6.2 UAS Model and Control

The key feature of *Movement Automaton* is to interface *UAS system* as the *discrete* command chain. Following topics are introduced in this section:

- 1. Movement Automaton Applications (sec. 6.2.1) the listing of related work and similar approaches to ours.
- 2. UAS Model (sec. 6.2.2) simple plane model used in this work as controlled plant.
- 3. UAS Movement Automaton (sec. 6.2.3) movement automaton for UAS Nonlinear Model constructed from scratch.

6.2.1 Movement Automaton Applications

Movement Automaton is basic interface approach for discretization of trajectory evolution or control input for any continuous or discrete system model.

Main function of Movement Automaton is for system given by equation state = f(time, state, input) with initial state $state_0$ to generate reference trajectory state(t) or control $signal\ input(t)$.

Using Movement Automaton as Control Proxy will provide us with discrete command chain interface. This will reduce the non deterministic element from Evasive trajectory generation, by reducing infinite maneuver set to finite movement set.

Non determinism of Avoidance Maneuver have been discussed as an issue in following works:

- 1. Newton gradient method for evasive car maneuvers [1].
- 2. Non-holistic methods for trajectory generation [2].
- 3. Stochastic approach to elliptic trajectories generation [3].

Examples of Movement Automaton Implementation as Control Element can be mentioned as follows:

- 1. Control of traffic flow [4].
- 2. Complex air traffic collision situation resolution system [5, 6].
- 3. SAA/DAA capable avoidance system [7].

6.2.2 UAS Model

Motivation: Simplified rigid body kinematic model will be used. This model have decoupled roll, yaw and pitch angles. The focus is on *reach set approximation methods*, therefore *UAS model* is simplified.

State Vector (eq. 6.1) defined as positional state in euclidean position in right-hand euclidean space, where x, y, z can be abstracted as latitude, longitude, altitude.

$$state = [x, y, z, roll, pitch, yaw]^{T}$$
(6.1)

Input Vector (eq. 6.2) is defined as linear velocity of UAS v and angular speed of rigid body $\omega_{roll}, \omega_{pitch}, \omega_{yaw}$.

$$input = [v, \omega_{roll}, \omega_{pitch}, \omega_{yaw}]^T$$
 (6.2)

Velocity vector function (eq. 6.3) is is defined through standard rotation matrix and linear velocity v, oriented velocity $[v_x, v_y, v_z]$ given by (eq. 6.4).

$$\begin{bmatrix} v_x \\ v_y \\ v_z \end{bmatrix} = \begin{bmatrix} v\cos(pitch)\cos(yaw) \\ v\cos(pitch)\sin(yaw) \\ -v\sin(pitch) \end{bmatrix}$$
(6.3)

UAS Nonlinear Model (eq. 6.4) is given by *first order equations:*

$$\frac{\mathrm{d}x}{\mathrm{d}time} = v \cos(pitch) \cos(yaw); \qquad \frac{\mathrm{d}roll}{\mathrm{d}time} = \omega_{roll};$$

$$\frac{\mathrm{d}y}{\mathrm{d}time} = v \cos(pitch) \sin(yaw); \qquad \frac{\mathrm{d}pitch}{\mathrm{d}time} = \omega_{pitch};$$

$$\frac{\mathrm{d}z}{\mathrm{d}time} = -v \sin(pitch); \qquad \frac{\mathrm{d}yaw}{\mathrm{d}time} = \omega_{yaw};$$
(6.4)

6.2.3 UAS Movement Automaton

Motivation: An *UAS Nonlinear Model* (eq. 6.4) can be modeled by *Movement Automaton* (def. ??).

Movement Primitives by (def. ??) are given as (eq. ??). To define primitives the minimal time is 1s. The maximal duration is also 1s.

Assumption 1. Let assume that transition time of roll, pitch, yaw, linear velocity is 0s.

Under the assumption (as. 1) the movement transitions (def. ??) have 0 duration.

Note. The assumption (as. 1) can be relaxed under condition that path tracking controller exists.

Movements (def. ??) for fixed step k we start with discretization of the input variables. The linear velocity in text step is given:

$$v(k+1) = v(k) + \delta v(k) \tag{6.5}$$

The roll, pitch, yaw for next step are given

$$roll(k+1) = roll(k) + \delta roll(k)$$

$$pitch(k+1) = pitch(k) + \delta pitch(k)$$

$$yaw(k+1) = yaw(k) + \delta yaw(k)$$
(6.6)

The $\delta v(k)$ is velocity change, $\delta roll(k)$, $\delta pitch(k)$, $\delta yaw(k)$, are orientation changes for current discrete step k. If the duration of transition is 0s (as. 1) then 3D trajectory evolution in discrete time is given as:

$$x(k+1) = x(k) + v(k+1)\cos(pitch(k+1))\cos(yaw(k+1)) = \delta x(k)$$

$$y(k+1) = y(k) + v(k+1)\cos(pitch(k+1))\sin(yaw(k+1)) = \delta y(k)$$

$$z(k+1) = z(k) - v(k+1)\sin(pitch(k+1)) = \delta z(k)$$

$$time(k+1) = time(k) + 1 = \delta time(k)$$
(6.7)

The $\delta x(k)$, $\delta y(k)$, $\delta z(k)$ are positional differences depending on *input vector* for given discrete time k:

$$input(k) = \begin{bmatrix} \delta x(k), \delta y(k), \delta z(k), \delta v(k), \\ \delta roll(k), \delta pitch(k), \delta yaw(k), \delta time(k) \end{bmatrix}^{T}$$
(6.8)

The *state vector* for discrete time is given:

$$state(k) = \begin{bmatrix} x(k), y(k), z(k), v(k), \\ roll(k), pitch(k), yaw(k), time(k) \end{bmatrix}^{T}$$

$$(6.9)$$

The nonlinear model (eq. 6.4) is then reduced to *linear discrete model* (eq. 6.10) given by *apply movements* function (eq. 6.5, 6.6, 6.7).

$$state(k+1) = applyMovement(state(k), input(k))$$
 (6.10)

Movement Set for linear discrete model (eq. 6.10) is defined as set of extreme unitary movements on main axes (tab. 6.1) and diagonal axes (tab. 6.2).

input(movement)	Straight	Down	Up	Left	Right
$\delta x(k)[m]$	1.00	0.98	0.98	0.98	0.98
$\delta y(k)[m]$	0	0	0	0.13	-0.13
$\delta z(k)[m]$	0	-0.13	0.13	0	0
$\delta roll(k)[^{\circ}]$	0	0	0	0	0
$\delta pitch(k)[^{\circ}]$	0	15°	-15°	0	0
$\overline{\delta yaw(k)[^{\circ}]}$	0	0	0	15°	-15°

Table 6.1: Input values for main axes movements.

input(movement)	Down-Left	Down-Right	Up-Left	Up-Right
$\delta x(k)[m]$	0.76	0.76	0.76	0.76
$\delta y(k)[m]$	-0.13	0.13	0.13	-0.13
$\delta z(k)[m]$	-0.13	-0.13	0.13	0.13
$\overline{\delta roll(k)[^{\circ}]}$	0	0	0	0
$\overline{\delta pitch(k)[^{\circ}]}$	-15°	-15°	15°	15°
$\delta yaw(k)[^{\circ}]$	15°	-15°	15°	-15°

Table 6.2: Input values for diagonal axes movements.

Note. Movement set in shorten form is given as

$$MovementSet = \begin{cases} Straight, Left, Right, Up, Down, \\ DownLeft, DownRight, UpLeft, UpRight \end{cases}$$
(6.11)

Trajectory by (def. ??) for initial time time = 0, initial state state(0) and Movement Buffer (from def. ??):

$$Buffer \in MovementSet^*(eq.6.11), \quad |Buffer| \in \mathbb{N}$$
 (6.12)

Trajectory (eq. 6.13) is then given as the time-series of discrete states:

$$Trajectory(state(0), Buffer) = \begin{cases} state(0) + \sum_{j=0}^{i-1} input(movement(j)) : \\ i \in \{1 \dots |Buffer| + 1\}, \\ movement(\cdot) \in Buffer \end{cases}$$

$$(6.13)$$

Trajectory (eq. 6.13) is ordered set of states bounded to discrete time $0 \dots n$, where n is member count of *Buffer*. Trajectory set has n+1 members:

Trajectory(state(0), Buffer) =

$$\begin{cases} state(0) = state(0) + \{\} \\ state(1) = state(0) + input(movement(1)) \\ state(2) = state(0) + input(movement(1)) + input(movement(2)) \\ \vdots = \vdots \\ state(n) = state(0) + input(movement(1)) + \dots + input(movement(n)) \end{cases}$$

$$(6.14)$$

State Projection (eq. 6.15) for the *Trajectory* (eq. 6.13) is given as follow:

$$StateProjection(Trajectory, time) = Trajectory.getMemberByIndex(time + 1)$$

$$(6.15)$$

Note. Movement Automaton for system (eq. 6.4) with given (as. 1) is established with all related properties (sec. ??).

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