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**Some Useful Results from Calculus** 

## Derivation of Gauss' Theorem

Let f(x, y, z) be a differentiable scalar function of (x, y, z).

$$\vec{f_1} \equiv \hat{i}f$$

By the divergence theorem,

$$\int_{V} \nabla \cdot \vec{f_{1}} \, dv = \int_{S} \vec{f_{1}} \cdot \vec{n} \, ds = \int_{S} n_{x} f \, ds$$

$$\nabla \cdot \vec{f_{1}} = \frac{\partial f}{\partial x}$$

$$\int_{V} \hat{i} \nabla \cdot \vec{f_{1}} \, dv = \int_{V} \hat{i} \frac{\partial f}{\partial x} \, dv = \int_{S} \hat{i} \, n_{x} f \, ds$$

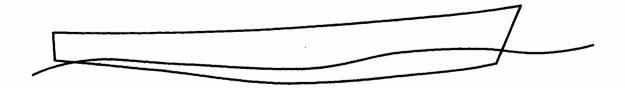
Similarly,

$$\int_{V} \hat{j} \nabla \cdot \vec{f_{2}} dv = \int_{V} \hat{j} \frac{\partial f}{\partial y} dv = \int_{S} \hat{j} n_{y} f ds$$
$$\int_{V} \hat{k} \nabla \cdot \vec{f_{3}} dv = \int_{V} \hat{k} \frac{\partial f}{\partial z} dv = \int_{S} \hat{k} n_{z} f ds$$

Now, add the last three equations together,

$$\int_V \nabla f \, dv = \int_S \vec{n} f \, ds$$

# Example of Use of Gauss Theorem: Froude Krylov Surge Force on a Ship



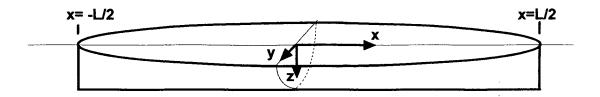
$$P = 
ho g A e^{-kz} \cos(kx - \omega t)$$
  $ec{F} = \int \int_S -p ec{n} \, dS$ 

$$egin{aligned} F_x &= \hat{i} \cdot \int \int_S -P ec{n} \, ds = -\hat{i} \int \int \int_V 
abla P \, dV \ \\ F_x &= -\int \int \int_V rac{\partial P}{\partial x} \, dV = 
ho g A k \int \int \int_V e^{-kz} \sin(kx - \omega t) dV \end{aligned}$$

$$egin{array}{ll} F_x &\simeq& 
ho g A k \int \int_V (1-kz) \sin(kx-\omega t) dV \ &=& 
ho g A k \int_L \left[ \int \int_{\mathrm{section}} dy \, dz 
ight] \sin(kx-\omega t) dx \ &- 
ho g A k^2 \int_L \left[ \int \int_{\mathrm{section}} z \, \, dy \, dz 
ight] \sin(kx-\omega t) dx \end{array}$$

$$F_x = 
ho g A k \int_L S(x) \sin(kx - \omega t) dx - 
ho g A k^2 \int_L z_{ca} S(x) \sin(kx - \omega t) dx$$

#### Example with Given Ship Shape



$$y = \frac{2W}{L^2D^2} \left(\frac{L}{2} - x\right)^2 (D - z)^2$$

For this shape:

$$S(x) = rac{2WD}{3L^2} \left(rac{L}{2} - x
ight)^2 \qquad \qquad z_{ca} = rac{D}{4}$$

Using these values:

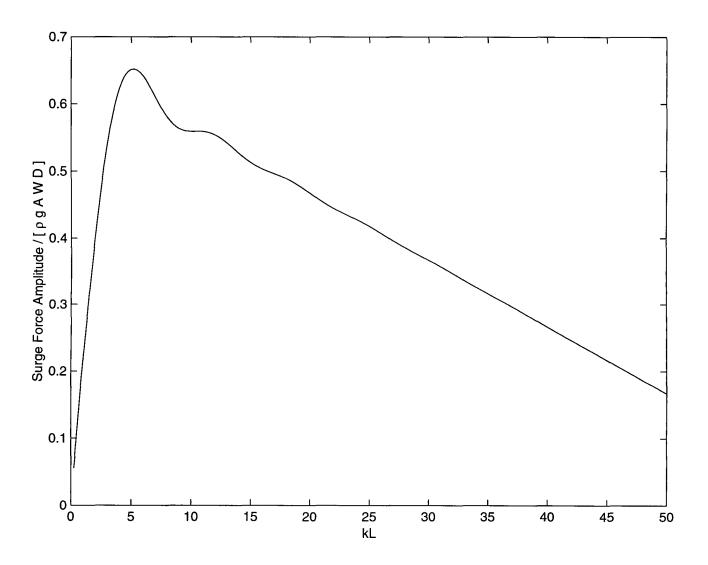
$$F_x = \frac{2}{3}\rho gAk\frac{WD}{L^2}\left(1 - \frac{kD}{4}\right)\left\{\cos\omega t\left[-\frac{2L}{k^2}\sin\frac{kL}{2} + \frac{L^2}{k}\cos\frac{kL}{2}\right] - \sin\omega t\left[\left(\frac{L^2}{k} - \frac{4}{k^3}\right)\sin\frac{kL}{2} + \frac{2L}{k^2}\cos\frac{kL}{2}\right]\right\}$$

$$\begin{split} \frac{F_x}{\rho gAWD} &= \frac{2}{3}kL\left(1-\frac{kL}{4}\frac{D}{L}\right)\left\{\cos\omega t\left[-\frac{2}{(kL)^2}\sin\frac{kL}{2}+\frac{1}{kL}\cos\frac{kL}{2}\right] \right. \\ &\left. -\sin\omega t\left[\left(\frac{1}{kL}-\frac{4}{(kL)^3}\right)\sin\frac{kL}{2}+\frac{2}{(kL)^2}\cos\frac{kL}{2}\right]\right\} \end{split}$$

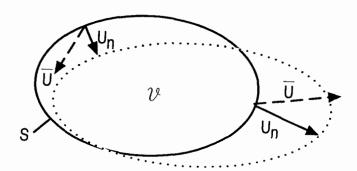
$$\left[\frac{F_x}{\rho g A W D}\right]_{\text{max}} = \frac{2}{3} k L \left(1 - \frac{k L}{4} \frac{D}{L}\right) \left\{ \left[ -\frac{2}{(k L)^2} \sin \frac{k L}{2} + \frac{1}{k L} \cos \frac{k L}{2} \right]^2 + \left[ \left(\frac{1}{k L} - \frac{4}{(k L)^3}\right) \sin \frac{k L}{2} + \frac{2}{(k L)^2} \cos \frac{k L}{2} \right]^2 \right\}^{1/2}$$

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```
% m-file script for gaussexp
tt = 2.0/3.0;
DoL = 0.06;
DoLf = DoL/4.;
m = 1:1:200;
kL = 0.25.*m;
kLi = 1.0 . / kL;
sn = sin(kL ./ 2.0);
cs = cos(kL ./ 2.0);
fnd = tt .* kL .* (1.0 - kL .* DoLf) .* ((( -2 ./ (kL .^ 2)) .* sn + kLi .* \checkmark
cs) .^ 2 ...
  + ( ( kLi -4.0 .* (kLi .^3)) .* sn + (2.0 ./ (kL.^2)).* cs) .^2) .^ 0.5;
q = [kL; fnd];
fid = fopen('surge.dat','w');
fprintf(fid,'%8.3f, %9.4f\n',q);
fclose(fid);
plot(kL, fnd)
xlabel('kL')
ylabel('Surge Force Amplitude / [ \rho g A W D ]')
```



# The Transport Theorem



Let  $f(\mathbf{x},t)$  be a differentiable scalar function of  $\mathbf{x}$  and t.

Consider the integral,

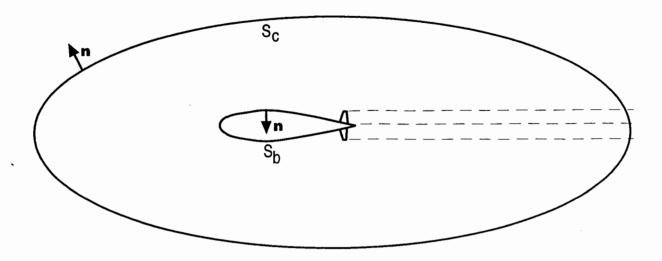
$$I(t) = \int \int \int_{\mathcal{V}(t)} f(\mathbf{x}, t) d\mathcal{V}$$

f is changing with time and  $\mathcal{V}$  is changing with time. The normal component of the velocity of any point on the surface, S of  $\mathcal{V}$  is called  $U_n$ .

$$\frac{dI}{dt} = \int \int \int_{\mathcal{V}} \frac{\partial f}{\partial t} d\mathcal{V} + \int \int_{S} f U_{n} dS = \int \int \int_{\mathcal{V}} \left\{ \frac{\partial f}{\partial t} + \nabla \cdot \left( f \vec{U} \right) \right\} d\mathcal{V}$$

Note that if  $\vec{U}$  is the fluid velocity, the surface S is a material surface and the Transport Theorem is simply the integral form of the Substantial Derivative.

## Pressure Forces and Moments on an Object



$$\mathbf{F} = \int \int_{S_b} p \, \mathbf{n} \, dS \qquad \qquad \mathbf{M} = \int \int_{S_b} p \, (\mathbf{r} \times \mathbf{n}) \, dS$$

Now use the (unsteady) Bernoulli equation:

$$\mathbf{F} = -\rho \int \int_{S_b} \left[ \frac{\partial \phi}{\partial t} + \frac{1}{2} \nabla \phi \cdot \nabla \phi \right] \mathbf{n} \, dS$$

$$\mathbf{M} = -\rho \int \int_{S_b} \left[ \frac{\partial \phi}{\partial t} + \frac{1}{2} \nabla \phi \cdot \nabla \phi \right] (\mathbf{r} \times \mathbf{n}) \, dS$$

The following results from applying Gauss, Transport and divergence theorems and boundary conditions:

$$\mathbf{F} = -\rho \frac{d}{dt} \int \int_{S_b} \phi \, \mathbf{n} \, dS - \rho \int \int_{S_c} \left[ \frac{\partial \phi}{\partial n} \nabla \phi - \mathbf{n} \frac{1}{2} \nabla \phi \cdot \nabla \phi \right] \, dS$$

$$\mathbf{M} = -\rho \frac{d}{dt} \int \int_{S_b} \phi \left( \mathbf{r} \times \mathbf{n} \right) dS - \rho \int \int_{S_c} \mathbf{r} \times \left[ \frac{\partial \phi}{\partial n} \nabla \phi - \mathbf{n} \frac{1}{2} \nabla \phi \cdot \nabla \phi \right] dS$$