File: C:\Program Files (x86)\GasTurb\GasTurb10\Demo_jet.CYJ - modified
Date: 17Aug22
Time: 21:44

Turbojet SL Mn=-0.137 ISA + 6 C 60% Relative Humidity

Station W amb 2 0.435 3 0.435 31 0.387 4 0.392	T 294.25 293.15 456.03 456.03 993.15	P 101.325 99.000 386.100 386.100 362.934	WRstd 0.450 0.144 0.204	FN = TSFC = FN/W2 = Prop Eff = eta core =	0.22 25.0605 506.04 0.0000 0.1974
41 0.414 49 0.414 5 0.436 6 0.436 8 0.436 P2/P1 = 0.9900 Efficiencies: Compressor Burner Turbine Spool mech	966.79 813.75 796.97 796.97 796.97 P6/P5 = 0. isentr po 0.8460 0. 0.9900 0.9000 0.	362.934 167.789 167.789 164.433 164.433 9800 CD8 = Plytr RNI 8719 0.949 8908 0.471 pminal Spd	0.212 0.439 0.448 0.9458 P/P 3.900 0.940 2.163 117000	WF = S NOX = XM8 = A8 = P8/Pamb = Ang8 = WCL/W2 = Loading %= e45 th = WBld/W2 = PWX = ZWBld =	0.00551 0.07292 0.8728 0.0020 1.6228 20.00 0.05000 100.00 0.87699 0.01000 0.00

hum [%] war0 FHV Fuel 60.0 0.00889 43.124 Generic

Iteration Variables:

Isentr.Compr.Efficiency (0...0.99) = 0.846046

Iteration Targets:
Net Thrust

= 0.22