



## *Drag Modelling Strategy*



## *Effect of Drag*

**Drag** in rockets is about an **order** of magnitude **lower** than **gravity** and is **tertiary** nonlinear effect.

Therefore, a simple **linearized** drag model based on non-drag trajectory **solution** can be used to predict its **effect**.



## *Simplified Drag Model*

In this regard, a **constant** average deceleration, based on total **energy loss**, gives reasonable performance **estimate**.

Under **vertical** motion assumption, the applicable **equation** is,

$$\frac{dV}{dt} = -\frac{g_0 I_{sp}}{m} \frac{dm}{dt} - \tilde{g} - \frac{D}{m} = -\frac{g_0 I_{sp}}{m} \frac{dm}{dt} - \tilde{g} - a_D$$

Here, **a<sub>D</sub>** is the constant drag **acceleration** term.



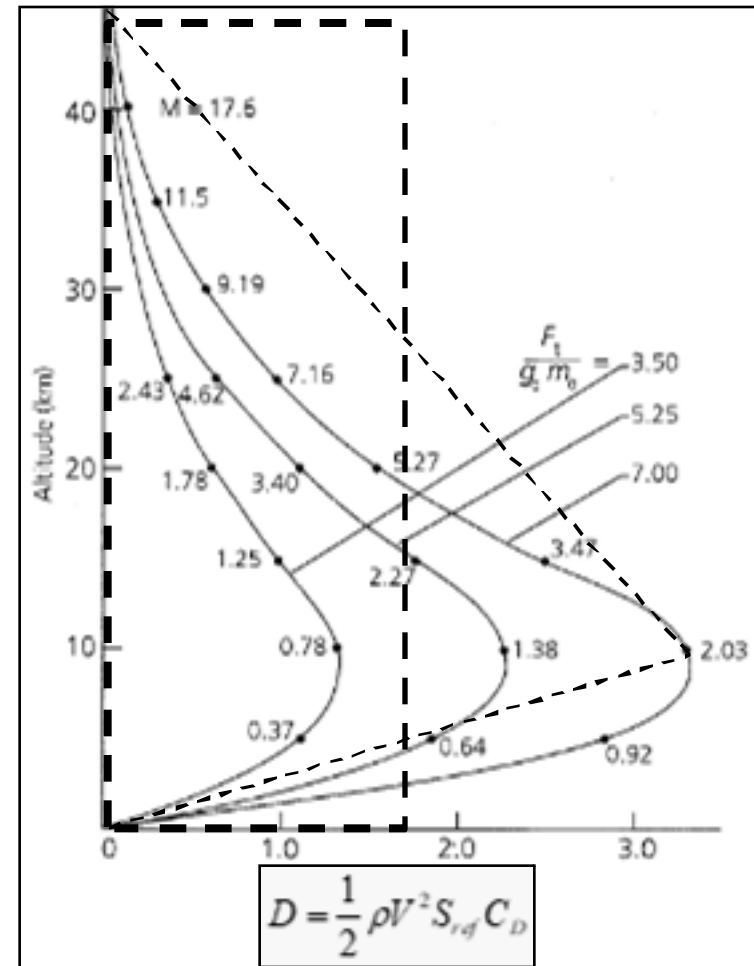
## *Constant Drag Deceleration Model*

Consider the **generic**  $D$  Vs.  $h$  plot, as presented **earlier**.

We see that a **straight** line model captures the **energy**, which is nothing but the **area** under the **curve**.

Further, as **area** of rectangle is **matched** with area of **triangle**, width of rectangle **denotes** the drag acceleration ' $a_D$ '.

Thus, ' $a_D$ ' is value of ' $D/2m$ ' ~10 km altitude.





## *Summary*

Therefore, to **summarize**, drag is a smaller **order** effect which can be **captured** from energy consideration.