



**i** FACULTEIT  
INGENIEURSWETENSCHAPPEN

## B-KUL-H04X3A: Control Theory

### Team 47:

Lefebure Tiebert (r0887630)  
Campaert Lukas (r0885501)

# Assignment 1: Identification of the cart

Professor:

Prof. Dr. Ir. Jan Swevers

Academic Year 2025-2026

## ***Declaration of Originality***

We hereby declare that this submitted draft is entirely our own, subject to feedback and support given us by the didactic team, and subject to lawful cooperation which was agreed with the same didactic team. Regarding this draft, we also declare that:

1. Note has been taken of the text on academic integrity <https://eng.kuleuven.be/studeren/masterproef-en-papers/documenten/20161221-academischeintegriteit-okt2016.pdf>.
2. No plagiarism has been committed as described on <https://eng.kuleuven.be/studeren/masterproef-en-papers/plagiaat#Definitie:%20wat%20is%20plagiaat?>.
3. All experiments, tests, measurements, ..., have been performed as described in this draft, and no data or measurement results have been manipulated.
4. All sources employed in this draft – including internet sources – have been correctly referenced.

# 1 Model structure selection for the cart

## 1.1 Discrete-time model structure selection (1a)

We assume that (i) the same voltage is applied to both motors, (ii) no wheel slip occurs, (iii) only longitudinal motion (no turning) is considered, and (iv) the cart mass and ground friction are lumped into the wheel inertia and viscous friction, respectively. Under these assumptions, the lumped-parameter model follows from the balance of angular momentum of the rotor and the conservation of electric charge in the armature circuit.

Taking the Laplace transform of these equations (see Figure 5 in Appendix B) and solving for  $\dot{\Theta}_m(s)/V_a(s)$  yields the following transfer function:

$$H(s) = \frac{\dot{\Theta}_m(s)}{V_a(s)} = \frac{K}{(J_m s + b)(L_a s + R_a) + K^2} \quad (1)$$

Symbol	Description
$\dot{\Theta}_m(s)$	Laplace transform of the angular velocity $\dot{\theta}_m$ [rad/s]
$V_a(s)$	Laplace transform of the applied voltage $v_a$ [V]
$J_m$	moment of inertia of the rotor [ $\text{kg m}^2$ ]
$b$	damping coefficient (viscous friction) [ $\text{N m s/rad}$ ]
$K$	motor torque and back EMF constant [ $\text{N m/A}$ ] or [ $\text{V s/rad}$ ]
$L_a$	armature inductance [H]
$R_a$	armature resistance [ $\Omega$ ]

This continuous-time transfer function relates the input voltage  $v_a(t)$  to the rotational speed of the motor shaft  $\dot{\theta}_m(t)$ . We can represent the transfer function with generic coefficients since the exact values of the physical constants are unknown and will be estimated later. Therefore,  $H(s)$  becomes:

$$H(s) = \frac{\dot{\Theta}_m(s)}{V_a(s)} = \frac{a}{bs^2 + cs + d} \quad (2)$$

### 1.1.1 Discrete-time transfer function

The discrete-time transfer function with computational delay is:

$$H(z) = \frac{b_1 z + b_2}{z^3 + a_1 z^2 + a_2 z} \quad (3)$$

This is a third-order model with numerator order 1, denominator order 3, and relative order 2 (indicating two delays: one from motor dynamics and one from microOS latency).

### 1.1.2 Model derivation

To obtain a discrete-time model, the continuous-time system

$$H(s) = \frac{a}{bs^2 + cs + d}$$

is sampled with sampling period  $T_s$  under a zero-order hold (ZOH) on the input. The ZOH assumption implies that the input voltage is piecewise constant on each sampling interval  $[kT_s, (k+1)T_s]$ . For a strictly proper second-order transfer function, ZOH discretization yields a second-order discrete-time transfer function

$$H_0(z) = \frac{b_1 z + b_0}{z^2 + a_1 z + a_0}, \quad (4)$$

with coefficients  $a_0, a_1, b_0, b_1$  that follow from the sampled state-space dynamics

$$x((k+1)T_s) = e^{AT_s}x(kT_s) + \int_0^{T_s} e^{A\tau} B d\tau u(kT_s),$$

with  $y(kT_s) = Cx(kT_s)$ , and then forming the corresponding transfer function.

The computational latency of the microOS introduces an additional one-sample delay between the computed control signal and the applied motor voltage. In the  $z$ -domain this corresponds to a multiplicative factor  $z^{-1}$ , so the overall discrete-time transfer function becomes

$$H(z) = z^{-1}H_0(z) = \frac{b_1z + b_2}{z^3 + a_1z^2 + a_2z}, \quad (5)$$

During identification these four parameters are estimated directly from the measured input-output data, so no explicit mapping back to the continuous-time coefficients is required.

### 1.1.3 Model structure motivation

We consider two candidate model structures: (i) the full third-order model above, which accounts for armature inductance, and (ii) a reduced second-order model obtained by neglecting the armature inductance  $L_a$ , leading to

$$H(z) = \frac{b_1}{z^2 + a_1z},$$

with only two parameters.

## 1.2 Model parameters estimation (1b)

We estimate the parameters of the third-order discrete-time transfer function  $H(z) = \frac{b_1z + b_2}{z^3 + a_1z^2 + a_2z}$  using linear least squares. The matrix formulation of the parameter estimation problem of the second-order model is presented in Appendix A.

### 1.2.1 Recursion expression and error criterion

The difference equation for the third-order model is [2]:

$$\dot{\theta}[k] = -a_1\dot{\theta}[k-1] - a_2\dot{\theta}[k-2] + b_1v[k-2] + b_2v[k-3] \quad (6)$$

This can be written in compact form as  $\dot{\theta}[k] = \boldsymbol{\theta}^T \boldsymbol{\phi}[k]$  with parameter vector and regression vector:

$$\boldsymbol{\theta} = \begin{bmatrix} a_1 \\ a_2 \\ b_1 \\ b_2 \end{bmatrix}, \quad \boldsymbol{\phi}[k] = \begin{bmatrix} -\dot{\theta}[k-1] \\ -\dot{\theta}[k-2] \\ v[k-2] \\ v[k-3] \end{bmatrix} \quad (7)$$

For the simplified second-order model, the recursion reduces to:

$$\dot{\theta}[k] = -a_1\dot{\theta}[k-1] + b_1v[k-2] \quad (8)$$

with corresponding parameter and regression vectors

$$\boldsymbol{\theta}_{\text{simp}} = \begin{bmatrix} a_1 \\ b_1 \end{bmatrix}, \quad \boldsymbol{\phi}_{\text{simp}}[k] = \begin{bmatrix} -\dot{\theta}[k-1] \\ v[k-2] \end{bmatrix}. \quad (9)$$

The least squares criterion minimizes the sum of squared prediction errors:

$$V_N(\boldsymbol{\theta}) = \sum_{k=1}^N \frac{1}{2} [\dot{\theta}[k] - \boldsymbol{\phi}^T[k]\boldsymbol{\theta}]^2 \quad (10)$$

### 1.2.2 Matrix formulation

For  $N$  measurements with maximum delay of 3 samples, the output vector (starting from  $k = 4$ ) and regression matrix are:

$$\mathbf{y} = \begin{bmatrix} \dot{\theta}[4] \\ \dot{\theta}[5] \\ \vdots \\ \dot{\theta}[N] \end{bmatrix}, \quad \Phi = \begin{bmatrix} -\dot{\theta}[3] & -\dot{\theta}[2] & v[2] & v[1] \\ -\dot{\theta}[4] & -\dot{\theta}[3] & v[3] & v[2] \\ \vdots & \vdots & \vdots & \vdots \\ -\dot{\theta}[N-1] & -\dot{\theta}[N-2] & v[N-2] & v[N-3] \end{bmatrix} \quad (11)$$

The least squares solution is:

$$\hat{\boldsymbol{\theta}}_N^{LS} = [\Phi^T \Phi]^{-1} \Phi^T \mathbf{y} = \Phi^+ \mathbf{y} \quad (12)$$

where  $\Phi^+$  is the Moore-Penrose pseudo-inverse.

## 2 Identification of the cart by exciting the motors while the cart is on the ground

### 2.1 Excitation signal selection (2a)

#### 2.1.1 Motivation for excitation signal

A periodic piecewise-constant multilevel excitation signal is used for system identification. Such signals are persistently exciting, cover multiple operating points, and generate sufficiently rich frequency content for reliable parameter estimation. The input alternates between four nonzero voltage levels (+6 V +5 V -6 V -5 V) and zero-voltage plateaus so that both transient and steady-state responses are observable. The use of multiple amplitudes facilitates verification of approximate linearity, guarantees input levels high enough to overcome static friction and excites the dominant modes of the motor-cart dynamics.

#### 2.1.2 Excitation signal design and implementation

The excitation has a period of 14 s with the sequence

+6 V (0–2 s), 0 V (2–3 s), +5 V (3–5 s), 0 V (5–7 s), -6 V (7–9 s), 0 V (9–10 s), -5 V (10–12 s), 0 V (12–14 s).

The signal is sampled at  $f_s = 100$  Hz and applied for two full periods (total duration 28 s), as shown in Figure 1. Repetition of the cycle increases the effective SNR via averaging over periods, yields a data record of sufficient length for parameter estimation and permits an empirical assessment of measurement noise by comparing responses to identical input segments.

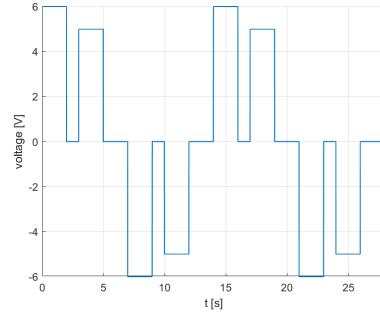


Figure 1: Applied excitation signal over 2 complete periods (28 seconds)

## 2.2 Parameter estimation and model validation for both motors (2b)

Using linear least squares estimation, we identified both the third-order model ( $H(z) = \frac{b_1 z + b_2}{z^3 + a_1 z^2 + a_2 z}$ ) and the simplified second-order model ( $H(z) = \frac{b_1}{z^2 + a_1 z}$ ) for motors A and B.

The simplified-model parameter vectors, identified on the unfiltered data set, are

$$\hat{\theta}_A^{\text{simp}} = \begin{bmatrix} -0.6819 \\ 0.6309 \end{bmatrix}, \quad \hat{\theta}_B^{\text{simp}} = \begin{bmatrix} -0.6806 \\ 0.6488 \end{bmatrix}. \quad (13)$$

The resulting discrete-time transfer functions are

$$\hat{H}_A^{\text{simp}}(z) = \frac{0.6309}{z^2 - 0.6819z}, \quad \hat{H}_B^{\text{simp}}(z) = \frac{0.6488}{z^2 - 0.6806z}. \quad (14)$$

The full model estimated parameter vectors are

$$\hat{\theta}_A = \begin{bmatrix} -0.5671 \\ -0.0970 \\ 0.0077 \\ 0.6589 \end{bmatrix}, \quad \hat{\theta}_B = \begin{bmatrix} -0.5762 \\ -0.0881 \\ 0.0081 \\ 0.6741 \end{bmatrix}. \quad (15)$$

Including the microOS delay yields the corresponding third-order models:

$$\hat{H}_A(z) = z^{-1} \hat{H}_{0,A}(z) = \frac{0.007672z + 0.6589}{z^3 - 0.5671z^2 - 0.09698z}, \quad (16)$$

$$\hat{H}_B(z) = z^{-1} \hat{H}_{0,B}(z) = \frac{0.008073z + 0.6741}{z^3 - 0.5762z^2 - 0.08815z}. \quad (17)$$

### 2.2.1 Measured vs simulated response and model error

Figures 2 and 3 collect the validation plots for both motors. The top row of each figure shows the third-order model, whereas the bottom row shows the simplified model (left column: motor A, right column: motor B). All panels are zoomed on the transient windows to expose the small mismatches between the measured data and the simulations.

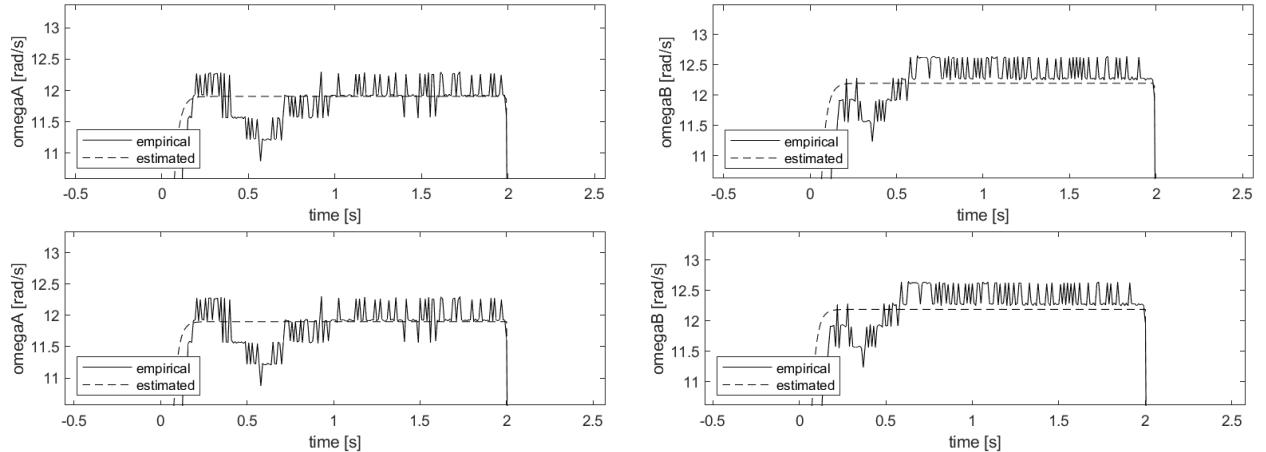


Figure 2: Measured (solid black) versus simulated (dashed black) responses for both motors. Top row: third-order model; bottom row: simplified model.

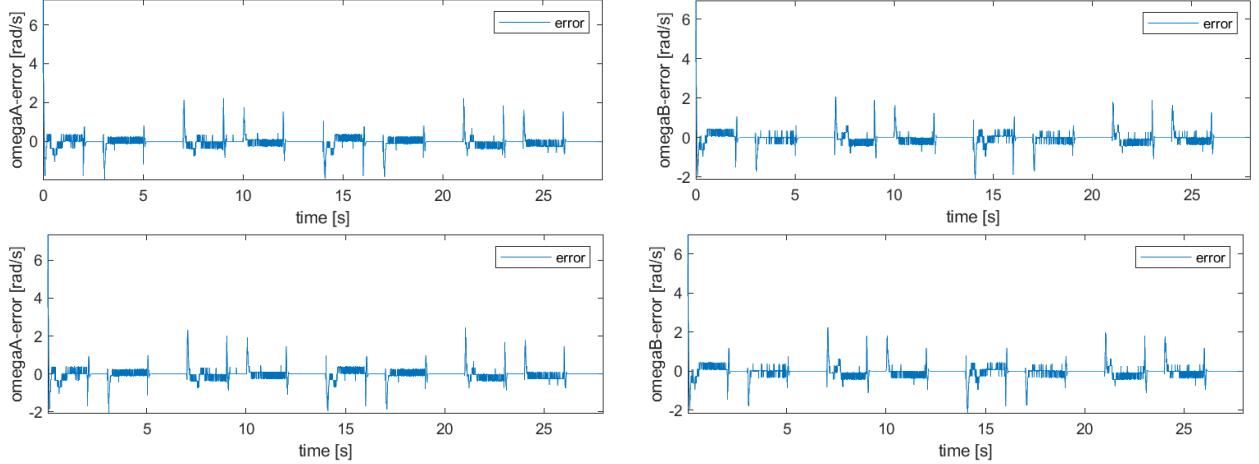


Figure 3: Prediction error (measurement minus simulation) for both motors. Top row: third-order model; bottom row: simplified model.

RMS error (motor A, full model): 0.3611 rad/s

RMS error (motor B, full model): 0.3644 rad/s

RMS error (motor A, simplified model): 0.3723 rad/s

RMS error (motor B, simplified model): 0.3758 rad/s

The two motors therefore behave almost identically: the RMS prediction errors differ by less than 0.004 rad/s (below 1%) for both model structures, and the identified parameters vary by less than 2% across motors. The model-measurement errors in Figure 3 overlap as well, showing that the remaining discrepancies stem from shared effects such as sensor resolution/quantization limitations. Because the validation metrics and visual inspections point to equivalent dynamics, subsequent sections display detailed plots for motor A only; whenever performance numbers are quoted, both motors are still reported to document their similarity.

### 2.2.2 Transient and steady-state characteristics

**Transient behavior:** Both models reproduce the measured step-to-step evolution of the angular velocity. The third-order model responds a fraction faster after each voltage transition because the additional pole associated with armature inductance gives a steeper initial slope. The velocity measurements are quantized, so visible level-to-level toggling appears that the continuous models do not mirror. That discrepancy excites the lightly damped high-frequency mode of the third-order fit and inflates its residuals. With the updated identification, the third-order model attains RMS prediction errors of 0.3611 rad/s for motor A and 0.3644 rad/s for motor B -only marginally below the simplified model's 0.3723 rad/s (motor A) and 0.3758 rad/s (motor B) while requiring more parameters.

**Steady-state behavior:** Both simulations settle to the measured steady-state angular velocities within the sensor resolution. The apparent steady-state ripple arises from the same quantization and is absent in the simulated trajectories, yet the mean levels match within negligible error.

Armature inductance therefore has no discernible influence on steady-state gain, and **the simplified second-order model is selected for subsequent analysis.**

## 2.3 Verification of data filtering to improve identification (2c)

### 2.3.1 Motivation for filtering

Pre-filtering measured signals before parameter estimation is a standard technique to suppress high-frequency measurement noise that corrupts the least-squares regression. In principle, applying a low-pass filter to both

input and output preserves the frequency-domain ratio  $\hat{H}(e^{j\omega})$  while attenuating noise components above the system bandwidth, thereby reducing parameter variance and improving model fidelity.

Filtering both the input voltage ( $v[k]$ ) and output angular velocity ( $\omega_A[k]$ ) with the same linear-phase filter maintains the correct phase relationship and gain ratio at all frequencies below the cutoff.

### 2.3.2 Filter design

To investigate whether pre-filtering improves identification accuracy, an eighth-order Butterworth low-pass filter was selected for its maximally flat passband characteristic. The filter was applied using zero-phase filtering (`filtfilt`) to eliminate phase distortion. We systematically swept the cutoff frequency from 20 Hz to 50 Hz (Nyquist limit) to assess the trade-off between noise suppression and transient preservation.

We found that every cutoff frequency attenuates useful transient content and phase information across the entire passband, causing the filtered signals to lose the sharp voltage-to-velocity steps that carry the identification information. This systematic loss of informative dynamics outweighs any noise suppression benefit, making filtering counterproductive in this case as seen in the figures below.

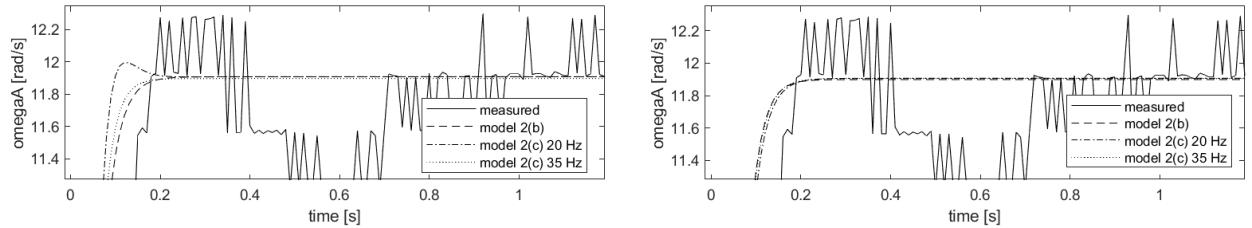


Figure 4: Left: full third-order model; Right: the same comparison for the simplified model

### 2.3.3 Re-identification with filtered data

After applying the Butterworth filter to both input and output signals using `filtfilt`, the linear least-squares estimation was repeated. For motor A, the estimated parameter vectors with 20 Hz and 35 Hz cutoffs are

$$\hat{\theta}_A^{\text{filt},20} = \begin{bmatrix} -1.3385 \\ 0.4738 \\ 0.3416 \\ -0.0730 \end{bmatrix}, \quad \hat{\theta}_A^{\text{filt},35} = \begin{bmatrix} -0.7993 \\ 0.0923 \\ -0.0127 \\ 0.5937 \end{bmatrix}, \quad (18)$$

These coefficients yield discrete-time transfer functions of the form

$$\hat{H}_{\text{filt}}(z) = \frac{\hat{b}_1 z + \hat{b}_2}{z^3 + \hat{a}_1 z^2 + \hat{a}_2 z}. \quad (19)$$

Re-identifying the simplified second-order model on the filtered data gives (motor A shown with both cutoffs)

$$\hat{\theta}_A^{\text{simp,filt},20} = \begin{bmatrix} -0.7007 \\ 0.5940 \end{bmatrix}, \quad \hat{\theta}_A^{\text{simp,filt},35} = \begin{bmatrix} -0.6873 \\ 0.6204 \end{bmatrix}, \quad (20)$$

with corresponding transfer functions

$$\hat{H}_A^{\text{simp,filt},20}(z) = \frac{0.5940}{z^2 - 0.7007z}, \quad \hat{H}_A^{\text{simp,filt},35}(z) = \frac{0.6204}{z^2 - 0.6873z}. \quad (21)$$

The explicit transfer functions are:

$$\hat{H}_A^{\text{filt},20}(z) = z^{-1} \hat{H}_{0,A}^{\text{filt},20}(z) = \frac{0.3416z - 0.07297}{z^3 - 1.338z^2 + 0.4738z}, \quad (22)$$

$$\hat{H}_A^{\text{filt},35}(z) = z^{-1} \hat{H}_{0,A}^{\text{filt},35}(z) = \frac{-0.01271z + 0.5937}{z^3 - 0.7993z^2 + 0.09234z}. \quad (23)$$

#### 2.3.4 Comparison and model selection

Direct comparison of RMS errors reveals that filtering does *not* improve identification accuracy for motor A. The unfiltered third-order model achieves an RMS error of 0.3611 rad/s, whereas the filtered third-order fits rise to 0.4255 rad/s (20 Hz) and 0.3865 rad/s (35 Hz). For the simplified motor A model, filtering yields 0.3682 rad/s (20 Hz) and 0.3706 rad/s (35 Hz), only a marginal change relative to the unfiltered 0.3723 rad/s.

The lack of improvement suggests that the dominant error source is *not* high-frequency additive noise but rather model mismatch (e.g., unmodeled nonlinearities, quantization effects, or the lightly damped pole artifact from encoder resolution). Because the simplified second-order model already achieves 0.3723 rad/s for motor A with two parameters and outperforms the filtered third-order alternatives, we conclude that the model structure, not the noise level, is the limiting factor.

Considering the simpler structure (reduced overfitting risk) and the better RMS error performance, we select the **unfiltered simplified second-order model** for all subsequent analysis.

The filtering exercise confirms that the measurement noise is secondary relative to structural mismatch; a model on unfiltered data best captures the true system dynamics under the given experimental conditions.

## References

- [1] Emama-Naeini Franklin, Powell. *Feedback Control of Dynamic Systems*. Pearson Education Limited, 2020.
- [2] Jan Swevers. *C4. Time domain system identification*. KU Leuven, 2022.

## A Simplified second-order model (armature inductance neglected)

The simplified discrete-time transfer function neglecting armature inductance  $L_a$  is:

$$H(z) = \frac{b_1}{z^2 + a_1 z} \quad (24)$$

This model has two parameters ( $a_1, b_1$ ) compared to four in the full model.

The recursion expression for the simplified model is given in the main text by (8), with the associated regression vector and least-squares criterion (10). For completeness, we retain the matrix formulation below.

### Matrix formulation

For  $N$  measurements with maximum delay of 2 samples (starting from  $k = 3$ ):

$$\mathbf{y} = \begin{bmatrix} \dot{\theta}[3] \\ \dot{\theta}[4] \\ \vdots \\ \dot{\theta}[N] \end{bmatrix}, \quad \Phi = \begin{bmatrix} -\dot{\theta}[2] & v[1] \\ -\dot{\theta}[3] & v[2] \\ \vdots & \vdots \\ -\dot{\theta}[N-1] & v[N-2] \end{bmatrix} \quad (25)$$

The least squares solution is:

$$\hat{\theta}_N^{LS} = [\Phi^T \Phi]^{-1} \Phi^T \mathbf{y} = \Phi^+ \mathbf{y} \quad (26)$$

## B Figures

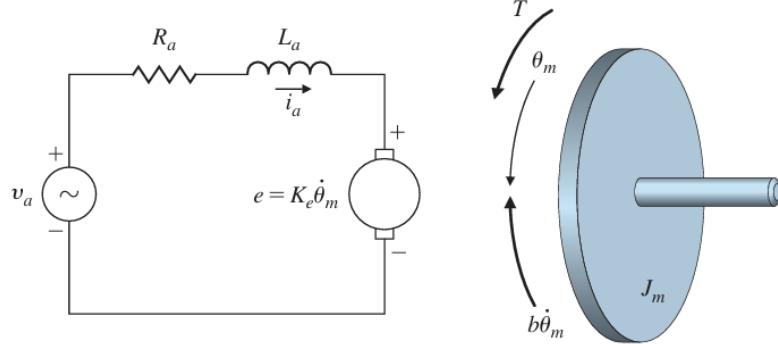


Figure 5: Model of the DC motor [1]