

# Sensor Subsystem Design

**Morris Blaustein** 



## **Sensor Subsystem Overview**



	Туре	Model	Connection	Function
	GPS	Locosys LS20031	Digital	Records UTC time, latitude and longitude, mean sea level altitude, and number of satellites tracked.
andras Bresnow	Altitude & Temp Sensor	Bosch BMP085	l <sup>2</sup> C Serial	Records altitude via non GPS and air temperature.
200	Video Camera	808 Spy Camera	Digital	Records video of the lander release

- The GPS and video camera will be used in the Carrier.
- The BMP085 will be used in both the Carrier and the Lander.



# **Sensor Subsystem Requirements**



ID	Requirement	Rationale	Priority	Parent(s)	Child(ren)	Verification			
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S-01	Cansat shall transmit UTC time every two (2) seconds	Base Mission Requirement	High	SYS-04	None			Х	Х
S-02	Cansat shall transmit latitude and longitude every two (2) seconds	Base Mission Requirement	High	SYS-04	None			X	Х
S-03	Cansat shall transmit number of satellites tracked every two (2) seconds	Base Mission Requirement	High	SYS-04	None			X	X
S-04	Cansat shall transmit altitude at accuracy of 2 m using sensor other than GPS every two (2) seconds	Base Mission Requirement	High	SYS-04	None	Х		X	X
S-05	Cansat shall transmit air temperature in Celsius every two (2) seconds	Base Mission Requirement	High	SYS-04	None			X	X
S-06	Cansat shall transmit remaining battery voltage in volts every two (2) seconds	Base Mission Requirement	High	SYS-04	EPS-05	Х		X	Х
S-07	Video camera shall record rocket in flight beginning no more than two (2) seconds before lander release	Selectable Objective Requirement	Medium	SYS-09	None			X	X

CanSat 2012 CDR: Team 1839 (Laser Wolverines)



# **Sensor Subsystem Requirements**



ID	Requirement	Rationale	Priority	Parent(s)	Child(ren)	Verification			
			, , , , ,	(-,		Α	I	T	D
S-08	Sensors must operate at 3.3V, 3.9V, or 5V	Cansat System regulates power at 3.3V, 3.9V, and 5V	High	EPS-01/02	None		X	X	
S-09	GPS Sensor shall withstand acceleration of the rocket	Able to transmit data following deployment	High	None	None		Х	X	



### **Sensor Changes Since PDR**



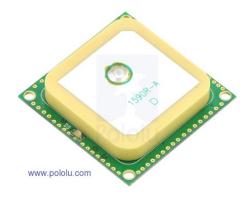
Video camera timing controlled by Arduino



### **Carrier GPS Summary**



Device	Accuracy (m)	Operating Voltage (V)	Current (mA)	Weight (g)	Update Rate (Hz)	Cold/Hot Start (s)	Dimensions (mm)	Cost (\$)
Locosys LS20031	2.5m	3.3V	41mA	14g	5Hz	35/2s	30x30mm	49.95



Presenter: Morris Blaustein

- Low current draw
- High accuracy
- Micro battery holds data for faster satellite acquisition
- 66 Channels
- TTL serial data

#### **GGA Fixed Data Example**

\$GPGGA,053740.000,2503.6319,N,12136.0099,E, 1,08,1.1,63.8,M,15.2,M,,0000\*64

GGA header UTC time Latitude N/S Longitude E/W #Satellites Altitude

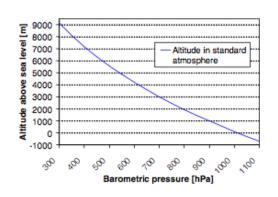


# **Carrier Non-GPS Altitude/ Temperature Sensor Summary**



Device	Operating Voltage (V)	Range (kPa)	Current (µA)	Pressure Accuracy (hPa)	Temperature Accuracy (C)	Dimensions (mm)	Cost (\$)
Bosch BMP085	1.8-3.6V	30kPa – 110kPa	0.1μΑ	1.5 hPa	2°C	16.5x16.5mm	19.95





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- Low power consumption
- Samples pressure and temperature
- I<sup>2</sup>C Interface

For pressure to altitude conversion

$$altitude = 44330 * (1 - (p/p_0)^{1/5.255})$$



# Lander Non-GPS Altitude Sensor Summary





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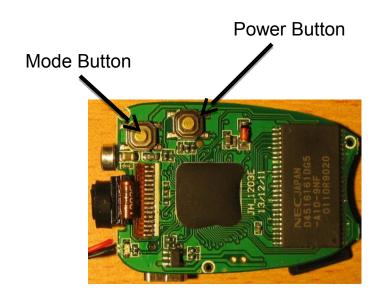
We will also be using the Bosch BMP085 for the Lander Non-GPS Altitude Sensor.



#### **Carrier Video Camera Summary**



Device	Operating Voltage	Resolution (pixels)	Weight (g)	Frame Rate (FPS)	Format	Current (mA)	Dimensions (mm)	Cost (\$)
808 Spy Camera	5V	720 x 480	8 g	30 FPS	M-JPEG	100- 140mA	63x100 mm	14.26



Presenter: Morris Blaustein

- Low cost
- Light weight
- High video resolution
- Records via micro SD card
- Outer case and built-in lithium polymer battery were removed
- Video timing controlled by Arduino



# **Electrical Power Subsystem Design**

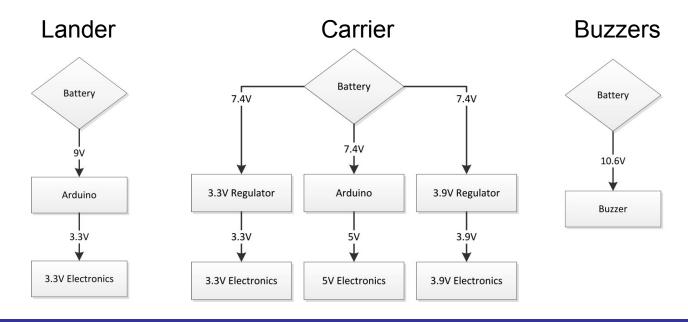
**Morris Blaustein** 



#### **EPS Overview**



	Model	Used For	Voltage	Purpose
	MH-9V250	Lander	9V	Supply power to Lander electronics
Batteries	Tenergy 14500	Carrier	7.4V	Supply power to Carrier electronics
	LIR2032	Lander/C arrier	10.6V	Supply power to the buzzers





### **EPS Changes Since PDR**



- 9V battery for Lander
- 3.6V Button cell batteries for buzzers
- Added 3.9V regulator
  - Video camera runs at 3.9V, not 5V
- Transceiver powered through 3.3V regulator, not Arduino



# **EPS Requirements**



ID	Requirement	Rationale	Priority	Parent(s)	Child(ren)	١	Verification			
	Roquiloment	Rationalo	1 Hority	i arom(s)	Jima(i cii)	Α	1	т	D	
EPS-01	Carrier battery shall output at least 5V	Arduino Uno requires 5V	High	CDH-11	S-08, EPS- 03		X	Х		
EPS-02	Lander battery shall output at least 5V	Arduino Nano requires 5V	High	CDH-12	S-08, EPS- 03		X	X		
EPS-03	Battery shall have high enough capacity to last the duration of the flight and recovery (approximately 3 hours)	Must power the system	High	None	None	X	X	X		
EPS-04	Battery shall have sufficient marginal voltage	Compensate for voltage drop	High	EPS-01, EPS- 02	None	X	Х			
EPS-05	Battery shall be light weight	Keep the system under the weight maximum	Medium	SYS-01	None		X			
EPS-06	Remaining battery voltage shall be measured	Base Mission Requirement	High	S-06	None	X		X		
EPS-07	Carrier and Lander shall have external power switch	Base Mission Requirement/Safet y/Convenience	High	SYS-05	None			X	X	

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# **EPS Requirements**

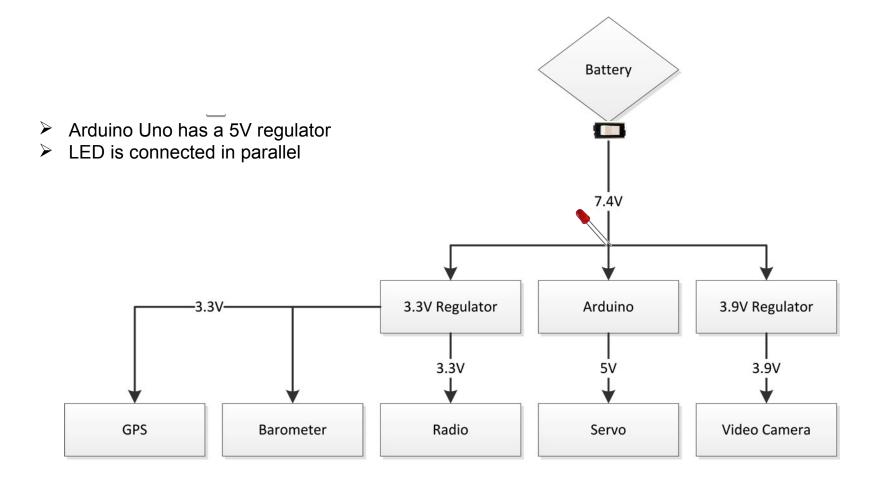


ID	Requirement	Rationale	Priority	Parent(s)	Child(ren)	V	Verification			
						A	ı	Т	D	
EPS-08	Buzzers in Carrier and Lander shall have independent power source	Base Mission Requirement	High	S-06	None	Х		Х	X	



## **Carrier Electrical Block Diagram**

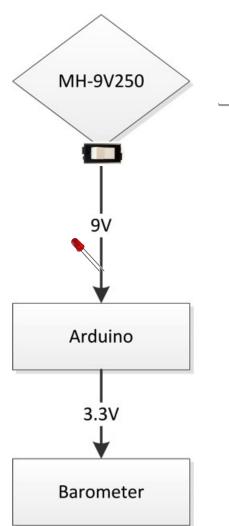






## **Lander Electrical Block Diagram**





- Arduino Nano has a 3.3V regulator
- ➤ LED is connected in parallel



# **Carrier Power Budget**



Device	Voltage (V)	Current (mA)	Expected Run Cycle (min)	Uncertainty (min)	Current Consumed (mAh)
Arduino Uno	5 V	140 mA	120 min	5 min	291 mAh
GPS	3.3 V	41 mA	60 min	5 min	44 mAh
Barometer	3.3 V	0.05 mA	60 min	5 min	.06 mAh
Video Camera	5 V	150 mA	20 min	5 min	63 mAh
Servo	5 V	500 mA	0.2 min	0.1 min	2.5 mAh
Transceiver	3.3 V	250 mA	60 min	5 min	271 mAh
Voltage Regulators					10% inefficiency

Total	Available	Margin
674 mAh	800 mAh	126 mAh



# **Lander Power Budget**



Device	Voltage (V)	Current (mA)	Expected Run Cycle (min)	Uncertainty (min)	Current Consumed (mAh)
Arduino Nano	5V	60 mA	120 min	5 min	125 mAh
Barometer	3.3 V	0.05 mA	60 min	5 min	.06 mAh

Total	Available	Margin
126 mAh	250mAh	124 mAh



# **Buzzer Power Budget**



Device	Voltage (V)	Current (mA)	Expected Run Cycle (min)	Uncertainty (min)	Current Consumed (mAh)	
Buzzer	10.6V	1 mA	180 min	60 min	4 mAh	

Total	Available	Margin
4 mAh	40 mAh	36 mAh



## **Power Source Summary**



	Battery	Used For	Туре	Voltage (V)	Weight (g)	Dimensio ns (mm)	Capacity (mAh)	Quantity	Price
+ HI-MH BATTERY - 90 ZSOWAH	MH- 9V250	Lander	NiMH	9V	19g	48 x 26 mm	250mAh	1	\$4.60
The state of the s	Tenergy 14500	Carrier	Lithium	3.7V	21g	14 x 49 mm	800 mAh	2, series	\$3.79
LIR2032  ETHOM ION BET	LIR2032	Buzzers	Li-lon	3.6V	3.1g	20 x 3 mm	40 mAh	3, series	\$3.59

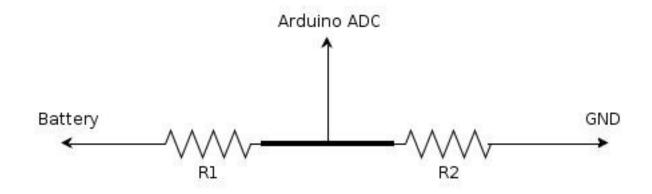
We are considering switching to a 9V battery for the Carrier because
 7.4V might drop too low



#### **Battery Voltage Measurement**



#### Arduino ADC (Analog to Digital Conversion)



- Simple and accurate
- 10 bit resolution
- Max input for ADC is 5V.
- Voltage will be scaled with resistors

$$V_{in} = V_{bat} \frac{R_1}{(R_1 + R_2)}$$



# **Sensor Subsystem Testing Overview**



#### Locosys LS20031

#### Sample Data

\$GPGGA,165256.000,4217.6726,N,08342.6620,W,1, 7,1.28,284.1,M,-34.0,M,,\*6E \$GPGGA,165256.200,4217.6726,N,08342.6620,W,1, 7,1.28,284.1,M,-34.0,M,,\*6C \$GPGGA,165256.400,4217.6726,N,08342.6620,W,1, 7,1.28,284.0,M,-34.0,M,,\*6B \$GPGGA,165256.600,4217.6726,N,08342.6620,W,1, 7,1.28,284.0,M,-34.0,M,,\*69

GGA header

**UTC** time

Latitude

N/S

Longitude

E/W #Satellites Altitude



- 7 Satellites tracked
- 284 m Altitude
  - Ann Arbor, MI (where data was aquired) is approximately 272m

#### **Conclusion**

- LS20031 is working properly
- Still need more tests



# Sensor Subsystem Testing Overview



#### Sample Data

Temperature = 23.00 \*C Pressure = 98059 Pa Altitude = 275.28 meters

Temperature = 23.00 \*C Pressure = 98053 Pa Altitude = 275.71 meters

Temperature = 23.00 \*C Pressure = 98061 Pa Altitude = 275.45 meters

Temperature = 23.00 \*C Pressure = 98064 Pa Altitude = 275.54 meters

#### Bosch BMP085

- 275.28 m Altitude verifies GPS accuracy (284 m)
- Temperature is highly accurate
  - Thermometer also showed 23° C

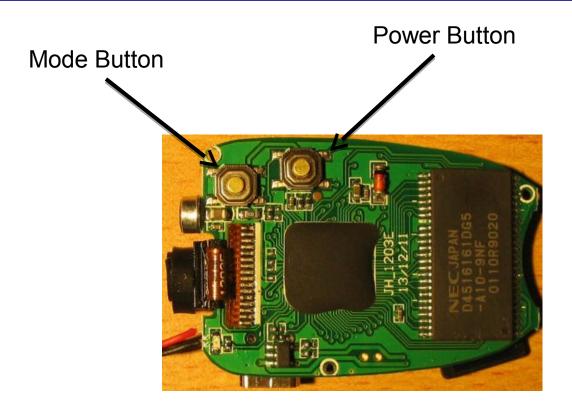
#### Conclusion

- BMP085 is working properly
- Still need more tests
  - Match altitude with GPS
- Consider appropriate placement in Cansat



# Imaging / Video Camera Testing Overview





Wires soldered to power and mode button give video control to Arduino Buttons activated based on 3.9V (high) or 0V (low)



### Imaging / Video Camera Testing Overview

### Current status of testing

- Video resolution is acceptable
- Video camera works on external power source (3.9V)
- Features of power/mode button not yet determined
  - To allow for Arduino control



Screenshot of Video



### **EPS Testing Overview**

### EPS testing when all circuits are complete

- Monitor average current consumption
  - Verify if 7.4V is sufficient for carrier
  - Account for regulator inefficiencies
- Arduino ADC voltage measurement
  - Compare with voltmeter
- Verify that power switch and LED work as expected
- Verify that transistor works as expected