

Control Systems Engineering (EYAG-1005):

Unit 03

Luis I. Reyes-Castro

Escuela Superior Politécnica del Litoral (ESPOL)
Guayaquil - Ecuador

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- 1 First-order Systems
- 2 Second-order Systems
- 3 Steady-state Errors
- 4 Evan's Root Locus
- 5 Bode Plots

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Definition:

We say that a system with transfer function $G(s)$ is of first-order if:

$$G(s) = \frac{K}{s + \sigma} \quad \text{for some } K, \sigma > 0$$

First-order Systems

Step response, i.e., $r(t) = u(t)$:

$$R(s) = \frac{1}{s} \quad \Rightarrow \quad c(t) = \mathcal{L}^{-1}[R(s) G(s)]$$

$$\Rightarrow c(t) = \mathcal{L}^{-1}\left[\frac{K}{s(s + \sigma)}\right]$$

$$\Rightarrow c(t) = \mathcal{L}^{-1}\left[\frac{K/\sigma}{s} - \frac{K/\sigma}{s + \sigma}\right]$$

$$\Rightarrow c(t) = \left[\frac{K}{\sigma} - \frac{K}{\sigma} e^{-\sigma t}\right] u(t), \quad \forall t \in \mathbb{R}$$

$$\Rightarrow c(t) = \frac{K}{\sigma} - \frac{K}{\sigma} e^{-\sigma t}, \quad \forall t \geq 0$$

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Second-order Systems

Recall the following results from elementary algebra:

- For any $\alpha, \beta, \gamma \in \mathbb{R}$ the roots of the second-order polynomial

$$D(s) = \alpha s^2 + \beta s + \gamma$$

are as follows:

$$s_1 = -\frac{\beta}{2\alpha} + \frac{\sqrt{\beta^2 - 4\alpha\gamma}}{2\alpha} \qquad s_2 = -\frac{\beta}{2\alpha} + \frac{\sqrt{\beta^2 - 4\alpha\gamma}}{2\alpha}$$

Therefore:

- If $\beta^2 > 4\alpha\gamma$ then the roots are repeated and different.
- If $\beta^2 = 4\alpha\gamma$ then the roots are repeated and identical.
- If $\beta^2 < 4\alpha\gamma$ then the roots are complex conjugates.

Second-order Systems

- For any $a, b \in \mathbb{R}$ the roots of the second-order polynomial

$$D(s) = s^2 + a s + b$$

are as follows:

$$s_1 = -\frac{a}{2} + \frac{\sqrt{a^2 - 4b}}{2} \qquad s_2 = -\frac{a}{2} - \frac{\sqrt{a^2 - 4b}}{2}$$

Therefore:

- If $a^2 > 4b$ then the roots are repeated and different.
- If $a^2 = 4b$ then the roots are repeated and identical.
- If $a^2 < 4b$ then the roots are complex conjugates.

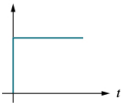


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Steady-state Errors

Recall the reference signals:

TABLE 7.1 Test waveforms for evaluating steady-state errors of position control systems

Waveform	Name	Physical interpretation	Time function	Laplace transform
	Step	Constant position	1	$\frac{1}{s}$
	Ramp	Constant velocity	t	$\frac{1}{s^2}$
	Parabola	Constant acceleration	$\frac{1}{2}t^2$	$\frac{1}{s^3}$

Recall the following property of the Laplace Transform:

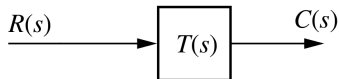
Final Value Theorem

For any function $f(t)$ defined for $t \geq 0$ with Laplace Transform $F(s)$ we have that:

$$\lim_{t \rightarrow \infty} f(t) = \lim_{s \rightarrow 0} s F(s)$$

Steady-state Errors

Steady-state errors for simple feedthrough systems:



$$E(s) = R(s) - C(s) = R(s) - T(s) R(s)$$

$$\Rightarrow E(s) = R(s) [1 - T(s)]$$

$$\Rightarrow e(\infty) = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s R(s) [1 - T(s)]$$

Steady-state Errors

- Step input $r(t) = u(t)$:

$$R(s) = \frac{1}{s} \quad \Longrightarrow \quad e_{step}(\infty) = \lim_{s \rightarrow 0} 1 - T(s)$$

- Ramp input $r(t) = t u(t)$:

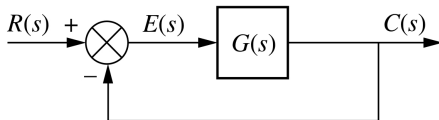
$$R(s) = \frac{1}{s^2} \quad \Longrightarrow \quad e_{ramp}(\infty) = \lim_{s \rightarrow 0} \frac{1 - T(s)}{s}$$

- Parabolic input $r(t) = (1/2) t^2 u(t)$:

$$R(s) = \frac{1}{s^3} \quad \Longrightarrow \quad e_{parabolic}(\infty) = \lim_{s \rightarrow 0} \frac{1 - T(s)}{s^2}$$

Steady-state Errors

Steady-state errors for unity-feedback systems:



$$E(s) = R(s) - C(s) = R(s) - G(s) E(s)$$

$$\Rightarrow E(s) [1 + G(s)] = R(s)$$

$$\Rightarrow E(s) = \frac{R(s)}{1 + G(s)}$$

$$\Rightarrow e(\infty) = \lim_{t \rightarrow \infty} e(t) = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} \frac{s R(s)}{1 + G(s)}$$

Steady-state Errors

- Step input $r(t) = u(t)$:

$$R(s) = \frac{1}{s} \quad \Rightarrow \quad e_{step}(\infty) = \lim_{s \rightarrow 0} \frac{1}{1 + G(s)}$$

- Ramp input $r(t) = t u(t)$:

$$R(s) = \frac{1}{s^2} \quad \Rightarrow \quad e_{ramp}(\infty) = \lim_{s \rightarrow 0} \frac{1}{s G(s)}$$

- Parabolic input $r(t) = (1/2) t^2 u(t)$:

$$R(s) = \frac{1}{s^3} \quad \Rightarrow \quad e_{parabolic}(\infty) = \lim_{s \rightarrow 0} \frac{1}{s^2 G(s)}$$

Steady-state Errors

Furthermore, for unity-feedback systems:

- Position constant K_p :

$$K_p = \lim_{s \rightarrow 0} G(s) \quad \Rightarrow \quad e_{step}(\infty) = \frac{1}{1 + K_p}$$

- Velocity constant K_v :

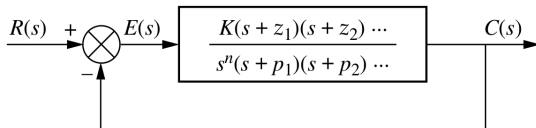
$$K_v = \lim_{s \rightarrow 0} s G(s) \quad \Rightarrow \quad e_{ramp}(\infty) = \frac{1}{K_v}$$

- Acceleration constant K_a :

$$K_a = \lim_{s \rightarrow 0} s^2 G(s) \quad \Rightarrow \quad e_{parabolic}(\infty) = \frac{1}{K_a}$$

Steady-state Errors

System Type: Number of integrators (poles at $s = 0$) on the forward path.

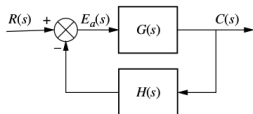


Steady-state error as a function of system type:

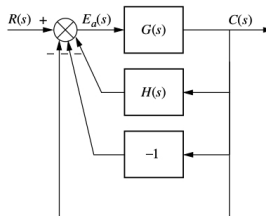
Input	Steady-state error formula	Type 0		Type 1		Type 2	
		Static error constant	Error	Static error constant	Error	Static error constant	Error
Step, $u(t)$	$\frac{1}{1 + K_p}$	$K_p = \text{Constant}$	$\frac{1}{1 + K_p}$	$K_p = \infty$	0	$K_p = \infty$	0
Ramp, $tu(t)$	$\frac{1}{K_v}$	$K_v = 0$	∞	$K_v = \text{Constant}$	$\frac{1}{K_v}$	$K_v = \infty$	0
Parabola, $\frac{1}{2}t^2u(t)$	$\frac{1}{K_a}$	$K_a = 0$	∞	$K_a = 0$	∞	$K_a = \text{Constant}$	$\frac{1}{K_a}$

Steady-state Errors

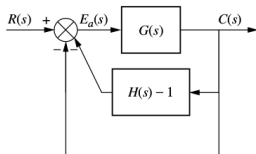
For non-unity non-unity-feedback systems, simply re-write the system in unity-feedback form:



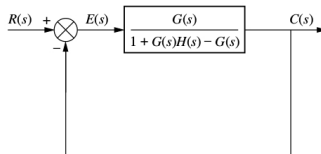
(b)



(c)



(d)



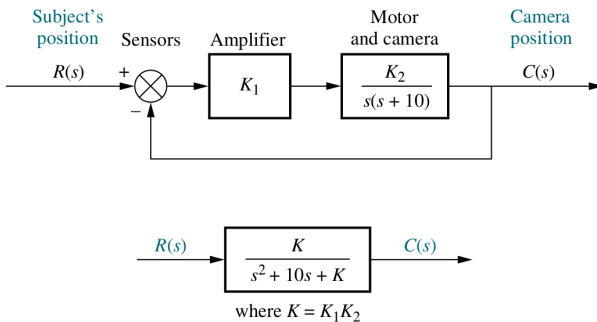
(e)

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Evan's Root Locus

Motivating example:

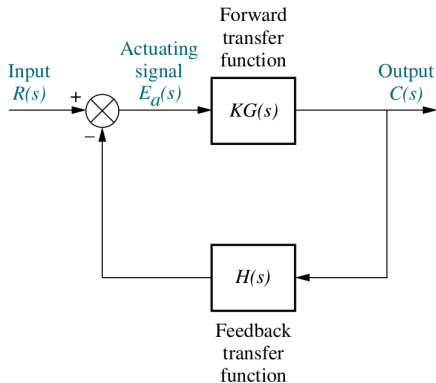


Evan's Root Locus

K	Pole 1	Pole 2
0	-10	0
5	-9.47	-0.53
10	-8.87	-1.13
15	-8.16	-1.84
20	-7.24	-2.76
25	-5	-5
30	$-5 + j2.24$	$-5 - j2.24$
35	$-5 + j3.16$	$-5 - j3.16$
40	$-5 + j3.87$	$-5 - j3.87$
45	$-5 + j4.47$	$-5 - j4.47$
50	$-5 + j5$	$-5 - j5$

Evan's Root Locus

The root locus concerns the design of closed-loop control systems with the following architecture:



Definitions:

- Loop gain: K
- Open-loop transfer function: $G(s) H(s)$

Objective:

- Sketch the roots of the closed-loop transfer function as the loop gain K ranges from near zero (*i.e.*, $K \rightarrow 0^+$) to infinity (*i.e.*, $K \rightarrow +\infty$).

Root locus sketching rules:

- The root locus is symmetric about the real axis.
- The number of branches, *i.e.*, pole trajectories, equals the number of poles of the open-loop transfer function.
- Each branch begins at an open-loop pole and ends either:
 - At an open-loop zero.
 - At infinity along an asymptote.
- Along the real line, root locus branches can be found to the left of any odd number of real open-loop poles or open-loop zeros.

- If the root locus has asymptotes, then the number of asymptotes is:

$$(\text{number of open-loop poles}) - (\text{number of open-loop zeros})$$

- If the root locus has asymptotes, then the centroid of the asymptotes is located along the real axis at the point:

$$\sigma_a = \frac{\sum (\text{open-loop pole locations}) - \sum (\text{open-loop zero locations})}{\text{number of asymptotes}}$$

- If the root locus has asymptotes, then their angles in radians are:

$$\theta_a = \frac{(2k + 1) \pi}{\text{number of asymptotes}} \quad \text{for } k = 0, \pm 1, \pm 2, \dots$$

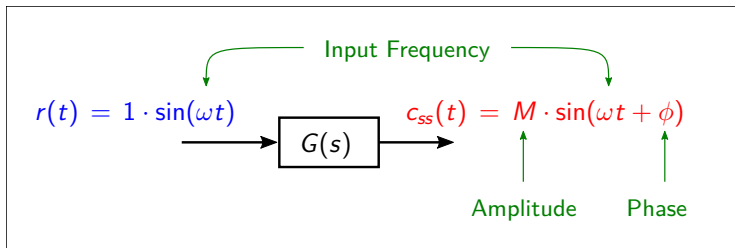
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Bode Plots

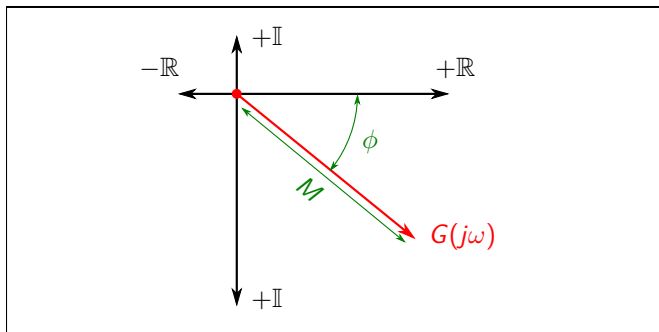
Consider a stable system with transfer function $G(s)$.

- Suppose the input $r(t)$ is a sinusoid with frequency ω and unit amplitude.
- Then the steady-state output $c_{ss}(t)$ must be a sinusoid with the same frequency ω but with a particular amplitude M and phase ϕ which depend on the transfer function $G(s)$ and on the input frequency ω .



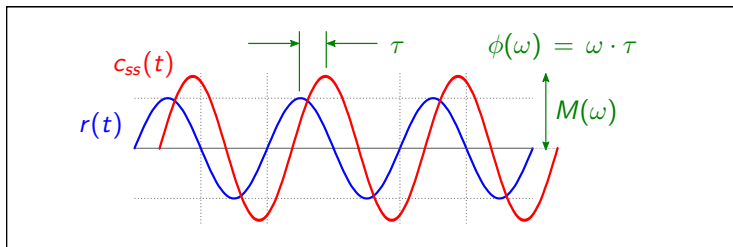
Bode Plots

- Given $G(s)$ and a frequency ω we can evaluate the amplitude and phase of the steady-state output by computing the phasor $G(j\omega)$. In particular:
 - The phasor's magnitude yields the amplitude $M(\omega)$.
 - The phasor's angle with $+\mathbb{R}$ yields the phase $\phi(\omega)$.



Bode Plots

- We can also estimate $M(\omega)$ and $\phi(\omega)$ experimentally:



- Furthermore, notice that:
 - Amplitude M is always positive.
 - If $M \in (0, 1)$ we get attenuation.
 - If $M = 1$ we get amplitude matching.
 - If $M > 1$ we get amplification.
 - Phase may be negative, zero or positive.
 - If $\phi < 0$ then the output lags the input.
 - If $\phi = 0$ then the output matches the input.
 - If $\phi > 0$ then the output leads the input.

Bode Plots are diagrams of $M(\omega)$ and $\phi(\omega)$. More precisely, they consist of the following two plots:

- **Magnitude Plot:** Amplitude $M(\omega)$ versus frequency ω .
 - The x-axis is frequency ω in decades, *i.e.*, $x = \log_{10}(\omega)$.
 - The y-axis is amplitude $M(\omega)$ in decibels, *i.e.*, $y = 20 \cdot \log(M(\omega))$.
- **Phase Plot:** Phase $\phi(\omega)$ versus frequency ω .
 - The x-axis is frequency ω in decades, *i.e.*, $x = \log_{10}(\omega)$.
 - The y-axis is phase in degrees, *i.e.*, $y = \phi(\omega)$.

Notice that when sketching Bode Plots by hand, we usually don't draw exactly the functions $M(\omega)$ and $\phi(\omega)$ but instead sketch *asymptotic approximations*.

Bode plot for a simple amplifier: $G(s) = K$

- Magnitude plot is constant at $y = 20 \cdot \log_{10}(K)$ decibels.
- Phase plot is constant at $y = 0$ degree.

Bode Plots

Bode plot for an integrator: $G(s) = \frac{1}{s}$

- Phasor as a function of ω :

$$G(j\omega) = \frac{1}{j\omega} = -\frac{j}{\omega} \quad \Rightarrow \quad M(\omega) = \frac{1}{\omega} \quad \& \quad \phi(\omega) = -90^\circ$$

- Magnitude plot is $y = -20 \cdot \log_{10}(\omega)$ decibels, *i.e.*, it is a line with slope of -20 decibels per decade which hits zero decibels at $\omega = 1$ rad/s.
- Phase plot is constant at $y = -90$ degree.

Bode plot for a differentiator: $G(s) = s$

- Phasor as a function of ω :

$$G(j\omega) = j\omega \quad \implies \quad M(\omega) = \omega \quad \& \quad \phi(\omega) = +90^\circ$$

- Magnitude plot is $y = +20 \cdot \log_{10}(\omega)$ decibels, *i.e.*, it is a line with slope of +20 decibels per decade which hits zero decibels at $\omega = 1$ rad/s.
- Phase plot is constant at $y = +90$ degree.