Mouser		\$3.02	✓	\checkmark	https://www.mouser.com/ProductDetail/JAE-Electronics/DX07S016JA1R1500?qs	~		
roSD Socket	SD card socket for Raspberry Pi		1		Adafruit		\$1.95		✓	https://www.adafruit.com/product/1660	✓		
disk 16GB Ultra MicroSD Card	MicroSD Card for Raspberry Pi	MSD	1	SanDisk	Amazon	9.92	\$6.18			https://www.amazon.com/dp/B073K14CVB/	✓	\checkmark	
33854-052FSLF	DDR2 SODIM slot (Raspberry Pi Cor	npute Module)	1	Amphenol ICC	Digikey		\$6.40	\checkmark	ightharpoonup	https://www.digikey.com/en/products/detail/amphenol-icc-fci/10033854-052FSLF/	✓		
BLV-3C-8M-000000 80MHz crystal	o 80MHz crystal oscillator		1		Digikey		\$1.35		\checkmark	https://www.digikey.com/en/products/detail/cts-frequency-controls/CB3LV-3C-8M	~		
DK-32.768kHz-LRT	32.768kHz crystal oscillator (smaller	ootprint)	1		Digikey		\$2.73		✓	https://www.digikey.com/en/products/detail/abracon-llc/ASDK1-32.768KHZ-LRT/2	✓	П	
EK-32.768kHz-LRT	32.768kHz crystal oscillator (reference		1		Digikey		\$1.33	ħ		https://www.digikey.com/en/products/detail/abracon-llc/ASEK-32-768KHZ-LRT/20		n	
L3115A2	Pressure/Temperature Sensor IC	BAR	1	NXP USA	Mouser	1.9	\$5.46			https://www.mouser.com/ProductDetail/841-MPI 3115A2R1	<u> </u>		
K 3339	GPS Module	GPS	1	Mediatek	Adafruit	4	\$29.95			https://www.adafruit.com/product/790	<u> </u>	Ħ	
10000	C. C. Modale	0.0		Mediator	7 tadii at		Ψ <u>2</u> 0.00			Ingo://www.ddulan.com/product/co			
-LTE-RPC-UFL	GPS Antenna	GPSANT	1	Adafruit	ChangHong		\$14.95			https://www.adafruit.com/product/960	☑		
A to u.FL cable adapter	GPS u.FL to SMA adapter		1	Adafruit	Adafruit		\$3.95			https://www.digikey.com/en/products/detail/adafruit-industries-llc/851/6051775?s=	Z		
	Ultrasonic	SON	3	Sparkfun	Sparkfun	25.5	\$12.00		-	https://www.sparkfun.com/products/15569		П	
~5KU4	Oltrasonic	SUN	3	Sparkiun	Sparkiun	25.5	\$12.00			nitus://www.sparkiun.com/producis/15569	<u> </u>		
M-20948	IMU 9DoF IC	IMU	1	TDK InvenSense	Digikey	13	\$10.47			https://www.digikey.com/en/products/detail/tdk-invensense/ICM-20948/6623535	✓		
	Logic Level FET 1.8V to 3.3V vice												
0.00	versa for IMU & PIC32MX340F512H			l			***			10 1 10 1 1000 DOC100 O	$ lap{}$		
S138	I2C communication wire	MPR	5	ON Semiconduct	Mouser	1.1	\$2.00 \$6.44			https://www.mouser.com/ProductDetail/863-BSS138-G		_	
RLS0001PG00001C	Pressure sensor IC	MPK	1	Honeywell	Digikey	1.1				https://www.digikey.com/en/products/detail/honeywell-sensing-and-productivity-se	<u> </u>	<u> </u>	
sky FS-i6X 6-10(Default 6)CH 2.4GI			1	Flysky	Amazon		\$55.99			https://www.amazon.com/gp/product/B0744DPPL8/ref=as_li_tl?ie=UTF8&tag=rc-	<u> </u>		
ial Telemetry Radio Kit - 915MHz, 1			1		Sparksfun		\$39.99			https://www.sparkfun.com/products/15007	<u> </u>		
erstar RS20A BLheli_S 4-in-1 ESC		ESC	1	Racerstar	getfpv	25	\$28.00			https://www.getfpv.com/racerstar-rs20a-blheli-s-4-in-1-esc.html	<u> </u>	~	
2822-1275kv	Motor	MOT	4	Turnigy	Hobbyking	120	\$78.12	\checkmark		https://hobbyking.com/en_us/turnigy-aerodrive-sk3-2822-1275kv-brushless-outru	\checkmark	\checkmark	
K KC03	Camera and Transmitter	CAM+CAMTX	1	AKK	Amazon	67.76?	\$27.99			https://www.amazon.com/AKK-Degree-800TVL-Switchable-Transmitter/dp/B	✓	\checkmark	
/ Receiver 5.8G 150CH OTG	Camera Receiver	CAMRX	1	SKYDROID	Amazon	50	\$27.99	П		https://www.amazon.com/Receiver-Female-Plastic-Android-Monitor/dp/B07Q5MF	\overline{v}	\overline{v}	
ceiver										And a second sec			
Sail Winch Servo 25T	Servo, 10.6kg / 0.9sec	SRVO	4	Corona	Ebay	60	\$64.00			https://www.ebay.com/itm/254648162225?chn=ps&mkevt=1&mkcid=28	~	\checkmark	
ter Airscrew 9 x 4.5 Multi Rotor celler Set White	Propeller	PROP	2	Master Airscrew	Hobbyking	19.4	\$11.78			https://hobbyking.com/en_us/master-airscrew-9-x-4-5-multi-rotor-propeller-set-wh	\checkmark		
	<u>'</u>							 	 				1
oz HyperD	Fabric to be used for envolope	ENV	18		Rip stop by the rol	500	\$85.00		12	https://ripstopbytheroll.com/collections/ultralight-nylon-fabric/products/1-0-oz-hyp		✓	-
Itipurpose 6061 Aluminum Sheet	Aluminum sheet	Al	1	McMaster-Carr	McMaster-Carr	32	\$4.68	<u> </u>			<u> </u>	\checkmark	-
sivated 18-8 Stainless Steel Pan d Phillips Screw	Screew for enviope attachment	SCREW1	50	McMaster-Carr	McMaster-Carr	TBD	\$11.90	\checkmark		https://www.mcmaster.com/91772A530/	\checkmark	\checkmark	
dium-Strength Steel Hex Nut	nut for envope attachment	NUT	100	McMaster-Carr	McMaster-Carr	TBD	\$4.88	<u> </u>	L			<u> </u>	1
	Ultrasonic screws	USCREW	25	McMaster-Carr		TBD	\$4.00 \$8.41	=	=				1
		USCREW		-			*****	~	Ш		<u> </u>	✓ <u> </u>	1
	constrain as leaded of the state			McMaster-Carr	McMaster-Carr	TBD	\$9.24	\checkmark		https://www.mcmaster.com/90131A101/	✓	\checkmark	
	washers on inside of envolope of external part connections	WASH	30					h	Ь				1
Pan Head Phillips Screw	washers on inside of envolope of external part connections 3D Printer Filament	WASH	30	NylonX	Matterhackers	TBD	\$58.00			https://www.matterhackers.com/store/l/nvlonx-carbon-fiber-nvlon-filament-1 75mrt			1
I Pan Head Phillips Screw	external part connections 3D Printer Filament	FIL		7.	Matterhackers scientific sales	TBD 200	\$58.00 \$35.00	H	Н	https://www.matterhackers.com/store/l/nylonx-carbon-fiber-nylon-filament-1.75mr https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823			
el Pan Head Phillips Screw con Fiber Filament - 1.75mm 8 Weather Balloon	external part connections 3D Printer Filament Helium Lift Bag	FIL LFTB		Hwoyee	scientific sales	200	\$35.00			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823	<u> </u>		
el Pan Head Phillips Screw con Fiber Filament - 1.75mm 8 Weather Balloon 8 Al He Cylinder W/ CGA580 Valve	external part connections 3D Printer Filament Helium Lift Bag Helium	FIL LFTB He	1 1 1	Hwoyee Varies	scientific sales Praxair	200 N/A	\$35.00 \$248.00			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823 https://www.praxairusa.com/			
el Pan Head Phillips Screw con Fiber Filament - 1.75mm I Weather Balloon I AI He Cylinder W/ CGA580 Valve Park PCB	external part connections 3D Printer Filament Helium Lift Bag Helium PCB printing	FIL LFTB He PCB	1 1 1 1	Hwoyee	scientific sales Praxair OSH Park	200 N/A TBD	\$35.00 \$248.00 \$260.10			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823 https://www.praxairusa.com/ https://oshpark.com/			
el Pan Head Phillips Screw con Fiber Filament - 1.75mm I Weather Balloon Is AI He Cylinder W/ CGA580 Valve Park PCB o to Shaft Clamping Coupler 0.250	external part connections 3D Printer Filament Helium Lift Bag Helium PCB printing P-Servo connector	FIL LFTB He PCB	1 1 1 1 1	Hwoyee Varies OSH Park	scientific sales Praxair OSH Park ServoCity	200 N/A TBD	\$35.00 \$248.00 \$260.10 \$28.00			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823 https://www.uraxairusa.com/ https://www.servocity.com/0-250-24-footh-c1-spline-serve-to-shaft-clamping-coup			
el Pan Head Phillips Screw bon Fiber Filament - 1.75mm 3 Weather Balloon 3 Al He Cylinder W/ CGA580 Valve 1 Park PCB 10 to Shaft Clamping Coupler 0.250 rofile Ends, 1045 Carbon Steel, 1/4	external part connections 3D Printer Filament Helium Lift Bag Helium PCB printing - Serve connector - D-Profile Rotary Shaft	FIL LFTB He PCB CUPLR SHAFT	1 1 1 1 4	Hwoyee Varies	scientific sales Praxair OSH Park ServoCity McMaster-Carr	200 N/A TBD	\$35.00 \$248.00 \$260.10 \$28.00 \$27.16			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823 https://www.praxairusa.com/ https://www.servodiy.com/0-250-24-looffi-c1-spline-servo-to-shaft-clamping-cour https://www.mcmaster.com/8632T132/	✓		
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sel Pan Head Phillips Screw ele Pan Head Phillips Screw rbon Fiber Filament - 1.75mm 88 Weather Balloon 13 Al He Cylinder W. CGA580 Valve H Park PCB rot to Shaft Clamping Coupler 0.250 Profile Ends, 1045 Carbon Steel, 1/4 1/20244 MS 002 0005 PH 57-440-AL-6 100 FL-R-SMT-1 pin male header 868	external part connections 3D Printer Filament Helium Lift Bag Helium PCB printing Gervo connector "D-Profile Rotary Shaft Votage Regulator Standoff Voltage Regulator Screew Hex Standoff Philips screw	FIL LFTB He PCB CUPLR SHAFT VR STAND VR SCREW PCB STAND PCB SCREW ANT1	1 1 1 1 1 4 4 10	Hwoyee Varies OSH Park McMaster-Carr	scientific sales Praxair OSH Park ServoCity McMaster-Carr Digikey Digikey Digikey Digikey	200 N/A TBD	\$35.00 \$248.00 \$260.10 \$28.00 \$27.16 \$6.27 \$7.47 \$3.04 \$0.44			https://www.scientificsales.com/8238-Weather-Balloon-350-Grams-Natural-p/823 https://www.praxairusa.com/ https://www.praxairusa.com/ https://www.servocity.com/0-250-24-toolth-c1-spline-servo-to-shaft-clamping-coup https://www.mennaster.com/86321132/ https://www.digikey.com/en/products/detail/w%C3%BCrth-elektronik/97012024/ https://www.digikey.com/en/products/detail/%-1-fastener-supply/MPMS-002-0005- ///www.digikey.com/en/products/detail/%-1-fastener-supply/MPMS-002-0005- ///www.digikey.com/en/products/detail/%-1-fastener-supply/MPMS-002-0005- ///www.digikey.com/en/products/detail/9900/36-9900-ND/3173217item-Seq-3-			

Version 5.3						State Based Cur	rrent(mA)			Time in State(hr	s/day)							
Name	Description	Quantity	Nominal Voltage	Input V Regulation(VIH)	Output V Variation(VOH)		High Measured	Low Expected	Low Measured	High	Low	Total Consumption/Cycle(mA*hrs)	Total Energy/Cycle (mW*hrs)	Power%	Datasheet link		Versioning	Power
LiPo Battery Charger	Charger	1	11.1	100-240V AC (From outlet)		.1-10A adjustable	9						100000 (when using AC input)		https://www.maxi	With current battery, should take 61 minutes to fully charge from 83% depleted while auto-balancing cel 100W charging power max	-Same as V5.2, but with the origin motors included instead of the replacements	Microcontroller, Microproccesor, and Oscillators
LIPO 11.1V 3S 11000mAh	Battery	1	11.1		9.6-12.6V							-9166.66	-101750	-28.109		83% of 11000mAh discharge for max current 40C discharge rating, 440A max discharge current 5C charge rating, 48.5A max charge current 3.2 per cell cutoff voltage(9.6V total) 4.2 per cell max voltage(12.6V total)		Sensors
uC32	Microcontroller	1	5	(VDD) 2.3-3.6V (Cannot go lower than 1.75' unless it will lose RAM data (I/O) 2.64-3.6V VIH 066V VIL	2.4 - 3.6V VOH 0-4V VOL	75	60	0	0	1	0	60	300	0.089	https://drive.goog	60mA current tested running servo code		Actuators
Raspberry Pi 3B+	Microprocessor (VBAT)	1	5	2.5-5.25V	N/A	1200	500	0	0	1	0	500	2500	0.699	https://drive.goog	Using 5V 1.2A maximum current draw specification datasheet	s	
AKK KC03	Camera/Transm itter	1	11.1	7-20V	Supplies 5V Vout for Camera	340	305	0	0	1	0	305	3385.5	0.939	6 https://www.ama	Supply current for transmitter too		
FS-iA6B Receive		1	5	4-6.5V	N/A	20	34	0	0	1	0	34	170	0.059		Tested receiving 30mA current constant for receiving	g	
Serial Telemetry Transmitter	Data Transmitter	1	5	5V	N/A	100	100	0	0	1	0	100	500	0.149	https://www.spark	100mA current needed for transmitting at 20dBm		
				(VDD)1.95-3.6V (VDDIO)1.62-3.6V														
MPL3115A2	Pressure/Temp erature Sensor	1	3.3	(I/O) 2.475-3.3V VIH 099V VIL	2.97-3.3V VOH 033V VOL	0.265	0.16	0	0	1	0	0.16	0.528	0.009	https://drive.goog	Typical current needed during Acquisition/Conversion of data in high resolution mode		
				(VDD) 5V										2.007		Current needed for each 4 sensors running for who		
HC-SR04	Ultrasonic	4	5	(Trigger)2-5V	2-5V	15	2	0	0	1	0	8	40.844	0.019	https://drive.goog	cycle, added power for PWM usage by sensors to r data at 16hz (.844mW)		
MTK 3339	GPS Module	1	3.3	(VDD)3-4.3V (VDDBackup)2-4.3V (I/O) 2-3.3V VIH 08V VIL	2.4-2.8V VOH 04V VOL	25	25	20	22	0.01	0.99	22.03	72.7099	0.029	https://drive.goog	25mA(Tested) Acquisition of GPS Signal takes 30s 20mA supply current for Tracking, adds 1.09mW for continuous UART power usage		
				(VDD)1.71-3.6V (VDDIO) 1.71-1.95V												· · · · · · · · · · · · · · · · · · ·		
ICM-20948	IMU 9DoF IC	1	3.3	(I/O) 1.35-2.3V VIH 054V VIL (VDD)1.8-3.6V	1.62-1.8V VOH 018V VOL	3	3	0	0	1	0	3	9.9	0.009	https://drive.goog	Typical current during data acquisition in 9-axis more	3	
MPRLS0001 PG00001C	Pressure sensor	1	3.3	(I/O) 066V VIL 2.64-3.3V VIH	066V VOL 2.64-3.3V VOH	4	4	0	0	1		4	13.2	0.009	https://sansing.bs	Tested to receive 3mA max while on		
1275kV	Motor	4	11.1	6.4-12.6V	N/A	3900	7840	1900	6550	1	0	31360	348096	96.149	runner-motor.htm	Depends on how much time drone is rising, 2x how at full throttle, .125g of thrust per motor to hover Also took into account 85% motor efficiency(Brushli See motor Power section	e e	
RC Sail Winch Servo			_													Depends on how much time drone is turning, Current ranges from 3-350mA, depending on how r the drone is turning	1	
	Servo	4	11.1	4.8-6V	N/A	350 Total Difference	350 -3190.895	3	1 1	1	0	1400 33796.19	7000 362088.6819	1.939	https://www.ebay			
			Max Voltage			. July Dimerellice	3044		1144	4		Total mAh	Total mWh					
Power Rails	Items	Voltage(V)	Average	Max Current(mA) Max Regu	Max at Dissipation() Due to Regulat lated Current(mA) Time Depende			Comments				Insert Flight Time Here(Hours)	Final Battery C	alculation	Results			
Rail 1	3	11.1 Nominal (Ranges from 12.6-9.6V)	0	12906	N/A 471	471	Heat comes from and from each m per 1000 feet 14	4ft wires to otor 1.588 ohms	max			1	Total Energy N		364,52	26.38		
							5V switching reg					Insert Percentage of time at high						
Rail 2	1	5	6.1	1400	5000 1235	1235	85% minimum et 3.3V Regulator u					throttle here(0-1)		scharge limit(mWh				
Rail 3	5	3.3	1.7	41	800 70	70	microcontroller Same 5V Switch					Insert Turning	Battery Conditi	on	Out of E	Battery		
Rail 6	5	5	6.1	750	5000 662	662	used for every of 85% efficient	her 5V part				time percentage here(0-1)	Power Used by		348,			
					2438 Maximum Tota							1	Power Used by Power Used by	Servos(mWh) Everything	7,01			
					lost to Heat(m	Wh)							Else(mWh) Power Lost to I	Heat	699			
													Else(mWh)		2,4	38 0.67%		

System Technical Requirements V.5

Requirement Number	er	Hierarchy				
	System	Subsystem	Component	Туре	Requirement	Verification
					Shall fly for at least 30 minutes with	
	Drone Flight				magnetometer payload during normal autopilot flight. May fly for one hour as a	Simulation - Test flights in simulations will model battery levels and estimate total obtainable flight time
1.0.0	Time			Performance	reach goal.	Physical Testing - The drone will be flown and have its flight time measured, as well as whole speeds and other factors.
						Design - Power budget will be created to assist in the design of the drone to select parts.
1.1.0		Power Usage		Performance	Component power should be optimized in part selection, design, and usage	Simulation - Current draw will be monitored in flight simulation to ensure power budget was accurate Physical Testing - Battery voltage will be monitored during flight to estimate power usage
1.1.0		1 Ower Osage		Chomance	Shall consume power within the power	Implation - Test power usage within simulator
1.1.1			Hardware	Functional	budget 11.4kW	Physical Testing - Power draw measured using multimeter
					Battery shall be capable of supplying all	Design - Power budget shall decide battery size
1.1.2			Battery	Functional	components listed in subsystem 1.1.1 for the amount of time listed in 1.0.0	Simulation - Battery levels will be monitored during simulated flights Physical Testing - The drone will have its flight time and battery levels measured
			Buttory	T dilottorial	The drone shall have an effective weight of	. Hydraci foliang this article might also date basely forms measured
		Drone			between 0 and 5 N. This requirement	
1.2.0		Effective Weight		Functional	must be balanced with 1.1.0, since the effective weight will affect power usage.	Design - Use solidworks to determine mass of drone, and use offset of estimated bouyancy of lift bag at STP to determine effective weight Testing - Weigh the drone after full construction and inflation to determine effective weight
1.2.0		Weight		i unctional	The weight of the drone body and	resulty - Weight the divide date full constitution to determine checker weight
					components shall be less then 15 N. The	
					payload weighs 9.8N. The estimated lifting force of the bag specified in 2.2.0 is 19.6N.	
					Weight should be minimized to assist on	
1.2.1			Drone Body	Functional	requirement 1.1.0, since effective mass is not affected.	Design - Weight budget will be used to estimate the weight of the final drone
1.2.1			Dione Body	runctional	The drone shall have a lift bag capable	Physical Testing - The drone body will be weighed on its own.
					of counteracting drone mass, resulting in	
					effective weight of < 5N. Currently specified at 4.07 cubic meter nominal volume, but	
					might change based on requirements 1.1.0	Design - The drone mass and required lift force will be calculated and adjusted to meet requirements
1.2.2			Lift Bag	Functional	and 2.1.0	Physical Testing - Will test effective weight
					System design may implement a throttle control in autopilot to remain in optimal	Design - See if this is a possible solution when motor choices and force analysis are finalized. Will use travel speed, not wind speed for optimization
		Control System			efficiency range of motors where possible,	Simulation - Test flight to see if there is a worthwhile difference
1.3.0		Optimization		Performace	but shall not conflict with 1.2.0	Physical Testing - Monitor throttle outputs during autopilot flight and battery voltage to estimate efficiency range of motors
	Minimal Drone				Drone shall be able to fly at least 5mph in	Design - All forces will be calculated, including drag, to estimate proper motor size Simulation - Flight Simulations will test flight speed under different conditions
2.0.0	Speed			Performance	winds up to 15mph	Similarion - Figin Similarions will east night speed unless different ordinations. Physical Testing - The drone will use its GPS to measure the flight speed of the drone.
					RC Control: Drone shall respond to user	
200	DC Control			Dorformonoo	commands with pitch and roll angles within	Simulation - Flight simulation will model responses to user input
3.0.0	RC Control	System		Performance	±0.1 radians and a height of 1±0.15m. The drone shall respond to user input in	Physical Testing - Motor angles will be tracked and used to confirm accuracy of RC response, and flight response in air will also be monitored
3.1.0		Response		Performance	< 0.5 seconds	Physical Testing - Time response time. Easiest test is from floating position to a hard acceleration.
					A controller should be chosen that is	
					capable of providing all neccesary commands (Foward, turn, ascend,	
					descend and others if deemed necessary)	
3.1.1			RC Controller	Functional	as well as relaying information to the user as specified in requirement 3.2.0	Physical Testing - The drone responds to the user's inputs, including forward, turn, ascend, and descend
0.1.1			TO CONTROLL	Tunctional	The software shall be fast enough to	Thysical results - The divise responds to the user's inputs, including forward, fully assected, and desected
					respond quickly to all user commands and	Simulation - Software speed will be measured in simulation tests.
3.1.2			Software	Functional	error handling.	Physical Testing - Responses will be measured for different situations in both normal flight and lab conditions (since we want to model a crash not actually crash)
					Some autonomous functions should be called while in the RC control state to assist the	
					user, such as terrain tracking 4.2.0 and	Simulation - Responses in simulation will be tested, with simulated terrain, simulated landing and other needed functions
3.1.3			Autonomous Functions	Functional/Performace	auto landing 4.5.0	Physical Testing - Using and switching between autonomous and RC FUNCTIONS will be tested for safety and reliability.
3.2.0		Data Feedback		Functional	The drone shall be able to send feedback to the user so the user can respond	Design - The software will be designed to send back drone data Physical Testing - Data will be monitored; this also assists with confirming other system requirements
					The drone shall send camera feedback to	
3.2.1			Camera	Functional	assist in user controlled flight	Physical Testing - The camera feed will be monitored and any cutouts or issues will be recorded
322			Crash Detection	Functional	The drone shall alert the user if it is determined to have crashed	Simulation - Crashes will be simulated and output will be monitored Physical Critical Acade will be disvoluted in a lab critical to acade or company
J.Z.Z			Grasii Detection	i uncudital	The drone shall alert the user when the	Physical Testing - A crash will be simulated in a lab setting to test drone response
					battery runs low, and autoland specified in	
3.2.3			Low Battery	Functional	requirement 4.5.0 if critically low. Specific values to be determined.	Simulation - Voltage levels will be simulated and fed to the system and response of the simulation will be recorded Physical Testing - The drone will be flown until it has a low battery, then pushed until it autolands. Tested away from users in case of crash
5.2.0			LOW DUTTER	- G. Gioriai	The drone should be able to switch from	Typical Totaling The World Third to Horn United Chapter of Valley, when positive united a duty adultations. Tested away from users in case of classification
					autonomous control to remote control	
		Switch from			but may need to move into a safe position first, and have preset settings, to	Design - Can switch control modes from remote controller
		Autonomous			prevent a crash. Component level will be	Design - Can switch control modes from remote controller Physical Testing - The drone will be switched to RC control from autonomous in both safe and unsafe conditions to measure effectiveness
3.3.0		Control to RC		Performance	determined as we find potential problems	Simulation - Switching between RC and autonomous in the simulation will be tested
					The system should be able to maintain pitch and roll angles within 0.1 radians of zero and a	
3.4.0		Closed-Loop RC		Functional	height of 1+-0.15 m	
						Design - A feedback control system will be used to ensure the drone succesfully flies on its own
4.0.0	Autonomous Control			Performance/Environmental	Autonomous Control: Drone shall maintain stable pitch/roll/height and follow a path.	Simulation - Paths and environments will be generated to test autonomous responses Physical Testing - Plan a path and test the drones ability to follow it accurately using GPS data and monitor height with ultrasonics
4.0.0	Control			- Chomiance/Environmental	The drone shall be able to follow a path	representating in the a path and test the divines ability to follow it accurately using GFS talk afformation fleight with disastrics
					specified by the user in up to 15mph wind	Simulation - Drone position will be kept track of in a simulated environment with wind
The second second						
4.1.0		Path Following		Performance/Environmental	with a positional accuracy of 5m	Physical Testing - GPS data will be used to confirm drone position compared to the preplanned path.
4.1.0		Path Following	GPS Sensor	Functional	The sensor shall be accurate to within 5m of its location with sampling of at least 3Hz	Physical Testing - GPS data will be used to confirm drone position compared to the preplanned path. Design - Check sensor data sheet for rated values Physical Testing - Test GPS location data compared with actual sensor position

					The drone shall maintain a constant height	
4.2.0		Terrain Tracking		Performance/Environmental	above the ground, approximately 1m, and adjust height as needed. < 15% overshoot	Simulation - A terrain will be generated and the drone's height will be monitored, as well as its altitude Physical Testing - The drone will be given a preplanned path over flat and rough terrain and its flight while ultrasonic data will be used to confirm actual height
424			Lilteragnia Communication	Functional	The sensors shall be able to monitor area in front of the drone in order to maintain constant height of 1 m. Physical location	Simulation - The field of view and ability of the sensors to update the system in time will be simulated
4.2.1			Ultrasonic Sensors	Functional	and sensor sensitivity affects effectiveness The sensor shall be able to monitor altitudes	Physical Testing - The range of the sensors, distance, and angle will be confirmed and adjusted for
4.2.2			Barometric Sensor	Functional	above 4m for drone altitude awareness. The drone should be able to detect flight	Physical Testing - Measurements in real-time will be confirmed and adjusted for
4.3.0		Error Handling		Performance/Environmental	errors and compensate accordingly, specified in component section	Simulation - Errors will be simulated and responses recorded Physical Testing - Errors will be performed in a lab setting and responses will be recorded
4.3.1			IMU Sensor	Functional	The IMU should be able to detect crashes and abnormal situations and feed the data back into the system	Simulation - Sensor reading will be simulated and fed into error handling functions to test functionality Physical Test - The data will be tested in flight and in a laboratory setting to ensure proper system response
4.3.2			Error Detection	Functional	Follows same requirements as 3.2.2 and 3.2.3	Same tests as in 3.2.2 and 3.2.3
					The drone should be able to detect a popped balloon with a pressure sensor and by calculating the force needed to stay afloat.	Simulation - The bouyancy of the drone will be quickly changed in a simulated flight and the drone response recorded Physical Test - The drone response will be tested with a sudden change in drone mass (attaching a weight) and pressure sensor
4.3.3			Popped balloon	Functional	Will make emergency landing if popped The drone shall send required data back to	manipulated to simulate deflation, and result will be recored. Test done in lab setting
4.4.0		Data Feedback		Functional	the user, as well as any errors specified in requirement 4.3.0	Simulation - Data will be fed to the simulator during simulated flights Physical Testing - Data will be tracked during flight and laboratory tests
4.5.0		Autolanding		Performance/Environmental	The drone should be able to safely land without damage to itself, the environment, or users.	Simulation - Simulated landing on several types of surfaces and wind conditions Physical Testing - The drone will be be autolanded throughout tests
460		Calculate New Path		Performance	If there is a system failure as specified in 4.3.0 that requires landing, the drone shall take the nearest path to a landing zone in <5s	Simulation - Partial system failure will be tested and the drone's ability to make it to a landing zone will be tested Physical Testing - TBD Need to plan safe way to test
		Switch from RC to Autonomous			The drone should be able to switch from remote control to autonomous control but may need to move into a safe position first, and have preset settings to prevent a crash. Specifics will be	Design - Can switch modes from remote controller Physical Testing - The drone will be switched to autonomous control from remote control in both safe and unsafe conditions to measure effectiveness
4.7.0		Control		Performance	determined as we find potential problems The drone should be able to determine its pitch, yaw roll, and NED states, as well as their	Simulation - Switching between RC and autonomous in the simulation will be tested
4.8.0		State Estimation		Functional	derivatives to use in the autonomous functionality and error detection systems	Design- Returns pitch, yaw, row, and state of microprocessor. Physical Testing - Drone must stabalize itself after a while.
					The drone shall cost less than \$10,000. Reach-The drone may cost less than	
5.0.0	Cost Magnetometer			Functional	\$6,000 The magnetometer interference shall be	Bill of materials will be recorded and manufactuing costs will be estimated
6.0.0	Interference			Performance	less than 10 nT	Physical Testing - The magnetometer will record magnetic field with the drone on and off and the difference will be compared
6.1.0		Motor Generated		Functional	The interference from the motors will be calculated and shall be less than requirement 6.0.0	Design -The magnetic contributions from motor locations will be added and summed together to estimate net interference
		menerence	Outputted Interference		The interference should be calculated for	Design - Magnetic Fields will be calculated for each motor
6.1.1			Interference	Functional	each motor, and balanced with 6.1.2 The physical location of motors on the	Physical Testing - Will use a magnetometer to record actual magnetic fields with the motor on
0.4.0			Dhusias I I sasting	Eventional	drone shall be decided with magnetic field outputs of motors in 6.1.1 in order to meet 6.0.0	Design - The positions will be calculated based off motor outputs in 6.1.1
6.1.2	Drone		Physical Location	Functional	The drone, its usage, and build should	Physical Testing - The total field will be measured at the point of the magnetometer
7.0.0	Safety			Performance/Functional/Safety	be safe to all individuals involved The drone should be designed to cause	Test for each factor will be designed, tests listed in subsystems
7.1.0		Collision Considerations		Performance	minimal damage to the user and environment in the event of a crash	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
				- II III II	The body of the drone should have no protruding or sharp componets that can	
7.1.1			Body Design	Functional/Safety	hurt the user The propellers should be safe if the user comes in close contact. May be enforced	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
7.1.2			Propellers	Functional/Safety	with propeller guards, propeller material, and other methods that will be determined.	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
7.2.0		Electrical Safeguards		Functional/Safety	The drone should have features to protect the drone and user from electrical malfunctions	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
7.2.1			Protected Electronics	Functional/Safety	The electronics and wires should be protected and securely mounted to prevent shorts, dropping components, etc	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
7.2.2			(Fuses, Other safeguards?)	Functional/Safety	Some electrical components may be implemented to protect electronics	Physical Testing - Since this is a safety test, we will need to find a safe way to test this. Specific test TBD
7.3.0		Helium		Safety	A method of handling shall be adopted or developed by the team to ensure the helium and helium tank are handled safely	A procedure will be written up in team documentation and followed by all people handling helium
	Helium	ricium		Environmental/Functional	The lift bag shall maintain 90% of its	
8.0.0	Leakage	Detect II "		Environmental/Functional	buoyancy over a one week period Pressure sensor shall send out an error	The drone will be weighed on a scale, then weighed again a week later. Helium loss will be calculated at the end of the week.
8.1.0		Detect Helium Leakage		Enviromental/Functional	message when pressure sensor is decreasing during flight	Physical Testing - Will test a slowly leaking balloon with the pressure sensor mounted. Can be tested with normal air
8.1.1			Pressure Sensor	Environmental/Functional	Sensor should detect volume drop of helium balloon	In software, check if error message is being received
9.0.0	Leagal Compliance			Legal/Functional	The drone and team shall abide by all applicable laws for drone flight.	

9.1.0		FAA Compliance		Legal/Functional	The drone and team shall abide by all applicable laws for drone flight	Get confirmation from someone who knows drone laws (TBD) so we abide by all rules					
9.1.0		Compliance	Follow FAA part	Legal/Functional	applicable laws for drone flight	Get confirmation from someone who knows drone laws (TBD) so we adde by all rules					
9.1.1			107 (flying small drones < 30kg)	Legal/Functional	Drone shall be less than 30kg when in flight outdoors.	Get confirmation from someone who knows drone laws (TBD) so we abide by all rules					
9.1.2			Statutory provision (PL 115-254, Section 350)	Legal/Functional	Drone shall follow PL 115-254, Section 350. "Use of Unmanned Aircraft Systems at Institutions of Higher Education."	Get confirmation from someone who knows drone laws (TBD) so we abide by all rules					
9.1.3			Registered Drone with FAA	Legal/Functional	We shall create an FAADroneZone account with email and password and register the drone through the account.	Receive Registration confirmation					
10.0.0	Noise Level			Performance	The drone should be quieter than 65dB	Physical Testing - Measure noise level of drone from 5 ft away					
11.0.0	Manufacturability			Functional	The drone should be able to be manufactured with equipment within our access, further decomposed in subsystem requirements	Can be manufactured					
11.1.0		3D Printing		Functional	All 3d printed components should fit within dimensions of the Lulbot Taz4 or Lulzbot Mini printer beds	Can be printed					
11.2.0		Soldering		Functional	All components must be able to be soldered with Soldering Iron and team member skill level						
11.3.0		Envelope		Funtional	The envelope must be able to be cut with standard scissors and sewed with sewing machine Dylan owns, for dimensions to be within 5% of design when inflated.						
11.4.0		PCB		Functional	The PCB design must fit within manufacturing manufacturing abilities of OSHAPark						
*All edits with	in a section, add at the	bottom of that se	ection, since interpender	icies refer to specific numbers	. If a new system requirements, add to the bot	tom of list					
1/13/2021 1/28/2020 2/4/2021	/13/2021 V.1 Created //28/2020 V.2 Cleaned up shall statements to make requirements more clear. Finalized for submission.										
2/8/2021	V.4 Updated shall s	hould and may s	statements to make more	e concrete and finalize for sub-	mission. 30min Flight time shall, and 60 move	d to may, but it is highly preferable.					
6/7/2021	V.5 Updated techinoa	ail requirmets to th	eir final state, some where	move around, and varification me	ethods						