

File: C:\Program Files (x86)\GasTurb\GasTurb10\RF_4off.CYJ - modified

Date: 17Aug22

Time: 22:23

Turbojet SL Mn=-0.137 ISA + 6 C 60% Relative Humidity

Station	W	T	P	WRstd	FN	=	0.22
amb		294.25	101.325		TSFC	=	36.0464
2	0.435	293.15	99.000	0.450	FN/W2	=	506.04
3	0.435	526.11	386.100	0.155	Prop Eff	=	0.0000
31	0.387	526.11	386.100		eta core	=	0.1380
4	0.395	1258.63	362.934	0.231			
41	0.417	1223.76	362.934	0.240	WF	=	0.00793
49	0.417	1015.61	150.916		s NOx	=	0.10457
5	0.438	993.15	150.916	0.548	XM8	=	0.7719
6	0.438	993.15	147.898		A8	=	0.0027
8	0.438	993.15	147.898	0.559	P8/Pamb	=	1.4596
P2/P1 = 0.9900		P6/P5 = 0.9800	CD8 = 0.9308		Ang8	=	20.00
Efficiencies:	isent	polytr	RNI	P/P	W NGV/W2	=	0.05000
Compressor	0.5888	0.6563	0.949	3.900	WCL/W2	=	0.05000
Burner	0.9900			0.940	Loading %	=	100.00
Turbine	0.9000	0.8900	0.319	2.405	e45 th	=	0.88033
Spool mech	0.9900	Nominal	Spd	117000	WBld/W2	=	0.01000
Bleed Air:	PBld =	386.10	TBld =	526.1	PWX	=	0.00
					ZWBld	=	0.00000

hum [%]	war0	FHV	Fuel
60.0	0.00889	43.124	Generic

Iteration Variables:

Isentr.Compr.Efficiency (0...0.99)	= 0.5888
Burner Exit Temperature K (0...1500)	= 1258.63

Iteration Targets:

Net Thrust	= 0.22
Nozzle Throat Temperature T8	= 993.15