

Given:

Air at 200 kPa and 950 K enters an adiabatic nozzle at low velocity and is discharged at a pressure of 110 kPa. Assume constant specific heat of air is 1.11 kJ/kgK and the specific heat ratio is 1.349.

$$P_1 := 200 \text{ kPa} \quad T_1 := 950 \text{ K} \quad P_2 := 110 \text{ kPa} \quad c_p := 1.11 \frac{\text{kJ}}{\text{kg K}} \quad k := 1.349$$

Required:

If the isentropic efficiency of the nozzle is 92%, determine the maximum possible velocity, the exit temperature of the air, and the actual velocity of the air.

Solution:

The isentropic efficiency of the nozzle is defined as

$$\eta_N := 92 \%$$

For an isentropic process the following is true

$$\left(\frac{T_2}{T_1} \right)_{s = \text{const}} = \left(\frac{P_2}{P_1} \right)^{\frac{k-1}{k}}$$

Solving for the temperature at state 2 for an isentropic process is then

$$T_{2s} := T_1 \cdot \left(\frac{P_2}{P_1} \right)^{\frac{k-1}{k}} = 813.9 \text{ K}$$

The 1st Law for a nozzle is shown below when $V_2 \gg V_1$ shows

$$\begin{aligned} \frac{d}{dt} E_{sys} &= \sum E'_{in} - \sum E'_{out} \\ 0 &= m' \cdot \left(h_1 + \frac{V_1^2}{2} \right) - m' \cdot \left(h_2 + \frac{V_2^2}{2} \right) \\ h_1 &= h_2 + \frac{V_2^2}{2} \end{aligned}$$

Solving for the velocity at the outlet when the process is an isentropic process shows

$$V_{2s} = \sqrt{2 \cdot (h_1 - h_{2s})}$$

Since the specific heat is constant, the maximum velocity at the exit can be expressed as

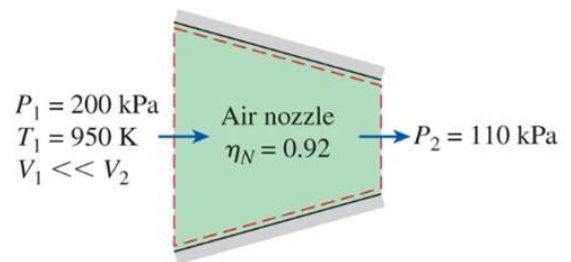
$$V_{2s} := \sqrt{2 \cdot c_p \cdot (T_1 - T_{2s})} = 549.7 \frac{\text{m}}{\text{s}}$$

Starting with the definition of the isentropic efficiency of a nozzle

$$\eta_N = \frac{h_1 - h_{2a}}{h_1 - h_{2s}}$$

Since the specific heat is constant, the isentropic efficiency of the nozzle can be expressed as

$$\eta_N = \frac{c_p \cdot (T_1 - T_{2a})}{c_p \cdot (T_1 - T_{2s})} = \frac{T_1 - T_{2a}}{T_1 - T_{2s}}$$



Solution (contd.):

Solving for the actual temperature at the exit yields

$$T_{2a} := T_1 - \eta_N \cdot (T_1 - T_{2s}) = 824.8 \text{ K}$$

The actual exit velocity may then be found by the alternate relation for the isentropic efficiency of a nozzle shown below.

$$\eta_N = \frac{V_{2a}^2}{V_{2s}^2}$$

Solving for the actual exit velocity yields

$$V_{2a} := \sqrt{\eta_N \cdot V_{2s}^2} = 527.3 \frac{\text{m}}{\text{s}}$$

$$V_{2a} := \sqrt{2 \cdot c_p \cdot (T_1 - T_{2a})} = 527.3 \frac{\text{m}}{\text{s}}$$

