Problem 4

```
clear all; close all; clc
ri = [8900 7200 0]; % intial state vectors
vi = [1.5 6.1 0];
hi = cross(ri,vi); % specific angular momentum
dTheta = 80*(2*pi)/360; % change in theta
mu = 398600;
ro = sqrt(dot(ri,ri)); %inital postion magnitude
vo = sqrt(dot(vi,vi)); % inital velocity magnitude
vro = dot(ri,vi)/ro; % inital radial velocity
h = ro*sqrt(vo^2-vro^2); % angular momentum magnitude
% new radial distance magnitdue
r = h^2/mu^*(1+(h^2/(mu^*ro)-1)^*cos(dTheta)-h^*vro/mu^*sin(dTheta))^-1;
f = 1 - mu*r/h^2*(1-cos(dTheta)); % lagrange coefficents
q = r*ro/h*sin(dTheta);
fdot = mu/h*(1-cos(dTheta))/sin(dTheta)*(mu/h^2*(1-cos(dTheta))-1/
ro-1/r);
gdot = 1-mu*ro/h^2*(1-cos(dTheta));
r_new = f*ri + g*vi; % new r vector per lagrange defintion
v new = fdot*ri + qdot*vi;
h_new = cross(r_new,v_new);
theta_new = atand(r_new(2)/r_new(1))+180; % new theta, add 180
because answer would be negative
e = norm(cross(v_new,h_new)/mu - r_new/(norm(r_new))); % eccentricity
magnitude from vector definition
fprintf('Inital True Anamoly: %0.3f \n', true_an)
fprintf('New Position Vector: [%.f, %0.f, %0.f] km \n', r_new)
fprintf('New Velocity Vector: [%0.3f, %0.3f, %0.3f] km/s \n', v_new)
fprintf('Eccentricity: %0.3f \n', e)
Inital True Anamoly: 38.972
New Position Vector: [-6371, 11506, 0] km
New Velocity Vector: [-0.754, -5.465, 0.000] km/s
Eccentricity: 0.800
```

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