

# Gravitation

(i) Gravitational law (force of attraction):-

$$F = G \frac{m_1 m_2}{r^2}$$

- \* Central
- \* Conservative
- \* Independent of medium

$$G = 6.67 \times 10^{-11} \text{ Nm}^2 \text{ kg}^{-2}$$

\* NOTE :- Two objects will attract maximum if  $m_1 = m_2$ .

(ii) Acceleration due to gravity :-

$$g = \frac{G M_e}{R_e^2}$$

(a) Variation at a height 'h' →

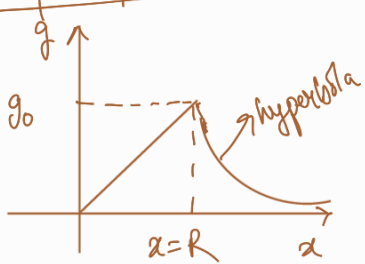
$$g' = g \left[ 1 + \frac{h}{R} \right]^{-2} \quad [\text{when } h > 600 \text{ km}]$$

$$g' = g \left[ 1 - \frac{2h}{R} \right] \quad [\text{when } h < 600 \text{ km}]$$

(b) Variation at a depth 'd' →

$$g' = g \left[ 1 - \frac{d}{R} \right]$$

Graph b/w the variations →



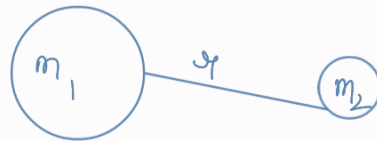
(iii) Gravity due to rotation :-

$$g' = g - \omega^2 R \cos^2 \theta \rightarrow \text{Angle}$$

(at equator,  $\theta = 0 \rightarrow g$  will be maximum)  
(at pole,  $\theta = 90 \rightarrow g$  will be minimum)

(iv) Gravitational P.E. →

$$U = -\frac{G m_1 m_2}{r}, \quad U_\infty = 0$$



(v) V<sub>escape</sub> :-

$$\frac{1}{2} m v^2 = \frac{G M m}{r}$$

$$\Rightarrow v = \sqrt{\frac{2GM}{R}} \quad \text{or} \quad v = \sqrt{2gR}$$

## Kepler's laws

(i) planets move in an elliptical path → with sun as one of the foci

(ii)  $\frac{dA}{dt} = \text{const}$  → Area Velocity

(iii)  $T^2 \propto a^3$

[Square of the time period is directly proportional to the cube of the semi-major axis]

## Motion of Satellites

$$V_0 = \sqrt{\frac{GM_e}{R}}$$

$$V_0 = \sqrt{\frac{GM_e}{(R_e + h)}}$$

Energy → short cut Trick

$$+B = -E = +K = -\frac{U}{2}$$

Geo-stationary Satellites  $\div \rightarrow T = 24 \text{ hrs}$

$$T = \frac{2\pi}{\sqrt{GM_e}} r^{3/2}, \quad r = 35786 \text{ km}$$

Learn

$V = V_0 \rightarrow$  Circular Motion

$V = V_e \rightarrow$  Parabolic

$V > V_e \rightarrow$  Hyperbolic

$V_e > V > V_0 \rightarrow$  Elliptical path

$V < V_0 \rightarrow$  elliptical and then falls on the planet earth.