

**Aerospace Engineering Department, IIT Bombay**  
**AE 308 & AE 775 - Control Theory**  
**Tutorial 4 Solution**

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**Q1**

For a unity feedback system with the forward transfer function

$$G(s) = \frac{K(s+20)}{s(s+2)(s+3)},$$

find the range of  $K$  to make the system stable.

**Solution:**

$$0 < K < 2.$$

**Q2**

Use the Routh-Hurwitz criterion to find how many poles of the following closed-loop system,  $T(s)$ , are in the rhp, in the lhp, and on the  $jw$ -axis:

$$T(s) = \frac{s^3 + 7s^2 - 21s + 10}{s^6 + s^5 - 6s^4 - s^2 - s + 6}.$$

**Solution:**

Two poles in the right-half plane, two in the left-half plane, and two on the  $jw$ -axis.

**Q3**

A unity feedback control system has the open-loop transfer function

$$G(s) = \frac{A}{s(s+a)}.$$

Compute the sensitivity of the closed-loop transfer function to changes in the parameters  $A$  and  $a$ .

**Solution:**

Closed-loop transfer function:

$$T(s) = \frac{A}{s^2 + as + A}.$$

Sensitivity of  $T(s)$  w.r.t  $A$  is denoted by  $\mathcal{S}_A^T$ , and defined as:

$$\mathcal{S}_A^T := \frac{\frac{\partial T}{\partial A}}{\frac{T}{A}}, \quad \mathcal{S}_A^T = \frac{s^2 + as}{s^2 + as + A}.$$

Similarly,

$$\mathcal{S}_a^T := \frac{\frac{\partial T}{\partial a}}{\frac{T}{a}}, \quad \mathcal{S}_a^T = \frac{-as}{s^2 + as + A}.$$

**Q4**

Consider the second-order plant with the transfer function

$$G(s) = \frac{1}{(s+1)(5s+1)},$$

in a unity feedback structure. Determine the system type and error constant with respect to tracking polynomial reference inputs of the system for PID  $[D_c = 19 + \frac{0.5}{s} + \frac{4}{19}s]$ .

**Solution:**

System type is 1. Error constant  $K_v = 0.5$ .

**Q5**

Let the transfer function of the plant is  $G(s) = \frac{K_0}{4s+1}$ . The system is in a unity feedback structure. Compute the steady-state error of the closed-loop plant when reference is a step input. Design a controller to make the steady-state error zero.

**Solution:**

$$\text{Steady-State error: } \frac{1}{1+K_0}.$$

An integral controller is used to make the steady-state error zero.

$$\text{Controller: } D_c = \frac{K_I}{s}, \text{ where } K_I \text{ is any non-zero real value.}$$

with this controller steady-state error is 0.

**Q6**

Consider a plant with nominal model given by  $G(s) = \frac{1}{s+2}$ . Compute the parameters of a PI controller so that the natural modes of the closed loop response decay at least as fast as  $e^{-5t}$ .

**Solution:**

A PI controller has transfer function given by:  $C(s) = \frac{as+b}{s}$ , Where  $a = K_p$ ,  $b = \frac{K_p}{T_r}$ ,

The closed loop characteristic polynomial,  $A_c(s) = \text{numerator of } 1+G_o(s)C(s) = s^2 + (2+a)s + b$

We choose the controller to obtain a pair of complex conjugate poles. To achieve a closed loop transient as fast as  $e^{-5t}$ , those poles must have real parts equal to -5. This requires  $a = -8$  and take  $b = 49$ . Hence

$$C(s) = \frac{(8s+49)}{s}$$

## Q7

Consider the system shown in Figure 1. This is a PID control of a second-order plant  $G(s)$ . Assume that disturbances  $D(s)$  enter the system as shown in the diagram. It is assumed that the reference input  $R(s)$  is normally held constant, and the response characteristics to disturbances are a very important consideration in this system. Design a control system such that the response to any step disturbance will be damped out quickly (in 2 to 3 sec in terms of the 2% settling time). Choose the configuration of the closed-loop poles such that there is a pair of dominant closed-loop poles.

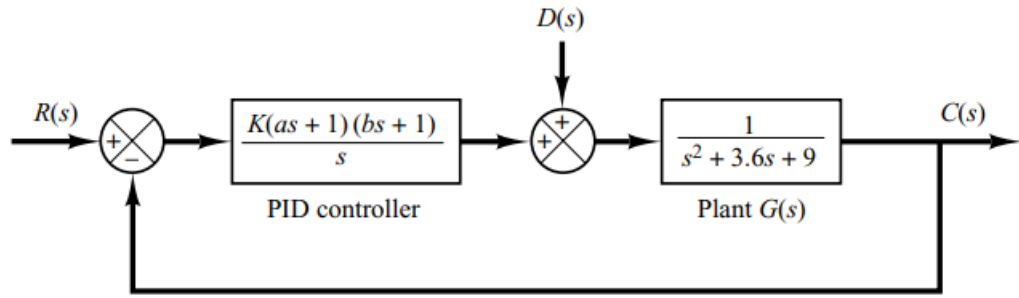


Figure 1:

**Solution:**

The PID controller has the transfer function  $G_c(s)$ :  $\frac{k(as+1)(bs+1)}{s}$ ,

For the disturbance input in the absence of the reference input, the closed-loop transfer function becomes:

$$\frac{C_d(s)}{D(s)} = \frac{s}{s(s^2+3.6s+9)+k(as+1)(bs+1)} = \frac{s}{s^3+(3.6+kab)s^2+(9+ka+kb)s+k}$$

The specification requires that the response to the unit-step disturbance be such that the settling time be 2 to 3 sec and the system have a reasonable damping. We may interpret the specification as  $\zeta = 0.5$  and  $\omega_n = 4$  rad/s for the dominant closed-loop poles. We may choose the third pole at  $s = -10$  so that the effect of the real pole on the response is small. Then the desired characteristic equation can be written as

$$(s+10)(s^2+2 \times 0.5 \times 4s+4^2) = (s+10)(s^2+4s+16) = s^3+14s^2+56s+160$$

The characteristic equation of the system is

$$s^3 + (3.6 + Kab)s^2 + (9 + Ka + Kb)s + K$$

Hence, we require

$$3.6 + Kab = 14$$

$$9 + Ka + Kb = 56$$

$$K = 160$$

Which yields  $ab = 0.065$ ,  $a + b = 0.29375$

The PID controller now becomes

$$G_c(s) = \frac{K(abs^2) + (a+b)s + 1}{s} = \frac{160(0.065s^2 + 0.29375s + 1)}{s} = \frac{10.4(s^2 + 4.5192s + 15.385)}{s}$$

## Q8

The block diagram of a control system with a series controller is shown in Figure 2. Find the transfer function of the controller  $G_c(s)$  so that the following specifications are satisfied:

1. The ramp error constant  $K_v$  is 5.
2. The closed loop transfer function is of the form  $M(s) = \frac{Y(s)}{R(s)} = \frac{K}{(s^2 + 20s + 200)(s + a)}$  where  $K$  and  $a$  are real constants. Find the values of  $K$  and  $a$ .

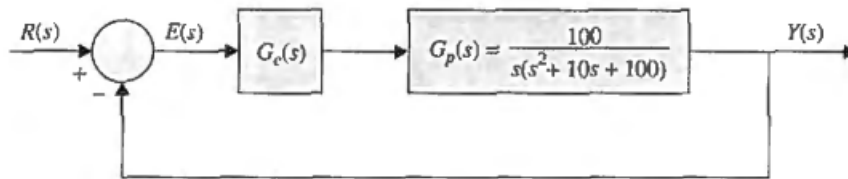


Figure 2:

**Solution:**

The forward-path transfer function

$$G(s) = \frac{M(s)}{1-M(s)} = \frac{K}{s^3 + (20+a)s^2 + (200+20a)s + 200a - K}$$

For type 1 system,  $200a - K = 0$

Thus  $K = 200a$

Ramp-error constant:

$$K_v = \lim_{s \rightarrow 0} sG(s) = \frac{K}{200+20a} = \frac{200a}{200+20a} = 5$$

Thus  $a = 10$ ,  $K = 2000$

The forward-path transfer function is

$$G(s) = \frac{2000}{s(s^2+30s+400)}$$

The controller transfer function is

$$G_c(s) = \frac{G(s)}{G_p(s)} = \frac{20(s^2+10s+100)}{s^2+30s+400}$$