## 7.5 Test Cases Conclusion

This section contains summary of performance evaluation (sec. ??), adversary behaviour impact on our approach (sec. ??), calculation load in (sec. ??).

## 7.5.1 Performance Evaluation

Performance of test cases were evaluated according to criteria given by (sec. ??). The performance for test cases from test plan (tab. ??) have been summarized in (tab. ??).

Scenario	Safety Margin			Trajectory tracking			
name	Distance		Breach	Waypoint	Reference	Acceptable	/
	min	max	Dreach	Reach	Deviation	Deviation	Fail
Building avoidance (sim. ??)	0.69 m UAS 1	24.98 m UAS 1	No (??)	Yes/UAS 1/(??)	$\mathbb{WP}_1: 107.05m$ $\mathbb{WP}_2: 86.20m$ $\mathbb{WP}_3: 28.70m$ $\mathbb{WP}_4: 32.84m$	Yes (??)	Pass
Slalom (sim. ??)	0.09 m UAS 1	3.74 m UAS 1	No (??)	Yes/UAS 1/(??)	$\mathcal{WP}_1$ : $20.06m$	Yes (??)	Pass
Maze (sim ??)	0.01 m UAS 1	2.95 m UAS 1	No (??)	Yes/UAS 1/(??)	$\mathcal{WP}_1: 28.06m$	Yes (??)	Pass
Storm (sim. ??)	0.04 m UAS 1	34.99 m UAS 1	No (??)	Yes/UAS 1/(??)	$WP_1: 15.76m$	Yes (??)	Pass
Emergency Converging (sim. ??)	1.67 m UAS 1-2	27.08 m UAS 1-1	No (??)	Yes/UAS 1/(??) Yes/UAS 2/(??)	$\begin{array}{ c c c c c c }\hline \mathcal{WP}_1: & 3.25m \\\hline \mathcal{WP}_1: & 0.00m \\\hline \end{array}$	Yes (??)	Pass
Emergency Head On (sim. ??)	0.38 m UAS 1-2	38.00 m UAS 1-2	No (??)	Yes/UAS 1/(??) Yes/UAS 2/(??)	$\frac{\mathcal{WP}_1:  3.25m}{\mathcal{WP}_1:  0.00m}$	Yes (??)	Pass
Emergency Multiple (sim. ??)	0.20 m UAS 2-4	45.46 m UAS 3-4	No (??)	$\frac{\text{Yes/UAS 1/(??)}}{\text{Yes/UAS 2/(??)}}$ $\frac{\text{Yes/UAS 3/(??)}}{\text{Yes/UAS 4/(??)}}$	$\begin{array}{c} \mathcal{WP}_1:  4.84m \\ \hline \mathcal{WP}_1:  1.83m \\ \hline \mathcal{WP}_1:  3.45m \\ \hline \mathcal{WP}_1:  2.05m \end{array}$	Yes (??)	Pass
Rule-based Converging (sim. ??)	1.22 m UAS 1-2	20.28 m UAS 1-2	No (??)	Yes/UAS 1/(??) Yes/UAS 2/(??)	$\frac{\mathcal{WP}_1:\ 10.22m}{\mathcal{WP}_1:\ 0.00m}$	Yes (??)	Pass
Rule-based Head On (sim. ??)	0.21 m UAS 1-2	36.33 m UAS 1-2	No (??)	Yes/UAS 1/(??) Yes/UAS 2/(??)	$\begin{array}{c cccc} \mathcal{WP}_1: & 5.40m \\ \hline \mathcal{WP}_1: & 5.40m \end{array}$	Yes (??)	Pass
Rule-based Multiple (sim. ??)	0.54 m UAS 2-3	32.24 m UAS 1-2	No (??)	$\frac{\text{Yes/UAS 1/(??)}}{\text{Yes/UAS 2/(??)}}$ $\frac{\text{Yes/UAS 3/(??)}}{\text{Yes/UAS 4/(??)}}$	$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	Yes (??)	Pass
Rule-based Overtake (sim. ??)	0.80 m UAS 1-2	48.85 m UAS 1-2	No (??)	Yes/UAS 1/(??)  Yes/UAS 2/(??)	$\mathcal{WP}_1: 24.00m$ $\mathcal{WP}_2: 0.00m$ $\mathcal{WP}_3: 4.00m$ $\mathcal{WP}_4: 5.00m$ $\mathcal{WP}_1: 0.00m$	Yes (??)	Pass

Table 7.1: Test cases performance evaluation.

**Highlights:** Each *scenario* contains reference to notable simulation moments and results. The scenarios were grouped according to *Operational Space* category and each category is separated by strike line.

Non cooperative test cases for Rural/Urban environment:

- 1. Static obstacle avoidance (Building/Slalom/Maze) the buildings were correctly avoided without security breach, navigation algorithm was sufficient for given scenarios and obstacle density.
- 2. Weather avoidance (Storm) the moving storm have been avoided in both soft constraint and hard constraint state. The assumption of early detection/notification is key in successful weather avoidance.

Non cooperative test cases for Intruder Avoidance - the key assumptions are early intruder detection in Avoidance Grid and non-adversarial behaviour. Each UAS was running own instance of Navigation loop (fig. ??). The summary of test cases is going like follow:

- 1. Emergency converging both UAS identified correct roles according rules of the air. The UAS 2 kept right of the way.
- 2. Emergency head on both UAS identified correct roles according rules of the air, both of them uses full separation with Combined Reach Set Approximation (sec. ??).
- 3. Emergency mixed all four UAS enters into emergency avoidance mode intermediately after intruders detection. The non-cooperative consensus of separation is reached (fig. ??)

Cooperative test cases with UTM supervision are working according to UTM architecture (fig. ??), where the UTM is considered as main authority. The key assumptions are UTM Resolution fulfillment and non-adversary behaviour. Each UAS was running own instance of Navigation loop (fig. ??) with enabled Rule Engine (sec. ??). The summary of test cases is going like follow:

- 1. Rule-based converging correct handling of converging maneuver (fig. ??), proper rule invocation (rule ??) on UAS side.
- 2. Rule-based head on correct handling of head on maneuver (fig. ??), proper rule invocation (rule ??) on UAS side.
- 3. Rule-based multiple proper Collision case Merge (tab. ??) with new collision point (eq. ??) and safety margin calculation (eq. ??).
- 4. Rule-based overtake correct handling of overtake maneuver (fig. ??), proper rule invocation (rule ??). Divergence/Convergence (eq. ??,??) for multiple waypoints calculation works for various speed difference (fig. ??).

## 7.5.2 Adversary Behaviour Impact

The *abuse* of UAS for *ill intentions* realization is expected. The UAS is cheap, disposable and does not have an ethic boundaries.

One of the assumptions was that there are only intruders whom does not actively look to harm our UAS. Breaking this assumption can be lethal for our system and also for other systems.

Let us take Rule-based Head on test case (sec. ??), changing only following aspects:

- 1. UAS 2 position spoofing the adversarial vehicle is faking its position according to expected behaviour.
- 2. UAS 2 Navigation goal set as UAS 1 position from intercepted position notifications (tab. ??).

**Simulation:** The *simulation* (fig. ??) have been run with defined condition. UAS 2 (magenta) has been chosen as the *adversary*. UTM sees expected trajectory of UAS 2 (grey plane/trajectory) based on spoofed *position notifications*. The *navigation/avoidance grid* range (black dashed line boundary) is shown. The notable moment of simulation are:

- 1. Deviation detection (UAS2 ↔ UTM) (fig. ??) the collision case (tab. ??) is active and enforced by UTM. The adversary UAS 2 (magenta) starts deviating from expected trajectory (grey). UAS 1 (blue) does not register any foreign object in avoidance grid range (black dashed line).
- 2. Adversary attacking (UAS2 → UAS1) (fig. ??) the adversary UAS 2 (magenta) starts actively pursuing UAS 1 (blue) by changing original heading. This can be considered as beginning of active pursuit. UAS 1 (blue) does not detect any foreign object in avoidance grid (black dashed line boundary). UTM is receiving expected UAS position (grey plane/line).
- 3. Emergency avoidance (UAS1  $\rightarrow$  UAS2) (fig. ??) following happens:
  - a. Adversary UAS 2 (magenta) is spotted by UAS 1(blue), it entered into UAS 1 avoidance grid (black dashed line boundary).
  - b. UAS 1 (blue) enters into Emergency Avoidance Mode, because there is an foreign objectin avoidance grid.
  - c. *UTM* notices a warning to *UAS 1* (blue), because it entered into *Emergency Avoid*ance *Mode*. UTM is not aware of any breach, because of expected UAS 2 position (grey plane/line)

	d. Adversary UAS 2 (magenta) has UAS 1 (blue) locked in navigation grid as goal (which guarantees optimal path).
4.	Blind spot ( $\circlearrowright UAS1$ ) (fig. $\ref{eq:spot}$ ) following happens:
	a. $UAS\ 1$ (blue) returns to $Navigation\ Mode$ , because there is no $foreign\ object$ in $avoidance\ grid$ (black dashed line boundary).
	b. $UTM$ receives the mode change and it starts enforcing resolutions for collision case, Adversary $UAS\ 2$ is considered clear due expected position (grey plane/line) compliance with resolution.
	c. Adversary UAS 2 (magenta) is on UAS 1 blind spot. The target UAS 1 (blue) is locked in UAS 2 navigation grid (black dashed line boundary).
5.	Collision detail (UAS1 $\leftrightarrow$ UAS2)(fig. ??) - Target UAS 1 (blue) is hit by Adversary

 $UAS\ 2$  (magenta) on left wing tip. Both UAS are going down. UTM will detect sudden

loss of both UAS systems.

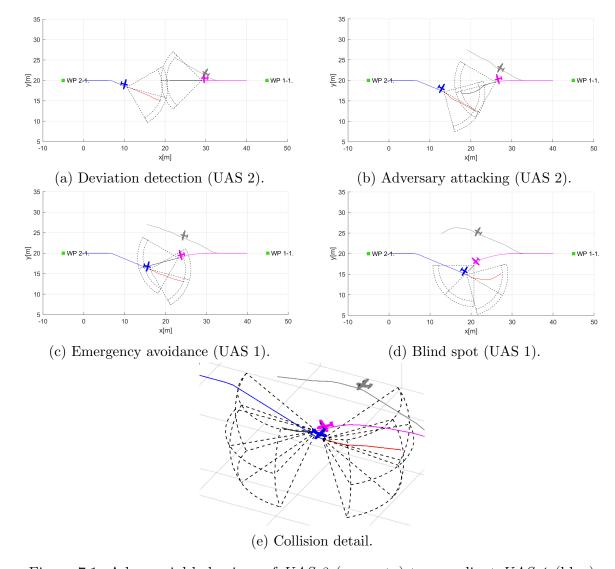


Figure 7.1: Adversarial behaviour of *UAS 2* (magenta) to compliant *UAS 1* (blue)

**Performance Parameters Evaluation:** Performance parameters (y-axis) are tracked over *UTM time* (x-axis). The evolution of *performance* (fig. ??) is tracking following parameters:

- 1. Expected crash distance (gray line) defined as (eq. ??) between UAS 1 (blue) and expected UAS 2 position (grey plane/line) over mission time  $t \in [0, 22]$ .
- 2. Crash distance (blue line) defined as (eq. ??) between UAS 1 (blue) and real UAS 2 position (magenta plane/line) over mission time  $t \in [0, 22]$ .
- 3. Safety margin (yellow line) constant value according to collision case (tab. ??) as value of 10 m. The safety margin is considered as soft constraint.
- 4. Body margin (red line) constant value according to (tab. ??) as value of 1.2 m. The body margin is considered as hard constraint. The breaking of body margin means an

effective collision UAS 1 and UAS 2.

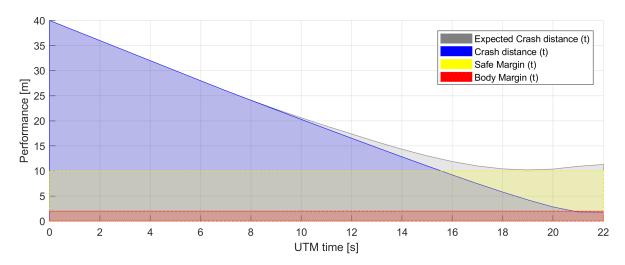


Figure 7.2: Expected/Real Distance to body/safety margin evolution for adversarial behaviour of UAS 2.

Safety criteria for both body and safety margins in case of expected behaviour are satisfied (eq. ??). This means that UAS 1 fulfilled the UTM directive despite the fact that it entered Emergency Avoidance Mode (fig. ??).

$$expectedDistanceToSafetyMargin(t) \ge 0, \quad \forall t \in [0, 22]$$

$$expectedDistanceToBodyMargin(t) \ge 0 \quad \forall t \in [0, 22]$$

$$(7.1)$$

Safety Margin is broken at UTM time 15 s, body margin is broken at UTM time 21 s, the collision happens at UTM time 22 s. This is summarized in Distance Condition Breach (eq. ??).

$$distance To Safety Margin(t) < 0, \qquad \forall t \in [21, 22]$$
$$distance To Body Margin(t) < 0 \qquad \forall t \in [15, 22]$$
 (7.2)

Note. An adversary behaviour needs to be addressed on:

- 1. UAS Traffic Management Level our UTM implementation failed to detect deviation (fig. ??) and start of attack (fig. ??). UAS 2 (magenta) had clean intention from beginning and did not change pursuit even when safety margin was breached.
- 2. Emergency Avoidance Level our navigation loop implementation does not consider the ill-intentions. The UAS 1 (blue) properly switched to Emergency avoidance mode (fig. ??) after detection of UAS2 (magenta). UAS 2 (magenta) then used the blind spot to exploit UAS 1 vulnerability.

## 7.5.3 Computation Footprint

The computation footprint is summarized in computation load (tab. ??). The computation load (eq. ??) was calculated for each time-frame in scenarios. There is summary of minimal, maximal, average and median values.

The computational load never exceed more than 55.95% in case of emergency Head On (eq. ??), which means that every path was calculated on time.

Scenario	Computation load				
Scenario	min.	max.	avg.	med.	
Building avoidance (fig. ??)	2.20%	27.40%	12.11%	13.20%	
Slalom (fig. ??)	12.20%	30.50%	21.42%	21.50%	
Maze (fig. ??)	24.90%	46.10%	31.51%	30.80%	
Storm (fig. ??)	2.60%	26.90%	11.57%	13.90%	
Emergency Converging (fig. ??)	2.75%	16.50%	5.84%	4.95%	
Emergency Head On (fig. ??)	3.90%	55.95%	13.19%	6.90%	
Emergency Multiple (fig. ??)	5.90%	52.35%	12.77%	8.56%	
Rule-based Converging (fig. ??)	3.60%	13.50%	7.32%	5.97%	
Rule-based Head on (fig. ??)	4.65%	41.60%	13.64%	9.30%	
Rule-based Multiple (fig. ??)	4.37%	23.30%	11.96%	10.93%	
Rule-based Overtake (fig. ??)	3.85%	13.40%	7.62%	6.70%	

Table 7.2: Computation load statistics for all test cases.

Following observations can be made:

- 1. Building avoidance, Slalom, and Maze scenarios the computation load is increasing with the amount of static obstacles. The average load for Emergency avoidance mode in clustered environment is 31.51% (Maze).
- 2. Storm scenario the overall computation load is very low due the moving constraint implementation (sec. ??).
- 3. Emergency Converging/Head On/Multiple scenarios the overall computation load is quite high due the ineffective body volume intersection (sec. ??) implementation.
- 4. Rule-based Converging/Head On/Multiple scenarios the median computational load is low, because of the linear rule implementation (sec. ??)
- 5. Rule-based Overtake the average computation load is very low, because only divergence/convergence (rule. ??) waypoints are calculated and UAS stays in navigation mode.