

# ME 6402 – Lecture 27

## FINAL EXAM REVIEW

April 17 2025

Overview:

- Backstepping
- Feedback Linearization
- Normal Form and Zero Dynamics
- Control Lyapunov functions
- Control Barrier functions

Additional Reading:

- Khalil Chapter 14.3 (Backstepping)
- Khalil Chapter 13 (Feedback Linearization of SISO Systems)
- Sastry Chapter 9.3 (Feedback Linearization of MIMO Systems)
- E. Sontag, 1983 (Control Lyapunov Functions)
- A. Ames et al. 2019 (Control Barrier Functions)

## Backstepping (Lecture 13)

Backstepping is a specific control design technique for a certain class of systems. The basic idea of backstepping is that we can stabilize the system

$$\begin{aligned}\dot{x}_1 &= F(x_1) + G(x_1)x_2 \\ \dot{x}_2 &= u\end{aligned}$$

through the coordinate shift  $z = x_2 - k(x_1)$ , where  $k(x_1)$  is a function that would result in stable dynamics for the  $x_1$  subsystem. This effectively shifts the equilibrium point for our  $x_1$  system and allows us to render it stable through the coordinate shift. Then, we can stabilize the remaining  $\dot{x}_2$  dynamics by choosing  $u$  such that  $\dot{z}$  is also a stable subsystem.

An example of backstepping is the following (Example 14.8 from Khalil):

**Example 1:** Consider the system

$$\begin{aligned}\dot{x}_1 &= x_1^2 - x_1^3 + x_2 \\ \dot{x}_2 &= u\end{aligned}$$

The  $\dot{x}_1$  dynamics can be stabilized through the “control law”

$$x_2 = k(x_1) = -x_1^2 - x_1$$

Applying this control would yield the system:

$$\dot{x}_1 = -x_1^3 - x_1$$

which is stable. Notably, the  $-x_1^3$  term provides an additional damping stabilization term compared to only having  $\dot{x}_1 = -x_1$ .

Next, we will shift our system to effectively *enforce* this control law when we drive the system to zero. This is done through the coordinate shift  $z = x_2 - k(x_1)$ , which results in the shifted system:

$$\begin{aligned}\dot{x}_1 &= x_1^2 - x_1^3 + (z + k(x_1)) = -x_1^3 - x_1 + z \\ \dot{z} &= u + \dot{k}\end{aligned}$$

where  $\dot{k} = (-2x_1 - 1)\dot{x}_1 = -(2x_1 + 1)(-x_1^3 - x_1 + z)$ .

The main idea of backstepping is that we can then prove stability of the  $x_1$  dynamics using the Lyapunov function  $V(x_1) = \frac{1}{2}x_1^2$  and then construct an augmented Lyapunov function to construct a control law that would also stabilize the  $x_2$  dynamics:

$$V_+ = V(x_1) + \frac{1}{2}z^2$$

This results in the control law:

$$u = \dot{k} - \frac{\partial V}{\partial x_1}G(x_1) - Kz$$

For our example,  $\frac{\partial V}{\partial x_1} = x_1$  and  $G(x_1) = 1$ . Taking  $K = 1$  for simplicity, we get the final control law:

$$\begin{aligned}u &= \dot{k} - \frac{\partial V}{\partial x_1}G(x_1) - z \\ &= -(2x_1 + 1)(-x_1^3 - x_1 + z) - x_1 - z\end{aligned}$$

If instead, we had had a system where  $\dot{x}_2$  was control affine, the approach would be mostly the same. A specific example is as follows.

### Example 2:

$$\begin{aligned}\dot{x}_1 &= x_1^2 - x_1^3 + x_2 \\ \dot{x}_2 &= x_2 + 2u\end{aligned}$$

We can convert the form to our previous form using the same approach as with input-output linearization:

$$u = \frac{1}{2}(-x_2 + v)$$

This results in the system:

$$\begin{aligned}\dot{x}_1 &= x_1^2 - x_1^3 + x_2 \\ \dot{x}_2 &= v\end{aligned}$$

Thus, following the same procedure as in Example 1, we would obtain:

$$v = -(2x_1 + 1)(-x_1^3 - x_1 + z) - x_1 - z$$

Plugging this back into our control law, we get:

$$u = \frac{1}{2} \left( -x_2 - (2x_1 + 1)(-x_1^3 - x_1 + z) - x_1 - z \right)$$

**Note:** While it's possible to recursively perform backstepping across multiple states, this typically results in very ugly and complex control laws and thus will likely not appear on the final exam.

### Feedback Linearization (Lectures 16-19)

#### Relative Degree

**Definition: Relative Degree for SISO.** A SISO system has relative degree  $r$  if, in a neighborhood of the equilibrium:

$$\begin{aligned} L_g L_f^{i-1} h(x) &= 0, \quad i = 1, 2, \dots, r-1 \\ L_g L_f^{r-1} h(x) &\neq 0 \end{aligned}$$

Informally, this is the same as saying that “A SISO system has relative degree  $r$  if the input does not appear until the  $r$ -th derivative of the output  $h(x)$ ”.

**Definition: Relative Degree for MIMO.** A MIMO system has relative degree  $r_i$  for each output  $h_i(x)$  if the  $i$ -th output needs to be differentiated  $r_i$  times before *some* input appears.

**Definition: Vector Relative Degree for MIMO.** A MIMO system has vector relative degree  $r = \{r_1, \dots, r_m\}$  if the matrix  $A(x)$  is nonsingular:

$$A(x) = \begin{bmatrix} L_{g_1} L_f^{r_1-1} h_1(x) & \cdots & L_{g_1} L_f^{r_1-1} h_1(x) \\ \vdots & \ddots & \vdots \\ L_{g_m} L_f^{r_m-1} h_m(x) & \cdots & L_{g_m} L_f^{r_m-1} h_m(x) \end{bmatrix}$$

**Example 3:** Consider the system:

$$\begin{aligned} \dot{x}_1 &= x_1 \\ \dot{x}_2 &= x_2 + u \\ y &= x_1 \end{aligned}$$

The system does not have a well-defined relative degree because  $\dot{y} = \dot{x}_1 = x_1 = y$ . Thus the input  $u$  will *never* appear.

**Example 4:** Consider the system:

$$\dot{x}_1 = -x_1 + \frac{2 + x_3^2}{1 + x_3^2}u$$

$$\dot{x}_2 = x_3$$

$$\dot{x}_3 = x_1x_3 + u$$

$$y = x_2$$

The system has relative degree 2 because:

$$\dot{y} = \dot{x}_2 = x_3$$

$$\ddot{y} = \dot{x}_3 = x_1x_3 + u$$

Notably, the relative degree is well-defined for all  $x \in \mathbb{R}^3$ .

**Example 5:** Consider the system (it is the controlled van der Pol equation):

$$\dot{x}_1 = x_2$$

$$\dot{x}_2 = -x_1 + \varepsilon(1 - x_1^2)x_2 + u$$

$$y = x_2$$

The system has relative degree 1 because  $\dot{y} = \dot{x}_2 = -x_1 + \varepsilon(1 - x_1^2)x_2 + u$ . This is also well-defined for all  $x \in \mathbb{R}^2$ .

**Example 6:** Consider the MIMO system:

$$\dot{x}_1 = \cos(x_3)u_1$$

$$\dot{x}_2 = \sin(x_3)u_1$$

$$\dot{x}_3 = u_2$$

$$y_1 = x_1$$

$$y_2 = x_2$$

The system has relative degree  $r_1 = r_2 = 1$  because

$$\dot{y}_1 = \dot{x}_1 = \cos(x_3)u_1$$

$$\dot{y}_2 = \dot{x}_2 = \sin(x_3)u_1$$

To check if the system has a valid vector relative degree, we need to check if the matrix  $A(x)$  is nonsingular. Explicitly, this matrix is:

$$A := \begin{bmatrix} \cos(x_3) & 0 \\ \sin(x_3) & 0 \end{bmatrix}$$

This matrix is NOT nonsingular, so it does not have a valid vector relative degree. This means that we could not perform feedback

linearization on this system. Instead, we would need to perform dynamic extension:

$$\begin{aligned}\dot{x}_1 &= x_4 \cos(x_3) \\ \dot{x}_2 &= x_4 \sin(x_3) \\ \dot{x}_3 &= u_2 \\ \dot{x}_4 &= u_1 \\ y_1 &= x_1 \\ y_2 &= x_2\end{aligned}$$

This would result in each output having relative degree 2, with the derivatives being:

$$\begin{aligned}\dot{y}_1 &= \dot{x}_1 = x_4 \cos(x_3) \\ \ddot{y}_1 &= u_1 \cos(x_3) - x_4 \sin(x_3) u_2 \\ \dot{y}_2 &= \dot{x}_2 = x_4 \sin(x_3) \\ \ddot{y}_2 &= u_1 \sin(x_3) + x_4 \cos(x_3) u_2\end{aligned}$$

Thus, the  $A$  matrix is now:

$$A := \begin{bmatrix} \cos(x_3) & -x_4 \sin(x_3) \\ \sin(x_3) & x_4 \cos(x_3) \end{bmatrix}$$

This matrix is only singular when  $x_4 = 0$ , so for any state such that  $x_4 \neq 0$ , the system has a valid vector relative degree  $r = \{2, 2\}$ .

### *Input-Output Linearization*

If a system has a well-defined relative degree (or a valid vector relative degree for MIMO systems) then it is input-output linearizable. Explicitly, this feedback linearizing control law is:

$$u = \frac{1}{L_g L_f^{r-1} h(x)} \left( -L_f^r h(x) + v \right)$$

or

$$u = A^{-1}(-B + v)$$

You can always think of this as the latter if you rearrange the system to be in the form:

$$y^{(r)} = B + Au$$

By selecting the auxiliary control law

$$v = -k_1 y - k_2 \dot{y} - \dots - k_r y^{(r-1)} \quad (1)$$

we can transform our input-output system to be:

$$\begin{bmatrix} \dot{y} \\ \ddot{y} \\ \vdots \\ y^{(r)} \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ 0 & 0 & 0 & \cdots & 1 \\ -k_1 & -k_2 & -k_3 & \cdots & -k_r \end{bmatrix} \begin{bmatrix} y \\ \dot{y} \\ \ddot{y} \\ \vdots \\ y^{(r-1)} \end{bmatrix}$$

### Full-State Feedback Linearization

If  $r = n$ , then there exists a diffeomorphism that transforms the system into the linear system

$$\dot{\eta} = A\eta$$

with the transformation being:

$$x \mapsto \begin{bmatrix} \eta_1 \\ \eta_2 \\ \vdots \\ \eta_n \end{bmatrix} = \begin{bmatrix} h(x) \\ L_f h(x) \\ \vdots \\ L_f^{n-1} h(x) \end{bmatrix} = \begin{bmatrix} y \\ \dot{y} \\ \vdots \\ y^{(n-1)} \end{bmatrix}$$

We have a theorem to verify when a system is provably full-state feedback linearizable. This theorem also provides us with tools to know how to select the output  $h(x)$  such that the system is full-state feedback linearizable.

**Theorem: Full-State Feedback Linearizable.** *The system  $\dot{x} = f(x) + g(x)u$  is full-state feedback linearizable around  $x_0$  if and only if the following two conditions hold:*

- C1)  $\begin{bmatrix} g(x_0) & \text{ad}_f g(x_0) & \cdots & \text{ad}_f^{n-1} g(x_0) \end{bmatrix}$  has rank  $n$ .
- C2) The distribution  $\Delta(x) = \text{span}\{g(x), \text{ad}_f g(x), \dots, \text{ad}_f^{n-2} g(x)\}$  is involutive in a neighborhood of  $x_0$ .

Importantly, by the Frobenius theorem, a nonsingular distribution is involutive if and only if it is completely integrable, which gives us the condition that there must exist a function  $h(x)$  such that:

$$\frac{\partial h}{\partial x} f_j = 0$$

where  $f_j$  represents each element in the span of the associated distribution  $\Delta$ .

**Example 7:** Consider the system

$$\begin{aligned}\dot{x}_1 &= x_2 \\ \dot{x}_2 &= \sin(x_1) + u\end{aligned}$$

First, to calculate the adjoint elements:

$$\begin{aligned}g(x) &= \begin{bmatrix} 0 \\ 1 \end{bmatrix} \\ \text{ad}_f g(x) &= [f, g] = \frac{\partial g}{\partial x} f(x) - \frac{\partial f}{\partial x} g(x) = - \begin{bmatrix} 0 & 1 \\ \cos(x_1) & 0 \end{bmatrix} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = \begin{bmatrix} 1 \\ 0 \end{bmatrix}\end{aligned}$$

Thus, the matrix of condition 1 is:

$$\begin{bmatrix} 0 & 1 \\ 1 & 0 \end{bmatrix}$$

which is full rank.

Second, we need to find an output  $h(x)$  such that

$$\frac{\partial h}{\partial x} g(x) = 0 \implies \frac{\partial h}{\partial x} \begin{bmatrix} 0 \\ 1 \end{bmatrix} = 0$$

This condition is satisfied for  $h(x) = x_1$ .

We can double check this by computing the relative degree associated with  $h(x) = x_1$ :

$$\begin{aligned}\dot{y} &= \dot{x}_1 = x_2 \\ \ddot{y} &= \dot{x}_2 = \sin(x_1) + u\end{aligned}$$

*Normal Form*

If the system is *not* full-state feedback linearizable, the system will have *zero dynamics*. The zero dynamics are those that remain when the feedback linearizing control law is applied (with  $v = 0$ ) and the outputs are consequently driven to zero.

**Example 8:** Consider the system

$$\begin{aligned}\dot{x}_1 &= x_2 \\ \dot{x}_2 &= -x_1 + x_3^2 + u \\ \dot{x}_3 &= -x_3 + x_1 \\ y &= x_1\end{aligned}$$

First, we analyze the relative degree of the system:

$$\begin{aligned}\dot{y} &= \dot{x}_1 = x_2 \\ \ddot{y} &= \dot{x}_2 = -x_1 + x_3^2 + u\end{aligned}$$

Thus, the system has relative degree  $r = 2$ . The associated outputs are  $y = x_1$  and  $\dot{y} = x_2$ . The feedback control law is:

$$u = x_1 - x_3^2 + v$$

The zero dynamics can then be derived as:

$$\begin{aligned}\dot{x}_1 &= 0 \\ \dot{x}_2 &= 0 + x_3^2 + (0 - x_3^2 + 0) = 0 \\ \dot{x}_3 &= -x_3 + 0\end{aligned}$$

Thus, the zero dynamics are  $\dot{x}_3 = -x_3$ .

To derive the zero dynamic coordinate transformation, we must find the transformation  $z$  such that  $z$  is independent of the outputs, and  $\dot{z}$  does not contain  $u$ . This is done by ensuring that  $\nabla z \cdot g(x) = 0$ .

**Example 8 continued:** The zero dynamic coordinates associated with our previous example can be derived by finding  $z$  to satisfy:

$$\frac{\partial z}{\partial x} \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} = 0 \quad \implies \quad z = x_3$$

Thus, our transformation to normal form is:

$$T : x \mapsto \begin{bmatrix} \eta_1 \\ \eta_2 \\ z \end{bmatrix} = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

**Note:** This example is trivial since the normal form is already decomposed as exactly our system state...

The full normal form dynamics are:

$$\begin{bmatrix} \dot{\eta}_1 \\ \dot{\eta}_2 \\ \dot{z} \end{bmatrix} = \begin{bmatrix} \eta_2 \\ v \\ -z \end{bmatrix}$$

We can check whether this map is a diffeomorphism (with a smooth inverse) by if its Jacobian has full rank.

**Note:** You should check the Jacobian if a question asks you to “specify the region over which the transformation to Normal Form is valid”



**Example 8 continued:** The Jacobian of the transformation is:

$$DT = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

Since this is full rank, our transformation is a diffeomorphism for all  $x \in \mathbb{R}^3$ .

### Control Lyapunov functions (Lecture 20)

**Definition: Control Lyapunov Function.** A positive definite function  $V(x)$  is a (global) control Lyapunov function for the system  $\dot{x} = f(x) + g(x)u$  if  $\forall x \neq 0, \exists u$  such that:

$$\dot{V}(x) = \frac{\partial V}{\partial x} (f(x) + g(x)u) < 0$$

One approach is to use Sontag's formula which is a closed-form solution to our inequality condition:

$$u = \begin{cases} - \left( \left( \frac{\partial V}{\partial x} f \right) + \sqrt{\left( \left( \frac{\partial V}{\partial x} f \right)^2 + \left( \frac{\partial V}{\partial x} g \right)^4} \right) / \left( \frac{\partial V}{\partial x} g \right)} & \text{if } \frac{\partial V}{\partial x} g \neq 0 \\ 0 & \text{if } \frac{\partial V}{\partial x} g = 0 \end{cases}$$

The alternative approach is to use convex optimization to solve the problem:

$$\begin{aligned} u^* = \underset{\mu}{\text{minimize}} \quad & \|\mu\|^2 \\ \text{subject to} \quad & L_f V(x) + L_g V(x)\mu < 0 \end{aligned}$$

### Control Barrier functions (Lectures 23-25)

The summary of control Lyapunov functions compared to control barrier functions is:

$$\underbrace{\dot{V} \leq -\alpha(V(x))}_{\text{Stability}} \quad \text{versus} \quad \underbrace{\dot{h} \geq -\alpha(h(x))}_{\text{Safety}}$$

**Definition: Barrier Function.** A function  $h$  with  $\mathcal{C} = \{x \mid h(x) \geq 0\}$  is a barrier function for  $\dot{x} = f(x)$  if there exists a locally Lipschitz function  $\alpha : \mathbb{R} \rightarrow \mathbb{R}$  satisfying  $\alpha(0) = 0$  such that

$$\dot{h}(x) \geq -\alpha(h(x)), \quad \text{for all } x \in \mathbb{R}^n$$

**Definition: Control Barrier Function.** A function  $h$  with  $\mathcal{C} = \{x \mid h(x) \geq 0\}$  is a control barrier function for  $\dot{x} = f(x) + g(x)u$  if there exists a locally Lipschitz function  $\alpha : \mathbb{R} \rightarrow \mathbb{R}$  satisfying  $\alpha(0) = 0$  such that

$$\sup_{u \in \mathbb{R}^m} \dot{h}(x) \geq -\alpha(h(x)), \quad \text{for all } x \in \mathbb{R}^n$$

As with control Lyapunov functions, we can use either a closed-form expression or convex optimization to find an input that satisfies our inequality condition. The closed-form expression is:

$$u = \begin{cases} 0 & \text{if } L_f h + \alpha(h(x)) \geq 0 \\ \frac{-(L_f h + \alpha(h(x)))L_g h^T}{\|L_g h\|^2} & \text{otherwise} \end{cases}$$

The convex optimization approach can take many forms, but we discussed two main ones. The minimum effort control barrier function is:

$$\begin{aligned} u^* &= \underset{\mu}{\text{minimize}} \quad \|\mu\|^2 \\ \text{subject to} \quad & L_f h(x) + L_g h(x)\mu \geq -\alpha(h(x)) \end{aligned}$$

The minimally-invasive control barrier function is:

$$\begin{aligned} u^* &= \underset{\mu}{\text{minimize}} \quad \|\mu - k(x)\|^2 \\ \text{subject to} \quad & L_f h(x) + L_g h(x)\mu \geq -\alpha(h(x)) \end{aligned}$$

Lastly, if  $L_g h(x) \equiv 0$ , we will need to instead use a higher-order barrier function.

**Example 9:** Consider the system:

$$\begin{aligned} \dot{x}_1 &= x_2 \\ \dot{x}_2 &= x_3 \\ \dot{x}_3 &= -x_1^2 + u \end{aligned}$$

Synthesize a control barrier function to keep the state  $x_1$  below a threshold of 2.

This desired behavior can be encoded by the function  $h(x) = 2 - x_1 \geq 0$ . This is associated with the safe setlength

$$\mathcal{C} = \{x \in \mathbb{R}^3 \mid h(x) = 2 - x_1 \geq 0\}$$

Taking the derivative, we get:

$$\dot{h} = -\dot{x}_1 = -x_2 \quad \implies \quad L_f h = -x_2, \quad L_g h = 0$$

Thus, this is an invalid control barrier function because  $L_g h \equiv 0$ .

Instead, we will need to use a higher-order barrier function of the form:

$$\Psi(x) := \dot{h}(x) + \alpha(h(x))$$

which is associated with its own safe set

$$\mathcal{C}_1 = \{x \in \mathbb{R}^3 \mid \Psi(x) \geq 0\}$$

To check if this higher order barrier function is valid. While doing this, we will assume  $\alpha(s) = \gamma_1 s$  for simplicity.

$$\begin{aligned}\dot{\Psi} &= \ddot{h}(x) + \alpha'(h(x))\dot{h} \\ &= -\dot{x}_2 + \gamma_1(-x_2) \\ &= -x_3 - \gamma_1 x_2\end{aligned}$$

Since this is *still* not valid, we will need to take another higher-order derivative:

$$\Psi_2(x) = \dot{\Psi}(x) + \alpha_2(\Psi(x))$$

which is associated with the safe set

$$\mathcal{C}_2 = \{x \in \mathbb{R}^3 \mid \Psi_2(x) \geq 0\}$$

To check if this higher order barrier function is valid, we need to again check whether  $L_g \Psi_2 \neq 0$ . Again, we will assume  $\alpha_2(s) = \gamma_2 s$  for simplicity.

$$\begin{aligned}\ddot{\Psi}_2 &= \ddot{\Psi}(x) + \alpha'_2(\Psi(x))\dot{\Psi} \\ &= (-\dot{x}_3 - \gamma_1 \dot{x}_2) + \gamma_2(-x_3 - \gamma_1 x_2) \\ &= -(-x_1^2 + u) - \gamma_1 x_3 + \gamma_2(-x_3 - \gamma_1 x_2)\end{aligned}$$

Here,  $L_g \Psi_2 = -1$  which means that the higher-order barrier function is valid everywhere. Finally, we will enforce this higher-order barrier function by finding  $u$  such that:

$$L_f \Psi_2(x) + L_g \Psi_2(x)u \geq -\alpha_2(\Psi_2(x))$$