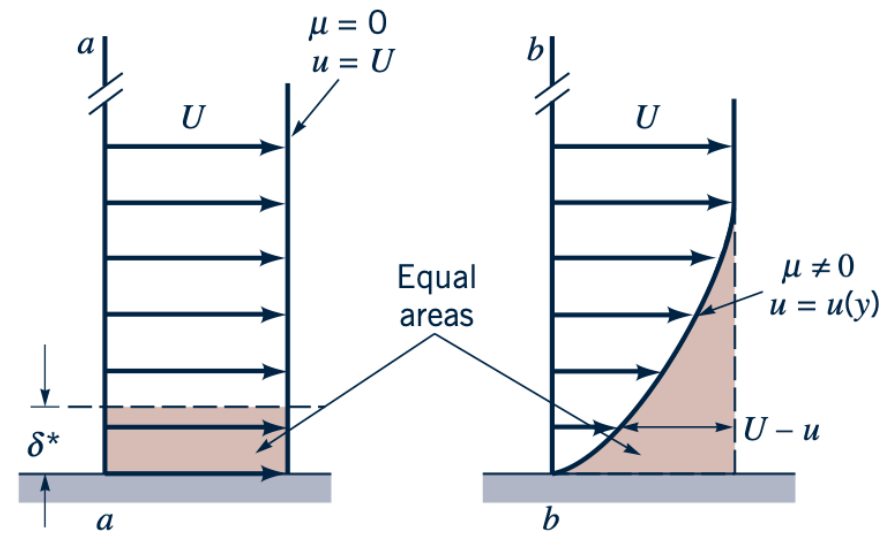
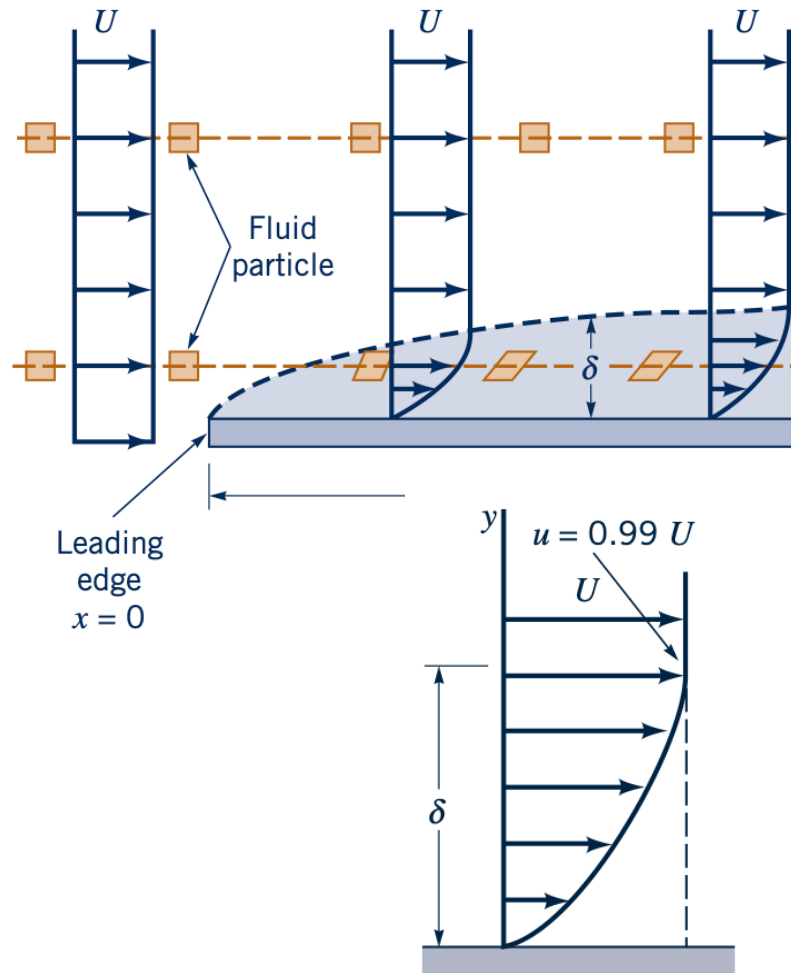


AE 330 Rocket Propulsion Boundary Layers & Turbulence

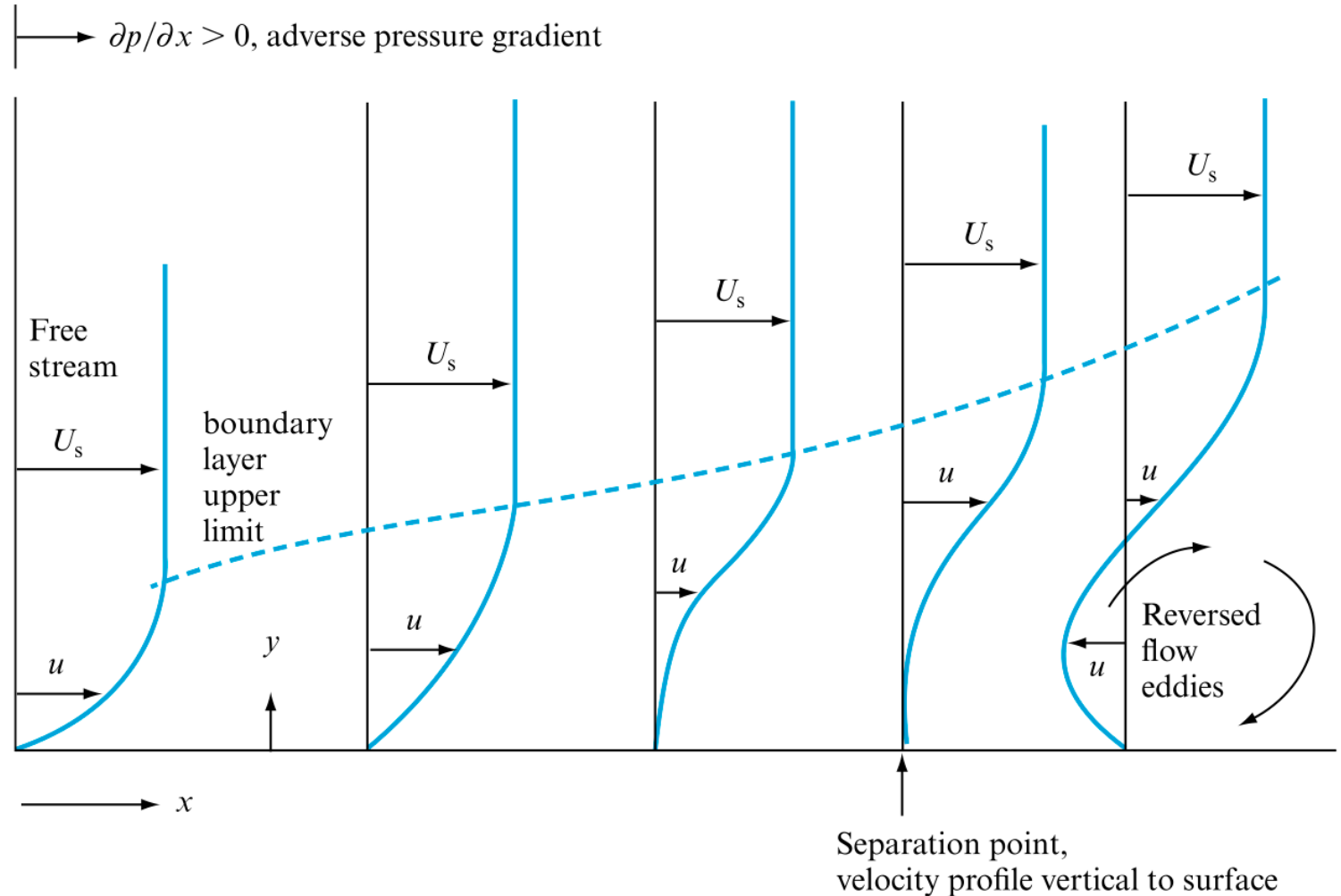
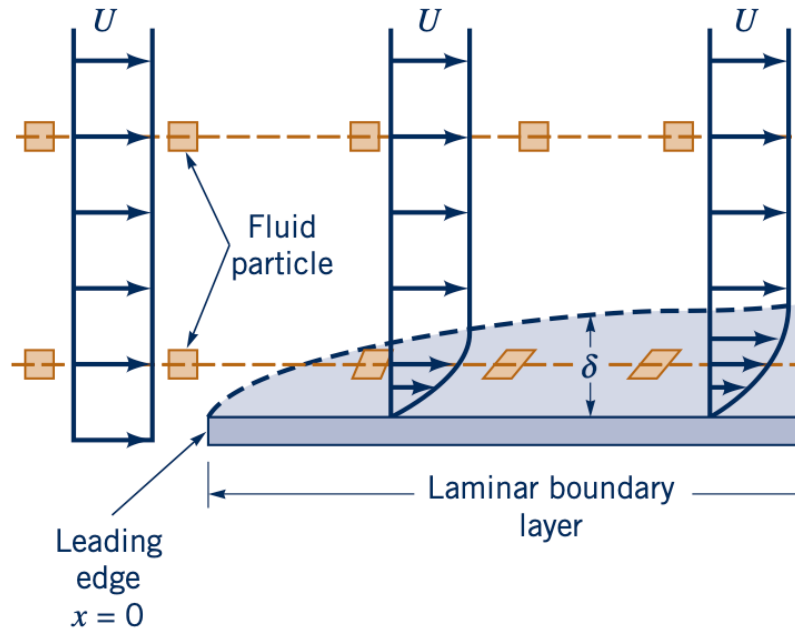
Kowsik Bodi

Aerospace Engineering, IIT Bombay

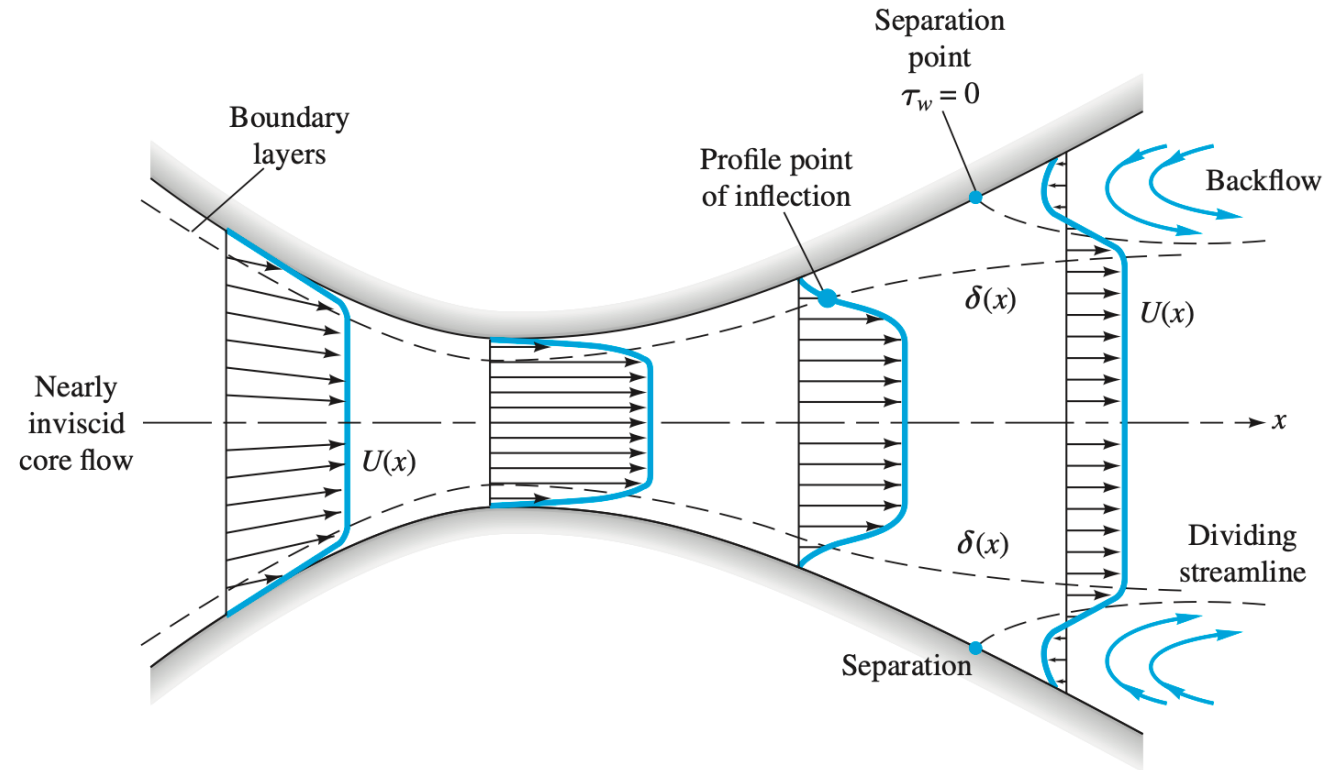
Displacement Thickness



Flow Separation



Flow Separation



Nozzle:
Decreasing
pressure
and area

Increasing
velocity

Favorable
gradient

Throat:
Constant
pressure
and area

Velocity
constant

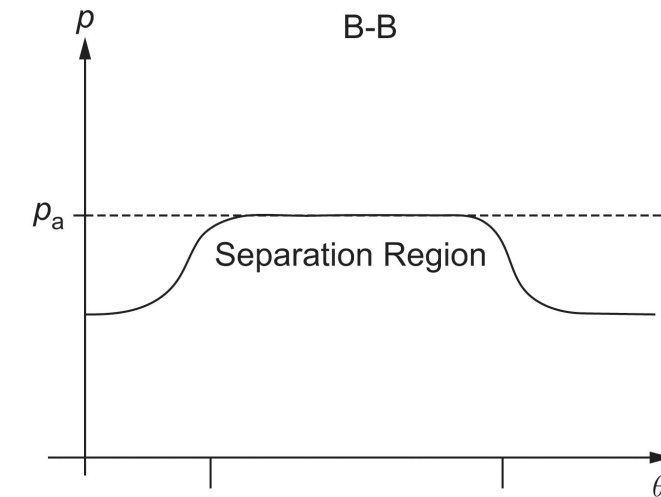
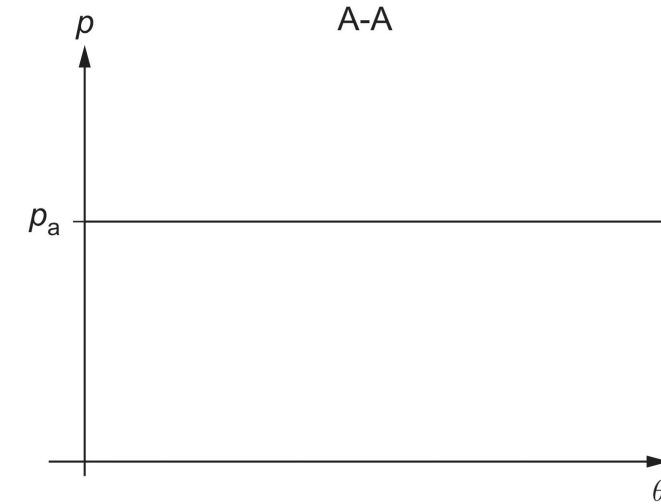
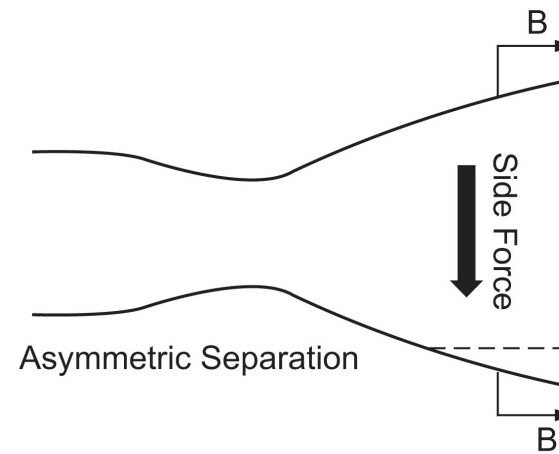
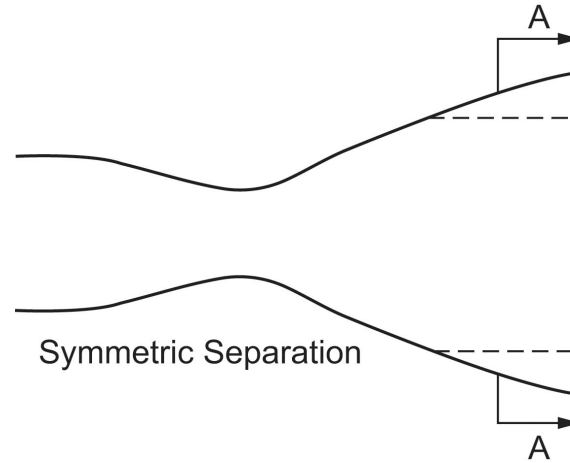
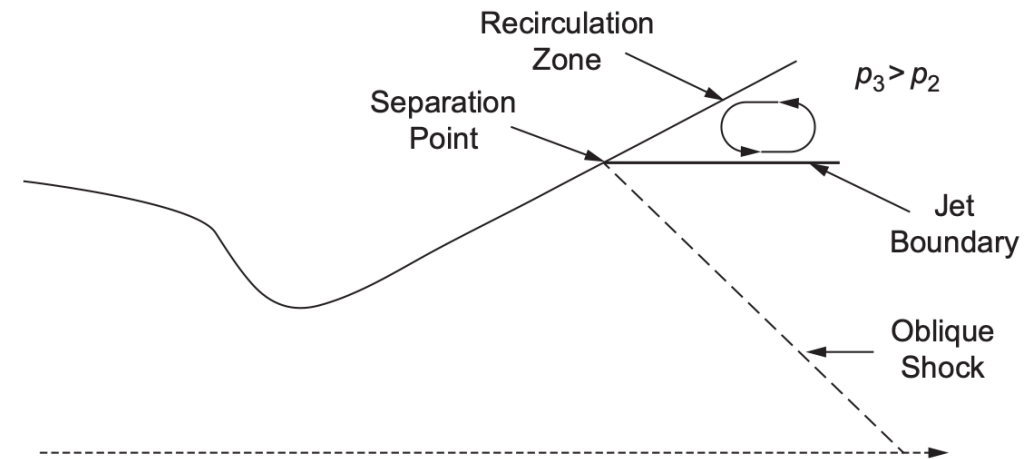
Zero
gradient

Diffuser:
Increasing pressure
and area

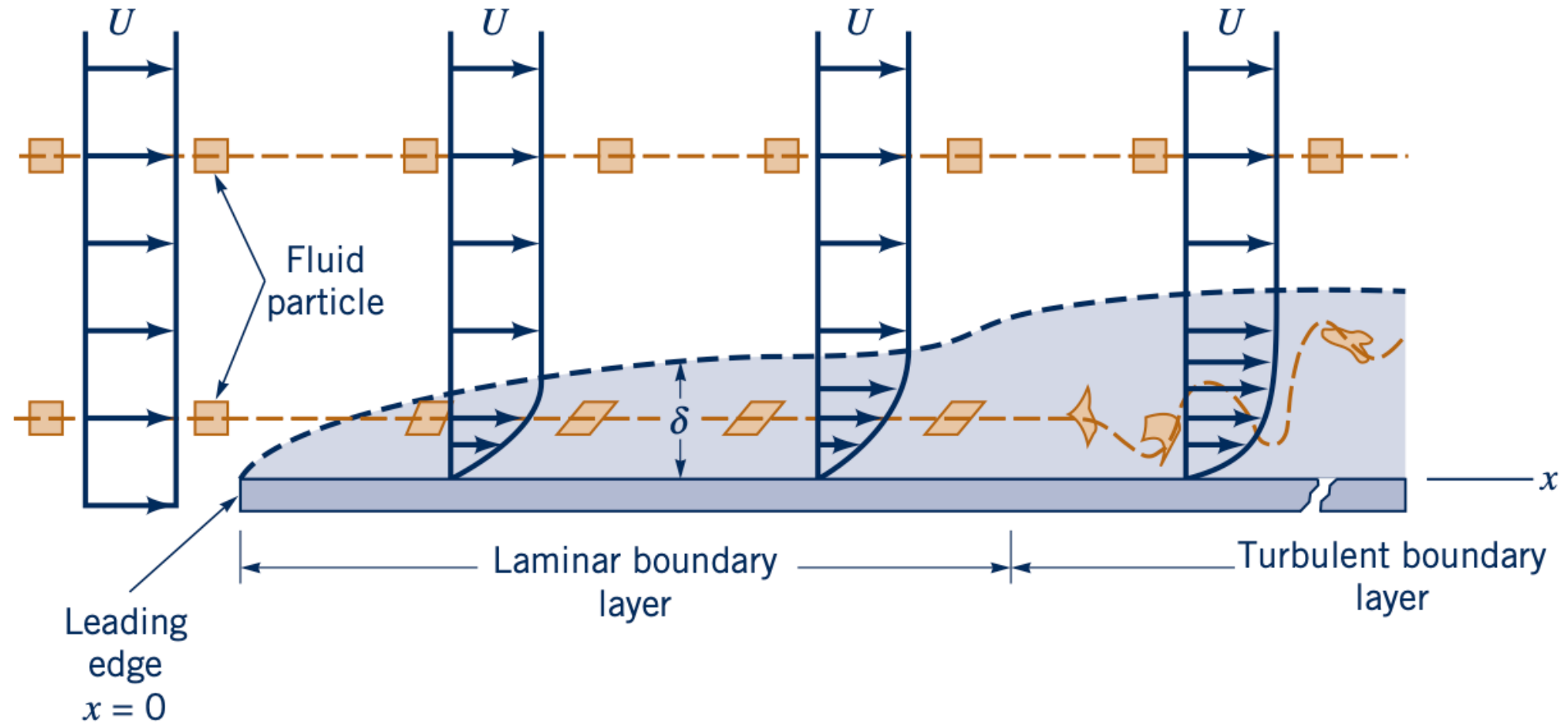
Decreasing velocity

Adverse gradient
(boundary layer thickens)

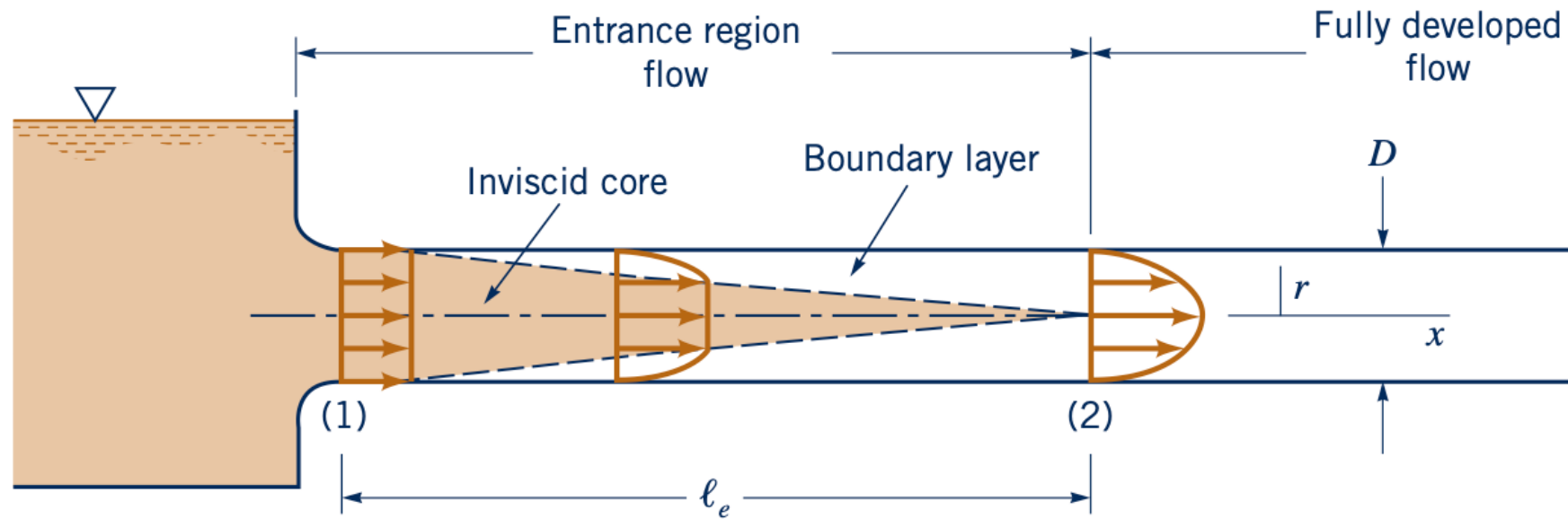
Flow Separation



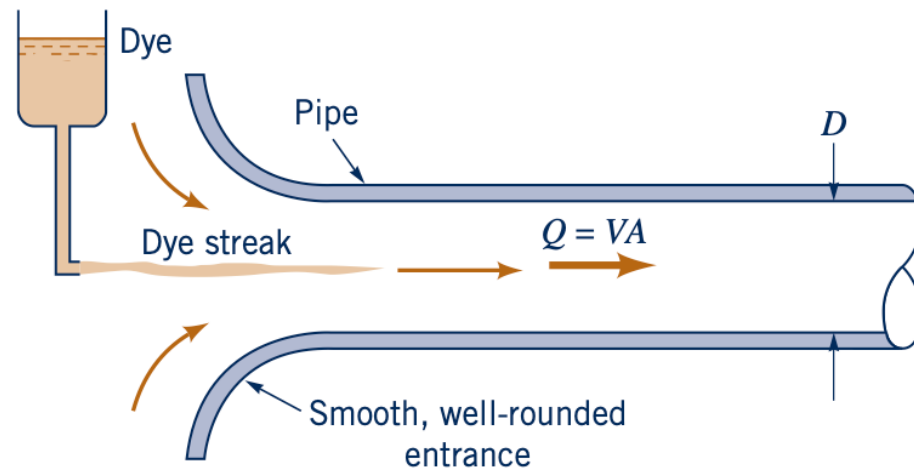
Flat-plate Boundary Layer



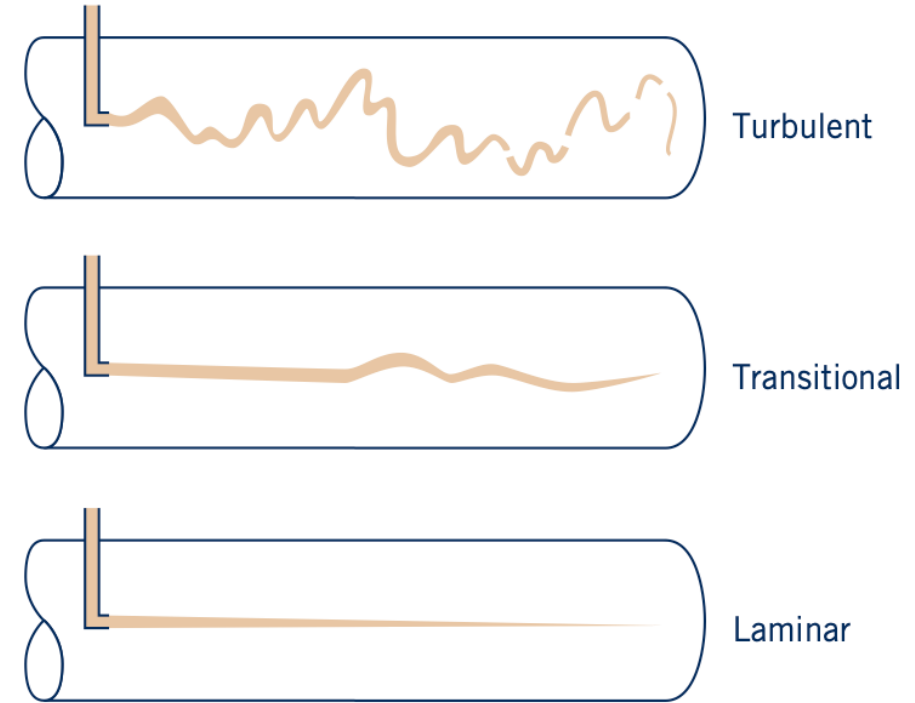
Pipe Flow



Pipe Flow – Reynolds Experiment

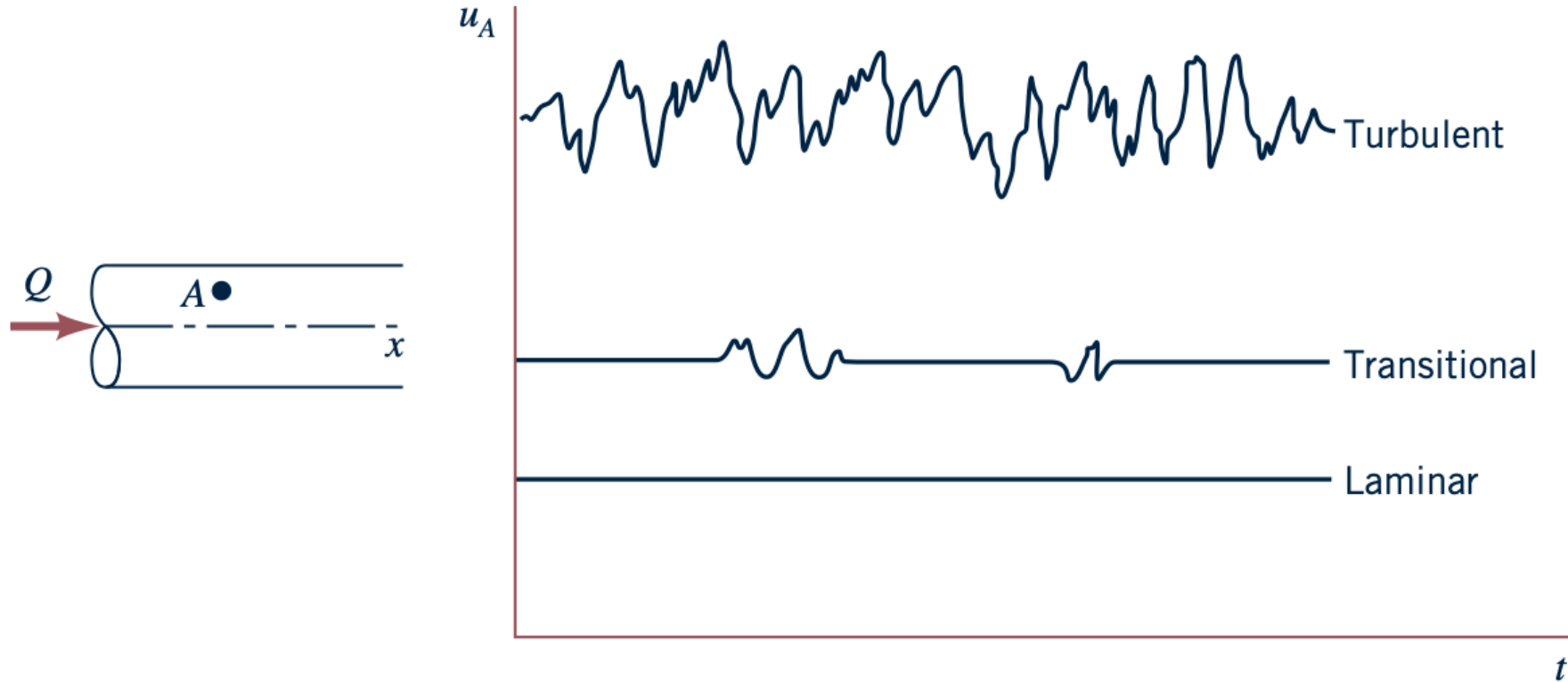


(a)

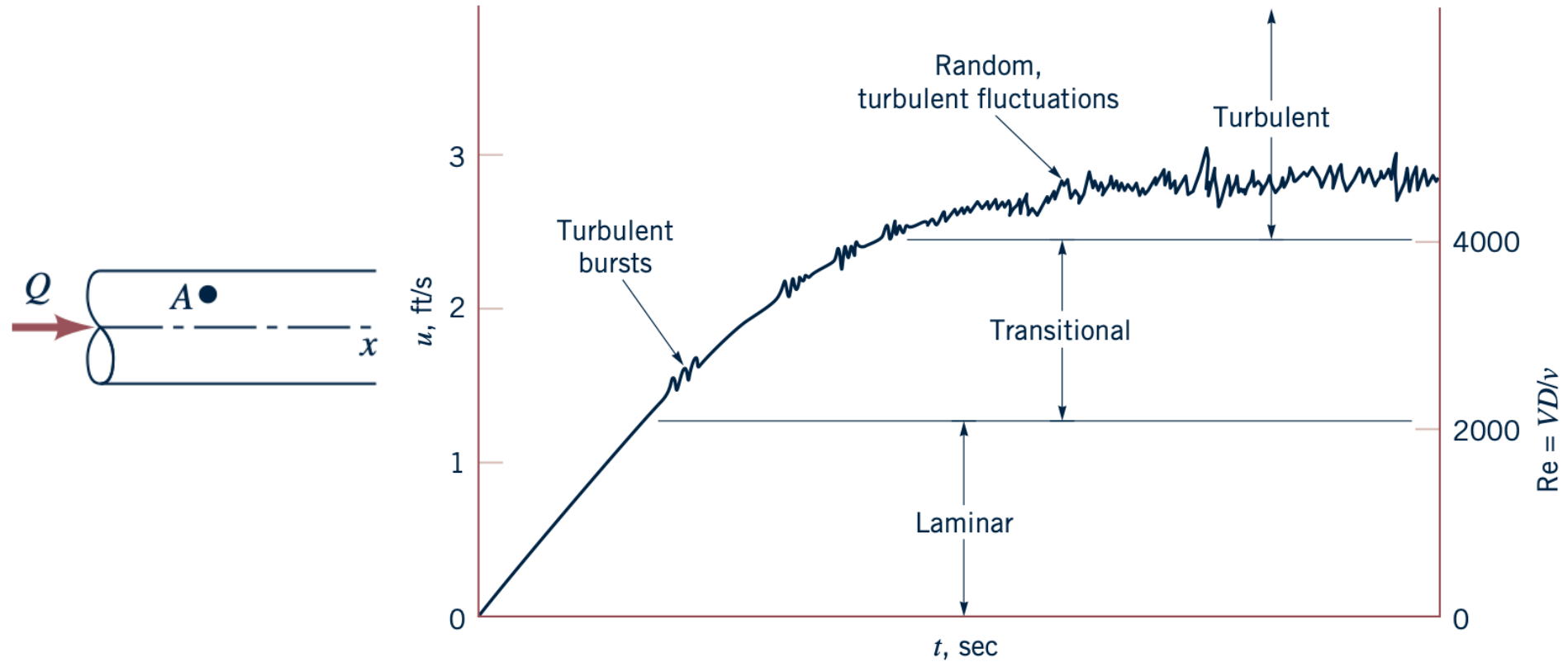


(b)

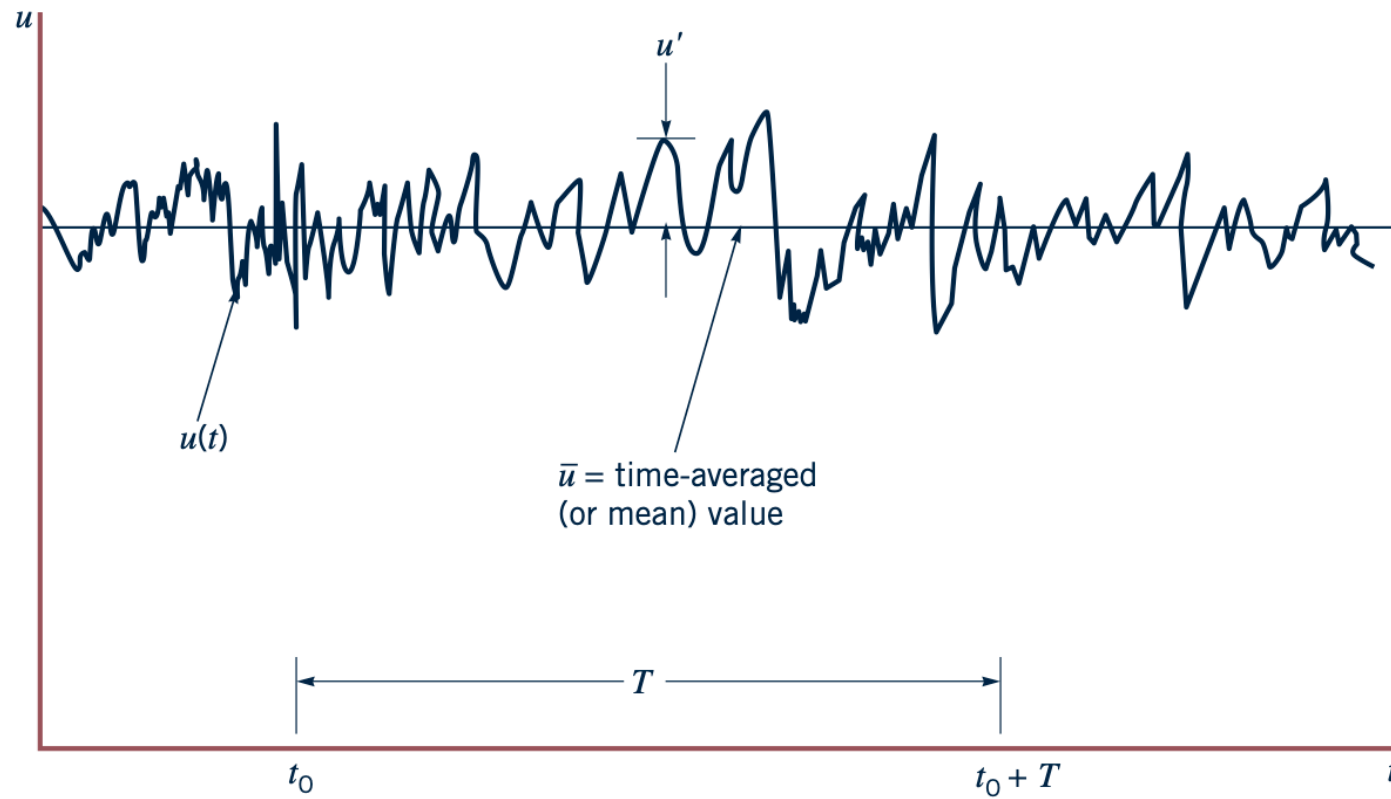
Pipe Flow – Reynolds Experiment



Pipe Flow – Reynolds Experiment

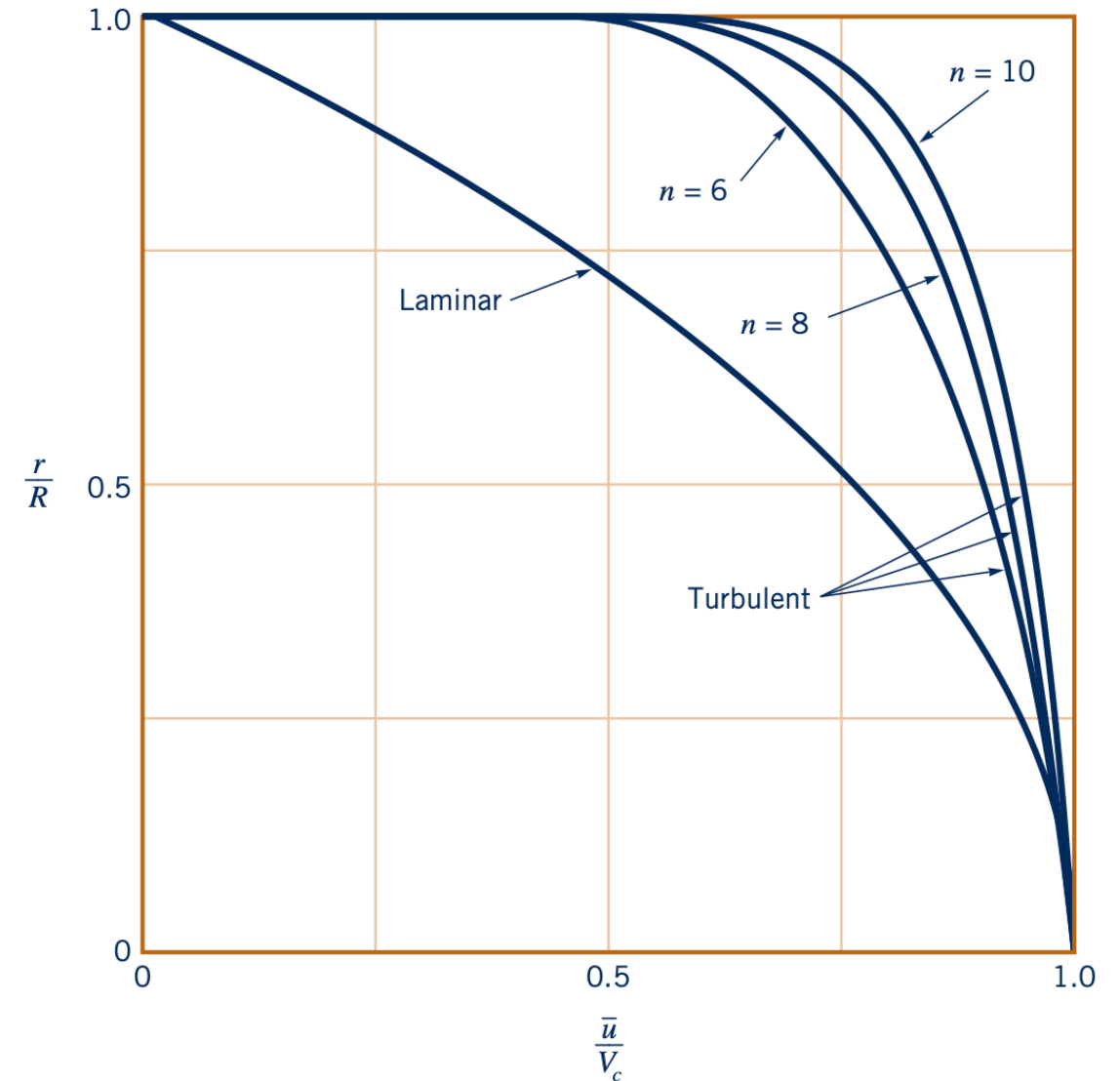
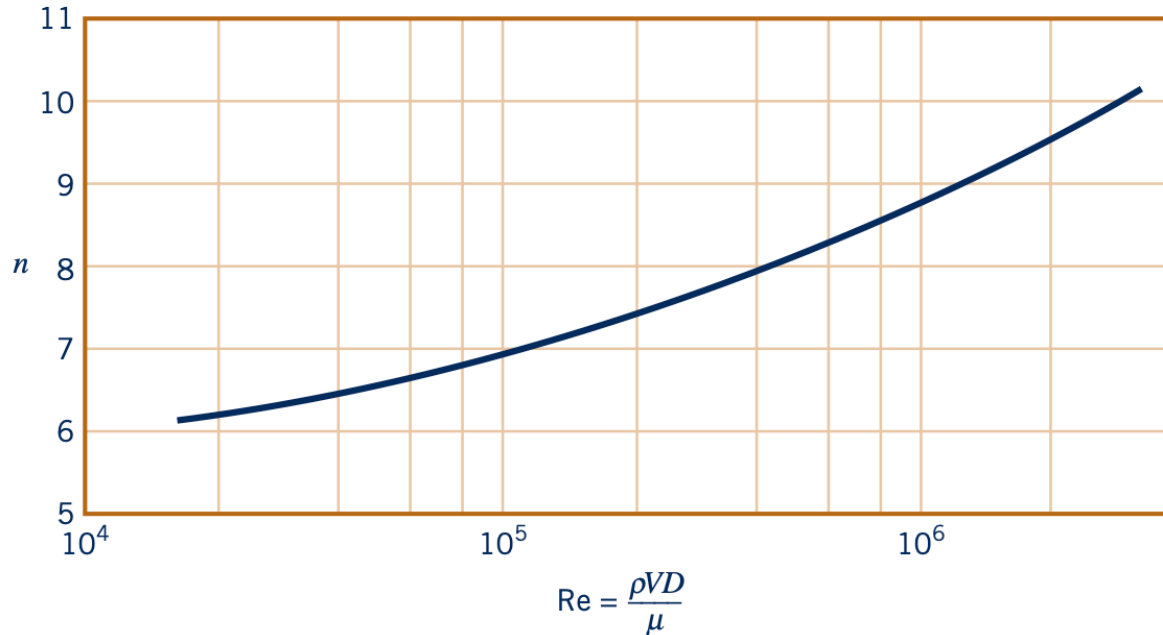


Turbulent Fluctuations

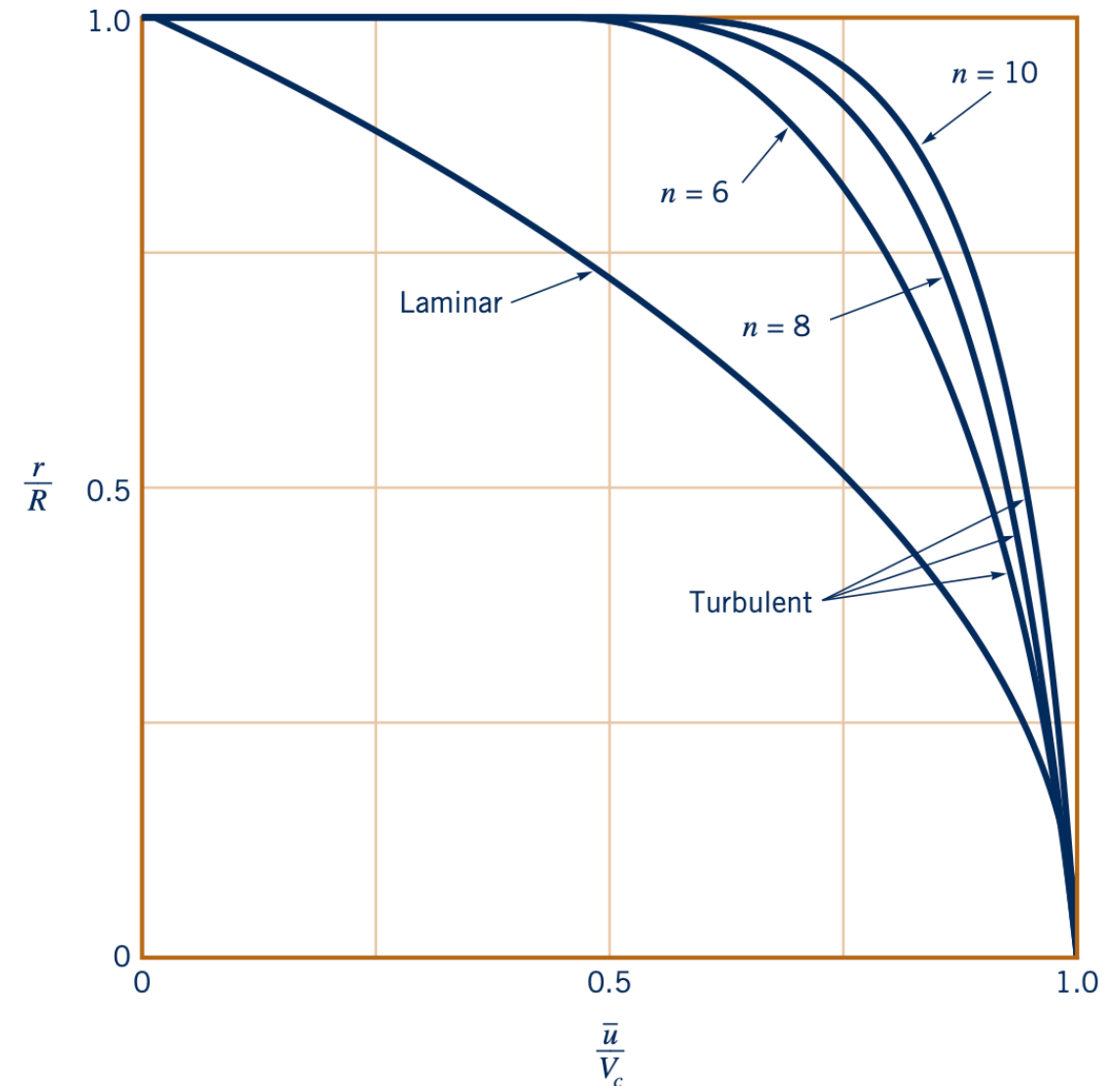
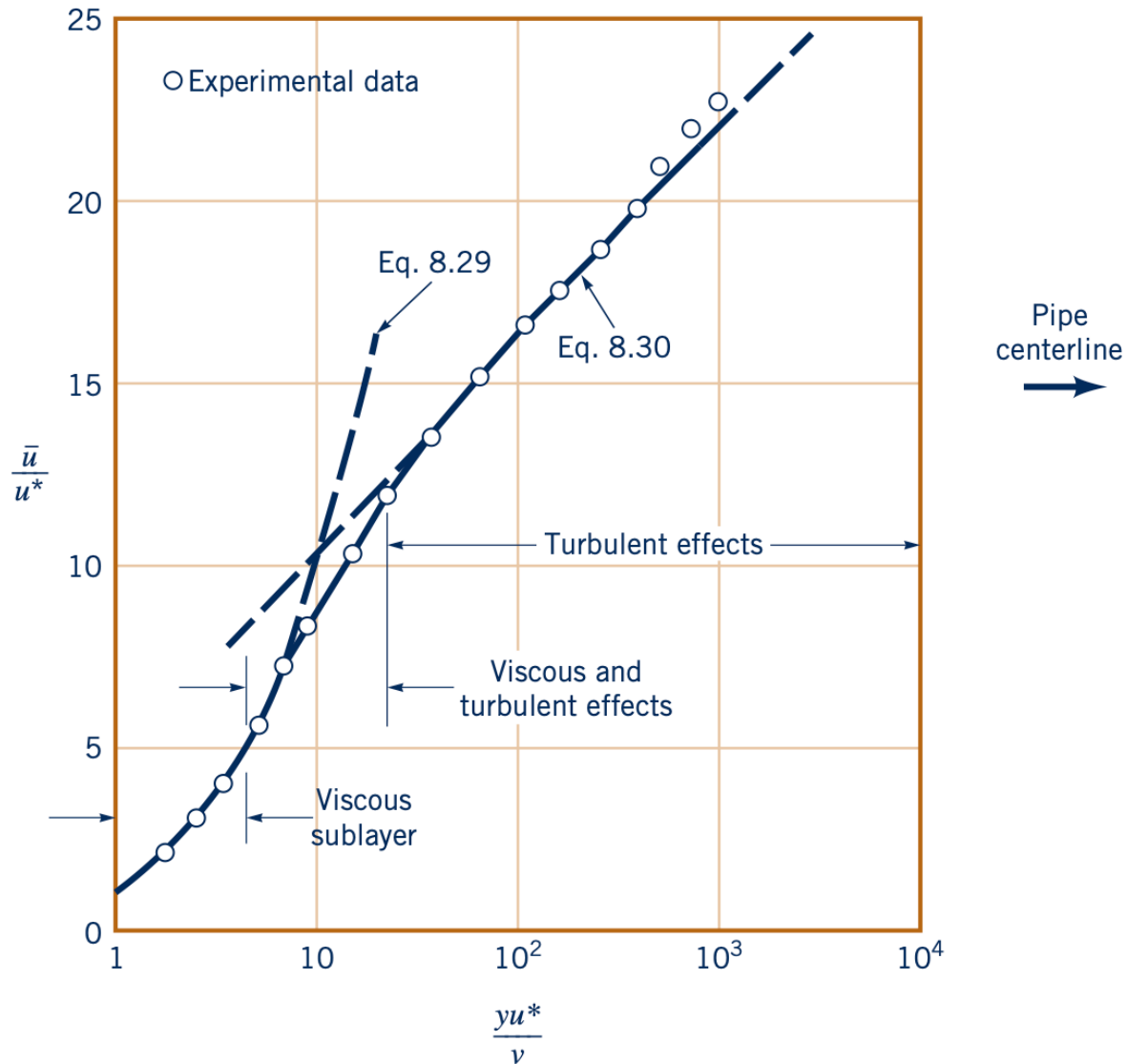


Pipe Flow – Mean Flow Profile

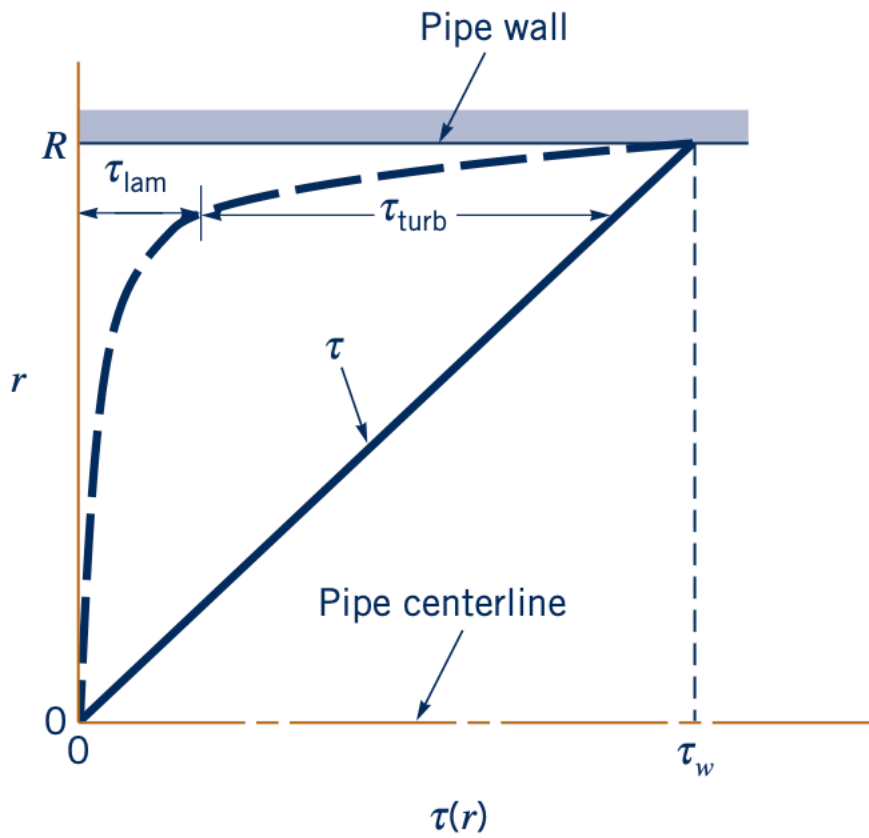
$$\frac{\bar{u}}{V_c} = \left(1 - \frac{r}{R}\right)^{1/n}$$



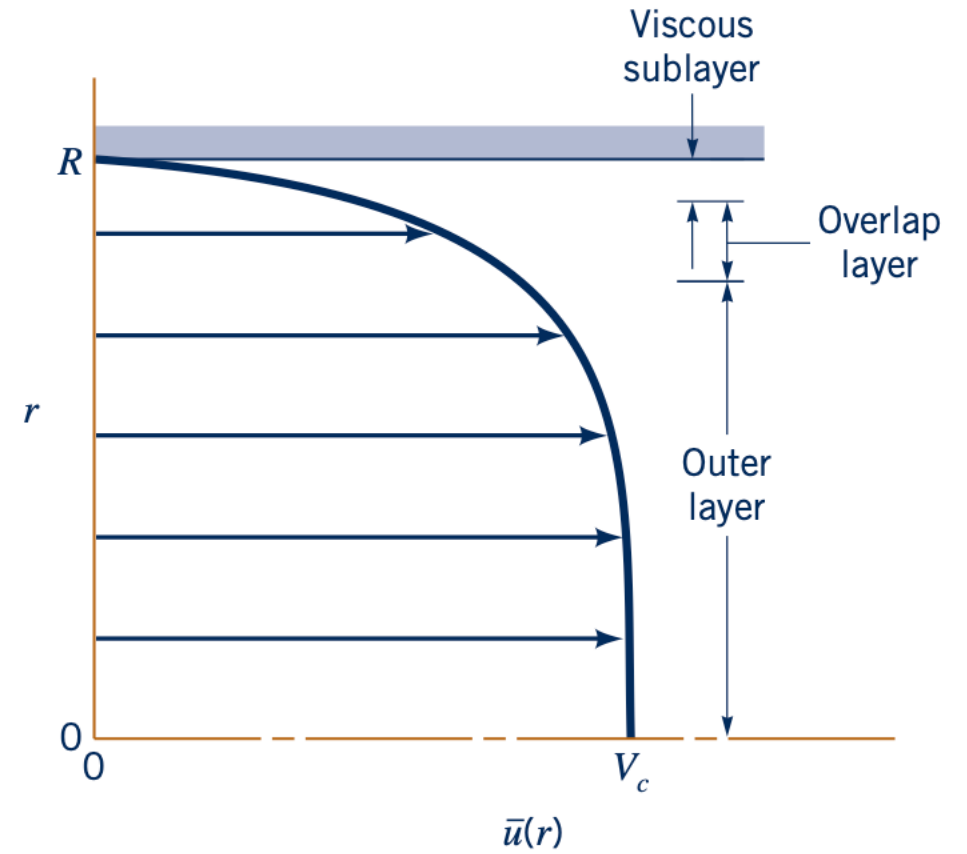
Pipe Flow – Mean Flow Profile



Pipe Flow – Mean Flow

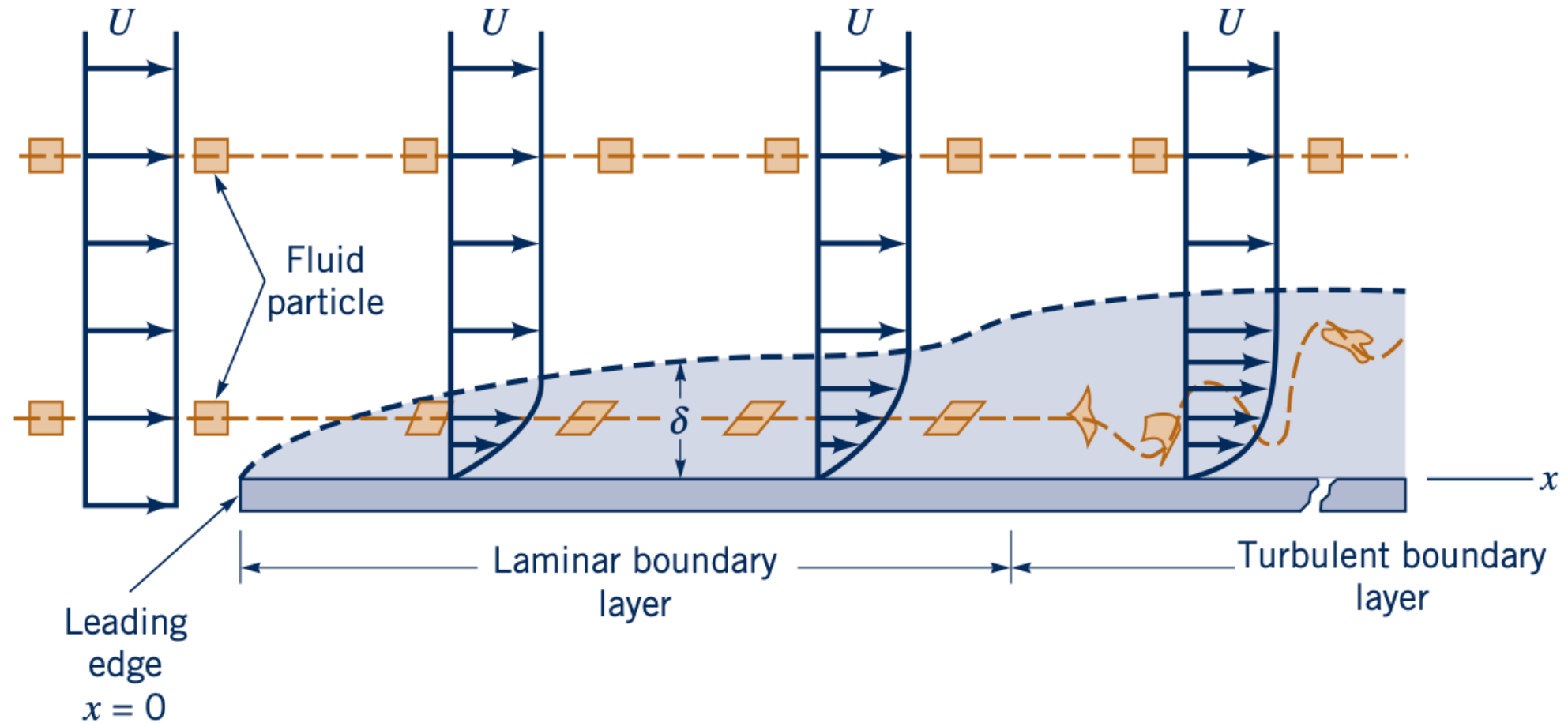


(a)

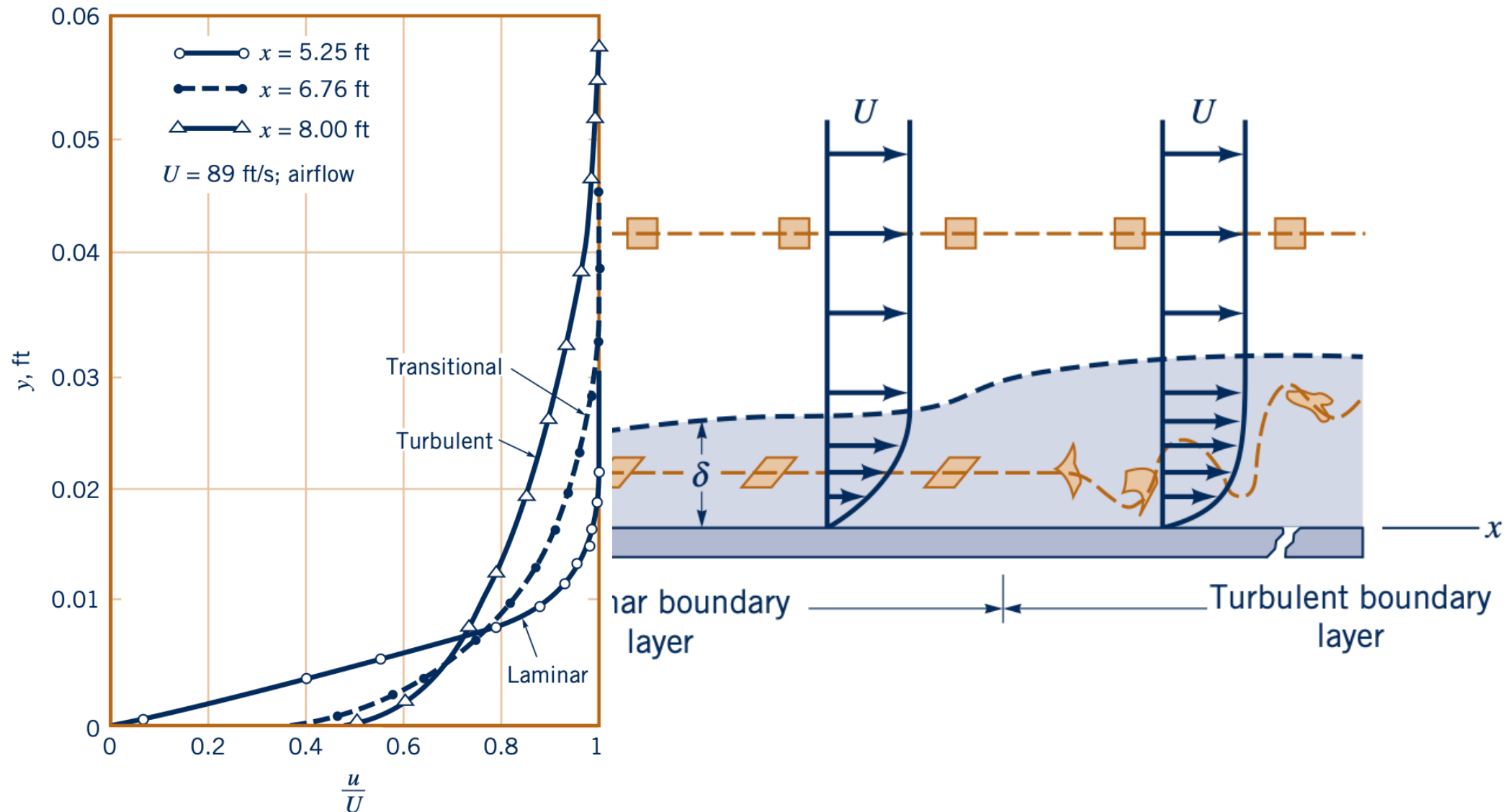


(b)

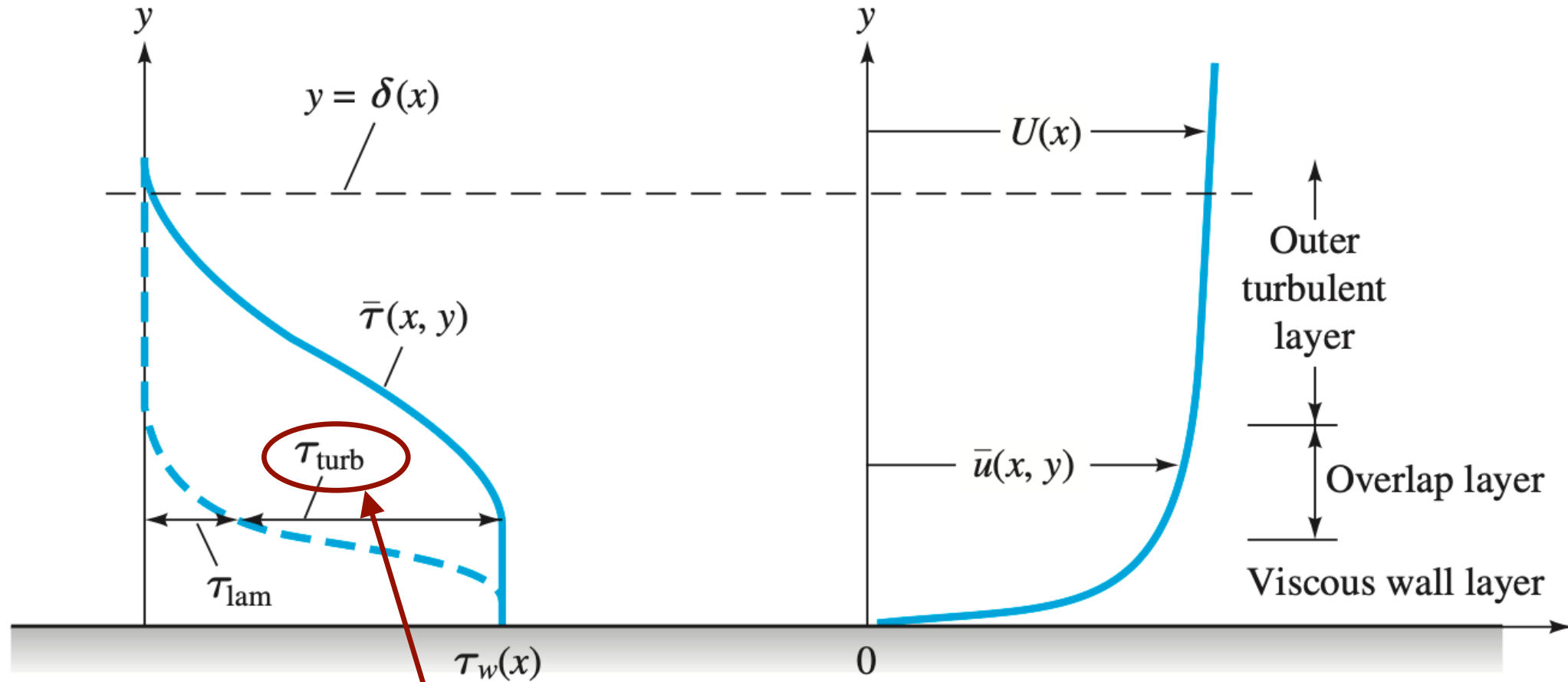
Flat-plate Boundary Layer



Flat-plate Boundary Layer

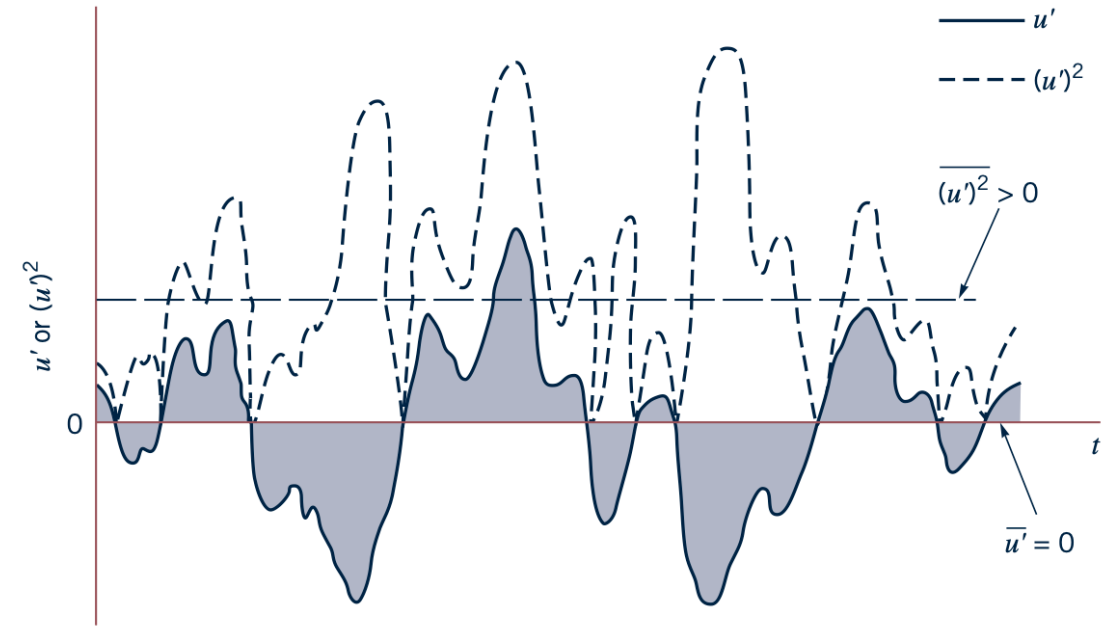
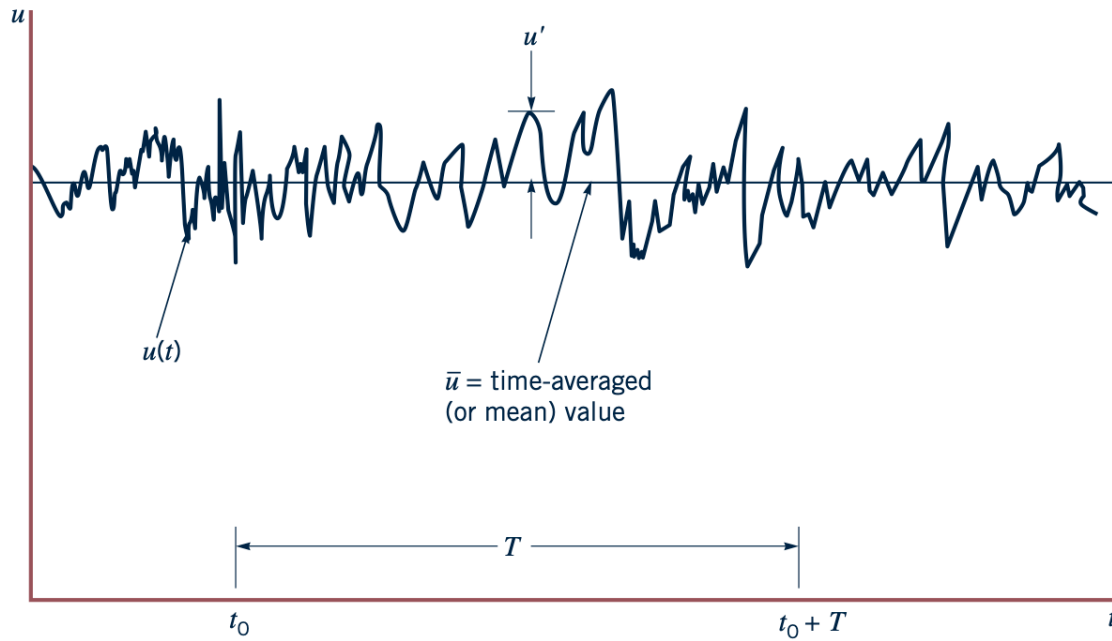


Flat Plate Flow – Mean Flow



How to model?

Turbulent Fluctuations



Turbulence Intensity $\mathcal{I} = \frac{\sqrt{\overline{(u')^2}}}{\bar{u}}$

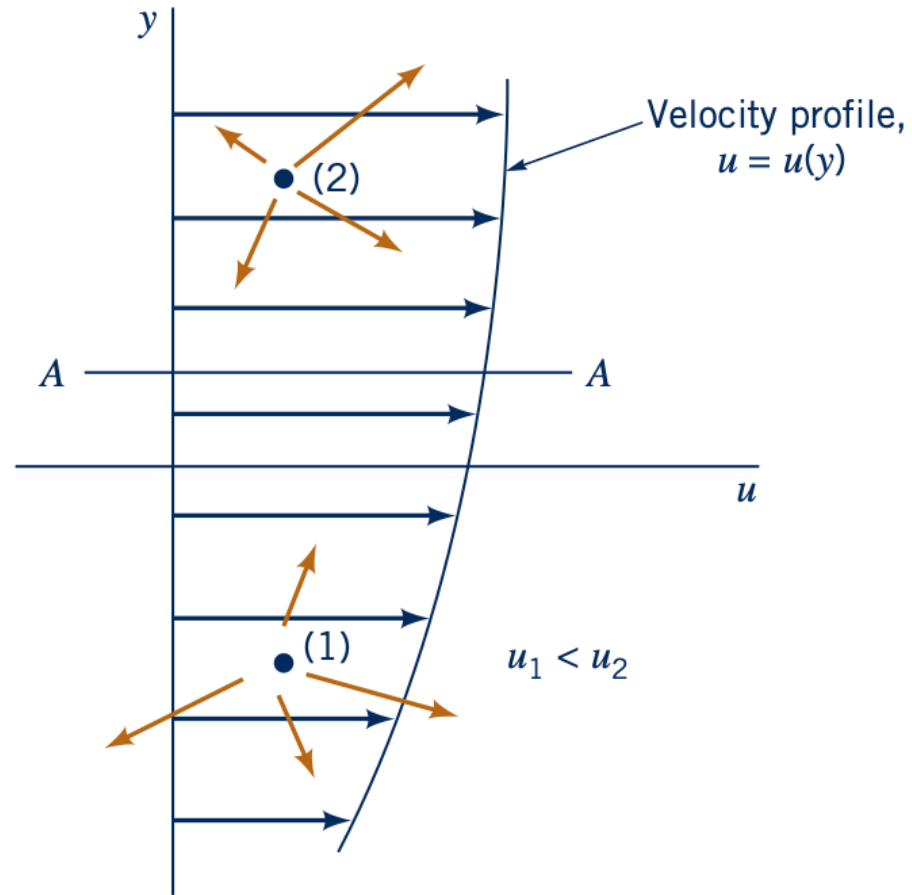
Reynolds Averaged Navier-Stokes (RANS)

$$\frac{\partial \bar{u}_i}{\partial t} + \frac{\partial}{\partial x_j} (\bar{u}_i \bar{u}_j) = -\frac{1}{\rho_o} \frac{\partial \bar{p}}{\partial x_i} - \frac{\partial}{\partial x_j} \left(\overline{u'_i u'_j} \right) + \nu \frac{\partial^2 \bar{u}_i}{\partial x_j \partial x_j}$$

Reynolds
Stress tensor

$$\tau_{t_{ij}} = -\overline{u'_i u'_j} = \nu_t \frac{\partial \bar{u}_i}{\partial x_j}$$

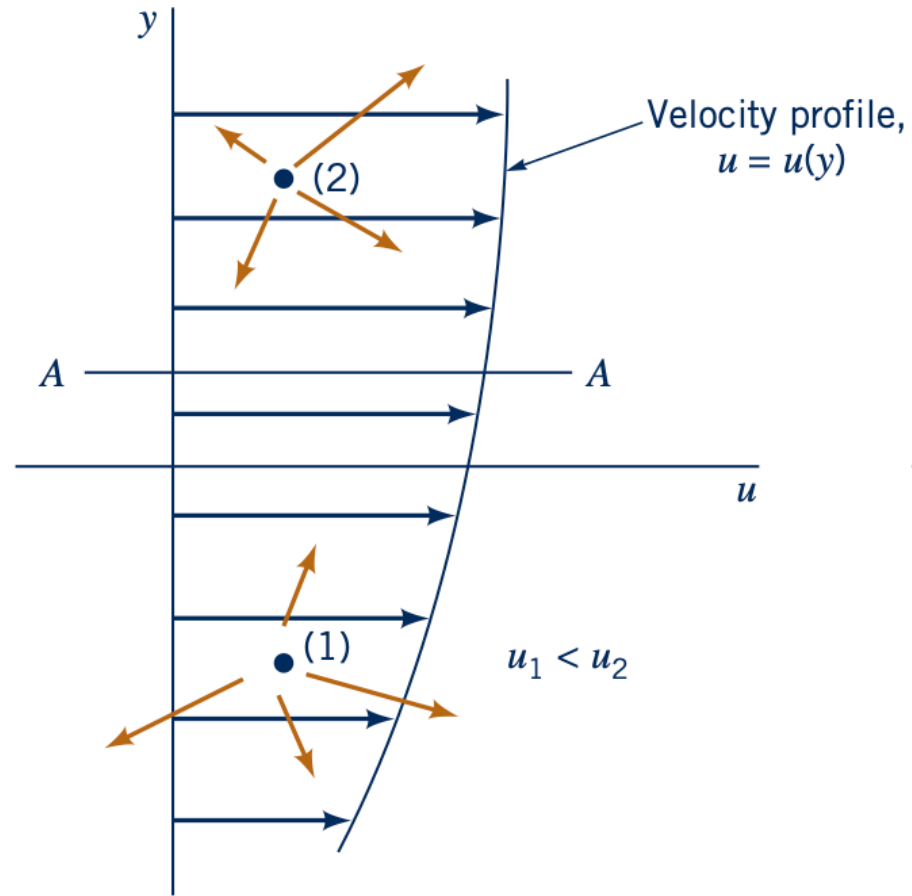
Concept of Eddy viscosity (ν_t)



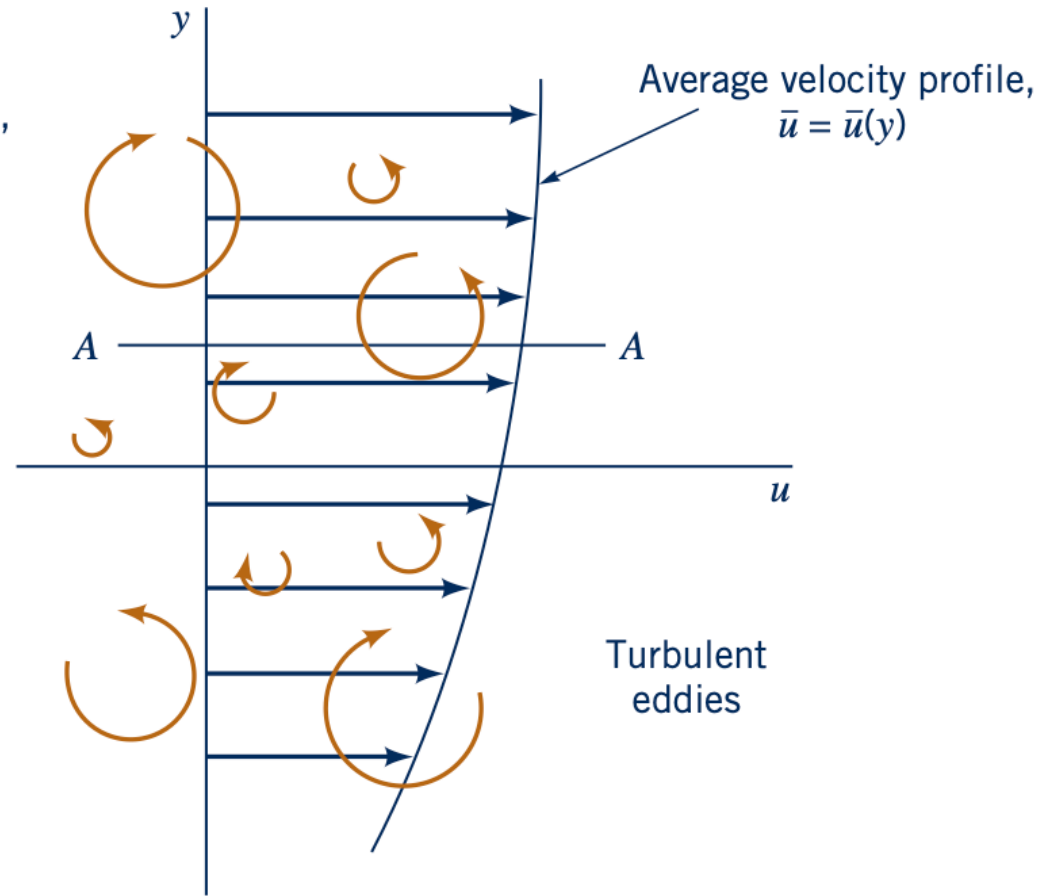
$$\nu \sim \nu_{th} \lambda_{mfp}$$

Classical Collisional transport

Concept of Eddy viscosity (ν_t)



Classical Collisional transport



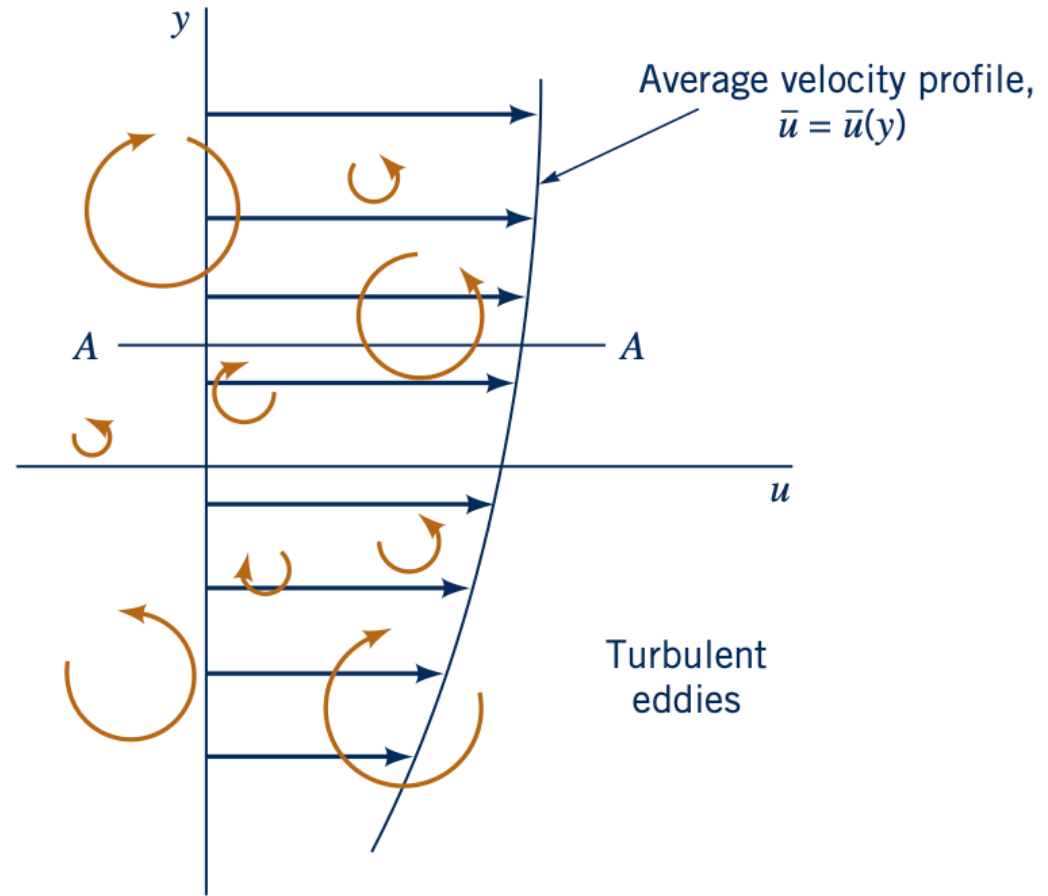
Turbulent transport

Concept of Eddy viscosity (ν_t)

l_m is “mixing length”

$$\nu_t \sim \lambda_t \sim \frac{l_m^2}{\tau_m} = l_m^2 \frac{\partial \bar{u}}{\partial y}$$

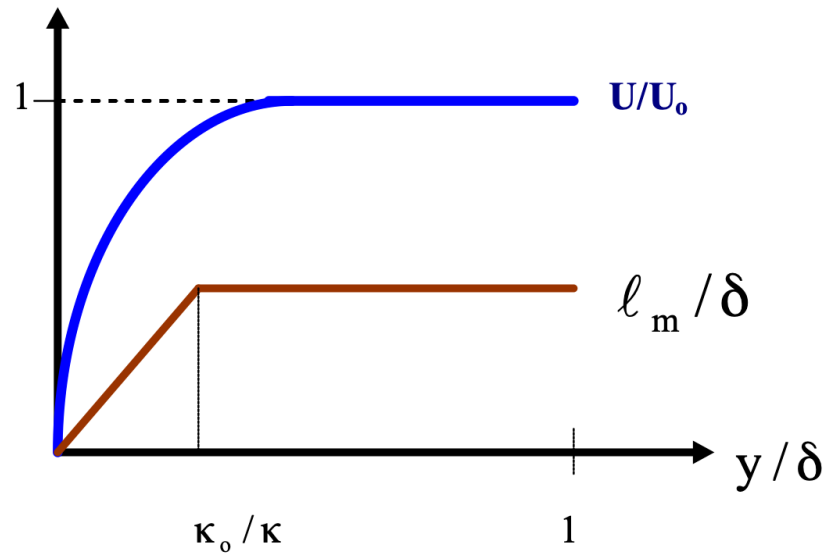
Prandtl



Turbulent transport

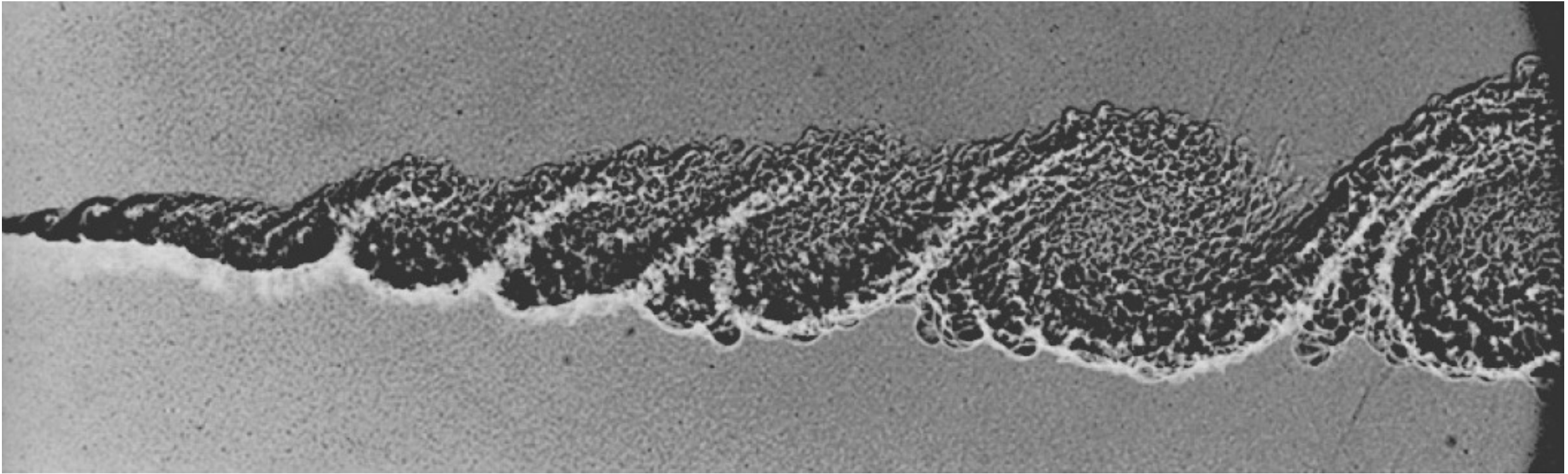
Concept of Eddy viscosity (ν_t)

$$\nu_t \sim v_t \lambda_t \sim \frac{l_m^2}{\tau_m} = l_m^2 \frac{\partial \bar{u}}{\partial y}$$



	l_m
plane jet	$0.09 \, l$
circular jet	$0.075 \, l$
mixing layer	$0.07 \, l$
wall boundary layer	$\kappa y, \quad \frac{y}{\delta} \leq \frac{\kappa_0}{\kappa}$ $\kappa \delta, \quad \frac{y}{\delta} \geq \frac{\kappa_0}{\kappa}$ $\kappa = 0.41, \kappa_0 = 0.09$

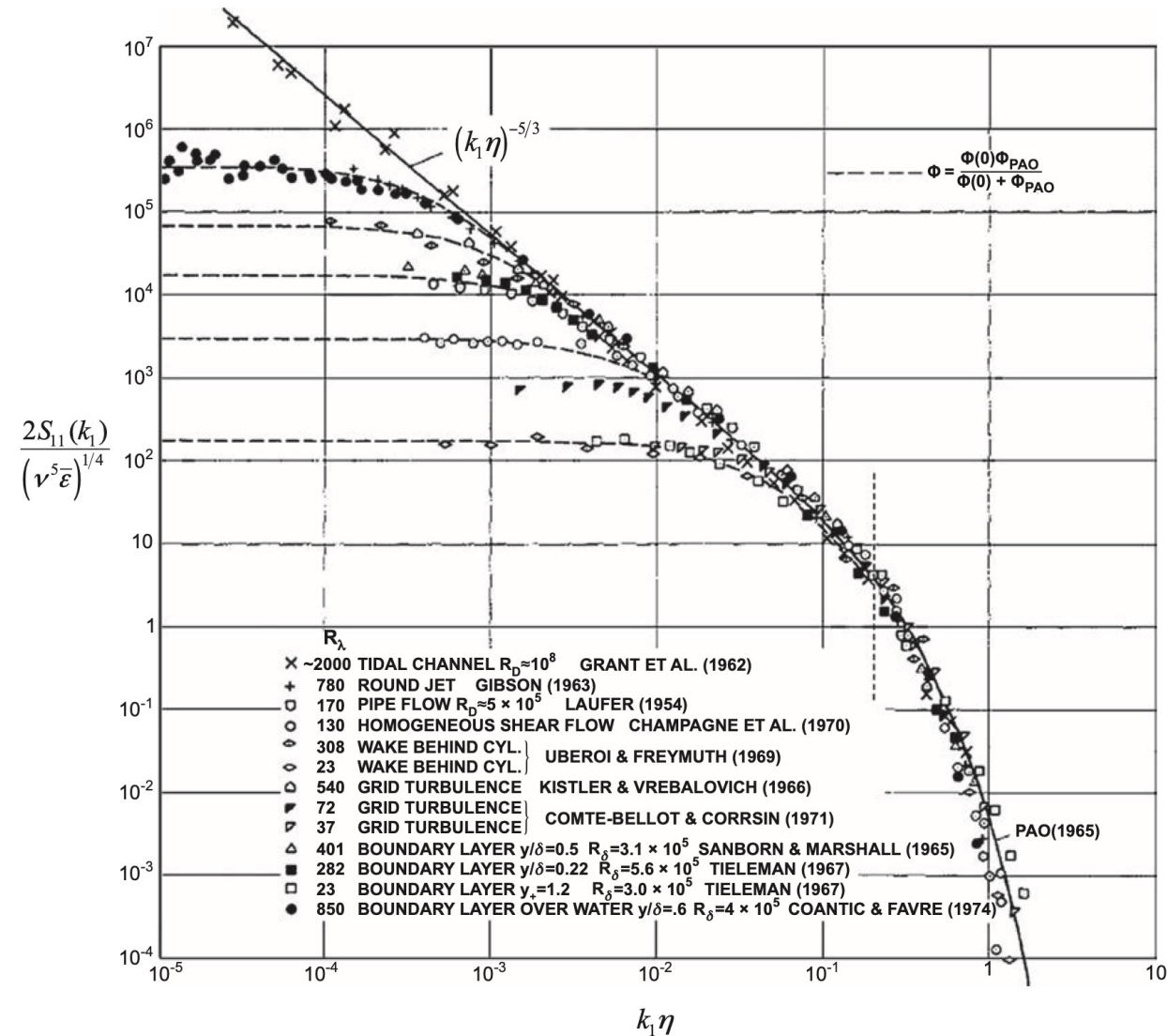
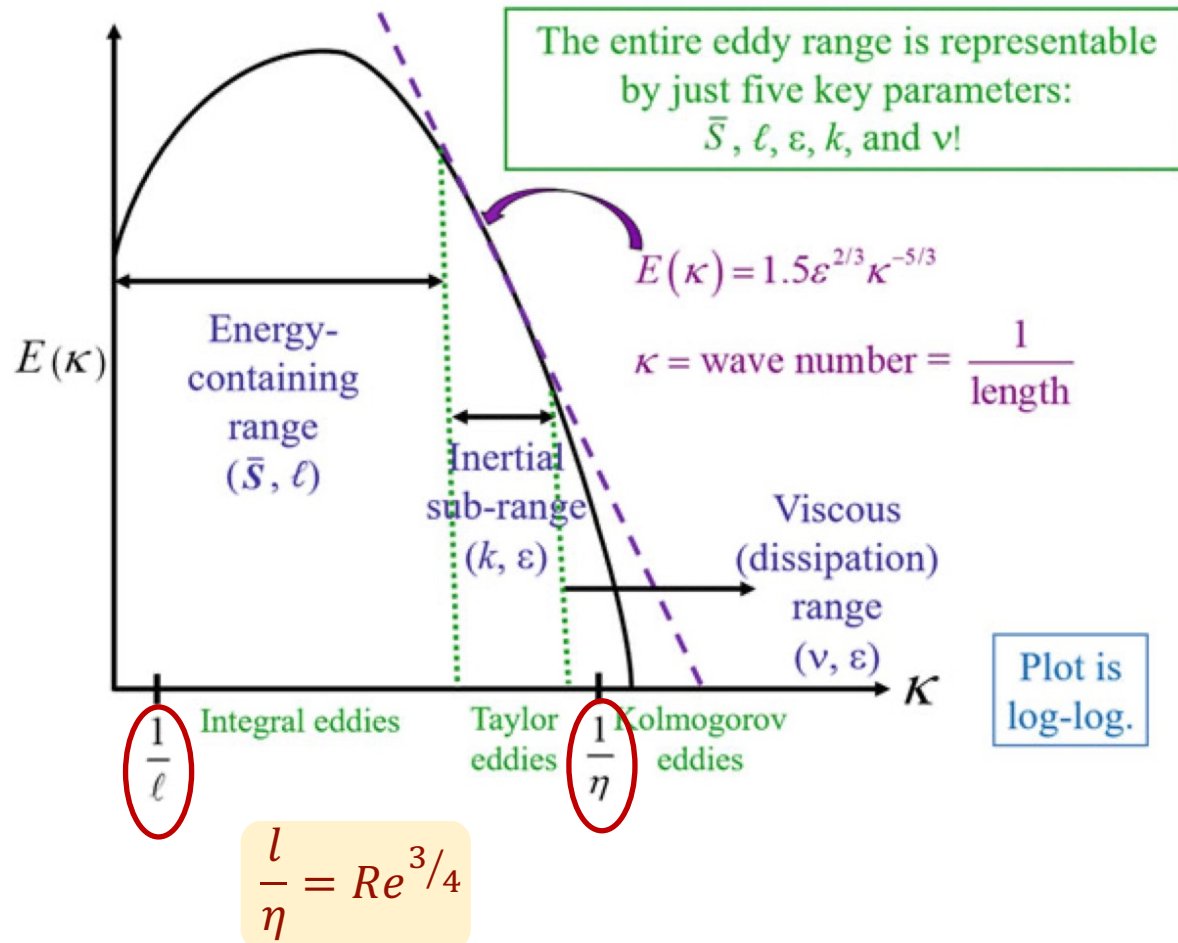
Turbulent Mixing Layer



Mixing Layer between Helium (upper) and Nitrogen (lower) streams

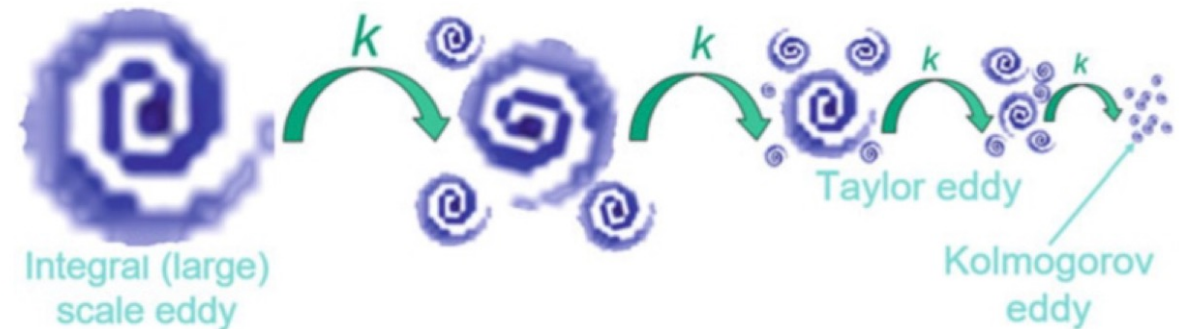
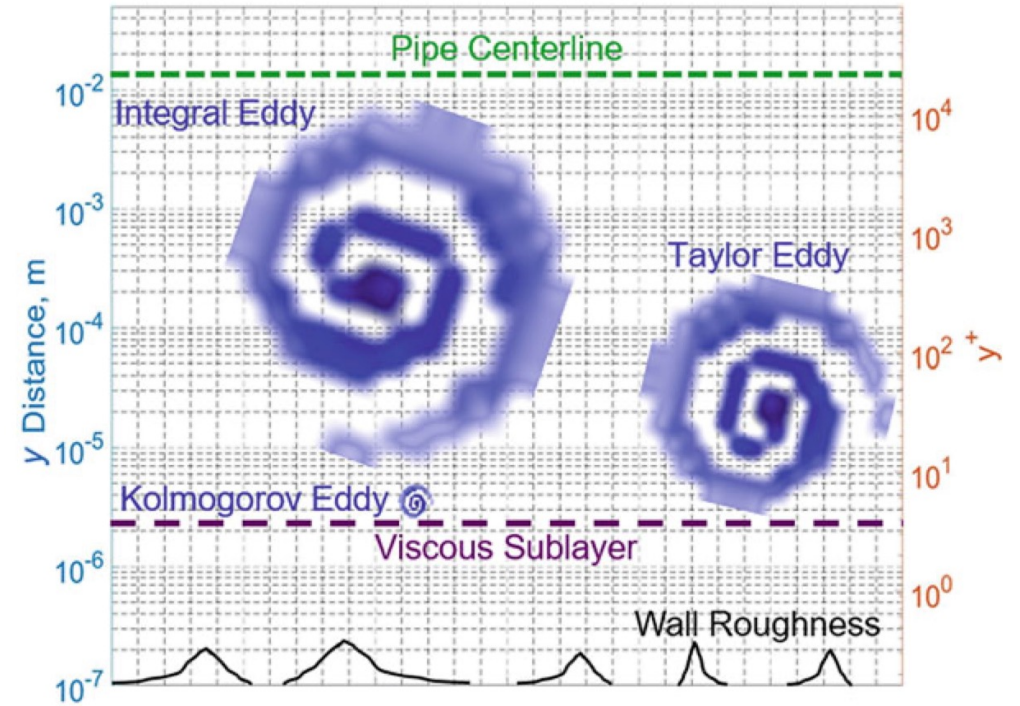
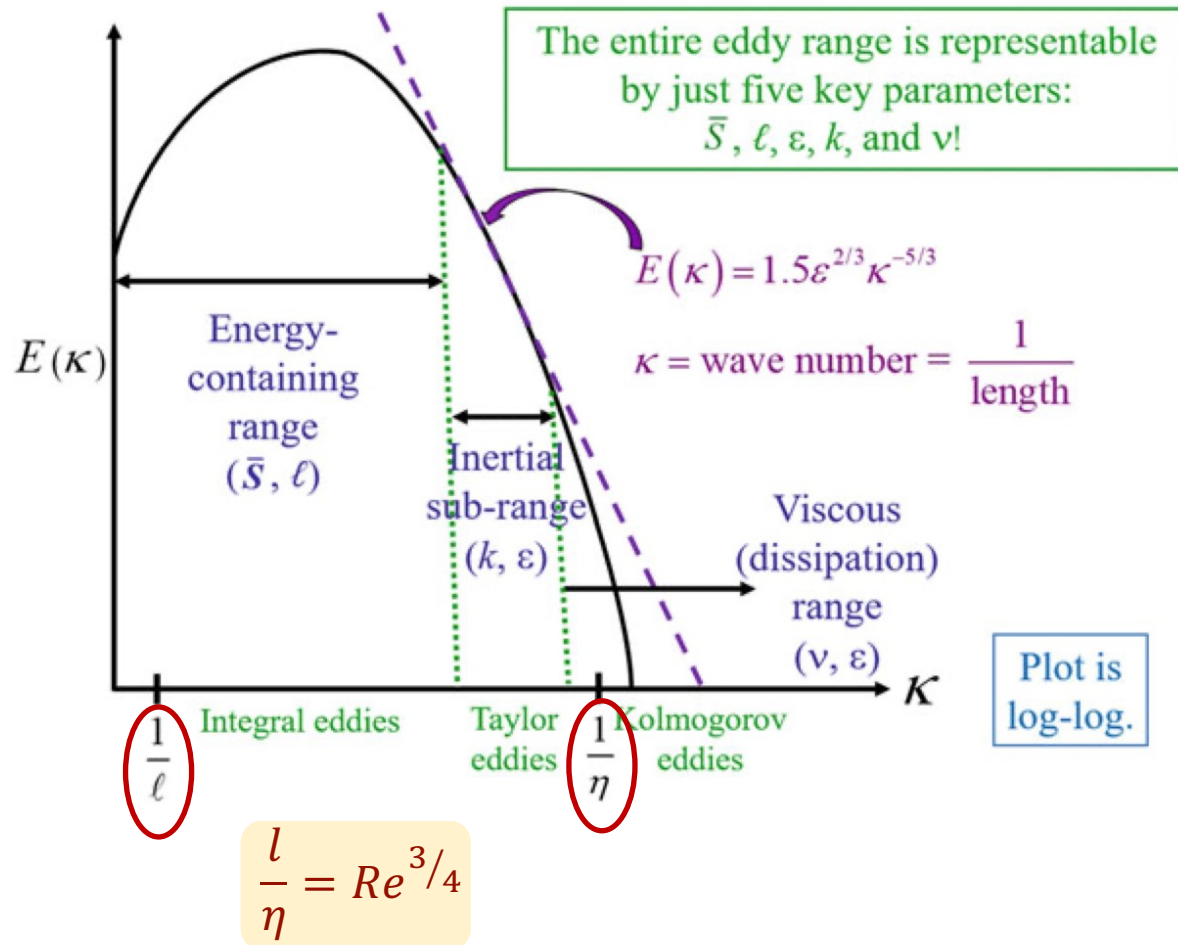
Turbulence

Kolmogorov's Energy Spectrum



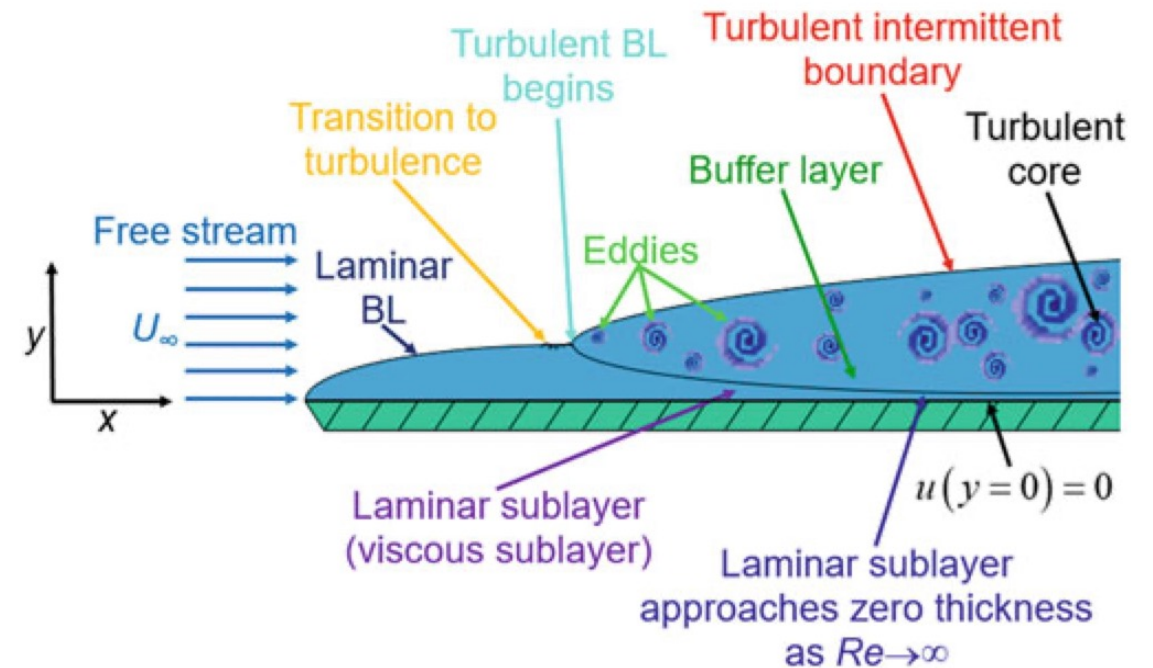
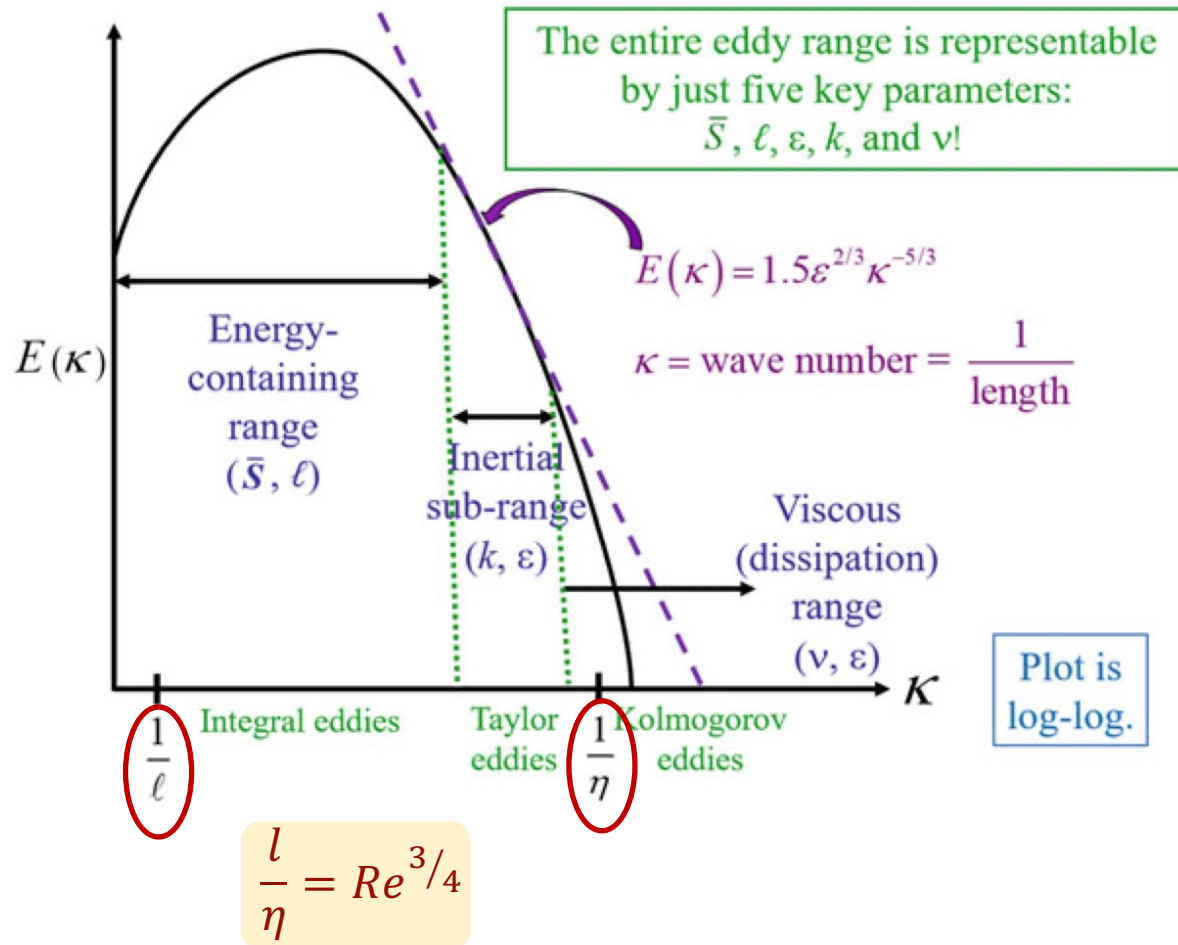
Turbulence

Kolmogorov's Energy Spectrum



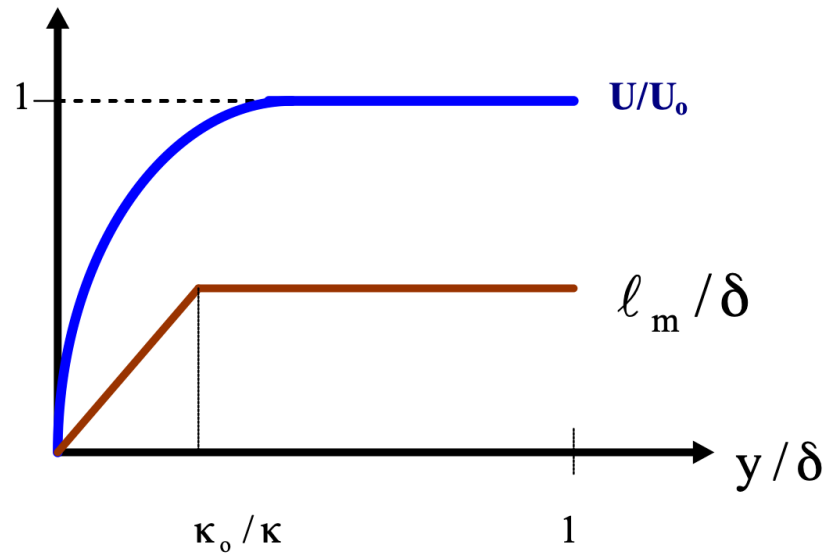
Turbulence

Kolmogorov's Energy Spectrum



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References

- Images from the books of White, Heister et al., Munson et al., Kundu et al., and Rodriguez