

Turbulent Non-Premixed Combustion


Combustion Summer School
2018

Prof. Dr.-Ing. Heinz Pitsch

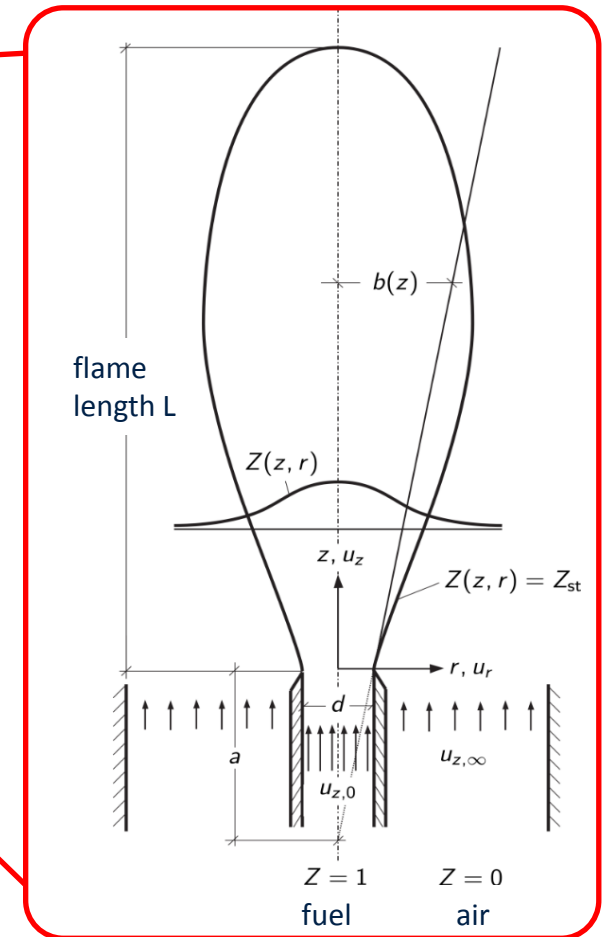


Course Overview

Part II: Turbulent Combustion

- Turbulence
- Turbulent Premixed Combustion
- Turbulent Non-Premixed Combustion 
- Laminar Jet Diffusion Flames
- Turbulent Jet Diffusion Flames
- Turbulent Combustion Modeling
- Applications

Laminar Jet Diffusion Flames

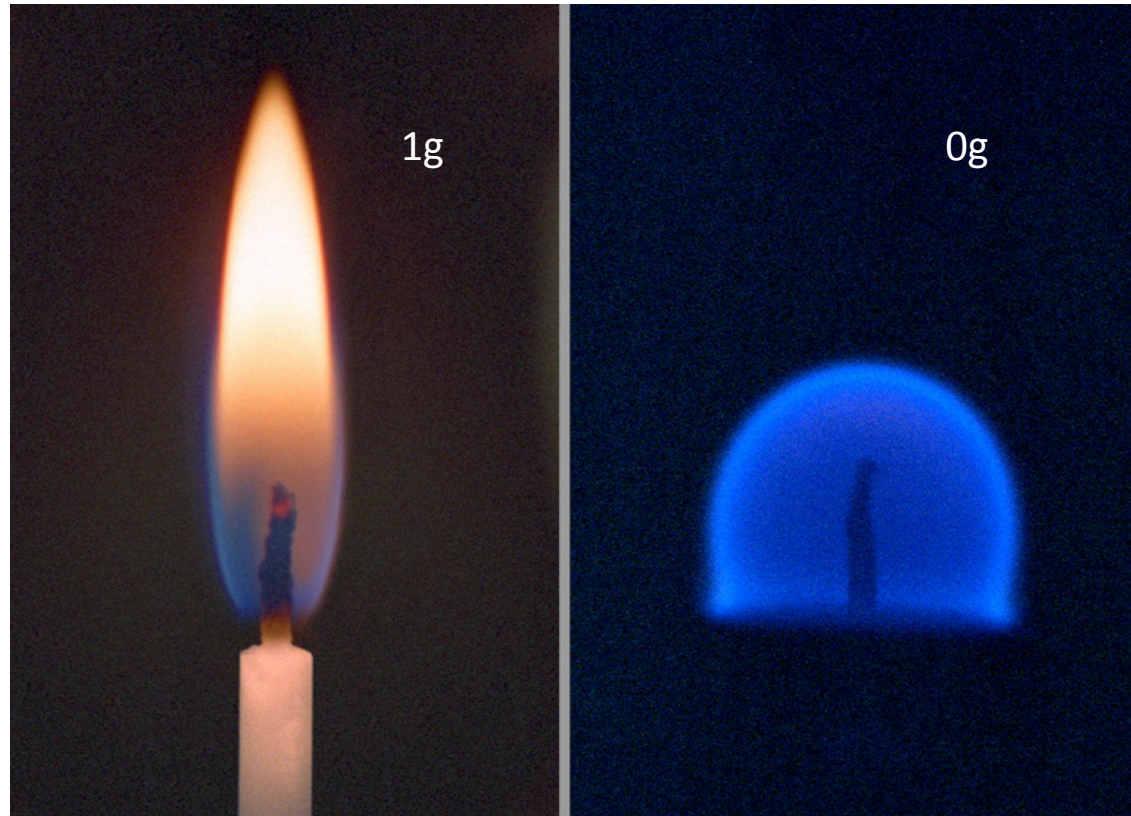


Laminar Jet Diffusion Flame

- Fuel enters into the combustion chamber as a round jet
- Forming mixture is ignited
- Example: Flame of a gas lighter
 - Only stable if dimensions are small
 - Dimensions too large: flickering due to influence of gravity
 - Increasing the jet momentum → Reduction of the relative importance of gravity (buoyancy) in favor of momentum forces
 - At high velocities, hydrodynamic instabilities gain increasing importance: laminar-turbulent transition



Laminar Diffusion Flame: Influence of Gravity



Laminar Jet Diffusion Flame (Governing Equations)

- Starting point: Conservation equations for stationary axisymmetric boundary layer flow without buoyancy
- Continuity:

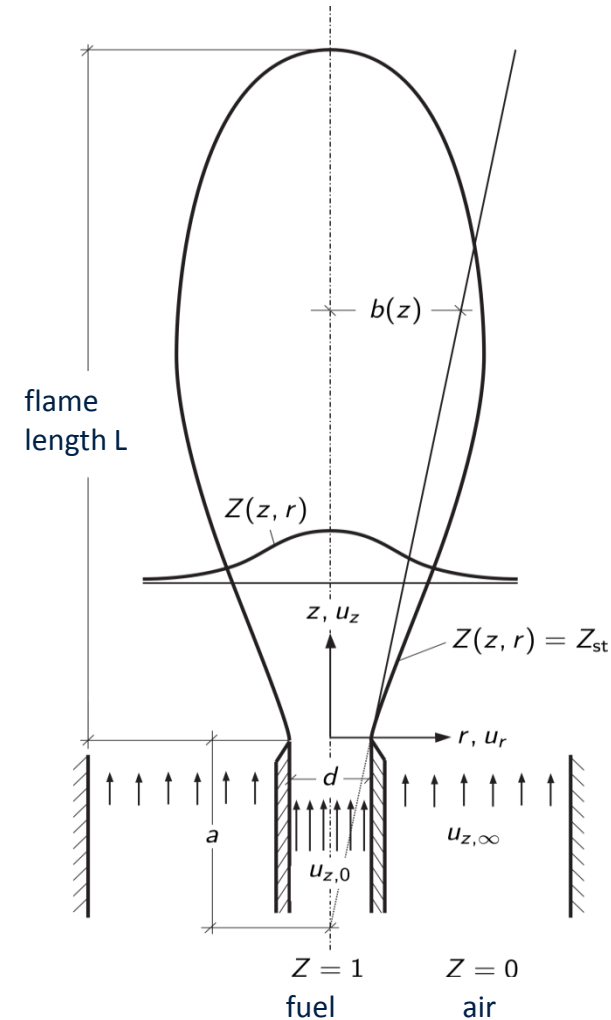
$$\frac{\partial(\rho u_z r)}{\partial z} + \frac{\partial(\rho u_r r)}{\partial r} = 0$$

- Momentum equation in z-direction

$$\rho u_z r \frac{\partial u_z}{\partial z} + \rho u_r r \frac{\partial u_z}{\partial r} = -r \frac{\partial p}{\partial z} + \frac{\partial}{\partial r} \left(\mu r \frac{\partial u_z}{\partial r} \right)$$

- Mixture fraction

$$\rho u_z r \frac{\partial Z}{\partial z} + \rho u_r r \frac{\partial Z}{\partial r} = \frac{\partial}{\partial r} \left(\frac{\mu}{Sc} r \frac{\partial Z}{\partial r} \right)$$

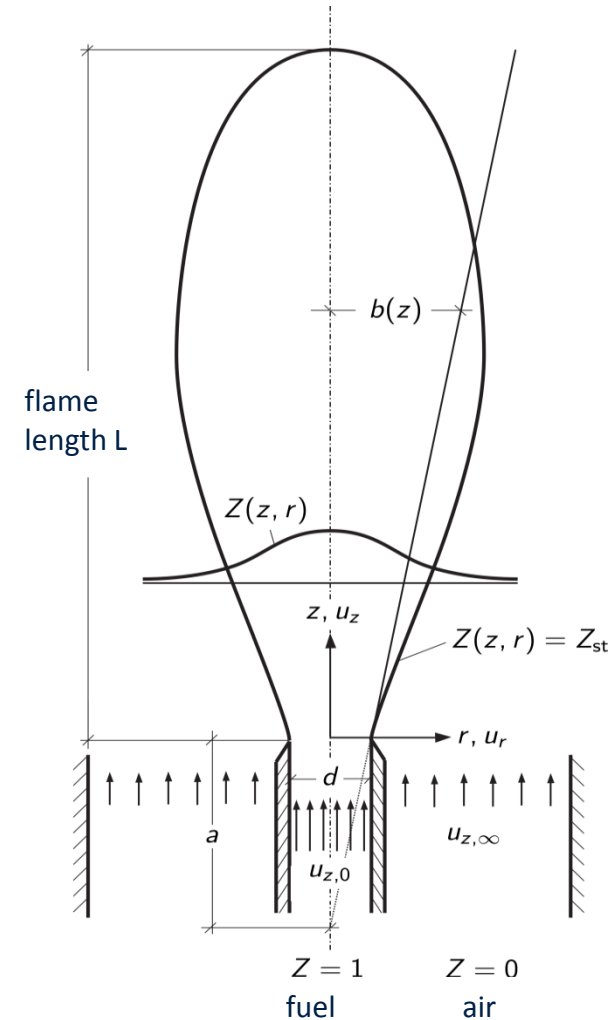


- Boundary layer flow:

- Incompressible round jet

- Similarity solution

- 7



Laminar Jet Diffusion Flame (Similarity Coordinates)

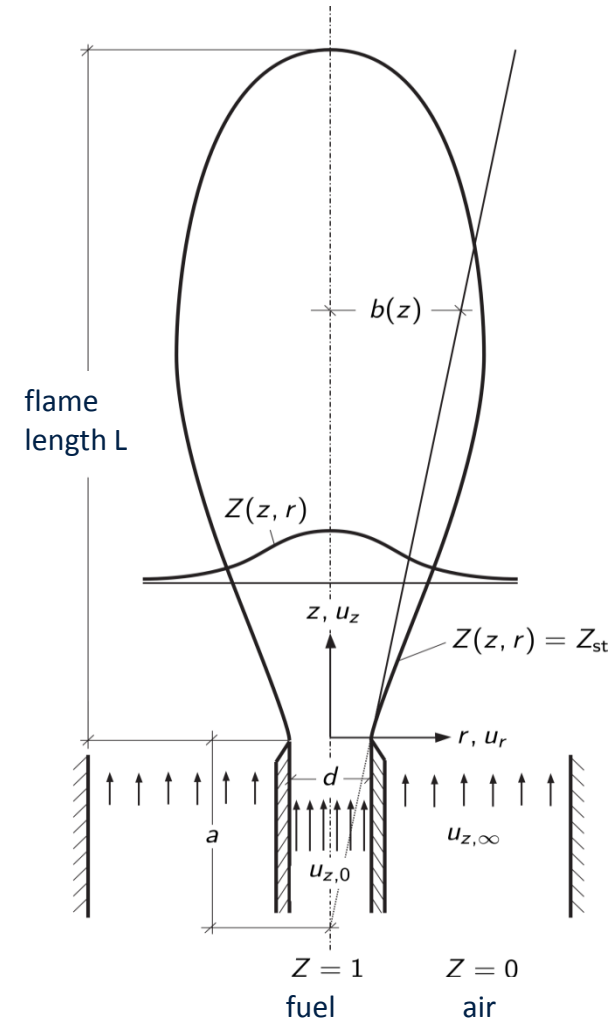
- If density not constant
→ Transformation

$$\zeta = z + a, \quad \eta = \frac{\sqrt{2 \int_0^r \frac{\rho}{\rho_\infty} r dr}}{\zeta}$$

- a : Distance of the virtual origin of the jet from the nozzle exit
- For $\rho = \text{const.}$ und $a \rightarrow 0$

$$\zeta = z, \quad \eta = \frac{r}{z}$$

- Implies linear spreading of the round jet



Laminar Jet Diffusion Flame (Stream Function)

- Introduction of a **stream function** Ψ

$$\rho u_z r = \frac{\partial \Psi}{\partial r}, \quad \rho u_r r = -\frac{\partial \Psi}{\partial z}$$

→ Continuity equation identically satisfied

- Applying the **transformation rules**

$$\zeta = z + a, \quad \eta = \frac{\sqrt{2 \int_0^r \frac{\rho}{\rho_\infty} r dr}}{\zeta} \rightarrow \frac{\partial}{\partial z} = \frac{\partial}{\partial \zeta} + \frac{\partial \eta}{\partial z} \frac{\partial}{\partial \eta}, \quad \frac{\partial}{\partial r} = \frac{\partial \eta}{\partial r} \frac{\partial}{\partial \eta}$$

to the **convective terms** in the momentum and mixture fraction equations yields

$$\rho u_z r \frac{\partial}{\partial z} + \rho u_r r \frac{\partial}{\partial r} = \frac{\partial \eta}{\partial r} \left(\frac{\partial \Psi}{\partial \eta} \frac{\partial}{\partial \zeta} - \frac{\partial \Psi}{\partial \zeta} \frac{\partial}{\partial \eta} \right)$$

Laminar Jet Diffusion Flame (Transformation Rules)

- Applying the **transformation rules**

$$\zeta = z + a, \quad \eta = \frac{\sqrt{2 \int_0^r \frac{\rho}{\rho_\infty} r dr}}{\zeta} \rightarrow \frac{\partial}{\partial z} = \frac{\partial}{\partial \zeta} + \frac{\partial \eta}{\partial z} \frac{\partial}{\partial \eta}, \quad \frac{\partial}{\partial r} = \frac{\partial \eta}{\partial r} \frac{\partial}{\partial \eta}$$

to the **convective terms** in the momentum and mixture fraction equations yields

$$\rho u_z r \frac{\partial}{\partial z} + \rho u_r r \frac{\partial}{\partial r} = \frac{\partial \eta}{\partial r} \left(\frac{\partial \Psi}{\partial \eta} \frac{\partial}{\partial \zeta} - \frac{\partial \Psi}{\partial \zeta} \frac{\partial}{\partial \eta} \right)$$

- The **diffusive terms** become

$$\frac{\partial}{\partial r} \left(\mu r \frac{\partial}{\partial r} \right) = \mu_\infty \frac{\partial \eta}{\partial r} \frac{\partial}{\partial \eta} \left(C \eta \frac{\partial}{\partial \eta} \right)$$

- C: Chapman-Rubensin-Parameter** $C = \frac{\rho \mu r^2}{2 \mu_\infty \int_0^r \rho r dr}$

- For constant density (with $\eta = r/\zeta$ and $\mu = \mu_\infty$): $C = 1$

- Formal transformation of the momentum and concentration equations and assumption that $C = f(\zeta, \eta)$
- With ansatz for non-dimensional stream function F

$$\Psi = \mu_{\infty} \zeta F(\zeta, \eta)$$

for the velocities follows

$$u_z = \frac{\partial F / \partial \eta}{\eta} \frac{\mu_{\infty}}{\rho_{\infty} \zeta}, \quad \rho u_r r = -\mu_{\infty} \left(\zeta \frac{\partial F}{\partial \zeta} + F - \eta \frac{\partial F}{\partial \eta} \right)$$

- u_z und u_r can be expressed as a function of the nondimensional stream function F and its derivatives

Laminar Jet Diffusion Flame (Transformation)

- From the momentum equation

$$\rho u_z r \frac{\partial u_z}{\partial z} + \rho u_r r \frac{\partial u_z}{\partial r} = \frac{\partial}{\partial r} \left(\mu r \frac{\partial u_z}{\partial r} \right)$$

→

$$\zeta \left(\frac{\partial F / \partial \eta}{\eta} \frac{\partial}{\partial \zeta} \frac{\partial F}{\partial \eta} - \frac{\partial F}{\partial \zeta} \frac{\partial}{\partial \eta} \frac{\partial F / \partial \eta}{\eta} \right) - \frac{\partial}{\partial \eta} \left(F \frac{\partial F / \partial \eta}{\eta} \right) = \frac{\partial}{\partial \eta} \left(C \eta \frac{\partial}{\partial \eta} \frac{\partial F / \partial \eta}{\eta} \right)$$

- Similarity solution only exists, if $F \neq f(\zeta)$
- Then, u_z is proportional to $1/\zeta$ (see previous slide)
→ velocity decreases linearly with $1/(z + a)$
- Prerequisites: Boundary conditions and C are independent of z
(e. g. $u_z = 0$ and $u_r = 0$ for $\eta \rightarrow 0$)

Laminar Jet Diffusion Flame (Resulting Equations)

- Equation for the **nondimensional stream function**

$$-\frac{\partial}{\partial \eta} \left(F \frac{\partial F / \partial \eta}{\eta} \right) = \frac{\partial}{\partial \eta} \left(C \eta \frac{\partial}{\partial \eta} \frac{\partial F / \partial \eta}{\eta} \right)$$

- Let $\omega = Z(z,r)/Z_a(z)$, ratio of the mixture fraction $Z_a(z)$ to its value at $r = 0$
- Applying the same transformations to the ω -equation yields

$$\zeta \left(\frac{\partial F}{\partial \eta} \frac{\partial \omega}{\partial \zeta} - \frac{\partial F}{\partial \zeta} \frac{\partial \omega}{\partial \eta} \right) + \zeta \frac{\partial F}{\partial \eta} \omega \frac{\partial \ln(Z_a)}{\partial \zeta} - F \frac{\partial \omega}{\partial \eta} = \frac{\partial}{\partial \eta} \left(\frac{C}{Sc} \eta \frac{\partial \omega}{\partial \eta} \right)$$

- In case of a similarity solution

$$-F \frac{\partial \omega}{\partial \eta} = \frac{\partial}{\partial \eta} \left(\frac{C}{Sc} \eta \frac{\partial \omega}{\partial \eta} \right)$$

Laminar Jet Diffusion Flame (Analytic Solution)

- Integration for $C = \text{const.}$ yields:

$$F = \frac{C(\gamma\eta)^2}{1 + (\gamma\eta)^2/4}, \quad \omega = \left(\frac{1}{1 + (\gamma\eta)^2/4} \right)^{2Sc}$$

where γ is integration constant

- The assumption $C = \text{const.}$ Holds if

$$C = \frac{\rho\mu r^2}{2\mu_\infty \int_0^r \rho r dr} \rightarrow C = \frac{\rho\mu}{\rho_m \mu_\infty}$$

and $\rho\mu/\rho_m \mu_\infty = \text{const.}$

- $C = \text{const.}$ Often not a good assumption, since $\mu \sim T^{0,7}$ und $\rho \sim T^{-1}$

- Constant of integration γ can be determined from the condition that the jet momentum is independent of ζ
- Substitution of the solution into the momentum balance

$$\int_0^{\infty} \rho u_z^2 r dr = \rho_0 u_{z,0}^2 \frac{d^2}{8}$$

yields

$$\gamma^2 = \frac{3}{64} \frac{\rho_0}{\rho_{\infty}} \frac{Re^2}{C^2}$$

- ρ_0 : density of the fuel stream
- Reynolds number $Re = u_{z,0} d / \nu_{\infty}$

- Analogously for the **mixture fraction** (with $Z_0 = 1$)

$$\int_0^{\infty} \rho u_z Z r dr = \rho_0 u_{z,0} \frac{d^2}{8}$$

→ **Mixture fraction on the centerline** $Z_a(z) = Z(z, r=0)$:

$$Z_a(z) = \frac{1 + 2Sc}{32} \frac{\rho_0}{\rho_{\infty}} \frac{Re}{C} \frac{d}{\zeta}$$

→ Z_a decreases with $1/\zeta$ (as the velocity)

Laminar Jet Diffusion Flame (Flame Length)

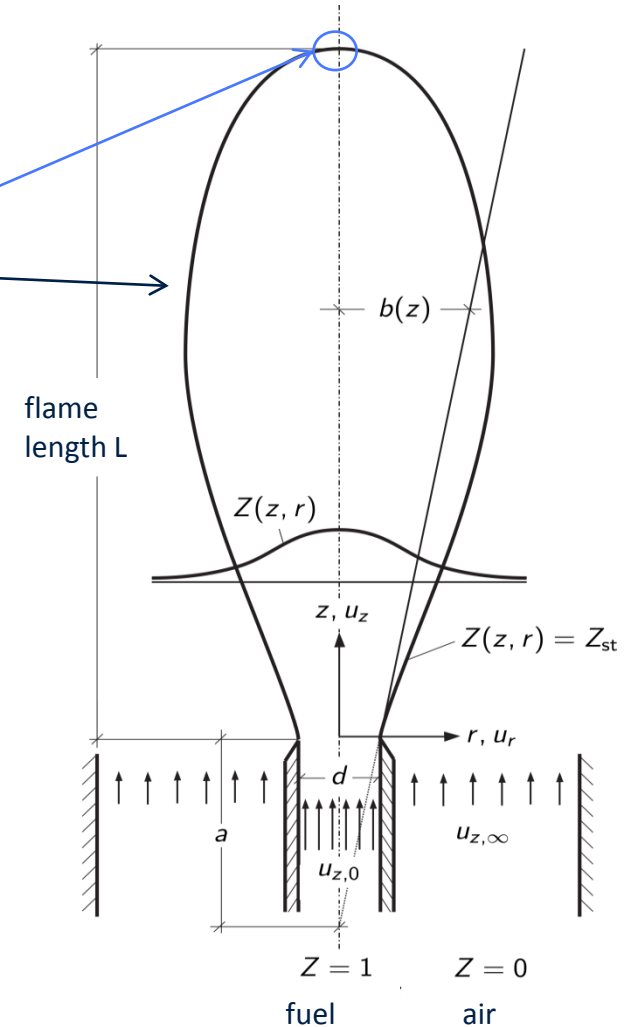
- Determination of the flame contour r as function of z from the condition

$$Z(z, r) = Z_a \omega(\eta) = Z_{st}$$

- Flame contour intersects centerline, $r = 0$, if $Z_a = Z_{st}$
- Corresponding value of z defines the flame length

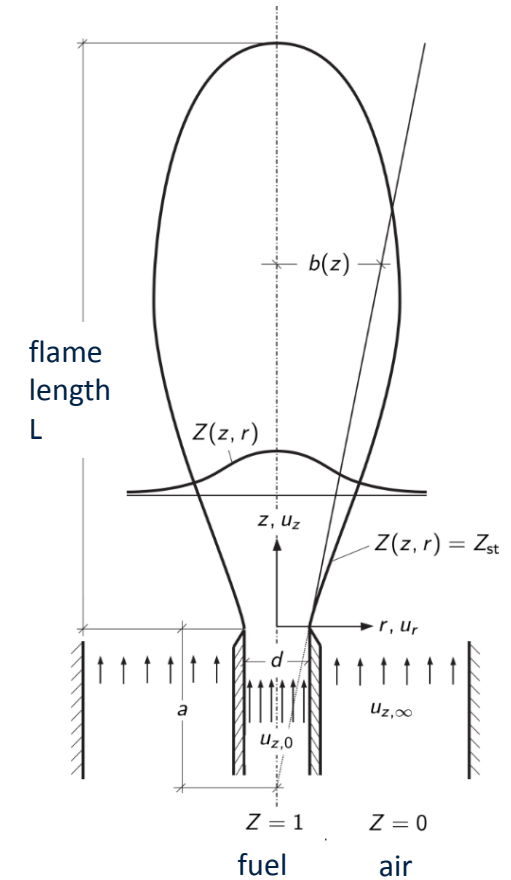
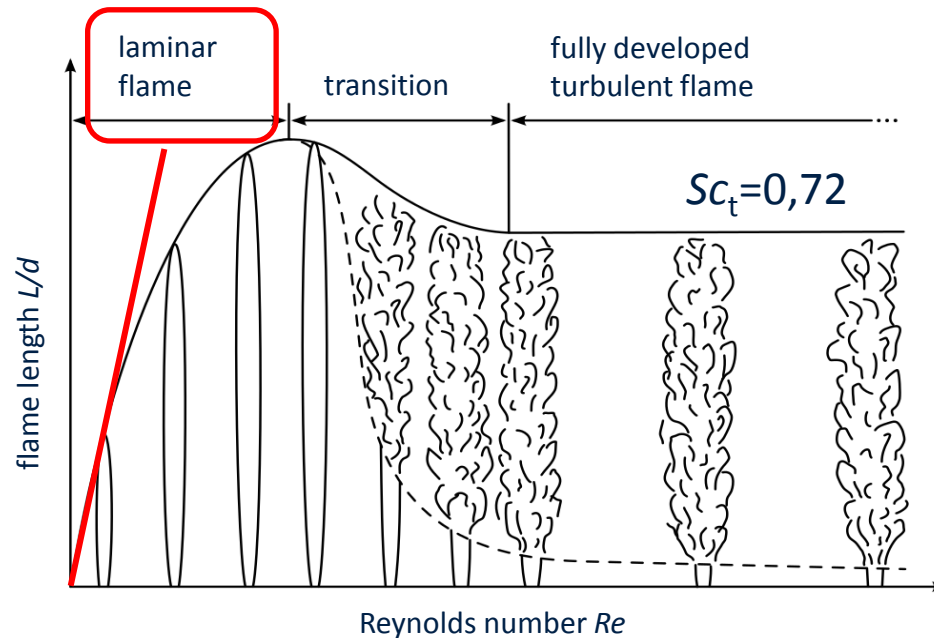
$$Z_a(z) = \frac{1 + 2Sc}{32} \frac{\rho_0}{\rho_\infty} \frac{Re}{C} \frac{d}{\zeta} \rightarrow L = \frac{1 + 2Sc}{32 Z_{st}} \frac{\rho_0}{\rho_\infty} \frac{u_0 d^2}{\nu} - a$$

- Valid for laminar jet flames without buoyancy




Laminar Jet Diffusion Flame

- For a given nozzle diameter, L increases linearly with the Reynolds number Re



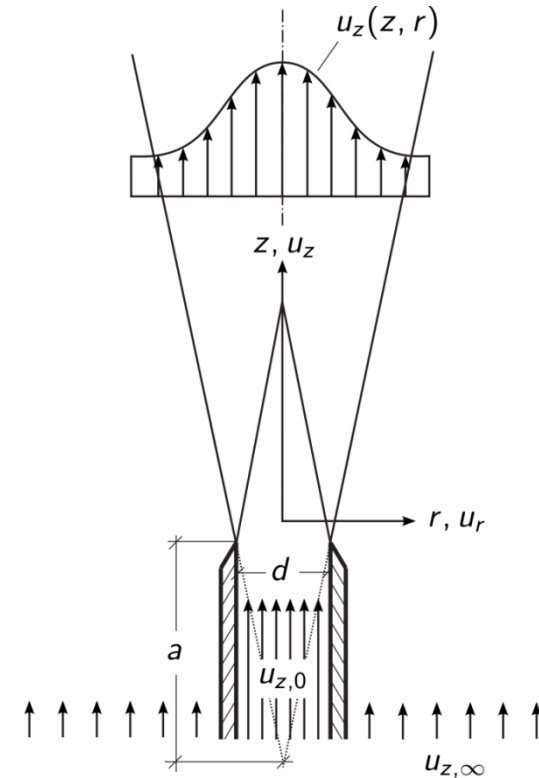
Course Overview

Part II: Turbulent Combustion

- Turbulence
 - Turbulent Premixed Combustion
 - Turbulent Non-Premixed Combustion
 - Turbulent Combustion Modeling
 - Applications
- Laminar Jet Diffusion Flames
 - Turbulent Jet Diffusion Flames
- 

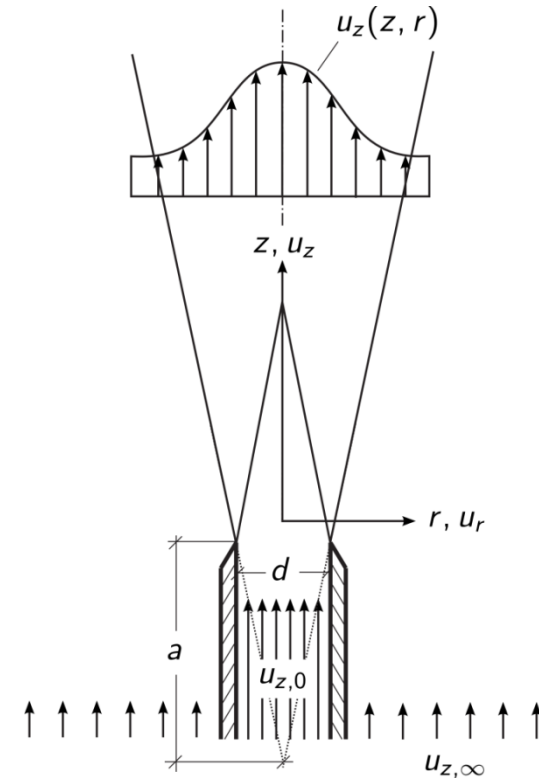
Turbulent Jet Diffusion Flame

- Shear flow at nozzle exit
- Flow instabilities (Kelvin-Helmholtz-instabilities) → laminar-turbulent transition
- Ring shaped turbulent shear layer propagates in radial direction
- Merging after 10 to 15 nozzle diameters downstream
- Streamlines are parallel in potential core
- Velocity profile reaches self similar state after 20-30 nozzle diameters



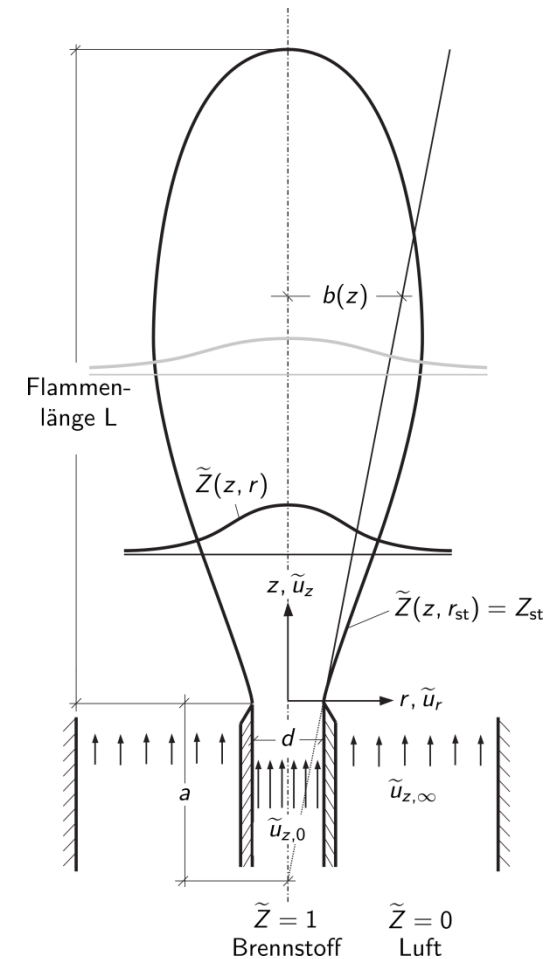
Round Turbulent Diffusion Flame

- Linear reduction of velocity along central axis
- Linear increase of jet width
- Assumption: fast chemical reaction
 - Scalar quantities such as temperature, concentration and density as function of mixture fraction Z
- Turbulent flow with variable density
 - Favre-averaged boundary layer equations





- 23



- Mean mixture fraction

$$\bar{\rho}\tilde{u}_z r \frac{\partial \tilde{Z}}{\partial z} + \bar{\rho}\tilde{u}_r r \frac{\partial \tilde{Z}}{\partial r} = \frac{\partial}{\partial r} \left(\frac{\bar{\rho}\nu_t}{Sc_t} r \frac{\partial \tilde{Z}}{\partial r} \right)$$



Round Turbulent Diffusion Flame

- Requires solving of equations for k and ε to determine ν_t
- Round turbulent jet: ν_t **approximately constant**
- Analogous for round laminar jet:

Laminar

$$\frac{\partial(\rho u_z r)}{\partial z} + \frac{\partial(\rho u_r r)}{\partial r} = 0$$

$$\rho u_z r \frac{\partial u_z}{\partial z} + \rho u_r r \frac{\partial u_z}{\partial r} = \frac{\partial}{\partial r} \left(\mu r \frac{\partial u_z}{\partial r} \right)$$

$$\rho u_z r \frac{\partial Z}{\partial z} + \rho u_r r \frac{\partial Z}{\partial r} = \frac{\partial}{\partial r} \left(\frac{\mu}{Sc} r \frac{\partial Z}{\partial r} \right)$$

Turbulent

$$\frac{\partial(\bar{\rho} \tilde{u}_z r)}{\partial z} + \frac{\partial(\bar{\rho} \tilde{u}_r r)}{\partial r} = 0$$

$$\bar{\rho} \tilde{u}_z r \frac{\partial \tilde{u}_z}{\partial z} + \bar{\rho} \tilde{u}_r r \frac{\partial \tilde{u}_z}{\partial r} = \frac{\partial}{\partial r} \left(\bar{\rho} \nu_t r \frac{\partial \tilde{u}_z}{\partial r} \right)$$

$$\bar{\rho} \tilde{u}_z r \frac{\partial \tilde{Z}}{\partial z} + \bar{\rho} \tilde{u}_r r \frac{\partial \tilde{Z}}{\partial r} = \frac{\partial}{\partial r} \left(\frac{\bar{\rho} \nu_t}{Sc_t} r \frac{\partial \tilde{Z}}{\partial r} \right)$$

Round Turbulent Diffusion Flame

- Special case: Jet in quiescent ambient
 - Treatment of turbulent equations like those in a laminar round jet case
 - Using the laminar theory
- Similarity coordinate

Laminar

$$\eta = \frac{\sqrt{2 \int_0^r \frac{\rho}{\rho_\infty} r dr}}{z + a}$$

$$C = \frac{\rho \mu r^2}{2 \mu_\infty \int_0^r \rho r dr}$$

Turbulent

$$\eta = \frac{\sqrt{2 \int_0^r \frac{\bar{\rho}}{\rho_\infty} r dr}}{z + a}$$

$$C = \frac{\bar{\rho}^2 \nu_t r^2}{2 \rho_\infty \nu_{t, \text{ref}} \int_0^r \bar{\rho} r dr}$$

- Chapman-Rubesin-Parameter

Round Turbulent Diffusion Flame

- Turbulent Chapman-Rubesin-Parameter approximately constant →

$$\tilde{u}_z = \frac{2C\gamma^2\nu_{t,\text{ref}}}{\zeta \left(1 + (\gamma\eta)^2 / 4\right)^2}$$

- Integration constant γ , containing fuel density and reference viscosity

$$\gamma^2 = \frac{3}{64} \frac{\rho_0}{\rho_\infty C^2} \left(\frac{u_{z,0}d}{\nu_{t,\text{ref}}} \right)^2 \quad \left(\text{laminar: } \gamma^2 = \frac{3}{64} \frac{\rho_0}{\rho_\infty} \frac{Re^2}{C^2} \right)$$

- The Favre-averaged velocity decreases proportional to $1/\zeta = 1/(z + a)$, just like in the laminar case

Round Turbulent Diffusion Flame

- Mean mixture fraction

$$\tilde{Z} = \frac{\tilde{Z}_a}{\left(1 + (\gamma\eta)^2 / 4\right)^{2Sc_t}}$$

with

$$\tilde{Z}_a = \frac{1 + 2Sc_t}{32} \frac{\rho_0}{\rho_\infty C} \left(\frac{u_{z,0} d}{\nu_{t,ref}} \right) \frac{d}{\zeta} \quad \left(\text{laminar: } Z_a = \frac{1 + 2Sc}{32} \frac{\rho_0}{\rho_\infty} \frac{Re d}{C \zeta} \right)$$

- Mixture fraction decreases proportional to $1/(z + a)$ on the jet axis
- Progression of profiles along jet axis resembles those of the laminar case
 - Also applies to the contour of the stoichiometric mixture

Round Turbulent Diffusion Flame

- Flame length L of round turbulent diffusion flame: Distance z from the nozzle, where the mean mixture fraction on the axis equals Z_{st}

$$\frac{L + a}{d} = \frac{1 + 2Sc_t}{32Z_{st}} \left(\frac{u_{z,0}d}{\nu_{t,ref}} \right) \frac{\rho_0}{\rho_\infty C}$$

- Comparison with experimental correlations (Hawthorne, Weddel and Hottel (1949))

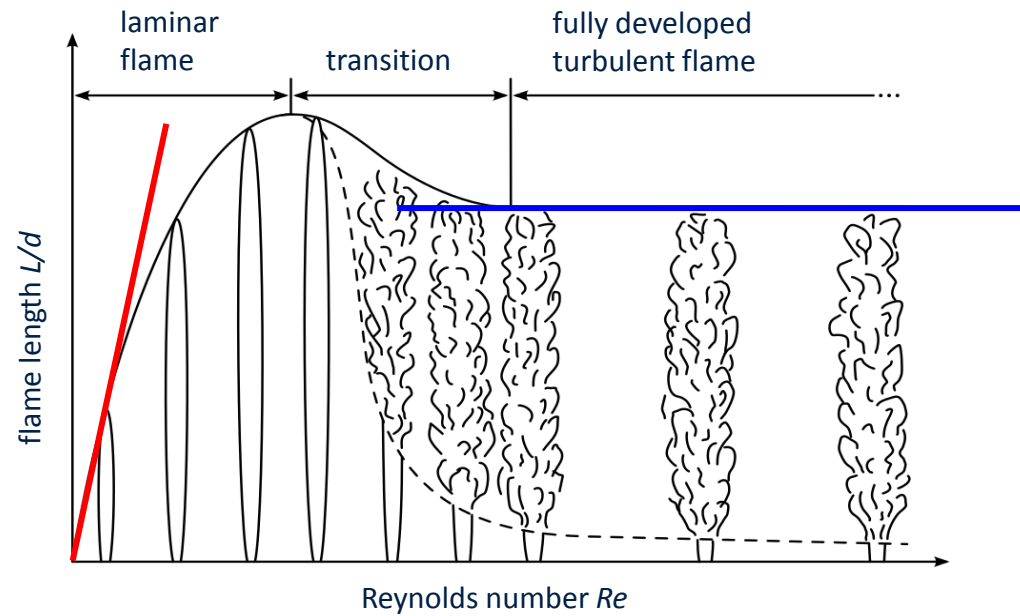
$$\frac{L + a}{d} = \frac{5,3}{Z_{st}} \sqrt{\frac{\rho_0}{\rho_\infty}}$$

- With $u_{z,0}d/\nu_{t,ref} = 70$ and $Sc_t=0,72$
- Complete agreement for $C = (\rho_0\rho_{st})^{1/2}/\rho_\infty$

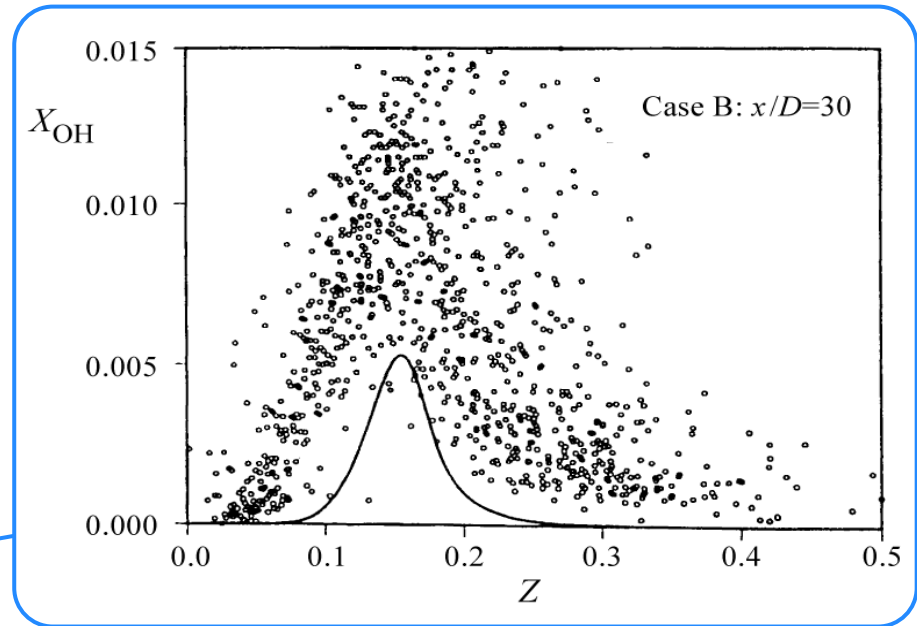
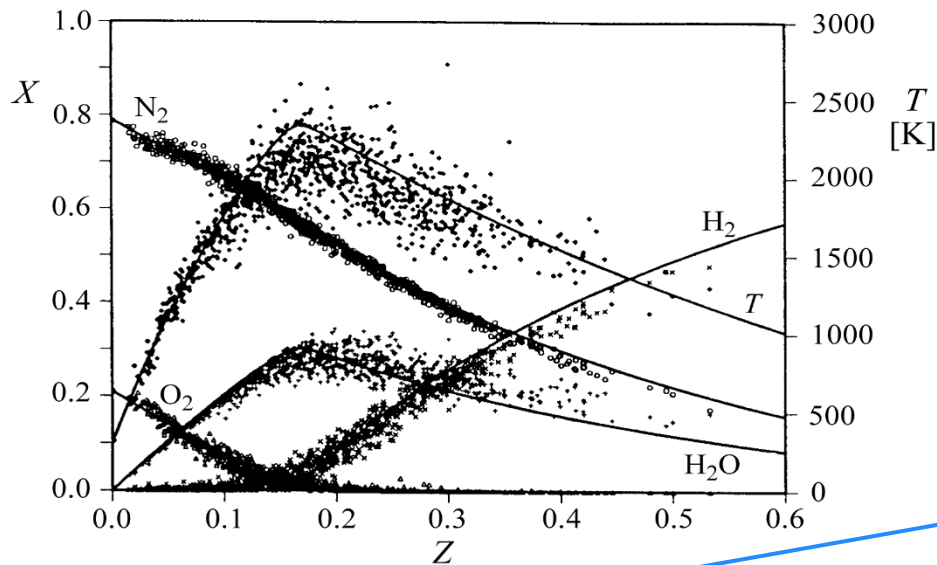
Round Turbulent Diffusion Flame

$$\frac{L + a}{d} = \frac{1 + 2Sc}{32Z_{st}} \frac{\rho_0}{\rho_\infty C} \overset{\text{linear}}{\boxed{\frac{u_0 d}{\nu}}}$$

$$\frac{L + a}{d} = \frac{1 + 2Sc_t}{32Z_{st}} \frac{\rho_0}{\rho_\infty C} \overset{\text{const.}}{\boxed{\frac{u_0 d}{\nu_{t,ref}}}} \approx 70$$



- Comparison of experimental results and simulations with chemical equilibrium



- Concentration of radicals and emissions cannot be described by infinitely fast chemistry

Summary

Part II: Turbulent Combustion

- Turbulence
- Turbulent Premixed Combustion
- **Turbulent Non-Premixed Combustion**
 - Laminar Jet Diffusion Flames
 - Turbulent Jet Diffusion Flames
- Turbulent Combustion Modeling
- Applications