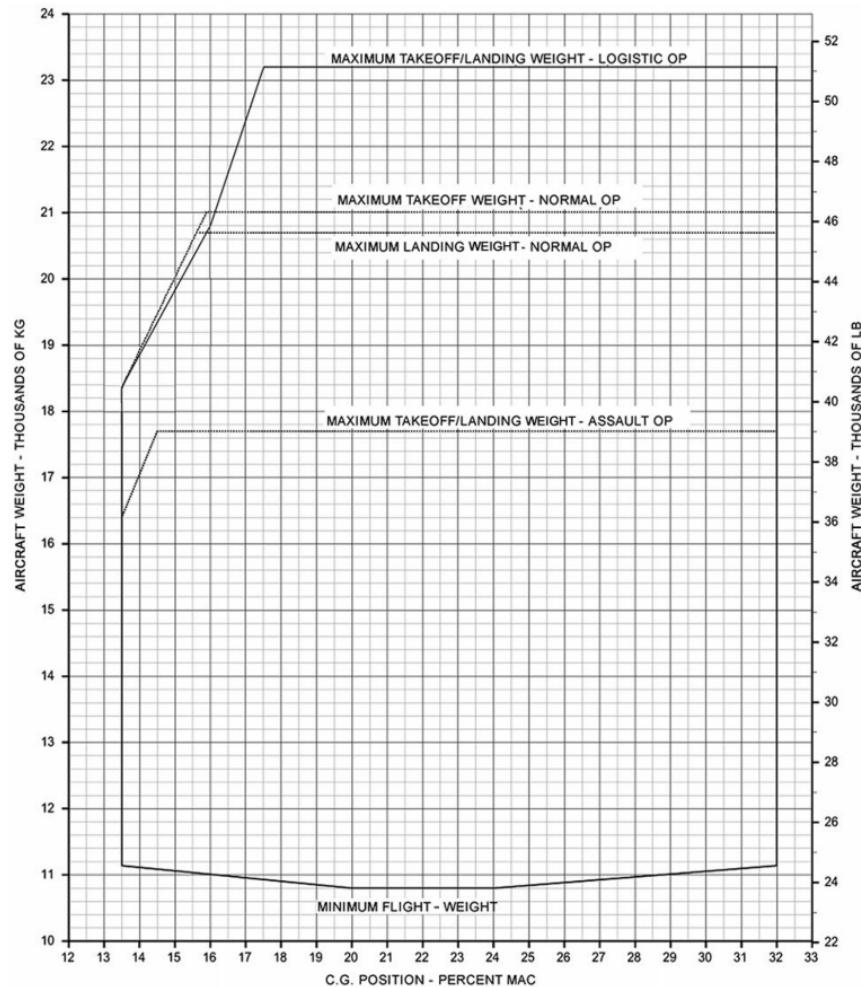


| REFERENCE DATA | | | | |
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| | | | | |
| | | | | |

LIMITATIONS

CG DIAGRAM

LONGITUDINAL DISPLACEMENT LIMITS



WEIGHT

| | LOGISTIC ROLE | NORMAL ROLE | ASSAULT ROLE |
|---|-------------------------|------------------------|-----------------------|
| MTW (Kg) | 23 250 | 21 050 | --- |
| MTOW (Kg) | 23 200 | 21 000 | 17 700 |
| MZFW (Kg) | 20 700 | 18 500 | 16 500 |
| MLW (Kg) (Max Sink Rate) | 23 200 (540 fpm) | 20 700 (600 fpm) | --- |
| MAX. LOAD FACTOR | -0.85 g up to 2.25 g | -1.0 g up to 2.50 g | -1.0 g up to 3.0 g |

ELECTRICAL

Maximum permissible continuous load on each generator is 400 Amps.

Generators may not be operated with loads greater than 400 Amps for periods longer than 5 minutes.

| | | |
|-----------------------------------|-------|-------|
| GPU Voltage Limits (AC) | MIN ➤ | 110 V |
| | MAX ➤ | 118 V |
| Minimum power of the GPU: 10 KVA. | | |

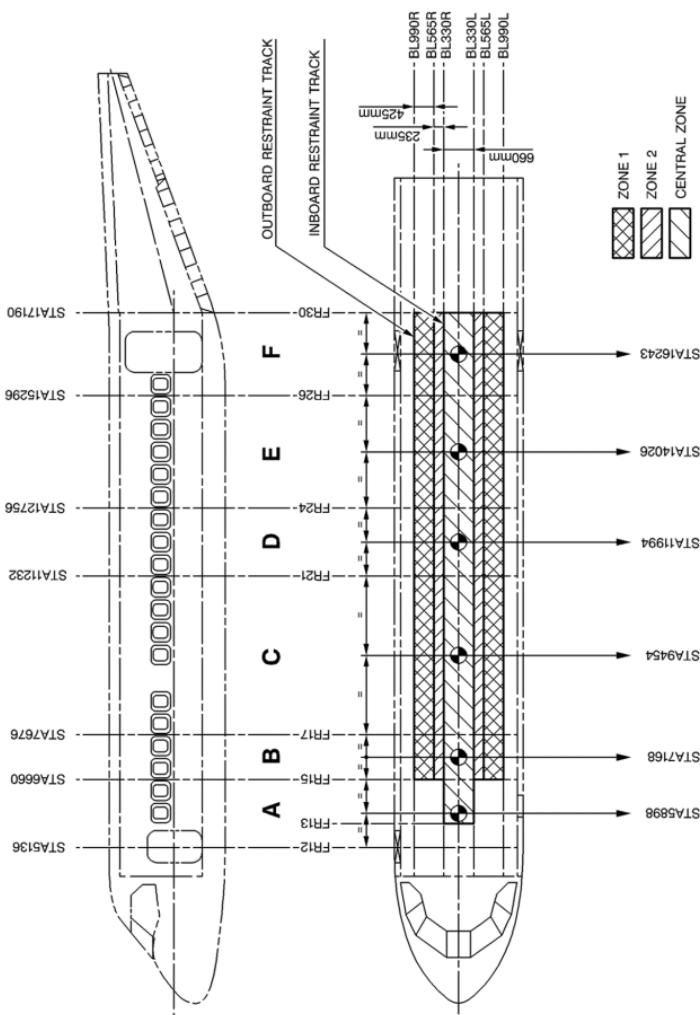
HYDRAULIC

When approved hydraulic fluid MIL-H-83282 (red) is used, whether it is mixed or not, lowest ambient temperature allowed is -40 °C.

DOORS

Refer to C/M-1 and /or C/M-2 for permission before opening any door.

CARGO CABIN ZONES

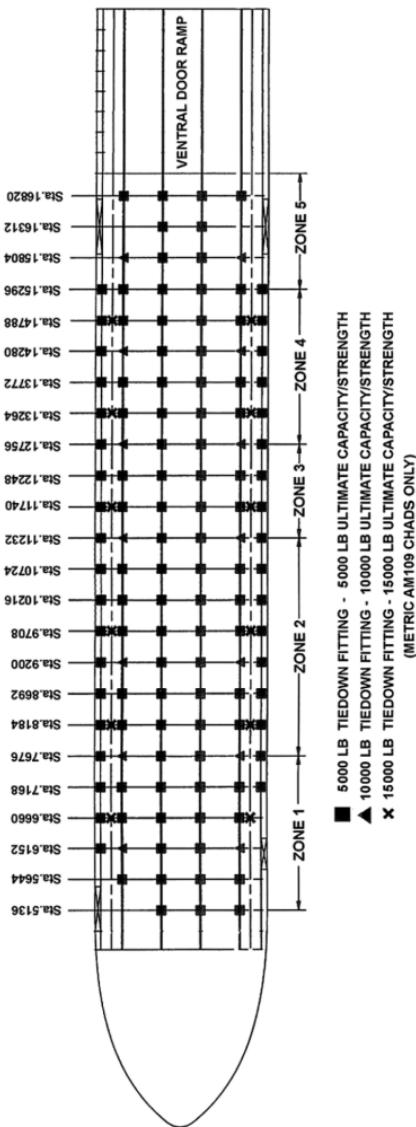


CARGO CABIN ZONES (cont.)

| ZONE | ZONE LIMITS Fus.Sta. to Fus.Sta. | Max. Linear Load | | Max. Permissible Load | |
|------|--|------------------|---------|-----------------------|------|
| | | kg / m | lb / in | kg | lb |
| I | 5.136 to 7.676 m (202.20 to 302.20 in.) | 1000 | 56.00 | 2540 | 5600 |
| II | 7.676 to 11.232 m (302.20 to 442.20 in.) | 1100 | 61.60 | 3912 | 8630 |
| III | 11.232 to 12.756 m (442.20 to 502.20 in.) | 1300 | 72.80 | 1981 | 4370 |
| IV | 12.756 to 15.296 m (502.20 to 602.20 in.) | 1100 | 61.60 | 2794 | 6160 |
| V | 15.296 to 17.190 m (602.20 to 676.77 in.) | 1000 | 56.00 | 1894 | 4180 |
| Ramp | 17.190 to 20.232 m (676.77 to 796.53 in.) | - | - | 1000 | 2210 |

NOTE: Lb-values above are rounded off to the nearest 10 lb.

TIE DOWN POINTS



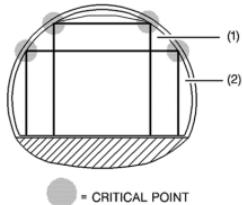
WINCH**LOADING CAPABILITIES DEPENDING ON RAMP SLOPE**

| | | | | | | |
|--|-----|-----|-----|-----|-----|-----|
| RAMP CREST HEIGHT (cm) | 115 | 120 | 125 | 130 | 136 | 140 |
| RAMP SLOPE (DEGREES) | 15° | 16° | 17° | 18° | 19° | 20° |
| NOTE: Ramp slope values are round-off to the nearest value in degrees. | | | | | | |

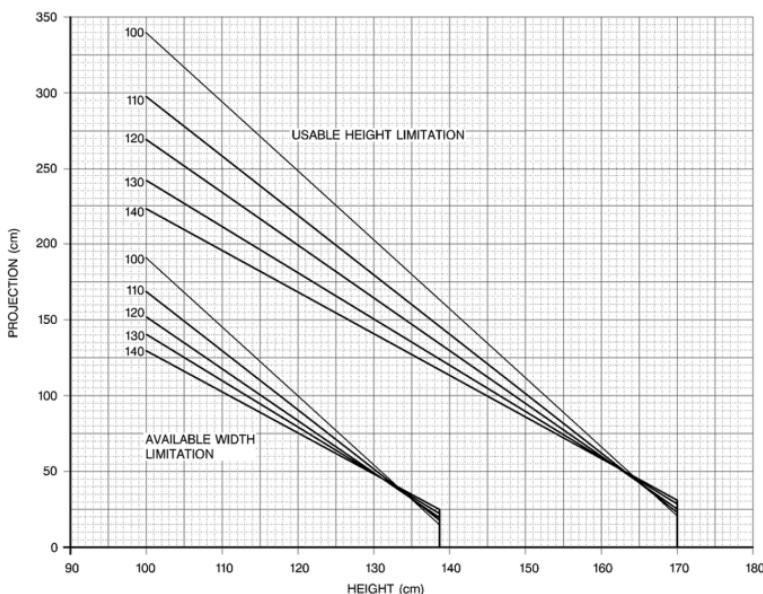
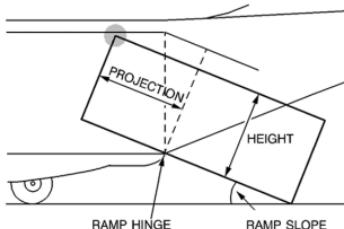
| ROLLING SUPPORT | FRICTION COEFFICIENT (FOR RAMP IN HORIZONTAL POSITION) | MAXIMUM WINCH CAPACITY (kg) IAW RAMP SLOPE | | | | | |
|--|---|---|------|------|------|------|------|
| | | 15° | 16° | 17° | 18° | 19° | 20° |
| SOLID TIRES (HARD RUBBER OR METALLIC) | 0.018 | 3620 | 3414 | 3230 | 3066 | 2919 | 2786 |
| ROLLER CONVEYORS | 0.020 | 3595 | 3391 | 3210 | 3048 | 2903 | 2772 |
| PNEUMATIC TIRES | 0.030 | 3475 | 3284 | 3115 | 2963 | 2825 | 2701 |

LOADS

OVERHEAD CRATED CARGO PROJECTION



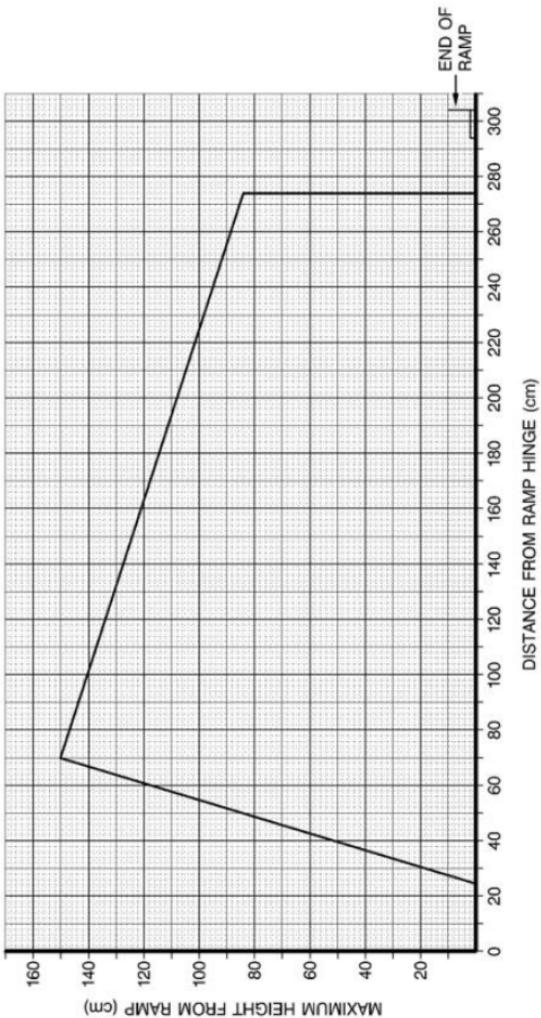
FLOOR HEIGHT AT
RAMP CREST (cm)



NOTE: LIMITATIONS OVER CABIN FLOOR

- (1) USABLE HEIGHT LIMITATION
- (2) AVAILABLE WIDTH LIMITATION

HEIGHT LIMITATION FOR LOADS ON THE RAMP

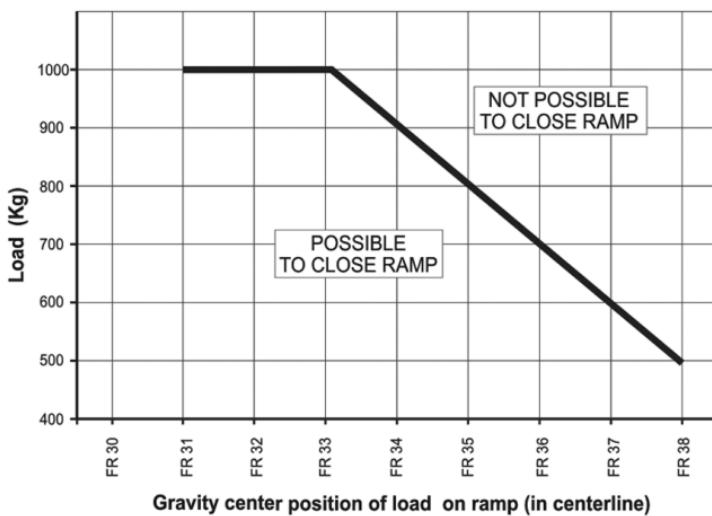


NOTE: DISTANCE FROM RAMP HINGE WITH THE RAMP OPENED, AND MEASURED FROM THE APP END OF THE CARGO CABIN FLOOR, SO LOAD DOES NOT PROTRUDE INTO THE CARGO HOLD WHEN CLOSING THE RAMP.

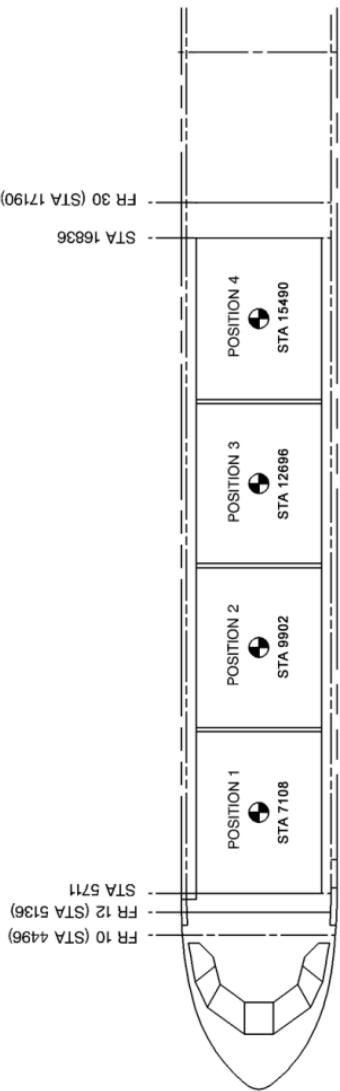
NOTE: FOR LOADS ON HCU-6/E PALLETS, SUBTRACT 11 CM FROM HEIGHT READING ON GRAPHIC.

NOTE: FOR LOADS ON THE INTERMEDIATE CONVEYOR TRAYS, SUBTRACT 5 CM FROM HEIGHT READING ON GRAPHIC.

LOAD CAPACITY ON RAMP



PALLET POSITIONS



CENTROIDS NOTED WITH PALLETS IN CENTER POSITION

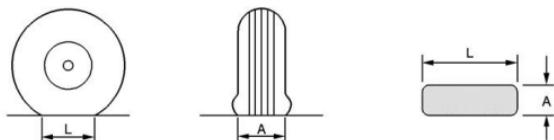
GAP BETWEEN PALLETS IN CONSECUTIVE POSITIONS 5cm (2 inches)

WITH A COMPLETE LOAD, PALLETS MAY BE LOCKED IN CENTROIDS SHOWN
+ 25.4 (10 inches)

WITH A LOAD OF LESS THAN 4 PALLETS ON THE CARGO CABIN,
PALLETS MAY BE LOCKED EVERY 25.4 cm (10 inches)

CALCULATING CONTACT PRESSURE AND SHORING EFFECT ON PNEUMATIC TIRES

CONTACT SURFACE AND WEIGHT PER WHEEL



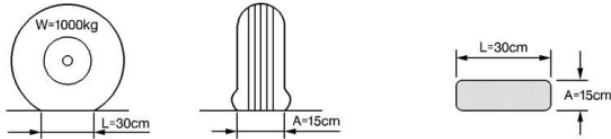
STEP 1: PLACE THE VEHICLE ON A FLAT SURFACE AND MARK WHEEL BORDER, AS SHOWN. MOVE THE VEHICLE.

STEP 2: CONTACT SURFACE WILL HAVE AN ELLIPTICAL SHAPE; MEASURE DIMENSIONS "L" AND "A".

STEP 3: COMPUTE CONTACT SURFACE BY MEANS OF THE FORMULA: $S = L \times A \times 0.785$.

STEP 4: COMPUTE WEIGHT PER WHEEL: $\text{WEIGHT PER WHEEL} = \frac{\text{WEIGHT ON AXLE}}{\text{NUMBER OF WHEELS}}$

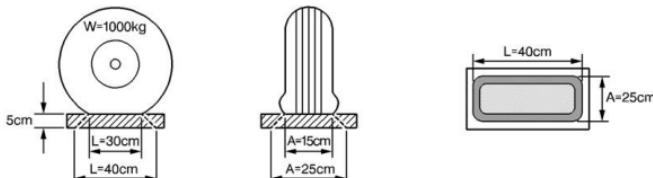
CONTACT PRESSURE WITHOUT SHORING



$$S = L \times A \times 0.785; S = 30 \times 15 \times 0.785 = 353.25 \text{ ROUND OFF} = 353 \text{ cm}^2$$

$$\text{PRESSURE} = \frac{W}{S}, P = \frac{1000}{353} = 2.83 \text{ ROUND OFF} = 2.9 \text{ Kg/cm}^2$$

CONTACT PRESSURE WITH SHORING

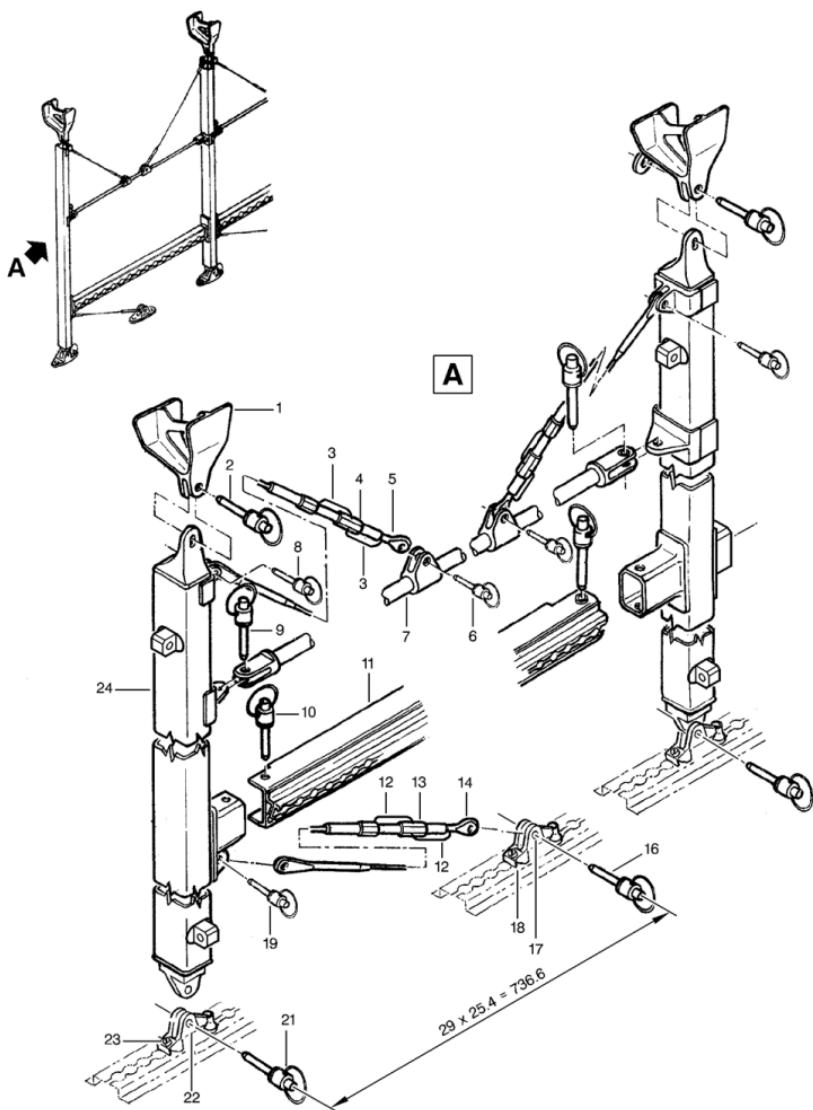


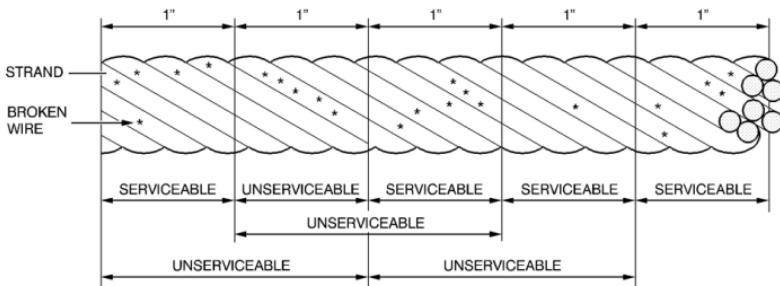
$$S = L \times A \times 0.785; S = 40 \times 25 \times 0.785 = 785 \text{ cm}^2$$

$$\text{PRESSURE} = \frac{W}{S}, P = \frac{1000}{785} = 1.27 \text{ ROUND OFF} = 1.3 \text{ Kg/cm}^2$$

C.L.Q.R.H. C-295M-VT01

INSTALLATION/REMOVAL OF CENTER SEAT SUPPORT FRAME

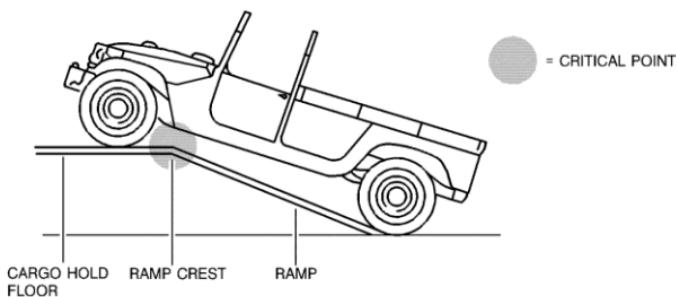
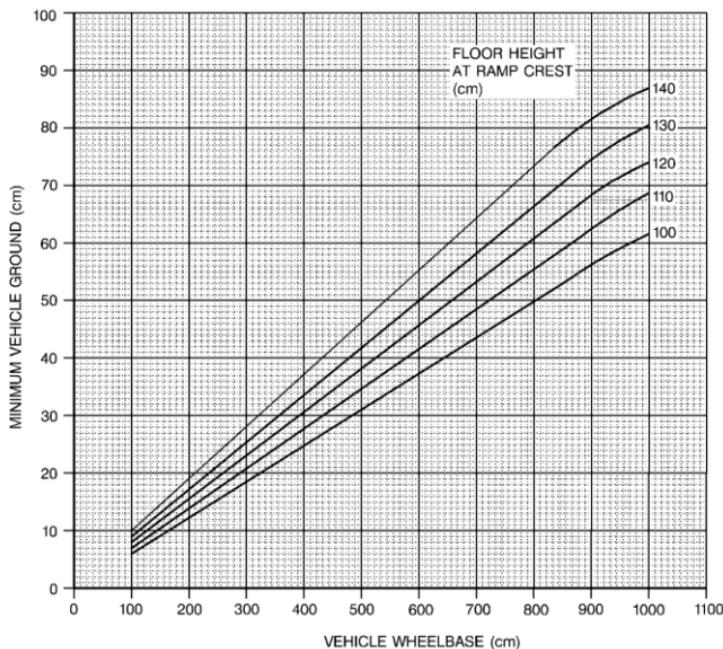


CABLE DAMAGE TOLERANCE

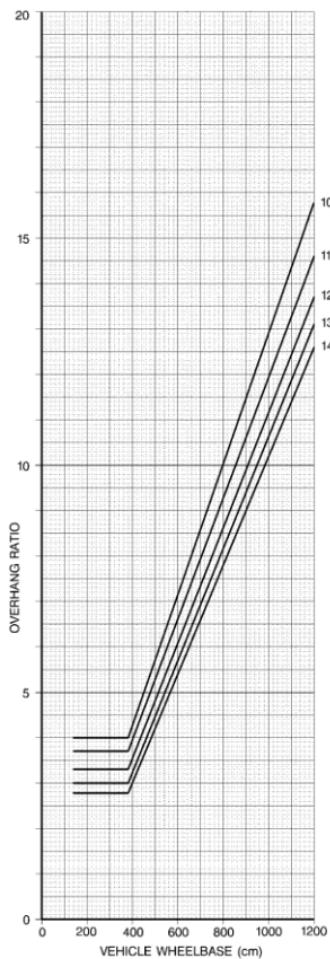
AIRDROP ANCHOR CABLES MUST
NOT HAVE KINKS, NOR MORE THAN SIX
BROKEN WIRES PER INCH, NEVER IN TWO
CONSECUTIVE INCHES, NOR MORE THAN
THREE BROKEN WIRES IN THE SAME STRAND

VEHICLES

RAMP CREST CONTACT

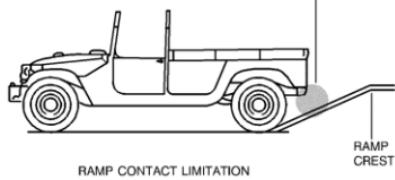


OVERHANG RAMP AND GROUND CONTACT LIMITATION

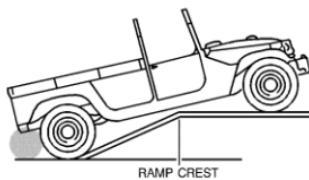


FLOOR HEIGHT
AT RAMP CREST
(cm)

MAXIMUM OVERHANG
RATIO WITH RAMP
AT MAXIMUM SLOPE IS 2.75

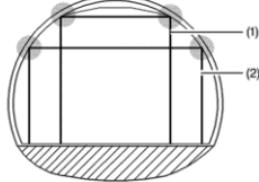
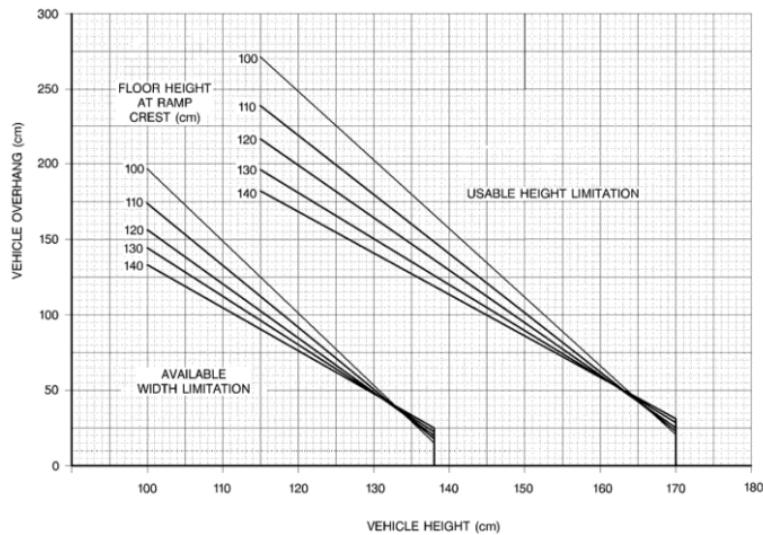


RAMP CREST
= CRITICAL POINT

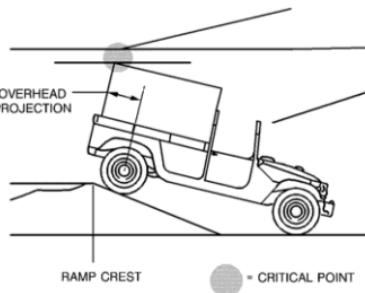


RAMP CREST
GROUND CONTACT LIMITATION

VEHICLE OVERHEAD PROJECTION



(1) USABLE HEIGHT LIMITATION
 (2) USABLE WIDTH LIMITATION



CARGO FLOOR – ON GROUND CAPACITIES

| | CARGO CABIN | | | | | | RAMP | |
|---|-------------|-------|-------|-------|-------|-------|-------|-------|
| COMPARTMENT | A | B | C | D | E | F | R1 | R2 |
| FRAME | 12 | 15 | 17 | 21 | 24 | 26 | 30 | 34 |
| STA | 5136 | 6680 | 7876 | 11232 | 12756 | 15286 | 17190 | 18542 |
| LENGTH (mm) | 1524 | 1016 | 3566 | 1524 | 2540 | 1894 | 1352 | 1352 |
| WIDTH (mm) | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 |
| AREA (cm ²) | 36037 | 24038 | 84134 | 36057 | 60096 | 44812 | 31988 | 31988 |
| MAXIMUM INDIVIDUAL CAPACITY (kg) | 6096 | 4064 | 15644 | 7624 | 11176 | 7576 | 4000 | 4000 |
| MAXIMUM LINEAR CAPACITY (kg/linear meter) | 4000 | 4000 | 4400 | 5200 | 4400 | 4000 | | |

| | COMPARTMENT | | | | | | | |
|-------------------------------|-----------------------------------|--|------|------|------|------|------|-------|
| UNITS | A | B | C | D | E | F | R1 | R2 |
| LATERAL STRIPS | AREA MORE THAN 80 CM ² | 2.1 | 2.1 | 2.1 | 2.1 | 2.1 | 2.1 | 1.7 |
| CONCENTRATED LOADS | AREA LESS THAN 80 CM ² | Kg/cm ² | 7.7 | 7.7 | 7.7 | 7.7 | 7.7 | 5.7 |
| | AREA MORE THAN 80 CM ² | | 1.7 | 1.7 | 1.7 | 1.7 | 1.7 | 1.7 |
| | AREA LESS THAN 80 CM ² | | 5.7 | 5.7 | 5.7 | 5.7 | 5.7 | 5.7 |
| CENTRAL STRIP | | | | | | | | |
| LATERAL STRIPS | LATERAL STRIPS | Kg/lineal meter | 4800 | 7200 | 7200 | 7200 | 4800 | |
| BULK LOADS | RUNNING LOAD PER PANEL | CENTRAL STRIP | 2700 | 2700 | 3000 | 3000 | 2700 | |
| | LATERAL AND CENTRAL STRIPS | | 5400 | 6600 | 6600 | 6600 | 5400 | |
| | | PNEUMATIC TIRES: MAXIMUM INTERNAL PRESSURE 7.03 kg/cm ² (100 PSI, 6.89 BAR), MINIMUM DISTANCE BETWEEN AXLES 90 cm | | | | | | |
| WHEELED LOADS | WEIGHT PER AXLE ON LATERAL STRIPS | ZONE 1 | Kg | 1900 | 5200 | 5200 | 3800 | 3200 |
| | ZONE 2 | | Kg | 1900 | 2800 | 2800 | 2800 | 3200 |
| | TONGUE LOAD ON CENTRAL STRIP | | Kg | 500 | 500 | 500 | 500 | 500 |
| PALLETTIZED LOADS AM109 CHADS | | | | | | | | |

SEE IN-FLIGHT CAPACITIES

LOADING/OFFLOADING CAPACITIES (ON THE GROUND)

WARNING: FIGURES SHOWN INDICATES FLOOR LOADING CAPACITIES ONLY.
AIRCRAFT WEIGHT AND BALANCE MUST BE COMPUTED IN ALL CASES, AND CG LIMITS MUST BE OBSERVED.

NOTE: Cargo floor between FR10 (STA 4496) and FR12 (STA 5136) is not usable for loading purposes.

CARGO FLOOR – IN FLIGHT CAPACITIES

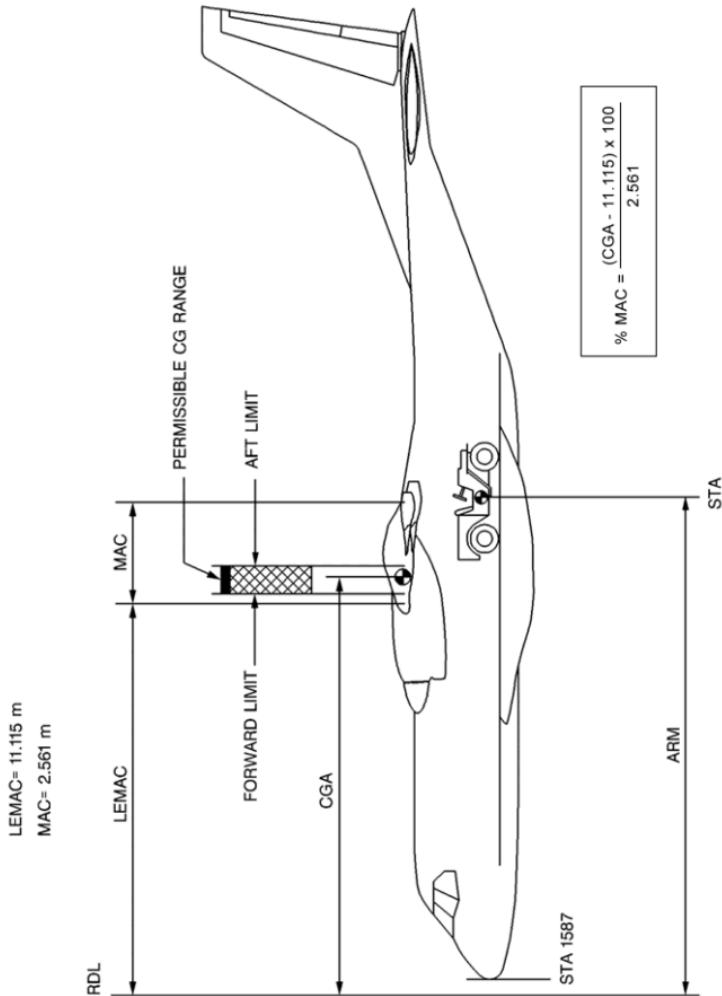
| COMPARTMENT | CARGO CABIN | | | | | | RAMP | | |
|---|-----------------------------------|-------|-------|-------|-------|-------|-------|-------|-------|
| | A | B | C | D | E | F | R1 | R2 | |
| STA | 5136 | 6660 | 7676 | 11232 | 12756 | 15296 | 17190 | 18542 | 19894 |
| LENGTH (mm) | 1524 | 1016 | 3556 | 1524 | 2540 | 1894 | 1352 | | |
| WIDTH (mm) | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 | 2366 | |
| AREA (cm ²) | 24038 | 54134 | 39057 | 50096 | 44812 | 31988 | | | |
| MAXIMUM INDIVIDUAL CAPACITY (kg) | 1524 | 1016 | 3911 | 1981 | 2794 | 1894 | 1000 | | |
| MAXIMUM LINEAR CAPACITY (kg/linear meter) | 1000 | 1000 | 1100 | 1300 | 1100 | 1000 | 1000 | 1000 | |
| COMPARTMENT | | | | | | | | | |
| UNITS | A | B | C | D | E | F | R1 | R2 | |
| LATERAL STRIPS | AREA MORE THAN 80 CM ² | | 1.2 | 1.2 | 1.2 | 1.2 | 1.2 | 1.2 | 1.0 |
| CENTRAL STRIP | AREA LESS THAN 80 CM ² | | 4.4 | 4.4 | 4.4 | 4.4 | 4.4 | 4.4 | 3.3 |
| BULK LOADS | AREA MORE THAN 80 CM ² | | 1.0 | 1.0 | 1.0 | 1.0 | 1.0 | 1.0 | 1.0 |
| | AREA LESS THAN 80 CM ² | | 3.3 | 3.3 | 3.3 | 3.3 | 3.3 | 3.3 | 3.3 |
| PNEUMATIC TIRES. MAXIMUM INTERNAL PRESSURE 703 kg/cm ² (100 PSI, 6.89 BAR). MINIMUM DISTANCE BETWEEN AXLES 90 cm | | | | | | | | | |
| WEIGHT PER AXLE ON LATERAL STRIPS | ZONE 1 | Kg | 800 | 2275 | 2275 | 2275 | 2275 | 1600 | 500 |
| TONGUE LOAD ON CENTRAL STRIP | ZONE 2 | Kg | 800 | 1000 | 1000 | 1000 | 1000 | 500 | 500 |
| PALLETIZED LOADS | ROLLER STATION | | Kg | 200 | 200 | 200 | 200 | 200 | 200 |
| AM109 CHADS (4 ROWS OF ROLLERS) | WEIGHT PER HC10-E PALLETT | | Kg | 300 | 300 | 300 | 300 | 300 | 300 |
| IN-FLIGHT CAPACITIES | | | | | | | | | |

WARNING: FIGURES SHOWN INDICATES FLOOR LOADING CAPACITIES ONLY.
AIRCRAFT WEIGHT AND BALANCE MUST BE COMPUTED IN ALL CASES, AND CG LIMITS MUST BE OBSERVED.

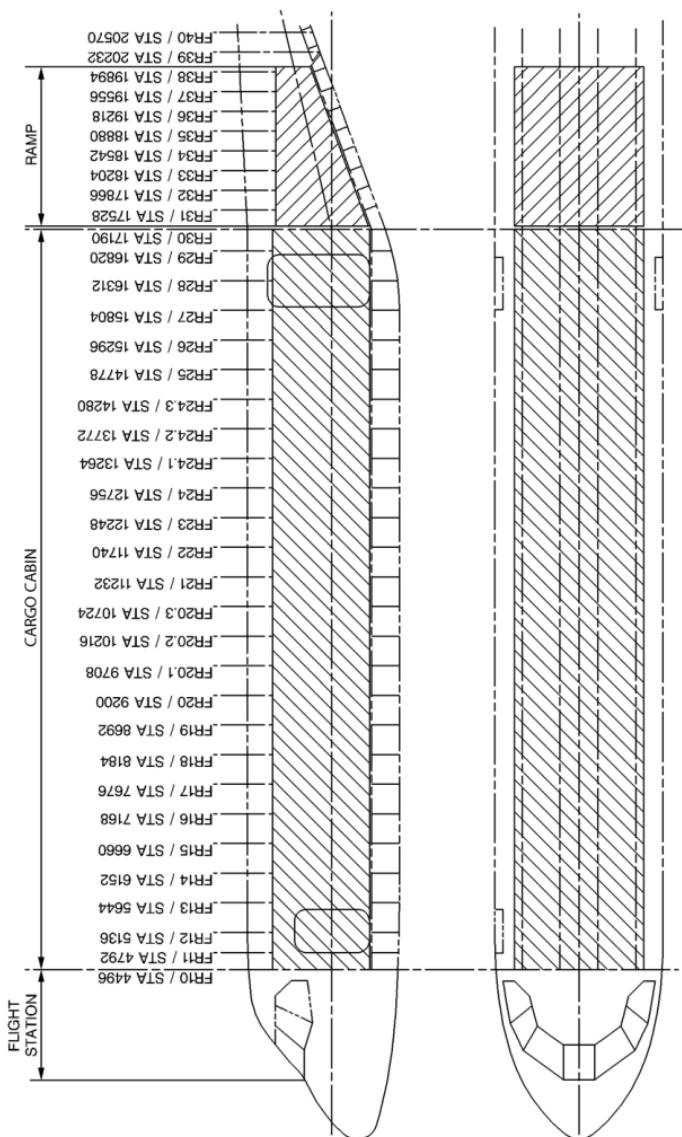
NOTE: Cargo floor between FR10 (STA 4496) and FR12 (STA 5136) is not usable for loading purposes.

CARGO CABIN DATA

MAC / LEMAC

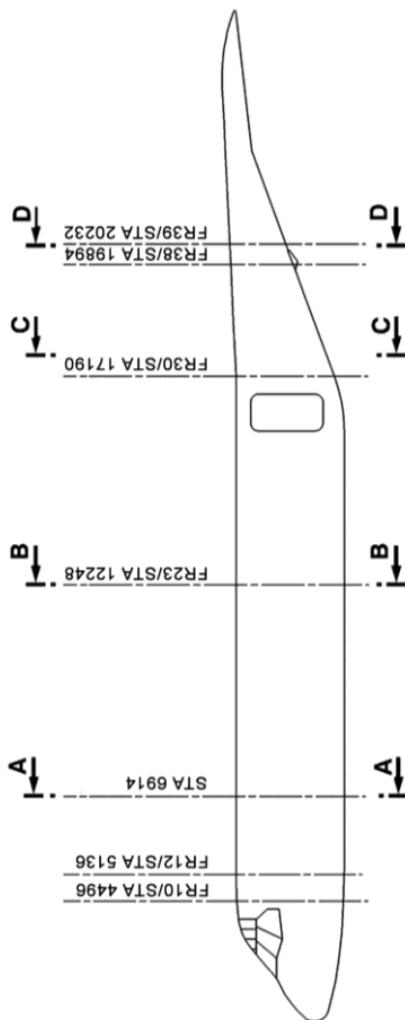


CARGO CABIN STATIONS

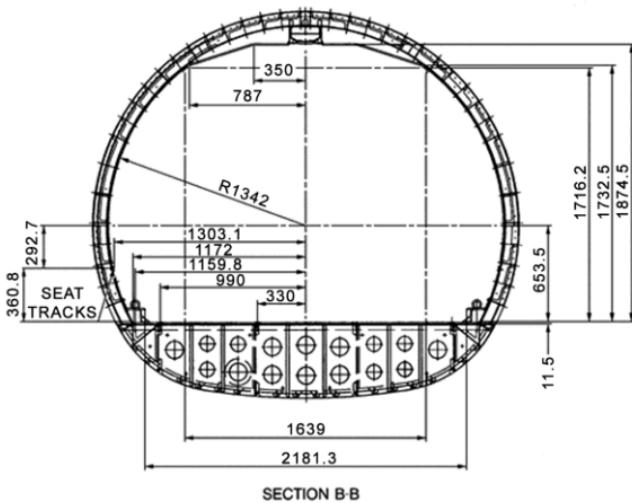
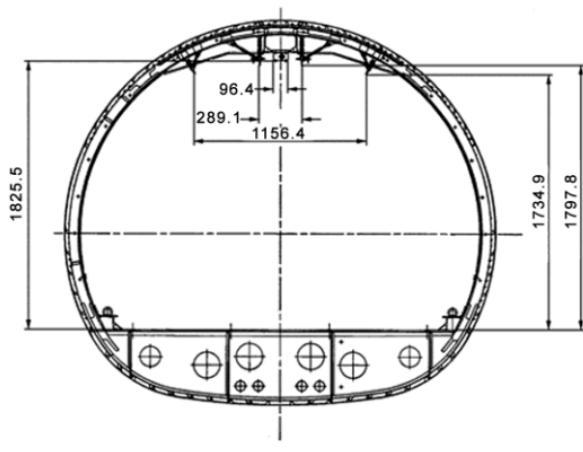


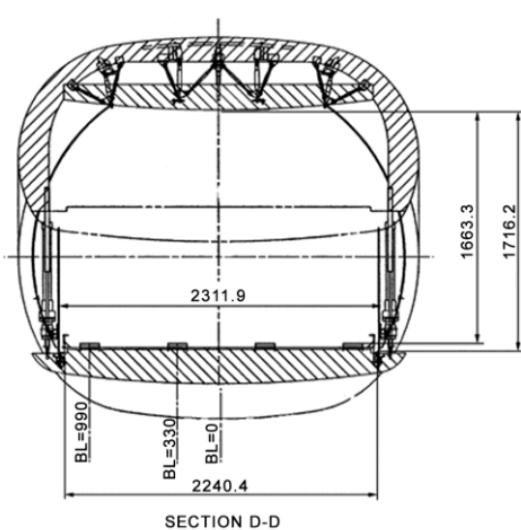
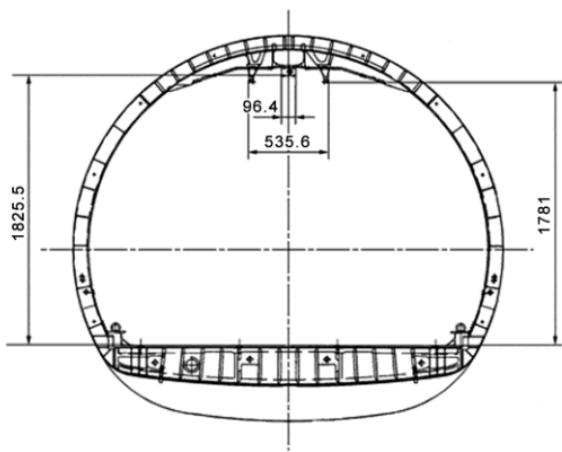
CARGO CABIN DIMENSIONS

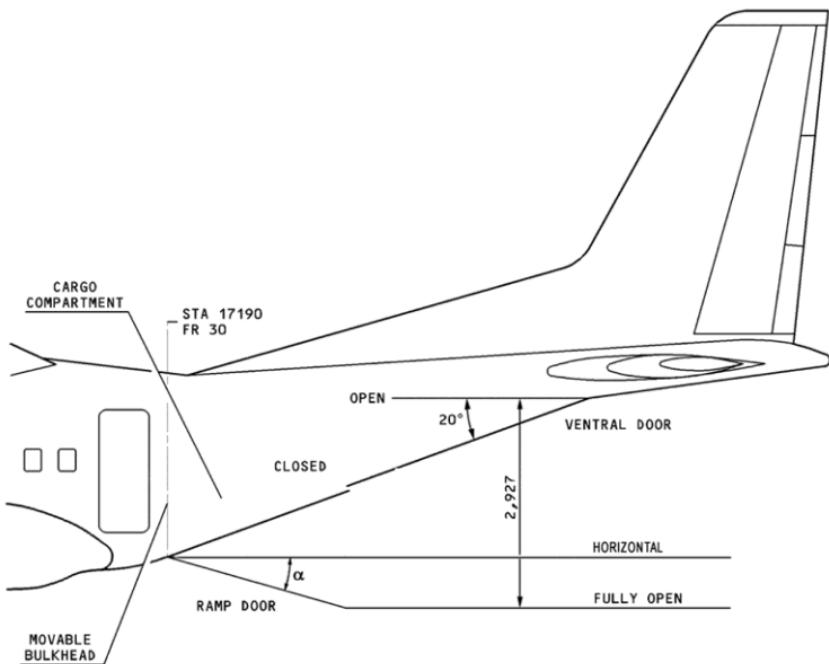
SECTIONS



SECTIONS (cont.)

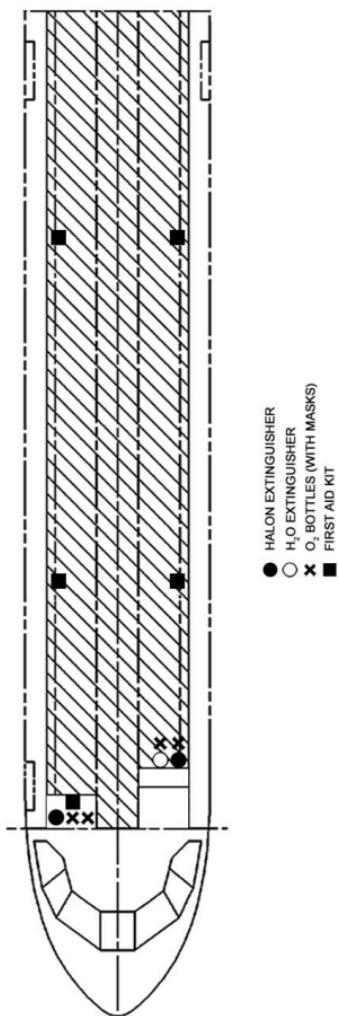


SECTIONS (cont.)

RAMP

| α (°) | Weight (kg) | CG (MAC) |
|--------------|-------------|----------|
| 15.1 | 23200 | 32% |
| 15.8 | 23200 | 17.6% |
| 18.9 | 11140 | 32% |
| 19.8 | 11140 | 13.5% |

EMERGENCY EQUIPMENT



OXYGEN

When flying above FL250 each crew-member oxygen mask must be connected to its oxygen socket, and masks must be immediately available.

OXYGEN DURATION (MANHOURS) – FIXED GASEOUS OXYGEN INSTALLATION

| ↓ CABIN ALTITUDE | COCKPIT MANOMETER READING (PSI) ↓ | | | | |
|---|-----------------------------------|------|------|------|------|
| | 1850 | 1500 | 1200 | 1000 | 800 |
| Fixed installation supplying diluted oxygen (NOR air/oxygen mixture). | | | | | |
| 5000 | 5.69 | 3.96 | 2.47 | 1.48 | 0.49 |
| 10000 | 4.52 | 3.14 | 1.96 | 1.18 | 0.39 |
| 15000 | 4.03 | 2.80 | 1.75 | 1.05 | 0.35 |
| 20000 | 3.73 | 2.59 | 1.62 | 0.97 | 0.32 |
| 25000 | 3.11 | 2.16 | 1.35 | 0.81 | 0.27 |
| Fixed installation supplying pure oxygen (100%). | | | | | |
| 5000 | 1.02 | 0.71 | 0.44 | 0.27 | 0.09 |
| 10000 | 1.25 | 0.87 | 0.54 | 0.33 | 0.11 |
| 15000 | 1.57 | 1.09 | 0.68 | 0.41 | 0.14 |
| 20000 | 1.96 | 1.37 | 0.85 | 0.51 | 0.17 |
| 25000 | 2.54 | 1.76 | 1.10 | 0.66 | 0.22 |

- NOTES:
- Duration in hours, to be divided by the number of users.
 - Oxygen pressure reading taken at 15°C (59°F) oxygen temperature.
 - Average breathing rate: 20 litres/minute BTPS.
 - Portable oxygen installation not taken into account.
 - 390 litres NTPD taken as reserve for fire fighting at 10000 ft.
 - Minimum bottle pressure at 200 psi.

OPERATIONAL ITEMS WEIGHT & MOMENTCREW WEIGHT & MOMENT DATA

| Crew | Weight (kg) (lb) | | Arm - H (m) | Moment (kg x m) |
|---------------|---------------------|-----|----------------|--------------------|
| Pilot | 70 | 154 | 3.549 | 248 |
| | 75 | 165 | 3.549 | 266 |
| | 80 | 176 | 3.549 | 284 |
| | 85 | 187 | 3.549 | 302 |
| | 90 | 198 | 3.549 | 319 |
| Copilot | 70 | 154 | 3.549 | 248 |
| | 75 | 165 | 3.549 | 266 |
| | 80 | 176 | 3.549 | 284 |
| | 85 | 187 | 3.549 | 302 |
| | 90 | 198 | 3.549 | 319 |
| Fwd. Observer | 70 | 154 | 4.336 | 304 |
| | 75 | 165 | 4.336 | 325 |
| | 80 | 176 | 4.336 | 347 |
| | 85 | 187 | 4.336 | 369 |
| | 90 | 198 | 4.336 | 390 |
| Load Master | 70 | 154 | 4.824 | 338 |
| | 75 | 165 | 4.824 | 362 |
| | 80 | 176 | 4.824 | 386 |
| | 85 | 187 | 4.824 | 410 |
| | 90 | 198 | 4.824 | 434 |

BAGGAGE WEIGHT & MOMENT DATA

| Situation | Weight (kg) (lb) | | Arm - H (m) | Moment (kg x m) |
|-----------|---------------------|----|----------------|--------------------|
| Pilot | 14 | 30 | 4.068 | 55 |
| Copilot | 14 | 30 | 4.068 | 55 |

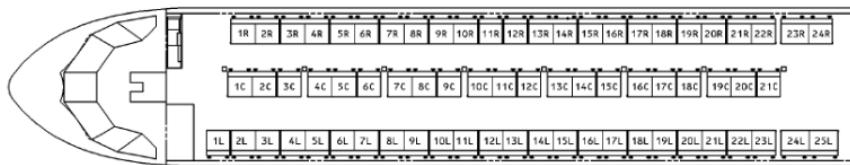
DOCUMENTATION WEIGHT & MOMENT DATA

| Situation | Weight (kg) (lb) | | Arm - H (m) | Moment (kg x m) |
|-----------|---------------------|-----|----------------|--------------------|
| Pilot | 1 | 2.2 | 3.705 | 3.7 |
| Copilot | 1 | 2.2 | 3.705 | 3.7 |

EMERGENCY RACK (MAX. LOAD)

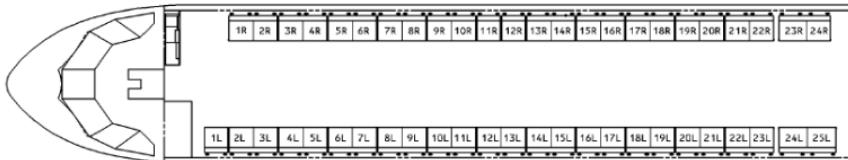
| Position | Weight (kg) (lb) | | Arm - H (m) | Moment (kg x m) |
|----------------|---------------------|-----|----------------|--------------------|
| Emergency Rack | 74 | 163 | 4.644 | 341 |

TROOP TRANSPORT CONFIGURATION



| Seat Nr. | Arm-H (m) | Moment (kg x m) | Seat Nr. | Arm-H (m) | Moment (kg x m) | Seat Nr. | Arm-H (m) | Moment (kg x m) |
|----------|-----------|-----------------|----------|-----------|-----------------|----------|-----------|-----------------|
| 1L | 5.494 | 467 | 1C | 5.891 | 501 | 1R | 5.993 | 509 |
| 2L | 5.993 | 509 | 2C | 6.285 | 534 | 2R | 6.387 | 543 |
| 3L | 6.387 | 543 | 3C | 6.744 | 573 | 3R | 6.907 | 587 |
| 4L | 6.907 | 587 | 4C | 7.364 | 626 | 4R | 7.301 | 621 |
| 5L | 7.301 | 621 | 5C | 7.758 | 659 | 5R | 7.821 | 665 |
| 6L | 7.821 | 665 | 6C | 8.247 | 701 | 6R | 8.215 | 698 |
| 7L | 8.215 | 698 | 7C | 8.837 | 751 | 7R | 8.736 | 743 |
| 8L | 8.736 | 743 | 8C | 9.231 | 785 | 8R | 9.130 | 776 |
| 9L | 9.130 | 776 | 9C | 9.720 | 826 | 9R | 9.650 | 820 |
| 10L | 9.650 | 820 | 10C | 10.311 | 876 | 10R | 10.044 | 854 |
| 11L | 10.044 | 854 | 11C | 10.705 | 910 | 11R | 10.533 | 895 |
| 12L | 10.565 | 898 | 12C | 11.193 | 951 | 12R | 10.991 | 934 |
| 13L | 10.959 | 932 | 13C | 11.784 | 1002 | 13R | 11.480 | 976 |
| 14L | 11.480 | 976 | 14C | 12.178 | 1035 | 14R | 11.874 | 1009 |
| 15L | 11.874 | 1009 | 15C | 12.667 | 1077 | 15R | 12.393 | 1053 |
| 16L | 12.393 | 1053 | 16C | 13.257 | 1127 | 16R | 12.787 | 1087 |
| 17L | 12.787 | 1087 | 17C | 13.651 | 1160 | 17R | 13.307 | 1131 |
| 18L | 13.308 | 1131 | 18C | 14.140 | 1202 | 18R | 13.702 | 1165 |
| 19L | 13.702 | 1165 | 19C | 14.730 | 1252 | 19R | 14.222 | 1209 |
| 20L | 14.222 | 1209 | 20C | 15.124 | 1286 | 20R | 14.616 | 1242 |
| 21L | 14.616 | 1242 | 21C | 15.613 | 1327 | 21R | 15.137 | 1287 |
| 22L | 15.137 | 1287 | | | | 22R | 15.531 | 1320 |
| 23L | 15.531 | 1320 | | | | 23R | 16.115 | 1370 |
| 24L | 16.115 | 1370 | | | | 24R | 16.509 | 1403 |
| 25L | 16.509 | 1403 | | | | | | |

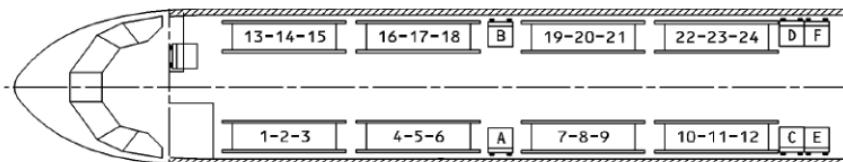
NOTE: In this table, the weight established for each trooper is 85 kg (187 lb), and is applied to each seat.

PARATROOP TRANSPORT CONFIGURATION

| Seat Nr. | Weight (kg) | Arm-H (m) | Moment (kg x m) | Seat Nr. | Weight (kg) | Arm-H (m) | Moment (kg x m) |
|----------|-------------|-----------|-----------------|----------|-------------|-----------|-----------------|
| 1L | 120 | 5.494 | 659 | 1R | 120 | 5.993 | 719 |
| 2L | 120 | 5.993 | 719 | 2R | 120 | 6.387 | 766 |
| 3L | 120 | 6.387 | 766 | 3R | 120 | 6.907 | 829 |
| 4L | 120 | 6.907 | 829 | 4R | 120 | 7.301 | 876 |
| 5L | 120 | 7.301 | 876 | 5R | 120 | 7.821 | 939 |
| 6L | 120 | 7.821 | 939 | 6R | 120 | 8.215 | 986 |
| 7L | 120 | 8.215 | 986 | 7R | 120 | 8.736 | 1048 |
| 8L | 120 | 8.736 | 1048 | 8R | 120 | 9.130 | 1096 |
| 9L | 120 | 9.130 | 1096 | 9R | 120 | 9.650 | 1158 |
| 10L | 120 | 9.650 | 1158 | 10R | 120 | 10.044 | 1205 |
| 11L | 120 | 10.044 | 1205 | 11R | 120 | 10.565 | 1268 |
| 12L | 120 | 10.565 | 1268 | 12R | 120 | 10.959 | 1315 |
| 13L | 120 | 10.959 | 1315 | 13R | 120 | 11.480 | 1378 |
| 14L | 120 | 11.480 | 1378 | 14R | 120 | 11.874 | 1425 |
| 15L | 120 | 11.874 | 1425 | 15R | 120 | 12.393 | 1487 |
| 16L | 120 | 12.393 | 1487 | 16R | 120 | 12.787 | 1534 |
| 17L | 120 | 12.787 | 1534 | 17R | 120 | 13.308 | 1597 |
| 18L | 120 | 13.308 | 1597 | 18R | 120 | 13.702 | 1644 |
| 19L | 120 | 13.702 | 1644 | 19R | 120 | 14.222 | 1707 |
| 20L | 120 | 14.222 | 1707 | 20R | 120 | 14.616 | 1774 |
| 21L | 120 | 14.616 | 1754 | 21R | 120 | 15.137 | 1816 |
| 22L | 120 | 15.137 | 1816 | 22R | 120 | 15.531 | 1864 |
| 23L | 120 | 15.531 | 1864 | 23R | 120 | 16.115 | 1934 |
| 24L | 120 | 16.115 | 1934 | 24R | 120 | 16.509 | 1981 |
| 25L | 120 | 16.509 | 1981 | | | | |

NOTE: In this table, the weight established for each paratrooper is 120 kg (265 lb), and is applied to each seat.

MEDEVAC CONFIGURATION



| Position | Weight (kg) | Arm-H (m) | Moment (kg x m) | Position | Weight (kg) | Arm-H (m) | Moment (kg x m) |
|-------------|----------------|--------------|--------------------|--------------|----------------|--------------|--------------------|
| Seat A | 85 | 10.724 | 912 | Stretcher 10 | 85 | 14.788 | 1256 |
| Seat B | 85 | 10.724 | 912 | Stretcher 11 | 85 | 14.788 | 1256 |
| Seat C | 85 | 16.058 | 1365 | Stretcher 12 | 85 | 14.788 | 1256 |
| Seat D | 85 | 16.058 | 1365 | Stretcher 13 | 85 | 6.660 | 566 |
| Seat E | 85 | 16.566 | 1408 | Stretcher 14 | 85 | 6.660 | 566 |
| Seat F | 85 | 16.566 | 1408 | Stretcher 15 | 85 | 6.660 | 566 |
| Stretcher 1 | 85 | 6.660 | 566 | Stretcher 16 | 85 | 9.200 | 782 |
| Stretcher 2 | 85 | 6.660 | 566 | Stretcher 17 | 85 | 9.200 | 782 |
| Stretcher 3 | 85 | 6.660 | 566 | Stretcher 18 | 85 | 9.200 | 782 |
| Stretcher 4 | 85 | 9.200 | 782 | Stretcher 19 | 85 | 12.248 | 1041 |
| Stretcher 5 | 85 | 9.200 | 782 | Stretcher 20 | 85 | 12.248 | 1041 |
| Stretcher 6 | 85 | 9.200 | 782 | Stretcher 21 | 85 | 12.248 | 1041 |
| Stretcher 7 | 85 | 12.248 | 1041 | Stretcher 22 | 85 | 14.788 | 1256 |
| Stretcher 8 | 85 | 12.248 | 1041 | Stretcher 23 | 85 | 14.788 | 1256 |
| Stretcher 9 | 85 | 12.248 | 1041 | Stretcher 24 | 85 | 14.788 | 1256 |

CARGO TRANSPORT CONFIGURATION

| CARGO CONFIGURATION – TABLE OF MOMENTS | | | | | |
|--|------------------|-------|--------|--------|--------|
| ZONE | I | II | III | IV | V |
| ARM (m) | 6.406 | 9.454 | 11.994 | 14.026 | 16.243 |
| CARGO (kg) | MOMENTS (kg x m) | | | | |
| 50 | 320 | 472 | 599 | 701 | 812 |
| 100 | 640 | 945 | 1 199 | 1 402 | 1 624 |
| 150 | 961 | 1 418 | 1 799 | 2 104 | 2 436 |
| 200 | 1 281 | 1 890 | 2 398 | 2 805 | 3 248 |
| 250 | 1 601 | 2 363 | 2 998 | 3 506 | 4 060 |
| 300 | 1 922 | 2 836 | 3 598 | 4 207 | 4 873 |
| 350 | 2 242 | 3 309 | 4 197 | 4 909 | 5 685 |
| 400 | 2 562 | 3 781 | 4 797 | 5 610 | 6 497 |
| 450 | 2 863 | 4 254 | 5 397 | 6 311 | 7 309 |
| 500 | 3 203 | 4 727 | 5 997 | 7 013 | 8 121 |
| 550 | 3 523 | 5 199 | 6 596 | 7 714 | 8 933 |
| 600 | 3 847 | 5 672 | 7 196 | 8 415 | 9 745 |
| 650 | 4 164 | 6 145 | 7 796 | 9 117 | 10 558 |
| 700 | 4 484 | 6 617 | 8 395 | 9 818 | 11 370 |
| 750 | 4 804 | 7 090 | 8 995 | 10 519 | 12 182 |
| 800 | 5 125 | 7 563 | 9 595 | 11 220 | 12 994 |
| 850 | 5 445 | 8 036 | 10 195 | 11 922 | 13 806 |
| 900 | 5 765 | 8 508 | 10 794 | 12 623 | 14 168 |
| 950 | 6 086 | 8 981 | 11 394 | 13 324 | 15 430 |
| 1 000 | 6 406 | 9 454 | 11 994 | 14 026 | 16 243 |
| NOTE: Moments for loads that exceed 1000 kg may be calculated by addition. | | | | | |

USABLE FUEL WEIGHT & MOMENT

| Main Wing Tanks | | | | Auxiliary Wing Tanks | | | |
|--|-------|----------------|--------------------|--|-------|----------------|--------------------|
| Capacity: 3 389 litres (895 U.S. Gal.) | | | | Capacity: 4 111 litres (1 087 U.S. Gal.) | | | |
| kg | lb | Arm - H (m) | Moment (kg x m) | kg | lb | Arm - H (m) | Moment (kg x m) |
| 91 | 200 | 12.192 | 1 109 | 91 | 200 | 12.111 | 1 102 |
| 181 | 400 | 12.139 | 2 197 | 181 | 400 | 12.076 | 2 186 |
| 272 | 600 | 12.098 | 3 290 | 272 | 600 | 12.054 | 3 279 |
| 363 | 800 | 12.068 | 4 381 | 363 | 800 | 12.046 | 4 373 |
| 454 | 1 000 | 12.039 | 5 466 | 454 | 1 000 | 12.038 | 5 466 |
| 544 | 1 200 | 12.019 | 6 538 | 544 | 1 200 | 12.035 | 6 547 |
| 635 | 1 400 | 12.007 | 7 624 | 635 | 1 400 | 12.032 | 7 640 |
| 726 | 1 600 | 11.994 | 8 707 | 726 | 1 600 | 12.030 | 8 733 |
| 816 | 1 800 | 11.985 | 9 780 | 816 | 1 800 | 12.029 | 9 816 |
| 907 | 2 000 | 11.979 | 10 865 | 907 | 2 000 | 12.029 | 10 910 |
| 998 | 2 200 | 11.973 | 11 949 | 998 | 2 200 | 12.028 | 12 004 |
| 1 088 | 2 400 | 11.968 | 13 021 | 1 088 | 2 400 | 12.029 | 13 086 |
| 1 134 | 2 500 | 11.967 | 13 570 | 1 134 | 2 500 | 12.029 | 13 641 |
| 1 361 | 3 000 | 11.959 | 16 276 | 1 361 | 3 000 | 12.031 | 16 374 |
| 1 587 | 3 500 | 11.954 | 18 971 | 1 587 | 3 500 | 12.033 | 19 096 |
| 1 814 | 4 000 | 11.951 | 21 679 | 1 814 | 4 000 | 12.036 | 21 833 |
| 2 041 | 4 500 | 11.949 | 24 388 | 2 041 | 4 500 | 12.039 | 24 571 |
| 2 268 | 5 000 | 11.947 | 27 096 | 2 268 | 5 000 | 12.042 | 27 311 |
| 2 494 | 5 500 | 11.946 | 29 793 | 2 494 | 5 500 | 12.044 | 30 037 |
| 2 721 | 6 000 | 11.944 | 32 500 | 2 721 | 6 000 | 12.046 | 32 777 |
| 2 881 | 6 351 | 11.943 | 34 408 | 2 948 | 6 500 | 12.049 | 35 520 |
| | | | | 3 175 | 7 000 | 12.052 | 38 265 |
| | | | | 3 401 | 7 500 | 12.056 | 41 002 |
| | | | | 3 494 | 7 702 | 12.058 | 42 131 |

NOTE: Fuel Density: 0.85 kg / litre (7.1 lb / U.S. Gal.)

FORM F – WEIGHT AND BALANCE CLEARANCE

| FORM F-WEIGHT AND BALANCE CLEARANCE | | | | | | |
|-------------------------------------|---|---------------------------------------|-------|----------------|------------|-------|
| DATE | | AIRPLANE | FROM: | HOME STATION | | |
| MISSION/TRIP/FLIGHT/No | | SERIAL No | TO: | PILOT | | |
| POS | ITEM | WEIGHT | H-ARM | MOMENT | LIMITATION | % MAC |
| 1 | BASIC EMPTY WEIGHT (BEW) | | | | | |
| 2 | PILOT AND COPILOT | | | | | |
| | 3rd CREW MEMBER | | | | | |
| | LOAD MASTER | | | | | |
| | AUXILIARY CREW MEMBER | | | | | |
| 3 | CREW'S BAGGAGE | | | | | |
| | MISSION EQUIPMENT | | | | | |
| | EMERGENCY EQUIPMENT | | | | | |
| | OTHERS EQUIPMENTS | | | | | |
| 4 | OPERATIONAL EMPTY WEIGHT (O.E.W) 1+2+3 | | | | | |
| 5 | PASSENGER/PARACHUT | | | | | |
| | LOAD | | | | | |
| 6 | MAX. ZERO FUEL WEIGHT | | | | | |
| | MAX. LOGISTICAL=20700 (4+5) | | | | | |
| | MAX. ASSAULT=16500 (4+5) | | | | | |
| 7 | MAIN TANKS | | | | | |
| | AUXILIARY TANKS | | | | | |
| 8 | TAKE-OFF WEIGHT (6+7) (Weight and CG Limitation) | | | | | |
| REMARK: | | | | | | |
| 10 | ZERO FUEL WEIGHT (6) | | | | | |
| 11 | JETBOMBERS PARACHUTISTS | | | | | |
| | LOAD | | | | | |
| 12 | LANDING FUEL MAIN TANKS | | | | | |
| | AUXILIARY TANKS | | | | | |
| 13 | LANDING WEIGHT (10-11+12) (Weight and CG Limitation) | | | | | |
| COMPUTED BY NAME: | | WEIGHT AND BALANCE AUTHORITY NAME: | | PILOT NAME: | | |
| SIGNATURE | | SIGNATURE | | SIGNATURE | | |

PERMISSIBLE CENTER OF GRAVITY 13,5%–32% MAC