

EI/IC-402

B.E. IV Semester

Examination, December 2016

Signals and Systems

Time : Three Hours

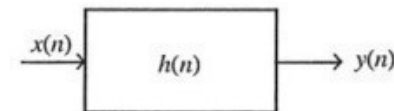
Maximum Marks : 70

- Note:** i) Answer five questions. In each question part A, B, C is compulsory and D part has internal choice.
ii) All parts of each question are to be attempted at one place.
iii) All questions carry equal marks, out of which part A and B (Max.50 words) carry 2 marks, part C (Max.100 words) carry 3 marks, part D (Max.400 words) carry 7 marks.
iv) Except numericals, Derivation, Design and Drawing etc.

Unit - I

1. a) What is the physical significance of Eigen function?
b) How do you characterize a LTI system?
c) Explain transmission of signals through LTI system.
d) Prove for the given system

$$y(n) = \sum_{k=-\infty}^{\infty} x(k)h(n-k)$$



OR

What do you mean by static and dynamic behaviour of the system? Explain with suitable example.

[4]

- c) What is VSAT system? How does a VSAT work? Who uses VSAT system?
- d) With the help of block diagram explain the operation of DBS system.

OR

What are VSAT networks? Explain various VSAT network topologies.

Total No. of Questions :5]

[Total No. of Printed Pages : 4

Roll No

EC-702

B.E. VII Semester

Examination, December 2016

Satellite Communication

Time : Three Hours

Maximum Marks : 70

- Note :** i) Answer five questions. In each question part A, B, C is compulsory and D part has internal choice.
- ii) All parts of each question are to be attempted at one place.
- iii) All questions carry equal marks, out of which part A and B (Max.50 words) carry 2 marks, part C (Max.100 words) carry 3 marks, part D (Max.400 words) carry 7 marks.
- iv) Except numericals, Derivation, Design and Drawing etc.

Unit - I

1. a) Why is it preferable for a remote sensing satellite to be in a sun-synchronous orbit?
- b) Describe the characteristics and uses of geostationary orbit.
- c) Define the following parameters with reference to satellite orbits :
- i) Apogee and perigee
- ii) Eccentricity
- iii) Ascending and descending nodes
- d) Define Kepler's laws of orbiting bodies and derive an equation to show that the third law is true for any orbiting satellite.