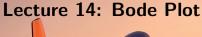
# AE 308: Control Theory AE 775: System Modelling, Dynamics & Control





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### **Bode Plot - Introduction**



#### **Bode Plot**

ullet In Bode plots or Bode diagrams, the log-magnitude and phase frequency response curves are as functions of  $\log \omega$ .

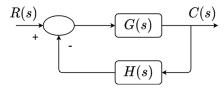
#### **Features**

- Because the magnitude of  $G(j\omega)$  in the Bode plot is expressed in dB, product and division factors in  $G(j\omega)$  become additions and subtractions, respectively. The phase relations are also added and subtracted from each other algebraically.
- ② The magnitude plot of the Bode plot of  $G(j\omega)$  can be approximated by straight-line segments, which allow the simple sketching of the plot without detailed computation.

### **Bode Plot - Introduction**



#### Closed loop control system



Closed loop transfer function:

$$T(s) = \frac{C(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

• Thus characteristic equation of the closed loop system is

$$CE = 1 + G(s)H(s) = 0$$

• G(s)H(s) is called as Loop Transfer Function.

### **Asymptotic Bode Plot - Introduction**



### **Asymptotic Bode Plots**

- Sketching Bode plots can be simplified, because they can be approximated as a sequence of straight lines.
- Straight-line approximation of the Bode plot is relatively easy to construct.
- The data necessary for the other frequency-domain plots, such as the polar plot and the magnitude-versus-phase plot, can be easily generated from the Bode plot.

# **Asymptotic Bode Plot - Introduction**



### Asymptotic Bode Plots (cont...)

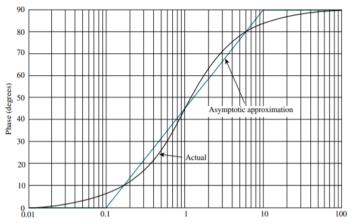


Figure: Source - "Control Systems Engineering", by N. S. Nise

### **Bode Plot - Example**



### Example 1

• Loop transfer function is

$$G(s)H(s) = \frac{(s+20)}{(s+2)(s+400)}$$

• Bring in standard form

$$G(s)H(s) = \frac{\left(\frac{s}{20} + 1\right)}{40\left(\frac{s}{2} + 1\right)\left(\frac{s}{400} + 1\right)}$$

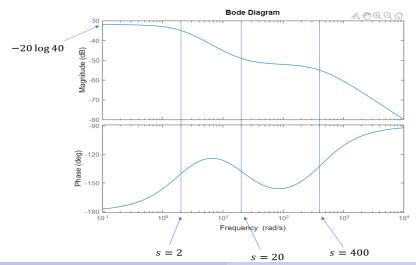
Here,

$$k = \frac{1}{40}$$

# **Bode Plot - Example**



### Example 1 (cont...)



### **Asymptotic Bode Plot - Analysis**



#### **Features**

- All bode plots show specific pattern for two limiting frequency points 0 and  $\infty$ .
- This is typically in the form of low and high frequency asymptotes, which provide DC gain G(0) and relative degree (i.e. n-m) of G(s).
- We also find that changes in the asymptote angles occur around frequencies that correspond to poles and zeros.

### **Asymptotic Bode Plot - Analysis**



### Features (cont...)

- Poles and zeros are seen as points where slope of magnitude plot changes and termed as corner frequencies.
- G(0) and k are seen as intercept, which is  $20\log k$ , and slope, which is -20L dB/decade for  $\omega=0$  where  $G(s)=k/s^L$ .
- (n-m) is seen as slope of curve for  $\omega \to \infty$ , which is -20(n-m) dB/decade.



### Question

Draw a bode plot for transfer function with unity feedback

$$G(s) = 100 \frac{(s+1)}{(s+10)(s+100)}$$

### Step 1: Rewrite the transfer function in proper form

$$G(s)H(s) = \frac{100}{10 \times 100} \frac{(s+1)}{\left(\frac{s}{10} + 1\right) \left(\frac{s}{100} + 1\right)}$$



### Step 2: Separate the transfer function into its constituent parts

- A constant of 0.1
- A pole at s=-10
- A pole at s=-100
- A zero at s=-1



### Step 3: Draw the Bode diagram for each part

- The constant is the cyan line (a quantity of 0.1 is equal to -20 dB). The phase is constant at 0 deg.
- Pole at 10 rad/sec is the green line. It is  $0~\mathrm{dB}$  up to the break frequency, then drops off with a slope of  $-20~\mathrm{dB/dec}$ . Phase is  $0~\mathrm{deg}$  up to  $1/10~\mathrm{times}$  of the break frequency, i.e.  $1~\mathrm{rad/sec}$ , then drops linearly down to  $-90~\mathrm{deg}$  at  $10~\mathrm{times}$  of the break frequency, i.e.  $100~\mathrm{rad/sec}$ .
- Pole at  $100~\rm rad/sec$  is the blue line. It is  $0~\rm dB$  up to the break frequency, then drops off with a slope of  $-20~\rm dB/dec$ . Phase is  $0~\rm deg$  up to  $1/10~\rm times$  of the break frequency, i.e.  $10~\rm rad/sec$ , then drops linearly down to  $-90~\rm deg$  at  $10~\rm times$  of the break frequency, i.e.  $1000~\rm rad/sec$ .
- Zero at 1 rad/sec is the red line. It is  $0~{\rm dB}$  up to the break frequency, then rises at  $20~{\rm dB/dec}$ . Phase is  $0~{\rm deg}$  up to  $1/10~{\rm times}$  of the break frequency, i.e.  $0.1~{\rm rad/sec}$ , then rises linearly to  $90~{\rm deg}$  at  $10~{\rm times}$  of the break frequency, i.e.  $10~{\rm rad/sec}$ .



Step 4: Add the results from step 3

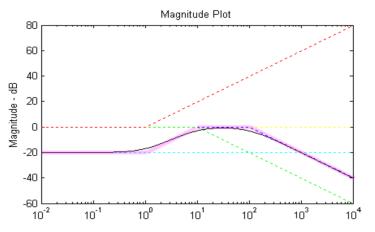


Figure: Source - "https://lpsa.swarthmore.edu/Bode/BodeExamples.html"



Step 4: Add the results from step 3

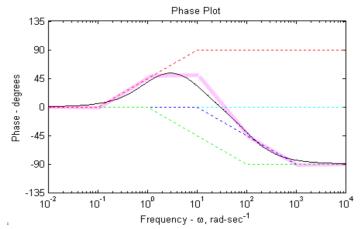


Figure: Source - "https://lpsa.swarthmore.edu/Bode/BodeExamples.html"

### **Bode Plot - Drawbacks**



#### **Drawbacks**

- Bode plot consists of two graphics, which need to be interpreted together.
- However, in some cases, there is a need to see complete frequency response in a single graphic and Nyquist plot addresses this need.

### **Concept of Crossover**



### Gain crossover (GCO) point

- The gain-crossover point on the frequency-domain plot of  $G(j\omega)$  is the point at which  $|G(j\omega)|=1$  or  $|G(j\omega)|_{dB}=0$ .
- The frequency at the gain-crossover point is called the gain-crossover frequency  $\omega_{GCO}$ .

### Phase crossover (PCO) point

- The phase-crossover point on the frequency-domain plot of  $G(j\omega)$  is the point at which  $\angle G(j\omega) = -180^{0}$ .
- The frequency at the phase-crossover point is called the phase crossover frequency  $\omega_{PCO}$ .

### **Bode Plot - Crossover Points**



#### Gain and Phase crossover points

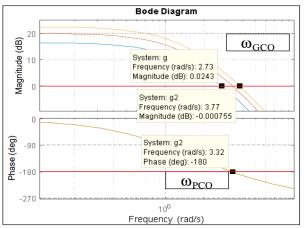


Figure: Source - "System Modeling Dynamics and Control, Lecture Notes", by Prof. Ashok Joshi

### **Crossover - Features**



#### **Features**

- $\omega_{GCO}$  and  $\omega_{PCO}$  contain stability information as much as that these are related to the corresponding location of closed loop poles in s-plane.
- ullet Therefore, magnitude and phase at PCO and GCO are used as quantitative measures of relative stability of the closed loop system, also called **margins**.

# **Stability Margins - Introduction**



### **Gain Margin**

- Gain Margin (GM) is the amount of gain that can be added to plant, before unity feedback closed loop system becomes unstable.
- ullet GM is the reciprocal of the magnitude at  $\omega_{PCO}$  and is given by

$$\frac{1}{|G(j\omega_{PCO})|}$$

• Gain margin is considered positive, if  $|G(j\omega_{PCO})| < 1$ .

# Stability Margins - Bode Plot



#### Gain Margin

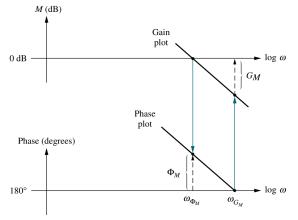


Figure: Source - "Control System Engineering", by Norman S. Nise

# **Stability Margins - Introduction**



### **Phase Margin**

- Similarly, phase margin (PM) is the amount of phase lag (or negative angle) that can be added to the plant before the closed loop system becomes unstable.
- As negative angle is measured clockwise from positive real axis, PM is defined as  $180^0 + \angle G(j\omega_{GCO})$ .
- PM is treated as positive if negative angle is  $< 180^{\circ}$ .

# Stability Margins - Bode Plot



### **Phase Margin**

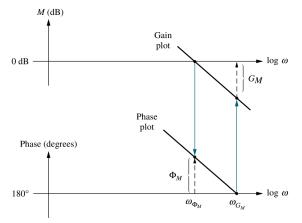


Figure: Source - "Control System Engineering", by Norman S. Nise



### Example 1

 Determine GCO, PCO, GM and PM for following plant and predict the stability of closed loop system. Also, correlate with closed loop pole locations

$$G(s) = \frac{10}{s(s+1)(s+5)}$$



### Example 1

 Determine GCO, PCO, GM and PM for following plant and predict the stability of closed loop system. Also, correlate with closed loop pole locations

$$G(s) = \frac{10}{s(s+1)(s+5)}$$

#### Solution

Let find GCO

$$|G(j\omega)| = \left| \frac{10}{j\omega(j\omega+1)(j\omega+5)} \right| = \frac{10}{\omega\sqrt{\omega^2+1}\sqrt{\omega^2+25}}$$
$$\frac{10}{\omega_{GCO}\sqrt{\omega_{GCO}^2+1}\sqrt{\omega_{GCO}^2+25}} = 1$$



### Example 1 (cont...)

Simplifying further

$$\omega_{GCO}^6 + 26\omega_{GCO}^4 + 25\omega_{GCO}^2 - 100 = 0$$

$$\omega_{GCO}^2 = 1.506 \to \omega_{\mathbf{GCO}} = \mathbf{1.227}$$

Let find PCO

$$\angle G(j\omega) = -90^{0} - \tan^{-1}(\omega) - \tan^{-1}(0.2\omega)$$
$$-90^{0} - \tan^{-1}(\omega_{PCO}) - \tan^{-1}(0.2\omega_{PCO}) = -180^{0}$$
$$\omega_{PCO} = 2.25$$



### Example 1 (cont...)

Let find GM

$$GM = \left| rac{1}{G(j\omega_{PCO})} 
ight|$$

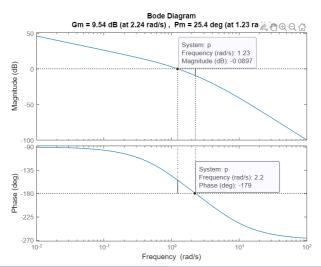
$$= \left| rac{1}{G(2.25j)} 
ight| = 3.03 = \mathbf{9.63 \ dB}$$

Let find PM

$$PM = 180^{0} + \angle G(j\omega_{GCO})$$
  
=  $180^{0} + \angle G(1.227j) = 25.4^{0}$ 



#### **Example 1 - Verification**



# **Infinite Stability Margins - Introduction**



#### Infinite GM

- There are situations where the phase plot either does not cross  $180^{0}$ , or phase cross over occurs at  $\omega=\infty$ .
- In such cases, the GM becomes undefined and is commonly interpreted to be infinite.
- Implication of such a situation is that no amount of increase in gain will destabilize the closed loop system.

#### Infinite PM

• There are situations where the magnitude plot either does not cross 0 dB, or gain cross over occurs at  $\omega = \infty$ .

# Infinite Stability Margins - GM



#### Infinite GM

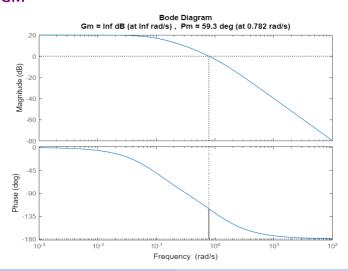
Consider a system

$$G(s) = \frac{1}{(s+0.1)(s+1)}$$

# Infinite Stability Margins - GM



#### Infinite GM



# Infinite Stability Margins - PM



#### Infinite PM

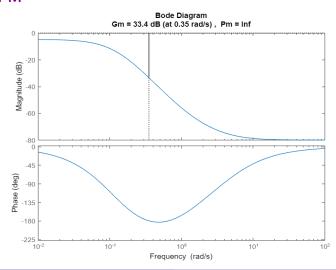
• Consider a system

$$G(s) = \frac{0.0001(s+1)(s+2)(s+5)}{(s+0.12)^3}$$

# Infinite Stability Margins - PM



#### Infinite PM



# Infinite Stability Margins - GM and PM



#### Infinite GM and PM

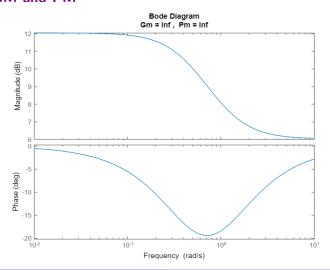
• Consider a system with unity feedback

$$G(s) = \frac{2(s+1)}{(s+0.5)}$$

# Infinite Stability Margins - GM and PM



#### Infinite GM and PM



### Non Minimum Phase System - GM and PM



### Non minimum phase - GM and PM

- Phase characteristics of non-minimum phase systems are significantly different and hence it is expected that both GM and PM would get affected.
- Let, consider a system

$$G(s) = \frac{2(s+1)e^{-s}}{(s+0.5)}$$

ullet We can employ  $1^{st}$  order Pade's approximation to get the rational transfer function as

$$G(s) = \frac{2(s+1)(1-0.5s)}{(s+0.5)(1+0.5s)}$$

# Non Minimum Phase System - GM and PM



### Non minimum phase - GM and PM (cont...)

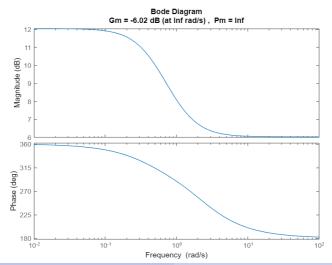
• After solving for the GCO, PCO, GM and PM, we get

$$\begin{split} \omega_{PCO} &= \infty \\ \omega_{GCO}^2 &= -0.083 \rightarrow \omega_{GCO} = \text{No} \\ GM &= -6.02 \text{dB} \end{split}$$

# Non Minimum Phase System - GM and PM



Non minimum phase - GM and PM (cont...)



### Summary



- Bode plot consists of two graphics, which need to be interpreted together.
- Magnitude plot changes at poles and zeros and these frequencies are called as corner frequency.
- Infinite gain and phase margins provide greater design freedom.
- Non-minimum phase behaviour has a significant impact on the stability margins.

### References 1



- Gene F. Franklin, J. David Powell, and Abbas Emami-Naeini: "Feedback Control of Dynamic Systems", Pearson Education, Inc., Upper Saddle River, New Jersey, Seventh Edition, 2015.
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