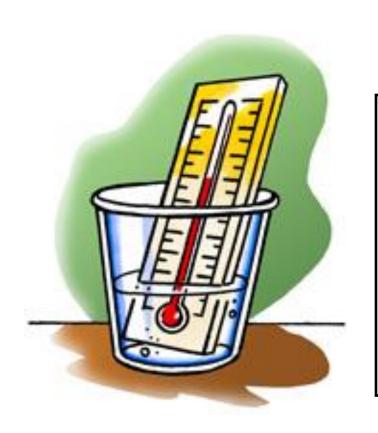
# AE 242 Aerospace Measurements Laboratory

#### Time varying measurements



When we pour hot or cold liquid, how much time will it take to show correct temperature? How fast temperature will rise / decrease in thermometer?

Will it be instantaneous or take time to show correct temperature?

#### Time varying measurements



After keeping the weight in the pan, how much time we should wait to get the correct measurement?

Can it be oscillatory?

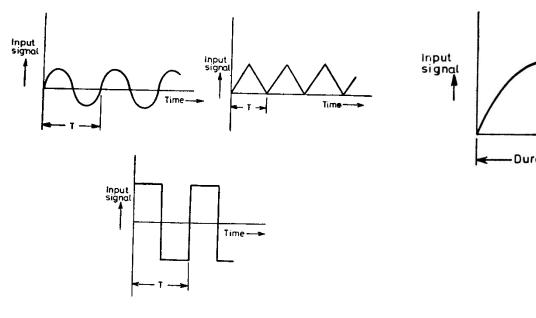
Time varying quantities to be measured.

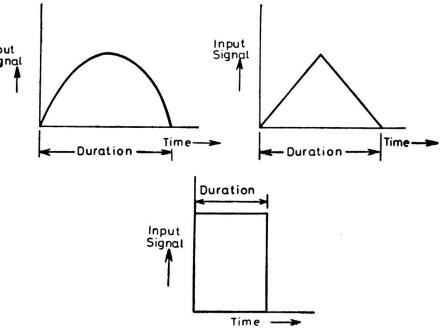
**Dynamic Inputs** 

Periodic input: Varying cyclically or repeating itself. Could be harmonic and non-harmonic

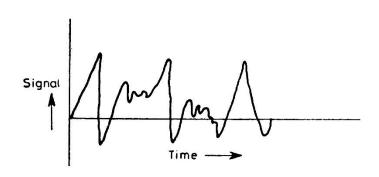
<u>Transient input</u>: Varying non-cyclically with time. It is of definite duration and reduces to zero after certain time.

Random input: Varying randomly with time, with no definite period and amplitude.





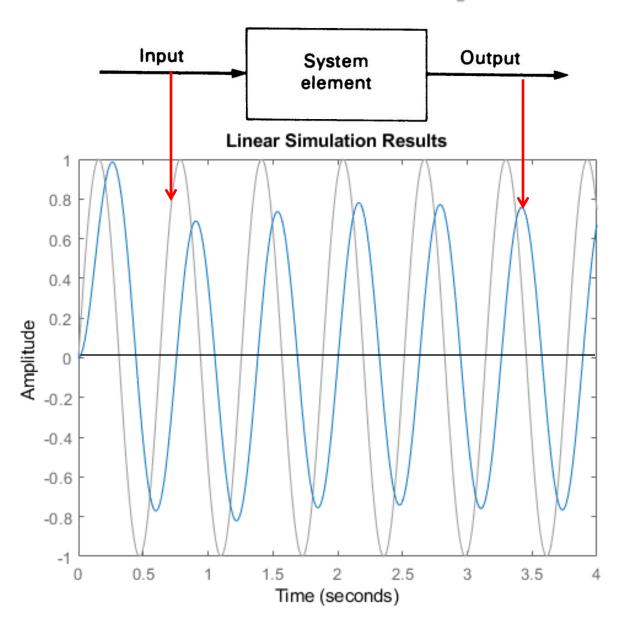
**Transient** 



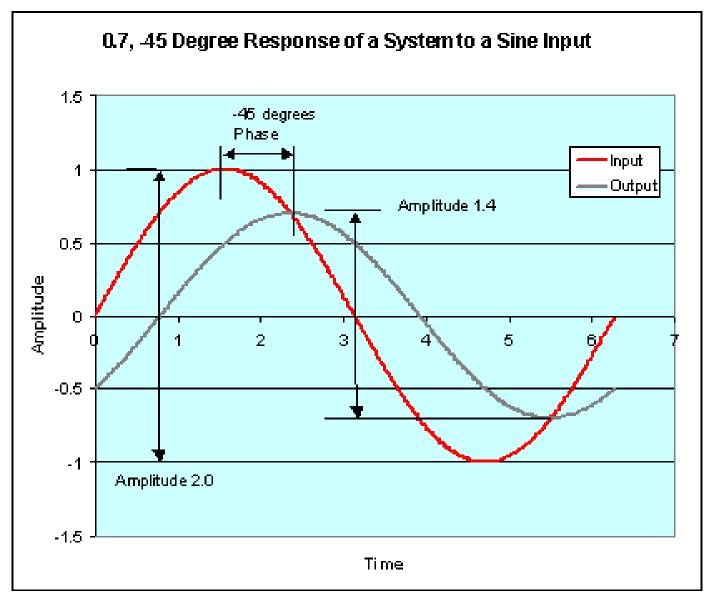
Periodic

Random

# Sinusoidal input



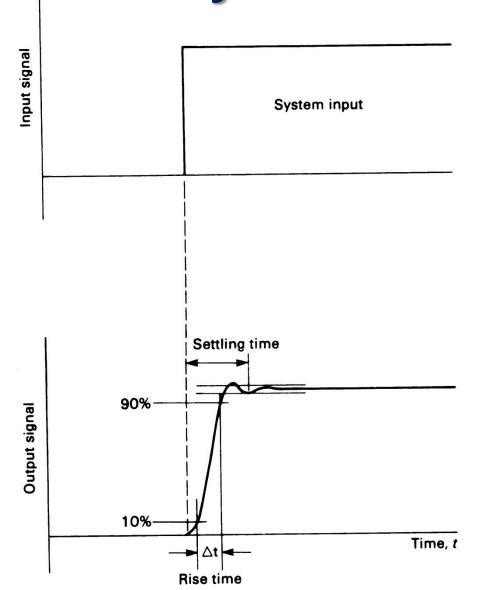
#### **Amplitude ratio and Phase difference**



Dynamic characteristics of an instrument can be obtained experimentally. Theoretical model is desirable for analysis and design.

#### Steps in understanding dynamic behavior

- a) Formulation of governing equation, relating input and output.
- b) Solution of governing equations, to study various input conditions
- c) To improve output response, if not satisfactory, by compensation.

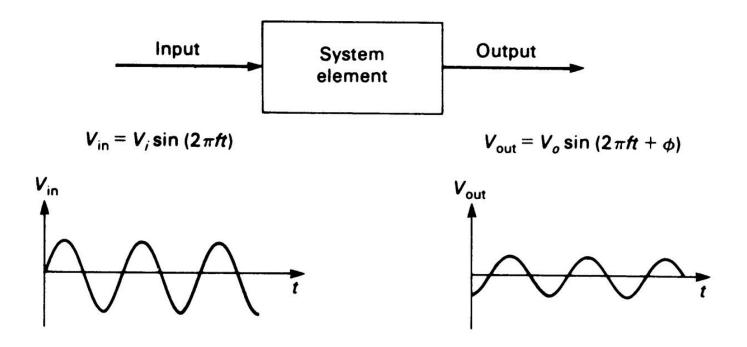


Rise Time: Time delay between the application of step input and proper output is reached. It can also be specified as time between 10% & 90% of the final output.

Overshoot: Maximum value of the output from the final output.

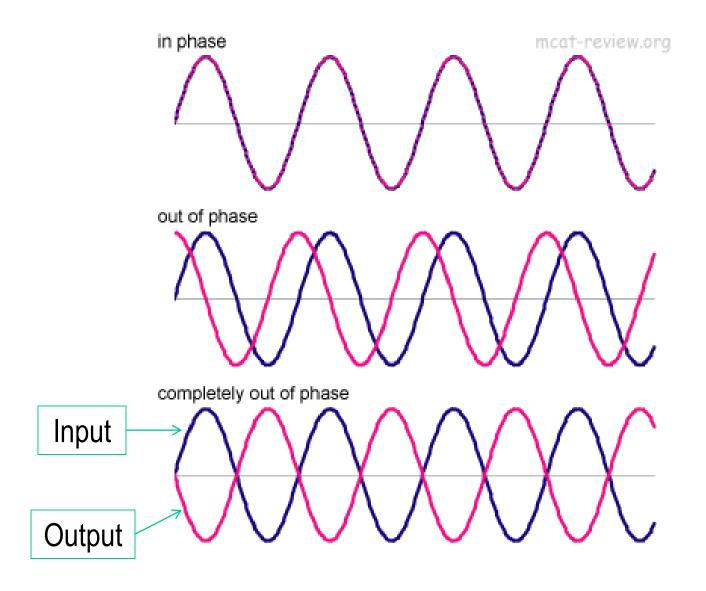
Settling time: Time required for the output to remain within some small percentage of final output.

Assumption: System is stable

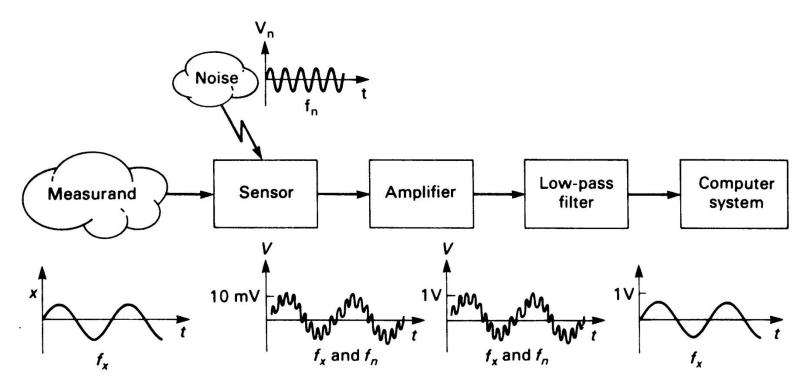


Output can differ in amplitude and phase. It is very important to study phase response of the measuring system when it is used for control, Amplitude and phase can be different for different input signal frequency.

# Out of phase signal



# Typical measuring system



Sensor output is a combination of sensor noise and input quantity. Noise is to be filtered such that original contents are preserved. System used for signal conditioning should have permissible noise and signal distortion parameters.

# **Dynamic System**

Input and output can be related in simplified form:

$$egin{aligned} & \mathbf{a}_{_{\mathbf{n}}} rac{\mathbf{d}^{_{\mathbf{n}}} \mathbf{q}_{_{_{0}}}}{\mathbf{d}t^{_{_{\mathbf{n}}}}} + \mathbf{a}_{_{\mathbf{n}^{-1}}} rac{\mathbf{d}^{_{\mathbf{n}^{-1}}} \mathbf{q}_{_{0}}}{\mathbf{d}t^{_{_{\mathbf{n}^{-1}}}}} + \dots + \mathbf{a}_{_{1}} rac{\mathbf{d} \mathbf{q}_{_{0}}}{\mathbf{d}t} + \mathbf{a}_{_{0}} \mathbf{q}_{_{0}} = \ & \mathbf{b}_{_{\mathbf{m}}} rac{\mathbf{d}^{_{\mathbf{m}}} \mathbf{q}_{_{\mathbf{i}}}}{\mathbf{d}t^{_{\mathbf{m}}}} + \mathbf{b}_{_{\mathbf{m}^{-1}}} rac{\mathbf{d}^{_{\mathbf{m}^{-1}}} \mathbf{q}_{_{\mathbf{i}}}}{\mathbf{d}t^{_{\mathbf{m}^{-1}}}} + \dots + \mathbf{b}_{_{1}} rac{\mathbf{d} \mathbf{q}_{_{\mathbf{i}}}}{\mathbf{d}t} + \mathbf{b}_{_{0}} \mathbf{q}_{_{\mathbf{i}}} \end{aligned}$$

 $q_0$  = Output quantity  $q_i$  = Input quantity a's, b's = system physical parameters assumed constant

Solution of above expression can be found out using standard mathematical techniques.

Most of the engineering systems can be simplified. No need to have such a complicated differential equation. Closed form or numerical methods can be used for solutions.

# **Dynamic System**

Input and output can be related in simplified form:

$$egin{aligned} & \mathbf{a}_{_{\mathbf{n}}} rac{\mathbf{d}^{_{\mathbf{n}}}\mathbf{q}_{_{0}}}{\mathbf{d}t^{_{\mathbf{n}}}} + \mathbf{a}_{_{\mathbf{n}^{-1}}} rac{\mathbf{d}^{_{\mathbf{n}^{-1}}}\mathbf{q}_{_{0}}}{\mathbf{d}t^{_{\mathbf{n}^{-1}}}} + \dots + \mathbf{a}_{_{1}} rac{\mathbf{d}\mathbf{q}_{_{0}}}{\mathbf{d}t} + \mathbf{a}_{_{0}}\mathbf{q}_{_{0}} = \ & \mathbf{b}_{_{\mathbf{m}}} rac{\mathbf{d}^{_{\mathbf{m}}}\mathbf{q}_{_{i}}}{\mathbf{d}t^{_{\mathbf{m}}}} + \mathbf{b}_{_{\mathbf{m}^{-1}}} rac{\mathbf{d}^{_{\mathbf{m}^{-1}}}\mathbf{q}_{_{i}}}{\mathbf{d}t^{_{\mathbf{m}^{-1}}}} + \dots + \mathbf{b}_{_{1}} rac{\mathbf{d}\mathbf{q}_{_{i}}}{\mathbf{d}t} + \mathbf{b}_{_{0}}\mathbf{q}_{_{i}} \end{aligned}$$

 $q_0$  = Output quantity  $q_i$  = Input quantity a's, b's = system physical parameters assumed constant

$$\mathbf{a}_{\scriptscriptstyle 0}\mathbf{q}_{\scriptscriptstyle 0}=\mathbf{b}_{\scriptscriptstyle 0}\mathbf{q}_{\scriptscriptstyle i}$$

Zeroth order system

$$\mathbf{a}_{\scriptscriptstyle 1} \, rac{\mathbf{d} \mathbf{q}_{\scriptscriptstyle 0}}{\mathbf{d} \mathbf{t}} + \mathbf{a}_{\scriptscriptstyle 0} \mathbf{q}_{\scriptscriptstyle 0} = \mathbf{b}_{\scriptscriptstyle 0} \mathbf{q}_{\scriptscriptstyle i}$$

First order system

$$\mathbf{a}_{2} \frac{\mathbf{d}^{2} \mathbf{q}_{0}}{\mathbf{d} \mathbf{t}^{2}} + \mathbf{a}_{1} \frac{\mathbf{d} \mathbf{q}_{0}}{\mathbf{d} \mathbf{t}} + \mathbf{a}_{0} \mathbf{q}_{0} = \mathbf{b}_{0} \mathbf{q}_{i}$$

Second order system

## Zero-order system

a<sub>0</sub> and b<sub>0</sub> are non zero in the generalised equation

$$\mathbf{a}_{\scriptscriptstyle 0}\mathbf{q}_{\scriptscriptstyle 0}=\mathbf{b}_{\scriptscriptstyle 0}\mathbf{q}_{\scriptscriptstyle 1}$$
 Output, input relationship

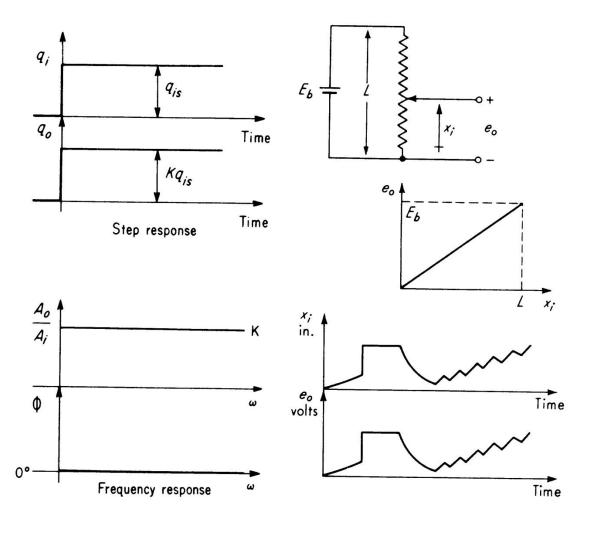
$$\mathbf{q}_{_0} = \frac{\mathbf{b}_{_0}}{\mathbf{a}_{_0}} \mathbf{q}_{_i} = \mathbf{K} \mathbf{q}_{_i}$$
 Output is related to input by a single quantity K

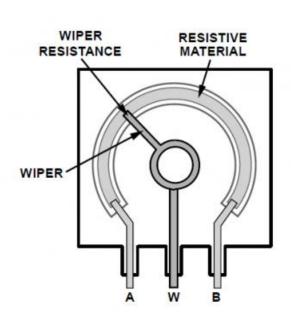
$$\mathbf{K} = \frac{\mathbf{b}_0}{\mathbf{a}_0}$$
 Static sensitivity, independent of input signal

Output follows input without any distortion and lag. Zero-order instrument represents ideal or perfect dynamic performance. Zero-order system can have non-linear gain.

## Zero-order system

Potentiometer is a zero-order instrument for all practical purpose. Potentiometer do have a very small capacitance and inductance can become critical for very high frequency applications.





a<sub>1</sub>, a<sub>0</sub> and b<sub>0</sub> are non zero in the generalised equation. Any instrument which can be represented using this equation is a first-order instrument.

$$\mathbf{a}_{\scriptscriptstyle 1} \frac{\mathbf{d}\mathbf{q}_{\scriptscriptstyle 0}}{\mathbf{d}t} + \mathbf{a}_{\scriptscriptstyle 0} \mathbf{q}_{\scriptscriptstyle 0} = \mathbf{b}_{\scriptscriptstyle 0} \mathbf{q}_{\scriptscriptstyle i}$$

 $\mathbf{a}_{1} \frac{\mathbf{dq}_{0}}{\mathbf{dt}} + \mathbf{a}_{0} \mathbf{q}_{0} = \mathbf{b}_{0} \mathbf{q}_{i}$  Output, input relationship, three parameters  $\mathbf{a}_{1, a_{0}}$  and  $\mathbf{b}_{0}$ 

$$\frac{\mathbf{a}_{_{\scriptscriptstyle 1}}}{\mathbf{a}_{_{\scriptscriptstyle 0}}}\frac{\mathbf{d}\mathbf{q}_{_{\scriptscriptstyle 0}}}{\mathbf{d}t}+\mathbf{q}_{_{\scriptscriptstyle 0}}=\frac{\mathbf{b}_{_{\scriptscriptstyle 0}}}{\mathbf{a}_{_{\scriptscriptstyle 0}}}\mathbf{q}_{_{\scriptscriptstyle i}}$$

$$\frac{\mathbf{a}_{_{0}}}{\mathbf{a}_{_{0}}}\frac{\mathbf{dq}_{_{0}}}{\mathbf{dt}} + \mathbf{q}_{_{0}} = \frac{\mathbf{b}_{_{0}}}{\mathbf{a}_{_{0}}}\mathbf{q}_{_{i}} \qquad \mathbf{K} = \frac{\mathbf{b}_{_{0}}}{\mathbf{a}_{_{0}}} \quad \text{Static sensitivity}$$

$$(\tau \mathbf{D} + 1)\mathbf{q}_{0} = \mathbf{K}\mathbf{q}_{i}$$

$$au = rac{{f a}_1}{{f a}_2}$$
 Time constant

$$\mathbf{D} = \frac{\mathbf{d}}{\mathbf{dt}}$$
 Operator

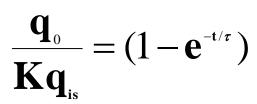
$$\frac{\mathbf{q}_{_0}}{\mathbf{q}_{_i}}(\mathbf{D}) = \frac{\mathbf{K}}{\tau \mathbf{D} + 1}$$
 Transfer function for the first order system

$$q_0 = 0;$$
 at  $t = 0$ 

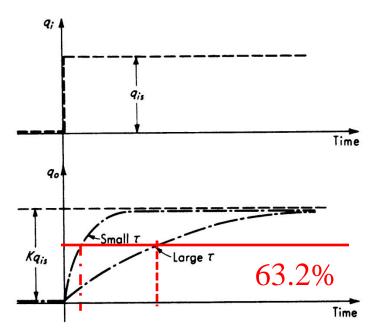
$$q_i = q_{is};$$
 for  $t > 0$ 

$$\mathbf{q}_{\scriptscriptstyle 0} = \mathbf{K}\mathbf{q}_{\scriptscriptstyle \mathbf{is}}(1 - \mathbf{e}^{-\mathbf{t}/\tau})$$

 $\mathbf{q}_{0} = \mathbf{K}\mathbf{q}_{10}(1-\mathbf{e}^{-\mathbf{t}/\tau})$  Solution of the equation for a step function



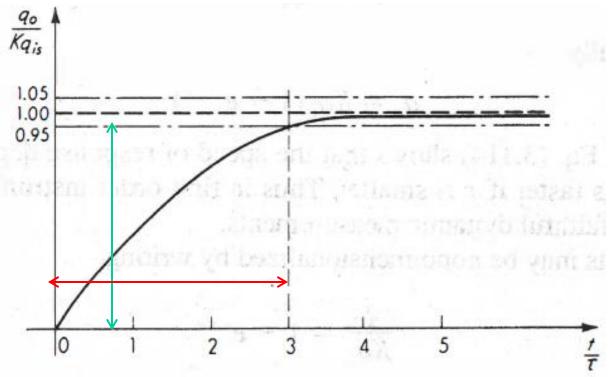
In non-dimensional form



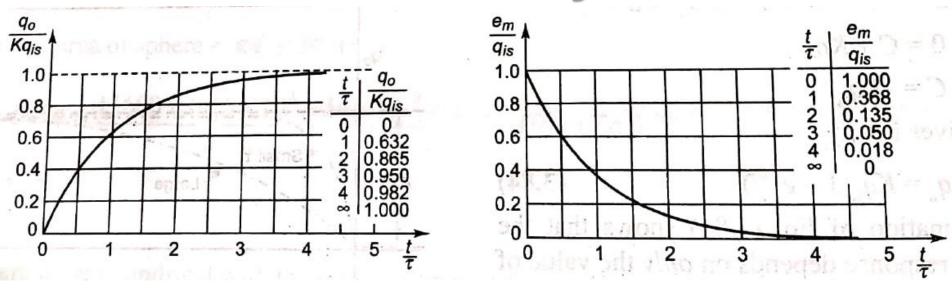
$$\mathbf{e}_{_{\mathbf{m}}} = \mathbf{q}_{_{\mathbf{i}}} - \frac{\mathbf{q}_{_{0}}}{\mathbf{K}}$$

Measurement error

$$\mathbf{e}_{_{\mathbf{m}}} = \mathbf{q}_{_{\mathbf{i}\mathbf{s}}} - \mathbf{q}_{_{\mathbf{i}\mathbf{s}}} (1 - \mathbf{e}^{^{-\mathbf{t}/\tau}}) \qquad \qquad \frac{\mathbf{e}_{_{\mathbf{m}}}}{\mathbf{q}_{_{\mathbf{i}\mathbf{s}}}} = \mathbf{e}^{^{-\mathbf{t}/\tau}}$$

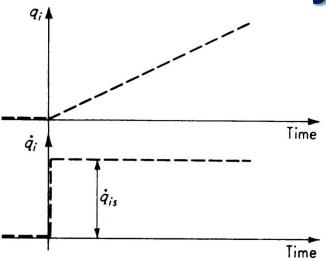


In one time constant output is 63.2% of input. To achieve output within 5% of settling time, 3 time constants are required. For fast response, time constant should be less



In one time constant output is 63.2% of input. To achieve output within 5% of settling time, 3 time constants are required. Error reduces as more time is given to achieve steady state. Ideally infinite time is required. For practical purpose four to five time constant is good enough. For fast response, time constant should be less.

# First-order system - Ramp response



$$\mathbf{q}_{_{0}} = \mathbf{K}\dot{\mathbf{q}}_{_{\mathbf{i}\mathbf{s}}}(\boldsymbol{\tau}\mathbf{e}^{^{-\mathbf{t}/\boldsymbol{\tau}}} + \mathbf{t} - \boldsymbol{\tau})$$

$$\frac{q_0}{K} = \dot{q}_{is}(\tau e^{-t/\tau} + t - \tau)$$

$$\frac{q_i}{K} = \dot{q}_{is}(\tau e^{-t/\tau} + t - \tau)$$

$$\frac{q_i}{K} = \dot{q}_{o}$$

$$\frac{q_o}{K}$$

$$\frac{\text{Steady-state}}{\text{time lag = }\tau}$$

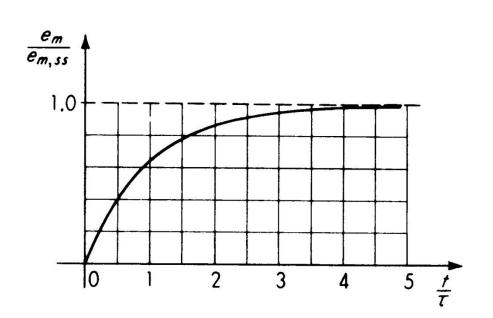
$$-e_{m, ss} = \dot{q}_{is}\tau$$
Time

$$\mathbf{q}_{i} = \begin{cases} \mathbf{q}_{0} = 0 & \mathbf{t} \leq 0 \\ \dot{\mathbf{q}}_{is} \mathbf{t} & \mathbf{t} \geq 0 \end{cases}$$

$$\mathbf{e}_{m} = -\dot{\mathbf{q}}_{is} \mathbf{\tau} \mathbf{e}^{-\mathbf{t}/\tau} + \dot{\mathbf{q}}_{is} \mathbf{\tau}$$
Transienterror Steadystate

First term gradually disappears – transient error. Second term persist for ever – steady state error.

error e<sub>m.ss</sub>

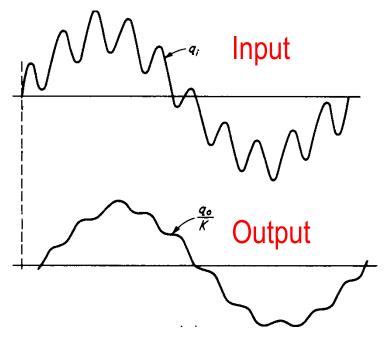


#### First-order system – sinusoidal inputs

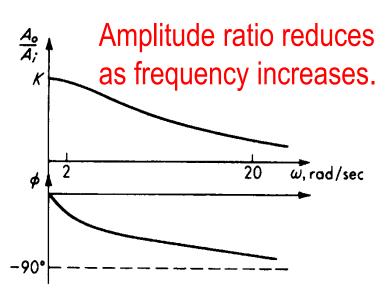
For sinusoidal input, input and output can be given as

$$\frac{q_0}{q_i}(i\omega) = \frac{K}{i\omega\tau + 1} = \frac{K}{\sqrt{\omega^2\tau^2 + 1}} \angle [\tan^{-1}(-\omega\tau)]$$

$$\frac{q_0}{q_i}(i\omega) = \frac{K}{i\omega\tau + 1} = \frac{K}{\sqrt{\omega^2\tau^2 + 1}} \angle \left[\tan^{-1}(-\omega\tau)\right] \qquad \frac{A_0}{A_i} = \left|\frac{q_0}{q_i}(i\omega)\right| = \frac{K}{\sqrt{\omega^2\tau^2 + 1}} \qquad \phi = \angle \frac{q_0}{q_i}(i\omega) = \tan^{-1}(-\omega\tau)$$



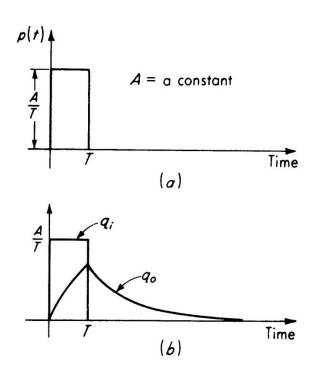
Inadequate frequency response



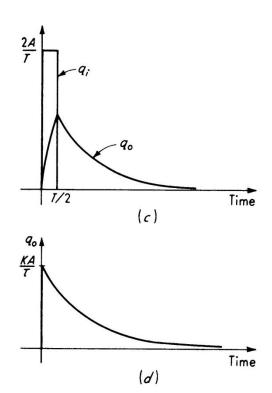
Output gain and phase depends on frequency.

For good frequency response time constant should be small.

#### First-order system – Impulse response



Same response as step input till the pulse is high. It never reaches the required output.



Exact impulse is infinite amplitude, derivative of step function

#### Second-order system

a<sub>2</sub> a<sub>1</sub>, a<sub>0</sub> and b<sub>0</sub> are non zero in the generalised equation. Any instrument which can be represented using this equation is a second - order instrument.

$$\mathbf{a}_{2} \frac{\mathbf{d}^{2} \mathbf{q}_{0}}{\mathbf{d} \mathbf{t}^{2}} + \mathbf{a}_{1} \frac{\mathbf{d} \mathbf{q}_{0}}{\mathbf{d} \mathbf{t}} + \mathbf{a}_{0} \mathbf{q}_{0} = \mathbf{b}_{0} \mathbf{q}_{i}$$
Roots of this equation can be imaginary, complex, real

Above equation can be written as

$$\mathbf{K} = \frac{\mathbf{b}_{\scriptscriptstyle 0}}{\mathbf{a}_{\scriptscriptstyle 0}}$$

$$\boldsymbol{\omega}_{\scriptscriptstyle \mathrm{n}} = \sqrt{\frac{\mathbf{a}_{\scriptscriptstyle 0}}{\mathbf{a}_{\scriptscriptstyle 2}}}$$

$$\left(\frac{\mathbf{D}^{2}}{\boldsymbol{\omega}_{n}^{2}} + \frac{2\boldsymbol{\xi}\mathbf{D}}{\boldsymbol{\omega}_{n}} + 1\right)\mathbf{q}_{0} = \mathbf{K}\mathbf{q}_{i}$$

$$\boldsymbol{\xi} = \frac{\mathbf{a}_{1}}{2\sqrt{\mathbf{a}_{0}\mathbf{a}_{2}}}$$

Static sensitivity

Undamped natural frequency

Damping ratio

Transfer function

$$\frac{\mathbf{q}_{0}}{\mathbf{q}_{i}}(\mathbf{D}) = \frac{\mathbf{K}}{\mathbf{D}^{2}/\boldsymbol{\omega}_{n}^{2} + 2\boldsymbol{\xi}\mathbf{D}/\boldsymbol{\omega}_{n} + 1}$$

#### Second-order system – an example

Force measuring spring

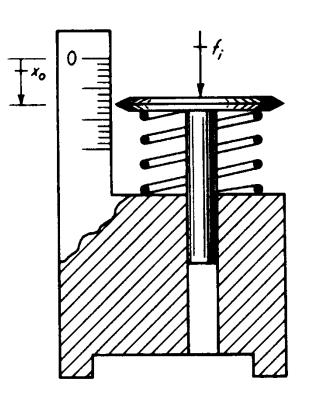
Total mass – M

Spring constant – K<sub>s</sub>

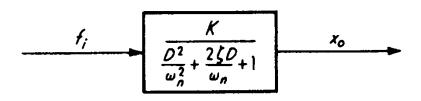
Damping - B

$$(\mathbf{M}\mathbf{D}^2 + \mathbf{B}\mathbf{D} + \mathbf{K}_{s})\mathbf{X}_{0} = \mathbf{f}_{s}$$

$$\mathbf{K} = \frac{1}{\mathbf{K}_{s}} \qquad \boldsymbol{\omega}_{n} = \sqrt{\frac{\mathbf{K}_{s}}{\mathbf{M}}} \qquad \boldsymbol{\xi} = \frac{\mathbf{B}}{2\sqrt{\mathbf{K}_{s}\mathbf{M}}}$$

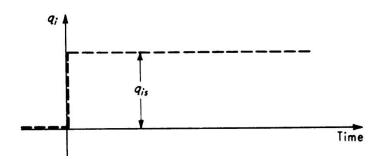


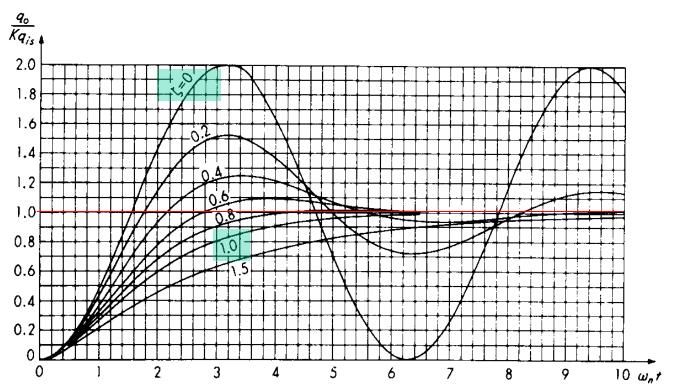
 $\omega_{\text{n}}$  is direct indication of speed of response



#### Second-order system – Step response

$$\left(\frac{\mathbf{D}^{2}}{\boldsymbol{\omega}_{n}^{2}} + \frac{2\boldsymbol{\xi}\mathbf{D}}{\boldsymbol{\omega}_{n}} + 1\right)\mathbf{q}_{0} = \mathbf{K}\mathbf{q}_{is}$$





Observe the time response for different damping ratio.

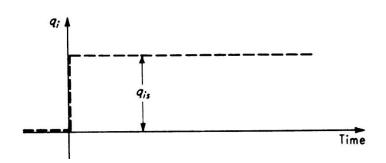
#### Second-order system – Step response

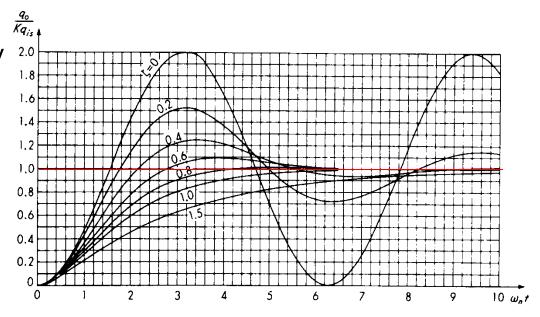
$$\left(\frac{\mathbf{D}^{2}}{\boldsymbol{\omega}_{n}^{2}} + \frac{2\boldsymbol{\xi}\mathbf{D}}{\boldsymbol{\omega}_{n}} + 1\right)\mathbf{q}_{0} = \mathbf{K}\mathbf{q}_{is}$$

Under damping – Damping ratio less than 1. This will always give overshoot before reaching steady state

Critical damping – Damping ratio equal to 1. this will take minimum time to achieve steady state output without overshoot

Over damping – Damping ratio greater than 1.this will take more time to achieve steady state compared to critical damping output without overshoot



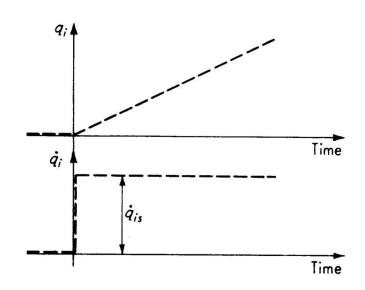


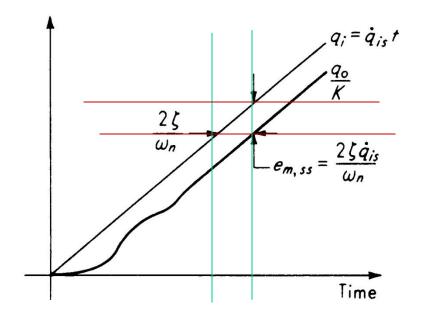
#### Second-order system – Ramp response

Steady state error  $2\xi\dot{\mathbf{q}}_{\scriptscriptstyle \mathrm{is}}/\boldsymbol{\omega}_{\scriptscriptstyle \mathrm{n}}$ 

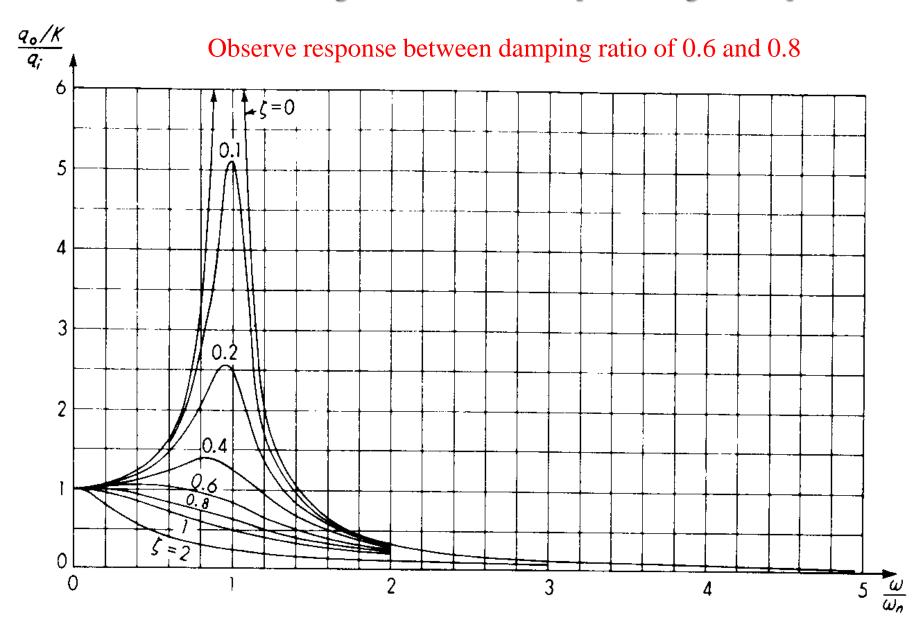
Time lag  $2\xi/\omega_{n}$ 

Steady state error can be reduced by reducing  $\xi$  and increasing  $\omega_n$  This will increase the oscillations in the output





#### Second-order system – frequency response



#### Second-order system – frequency response

Amplitude ratio strongly related to small and large damping ratio.

Flat response for damping of 0.6-0.7

Linear variation of phase for above damping

Damping in the above range is a good choice for second order system

Frequency up to 0.4 of natural frequency

