# P-51 Mustang Flight Simulation Data

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Author: Marek M. Cel

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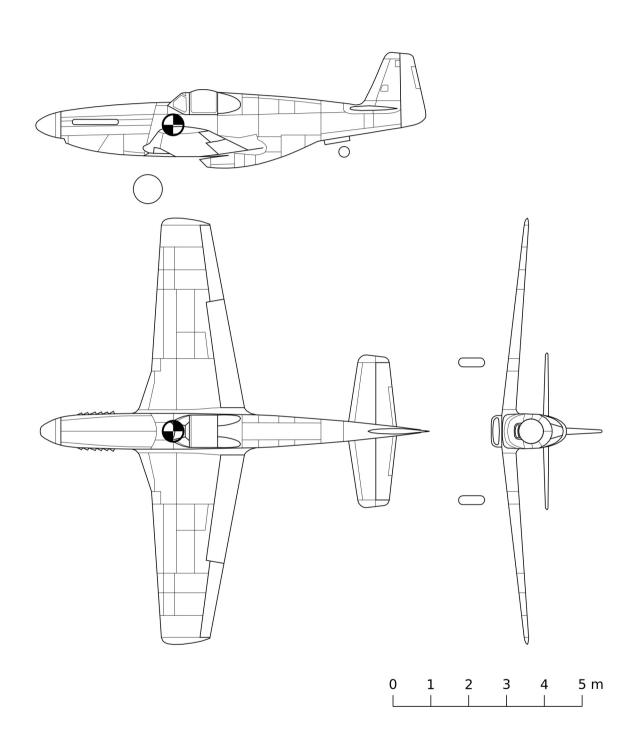
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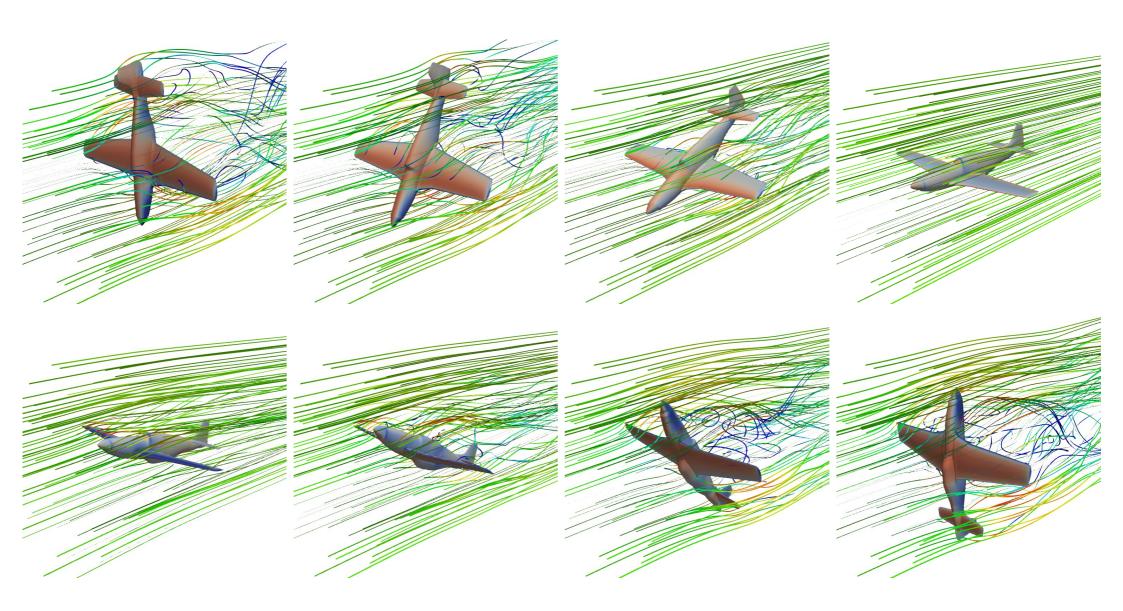
# 1. General Data

Parameter	Value	Reference		
Length	9.83 m	[1]		
Wingspan	11.28 m	[1]		
Height	3.71 m	[1]		
Wing area	22.3 m <sup>2</sup>	[1]		
Mean aerodynamic chord	2.02 m	[1]		
Wing airfoil	NAA-NACA 45-100	[2]		
Wing incidence	+1°	[1]		
Horizontal tail area	3.81 m <sup>2</sup>	[1]		
Horizontal tail incidence	+2°	[1]		
Vertical tail area	1.86 m <sup>2</sup>	[1]		
Ailerons area (total)	1.18 m <sup>2</sup>	[1]		
Ailerons deflection limit	±10, 12 or 15°	[1]		
Ailerons trim tab deflection limit	±10°	[1]		
Elevator area (including tabs)	1.21 m <sup>2</sup>	[1]		
Elevator trim tabs area	0.18 m <sup>2</sup>	[1]		
Elevator deflection limit	up 30°, down 20°	[1]		
Elevator trim tab deflection limit	up 10°, down 25°	[1]		
Rudder area (including tabs)	0.97 m <sup>2</sup>	[1]		
Rudder trim tabs area	$0.07 \text{ m}^2$	[1]		
Rudder deflection limit	±30°	[1]		
Rudder trim tab deflection limit	±10°	[1]		
Flaps area (total)	2.99 m <sup>2</sup>	[1]		
Flaps deflection limit	47°	[1]		
Empty weight	3 311 kg	[3]		
Loaded weight	4 540 kg	[4]		
Internal fuel tanks capacity (wings)	696 l	[1]		
Internal fuel tanks capacity (fuselage)	322 l	[1]		
Engine model	Packard V-1650-7	[1]		
Engine rated power (at 3 000 RPM) - take-off	1 186 kW	[5]		
Engine displacement	27.02 l	[5]		

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Parameter	Value	Reference		
Engine height	1.056 m	[5]		
Engine width	0.762 m	[5]		
Engine length	2.212 m	[5]		
Engine weight	767 kg	[5]		
Specific fuel consumption	301.7 g/(kW·h)	[5]		
Engine gear ratio	0.479	[1], [5]		
Propeller manufacturer	Hamilton Standard	[1]		
Propeller hub model no.	24D50-65	[1]		
Propeller blade model no.	J6523A-24	[1]		
Propeller diameter	3.40 m	[1]		
Minimum propeller pitch	23°	[1]		
Maximum propeller pitch	65°	[1]		
Supercharger low speed gear ratio	5.80	[5]		
Supercharger high speed gear ratio	7.35	[5]		

# 2. Aerodynamic Characteristics



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# 3. Mass Data

Parameter	Value
Center of mass x-coordinate	-0.20 m
Center of mass y-coordinate	0.00 m
Center of mass z-coordinate	0.21 m
Moment of inertia I <sub>x</sub>	11 022.0 kg·m²
Moment of inertia I <sub>y</sub>	15 384.2 kg·m²
Moment of inertia I <sub>z</sub>	24 937.5 kg·m²
Cross product of inertia I <sub>xy</sub>	0.0 kg⋅m²
Cross product of inertia I <sub>xz</sub>	-227.9 kg·m²
Cross product of inertia I <sub>yz</sub>	0.0 kg·m²

Mustang empty aircraft inertia tensor and center of mass coordinates

Structure group	Weight Coordinates [kg] [m]			First moment of mass [kg·m]			Moment of inertia [kg·m²]			Moment of inertia (Body Axis System) [kg·m²]						
		Χ	y	Z	$S_X$	$S_{Y}$	$S_z$	$I_{x,0}$	$I_{y,0}$	$I_{z,0}$	$I_{x}$	$I_y$	$I_z$	$I_{xy}$	$I_{xz}$	$I_{yz}$

Mustang structure groups breakdown

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