ME 104 Lec 9

Recall: Two equivalent versions of the workenergy theorem:

$$\begin{array}{c}
\text{(1)} \quad W_{AB}^{\text{Tot}} = \sum_{A11}^{\infty} \int_{Path}^{E^{i}} \frac{dr}{r} = \frac{1}{2} M V_{B}^{2} - \frac{1}{2} M V_{A}^{2} \\
\text{forces} \quad \underline{r}_{A} \Rightarrow \underline{r}_{B}
\end{array}$$

where
$$E = \frac{1}{2}mv^2 + \sum_{c}U^{c}$$
 energy of all conservative forces. $(E^{c} = -\nabla U^{c})$

Benefits of using work-energy:

Can determine certain features of the motion without having to

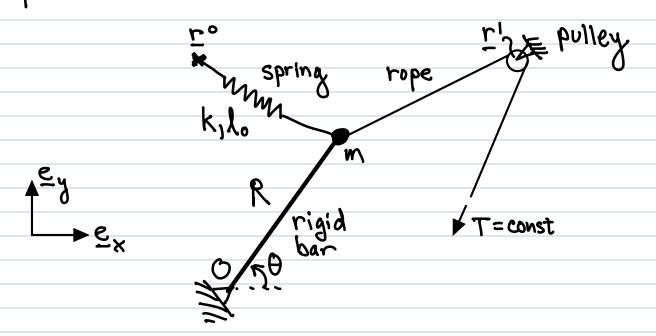
- · solve first for (powerless) constraint forces, e.g. normal wall forces or rigid bar forces,
- solve a system of equations,
 solve differential equations.

Issues using energy balance include:

- · hard to use if W_{AB}^{nc} is difficult to calculate (best if $W_{AB}^{nc} = 0 = \Delta E$, i.e. energy conserved)
- · since it is just one equation it can only tell you one thing about the motion.

Using energy balance, we can often infer unknown forces after the fact.

Example from last time:



We previously calcutated $|\dot{\theta}_f|$ at some $\dot{\theta}_f$ assuming mass started stationary at $\dot{\theta}_i$. What is E_{bar} at $\dot{\theta}_f$?

Ans: ZF=ma gives

Thus, for
$$\theta = \theta_f$$
 and $\dot{\theta}^2 = |\dot{\theta}_f|^2$, we have

$$F_{bar} = -mR |\dot{\theta}_{f}|^{2} - \left[F_{s}(\theta_{f}, R, r^{o}) + F_{rope}(\theta_{f}, R, r') \right]$$

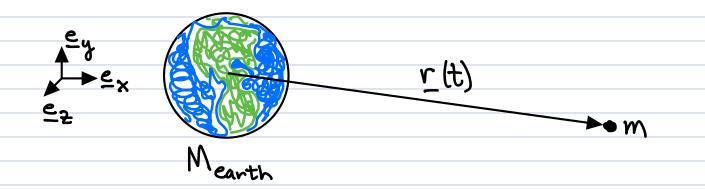
$$\cdot \left(\cos\theta_{f} \, \varepsilon_{x} + \sin\theta_{f} \, \varepsilon_{y} \right)$$

Since we know R, θ_f , $|\dot{\theta}_f|$, \underline{r}^o , and \underline{r}' we can compute boxed formula for F_{bar} .

In general, dotting $\Sigma F = ma$ into the direction normal to a wall let's you compute the normal force knowing only the particle's speed, position, and path curvature.

Orbital mechanics

Assume an object interacting with earth's gravity with m< Mearth. \Rightarrow The center of the earth can be assumed stationary. Let the origin be at earth's center.



$$\frac{2}{E_g} = \frac{GM_{earth}m}{|\underline{r}|^2} \left(\frac{-\underline{r}}{|\underline{r}|}\right)$$

(3) Defining
$$B = GM_{earth}$$
, $\Sigma F = m\underline{a}$ implies

$$a = -\beta r/|r|^3$$
 "Trajectory equation".

In terms of work-energy, F_g is conservative as a power-law force with $U_g = -B/|\underline{r}|$. Since no nc forces, energy is conserved so:

$$\frac{1}{2}mv^2 - \beta m/|\underline{r}| = const$$

4) Calculate the "escape velocity"— the speed that the mass must be launched with from earth's surface to escape earth's pull without any extra propulsion.

 $E(right after launch) = \frac{1}{2}mv_e^2 - \beta m/R_{earth}$

 $E(at escape) = \frac{1}{2}m \cdot 0^2 - \beta m/|r \rightarrow \infty| = 0$

 $\Rightarrow \frac{1}{2}mv_e^2 - \beta m/R_{earth} = 0 \Rightarrow v_e = \sqrt{\frac{2\beta}{R_{earth}}}$ = 11,200 m/s.

Need more than just energy balance to learn more trajectory behaviors. Must solve trajectory eq directly or use other conservation laws to augment energy balance.

Angular momentum: Let H=mr×v.

Note that H=mr×y+mr×y

$$= w_{\tilde{\Lambda}} \times \tilde{\Lambda} + L_{x}(\tilde{W}\tilde{a})$$

So H= "Angular momentum" = H_o is conserved.

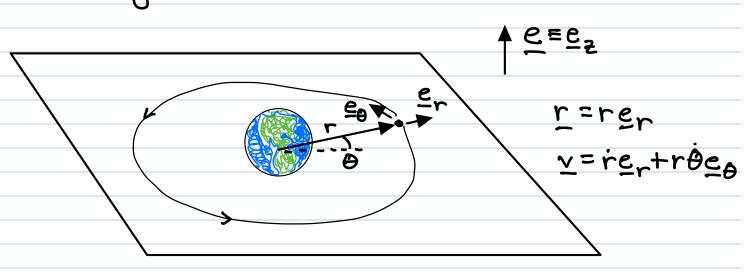
Generalizing:

H is hence conserved whenever a particle is subject to a net force || r. (2)

H=mrxv=Ho=Hoe for some fixed Ho and e.

Fixed e implies r and v must always lie in one fixed plane L to e. But this implies the trajectory of r(t) must lie in a single plane. Orbits always lie in one plane!

Knowing this, the problem is now 2D.



 $H = m(re_r) \times (re_r + r\theta e_{\theta}) = mr^2\theta e_z = H_0 e_z$ $\implies mr^2\theta = H_0$ where $H_0 = mr_i^2\theta_i$ $\implies \theta = H_0/mr^2$.

Now conserve energy:
$$\frac{1}{2}mv^2 - \beta m/r = E_0 = const$$

$$\Rightarrow \frac{1}{2}m(\dot{r}^2+r^2\dot{\Theta}^2)-\beta m/r=E_0$$

$$\Rightarrow \frac{1}{2}m\left(\dot{r}^2 + \frac{H_0^2}{m^2r^2}\right) - \beta m/r = E_0 \leftarrow ODE \text{ for } r(t).$$

Rather than solve for r(t), we can use this to learn the trajectory $r(\theta)$.

$$\dot{r} = \frac{dr}{d\theta} \dot{\theta} = \frac{dr}{d\theta} \frac{H_0}{mr^2}$$
. Plug in above:

$$\Rightarrow \frac{1}{2}m\left(\left(\frac{dr}{d\theta}\right)^2\left(\frac{H_0}{mr^2}\right)^2 + \frac{H_0^2}{m^2r^2}\right) - \beta m/r = E_0$$

$$\Rightarrow \frac{dr}{d\theta} = \left(\frac{2}{m}\left(E_0 + \beta m/r\right) - \frac{H_0^2}{m^2r^2}\right)\frac{m^2r^4}{H_0^2}$$

Solve by separation of variables. We get:

$$r(\theta) = \frac{p}{1 + e \cdot \cos(\theta - \phi)}$$

$$p = "trajectory parameter" = \frac{H_0^2}{m^2 \beta}$$

$$e = \text{"eccentricity"} = \left(1 + \frac{2E_0H_0^2}{m^3\beta^2}\right)^{1/2}$$

$$\phi = \text{"offset angle"} = \theta_i - \cos^{-1}\left(\frac{p-r_i}{er_i}\right)$$

What are these shapes? Conic sections! 0<e<1 e=0 ellipse circle e=1 e>1 Parabola hyperbola