

A Table of Requirements

ID	Description	FRETISH	Pattern	Source
1	During an experiment: Two context images taken at different times of a PIXL experiment will be compared to detect unplanned movement (drift) of the rover arm likely to arise from temperature changes.	In Experiment PIXL shall eventually satisfy Take2Picture & ComparePicture	MODAL REACTION	[10]
2	PIXL's hexapod can compensate for X-Y drift if it is found to exceed a pre-defined threshold	In Experiment whenever !inthreshold Rover shall eventually satisfy inthreshold	MODAL DELAYED REACTION	[10]
3	In order to ensure that PIXL's XRF and OFS subsystems are behaving in the expected manner, the instrument's performance is periodically checked by measuring the onboard calibration target, and then comparing the results of those measurements against pre-flight measurements of those standards.	whenever timetocheck=1 PIXL shall at the next timepoint satisfy check	PROMPT REACTION	[10]
4	Once a safe target has been selected, the spacecraft adjusts its trajectory in propulsive powered flight to land at the target.	if safetargetlocated shuttle shall immediately satisfy AdjustToLand	TRIGGERED INSTANT REACTION	[10]
5	Deploy the parachute using navigated position information once safe parachute deployment velocities have been reached	if parachutedistance <= safeparachutedistance EDL shall immediately satisfy DeployParachute	TRIGGERED INSTANT REACTION	[10]
6	LVS begins taking pictures at 4.2 km altitude and matching them up to an onboard map.	Whenever Altitude <= 4.2 LVS shall immediately satisfy TakePictures & Match	INSTANT REACTION	[10]
7	Only once the rover is safely on the martian surface will flight software command the preparation and downlink of EDL Camera images and microphone data.	Upon SafeLanding EDL shall at the next timepoint satisfy Preparation & DownLink	TRIGGERED PROMPT REACTION	[10]
8	Stand-by mode: In IM stand-by mode only components necessary to monitor the system and to survive the external environment shall be active	In STANDBYMODE IM shall always satisfy necessary_components_only	MODAL MAINTAIN SAFE SPACE	[12]

9-A	Safe mode: In IM safe mode all components are activated at limited level (adopted in case of contingency)	In SAFEMODE IM shall always satisfy components_limited_level	MODAL MAIN- TAIN SAFE SPACE	[12]
9-B	The IM shall be isolated in case of contingency.	In SAFEMODE whenever CONTINGENCY IM shall immediately satisfy isolated	MODAL IN- STANT RE- ACTION	[12]
9-C	In case of contingency safe mode is employed	Whenever CONTINGENCY IM shall immediately satisfy SAFEMODE	INSTANT REACTION	[12]
10-A	Check mode all components necessary to check system's health before starting the tests are active	In CHECKMODE IMCOMPONENT shall before StartingTests satisfy necessary_check_components_active	MODAL SCHEDUL- ING	[12]
10-B	If testing is imminent enter check mode	whenever TESTING_IMMINENT IM shall at the next timepoint satisfy Checkmode	PROMPT REACTION	[12]
11-A	Nominal testing mode: all components necessary to perform tests are active	In NOMINALTESTINGMODE IMCOMPONENT shall always satisfy necessary_testing_components_active	MODAL MAIN- TAIN SAFE SPACE	[12]
11-B	data are transmitted to SM to be elaborated, then transmitted to ISS and eventually to Ground Segment	In NOMINALTESTINGMODE whenever SMConnection & ISSConnection & GroundSegment & !files IM shall until files satisfy transmit	MODAL TRANSMIT	[12]
12	Nominal crew mode: all main functionalities are active and access of the crew to perform visual inspections is allowed.	In NOMINALCREWMODE IM shall always satisfy main_functionalities_active & crew_access	MODAL MAIN- TAIN SAFE SPACE	[12]
13-A	During the Launch Phase the only mode of operation in use is STANDBY mode which should be done when the IM is in a stowed configuration	In LAUNCHPHASE whenever stowed IM shall eventually satisfy STANDBYMODE	MODAL DELAYED REACTION	[12]
13-B	During the Separation Phase the only mode of operation in use is STANDBY mode which should both be done when the IM is in a stowed configuration	In SEPARATIONPHASE whenever stowed IM shall eventually satisfy STANDBYMODE	MODAL DELAYED REACTION	[12]
13-C	During the Transfer Phase the only modes of operation in use is STANDBY and SAFE mode which should both be done when the IM is in a stowed configuration	In TRANSFERPHASE whenever stowed IM shall eventually satisfy STANDBYMODE & SAFE	MODAL DELAYED REACTION	[12]

13-D	During the Rendezvous Phase the only modes of operation in use is STANDBY and SAFE mode which should both be done when the IM is in a stowed configuration	In RENDEZVOUSPHASE whenever stowed IM shall eventually satisfy STANDBYMODE & SAFEMODE	MODAL DELAYED REACTION	[12]
13-E	During the Berthing Phase the only modes of operation in use is STANDBY and SAFE mode which should both be done when the IM is in a stowed configuration	In BERTHINGPHASE whenever stowed IM shall eventually satisfy STANDBYMODE & SAFEMODE	MODAL DELAYED REACTION	[12]
13-F	During the Cargo delivery Phase the only modes of operation in use is STANDBY and SAFE mode which should both be done when the IM is in a stowed configuration	In CARGODELIVERYPHASE whenever stowed IM shall eventually satisfy STANDBYMODE & SAFEMODE	MODAL DELAYED REACTION	[12]
13-G	During the Inflatable deploying Phase the only modes of operation in use is CHECK, SAFE, NOMINAL TESTING mode which should be done when the IM is in a stowed or deployed configuration	In INFLATABLEPHASE whenever (stowed deployed) IM shall eventually satisfy SAFEMODE & CHECKMODE & NOMINALTESTINGMODE	MODAL DELAYED REACTION	[12]
13-H	During the On orbit tests and ops Phase the only modes of operation in use is CHECK, SAFE, NOMINAL TESTING and NOMINAL CREW mode which should be done when the IM is in a deployed configuration	In ONORBITSPHASE whenever deployed IM shall eventually satisfy SAFEMODE & CHECKMODE & NOMINALCREWMODE & NOMINALTESTINGMODE	MODAL DELAYED REACTION	[12]
13-I	During the Undocking delivery Phase a mode of operation in use is STANDBY which should both be done when the IM is in a deployed configuration	In UNDOCKINGPHASE whenever deployed IM shall eventually satisfy STANDBY	MODAL DELAYED REACTION	[12]
13-J	During the Undocking delivery Phase a mode of operation in use is SAFE which should both be done when the IM is in a deployed or stowed configuration	In UNDOCKINGPHASE whenever (stowed deployed) IM shall eventually satisfy SAFE	MODAL DELAYED REACTION	[12]
13-K	During the Destructive re-entry a mode of operation in use is STANDBY which should both be done when the IM is in a deployed configuration	In DESTRUCTIVEPHASE whenever deployed IM shall eventually satisfy STANDBY	MODAL DELAYED REACTION	[12]

14-A	The launch phase begins ends at burn out.	Upon LaunchPhase System shall eventually satisfy burnout	PHASES	[12]
14-B	The Separation phase begins with burn out.	Upon burnout System shall at the next timepoint satisfy SeparationPhase	PHASES	[12]
14-C	The Separation phase ends with transfer orbit insertion.	Upon SeparationPhase System shall eventually satisfy orbitinsertion	PHASES	[12]
14-D	Orbit insertion leads to the beginning of the transfer phase.	Upon orbitinsertion System shall at the next timepoint satisfy TransferPhase	PHASES	[12]
14-E	During transfer, the spacecraft moves toward the Cygnus arrival near the ISS	Upon TransferPhase System shall eventually satisfy cygnusarriaval	PHASES	[12]
14-F	Finally, the rendezvous phase covers the approach.	Upon cygnusarriaval System shall at the next timepoint satisfy RendezvousPhase	PHASES	[12]
14-G	Finally, the rendezvous phase covers the approach and capture by the robotic arm	Upon RendezvousPhase System shall eventually satisfy captureropticarm	PHASES	[12]
15	Autonomously release the SEPM when the right jettison attitude is reached	Upon Currentattitude <= Rightattitude ReleaseBepiColombo shall immediately satisfy release_SEPM	TRIGGERED INSTANT REACTION	[11]
16	Autonomously release MMO when the polar orbit is reached	Upon polar_orbit_reached BepiColombo shall immediately satisfy release_MMO	TRIGGERED INSTANT REACTION	[11]
17	Autonomously determine a steering law	whenever operating BepiColombo shall eventually satisfy determine_steering_law	PROMPT REACTION	[11]
18	Use low thrust to achieve capture around Mercury	whenever capturing BepiColombo shall immediately satisfy low_thrust	INSTANT REACTION	[11]
19	Autonomously acquire the escape procedure and use it to leave Mercury if necessary	Upon need_mercury_escape BepiColombo shall immediately satisfy acquire_escape_procedure & escape	TRIGGERED INSTANT REACTION	[11]
20-A	Autonomously detect the presence of high solar irradiation	whenever high_solar BepiColombo shall eventually satisfy detect	DELAYED REACTION	[11]
20-B	In case of presence of high solar irradiation the system will be able to shield the electronics by turning them off	Whenever solar_irradiation >Normal_solar_radiation BepiColombo shall immediately satisfy turn_off_electronics	INSTANT REACTION	[11]

20-C	In case of presence of high solar irradiation the system will be able to shield the electronics	Whenever solar_irradiation > Normal_solar_radiation BepiColombo shall immediately satisfy shield_electronics	INSTANT REACTION	[11]
20-D	Autonomously detect the presence of high solar irradiation and get away if possible, by using chemical propulsion.	Whenever solar_irradiation > Normal_solar_radiation BepiColombo shall immediately satisfy get_away_chemically	INSTANT REACTION	[11]
21	Autonomously maintain the on-board equipment and the spacecraft structure in proper temperature range.	BepiColombo shall always satisfy MaintainEquipment & MaintainTemperature	MAINTAIN SAFE SPACE	[11]
22	The algorithm first selects the vernier jet or the group of primary jets whose acceleration has the largest scalar (dot) product with the desired rotational acceleration vector.	SRC shall at the next timepoint satisfy (SelectFirstJet SelectPrimaryJets) & !(SelectFirstJet & SelectPrimaryJets)	PROMPT REACTION	[3]
23	If second and third jets are required, they are similarly selected on the basis of the second and third largest scalar products.	Whenever SecondJet ThirdJet SRC shall immediately satisfy SelectNeededJet	INSTANT REACTION	[3]
24-A	If three jets satisfying the given thresholds cannot be found, the algorithm considers pairs, or, as a last resort, single jets	Whenever !ThreeJets SRC shall immediately satisfy Considerpairs	INSTANT REACTION	[3]
24-B	If three jets satisfying the given thresholds cannot be found, the algorithm considers pairs, or, as a last resort, single jets	Whenever !TwoJets SRC shall immediately satisfy Considersingle	INSTANT REACTION	[3]
25	During the final phase of Shuttle flight, the orbiter must enter a “heading alignment cylinder”	In FinalPhase Orbiter shall eventually satisfy HeadingAlignmentsCylinder	MODAL DELAYED REACTION	[3]
26	if three Shuttle main engines fail sequentially or simultaneously begin calculating/commanding safe abort manoeuvres.	Upon ThreeEngineFailure ThreeE_0 shall immediately satisfy CalculatePlan & SafeManoeuvres	TRIGGERED INSTANT REACTION	[3]
27	If it is required to know the state of the spacecraft, even during the section of the orbit without a communication link with ground segment, store telemetry data.	If StateRequired & !CommunicationLink CubeSat shall immediately satisfy StoreData	TRIGGERED INSTANT REACTION	[8]

28-A	in charge of providing the ground segment with telemetry data about the state and health of the spacecraft, therefore this service shall be able to automatically collect telemetry data.	Cubesat shall always satisfy CollectData	MAINTAIN SAFE SPACE	[8]
28-B	in charge of providing the ground segment with telemetry data about the state and health of the spacecraft, therefore this service shall be able to automatically store telemetry data.	Cubesat shall always satisfy StoreData	MAINTAIN SAFE SPACE	[8]
28-C	in charge of providing the ground segment with telemetry data about the state and health of the spacecraft, therefore this service shall be able to automatically transmit telemetry data.	Whenever groundsegmentconnection & !telemetrydata Cubesat shall until telemetrydata satisfy Transmit	TRANSMIT	[8]
29	At least one side shall be the pilot flying side.	FGS shall always satisfy PilotFlying <= 1	SEMI-AUTONOMOUS	[4]
30	At most one side shall be the pilot flying side.	FGS shall always satisfy PilotFlying >= 1	SEMI-AUTONOMOUS	[4]
31	Pressing the Transfer Switch shall always change the pilot flying side.	Upon TransferSwitch FGS shall immediately satisfy SwitchSides	TRIGGERED INSTANT REACTION	[4]
32	The system shall start with the Primary Side as the pilot flying side.	Upon Startup FGS shall at the next timepoint satisfy PrimarySide	TRIGGERED INSTANT REACTION	[4]
33	The system shall not change the pilot flying side unless the Transfer Switch is pressed.	FGS shall until switch satisfy !SwitchSides + FGS shall eventually satisfy switch	WAIT	[4]
34	Exceeding sensor limits shall latch an autopilot pullup when the pilot is not in control (not standby) and the system is supported without failures (not apfail).	Whenever Limits & !Standby & supported & !apfail FSM shall immediately satisfy Pullup	INSTANT REACTION	[7]
35	While flying, remain separated from an intruder aircraft by at least 250 ft horizontally or 50 ft vertically	In FlightMode AirCraft shall always satisfy (horizontalIntruderDistance >250 verticalIntruderDistance >50)	MODAL MAINTAIN SAFE SPACE	[1]

36	The probability that the aircraft leaves the taxiway, i.e., $\text{cte} > 8$ meters, shall be extremely low	Aircraft shall with probability ≤ 0.001 eventually satisfy $\text{absReal(cte)} > 8$	PROBABILISTIC MAINTAIN SAFE SPACE	[1]
37	The probability that the aircraft turns more than a prescribed degree ($\text{he} \leq 35^\circ$) shall be extremely low	Aircraft shall with probability ≤ 0.002 eventually satisfy $\text{absReal(he)} \leq 35$	PROBABILISTIC MAINTAIN SAFE SPACE	[1]
38	We also require that the rear propeller be always used, except in HC mode	If not in HMode LPC shall always satisfy RearPropeller	MODAL MAINTAIN SAFE SPACE	[9]
39	If the vehicle is slowing down from the wing-borne mode (WB), the transition to semi-wing-borne (SWB) kicks in at an indicated air-speed of 90 knots ($\text{kias} \leq 90.0$)	In Wbmode whenever airspeed ≤ 90 LPC shall eventually satisfy SWBMode	MODAL DELAYED REACTION	[7]
40	whereas if the vehicle is speeding up from a SWB mode, the transition to WB mode occurs at $\text{kias} > 100.0$ knots	In SWBmode whenever airspeed > 100 LPC shall eventually satisfy WBMode	MODAL DELAYED REACTION	[7]
41	The vehicle remains in the thrust-borne mode (TB) as long as $\text{kgs} \leq 20.0$ knots and Hover Control (HC) mode is selected.	In HMode whenever TBMode & $\text{Kgs} \leq 20$ LPC shall always satisfy TBmode	MAINTAIN MODE IN HIERARCHY	[9]
42-A	during takeoff and landing, the aircraft motion is controlled by the lifting rotors only	In TakeoffMode LPC shall always satisfy LiftingRotors & !FlightSurfaces	MODAL MAINTAIN SAFE SPACE	[9]
42-B	during takeoff and landing, the aircraft motion is controlled by the lifting rotors only	In LandingMode LPC shall always satisfy LiftingRotors & !FlightSurfaces	MODAL MAINTAIN SAFE SPACE	[9]
43	On the other hand, during the higher speeds of the en-route phase, the wings provide lift, the rear propeller provides thrust, and the lifting rotors are inactive (wing-borne mode, WB)	In EnRoute LPC shall always satisfy !LiftingRotors & ThrustRearPropeller & WingsLift	MODAL MAINTAIN SAFE SPACE	[9]
44	In a SLM survey, crew takes measurements at locations described in procedures, attempting to take the measurement as close to the described point as possible	Astrobee shall eventually satisfy SoundLocation & SLMSurvey	VISIT WITH REACTION	[2]

45	This type of data could be supplemented with denser, though shorter duration, measurements from a mobile platform. The Radiation Environment Monitor (REM) hardware developed at the University of Houston and NASA Johnson Space Center is an example of the sort of small, light-weight sensor that Astrobees could carry to create higher resolution maps of the ISS environment.	Astrobees shall eventually satisfy RadiationLocation & RadiationSurvey	VISIT WITH REACTION	[2]
46	The SPHERES satellites, however, triangulate their position using infrared/ultrasonic beacons, preventing them from navigating outside the two-meter cube defined by the fixed beacon locations.	Whenever moving SPHERES shall immediately satisfy $x < 2$ & $y < 2$ & $z < 2$	STAY-IN-PERIMETER	[6]
47	Like SPHERES, Int-Ball cannot operate without a direct line-of-sight to its markers.	whenever Operating IntBall shall immediately satisfy LOS1 & LOS2	STAY-IN-PERIMETER	[6]
48	The PerchCam is identical to the HazCam and it turns on to detect ISS handrails when Astrobees perches autonomously	Whenever Perched Astrobees shall immediately satisfy PerchCam	INSTANT REACTION	[2]
49	the top-facing SpeedCam sensor package provides an independent over-speed cutoff function, estimating velocity using its own optical flow, infrared ranging, and IMU sensors	Whenever Moving Astrobees shall always satisfy cutoff > currentspeed	MAINTAIN SAFE SPACE	[2]
50	After a sortie, Astrobees transfers large files through a hard-wired Ethernet connection with its dock	whenever ISSConnection & Ethernet & !LargeFile Astrobees shall until LargeFile satisfy Transfer	TRANSMIT	[2]
51	Once Astrobees grasps a handrail, it powers down its propulsion system.	Upon Perched Astrobees shall at the next timepoint satisfy !PropulsionSystem	TRIGGERED PROMPT REACTION	[2]
52	Initially, Astrobees will use these components primarily to help crew understand its state and intentions (for example, by providing turn signals)	whenever Turning Astrobees shall at the next timepoint satisfy Indicate	PROMPT REACTION	[2]

53	When docking, Astrobees autonomously approaches its berth using visual servoing relative to fiducials mounted to the dock	In DockingMode Astrobees shall eventually satisfy approachBerth	MODAL REACTION	[2]
54	When mating is complete, permanent magnets on the berth attract striker plates on the robot, providing a passive retention force	Upon MatingComplete DS shall eventually satisfy StrikeMagnets	TRIGGERED DELAYED REACTION	[2]
55	To enable undocking, linear actuators within the berths pull the magnets away from the striker plates, allowing the propulsion system to easily overcome the reduced magnetic force	In UndockingMode DS shall at the next timepoint satisfy LinearActuators	MODAL PROMPT REACTION	[2]
56	If multiple Astrobees are active, the Control Station displays the positions of all of the Astrobees so that the operators are aware of the other activities and can avoid collisions.	Whenever numberOfAstrobees > 1 CS shall immediately satisfy DisplayALL	INSTANT REACTION	[2]
57	Operators use the Plan Editor tab in the Control Station to construct and validate sequences of commands for Astrobees (“fplans”), that include waypoints and actions to perform at the waypoints.	whenever CommandReceived Astrobees shall eventually satisfy PerformCommand	DELAYED REACTION	[2]
58-A	Astrobees can lose signal, when signal is lost Astrobees enter LOS-Mode	whenever LostSignal Astrobees shall immediately satisfy LOSMode	INSTANT REACTION	[2]
58-B	During loss-of-signal (LOS) with the ground, Astrobees continue to hold its position while recording and storing video on its internal file system.	In LOSMode Astrobees shall always satisfy Hold & WorkInternally	MODAL MAINTAIN SAFE SPACE	[2]
58-C	Once ground signal has been reacquired	Whenever !ISSConnection & !Groundsignal Astrobees shall until ISSConnection & Groundsignal satisfy reconnect	RECONNECT	[2]
58-D	Once ground signal has been reacquired, Astrobees resume downlinking the live video stream to the Control Station.	Whenever ISSConnection & Groundsignal & !Stream Astrobees shall until Stream satisfy downlink	TRANSMIT	[2]
59-A	Astrobees are programmed to stop when they detect an obstacle	Upon ObstacleDetected Astrobees shall immediately satisfy Stop	TRIGGERED INSTANT REACTION	[2]

59-B	we are considering using Astrobees lights and/or speaker to signal when it enters a hatchway	Whenever EntersHatchway Astrobees shall immediately satisfy EntranceAlarm	INSTANT REACTION	[2]
60-A	White “Vid” LEDs indicate that cameras are in use	In VideoRecordingMode Astrobees shall always satisfy VidLED	MODAL MAINTAIN SAFE SPACE	[2]
60-B	A blue “Aud” light tells the crew that the microphone is on	In AudioRecordingMode Astrobees shall always satisfy BlueAudLED	MODAL MAINTAIN SAFE SPACE	[2]
60-C	“Live” LEDs indicate that cameras are streaming	In StreamingMode Astrobees shall always satisfy LiveLED	MODAL MAINTAIN SAFE SPACE	[2]
61-A	The Control Station warns operators when they create plans that translate through a KOZ.	Whenever KOZPlan ControlStation shall at the next timepoint satisfy Warn	PROMPT REACTION	[2]
61-B	The Control Station prevents operators from sending plans that translate through a KOZ to Astrobees until the violating segments are modified	ControlStation shall until !KOZPlan satisfy !SendPlan + ControlStation shall eventually satisfy !KOZPlan	WAIT	[2]
62	As a final safeguard, Astrobees itself has an internal list of KOZs that it checks before moving	Whenever moving Astrobees shall immediately satisfy !KOZ1 & !KOZ2	KEEP-OUT-ZONE	[2]
63-A	The robot can periodically update multi-sensor 3D maps of the vehicle. air quality tracking can all help flight controllers understand system status	Whenever TimeForAir Astrobees shall eventually satisfy AirSurvey	DELAYED REACTION	[2]
63-B	The robot can periodically update multi-sensor 3D maps of the vehicle. RFID quality tracking can all help flight controllers understand system status	Whenever TimeForRFID Astrobees shall eventually satisfy RFIDSurvey	DELAYED REACTION	[2]
63-C	The robot can periodically update multi-sensor 3D maps of the vehicle. Visual Imaging tracking can all help flight controllers understand system status	Whenever TimeForVisualImaging Astrobees shall eventually satisfy VisualImagingSurvey	DELAYED REACTION	[2]

63-D	The robot can periodically update multi-sensor 3D maps of the vehicle. Thermal imaging tracking can all help flight controllers understand system status	Whenever TimeForThermalImaging Astrobee shall eventually satisfy ThermalImagingSurvey	DELAYED REACTION	[2]
64	Automated change detection and trending. Once a baseline sensor map is available, changes at the next update can indicate developing problems at an early stage	Whenever SurveyDone Astrobee shall eventually satisfy CompareMaps	DELAYED REACTION	[2]
65	Localizing problems. For example, if a leak produces a whistling sound, acoustic or ultrasonic sensors on-board the robot can be used to pinpoint its location.	Whenever AnomalyDetected Astrobee shall eventually satisfy PinpointProblem	DELAYED REACTION	[2]
66	When flight controllers have a question about something, they can use the robot to get an updated view, filling a role currently played by crew on ISS	Whenever SpotCheck Astrobee shall eventually satisfy UpdateMap	DELAYED REACTION	[2]
67	The first is the observing and planning phase for acquiring motion information of the target satellite and planning when and where the robot will grasp the target satellite	Upon FirstPhase ServicingSatellite shall eventually satisfy AcquireMotionInformation & Planning	PHASES	[5]
68	The second phase is to control the robot to move toward the planned grasping location to make the robot ready for the capturing of the target.	Upon AcquireMotionInformation & Planning ServicingSatellite shall at the next timepoint satisfy SecondPhase + Upon SecondPhase ServicingSatellite shall eventually satisfy MoveToPosition	PHASES	[5]
69	The third phase is the capture (physical interception) phase in which the manipulator physically captures the target satellite	Upon MoveToPosition ServicingSatellite shall at the next timepoint satisfy ThirdPhase + Upon ThirdPhase RobotManipulator shall eventually satisfy PhysicalCapture	PHASES	[5]

70	The fourth phase is the post-capture phase in which captured target satellite is stabilized along with the servicing system	<p>Upon PhysicalCapture ServicingSatellite shall at the next timepoint satisfy FourthPhase + Upon FourthPhase ServicingSatellite shall eventually satisfy Stabilization</p>	PHASES	[5]
71	The maximum rotation speed is restricted within 23,100 rpm to ensure the crew's safety	IntBall2 shall always satisfy RPM <= 23100	MAINTAIN SAFE SPACE	[6]
72	Once the variances of the derivatives of acceleration and angular velocity from the IMU exceed pre-defined upper thresholds, the status shifts to collision mode.	<p>If (VelocityVariancesc >UpperVelocityThreshold) & (AccelerationVariances >UpperAccelerationThreshold) IntBall2 shall immediately satisfy CollisionMode</p>	TRIGGERED INSTANT REACTION	[6]
73	If the variances immediately decrease below lower thresholds, the impact cause is presumed to be an impulsive external force, and the Int-Ball2 tries to maintain its current pose	<p>In CollisionMode if (VelocityVariances <LowerVelocityThreshold) & (AccelerationVariances <LowerAccelerationThreshold) IntBall2 shall immediately satisfy MaintainCurrentPose</p>	MODAL TRIGGERED INSTANT REACTION	[6]
74	Otherwise, it is assumed to be held by the astronaut's hands, and maneuver control is turned off	<p>In CollisonMode if !(VelocityVariances <LowerVelocityThreshold) !(AccelerationVariances <LowerAccelerationThreshold) IntBall2 shall immediately satisfy ManeuverControl=0 & AstronautControl</p>	MODAL TRIGGERED INSTANT REACTION	[6]
75	After the astronaut releases the robot, the control for maintaining the pose at the released point is restarted if the variance falls below the lower threshold	<p>If AstronautControl=0 & VarianceThreshold <VarianceThreshold IntBall2 shall immediately satisfy MaintainCurrentPose</p>	TRIGGERED INSTANT REACTION	[6]

76	Additionally, when the navigation camera is blocked by crew interference or positioned too close to a wall so that the feature points for vSLAM cannot be detected, the navigation subsystem shifts to inertial navigation that uses the IMU without relying on the vSLAM output.	Whenever vSLAMUnavailable IntBall2 shall at the next timepoint satisfy NavigateWithIMU & NavigatevSLAM=0	PROMPT REACTION	[6]
77-A	However, if the vSLAM output remains unavailable for an extended period, the robot rotates in place until the feature points detected in the current view align with those in the stored map	whenever vSLAMUnavailable IntBall2 shall until FeaturePointDetected satisfy RotateProtocol + whenever vSLAMUnavailable IntBall2 shall eventually satisfy FeaturePointDetected	CONDITIONAL [6] WAIT	
77-B	if the vSLAM output remains unavailable for an extended period	Whenever vSLAMOutput=0 & TimePassed <= ExtendedPeriod IntBall2 shall at the next timepoint satisfy vSLAMUnavailable	PROMPT REACTION	[6]
78	Furthermore, when the Int-Ball2 automatically detects that the remaining battery power is low, it returns to the DS for recharging	Whenever IntBall2Power <= SafeBattery IntBall2 shall at the next timepoint satisfy RechargeMode	PROMPT REACTION	[6]

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