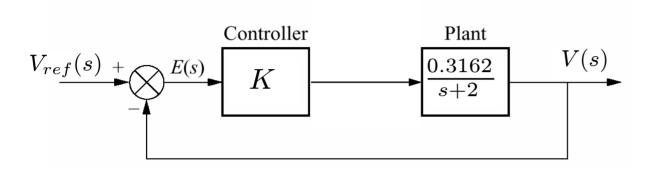
# Cranking up the gain ©

#### Type **0** system (no disturbance)

Steady-state error due to step input:

$$e_R(\infty) = \frac{2}{2 + 0.3162K}$$

$$e_R(\infty) \to 0$$
 as  $K \to \infty$ 



D(s)

#### Type 1 system with disturbance

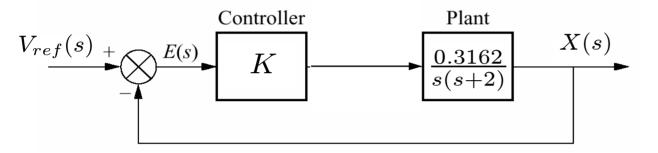
Steady-state error due to step input:

$$e_R(\infty) = 0$$

Steady-state error due to step disturbance:

## Cranking up the gain ⊗

#### Type 1 system (no disturbance)



Closed-loop transfer function

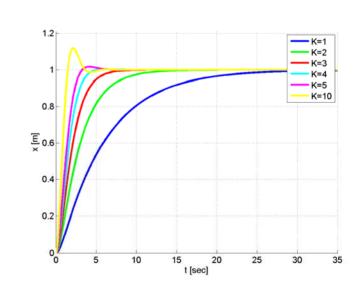
$$\frac{X(s)}{V_{ref}(s)} = \frac{0.3162K}{s^2 + 2s + 0.3162K}$$

Pole locations

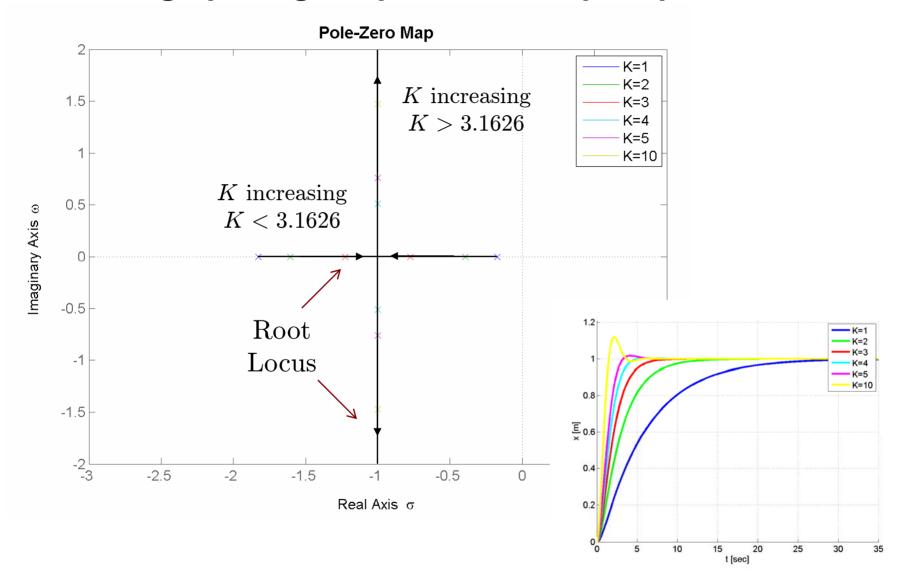
$$p_1 = -1 + \sqrt{1 - 0.3162K}$$
  $p_2 = -1 - \sqrt{1 - 0.3162K}$ 

System becomes **underdamped**⇒ ⇒ step response **overshoots** if

$$1 - 0.3162K < 0 \Leftrightarrow K > 3.1626$$

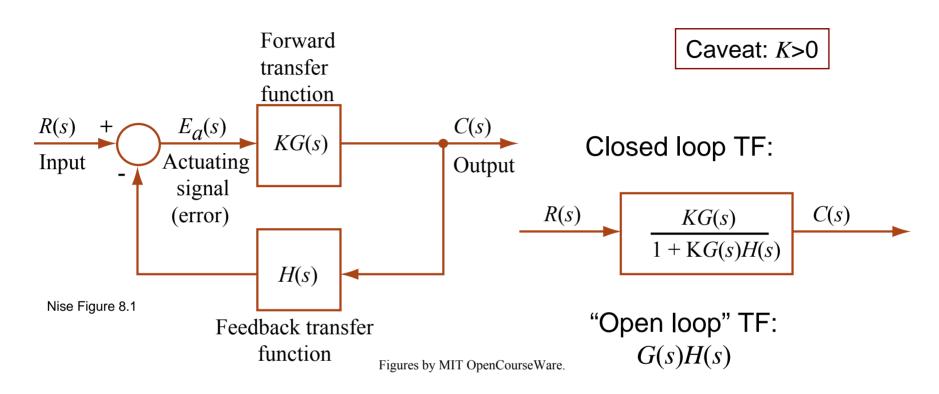


## Cranking up the gain: poles and step response





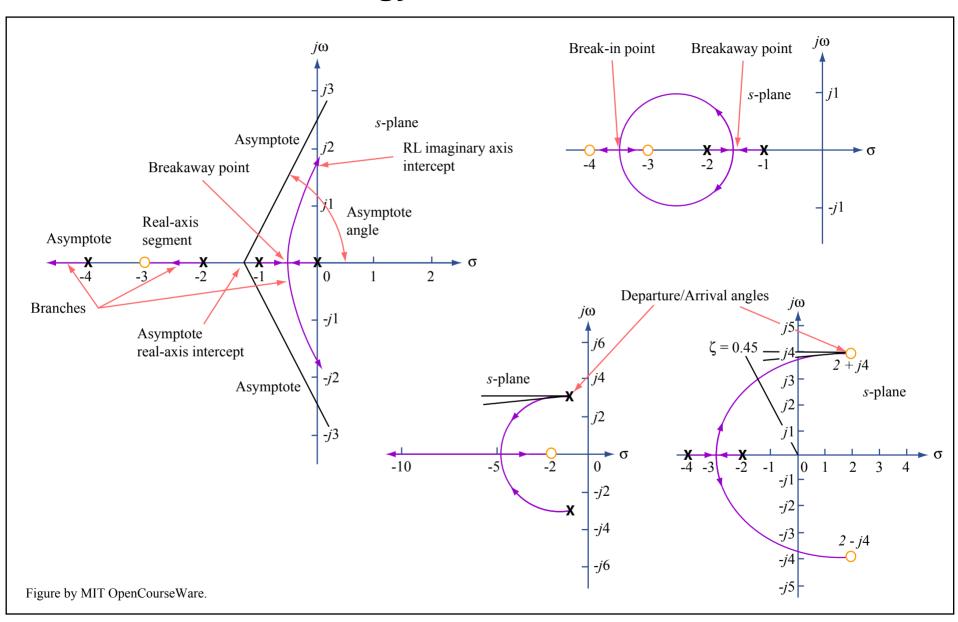
### Root locus for nonunity feedback systems



Closed-loop pole locations

$$1 + KG(s)H(s) = 0 \Rightarrow \begin{cases} K = 1/|G(s)H(s)|; \\ \angle KG(s)H(s) = (2n+1)180^{\circ}. \end{cases}$$

## **Root locus terminology**





### **Root-locus sketching rules**

- Rule 1: # branches = # poles
- Rule 2: symmetrical about the real axis
- Rule 3: real-axis segments are to the left of an odd number of realaxis finite poles/zeros

Let 
$$G(s) = \frac{N_G(s)}{D_G(s)}$$
,  $H(s) = \frac{N_H(s)}{D_H(s)}$ .

$$\Rightarrow \angle G(s)H(s) = \sum \angle (\text{poles}) - \sum \angle (\text{zeros}).$$

Recall angle condition for closed-loop pole:

$$\angle KG(s)H(s) = (2n+1)180^{\circ}.$$

Complex-pole/zero contributions: **cancel**because of symmetry
Real-pole/zero contributions: each is
0° from the left, 180° from the right;
total contributions from right must be
odd number of 190°'s to satisfy angle condition.

Image removed due to copyright restrictions.

Please see: Fig. 8.8 in Nise, Norman S. Control Systems Engineering. 4th ed. Hoboken, NJ: John Wiley, 2004.

#### **Root-locus sketching rules**

Rule 4: RL begins at poles, ends at zeros

Let 
$$G(s) = \frac{N_G(s)}{D_G(s)}$$
,  $H(s) = \frac{N_H(s)}{D_H(s)}$ .  
 $\Rightarrow$  Closed-loop TF $(s) = \frac{KN_G(s)D_H(s)}{D_G(s)D_H(s) + KN_G(s)N_H(s)}$ .

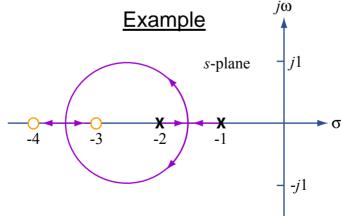
If  $K \to 0^+$  (small gain limit)

If  $K \to +\infty$  (large gain limit)

Closed-loop 
$$TF(s) \approx \frac{KN_G(s)D_H(s)}{D_G(s)D_H(s) + \epsilon} \Rightarrow$$

Closed-loop 
$$TF(s) \approx \frac{KN_G(s)D_H(s)}{\epsilon + KN_G(s)N_H(s)} \Rightarrow$$

closed–loop denominator is denominator of G(s)H(s) $\Rightarrow$ closed–loop poles are the poles of G(s)H(s). closed-loop denominator is numerator of G(s)H(s)  $\Rightarrow$  closed-loop poles are the zeros of G(s)H(s).



Nise Figure 8.10

Figure by MIT OpenCourseWare.

#### Please see the following selections from

Mathworks, Inc. "Control System Toolbox Getting Started Guide."

 $http://www.mathworks.com/access/helpdesk/help/pdf\_doc/control/get\_start.pdf \\ \square$ 

Ch. 4, pp. 3-18