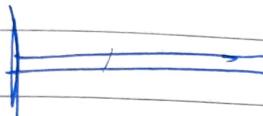


* Segue into 1- and 2-DOF systems

- designer's tool
- abstraction - reality \rightarrow model approximation of cont sys.



make 1 DOF

$$w(u, t) = f(u)g(t)$$

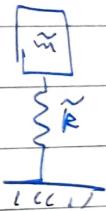
$$T = \int_0^L \frac{1}{2} m \left(\frac{dw}{dt} \right)^2 = \int_0^L \frac{1}{2} m (g(t))^2 (f'(u))^2 du$$

$$T = \frac{1}{2} \int_0^L \tilde{m}(g(t))^2 - \text{equiv mass}$$

equivalent mass

$$V = \int_0^L \frac{1}{2} EI \left(\frac{d^2 w}{du^2} \right)^2 du$$

now,



$$= \int_0^L \frac{1}{2} EI (g^2(t)) (f''(u))^2 du$$

$$= \frac{1}{2} EI g^2(t) \int_0^L (f''(u))^2 du$$

$$V = \frac{1}{2} \tilde{k} g^2(t) \rightarrow \text{equiv diss}$$

spring const

quantitative + qualitative

deck
 reality \rightarrow model \rightarrow approx \rightarrow soln. \leftarrow engg
 deflection (D.F) soln method?
 we dealing time invariant system

for 2 D.o.F
 $y(u, t) = a_1 f_1(u) g_1(t) + a_2 f_2(u) g_2(t)$

sub in T & V: fastig

$$T = \frac{1}{2} \int_0^L m \dot{y}^2 dx$$

$$= \frac{1}{2} m \left[A_1 \dot{f}_1^2(t) + A_2 \dot{f}_2^2(t) \right] + A_3 \dot{f}_1 \dot{f}_2$$

$$= \frac{1}{2} [\dot{f}_1 \dot{f}_2] \begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \begin{bmatrix} \dot{f}_1 \\ \dot{f}_2 \end{bmatrix} \approx \frac{1}{2} M \dot{v}^2$$

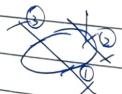
general form of 2 D.o.F system

general exp. $E = \frac{1}{2} \begin{bmatrix} & & & \end{bmatrix}_{nn} \begin{bmatrix} & & & \end{bmatrix}_{nn} \begin{bmatrix} & & & \end{bmatrix}_{nn}$

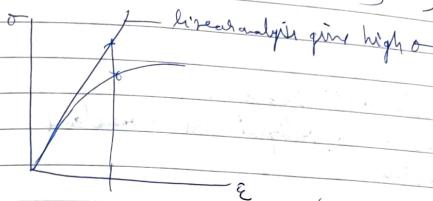
Domain $[f(u)] \in [0, L]$

global shape function

working of parts is Ritz method



$$V = \frac{1}{2} \begin{bmatrix} s & s \\ 0 & 1 \end{bmatrix}_{22} \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} \begin{bmatrix} \dot{f}_1 \\ \dot{f}_2 \end{bmatrix}$$



29/9/22

1 DoF system



- formulate the 1DoF problem
- solve said problem, interpret and understand
- interpret / validate

general form of 1DoF

No damping

- $m\ddot{x} + kx = F$
- where m, F are δ^m of t
- m, k may be δ^m of t
- m, k const unless specified otherwise

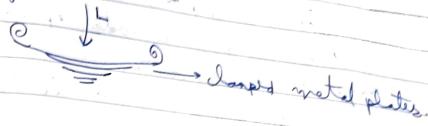
Damping (Estimate)

- 1) viscous : solid - fluid friction
- 2) Friction : solid - solid external (contact)
- 3) Material : solid - solid (internal to material)

more damping affects soln



Lab springs



Most of the times we use viscous models

deal w/ "small" levels / amounts of damping
much nicer mathematically

8

Equivalent viscous damping

$$m\ddot{x} + c\dot{x} + kx = F \quad m, c, k = \text{const}$$

$$n = \text{homogeneous} + \text{particular}$$

$$F=0 \quad F=F$$

$$m\ddot{x} + c\dot{x} + kx = 0$$

$$\text{initial cond} \quad x(0)=x_0, \quad \dot{x}(0)=\dot{x}_0=v_0$$

Laplace \checkmark state space \checkmark general form of soln
 $x(t) = A_0 e^{st}$

$$(m\ddot{x} + c\dot{x} + kx) A_0 e^{st} = 0$$

$$\text{non trivial soln} \Rightarrow m\ddot{x} + c\dot{x} + kx = 0$$

$$s_2 = \frac{-C \sqrt{C^2 - 4mk}}{2m}$$

$$C=0 \Rightarrow \text{underdamped}$$

$c^2 > 4 \text{ km}$

$c^2 = 4 \text{ km}$

$c^2 < 4 \text{ km}$

real purely

degenerate roots \rightarrow purely real

complex roots.

{most oscillatory part converges}

more interested here.

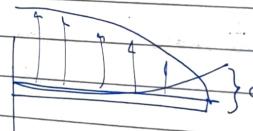
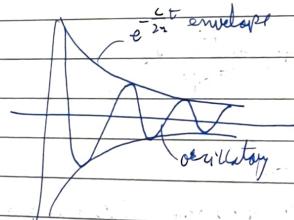
$$s_{1,2} = -c \pm \sqrt{c^2 - 4 \text{ km}} / 2m$$

$$= -\frac{c}{2m} \pm \sqrt{\left(\frac{c}{2m}\right)^2 - \left(\frac{k}{m}\right)}$$

natural ω for undamped sys.

$$\begin{aligned} s(t) &= A_1 e^{s_1 t} + A_2 e^{s_2 t} \\ &= A_1 e^{-\frac{c}{2m}t} [A_1 e^{-\frac{i\omega}{2m}t} + A_2 e^{-\frac{i\omega}{2m}t}] \end{aligned}$$

exponentially decaying oscillatory nature (purely)



$e^{(i\omega)t}$ \rightarrow frequency

freq of the damped system is

$$\omega_d = \sqrt{\frac{k}{m} - \left(\frac{c}{2m}\right)^2}$$

$$\omega_d = \sqrt{1 - \left(\frac{c}{2m}\right)^2} \sqrt{\frac{k}{m}}$$

$$\omega_d = \omega_n \left(\frac{c}{2m} \right)$$

$$\omega_d < \omega_n$$

$$\frac{c^2}{4m^2} \frac{m}{k} = \frac{c^2}{4 \text{ km}}$$

$$\text{damping case } \xi^2 = \frac{c^2}{9 \text{ km}}$$

$$\xi^2 = \frac{c^2}{4 \text{ km}} = \frac{c}{2\sqrt{km}} \text{ damping ratio}$$

$$\therefore \omega_d = \omega_n \sqrt{1 - \xi^2}$$

value of

need to find good sub for damping.

$$\frac{c^2}{4m^2} - \frac{k}{m} = 0$$

$$c^2 = 4 \text{ km}$$

$c^2 = 4 \text{ km} \rightarrow$ non-oscillatory

$c^2 = 4 \text{ km} \approx$

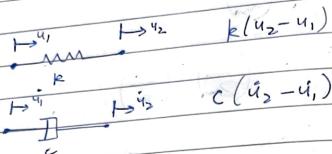
derivation of oscillatory pt.

critically damping.

3/10/22

→ viscous damping

$$m\ddot{u} + c\dot{u} + ku = 0$$

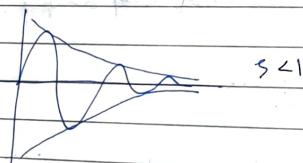


$$y(t) = e^{-\zeta \omega_n t} (A_1 e^{i\sqrt{\omega_n^2 - \zeta^2} t} + A_2 e^{-i\sqrt{\omega_n^2 - \zeta^2} t})$$

$$\zeta^2 = \frac{c^2}{m}$$

$$\omega_n = \sqrt{\frac{k}{m}}, \quad \omega_d = \omega_n \sqrt{1 - \zeta^2}$$

$$= e^{-\zeta \omega_n t} [A_1 e^{i\omega_d t} + A_2 e^{-i\omega_d t}]$$

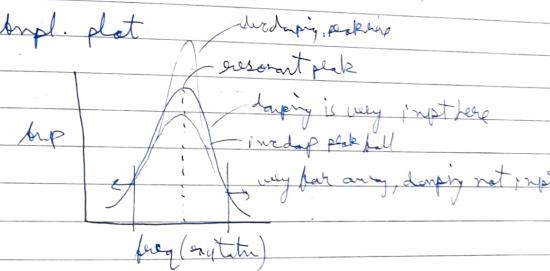


$S = 1$ converges fastest
 $S > 1$ gradual exponential decay

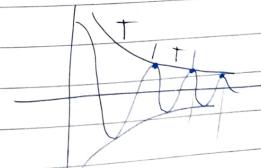
- in practice, damping is intro. as part of crit.
- in practice, easiest to obtain equivalent viscous damping estimate damping as a purely viscous damper.

- + free vibration - easy but more approx.
- + forced vibration - not so easy requires specific equipment & knowledge of the system but more accurate
- + brief pause, bandwidth, resonance peak, ...

→ freq. resp. plot



$$y(t) = e^{-\zeta \omega_n t} [A_1 \cos \omega_d t + A_2 \sin \omega_d t]$$



→ how to obtain estimate of ζ ?

rate of change at peaks $\rightarrow S$ info get

$$e^{-\zeta \omega_n t} \cos(\omega_n t + \phi)$$

for given initial co., A & ω_n

$$\begin{aligned} y_1 \\ y_2 = \frac{y_1}{e^{-\zeta \omega_n t}} \end{aligned}$$

for only peaks $\cos = 1$

$$\begin{aligned} y_1 \\ y_n = \frac{e^{-\zeta \omega_n t}}{e^{-\zeta \omega_n n T}} = e^{S \omega_n (n-1) T} \end{aligned}$$

take log get estimate

T = period of oscillation of damped system

2nd freq of damped system is ω_d

$$T = \frac{2\pi}{\omega_d} = \frac{2\pi}{\omega_n \sqrt{1-\zeta^2}}$$

$$S \omega_n (n-1) T = S \omega_n (n-1) \frac{2\pi}{\omega_n \sqrt{1-\zeta^2}} = \frac{2\pi(n-1)\zeta}{\sqrt{1-\zeta^2}}$$

take log value of response

$$\ln\left(\frac{y_j}{y_{j+n}}\right) = \frac{2\pi(n-1)\zeta}{\sqrt{1-\zeta^2}}$$

$\Rightarrow j \Rightarrow$ peak of cycle j

for small S ,

$$\frac{1}{\sqrt{1-S^2}} = (1-S^2)^{-1/2} = 1 + \frac{S^2}{2}$$

or $S \ll 1 \quad 1-S^2 \approx 1$

$$S = \frac{\delta}{2\pi(n-1)} \quad \text{where } \delta = \ln\left(\frac{y_j}{y_{j+n}}\right)$$

this approach valid, logarithmic damping

→ assignment

IDOF system

M, K give to us
estimate S acc.

Harder on.

double using simple constructs
estimate S , prove it is correct.

2nd there is a deviation from expectation, discuss

6/10/22

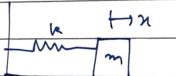
- friction damping
- Fouad subspace (either on 17th or 31st)
- 17th quiz #2
- class end on 8th Nov

- prev lecture
- viscous damping
-  dashpots

- friction
-  wedge scratch the surface

- material damping
- no symbol, but  seen in books

→ Friction damping



friction force F_d acts on the surface. force is F_d .
frictn(F_d) always opposes motion $\dot{x}(t)$

$$F_d = \mu_k n$$

$$\ddot{x}(t)$$

$$F_d = \mu_k n \quad \text{normal exn, coeff of fric}$$

dynamic/kinetic friction

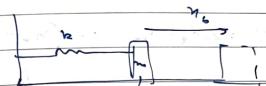
$$F_d = M \cdot N (-\dot{x})$$

Signum function

- note, signum $\text{sgn } \dot{x}$ is not linear over the domain



- split the problem into 2 pieces based on the sign of \dot{x}



$$kx - F_d = m\ddot{x}$$

$$\text{initial cond} \quad x(0) = x_0 \quad \dot{x}(0) = 0$$

$$\begin{aligned} m\ddot{x} + kx - F_d &= 0 \quad \text{if } \dot{x} < 0 \text{ moving left} \\ m\ddot{x} + kx &= F_d \end{aligned}$$

$$m\ddot{x} + kx + F_d = 0 \quad \text{if } \dot{x} > 0 \text{ moving right}$$

once $\dot{x} > 0$, moving right

$$m\ddot{x} + kx + F_d = 0$$

$$m\ddot{x} + kx = F_d \quad (2)$$

$$m\ddot{x} + kx + F_d \text{ sign}(\dot{x})$$

$$\ddot{x} + \frac{k}{m}x = \frac{F_d}{m} \cdot \frac{k}{k}$$

$$\ddot{x} + \omega_n^2 x = \omega_n^2 \cdot \frac{F_d}{m} \quad \text{non-dimensionalize let } \frac{F_d}{m} = \frac{f_d}{k}$$

so mechanical energy

$$\frac{1}{2} m \dot{x}^2 = \frac{1}{2} m \dot{x}_{\text{eff}}^2$$

$$n(t) = A \cos \omega_n t + B \sin \omega_n t + f_d$$

$$n(0) = n_0 = A + f_d$$

$$\dot{n}(t) = -A \omega_n \sin \omega_n t + B \omega_n$$

$$A = n_0 - f_d, B = 0$$

$$\begin{aligned} n(t) &= (n_0 - f_d) \cos \omega_n t + f_d \\ \dot{n}(t) &= n_0 (\omega_n - f_d) \sin \omega_n t \\ t &= \pi \frac{\theta}{\omega_n} \quad \dot{n}(t) \text{ versus direct} \end{aligned}$$

This solⁿ valid for $t \in [0, \pi/\omega_n]$

$$n(\pi/\omega_n) = f_d - n_0 + f_d = 2f_d - n_0 = -(n_0 - 2f_d)$$

$$\dot{n}(\pi/\omega_n) = 0$$

amplitude dec on at the
side due to fricⁿ.

nextual

$$m\ddot{n} + kn = -F_d$$

$$\frac{\pi}{\omega_n} \dot{n}(0) = -(n_0 - 2f_d)$$

$$\frac{\pi}{\omega_n} \dot{n}(0) = 0$$

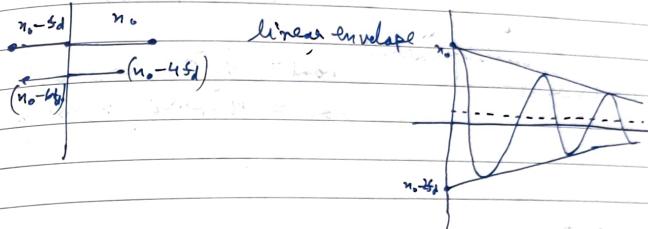
$$n(t) = (n_0 - 3f_d) \cos \omega_n t - f_d$$

again at $2\pi/\omega_n$ change of direct.

undamped system sys at $\omega = \frac{2\pi}{\omega_n}, \frac{2\pi}{\omega_n} \dots$ rouser

friction damping does not affect frequency of oscillation

→ amplitude



this cause difficulty when we approx using viscous.

$$e^{-n} = 1 - \frac{n}{1!} + \frac{n^2}{2!} - \dots$$

$$e^{-n} = 1 - n + O(n^2)$$

expantial fit as linear

$$n = \omega_n t$$

$\therefore n$ changes, progressing loss of accuracy

→ Forced vibration

- nature/duration of force (static, dynamic) finite duration
 - duration of load periodic non-periodic
 - type of response also cont or discontin.
- (decay, oscillatory, steady)
direct \rightarrow short
- $\text{load} = 1/\text{load period}$
- $\text{load} = n \text{ load period} \rightarrow \text{long}$
- (cont T period)

→ go to

- Fourier analysis/series
- Laplace
- convolution integral

10/10/22

→ Forced vibration

- Periodic forcing: trigonometric / harmonic
- non-periodic: long duration, short duration etc
convolution integral
fourier also work

→ Damped system subject to harmonic loading

$$m\ddot{x} + c\dot{x} + kx = F_0 \cos \omega t$$

$$\ddot{x} + \frac{c}{m}\dot{x} + \frac{k}{m}x = \frac{F_0}{m} \cos \omega t$$

$$\zeta = \frac{c}{2\sqrt{km}} = \frac{c}{m\sqrt{\frac{1}{k}}\frac{1}{2}} = \frac{1}{2} \frac{c}{m\omega_n}$$

$$c = 2\zeta m \omega_n$$

$$\ddot{x} + 2\zeta \omega_n \dot{x} + \omega_n^2 x = \frac{F_0}{m} \cos \omega t$$

$$\frac{F_0}{m} = \frac{F_0}{k} \frac{k}{m} = \frac{F_0 \omega_n^2}{k}$$

$$\ddot{x} + 2\zeta \omega_n \dot{x} + \omega_n^2 x = \frac{F_0}{k} \omega_n^2 \cos \omega t$$

$$x(t) = \text{homogeneous} + \text{particular}$$

$$= e^{-\zeta \omega_n t} [A \cos \omega_n t + B \sin \omega_n t] + x_p(t)$$

$$n_p(t) >$$

will be of the form
 $\alpha \cos \omega t + \beta \sin \omega t$

$$\rightarrow \omega^2 n_p(t) + -\alpha \omega \sin \omega t + \beta \omega \cos \omega t$$

$$\omega^2 n_p(t) + (\alpha \omega \sin \omega t + \beta \omega \cos \omega t) = F_0 w_n^2 \cos \omega t$$

group all 'sin' terms together
all 'cos' terms together & equate.

$$n_p(t) = \frac{A_0}{\sqrt{1 - \left(\frac{\omega}{\omega_n}\right)^2}}$$

$$n_p(t) = \frac{F_0}{k G_1} \cdot \frac{1}{\omega_n} \left[2\beta \omega \sin \omega t + \left(1 - \left(\frac{\omega}{\omega_n} \right)^2 \right) \cos \omega t \right]$$

$$\text{where } G_1 = \left(1 - \left(\frac{\omega}{\omega_n} \right)^2 \right)^2 + \left(\frac{2\beta\omega}{\omega_n} \right)^2$$

- we don't care about un-damped system (damped)
cos it $\rightarrow 0$ as $t \rightarrow \infty$.

keeping at sys. for long t , particular dominates.

- Damped system $n_p(t) \rightarrow 0$ as $t \rightarrow \infty$ only $n_p(t)$ remains
- response depends only on forcing, not on initial cond?
(long time record).

$$\frac{F_0}{k} = \omega_n \text{ static disp.} = n_s$$

$$n_p(t) = \frac{(F_0/k)}{\sqrt{G_1}} \sin(\omega_n t - \phi) \quad \text{delay due to damping}$$

scaled static response

- what happens when $\omega \rightarrow \omega_n$

$$G_1 = 45^2$$

for undamped $G_1 = \infty$

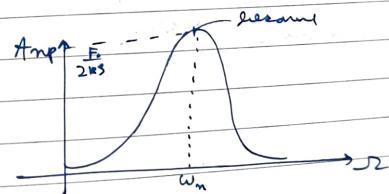
$n_p(t) \rightarrow \infty$ for undamped sys.

$\omega \rightarrow \omega_n$ called 'resonance'

$\omega = \omega_n$ 'at' pt of resonance

amplitude at resonance $\left(\frac{F_0/k}{G_1} \right) = n_{p_{max}}$

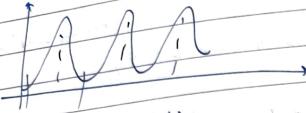
another way to measure ω
damping.



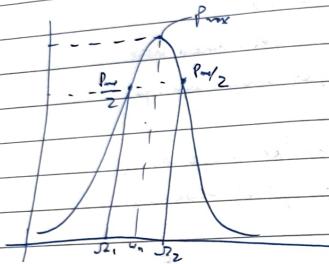
plots must be non-dimensionalised

$$\omega \rightarrow \frac{\omega}{\omega_n} \quad \text{and} \quad \frac{n_p}{n_s} \rightarrow \frac{F_0 k}{2G_1} \rightarrow \frac{1}{2}$$

free and nat



In chapter you get one of the



- using panel lectures error due to integration.
- half panel band width method.

- Integration reduces errors. $f(n) \Delta n$
- differentiation inc. errors, $f(n)/\Delta n$

→ work done over a cycle (ω)

$$m\ddot{x} + c\dot{x} + kx = F_0 \cos \omega t$$

$$\begin{aligned} x_p(t) &= A \sin \omega t + B \cos \omega t \\ &= A \cos(\omega t - \phi) \end{aligned}$$

- work done by forces in system

inertial, stiffness, damping, applied force

→ multiple terms in the force freq.)

$$F(t) = F_1 \cos \omega_1 t + F_2 \cos \omega_2 t + \dots$$

L.T. I. → superpos.

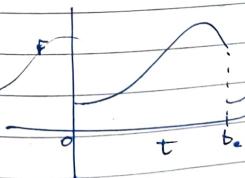
or fourier series expand.

→ periodic but not harmonic forces

$$F(t) = \sum \text{fourier series terms}$$

- Gibbs' phenomenon

→ non-periodic



- heat the system to being periodic

13/10/22

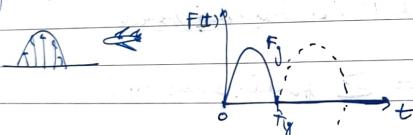
→ Methods to take:
1) POF system.

- periodic loads
 - ↳ one ton harmonic
 - ↳ multiple tones (Fourier)
 - ↳ general smooth
 - ↳ general non-smooth
- Fourier transform / series expansion
- can use Fourier but gives errors
(aliasing phenomenon)

- Non-periodic loads
 - ↳ smooth
 - ↳ general
 - don't treat the system into periodic 1 cycle
 - (tricky) still Fourier usable. but avoid at certain maybe unacceptable.

superposition & linear combos for multiple tones.
 $y(t) = \sum y_i(t) + y_p(t)$

- quiet problem in aerospace



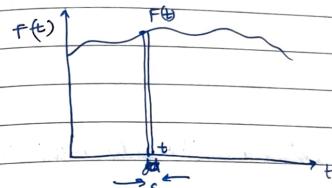
$$m\ddot{x} + c\dot{x} + kx = F_g(t) \quad t \in [0, T_g]$$

$$m\ddot{x} + c\dot{x} + kx = 0 \quad t > T_g$$

we'll talk about convolution

convolution integral

- unit impulse Force \times time (N_s)
- change in momentum metric (linear)
- small time segment "small" compared to duration of force's application.



$$F(t) \cdot \epsilon = I(t)$$

- force \times seg as "impulse" is the entire idea.

- determine response of the system to a unit impulse at some arbitrary time.
- scale to match impulse due to force (linearity)
- sum over all impulses that make up the force.
(superposition)

- define unit impulse based on the Dirac delta δ .

$$\delta(t - \tau) = 0 \quad t \neq \tau$$

$$\int_{-\infty}^{\infty} \delta(t - \tau) dt = 1$$

- we this don't make sense. consider the -
time varying δ .

$$\int_{-\infty}^{\infty} \delta(t - \tau) dt = 1$$

Define $\hat{F} = \int_t^{t+\Delta t} F(t) dt$ impulse at $F(t)$ at time t
 $= \hat{f} \cdot \Delta t$

unit impulse when $\hat{F} = 1$

response to unit impulse at time t' is

$$\lim_{\Delta t \rightarrow 0} \hat{f} = \hat{f} \cdot \Delta t$$

Suppose we want to obtain a unit impulse at time t' .

$$\hat{f} = \lim_{\Delta t \rightarrow 0} \int_t^{t+\Delta t} F dt \quad \leftarrow \text{this } \Rightarrow \text{ needed.}$$

$$\text{as } \Delta t \rightarrow 0, \int_t^{t+\Delta t} F(t) dt = 1 \quad \text{yields a delta function.}$$

$$m\ddot{x} + c\dot{x} + kx = \delta(t-t')$$

\leftarrow this is incorrect way to think.

$$\text{instead } \int_t^{t+\Delta t} (m\ddot{x} + c\dot{x} + kx = \delta(t-t')) dt$$

impulse causes change in momentum

$$\begin{aligned} \int_t^{t+\Delta t} m\ddot{x} &= m\dot{x}(t+\Delta t) - m\dot{x}(t) \\ \int_t^{t+\Delta t} c\dot{x} dt &= c\dot{x}(t+\Delta t) - c\dot{x}(t) \\ \int_t^{t+\Delta t} kx dt &= 0 \quad \text{based on property of a position.} \end{aligned}$$

$$\lim_{\Delta t \rightarrow 0} m\dot{x}(t+\Delta t) = \int_t^{t+\Delta t} F(t) dt = 1$$

unit impulse

$$\dot{x}(t^+) = 1$$

$$\dot{x}(t^+) = \frac{1}{m}$$

* impulse at time t' produces an initial $\dot{x}(t^+) = 1/m$ to a system at rest.

* what is response of the sys to the unit impulse.

$$m\ddot{x} + c\dot{x} + kx = 0 \quad \text{will be solved after this of 1/m}$$

$$\dot{x}(0) = 0$$

$$\dot{x}(0) = 1/m$$

unit impulse applied at $t=0$. $\dot{x}(0) = 1/m$
 will change later to unit impulse applied at time t .

$$x(t) = e^{-\beta w_m t} [A \cos \omega_m t + B \sin \omega_m t]$$

$$A = 0$$

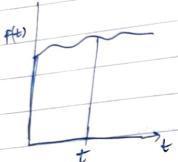
$$x(t) = -\beta w_m e^{-\beta w_m t} B \sin \omega_m t + \omega_m e^{-\beta w_m t} B \cos \omega_m t$$

$$= -\omega_m^2 B \cos \omega_m t = 1/m$$

$$B = 1/w_m$$

$$A=0, B=\frac{1}{m\omega_d}$$

$$n(t) = \frac{e^{-\frac{\omega_d}{2}t} \sin \omega_d t}{m \omega_d}$$



unit impulse applied at $t=t$, system at rest prior to impulse. $t \rightarrow t-t$

$$n(t-t)$$

L

$$n(t-t) = \frac{e^{-\frac{\omega_d}{2}(t-t)}}{m \omega_d} \sin(\omega_d(t-t))$$

If \hat{F} is the magⁿ of impulse due to force at time $t-t$. Scale up $n(t-t)$ by \hat{F}

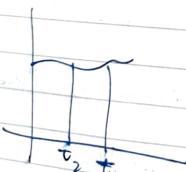
$n(t-t)$ is unit impulse response at time t .

$g(t-t)$ formal defⁿ.

Scale up by $\hat{F}(t)$

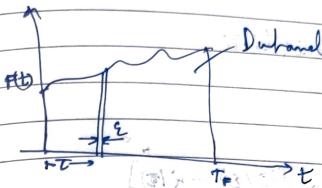
$$\hat{F}(t) g(t-t)$$

$$\int_0^t \hat{F}(t) g(t-t) dt \text{ total response after } \cancel{\text{impulse}}$$



$$\text{response only } R(t_1) + R(t_2) - \dots$$

second attempt at convolution.



Duhamel's integral

Let's apply $F(t)$ as a sequence of impulses

- 1) obtain response to a unit impulse at some arbitrary time $t \in [0, T_f]$
- 2) scale response for impulse due to $F(t)$
- 3) Prepare to handle summation of responses to series of impulses.

$$\text{Data: } \hat{F}(t) = \text{impulse of interest}$$

$$\hat{F} = F(t) \epsilon$$

impulse of interest
impulse from t to $t+\epsilon$

assumption

- system at rest prior to impulse.
- $n(t)$ can't be dis. continuous.
- ii, ii can be discontinuous.

$$\min + c_i \epsilon + b_n = F(t)$$

\min (from $F(t)$ width ϵ)

$$\int_{t-\epsilon}^{t+\epsilon} \min + c_i \epsilon + b_n = F(t)$$

$$= F(t) \epsilon$$

ϵ is same as $\Delta t, dt$

$$\int_{\varepsilon}^{T+\varepsilon} m \ddot{x}(t) dt$$

say at next period to t

$$\tilde{t} = t - \varepsilon$$

$$\int_{\varepsilon}^{\tilde{t}} m \ddot{x}(\tilde{t}) d\tilde{t} = m[\dot{x}(\varepsilon) - \dot{x}(0)]$$

$$= m \ddot{x}(t+\varepsilon) |_{t=0} \quad (\text{using } \ddot{x}(t) \text{ is constant})$$

$$= m \ddot{x}(\varepsilon), \quad t=0$$

\therefore response period to impulse is 0.

$$m \ddot{x}(\varepsilon) = \int_{\varepsilon}^{\tilde{t}} F(\tilde{t}) d\tilde{t} = F(0) \varepsilon$$

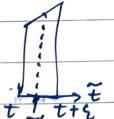
(simplified,

$$\int_{\varepsilon}^{\tilde{t}} c_i(\tilde{t}) d\tilde{t} = c_i[n(\varepsilon) - n(0)]$$

$$\lim_{\varepsilon \rightarrow 0} c_i(n(\varepsilon)) \rightarrow 0$$

$$\int_{\varepsilon}^{\tilde{t}} k_n(\tilde{t}) d\tilde{t} = k_n(\varepsilon) \Delta \tilde{t}$$

$$\lim_{\varepsilon \rightarrow 0} k_n(\tilde{t}) \frac{d\tilde{t}}{\varepsilon} \rightarrow 0$$



$$m \ddot{x}(\varepsilon) = \int_{\varepsilon}^{\tilde{t}} F(\tilde{t}) d\tilde{t}$$

$$\lim_{\varepsilon \rightarrow 0} m \ddot{x}(\varepsilon) = \int_{\varepsilon}^{\tilde{t}} F(\tilde{t}) d\tilde{t} |_{\varepsilon=0}$$

$$m \ddot{x}(0) = 1$$

$$\ddot{x}(0^+) = 1/m$$

$$m \ddot{x}(\tilde{t}) + c_i(\tilde{t}) + b_n(\tilde{t}) = 0$$

$$n(0) = 0$$

$$\ddot{x}(0^+) = 1/m$$

$$n(\tilde{t}) = \frac{1}{m} e^{-\frac{c_i \tilde{t}}{m}} \sin \omega_m \tilde{t} = g(\tilde{t}) \text{ called}$$

response to unit impulse at $\tilde{t}=0$
response to impulse of any $F(0) \varepsilon$

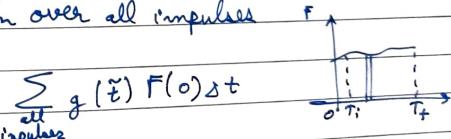
$$\text{scale, } n(\tilde{t}) = F$$

$$n(\tilde{t}) = F(0) \cdot \varepsilon$$

$$n(\tilde{t}) = F(0) \cdot \Delta t$$

$$g(\tilde{t}) = F(0) \Delta t \quad \tilde{t} > 0$$

summation over all impulses



$$\int_{\varepsilon}^{T_f} g(t-\varepsilon) F(t) dt \text{ or } \int_{\varepsilon}^{T_f} g(t-T) F(t) dt$$

total response for any force.

Most basic way is $F(t)$ to be integrable $\int F(t) \varepsilon$

→ random signals

- noise

- earth quake

- shocks

- turbulence

randomized distribution of freq.
amp.

→ No bias or 'unknown bias'

such loads known as transient load.

→ applications

- response calculations

 |
 | ICS

 |
 | shock

 |
 | Freq. func.

 |
 | steady state loads
(long duration)

2d/10/22

$$m\ddot{x} + c\dot{x} + kx = F(t)$$

$$F(t) = F_0 \sin \omega t$$

$m\ddot{x} + kx = 0$ free vibration.

$c = mN$ column damping
 $c = h/w$ structural damping

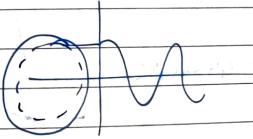
only thing to excite the system is initial cond² in frequencies

$$x(t) = \frac{v_0}{\omega_n} \sin \omega_n t + u_0 \cos \omega_n t$$

$$\cancel{x} = v(t) = R \sin(\omega t + \phi)$$

$$R = \sqrt{v_0^2 + u_0^2} \quad \phi = \tan^{-1} \frac{u_0}{v_0}$$

$$\omega_n = \sqrt{\frac{k}{m}}$$

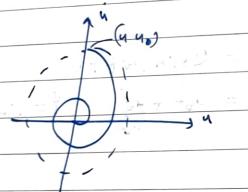
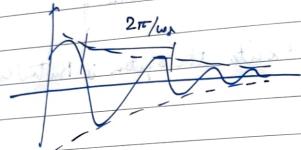


initially damped → spiderman 2 train scene.

$$u(t) = e^{-\omega_0 t} \left[u_0 \cos \omega_0 t + \frac{(u_0 + \omega_0 s_0)}{\omega_0} \sin \omega_0 t \right] = R e^{-\omega_0 t} \sin(\omega_0 t + \phi)$$

$$R = \sqrt{\left(\frac{u_0}{\omega}\right)^2 + 2u_0 \left(\frac{u_0}{\omega}\right)s + s^2}$$

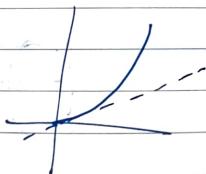
$$\omega_0^2 = \frac{u_0^2/\omega^2 + u_0^2}{u_0^2/(\omega^2)}$$



$$\delta = \frac{2\pi s}{\sqrt{1-s^2}}$$

linearized

$$\delta = 2\pi s$$



$$x(t) = (C_1 + C_2 t) e^{-\omega_0 t}$$

critically damped

$$C_1 = x_0$$

$$C_2 = u_0 + \omega_0 x_0$$

$$f(t) = \frac{a_0}{2} + a_1 \cos \omega_0 t + a_2 \cos 2\omega_0 t + \dots + a_m \sin \omega_0 t + b_m \sin 2\omega_0 t$$

$$= \frac{a_0}{2} + \sum_{j=1}^{\infty} a_j \cos j\omega_0 t + b_j \sin j\omega_0 t$$

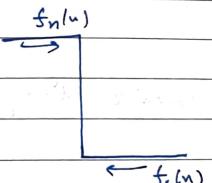
$$a_j = \frac{2}{T} \int_0^T f(t) \cos jt dt$$

$$\sum_{n=1}^{\infty} F_n(u) = f(u)$$

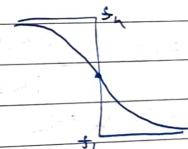
Fourier convergence

valid when $f(u)$ smooth & cont.

$$\lim_{n \rightarrow \infty} |F_n(u) - f(u)| < \epsilon$$



$$\lim_{n \rightarrow \infty} F_n(x) = \frac{1}{2} (f_L + f_R)$$



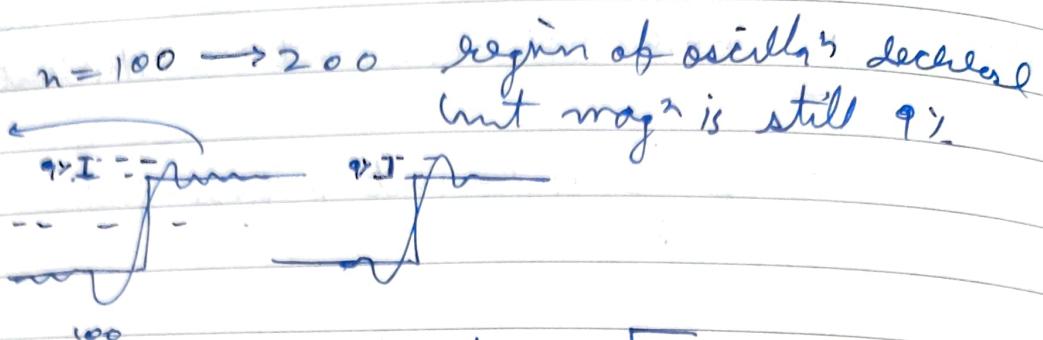
Gibbs' oscillations

- i) every 'n' where f is discont.
- ii) magnitude of error - size of discont
- iii) gibbs - const

$$\lim_{n \rightarrow \infty} \frac{\int_0^T f(t) dt}{T} = 0.089 \dots$$

overperiodic

97.8% error at discont and



in aerospace, shocks have \square type of discontin.

gives oscillation \rightarrow high freq comp.

\downarrow
 use low pass filter

\downarrow
 remove this oscill.

gives oscillation occurs when f is discontinuous & non diff



$$x(t) = g(t) = \frac{e^{-j\omega_n t}}{m \omega_d} \sin \omega_d t$$

$$n_0 = 0 \quad n_0 = 1/m$$

Damping - viscous

Forcing - Lateral, non-linear (periodic) cont. gear box