

Spacecraft Motion Basics



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End point of an ascent mission is the start point for the orbital mission.

Similar to ascent mission, orbital mission also can be divided into segments.

First segment is immediately after separation, when it is normally put into a temporary orbit called LEO or GTO.



Orbital Mission Phases

Next, a **series** of orbital **manoeuvres** are carried out in order to **put** spacecraft into the **final** or parking **orbit**.

However, for **inter-planetary** missions, manoeuvres **aim** to put spacecraft on a **trajectory** for other planets.



Spacecraft Motion Scenarios

Upon **injection**, spacecraft **moves** as per its **initial** conditions and **forces** experienced by it.

Thus, in order to **predict** motion of spacecraft from this point **onwards**, there a need to **create** mathematical **models**, that represent the above **motion** scenario.



Context of Spacecraft Motion

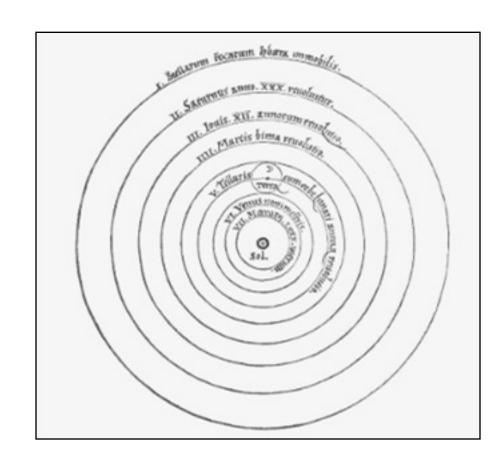
As **spacecrafts** operate mainly in our **solar system**, forces present in it **need** to be modelled **suitably.**

In this regard, it is **useful** to understand the **nature** of motion of **planets**, as these are also under **similar** forces.



Copernican Motion Model

Copernicus in 1543, established **sun** as centre, with all the planets **orbiting** around it.





Kepler's Motion Hypothesis

Kepler subsequently formulated the **laws** of planetary **motion**, which are given **alongside**.

These laws are generic and can be applied to any orbit, including Earth-satellite system.

- 1. All **planets** move around the **sun** in planar and **elliptic** orbits, with sun as one **focus**.
- 2. Radius vector from sun centre to planet centre sweeps equal areas in equal time.
- 3. (**Time period** of the orbit)²
- \propto (Semi-major axis)³



Planetary Motion Features

Orbital solution, as proposed by **Kepler**, is based on the **premise** that all **objects** in space form an **orbit**, which can be **observed**.

However, there are no **explicit** guarantees in Kepler's laws that all **objects**, if left in space, will form **orbits**.



Planetary Motion Features

This **issue** was settled by **Newton**, who formally derived the **Kepler's laws**, using his own three **laws** of motion.

This also led to the **evolution** of particle motion **theory**, which is central to the **orbital** mechanics.



Space Force System



Force Field in Space

In outer **space**, there is practically no **atmosphere** and therefore no **aerodynamic** forces are present.

Also, spacecraft has **limited** propulsion capability, which is used only for short-duration **manoeuvres**, resulting in no **propulsive** forces either.



Force Field in Space

Further, forces due to **magnetic** field and solar **radiation** are very **small** and have **impact** only on the long term **drift** of the satellite from its **path**, and hence, ignored.

Thus, **gravity** is the only **important** force, which **primarily** determines the spacecraft **motion**.



Gravitational Model

In **space**, general form of the **universal** law of gravitation is **applicable**.

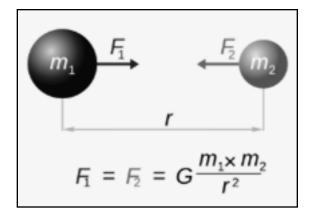
Further, as **sizes** of planets are much **smaller** in relation to their **mutual** distances, (as indicated by fact sheet), we invoke **spherical** symmetry and **point mass** assumptions.

	MERCURY	<u>VENUS</u>	EARTH	MOON	MARS	<u>JUPITER</u>	SATURN
Dia (km)	4879	12,104	12,756	3475	6792	142,984	120,536
Sun Distance (10 ⁶ km)	57.9	108.2	149.6	0.384*	227.9	778.6	1433.5



Applicable Gravitational Model

Space gravity model is also based on the **universal** law of gravitation, which is **defined** as follows.



$$\vec{F}_g = -\frac{Gm_1m_2}{r^3}\vec{r} = -\frac{\mu m_2}{r^3}\vec{r}$$

$$\mu = Gm_1; \quad \Phi_g = -\frac{\mu}{r}$$

$$g = \text{grad}(\Phi_g)$$



Summary

The **motion** of spacecraft upon **injection** is similar to the **planetary** motion features and is broadly **governed** by the Kepler's laws.

Further, the **force** system, for most **space** based objects, consists only of the **gravitational** force.