

NSE (A)

Equation
analysis

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Laminar flow
between plates
(A)

Flow down
inclined plane
(A)

Tips (A)

Navier-Stokes Equations – 2d case

SOE3211/2 Fluid Mechanics lecture 3

NSE (A)

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Equation analysis

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Flow down inclined plane (A)

Tips (A)

- conservation of mass, momentum.
- often written as set of pde's
- differential form – fluid flow at a point
- 2d case, incompressible flow :

Continuity equation :

$$\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0$$

- conservation of mass
- seen before – potential flow

Momentum equations :

$$\begin{aligned}\frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} &= -\frac{1}{\rho} \frac{\partial p}{\partial x} \\ &+ \nu \left[\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right] + f_x \\ \frac{\partial u_y}{\partial t} + u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} &= -\frac{1}{\rho} \frac{\partial p}{\partial y} \\ &+ \nu \left[\frac{\partial^2 u_y}{\partial x^2} + \frac{\partial^2 u_y}{\partial y^2} \right] + f_y\end{aligned}$$

- (x and y cmpts)
- 3 variables, u_x , u_y , p
- linked equations
- need to simplify by considering details of problem

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Tips (A)

The NSE are

- Non-linear – terms involving $u_x \frac{\partial u_x}{\partial x}$
- Partial differential equations – u_x, p functions of x, y, t
- 2nd order – highest order derivatives $\frac{\partial^2 u_x}{\partial x^2}$
- Coupled – momentum equation involves p, u_x, u_y

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Two ways to solve these equations

- ① Apply to *simple* cases – simple geometry, simple conditions – and reduce equations until we can solve them
- ② Use computational methods – CFD (SOE3212/3)

Equation analysis (A)

Consider the various terms :

$$\frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} = - \frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left[\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right] + f_x$$

$$\frac{\partial u_x}{\partial t}$$

- change of u_x at a point

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$$u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y}$$

- *transport/advection* term
- how does flow (u_x, u_y) move u_x ?
- non-linear

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$$- \frac{1}{\rho} \frac{\partial p}{\partial x}$$

- pressure gradient –
usually drives flow

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$$\nu \left[\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right]$$

- viscous term – effect of viscosity ν on flow
- has a diffusive effect

Equation analysis (A)

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f_x

- external *body forces* – eg. gravity

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Laminar flow between plates (A)

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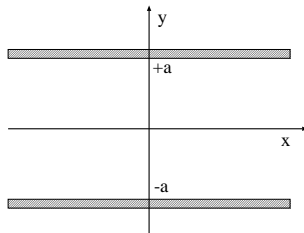
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Tips (A)

Fully developed laminar flow
between infinite plates at
 $y = \pm a$

What do we expect from the
flow?

- $\underline{u} = 0$ at walls
- Flow symmetric around $y = 0$
- Flow parallel to walls



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Flow parallel to walls – we expect

$$u_y = 0, \quad \frac{dp}{dy} = 0 \quad \text{and} \quad u_x = u_x(y)$$

$$\frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left[\frac{\partial^2 u_x}{\partial y^2} \right]$$

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Tips (A)

Flow fully developed – no change in profile in streamwise direction

$$\text{i.e. } \frac{\partial}{\partial t} = 0, \quad \frac{\partial}{\partial x} = 0$$

$$= -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left[\frac{\partial^2 u_x}{\partial y^2} \right]$$

Flow fully developed – no change in profile in streamwise direction

$$\text{i.e. } \frac{\partial}{\partial t} = 0, \quad \frac{\partial}{\partial x} = 0$$

So momentum equation becomes

$$0 = -\frac{1}{\rho} \frac{dp}{dx} + \nu \frac{d^2 u_x}{dy^2}$$

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So momentum equation becomes

$$0 = -\frac{1}{\rho} \frac{dp}{dx} + \nu \frac{d^2 u_x}{dy^2}$$

Integrate once :

$$y \frac{dp}{dx} = \rho \nu \frac{du_x}{dy} + C_1$$

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But at $y = 0$, $\frac{du_x}{dy} = 0$ (symmetry), so $C_1 = 0$.

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Integrate again

$$\frac{1}{2} y^2 \frac{dp}{dx} = \rho \nu u_x + C_2$$

So momentum equation becomes

$$0 = -\frac{1}{\rho} \frac{dp}{dx} + \nu \frac{d^2 u_x}{dy^2}$$

Integrate once :

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But at $y = 0$, $\frac{du_x}{dy} = 0$ (symmetry), so $C_1 = 0$.

Integrate again

$$\frac{1}{2} y^2 \frac{dp}{dx} = \rho \nu u_x + C_2$$

But at $y = \pm a$, $u_x = 0$, so

$$C_2 = \frac{1}{2} a^2 \frac{dp}{dx}$$

Final solution

$$u_x(y) = \frac{1}{2\rho\nu} (y^2 - a^2) \frac{dp}{dx}$$

– equation of a parabola

Also, remember that

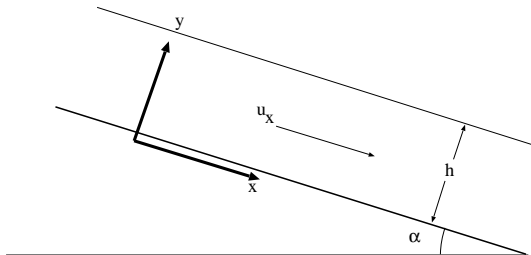
$$\tau = \mu \frac{\partial u_x}{\partial y}$$

So from this we see that in this case

$$\tau = y \frac{dp}{dx}$$

Flow down inclined plane (A)

- Flow of liquid down inclined plane



Take x -component momentum equation

$$\begin{aligned} \frac{\partial u_x}{\partial t} + u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} &= -\frac{1}{\rho} \frac{\partial p}{\partial x} \\ &+ \nu \left[\frac{\partial^2 u_x}{\partial x^2} + \frac{\partial^2 u_x}{\partial y^2} \right] + f_x \end{aligned}$$

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Note :

- 1 Steady flow
- 2 $u_x(y)$ only
- 3 *No pressure gradient*
- 4 $f_x = g \sin \alpha$

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- 1 Steady flow
- 2 $u_x(y)$ only
- 3 *No pressure gradient*
- 4 $f_x = g \sin \alpha$

Equation becomes

$$\frac{d^2 u_x}{dy^2} = -\frac{g}{\nu} \sin \alpha$$

which we can integrate easily.

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Boundary conditions :

- lower surface – $u_x(0) = 0$
- upper surface – $\frac{du_x}{dy} = 0$

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- lower surface – $u_x(0) = 0$
- upper surface – $\frac{du_x}{dy} = 0$

Solution

$$u_x = \frac{g}{\nu} \sin \alpha \left(hy - \frac{y^2}{2} \right)$$

Tips (A)

Most NSE problems will be time-independent. They will probably only involve one direction of flow, and one coordinate direction. They will probably be either *pressure driven* (so no viscous term) or *shear driven* (ie. viscous related, so no pressure term).

Thus, most NSE problems will lead to a 2nd order ODE for a velocity component (u_x or u_y) as a function of one coordinate (x or y).

Thus we would expect to integrate twice, and to impose two boundary conditions.

- A *wall* boundary condition produces a fixed value : eg.
 $u_x = 0$.
- A *free surface* produces a *zero gradient* condition, eg.
 $\frac{du_x}{dy} = 0$.