Manufactured Solution for the 1D Compressible Transient Navier—Stokes Equations with Sutherland Viscosity Model using MapleTM

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Abstract

The Method of Manufactured Solutions is a valuable approach for code verification, providing means to verify how accurately the numerical method solves the partial differential equations of interest. This document presents the source terms generated by the application of the Method of Manufactured Solutions (MMS) on the 1D transient Navier–Stokes equations with Sutherland Viscosity Model using the analytical manufactured solutions for density, velocity and pressure presented by Roy et al. (2002).

1 Mathematical Model

The conservation of mass, momentum, and total energy for a compressible transient viscous fluid may be written as:

$$\frac{\partial(\rho)}{\partial t} + \frac{\partial(\rho u)}{\partial x} = 0,\tag{1}$$

$$\frac{\partial(\rho \boldsymbol{u})}{\partial t} + \nabla \cdot (\rho \boldsymbol{u} \boldsymbol{u}) = -\nabla p + \nabla \cdot (\boldsymbol{\tau}), \tag{2}$$

$$\frac{\partial(\rho e_t)}{\partial t} + \nabla \cdot (\rho \mathbf{u} H) = -\nabla \cdot (p \mathbf{u}) - \nabla \cdot \mathbf{q} + \nabla \cdot (\boldsymbol{\tau} \cdot \mathbf{u}). \tag{3}$$

Equations (1)–(3) are known as Navier–Stokes equations and, ρ is the density, $\mathbf{u} = u$ is the velocity in x-direction, and p is the pressure. The total enthalpy, H, may be expressed in terms of the total energy per unit mass e_t , density, and pressure:

$$H = e_t + \frac{p}{\rho}.$$

For a calorically perfect gas, the Navier-Stokes equations are closed with two auxiliary relations for energy:

$$e_t = e + \frac{\boldsymbol{u} \cdot \boldsymbol{u}}{2}, \quad \text{and} \quad e = \frac{1}{\gamma - 1} RT,$$
 (4)

and with the ideal gas equation of state:

$$p = \rho RT. \tag{5}$$

The shear stress tensor is:

$$\tau_{xx} = \frac{2}{3}\mu \left(2\frac{\partial u}{\partial x} - \frac{\partial v}{\partial y} \right),\tag{6}$$

where μ is the absolute viscosity. The heat flux vector $\mathbf{q} = q_x$ is given by:

$$q_x = -\kappa \frac{\partial T}{\partial x},\tag{7}$$

where κ is the thermal conductivity, which can be determined by choosing the Prandtl number:

$$\kappa = \frac{\gamma R \mu}{(\gamma - 1) \text{Pr}}.$$
 (8)

1.1 Sutherland viscosity model

Sutherland (1893) published a relationship between the absolute temperature of an ideal gas, T, and its dynamic (absolute) viscosity, μ . The model is based on the kinetic theory of ideal gases and an idealized intermolecular-force potential. The general equation is given as:

$$\mu = \frac{A_{\mu} T^{\frac{3}{2}}}{T + B_{\mu}} \tag{9}$$

with

$$A_{\mu} = \frac{\mu_{\text{ref}}}{T_{\text{ref}}^{\frac{3}{2}}} (T_{\text{ref}} + B_{\mu}), \tag{10}$$

where B_{μ} is the Sutherland temperature, T_{ref} is a reference temperature, and μ_{ref} is the viscosity at the reference temperature T_{ref} .

2 Manufactured Solution

The Method of Manufactured Solutions (MMS) applied to Navier–Stokes equations consists in modifying Equations (1) – (3) by adding a source term to the right-hand side of each equation, so the modified set of equations conveniently has the analytical solution chosen *a priori*.

Roy et al. (2002) introduce the general form of the primitive manufactured solution variables to be a function of sines and cosines in x, y and z only. In this work, Roy et al. (2002)'s manufactured solutions are modified in order to address temporal accuracy as well:

$$\phi(x,t) = \phi_0 + \phi_x f_s \left(\frac{a_{\phi x} \pi x}{L} \right) + \phi_t f_s \left(\frac{a_{\phi t} \pi t}{L} \right), \tag{11}$$

where $\phi = \rho, u$ or p, and $f_s(\cdot)$ functions denote either sine or cosine function. Note that in this case, ϕ_x , and ϕ_t are constants and the subscripts do not denote differentiation.

Although Roy et al. (2002) provide the constants used in the manufactured solutions for the 2D supersonic and subsonic cases for Euler and Navier-Stokes equations, only the source term for the 2D mass conservation equation (1) is presented.

Source terms for mass conservation (Q_{ρ}) , momentum (Q_u) and total energy (Q_{e_t}) equations are obtained by symbolic manipulations of compressible transient Navier–Stokes equations above using Maple 15 (Maplesoft, 2011) and are presented in the following sections for the one, two and three-dimensional cases.

The manufactured analytical solution for each one of the variables in the 1D transient Navier–Stokes equations are:

$$\rho(x,t) = \rho_0 + \rho_x \sin\left(\frac{a_{\rho x}\pi x}{L}\right) + \rho_t \sin\left(\frac{a_{\rho t}\pi t}{L_t}\right),$$

$$u(x,t) = u_0 + u_x \sin\left(\frac{a_{ux}\pi x}{L}\right) + u_t \cos\left(\frac{a_{ut}\pi t}{L_t}\right),$$

$$p(x,t) = p_0 + p_x \cos\left(\frac{a_{px}\pi x}{L}\right) + p_t \cos\left(\frac{a_{pt}\pi t}{L_t}\right).$$
(12)

The MMS applied to 1D Navier–Stokes equations with Sutherland viscosity model consists in modifying Equations (1)–(3) by adding a source term to the right-hand side of each equation:

$$\frac{\partial(\rho)}{\partial t} + \frac{\partial(\rho u)}{\partial x} = Q_{\rho}$$

$$\frac{\partial(\rho u)}{\partial t} + \frac{\partial(\rho u^{2})}{\partial x} + \frac{\partial(p)}{\partial x} - \frac{\partial(\tau_{xx})}{\partial x} = Q_{u},$$

$$\frac{\partial(\rho e_{t})}{\partial t} + \frac{\partial(\rho u e_{t})}{\partial x} + \frac{\partial(p u)}{\partial x} + \frac{\partial(q u)}{\partial x} - \frac{\partial(u \tau_{xx})}{\partial x} = Q_{e_{t}},$$
(13)

so this modified set of equations has for analytical solution Equation (12).

The source terms Q_{ρ} , Q_{u} , Q_{v} and $Q_{e_{t}}$ are presented with the use of the following auxiliary variables:

$$\begin{aligned} &\operatorname{Rho} = \rho_0 + \rho_x \sin\left(\frac{a_{\rho x} \pi x}{L}\right) + \rho_t \sin\left(\frac{a_{\rho t} \pi t}{L}\right), \\ &\operatorname{U} = u_0 + u_x \sin\left(\frac{a_{ux} \pi x}{L}\right) + u_t \cos\left(\frac{a_{ut} \pi t}{L}\right), \\ &\operatorname{P} = p_0 + p_x \cos\left(\frac{a_{px} \pi x}{L}\right) + p_t \cos\left(\frac{a_{pt} \pi t}{L}\right), \\ &\operatorname{T} = \frac{\operatorname{P}}{R \operatorname{Rho}}, \\ &\operatorname{Mu} = \frac{A_{\mu} \operatorname{T}^{\frac{3}{2}}}{\operatorname{T} + B_{\mu}}, \end{aligned} \tag{14}$$

which simply are the manufactured solutions for ρ , u, v and p, the temperature (5), and the fluid viscosity according to Sutherland model (9), respectively. The following expression for the derivative of the viscosity is also used:

$$\frac{\partial \operatorname{Mu}}{\partial x} = \frac{a_{\rho x} \pi \rho_x \operatorname{Mu}^2}{A_{\mu} L \operatorname{Rho} \sqrt{T}} \cos \left(\frac{a_{\rho x} \pi x}{L}\right) - \frac{3}{2} \frac{a_{\rho x} \pi \rho_x \operatorname{Mu}}{L \operatorname{Rho}} \cos \left(\frac{a_{\rho x} \pi x}{L}\right) + \frac{a_{p x} \pi p_x \operatorname{Mu}^2}{A_{\mu} L R \operatorname{Rho} T^{3/2}} \sin \left(\frac{a_{p x} \pi x}{L}\right) - \frac{3}{2} \frac{a_{p x} \pi p_x \operatorname{Mu}}{L R \operatorname{Rho} T} \sin \left(\frac{a_{p x} \pi x}{L}\right) \tag{15}$$

2.1 1D Mass Conservation

The mass conservation equation may be written as an operator \mathcal{L}_{ρ} :

$$\mathcal{L}_{\rho} = \mathcal{L}_{\rho \, \mathrm{time}} + \mathcal{L}_{\rho \, \mathrm{convection}}$$

where:

$$\mathcal{L}_{\rho \text{ time}} = \frac{\partial(\rho)}{\partial t}
\mathcal{L}_{\rho \text{ convection}} = \frac{\partial(\rho u)}{\partial x}.$$
(16)

The operators defined in Eq. (16) are applied into Equation (12), providing respective source terms that will compound source term Q_{ρ} :

$$Q_{\rho} = Q_{\rho \text{ time}} + Q_{\rho \text{ convection}}.$$

They are:

$$\begin{split} Q_{\rho\,\mathrm{convection}} &= \frac{a_{\rho x}\pi \rho_x\, \mathtt{U}}{L} \cos\left(\frac{a_{\rho x}\pi x}{L}\right) + \frac{a_{ux}\pi u_x\, \mathtt{Rho}}{L} \cos\left(\frac{a_{ux}\pi x}{L}\right) \\ Q_{\rho\,\mathrm{time}} &= \frac{a_{\rho t}\rho_t\pi}{L_t} \cos\left(\frac{a_{\rho t}\pi t}{L_t}\right) \end{split}$$

where Rho and U are defined in Equation (14).

2.2 1D Momentum

2.2.1 Velocity u

For the generation of the analytical source term Q_u , x-momentum equation (2) is written as an operator \mathcal{L}_u :

$$\mathcal{L}_u = \mathcal{L}_{u \text{ time}} + \mathcal{L}_{u \text{ convection}} + \mathcal{L}_{u \text{ pressure}} + \mathcal{L}_{u \text{ viscous}}$$

with each one of the sub-operators defined as follows:

$$\mathcal{L}_{u \text{ time}} = \frac{\partial(\rho u)}{\partial t}$$

$$\mathcal{L}_{u \text{ convection}} = \frac{\partial(\rho u^2)}{\partial x}$$

$$\mathcal{L}_{u \text{ pressure}} = \frac{\partial(p)}{\partial x}$$

$$\mathcal{L}_{u \text{ viscous}} = -\frac{\partial(\tau_{xx})}{\partial x}$$

Source term Q_u is obtained by operating \mathcal{L}_u on Equations (12) together with the use of the auxiliary relations given in Equations (14) and (15). It yields:

$$Q_u = Q_{u \text{ time}} + Q_{u \text{ convection}} + Q_{u \text{ pressure}} + Q_{u \text{ viscous}}$$

with

$$\begin{split} Q_{u\,\text{convection}} &= \frac{a_{\rho x}\pi\rho_x\, \mathbf{U}^2}{L}\cos\left(\frac{a_{\rho x}\pi x}{L}\right) + \frac{2a_{ux}\pi u_x\, \mathrm{Rho}\, \mathbf{U}}{L}\cos\left(\frac{a_{ux}\pi x}{L}\right) \\ Q_{u\,\text{pressure}} &= -\frac{a_{px}\pi p_x}{L}\sin\left(\frac{a_{px}\pi x}{L}\right) \\ Q_{u\,\text{viscous}} &= \frac{4}{3}\, \mathrm{Mu}\frac{a_{ux}^2\pi^2 u_x}{L^2}\sin\left(\frac{a_{ux}\pi x}{L}\right) - \frac{4}{3}\frac{\partial\, \mathrm{Mu}}{\partial x}\frac{a_{ux}\pi u_x}{L}\cos\left(\frac{a_{ux}\pi x}{L}\right) \\ Q_{u\,\text{time}} &= \frac{a_{\rho t}\pi\rho_t\, \mathbf{U}}{L_t}\cos\left(\frac{a_{\rho t}\pi t}{L_t}\right) - \frac{a_{ut}\pi u_t\, \mathrm{Rho}}{L_t}\sin\left(\frac{a_{ut}\pi t}{L_t}\right) \end{split}$$

with Mu, Rho, and U defined in Equation (14) and the derivative $\frac{\partial Mu}{\partial x}$ defined in Equation (15).

2.3 Total Energy Conservation Equation

The total energy equation is written as an operator:

$$\mathcal{L}_{e_t} = \mathcal{L}_{e_t ext{time}} + \mathcal{L}_{e_t ext{ convection}} + \mathcal{L}_{e_t ext{ pressure work}} + \mathcal{L}_{e_t ext{ viscous work}} + \mathcal{L}_{e_t ext{ conduction}}$$

with

$$\mathcal{L}_{e_t \text{time}} = \frac{\partial(\rho e_t)}{\partial t},$$

$$\mathcal{L}_{e_t \text{ convection}} = \frac{\partial(\rho u e_t)}{\partial x},$$

$$\mathcal{L}_{e_t \text{ pressure work}} = +\frac{\partial(\rho u)}{\partial x},$$

$$\mathcal{L}_{e_t \text{ conduction}} = +\frac{\partial(q_x)}{\partial x},$$

$$\mathcal{L}_{e_t \text{ viscous work}} = -\frac{\partial(u \tau_{xx})}{\partial x},$$

Therefore, source term Q_{e_t} is given by

$$Q_{e_t} = Q_{e_t \text{time}} + Q_{e_t \text{ convection}} + Q_{e_t \text{ pressure work}} + Q_{e_t \text{ viscous work}} + Q_{e_t \text{ conduction}},$$

where:

$$\begin{split} Q_{e\,\text{convection}} &= \frac{1}{2} \frac{a_{\rho x} \pi \rho_{x} \, \mathbb{U}^{3}}{L} \cos \left(\frac{a_{\rho x} \pi x}{L} \right) + \frac{3}{2} \frac{a_{u x} \pi u_{x} \, \text{Rho} \, \mathbb{U}^{2}}{L} \cos \left(\frac{a_{u x} \pi x}{L} \right) + \\ &- \frac{a_{p x} \pi p_{x} \, \mathbb{U}}{(\gamma - 1)L} \sin \left(\frac{a_{p x} \pi x}{L} \right) + \frac{a_{u x} \pi u_{x} \, \mathbb{P}}{(\gamma - 1)L} \cos \left(\frac{a_{u x} \pi x}{L} \right) \\ Q_{e\,\text{pressure work}} &= - \frac{a_{p x} \pi p_{x} \, \mathbb{U}}{L} \sin \left(\frac{a_{p x} \pi x}{L} \right) + \frac{a_{u x} \pi u_{x} \, \mathbb{P}}{L} \cos \left(\frac{a_{u x} \pi x}{L} \right) \\ Q_{e\,\text{conduction}} &= - \frac{2 \, \text{k} a_{\rho x}^{2} \pi^{2} \rho_{x}^{2} \, \mathbb{P}}{L^{2} R \, \text{Rho}^{3}} \cos \left(\frac{a_{\rho x} \pi x}{L} \right)^{2} - \frac{2 \, \text{k} a_{\rho x} a_{p x} \pi^{2} \rho_{x} p_{x}}{L^{2} R \, \text{Rho}^{2}} \cos \left(\frac{a_{\rho x} \pi x}{L} \right) \sin \left(\frac{a_{p x} \pi x}{L} \right) + \\ &- \frac{\text{k} a_{\rho x}^{2} \pi^{2} \rho_{x} \, \mathbb{P}}{L^{2} R \, \text{Rho}^{2}} \sin \left(\frac{a_{\rho x} \pi x}{L} \right) + \frac{\text{k} a_{p x}^{2} \pi^{2} p_{x}}{L^{2} R \, \text{Rho}} \cos \left(\frac{a_{p x} \pi x}{L} \right) \\ Q_{e\,\text{viscous work}} &= -\frac{4}{3} \frac{a_{u x}^{2} \, \text{Mu} \pi^{2} u_{x}^{2}}{L^{2}} \cos \left(\frac{a_{u x} \pi x}{L} \right)^{2} + \frac{4}{3} \frac{a_{u x}^{2} \, \text{Mu} \pi^{2} u_{x} \, \mathbb{U}}{L^{2}} \sin \left(\frac{a_{u x} \pi x}{L} \right) \\ Q_{e\,\text{time}} &= -\frac{a_{u t} u_{t} \pi \, \text{Rho} \, \mathbb{U}}{L_{t}} \sin \left(\frac{a_{u t} \pi t}{L_{t}} \right) + \frac{1}{2} \frac{a_{\rho t} \rho_{t} \pi \, \mathbb{U}^{2}}{L_{t}} \cos \left(\frac{a_{\rho t} \pi t}{L_{t}} \right) - \frac{a_{p t} p_{t} \pi}{(\gamma - 1) L_{t}} \sin \left(\frac{a_{p t} \pi t}{L_{t}} \right). \end{split}$$

Again, Mu, Rho, U, and P are defined in Equation (14) and the derivative $\frac{\partial Mu}{\partial x}$ are given in Equation (15). Accordingly,

$$\mathtt{k} = rac{\gamma R \, \mathtt{Mu}}{(\gamma - 1) \mathrm{Pr}}.$$

3 Boundary Conditions

Additionally to verifying code capability of solving the governing equations accurately in the interior of the domain of interest, one may also verify the software capability of correctly imposing boundary conditions. Therefore, the gradients of the analytical solutions (11) have been calculated and translated into C codes. For the 1D case, they are:

$$\frac{\partial(\rho)}{\partial x} = \frac{a_{\rho x}\pi\rho_x}{L}\cos\left(\frac{a_{\rho x}\pi x}{L}\right), \quad \frac{\partial(u)}{\partial x} = -\frac{a_{p x}\pi p_x}{L}\sin\left(\frac{a_{p x}\pi x}{L}\right), \quad \frac{\partial(p)}{\partial x} = \frac{a_{u x}\pi u_x}{L}\cos\left(\frac{a_{u x}\pi x}{L}\right).$$

4 List of model / manufactured solution parameters

There are a variety of parameters present in the flow described by Navier-Stokes equations with the passive transport of a generic scalar and the Sutherland viscosity model, due to both fluid properties and the constants arising from the chosen manufactured solutions.

Table 1 shows the constants arising from fluid properties and their representation in the C code and Table 2 shows the constants present in the manufactured solutions.

For air, the Sutherland's law coefficients are: $B_{\mu} = 110.4 \, [K]$, $T_{ref} = 273.15 \, [K]$, $\mu_{ref} = 1.716 \times 10^{-5} \, [kg/ms]$ and, therefore, $A_{\mu} = 1.458 \times 10^{-6} \, [kg/ms\sqrt{K}]$. Additionally, $\gamma = 1.4$, $R = 287 \, [J/kgK]$, Pr = 0.7.

5 C Files

The C files for both source terms and gradients of the manufactured solutions are:

- NavierStokes_1d_transient_Sutherland_e_code.C
- NavierStokes_1d_transient_Sutherland_rho_code.C
- NavierStokes_1d_transient_Sutherland_u_code.C
- NavierStokes_1d_transient_Sutherland_manuf_solutions_grad_code.C

```
#include <math.h>
double SourceQ_rho (double x, double t)
{
```

Listing 1: File NavierStokes_1d_transient_Sutherland_rho_code.C

```
#include <math.h>
double SourceQ_u (double x, double t)
    double RHO;
    double P;
    double U;
    double T;
    double MU;
    double DMu_Dx;
    double Q_u;
    double Q_u_convection;
    double Q_u_pressure;
    double Q_u_viscous;
    double Q_u_time;
    RHO = rho_0 + rho_x * sin(a_rhox * PI * x / L) + rho_t * sin(a_rhot * PI * t / Lt);
    U = u_0 + u_x * \sin(a_u x * PI * x / L) + u_t * \cos(a_u t * PI * t / Lt);
    P = p_0 + p_x * cos(a_px * PI * x / L) + p_t * cos(a_pt * PI * t / Lt);
    T = P / RHO / R;
    MU = A_mu * pow(T, 0.3e1 / 0.2e1) / (T + B_mu);
     \texttt{DMu\_Dx} = \texttt{a\_rhox} * \texttt{PI} * \texttt{rho\_x} * \texttt{MU} * \texttt{MU} * \texttt{cos(a\_rhox} * \texttt{PI} * \texttt{x} / \texttt{L}) / \texttt{A\_mu} / \texttt{L} / \texttt{RHO} / \texttt{sqrt(T)} - \texttt{RHO} / \texttt{Sqrt(T)} ) 
            0.3e1 / 0.2e1 * a_rhox * PI * rho_x * MU * cos(a_rhox * PI * x / L) / L / RHO + a_px * PI *
            p_x * MU * MU * sin(a_px * PI * x / L) / A_mu / L / R / RHO * pow(T, -0.3e1 / 0.2e1) - 0.3e1
              / 0.2e1 * a_px * PI * p_x * MU * sin(a_px * PI * x / L) / L / R / RHO / T;
    Q_u_convection = a_rhox * PI * rho_x * V * V * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * v * 
            * u_x * RHO * U * cos(a_ux * PI * x / L) / L;
    Q_u_pressure = -a_px * PI * p_x * sin(a_px * PI * x / L) / L;
    Q_u_viscous = 0.4e1 / 0.3e1 * MU * a_ux * a_ux * PI * PI * u_x * sin(a_ux * PI * x / L) * pow(L,
              -0.2e1) - 0.4e1 / 0.3e1 * DMu_Dx * a_ux * PI * u_x * cos(a_ux * PI * x / L) / L;
    Q_u_time = a_rhot * PI * rho_t * U * cos(a_rhot * PI * t / Lt / Lt - a_ut * PI * u_t * RHO *
           sin(a_ut * PI * t / Lt) / Lt;
    Q_u = Q_u_convection + Q_u_pressure + Q_u_viscous + Q_u_time;
    return(Q_u);
}
```

Listing 2: File NavierStokes_1d_transient_Sutherland_u_code.C

```
#include <math.h>

double SourceQ_e (double x, double t)
{
   double RHO;
   double P;
   double U;
   double T;
   double MU;
   double DMu_Dx;
   double kappa;
   double Q_e;
```

```
double Q_e_convection;
double Q_e_work_pressure;
double Q_e_work_viscous;
double Q_e_conduction;
double Q_e_time;
RHO = rho_0 + rho_x * sin(a_rhox * PI * x / L) + rho_t * sin(a_rhot * PI * t / Lt);
U = u_0 + u_x * \sin(a_u x * PI * x / L) + u_t * \cos(a_u t * PI * t / Lt);
P = p_0 + p_x * cos(a_px * PI * x / L) + p_t * cos(a_pt * PI * t / Lt);
T = P / RHO / R;
MU = A_mu * pow(T, 0.3e1 / 0.2e1) / (T + B_mu);
 DMu_Dx = a_rhox * PI * rho_x * MU * MU * cos(a_rhox * PI * x / L) / A_mu / L / RHO / sqrt(T) - A_mu
           0.3e1 / 0.2e1 * a_rhox * PI * rho_x * MU * cos(a_rhox * PI * x / L) / L / RHO + a_px * PI *
           p_x * MU * MU * sin(a_px * PI * x / L) / A_mu / L / R / RHO * pow(T, -0.3e1 / 0.2e1) - 0.3e1
              / 0.2e1 * a_px * PI * p_x * MU * sin(a_px * PI * x / L) / L / R / RHO / T;
Q_{e} convection = a_r hox * PI * rho_x * pow(U, 0.3e1) * cos(a_r hox * PI * x / L) / L / 0.2e1 +
           0.3e1 / 0.2e1 * a_ux * PI * u_x * \overline{\text{RH0}} * U * U * \cos(a_ux * PI * x / L) / L - a_px * PI * p_x
              * U * sin(a_px * PI * x / L) / (Gamma - 0.1e1) / L + a_ux * PI * u_x * P * cos(a_ux * PI *
           x / L) / (Gamma - 0.1e1) / L;
Q_e_work_pressure = -a_px * PI * p_x * U * sin(a_px * PI * x / L) / L + a_ux * PI * u_x * P *
           cos(a_ux * PI * x / L) / L;
Q_{e} conduction = -0.2e1 * kappa * a_rhox * a_rhox * PI * PI * rho_x * rho_x * P * pow(cos(a_rhox
              * PI * x / L), 0.2e1) * pow(L, -0.2e1) / R * pow(RHO, -0.3e1) - 0.2e1 * kappa * a_rhox *
           a_px * PI * PI * rho_x * p_x * cos(a_rhox * PI * x / L) * sin(a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * pow(L, a_px * PI * x / L) * p
           -0.2e1) / R * pow(RHO, -0.2e1) - kappa * a_rhox * a_rhox * PI * rho_x * P * sin(a_rhox
           * PI * x / L) * pow(L, -0.2e1) / R * pow(RHO, -0.2e1) + kappa * a_px * a_px * PI * PI * p_x
           * cos(a_px * PI * x / L) * pow(L, -0.2e1) / R / RHO;
Q_e_work_viscous = -0.4e1 / 0.3e1 * a_ux * a_ux * MU * PI * PI * u_x * u_x * pow(cos(a_ux * PI *
             x / L), 0.2e1) * pow(L, -0.2e1) + 0.4e1 / 0.3e1 * a_ux * a_ux * MU * PI * PI * u_x * U *
           sin(a_ux * PI * x / L) * pow(L, -0.2e1);
Q_e_time = -a_ut * u_t * PI * RHO * U * \sin(a_ut * PI * t / Lt) / Lt + a_rhot * rho_t * PI * U *
              \label{eq:cos}       U \ * \ cos(a\_rhot \ * \ PI \ * \ t \ / \ Lt \ / \ 0.2e1 \ - \ a\_pt \ * \ p\_t \ * \ PI \ * \ sin(a\_pt \ * \ PI \ * \ t \ / \ Lt) \ / \ (
           Gamma - 0.1e1) / Lt;
Q_e = Q_e_convection + Q_e_work_pressure + Q_e_work_viscous + Q_e_conduction + Q_e_time;
return(Q_e);
```

Listing 3: File NavierStokes_1d_transient_Sutherland_e_code.C

```
rho_an = rho_0 + rho_x * sin(a_rhox * pi * x / L) + rho_t * sin(a_rhot * pi * t / Lt);
p_an = p_0 + p_x * cos(a_px * pi * x / L) + p_t * cos(a_pt * pi * t / Lt);
u_an = u_0 + u_x * sin(a_ux * pi * x / L) + u_t * cos(a_ut * pi * t / Lt);
grad_rho_an[0] = rho_x * cos(a_rhox * pi * x / L) * a_rhox * pi / L;
grad_rho_an[1] = 0;
grad_p_an[2] = 0;
grad_p_an[0] = -p_x * sin(a_px * pi * x / L) * a_px * pi / L;
grad_p_an[1] = 0;
grad_p_an[2] = 0;
grad_u_an[0] = u_x * cos(a_ux * pi * x / L) * a_ux * pi / L;
grad_u_an[1] = 0;
grad_u_an[2] = 0;
```

Listing 4: File NavierStokes_1d_transient_Sutherland_manuf_solutions_grad_code.C

References

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Roy, C., T. Smith, and C. Ober (2002). Verification of a Compressible CFD Code using the Method of Manufactured Solutions. In 32nd AIAA Fluid Dynamics Conference and Exhibit, Number AIAA 2002-3110.

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Table 1: Relations between fluid properties in the model documentation and in the ${\bf C}$ code.

Variable	Description	Equation	Representation in C
B_{μ}	Sutherland temperature	(9)	B_mu
T_{ref}	reference temperature	(9), (10)	$T_\mathtt{ref}$
μ_{ref}	reference viscosity	(10)	${\tt mu_ref}$
γ	ratio of specific heats	(8)	Gamma
\Pr	Prandtl number	(8)	Pr
R	gas constant	(5)	R
κ	thermal conductivity coefficient	(8)	kappa

Table 2: Constants in the manufactured solutions and their representation in C codes.

Constant	Representation in C	Constant	Representation in C
L	L	p_0	p_0
L_t	Lt	p_x	p_x
$a_{ ho x}$	a_rhox	p_t	pt
$ ho_0$	rho_O	a_{ux}	a_ux
$ ho_x$	${\tt rho_x}$	a_{ut}	a_ut
$ ho_t$	${\tt rho_t}$	u_0	u0
a_{px}	a_px	u_x	$u_{-}x$
$a_{px} \ a_{pt}$	a_pt	u_t	$\mathtt{u}_{-}t$
		π	PI